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#### 1

# Penetration of a spherical vortex into turbulence

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The penetration of a spherical vortex into turbulence is studied theoretically and experimentally. The characteristics of the vortex are first analysed from an integral perspective that reconciles the far-field dipolar flow with the near-field source flow. The influence of entrainment on the vortex drag force is elucidated, extending the Maxworthy (1977) model to account for turbulent entrainment into the vortex movement and vortex penetration into an evolving turbulent field.

The physics are explored numerically using a spherical vortex (initial radius  $R_0$ , speed  $U_{v0}$ , characterised by a Reynolds number  $Re_0 (= 2R_0 U_{v0} / \nu, \nu$  is the kinematic viscosity) of 2000, moving into decaying homogeneous turbulence (rms  $u_0$ , integral scale L). When the turbulence is absent  $(I_t = u_0/U_{v0}) = 0)$ , a wake volume flux leads to a reduction of vortex impulse that causes the vortex to slow down. In the presence of turbulence  $(I_t > 0)$ , the loss of vortical material is enhanced and the vortex speed decreases until it is comparable to the local turbulent intensity and quickly fragments, penetrating a distance that scales as  $I_t^{-1}$ .

An experimental study is performed on a vortex ( $Re_0 \sim 5000 - 20000$ ) propagating into statistically steady, spatially varying turbulent field ( $I_{ve} = 0.02$  to 0.98). The penetration distance is observed to scale with the inverse of the turbulent intensity.

Incorporating the spatially and temporally varying turbulent fields into the integral model gives a good agreement with the predicted trend of the vortex penetration distance with turbulent intensity and insight into its dependence on the structure of the turbulence.

Key words: vortex dynamics, turbulence, entrainment

#### 1. Introduction

Vortex rings, dipoles and line vortices (see Pullin & Saffman 1998) are fundamental 31 components of turbulence (see Synge & Lin 1943; Saffman 1997). As these vortical struc-32 tures are basic elements of complex flows, their interactions with flow inhomogeneities 33 provide a crucial starting point for exploring fundamental flow processes. In this paper, 34 we investigate the interaction between a vortex ring and turbulence as an initial step 35 toward examining critical layers between regions of turbulence. 36

The pivotal advance in understanding the dynamics of turbulent vortex rings came 37 from Maxworthy's (1974) interpretation of Osborne Reynolds' observation that vortices

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lose mass and slow down. Figure 1(a) (see also figure 10, discussed later) depicts a 39 turbulent vortex propagating into a quiescent flow, with the mass loss evident from the 40 detrainment of dved fluid behind the vortex, as is discussed in more detail in §6. This 41 mass loss leads to a decrease in the vortex impulse and a reduction in the vortex speed, 42 which is often incorrectly interpreted as a drag force. This interpretation has a wider 43 significance beyond the dynamics of vortices, such as in problems where entrainment 44 exerts an equivalent force (or impulse decrease) that occurs even in the absence of solid 45 boundaries. Maxworthy (1974) noted that vortex progression in stationary fluids is not 46 universally consistent over time but develops through well-defined sequential steps, each 47 with distinct scaling laws. 48

While most idealised studies have considered a vortex propagating in a quiescent 49 flow, real-world scenarios typically involve some level of ambient turbulence affecting the 50 longevity and persistence of these structures. Many related studies have considered well-51 behaved laminar vortex rings in an ambient fluid at rest, dominated by diffusive dynamics 52 and relatively small entrainment effects (see e.g. recent works such as Tinaikar et al. 53 2018, and references therein). Maxworthy (1977) demonstrated how external processes 54 like buoyancy, stratification and turbulence, dramatically influence a vortex's lifetime. 55 Arnold (1974) conducted the first detailed experimental study of a vortex ring moving 56 into turbulence created in a water channel. The vortices were observed to grow faster 57 in size and slow down more rapidly due to the background turbulence (Arnold 1974; 58 Maxworthy 1977). Although the reduction of the vortex speed and increase in vortex 59 radius was evident, the final stages of the vortex propagation and arrestment were not 60 observed. This forms an essential component to this study. 61

Several relevant studies have examined vortices interacting with turbulence, including 62 experimental and theoretical studies of line vortices created by a lifting surface in a 63 turbulent flow (see van Jaarsveld et al. 2011), and a single line vortex interacting with 64 turbulence (Marshall & Beninati 2005). The latter example demonstrated the critical 65 change in the external turbulence structure caused by the differential rotation imparted 66 by the line vortex. The case of a propagating three-dimensional vortex is quite different 67 from monopolar and line vortices owing to the flow perturbation decaying more rapidly 68 with distance in three-dimensions, and the vorticity maximum (within the vortex) lying 69 close to the bounding streamline. 70

In this paper, we address a key missing element in the vortex dynamics: the interaction 71 between a spherical vortex and turbulence. The dramatic effect of turbulence on a 72 vortex is shown by contrasting figure 1(b) with figure 1(a). The turbulence significantly 73 decreases the vortex speed by enhancing the growth of the vortex size through enhanced 74 entrainment by the external turbulence and loss of vortical fluid in the vortex wake. Part 75 of the challenge of incorporating the effect of turbulence into the description of the vortex 76 is to understand the distinction between enhanced entrainment across the vortex edge and 77 its influence on the overall 'force' experienced by the vortex. An integral perspective gives 78 greater clarity about the different effects on entrainment and detrainment on the vortex 79 properties, helping to reconcile the dipolar nature of the vortex with the monopolar flow 80 created by the loss of impulse. Drawing on reanalysis of Maxworthy (1977), the additional 81 effect of a spatially and temporally varying turbulent field on the vortex is proposed and 82 analysed. 83 Numerical simulations are performed on the idealised interaction of a Hill's spherical 84

Numerical simulations are performed on the idealised interaction of a Hill's spherical
 vortex moving into decaying homogeneous turbulence. An analysis of the dynamics
 requires the development of both Lagrangian and Eulerian measures of impulse and
 momentum, which are intrinsically linked to the integral perspective offered.

