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Numerical Simulations: PIC models

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- PIC algorithms
- Electrostatic thrusters
- Electromagnetic thrusters
- Acceleration of the PIC schemes
- Conclusions and perspectives













Kinetic vs fluid approaches







Particle-In-Cell (PIC) model

 Sampling of f_s with a finite number p of particles (superparticules)

$$f_s(\vec{r}, \vec{v}, t) = \sum_p f_{s,p}(\vec{r}, \vec{v}, t)$$

- Time evolution of the particles defined by position and velocity in an electric and magnetic fields generated and modified by themselves (with the potential action of an external B field), including collisions
- Introduction of a time step, a space grid mesh and a finite number of particles (and constrains associated)
- Self-induced B field negligible

Electrostatic PIC model – ES-PIC









J. P. Verboncoeur, Plasma Phys. Control. Fusion 47, A231 (2005)





Hall thruster







Hall thruster, instabilities?

Analytical resolution – no gradients

- Electrons are magnetized, Maxwellian and collisionless
- Cold ions, not magnetized and collisionless ٠
- Coupling with Poisson's equation ٠
- Solution $\phi = \phi_0 \exp[i(\vec{k}.\vec{r} \Omega t)]$ •
- Linearize the system of equations •
- $\Omega(k)$: dispersion relation ٠
- Instabilities appear if $\gamma < 0$ ٠

Dispersion relation, electrostatic waves

real imaginary (growth rate)
with
$$\Omega = \omega + i\gamma$$
: frequency; \vec{k} : wave number

$$g(\Omega, X, Y): \text{ Gordeev function}$$

$$1 + k^2 \lambda_{De}^2 + g\left(\frac{\Omega - k_y V_E}{\omega_{ce}}, \left(k_x^2 + k_y^2\right)\rho^2, k_z^2\rho^2\right) - \frac{k^2 \lambda_{De}^2 \omega_{pi}^2}{(\Omega - k_x V_{ib})^2} = 0$$

$$E = \sqrt{\epsilon_x k_z T}$$

$$B = \frac{k_z V_E}{(\Omega - k_x V_{ib})^2} = 0$$

$$S. P. \text{ Gary and J. J. Sanderson, J. Plasma Physics 4, 739 (1)}$$

$$V_E = \frac{E}{B} \qquad \lambda_{De} = \sqrt{\frac{\varepsilon_0 k_B T_e}{e^2 n_e}} \qquad \omega_{ce} = \frac{e}{n}$$

 $ho = rac{\mathbf{v}_{the}}{\omega_{ce}}$

$$\mathbf{v}_{the} = \sqrt{\frac{8k_BT_e}{\pi m_e}}$$



970)

- D. W. Forslund et al., Phys. Rev. Lett. 25, 1266 (1970)
- M. Lampe et al., Phys. Rev. Lett. 22, 1221 (1971)
- A. Ducrocq et al., Phys. Plasmas 13, 102111 (2013)
- J. Cavalier et al., Phys. Plasmas 20, 082107 (2013)





Transition to MIA instability?

Transition to the modified acoustic instability

- If $\lambda_{De} < \rho$ and $V_E < V_{the}$
- Transition to MIA
- Discrete modes

$$\omega \approx k_x V_{ib} + \frac{kc_s}{\left(1 + k^2 \lambda_{De}^2\right)^{1/2}}; \gamma \approx \sqrt{\frac{\pi m_e}{8m_i}} \frac{k_y V_d}{\left(1 + k^2 \lambda_{De}^2\right)^{3/2}}$$

M. Lampe et al., Phys. Rev. Lett. 22, 1221 (1971)

MIA Characteristics

Maximum growth rate

$$\gamma_{max} pprox a \omega_{pi} rac{V_E}{V_{the}}$$
 , $a \sim 0.2$

- Wave vector at maximum growth rate
- Angular frequency at maximum growth rate
- Amplitude of field oscillations
- Assuming saturation due to jon-wave trapping

$$|\delta\phi| \approx \frac{1}{2} \frac{M}{e} \left(\frac{\omega_{R,max}}{k_{y,max}} \right)^2$$

$$k_{y,max} \approx \frac{1}{\sqrt{2}\lambda_{De}}, \lambda_{y,max} \approx 2\pi\sqrt{2}\lambda_{De}$$