An experimental study is developed to examine the vortex penetration into a turbulent



Figure 1. Visualisation of a ring vortex moving steadily into (a,b) a quiescent flow and (c,d) a turbulent field generate by an oscillating grid, both at 4.6 and 17.0 seconds, respectively, after vortex generation. The Reynolds number of the initial vortex, based on velocity and distance between the vortices, is 3200. Experimental parameters correspond to experiment T in Table 1. The image is 50 cm wide.

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flow (see figure 1) with the fundamental difference between the numerical and experimental study being the turbulence distribution. In the experimental study a piston-driven vortex is generated that moves into a grid generated turbulent flow field. The strength of the turbulence is sufficiently high that a finite penetration distance of the vortices is realised, in contrast to the experiments of Arnold *et al.*(2012), where the level of turbulence is weak. The two strands of study are brought together through a comparison with the integral vortex model.

The paper is structured as follows. In continuation of the introduction, to analyse the 96 gross features of the interaction, integral measures are introduced (in  $\S 2$ ) which provides 97 an opportunity to elaborate on Maxworthy's model, particular the role of entrainment 98 in the impulse analysis and extension to incorporate the effect of external turbulence 99 on their dynamics. The computational and experimental methodologies are described in 100 §3 and §4 respectively. The basic building block of a spherical vortex moving through a 101 quiescent region is analysed in §5, focusing not just on the vortical dynamics but also how 102 fluid elements external to the vortex are displaced and distorted. In §6, the numerical and 103 experimental results are analysed. The primary results concern the relationship between 104 penetration distance into turbulence and the turbulent structure, which is discussed using 105 the integral model. Finally, the results are put into a general context in §7. 106

#### <sup>107</sup> 2. Integral measures of the flow processes

#### 2.1. Integral measures

Integral measures of momentum, impulse, circulation, and kinetic energy are useful for analysing the gross dynamics of complex flows, even for processes involving changes in flow topology (Benjamin & Ellis 1966; Eames 2008). A challenge of using these measures is that they may need to be reconciled for the presence of flow boundaries (Theodorsen 1941; Darwin 1953; Eames *et al.* 1994). A key measure is the impulse associated with a region of the flow. The specific impulse of a vortical element within the region is proportional to the weighted integral of the moment of vorticity ( $\omega$ )

$$\boldsymbol{I} = \frac{1}{2}\boldsymbol{x} \times \boldsymbol{\omega}, \tag{2.1}$$

where the vorticity is defined as  $\boldsymbol{\omega} = \boldsymbol{\nabla} \times \boldsymbol{u}$ , where  $\boldsymbol{u}$  is the velocity field and  $\boldsymbol{x}$  is position. The impulse changes at a rate

$$\frac{\mathrm{D}\boldsymbol{I}}{\mathrm{D}t} = \boldsymbol{u} \times \boldsymbol{\omega} + \frac{1}{2}\boldsymbol{\omega} \cdot \boldsymbol{\nabla}(\boldsymbol{x} \times \boldsymbol{u}) + \frac{1}{2}\boldsymbol{x} \times \boldsymbol{\nu}\boldsymbol{\nabla}^{2}\boldsymbol{\omega}, \qquad (2.2)$$

which includes the diffusion of vorticity with  $\nu$ , the kinematic viscosity. Half the contribution to the vortex force ( $\boldsymbol{u} \times \boldsymbol{\omega}$ ) comes from the advection of the vorticity while the other half comes from stretching. The two remaining terms comes from the production of stretching of vorticity by angular momentum and the diffusion of vorticity. The specific impulse associated with free vorticity in a region V is

$$\boldsymbol{I}_{v} = \int_{V} \frac{1}{2} \boldsymbol{x} \times \boldsymbol{\omega} \mathrm{d}V.$$
(2.3)

The rate of change of total impulse in a control volume V, whose boundary S moves with velocity  $u_b$ , can be evaluated by using the result of Saffman (1995, ch.3.2 eqn. 10 and 11) and the Reynolds Transport Theorem,

$$\frac{\mathrm{d}\boldsymbol{I}_{\boldsymbol{v}}}{\mathrm{d}t} = \int_{V} \boldsymbol{u} \times \boldsymbol{\omega} \mathrm{d}V + \int_{S} \frac{1}{2} \left( (\boldsymbol{\omega} \cdot \hat{\boldsymbol{n}}) \boldsymbol{x} \times \boldsymbol{u} - (\boldsymbol{u} \cdot \hat{\boldsymbol{n}}) (\boldsymbol{x} \times \boldsymbol{\omega}) \right) \mathrm{d}S$$

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$$+ \int_{S} (\boldsymbol{u}_{b} \cdot \hat{\boldsymbol{n}}) \Big( \frac{1}{2} \boldsymbol{x} \times \boldsymbol{\omega} \Big) \mathrm{d}S + \int_{V} \frac{1}{2} \boldsymbol{x} \times \nu \nabla^{2} \boldsymbol{\omega} \mathrm{d}V, \qquad (2.4)$$

where  $\hat{n}$  is a unit vector pointing out of the control volume. For a large domain that encloses a compact region of vorticity, the second and third integrals are zero and the change in impulse is then expressed in terms of the total vortex force imposed on the flow. As Saffman discusses, the consequence of the far field flow being dipolar is that the impulse and dipole moment - which are intrinsically linked – are conserved/constant.

Our attention is on the behaviour of a vortex propagating and disintegrating as <sup>129</sup> it penetrates a turbulent flow. To distinguish between the vortex and the external <sup>130</sup> turbulence, we make use of a passive scalar field C and Lagrangian fluid particles to <sup>131</sup> tag the vortex. The passive scalar is set to be C = 1 within the vortex (of volume  $V_0$ ) <sup>132</sup> and zero outside. The Lagrangian perspective is developed by tagging  $N_p$  fluid particles <sup>133</sup> initially within the vortex. The integral measures of A, based on either a Lagrangian or <sup>134</sup> Eulerian approach respectively, are defined as <sup>135</sup>

$$A^{(L)} = \frac{1}{N_p} V_0 \sum_{i=1}^{N_p} A(\boldsymbol{X}_i), \qquad A^{(E)} = \int_V CA(\boldsymbol{x}, t) \mathrm{d}V,$$
(2.5)

respectively and distinguished by the superscript L or E (see Eames *et al.* (2004)). The <sup>136</sup> Eulerian and Lagrangian measures of vortex impulse,  $I_v^{(E)}$  and  $I_v^{(E)}$  respectively, are <sup>137</sup> defined in terms of  $A = \frac{1}{2} \boldsymbol{x} \times \boldsymbol{\omega}$ .

#### 2.2. Integral vortex model 139

In this section, the vortex dynamics are analysed from an integral perspective that 140 requires distinguishing between the vorticity that forms the main body of the vortex 141 and that which is deposited downstream. This distinction requires a separation of scales 142 between the size of the vortex and the distance over which the vortex has travelled. There 143 are several ways to identify the position of the vortex, including weighted measures of 144 the initial constituent material of the vortex and positions of local maxima of vorticity 145 magnitude; for consistency with the numerical calculations and experiments, the vortex 146 centre  $X_v$  is defined by following the initial fluid within the vortex (with A = x in (2.5)); 147 the superscript (L) is dropped. For the numerical simulation, the position is determined 148 by following (Lagrangian) fluid particles in time, while for the experiments, a dye is 149 followed in time. The vortex velocity,  $U_v$ , is defined as  $dX_v/dt$ . 150

For unbounded flows, the dipole moment in the far field is unchanged in time, but this 151 needs to be reconciled with changes that occur in the vicinity of the vortex. Maxworthy 152 (1974) argued that the progressive loss of mass, evident through a visible wake loss, leads 153 to the vortex impulse decreasing in time. The integral analysis is applied over a volume 154 V that has a size larger than the size of the vortex, e.g.  $O(V_v^{1/3})$  (where  $V_v$  is the volume 155 of the vortex), and moving with velocity  $\boldsymbol{u}_b = U_v \hat{\boldsymbol{x}}$  (figure 2(a)). Identifying  $I_v = \boldsymbol{I}_v \cdot \hat{\boldsymbol{x}}$ 156 as the local impulse associated with the vortex, Maxworthy (1974) argued that, at high 157 Reynolds numbers, the third term in (2.4) was dominant, with the rate of change of the 158 vortex impulse being 159

$$\frac{\mathrm{d}I_v}{\mathrm{d}t} \approx -\frac{1}{2} \int_{S_d} U_v \sigma \omega_\phi \mathrm{d}S,\tag{2.6}$$

where  $\sigma$  is the radial distance from the centerline,  $\omega_{\phi}$  is the azimuthal vorticity component and the integration is taken over the downstream surface  $S_d$  (see figure 2(a)). Equation 161



Figure 2. (a) Schematic of the control volume V and surface S, moving with speed  $U_v$  with the vortex, that is used in the analysis. (b) The flow perturbation associated with the vortex moving from rest is characterised by a sink at the start and local source flow (due to the loss of impulse) which reconciles the far field dipolar flow with the near field monopolar flow and this is shown as a schematic.

#### $_{162}$ (2.6) can be written as

$$\frac{\mathrm{d}I_v}{\mathrm{d}t} \approx -Q_w U_v, \qquad Q_w = \int_{S_d} \frac{1}{2} \sigma \omega_\phi \mathrm{d}S, \qquad (2.7\mathrm{a,b})$$

where  $Q_w$  is the volume flux (deficit) behind the vortex. The volume flux generates a monopolar flow in the near-field (figure 2(b)). The combination of the sink flow created by the vortex starting, and the source flow created by drag, yields a far field flow which is dipolar in character. These types of source-sink flows are also created by rigid bodies impulsively set into motion; Hinch provides an extremely clear physical interpretation of the consequence of these impulsively driven flows on the forces experienced by moving bodies in Appendix D of Lovalenti & Brady (1993).