 $\omega_{R,max} \approx \frac{\omega_{pi}}{\sqrt{3}}$
 $|\delta E| \approx \frac{1}{3\sqrt{2}}\frac{T_e}{\lambda_{De}}$

T. Lafleur *et al.*, Phys. Plasmas 23, 053503 (2016)

Wavelength

 $\lambda_{y,MIA} \approx 2\pi \sqrt{2} \lambda_{De}$ E = 200 V/cm, B = 200 G, n_e ~ 10¹⁷ m⁻³, T_e ~ 50 eV, $\lambda_{y,MIA} \sim 1 - 2$ mm





Hall thruster: E × B configuration



Hall thruster characteristics

- Electrons are magnetized
- lons are not magnetized
- Closed electron drift in the azimuthal direction
- Instabilities due to different velocity between electrons and ions in the azimuthal direction





2D ES-PIC simplified model – (x,y) plane (1/2)



- Explicit PIC method with OpenMP/MPI parallelization techniques
- Rectangular simulation domain (x,y) directions
- B ⊥ to simulation plane (with fixed profile)
- Periodic boundary conditions along y (angular sector, w = 1 cm)
- d= 2.5 cm, cathode @ 2.4 cm
- Applied voltage 200 V

J. P. Boeuf and L. Garrigues, Phys. Plasmas 25, 061204 (2018)





2D PIC simplified model – (x,y) plane (2/2)





Given source of injection of charged particles

$$J_M = e \int_0^d S(x) dx$$

- Changing J_M means changing plasma density
- No collisions



1D plasma properties – aver. along y and t



$J_{M} = 400 \text{ A/m}^{2}$

Axial electric field

- Maximum ~ 4 × 10⁴ V/m
- Consistent profile with LIF measurements

S. Mazouffre et al., Plasma Sources Sci. Technol. 22, 013001 (2013)

Plasma properties

- Density ~ 3 × 10¹⁷ m⁻³
- Electron temperature ~ 50 eV consistent with Thomson scattering

T. Dubois et al., IEPC, Boston, MA, 2022





EDI instability

 $J_{M} = 400 \text{ A/m}^{2}$





 Electron transport ensured by electron-wave interactions

J. P. Boeuf and L. Garrigues, Phys. Plasmas 25, 061204 (2018)





Saturation mechanism – ion wave trapping



Experimental validation from LIF measurements?





Parametric study – plasma densities



J. P. Boeuf and L. Garrigues, Phys. Plasmas 25, 061204 (2018)





How the instability affects the transport (1/2)?

T. Lafleur et al., Plasma Sources Sci. Technol. 27, 015003 (2018)

Fluid momentum equation

"e-i friction"

 $\frac{d}{dt}(m_e n_e \mathbf{v}_e) + \nabla (m_e n_e \mathbf{v}_e \mathbf{v}_e) = e n_e (\mathbf{E} + \mathbf{v}_e \times \mathbf{B}) - \nabla (\mathbf{\Pi}_e) + \mathbf{R}_{en} + \mathbf{R}_{ei}$

Calculations of moments from PIC simulations

$$n_{e} = \int_{-\infty}^{\infty} f_{e}(\mathbf{w}) d^{3}w \qquad \mathbf{R}_{en} = -m_{e}n_{g} \int_{-\infty}^{\infty} \sigma_{en}(w) w \mathbf{w} f_{e}(\mathbf{w}) d^{3}w
n_{e} \mathbf{v}_{e} = \int_{-\infty}^{\infty} \mathbf{w} f_{e}(\mathbf{w}) d^{3}w \qquad \mathbf{R}_{ei} = e \langle \delta n_{e} \delta \mathbf{E} \rangle = e \langle [\langle n_{e} \rangle_{t} - n_{e}(t)] [\langle \mathbf{E} \rangle_{t} - \mathbf{E}(t)] \rangle_{t}
\Pi_{e} = m_{e} \int_{-\infty}^{\infty} (\mathbf{w} - \mathbf{v}_{e}) (\mathbf{w} - \mathbf{v}_{e}) f_{e}(\mathbf{w}) d^{3}w$$





How the instability affects the transport (1/2)?