The importance of the remaining terms in (2.4) was not discussed by Maxworthy 170 and needs more analysis to unpick their link to the flow physics. Entrainment is in 171 general mechanistically distinguished as having either an inviscid or a viscous component 172 (da Silva *et al.* (2014)). The (inviscid) enguliment process leads to a net flow across an 173 interface, while the viscous mechanism leads to boundary movement with a generally 174 weaker boundary velocity. Without defining the features of such processes, we ascribe 175 the flow due to entrainment as a volume flux,  $Q_e$ , and the loss of vorticity in the wake 176 being equivalent to a volume flux  $Q_w$ . Due to the combination of the wake flow and 177 entrainment, the flow in the region moving with the vortex  $(R_0 \ll |\boldsymbol{x} - \boldsymbol{X}_v| \ll |\boldsymbol{X}_v|)$  is 178

to leading order

$$\boldsymbol{u} \sim -U_v \hat{\boldsymbol{x}} + \frac{Q_w - Q_e}{4\pi} \frac{\boldsymbol{x} - \boldsymbol{X}_v}{|\boldsymbol{x} - \boldsymbol{X}_v|^3},\tag{2.8}$$

where the vortex position  $X_v$  may be defined in terms of the position of the original constitutent fluid. We can use (2.8) to evaluate (2.4) more precisely than (2.6). The components of the control surfaces that bound the sides of the cylinder  $S_s$ , the upstream surface  $S_u$  and the surface for the downstream wake,  $S_d$ , can be distinguished (see figure 2(a)). The vorticity is zero everywhere on  $S_s$  except on the surface  $S_d$ . Typically,  $\boldsymbol{\omega}$  is perpendicular to the direction of motion so that ( $\boldsymbol{\omega} \cdot \hat{\boldsymbol{n}} = 0$ ). Using

$$\boldsymbol{u} \times \boldsymbol{\omega} = -(\boldsymbol{u} \cdot \boldsymbol{\nabla})\boldsymbol{u} + \boldsymbol{\nabla} (\frac{1}{2}q^2),$$
 (2.9)

where  $q = |\boldsymbol{u}|$ , the rate of change of impulse can be written as

$$\frac{\mathrm{d}\boldsymbol{I}_{v}}{\mathrm{d}t} = \int_{S_{s}+S_{u}+S_{d}} -(\boldsymbol{u}\cdot\hat{\boldsymbol{n}})\boldsymbol{u}\mathrm{d}S + \int_{S_{d}} \left(\frac{1}{2}q^{2}\hat{\boldsymbol{n}} - \frac{1}{2}(\boldsymbol{u}\cdot\hat{\boldsymbol{n}})(\boldsymbol{x}\times\boldsymbol{\omega})\right)\mathrm{d}S.$$
 (2.10)

For an axisymmetric flow characterised by a large gradient in the cross-stream direction  $\omega_{\phi} = -\partial u_x/\partial\sigma$ , which means that the last integral is zero. Using (2.9), the first integral use of change of the vortex impulse is  $(Q_w - Q_e)U_v$  term. In total, the use of change of the vortex impulse is  $(Q_w - Q_e)U_v$  term. In total, the use of change of the vortex impulse is  $(Q_w - Q_e)U_v$  term. In total, the use of change of the vortex impulse is  $(Q_w - Q_e)U_v$  term. In total, the use of change of the vortex impulse is  $(Q_w - Q_e)U_v$  term. In total, the use of change of the vortex impulse is  $(Q_w - Q_e)U_v$  term. In total, the use of change of the vortex impulse is  $(Q_w - Q_e)U_v$  term. In total, the use  $(Q_w - Q_e)U_v$  term. In total  $(Q_w - Q_e)U_v$  term.

$$\frac{\mathrm{d}I_v}{\mathrm{d}t} = (Q_w - Q_e)U_v - 2Q_w U_v. \tag{2.11}$$

The physical interpretation follows the discussion of Hunt & Eames (2002) and Eames <sup>191</sup> & Hunt (2004) – the source term gives rises to a Lamb thrust force  $(Q_w - Q_e)U_v$ , while <sup>192</sup> the momentum flux in the wake generates a drag  $-2Q_wU_v$ . Thus, the change in the local <sup>193</sup> vortex impulse (2.11) becomes <sup>194</sup>

$$\frac{\mathrm{d}I_v}{\mathrm{d}t} = -(Q_w + Q_e)U_v,\tag{2.12}$$

with entrainment now accounted for in this new description. This explanation provides 195 the first clear and consistent link between the wake-drag model of Maxworthy and the 196 entrainment drag-law discussed in the context of cross-flow jets (see Coelho & Hunt 197 (1989) for a discussion of the key physics for strong cross-flow jets). The consequence 198 of the local flow being dominated by a source (2.8) is still consistent with the global 199 perspective of a far field dipole moment. Taking into account the loss of impulse and its 200 change along the vortex path is equivalent to a line distribution of sinks whose total sum 201 is equal to  $Q_e - Q_w$ , as shown in figure 2(b). The global impulse, created by the trailing 202 wake and the head of the vortex, is thus conserved. 203

The volume flux can be expressed in terms of an average wake flow  $U_w$  and equivalent wake radius  $R_w$  204

$$Q_w = \pi R_w^2 U_w. \tag{2.13}$$

Maxworthy (1974) interpreted the volume flux as an equivalent drag force  $F_D = \rho U_v Q_w$  to which he ascribed an equivalent drag coefficient  $C_Q = 2Q_w/\pi R_v^2 U_v$ . The wake-width behind a vortex  $R_w$  is expected to be smaller than that behind a solid sphere because the growth of the vortex-wake-width is due to significant negative strain at its rear rather than due to boundary layer separation, so that  $R_w/R_0 = \beta$ . With the velocity deficit  $U_w/U_v = \eta$ , these estimates can be combined into 211

$$Q_w = \beta^2 \eta \pi R^2 U_v, \qquad (2.14)$$

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<sup>212</sup> so that  $C_Q = 2\beta^2 \eta$ . The wake width is still much narrower than that of a rigid body with <sup>213</sup> estimated values  $\beta \sim 1/3$  and  $\eta \sim 1/4$ . An *a priori* estimate of  $C_Q \sim 0.12$  is close to <sup>214</sup> the value  $C_Q = 0.1$  that Maxworthy inferred from lines-of-fit to his experimental data. <sup>215</sup> To account for the growth of the vortex, Maxworthy (1974) introduced an additional <sup>216</sup> integral equation describing the increase in volume due to an effective entrainment rate <sup>217</sup>  $\alpha$ 

$$\frac{\mathrm{d}V_v}{\mathrm{d}t} = Q_w = 4\pi\alpha R_v^2 U_v,\tag{2.15}$$

based on generalising the entrainment concept of Morton et al. (1956). Here entrainment 218 is ascribed over the whole surface of the vortex and  $C_Q = 8\alpha$ . Maxworthy (1974) defined 219 a lumped parameter  $C_D = C_Q/4\alpha$  which was determined to have a value that spans 220 from 1.9 to 2.7 in his tests. The growth of the vortex arises due to the wake volume 221 flux  $(Q_w)$  and entrainment  $(Q_e + Q_w)$ , that gives an effective entrainment coefficient 222  $\alpha = C_Q/8$ . This closure gives a fixed value of  $C_D = 2$  which sits in the range estimated 223 by Maxworthy (1974). This also explains why the effective entrainment rate is in the 224 range  $\alpha \sim 0.010 - 0.0012$  (see Maxworthy 1977, figure 24, and table 1), i.e. by an 225 order of magnitude smaller than the entrainment coefficient of a jet which is typically 226  $\alpha \sim 0.065 - 0.08$  (see van Reeuwijk & Craske (2015)). 227

#### 2.3. Interaction with turbulence

The bulk model of vortex movement, described by Maxworthy (1974), is developed 229 to include the approximate effects of external turbulence on the vortex dynamics. This 230 proves a fruitful avenue to pursue because it enables the influence of a spatially and 231 temporally varying turbulent field on the gross dynamics of a vortex to be assessed. 232 The external turbulence is effective at dispersing material shed behind the vortex but 233 this influence on the vortex impulse is of secondary importance compared to the growth 234 of the vortex through entrainment. The effect of external turbulence on the vortex is 235 parameterised in terms of its entrainment through the vortex surface  $(S_v = 4\pi R_v^2)$ , 236

$$Q_e = S_v \alpha_e u_e. \tag{2.16}$$

The growth of the vortex occurs through entrainment caused by the turbulence exterior to the vortex surface and the wake volume flux, a process which is parameterised by

$$\frac{\mathrm{d}V_v}{\mathrm{d}t} = Q_e + Q_w. \tag{2.17}$$

<sup>239</sup> The impulse of the vortex is

$$I_v = (1 + C_m) U_v V_v, (2.18)$$

<sup>240</sup> and for an approximately spherical vortex,  $C_m = 1/2$  (Saffman 1995). Combined with <sup>241</sup> (2.12), this means that the dynamics of the vortex are described by

$$\frac{\mathrm{d}R_v}{\mathrm{d}t} = \alpha_e u_e(x,t) + \alpha U_v \tag{2.19}$$

$$\frac{\mathrm{d}U_v}{\mathrm{d}t} = -\frac{2+C_m}{1+C_m} \frac{U_v(Q_w + Q_e)}{V_v}$$
(2.20)

$$\frac{\mathrm{d}X_v}{\mathrm{d}t} = U_v. \tag{2.21}$$

The model (2.19-2.21) differs from Maxworthy (1974) with the inclusion of effect of entrainment caused by external turbulence that is included in (2.19) and (2.20). Combining

(2.20) and (2.17) gives

$$\frac{U_v}{U_{v0}} = \left(\frac{R_v}{R_0}\right)^{-5},\tag{2.22}$$

regardless of how the turbulent intensity varies in time or space. The influence of external turbulence is parameterised to have an *external* entrainment coefficient,  $\alpha_e \sim 0.12$ , that is more typical of turbulent jets (eg van Reeuwijk & Craske (2015)), and the *internal* entrainment coefficient taken to be  $\alpha \sim 0.012$ , which was inferred by Maxworthy (1977) from the growth of the size of the turbulent vortex. 249

The dynamics of a dipolar vortex can be understood by integrating the coupled system of equations (2.19-2.21). This system of equations admits a number of useful analytical results for the case of  $u_e = 0$  (from Maxworthy) or  $u_e$  constant. When  $u_e = 0$ , 252

$$\frac{U_v}{U_{v0}} = \left(1 + \frac{6\alpha U_{v0}t}{R_0}\right)^{-\frac{5}{6}},\tag{2.23}$$

which on integration gives

$$\frac{X_v}{R_0} = \frac{1}{\alpha} \left( \left( 1 + \frac{6\alpha U_{v0}t}{R_0} \right)^{\frac{1}{6}} - 1 \right).$$
(2.24)

For a field of turbulence maintained at a constant turbulent intensity  $u_0$ , the penetration distance can be evaluated from 255

$$X_{max} = \int_{U_v}^0 \frac{\mathrm{d}U_v}{\frac{\mathrm{d}U_v}{\mathrm{d}t} \frac{\mathrm{d}t}{\mathrm{d}X_v}} = \frac{R_0}{5\alpha} \left(\frac{\alpha}{\alpha_e I_t}\right)^{\frac{1}{5}} \int_0^{\frac{\alpha}{\alpha_e I_t}} \frac{\mathrm{d}\tilde{U}}{\tilde{U}^{1/5}(1+\tilde{U})}.$$
 (2.25)

When  $\alpha_e I_t / \alpha \gg 1$ ,

$$\frac{X_{max}}{R_0} \sim \frac{1}{4\alpha_e I_t} \left( 1 - \frac{4}{9} \frac{\alpha}{\alpha_e I_t} \right).$$
(2.26)

The vortex model, based on the enhancement of boundary movement caused by entrainment, predicts a maximum penetration distance that varies inversely with  $I_t$ . Arnold *et al.* (2013) developed a ballistic model, based on large-scale vortices leading to differential advection of the vortex, with the maximum penetration distance scaling as  $X_{max}/R_0 \sim 1/((L/R)I_t)$ .