Projection along the y – azimuthal direction







Landmark project

https://www.landmark-plasma.com/

LANDMARK



Three first test cases have been defined and are described below. These cases correspond to typical conditions of Hall thrusters. The problem has however been oversimplified in order to address specific issues related to the numerical method **and**/or to physics questions. Other test cases will be proposed in the future

Test Case 1 - 1D azimuthal PIC simulation of the ExB EDI

This test case is aimed at studying the possible development of microinstabilites induced by the large ExB electron drift in the magnetic barrier of a Hall thruster. The Electron Cyclotron Drift Instability (which we call ExB Drift Instability, ExB EDI, in the present context) studied in the 1970's in space plasmas, has been evidenced in axial-azimuthal PIC simulations of Hall thrusters. The proposed benchmark is aimed at studying the ExB EDI under 1D conditions (constant E and B, and description of the ExB, azimuthal direction) and the conditions of the possible transition to an ion acoustic instability. The effects of numerical noise, accuracy of the simulations, and spatial periodicity in the ExB direction must also be evaluated

Test Case 2a - 2D axial-azimuthal PIC simulation of the ExB EDI

Here we study the ExB EDI under slightly more realistic conditions that can naturally take into account the finite transit time of electrons in the acceleration region, and the effects of density and magnetic field gradients. We simplify the problem by assuming a given ionization rate profile. This suppresses the non-linear coupling of the plasma density and atom density leading to the well know low frequency ("breathing") oscillations of Hall thrusters. The proposed benchmark allows to study the formation, convection and saturation of the ExB EDI. The issues are the transition to an ion acoustic instability, the effect of the instability on the anomalous electron cross-field mobility, the possible effect of numerical noise and of the finite period of the model in the azimuthal direction.

Test Case 2b - 2D radial-azimuthal PIC simulation of the ExB EDI

In this case the ExB EDI is studied in the radial and azimuthal directions. The electron and ion transport in the axial direction is described as in Test Case 1 (constant electric field in the axial direction, constant magnetic field in the radial direction).



Test Case 2a results

T. Charoy et al., Plasma Sources Sci. Technol. 28, 105010 (2019)

W. Villafana et al., Plasma Sources Sci. Technol. 30, 075002 (2021)



details



Numerical models – Test Case 2a

	Tbb	LAPLACE	CERFACS	RUB	USask	TAMU	PPPL
Algorithms							
Pusher solver	Explicit	Explicit	Explicit	Implicit	Explicit	Explicit	Explicit
Poisson solver	Hypre	Pardiso	Maphys	FFT Thomas	FFT	Hypre	Hypre
Floating- point precision	Double	Single(pusher) Double (Poisson)	Double	Single(pusher) Double (Poisson)	Double	Double	Double
Code acceleration							
Architecture	CPU	CPU	CPU	GPU	CPU	CPU	CPU
Parallelization	MPI	MPI/OpenMP	MPI	CUDA	MPI	MPI	MPI/OpenMP
Decomposition	Domain	Particle	Domain	Both	Domain	Particle	Particle
Language	Fortran	Fortran	Fortran	C+Cuda C	Fortran	C++	С
Simulation times (days)							
Case 1							
$(N_{ppc,ini} = 150)$	8 (360 CPU)	5 (108 CPU)	7 (360 CPU)	14 (1 GPU)	21 (32 CPU)	15 (300 CPU)	2.5 (224 CPU)
Case 2 $(N_{ppc,ini} = 75)$	5 (360 CPU)	3 (108 CPU)	4 (360 CPU)	9 (1 GPU)	11 (32 CPU)	11 (200 CPU)	2.5 (112 CPU)
Case 3 ($N_{ppc,ini} = 300$)	14 (360 CPU)	6 (180 CPU)	13 (360 CPU)	14 (2 GPU)	20 (64 CPU)	22 (400 CPU)	2.5 (448 CPU)

Table 2: Main codes specificities.

T. Charoy et al., Plasma Sources Sci. Technol 28, 105010 (2019)





3D PIC simplified model (1/2)







3D PIC simplified model (2/2)



- Very long time consuming 1-2 months on 3000 cores (1Mhrs)
- Very close results than standard 3D Cartesian PIC approach





Gridded ion engine (GIE)



- 3D hybrid model electron-fluid and ion-PIC
- Beamlet extraction



J. Perales-Diaz et al., Plasma Sources Sci. Technol 30, 105023 (2021)





Inductively coupled plasma thrusters

Plasma coupling with the antenna - GIE - RIT

$$\left(\Delta - \mu_0 \varepsilon_0 \frac{\partial^2}{\partial t^2}\right) \boldsymbol{E}\left(\boldsymbol{r}, t\right) = \mu_0 \frac{\partial}{\partial t} \boldsymbol{j}(\boldsymbol{r}, t)$$

Harmonic expansion – azimuthal component

 $(\Delta + \mu_0 \varepsilon_0 \omega^2) \tilde{E}_{\theta} = i \omega \mu_0 \tilde{j}_{\theta} = i \omega \mu_0 [\tilde{j}_{\theta,\text{coil}} + \tilde{j}_{\theta,\text{electron}}]$

R. Henrich et al., IEPC (2017)

3D parallelized EM-PIC applied to the μN-RIT operation





grid 2



Figure 3: Electron density in 1/m³ for a µN-RIT 1.0: x-z plane (left), y-z plane (middle) and x-y plane at position 4 mm above grid 1 (right)



Figure 8: Power deposition in W/m³ for a μ N-RIT 1.0: x-z plane (left), y-z plane (middle) and x-y plane at position 4 mm above grid 1 (right)



thickness

(0.4 mm)



μW driven ECR plasma thrusters



- Magnetostatic field + μW to enhance electron heating
- Feedback of the plasma on EM ignore if plasma density < n_c



Coupling waves and plasma, 1D

J. Porto et al., Phys. Plasmas 30, 023506 (2023)





Methods to overcome explicit constrains

- Easy to implement but constrains on
 - Time step (plasma frequency) and grid spacing (Debye length)
 - High computational time: large density and/or large dimensions
 - Coupling with EM waves: FDTD scheme needs with CFL condition with c (light velocity)
- High Performance Computing techniques
- Overcome time step and grid mesh size constrains?
 - "scaling" method, reduced plasma density G. Fubiani et al., New J. Phys. 19, 015002 (2017)
 - Fully implicit method
 - Energy conserving scheme (ECSIM)
- G. Chen et al., J. Comput. Phys. 230, 7018 (2011)
 - G. Lapenta, J. Comput. Phys. 334, 349 (2017)
- Reduce statistical error numerical noise?

Use of a sparse grid approach





Sparse grid methods

- Classical discretization on uniform Cartesian grid meshes: the total number of unknowns is proportional to M^d (M degree of freedom in each direction, d dimensionality of the problem)
- Sparse grid methods: reduction of the complexity of discrete problem by breaking the exponential increase of the number of degrees of freedom with respect to the dimensionality of the problem

Sparse grid techniques

- Define a hierarchy of anisotropic grids with a coarser resolution 0
- Reconstruction of the solution on the initial Cartesian grid using combination techniques 0
- Preserving second order approximation (for d > 1) 0

Examples

- A. Rütgers and M. Griebel, Applied Math. and Comp. 319, 425 (2018) Resolution of Navier-Stokes equation 0
- Quantum mechanics: Schrödinger equation J. Garcke and M. Griebel, JCP 165, 694 (2000) Ο
- Plasma physics: gyrokinetic Vlasov equation Ο
- D. Pflüger et al., Euro-Par 2014: Parallel Processing (2014)
- Financial mathematics: Black-Scholes equation H. J. Bungartz et al., J. Comput. Applied Math. 236, 3741 (2012) Ο





Construction of sparse grid domains



Aplace 📈



10th-12th October, 2023 Noordwijk, The Netherlands

ES-PIC cycle revisited with sparse approach







Error: Sparse vs Std PIC approaches (1/2)

Numerical noise

Reduction with sparse grids thanks to large volume

• Error associated
$$\varepsilon_s = |\log h_n|^{d-1} \left(\frac{1}{Nh_n}\right)^{1/2} \le \left(\frac{1}{Nh_n^d}\right)^{1/2}$$

sparse standard







Component grid

ESAIM: M2AN 56 (2022) 1809–1841 https://doi.org/10.1051/m2an/2022055

SPARSE GRID RECONSTRUCTIONS FOR PARTICLE-IN-CELL METHODS

Fabrice Deluzet^{1,2}, Gwenael Fubiani³, Laurent Garrigues³, Clément Guillet^{1,2,3,*} and Jacek Narski³





Error: Sparse vs Std ES-PIC approaches (2/2)



https://doi.org/10.1051/m2an/2022055

SPARSE GRID RECONSTRUCTIONS FOR PARTICLE-IN-CELL METHODS

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Momentum and total charged are conserved, not energy (as explicit standard ES-PIC)

- Others benefits for sparse grid PIC algorithms
 - Gain in memory footprint with less total number of particles
 - Saving time in the resolution of Poisson's equation



First plasma simulations

Applicability to ES-PIC techniques

- lons at rest
- Periodic boundary conditions
- No collisions

Numerical tests

Linear/non linear
 Landau damping

Diocotron instability



Extension to low temperature plasmas?