## 3. Numerical approach

#### 3.1. Problem definition

A Hill's (1894) spherical vortex, with an initial radius  $R_0$  and speed  $U_{v0}$ , moves into homogeneous turbulence, characterised by an initial rms-velocity  $u_0$  and integral lengthscale L. The incompressible flow evolves according to the Navier-Stokes equation and the

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#### <sup>274</sup> differential form of the conservation of mass, or

$$\rho\left(\frac{\partial \boldsymbol{u}}{\partial t} + \boldsymbol{u} \cdot \boldsymbol{\nabla} \boldsymbol{u}\right) = -\boldsymbol{\nabla} p + \mu \boldsymbol{\nabla}^2 \boldsymbol{u}, \qquad \boldsymbol{\nabla} \cdot \boldsymbol{u} = 0.$$
(3.1)

<sup>275</sup> The fluid has a density  $\rho$  and dynamic viscosity  $\mu$ . It is pertinent to evaluate the properties <sup>276</sup> of a vortex evolving in time. Since the vortex moves through turbulence, the use of <sup>277</sup> vorticity to discriminate the vortex edge, commonly applied to discriminate interfaces <sup>278</sup> that separate turbulent from non-turbulent regions (see da Silva *et al.* 2014) is not viable. <sup>279</sup> Experimentally, interfaces are typically discriminated using a passive dye represented as <sup>280</sup> a scalar  $C(\mathbf{x}, t)$  whose evolution in time is described by

$$\frac{\partial C}{\partial t} + \boldsymbol{\nabla} \cdot (\boldsymbol{u}C) = D\boldsymbol{\nabla}^2 C, \qquad (3.2)$$

where D is the diffusivity of the scalar. In experimental studies with water, the diffusivity of dyes are 3 orders of magnitude smaller than the kinematic viscosity, the computational challenge of dealing with high Schmidt numbers generally precludes using scalars as a diagnostic to analyse vortex movement. An alternative approach is to use a Lagrangian method where fluid particles are tracked in time. A fluid particle located at  $X_0$  at t = 0moves to X in time t, where

$$\frac{\mathrm{d}\boldsymbol{X}}{\mathrm{d}t} = \boldsymbol{u}(\boldsymbol{X}, t), \qquad \boldsymbol{X}(0) = \boldsymbol{X}_0, \tag{3.3}$$

and this has the advantage of enabling the fluid exchange between the vortex and external
 flow to be assessed.

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#### 3.2. Flow initialisation

The initial flow  $\boldsymbol{u}(\boldsymbol{x},0) = \boldsymbol{u}_v + \boldsymbol{u}_t$  is created by the superposition of a spherical vortex  $(\boldsymbol{u}_v)$  and a homogeneous turbulent field  $(\boldsymbol{u}_t)$  defined in a region starting a normal distance  $X_f$  from the vortex centre. The flow generated by the Hill's spherical vortex (Hill 1894),  $\boldsymbol{u}_v = (\boldsymbol{u}_r, \boldsymbol{u}_\theta)$  is set internally by

$$(u_{vr}, u_{v\theta}) = U_{v0} \left( \left( \frac{5}{2} - \frac{3}{2} \frac{r^2}{R_0^2} \right) \cos \theta, \left( -\frac{5}{2} + \frac{3r^2}{R_0^2} \right) \sin \theta \right),$$
(3.4)

where  $r = |\mathbf{x}|, \theta = \cos^{-1}(\mathbf{x} \cdot \hat{\mathbf{x}}/r)$  and the internal azimuthal vorticity field is

$$\omega_{v\phi} = \frac{15U_{v0}}{2R_0^2} r \sin \theta.$$
 (3.5)

<sup>295</sup> Outside the vortex, the flow is described by

$$(u_{vr}, u_{v\theta}) = U_{v0} \left( \frac{R_0^3}{r^3} \cos \theta, \frac{R_0^3}{2r^3} \sin \theta \right),$$
(3.6)

and corresponds to the irrotational flow past a sphere. The Eulerian impulse and momentof the vortex are

$$I_{v0}^{(E)} = (1 + C_m)U_{v0}V_0, \qquad M_{v0}^{(E)} = U_{v0}V_0, \qquad (3.7)$$

respectively, where  $V_0 = 4\pi R_0^3/3$  and  $C_m$ , the added-mass coefficient, takes a value of 1/2.