3D non linear Landau damping Comparison of density reference solution computed ($128 \times 128 \times 128$) using regular PIC (left), ~ 30 simulations, sparse solution on grid with $N_{\rm pc} = 800$ (center), and regular-PIC solution on grid with $N_{\rm pc} = 800$ (right)



Laurent Garrigues



L. F. Ricketson and A. J. Cerfon, PPCF 59, 024002 (2017)

RF capacitive discharges

- Self-consistent description of the discharge
- Motion of ions
- Time and space evolution of sheaths
- Collisions between electrons/ions and neutrals
- Single-frequency RF capacitive discharges







L. Garrigues et al.,

J. Appl. Phys. 129, 153303 (2021)

HT - 2D benchmark, axial-azimuthal directions







Number of cells in 3D





large speed up expected for 3D simulations





Conclusions

- Applicability of PIC simulations to model plasma thrusters
 - More studies with the electrostatic approximation (Hall Thruster, GIE)
 - Electromagnetic PIC algorithms more difficult to implement
 - Very time consuming
- PIC schemes take advantages of efficient parallelization techniques (OpenMP-MPI hybrid parallelization)

Acceleration of PIC schemes

- Sparse approach seems promising (cusp magnetic field can be included)
- Others approaches exist (implicit version, reduced dimensional order method, etc.)



Open questions/Perspectives

- A full coupling between Maxwell's equations and the plasma is difficult to achieve but a self-consistent calculation of power absorption is reachable
- Heterogeneous architecture of computers (GPU/CPU)
- Plasma-wall interactions and necessary data
- Elementary data cross sections for new propellant gases (iodine)



For more details

To appear in Journal of Applied Physics

https://arxiv.org/abs/2304.05103

Plasma propulsion simulation using particles

Plasma propulsion simulation using particles

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(Dated: 29 September 2023)

This perspective paper deals with an overview of particle-in-cell / Monte Carlo collision models applied to different plasma-propulsion configurations and scenarios, from electrostatic ($E \times B$ and pulsed arc) devices to electromagnetic (RF inductive, helicon, electron cyclotron resonance) thrusters, with an emphasis on plasma plumes and their interaction with the satellite. The most important elements related to the modeling of plasma-wall interaction are also presented. Finally, the paper reports new progress in the particle-in-cell computational methodology, in particular regarding accelerating computational techniques for multi-dimensional simulations and plasma chemistry Monte Carlo modules for molecular and alternative propellants.





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Super calculator CALMIP/Olympe (Toulouse University)







Additional slides





PIC cycle with the electrostatic approach







More about the grid-based error

component grid Grid-based error in 2D Terms along x and y directions cancel out with 0 combination technique ... but not along mixed directions Error associated with ρ Ο $\varepsilon_s = b_1 \frac{\partial^2 \rho}{\partial x^2} h_x^2 + b_2 \frac{\partial^2 \rho}{\partial y^2} h_y^2 + b_{1,2} \frac{\partial^4 \rho}{\partial x^2 \partial y^2} h_x^2 h_y^2 + O(h_x^4, h_y^4)$ h_{v} h_x error along y error along mixed directions error along x 10 n (10¹⁵ m³) Contents lists available at ScienceDirect 3.5 Position y (mm) 3.0 Journal of Computational Physics 2.3 journal homepage: www.elsevier.com/locate/jcp 1.5 0.75 0.0 Efficient parallelization for 3d-3v sparse grid Particle-In-Cell: Shared memory architectures 0 10 15 0 5 20 Fabrice Deluzet^a, Gwenael Fubiani^b, Laurent Garrigues^b, Clément Guillet^{a,b,*}, Position x (mm) Jacek Narski Y. Jiang et al., Phys. Plasmas 27, 113506 (2020)

Oversampled method to reduce the grid-based error in the mixed directions

S. Muralikrishnan et al., J. Comput. Phys. X 11, 100094 (2021)





The offset sparse method – component grids







Simplified 2D modeling of a cusp discharge







Results @ 0.1 mTorr - 400 G