An incompressible homogeneous random flow is constructed by a summation of Fourier modes chosen to satisfy a specified energy spectrum E(k) (see Kraichnan 1959; Fung *et al.* 1992). The turbulent field is defined to start a distance  $X_f$  in front of the vortex center so

that it does not interfere with the initial stability of the vortex. This standard technique is based on an incompressible velocity field of the form 304

$$\boldsymbol{u}_t(\boldsymbol{x}) = \sum_{n=1}^{N_m} \boldsymbol{a}_n \times \hat{\boldsymbol{k}}_n \cos(\boldsymbol{k}_n \cdot \boldsymbol{x} + \phi_n) H(\boldsymbol{x} - X_f), \qquad (3.8)$$

where  $\phi_n$  is a random phase,  $\mathbf{k}_n$  is a wave number vector,  $\mathbf{k}_n$  is a unit vector with random direction, and  $\mathbf{a}_n$  is chosen so that the velocity field has the statistical properties of a prescribed energy spectrum E(k). A range of wave numbers are chosen to span the spectrum. Here we set  $N_m = 300$  and wave numbers  $k = k_n$  range from  $0.2k_0$  to  $10k_0$ (where  $k_0 = 2\pi/L$ ). The wave vectors  $\mathbf{k}_n = k_n \mathbf{\hat{k}}_n$  and  $\mathbf{a}_n$  are chosen from a distribution that satisfies

$$\langle \boldsymbol{a}_n \rangle = \boldsymbol{0}, \qquad \langle |\boldsymbol{a}_n|^2 \rangle = 2 \int_{k_n - \delta k/2}^{k_n + \delta k/2} E(k) \mathrm{d}k.$$
 (3.9)

We follow Fung *et al.* (1992) and prescribe an energy spectrum of the form

$$E(k) = \frac{\gamma g_2(k/k_0)^4}{\left(g_1 + (k/k_0)^2\right)^{17/6}} \frac{u_0^2}{k_0}, \qquad k_0 = \frac{2\pi}{L}, \tag{3.10}$$

where

$$g_1 = \frac{\pi \Gamma^2(5/6)}{\Gamma^2(1/3)}, \quad g_2 = \frac{55}{9} \frac{g_1^{5/6}}{\pi}.$$
 (3.11)

 $\gamma$  is a correction factor caused by limiting the wave number range and  $\Gamma$  is the Gamma function (Abramowitz & Stegun 1964). The vorticity field associated with (3.8) is 314

$$\boldsymbol{\omega}_t(\boldsymbol{x}) = \boldsymbol{\nabla} \times \boldsymbol{u}_t = \sum_{n=1}^{N_m} \boldsymbol{a}_n k_n \sin(\boldsymbol{k}_n \cdot \boldsymbol{x} + \phi_n). \tag{3.12}$$

The initial rms speed,  $u_{rms(0)}$ , was evaluated from the initialised velocity field with the ratio  $u_{rms(0)}/u_0$  varying from 0.975 to 1.02.

#### 3.3. Numerical solution

Equations (3.1, 3.2) were solved using OpenFOAMv1706, which is a general computational framework for object orientated solutions to multiphysics problems through finite volume and Lagrangian particle tracking. 320

To study the vortex-turbulence interaction, the domain size was chosen to be  $30R_0 \times$ 321  $8R_0 \times 8R_0$ . The effect of flow boundedness is expected to have a  $1/8^2$  effect on the 322 mean vortex speed. The vortex is initialised at a distance of  $3R_0$  from the inlet plane, 323 with  $X_f = 5R_0$ . A uniform block mesh was used - a variety of sizes were tested and 324 settled for  $1000 \times 260 \times 260$ . Slip boundary conditions were applied to the walls of 325 the domain and the simulations ran to time  $t = U_{v0}t/R_0 = 50$ . Mesh convergence was 326 tested by analysing the vortex position as a function of time and this showed less than 2%327 difference in vortex position between simulations running at 8.45 million cells, compared 328 to 67.6 million cells. The evolution of the rms speed of the homogeneous turbulence was 329 calculated as a function of time and fitted with a power-law of the form 330

$$\frac{u_e}{u_0} = \left(1 + \frac{2.0u_0t}{L}\right)^n,$$
(3.13)

with a decay index of n = -0.7, which is typical for decaying homogeneous turbulence (Wilczek *et al.* 2011). 332

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Figure 3. (a) Schematic of the experimental setup, showing the piston dimensions (diameter  $D_p$ , stroke length  $L_p$ ) and grid dimensions (stroke distance S and bar separation M). The grid mass was increased with metal weights (not shown) to stabilise its motion. The box turbulence is characterised in terms of (b) the scaled turbulent velocity,  $u_{rms}/fS$ , fitted with an exponential  $a + b \exp(-cz)$ , (see legend and (4.1)) and (c) turbulent lengthscale,  $\ell$ , plotted against height z above the grid. The origin z = 0 corresponds to the maximum height of the grid (see Hopfinger & Toly 1976), i.e.  $\approx 5$  cm above the bottom of the tank.

The vortex center,  $X_v$ , was calculated from the Lagrangian position of fluid particles that were initially located within the vortex:

$$X_v = \frac{1}{N_p} \sum_{i=1}^{N_p} X_i.$$
 (3.14)

Since the fluid particles are advected by the local fluid velocity (see (3.3)), the vortex velocity is

$$U_v = \frac{1}{N_p} \sum_{i=1}^{N_p} \frac{\mathrm{d}X_i}{\mathrm{d}t} = \frac{M_x^{(L)}}{V_0}.$$
(3.15)

The vortex radius was defined in terms of the standard deviation from the vortex centre, so that  $R_{std} = \overline{(X_i - X_v)^2}^{\frac{1}{2}}$ . The vortex was seeded with  $N_p = 10^4$  particles.

#### 4. Experimental setup and measurements

In the laboratory, the interaction of a vortex with turbulence is studied in a cubic 340 Plexiglass tank of  $1.0 \times 1.0 \times 0.90$  m<sup>3</sup> filled with tap water to a height 89 cm as shown in 341 figure 3(a). About thirty minutes prior to the vortex generation near the fluid surface, 342 the turbulence generation is started near the bottom of the tank by the vertical (and 343 sinusoidal) oscillation of a grid with a mesh size close to that used in Hopfinger & Toly 344 (1976). To visualise the flow, the fluid is seeded with spherical particles of a diameter of 345 60  $\mu m$  and density of approximately  $\rho = 1.03 \mathrm{g \ cm^{-3}}$ , while the vortex is coloured with a 346 fluorescent dye (Rhodamine 6G). A small amount of sodium hydrochlorate is added to the 347 water in order to slowly bleach the dye and to be able to repeat the experiment in the same 348 fluid. The flow is illuminated by a vertical laser sheet passing through the centre of the 349 tank, generated with a 2 W Spectra Physics Millenia Pro laser and cylindrical lens. For 350 the recording and flow measurements a Dalsa 1M60 Pantera camera is used with a frame 351 rate between 5 to 20 Hz (see figure 3) and in addition a Sony Alpha III camera. These 352 recordings are processed with the Particle Image Processing program UVmat (http:// 353 servforge.legi.grenoble-inp.fr/projects/soft-uvmat), developed at LEGI, which 354 allowed for the measurements of the rms velocity of the turbulence as well as the vorticity 355 fields, whereas the dye visualisations are processed with ImageJ (https://imagej.net/). 356 The distance of the vortex centre from its start,  $X_v$ , and vortex radius,  $R_v$ , defined in 357 terms of the separation of the vortex cores are determined from vorticity maxima and dye 358 visualisation, as functions of time. The vortex speed,  $U_v$ , is deduced from the displacement 359 of the vortex centres. The turbulence of the grid is calibrated, such that the turbulent 360 velocity,  $u_e$ , and length scale  $\ell$ , are known from the forcing parameters. 361

The vortices are injected a few cm below the tank free surface, at about z = 84 cm, 362 such that the vortex centres of the formed vortex are at about z = 73 cm above the 363 virtual origin in figures 3(b,c). The translation velocity  $U_v$  of the vortex is measured 364 from the dye and for some cases, verified with the displacement of the vorticity maxima 365 from PIV measurements. Below the details of the vortex generation, and the turbulence 366 generation are discussed. 367

#### 4.1. Vortex generation

The vortex was generated by the movement of a piston, controlled by a stepper motor, 369 which drives flow through a pipe with internal diameter  $D_p = 11$  cm or  $D_p = 4$  cm 370 diameter. The initial speed of this vortex,  $U_{v0}$ , is proportional to the piston velocity, 371 whereas the circulation is determined by the distance travelled by the piston, and the 372 size of the vortex is mainly determined by the diameter of the cylinder (see Gharib 373 et al. 1998, for details). The displacement distance of the piston to generate the vortex 374 ring,  $L_p$ , was kept constant and is equal to 13.65 cm during each experiment, so that the 375 vortex formation number,  $L_p/D_p$ , is fixed at a value of 1.24, and 3.41 in case of the small 376 vortex. Since this vortex formation number is below 4 in this study, only isolated vortex 377 rings are generated with no (or hardly any) trailing wake (e.g. Gharib et al. 1998). The 378 Reynolds number of the vortex, based on the initial velocity  $U_{v0}$  and diameter  $D_0$  just 379 after generation,  $Re_0 = D_0 U_{v0} / \nu$ , was varied between 1490 and 5660 (see Table ??). 380

#### 4.2. Turbulence generation and properties

A turbulent field is generated with a vertically oscillating grid, made of square bars of 382 thickness d = 1.5 cm, separation M = 6 cm covering an area 97 cm  $\times$  97 cm. The gap 383 between the grid and the sidewall of the tank is minimized with 5mm thick matt black 384

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Perspex plates. This also improves the quality of the recordings. The stroke of the grid,
 S, was varied at the values 5.4, 7.4, and 10.4 cm.

The characteristics of the turbulence are set by the grid forcing frequency (f) and amplitude of forcing (S), with the characteristic velocity and length-scale set by velocity and the dimensions of the grid. Turbulence statistics are obtained from long time statistics of planar PIV measurements in a vertical plane. The in-plane rms velocity  $u_{0,2D}$  and integral length-scale L are expressed in terms of the in-plane velocity fluctuation u', through

$$u_{0,2D} = \overline{(u'^2)}^{1/2}$$

and

$$L = \int_0^{R_L} \overline{u'(r)u'(r+x)} \mathrm{d}r/\overline{(u'^2)},$$

where  $R_L$  upper bound for the correlation integral. The results for  $u_{0,2D}$  and L are 387 shown in figure 3(b,c) as a function of distance z. At larger distances from the grid, the 388 turbulence intensity and integral lengthscale tend to a constant where the small structures 389 between the plumes that are generated by the moving bars, are dissipated. Similar results 390 of a constant  $u_{0,2D}$  and length scale L were also obtained for faster grid motion ( $f \sim 1$ 391 Hz) and a smaller stroke S (personal communication with J. Sommeria). Close to the 392 moving grid, the turbulent intensity increases and the integral scale decreases as the 393 flow is dominated by the shedding of small intense vortices from the grid. In view of the 394 present interest of knowing the turbulence intensity at a particular level, an exponential 395 fit was employed 396

$$\frac{u_{0,2D}}{fS} = 0.036 + 0.4 \exp\left(-\frac{z}{(SM)^{1/2}}\right),\tag{4.1}$$

which is shown in figure 3(b). The saturation of the turbulence near a constant value of 397  $u_{rms}$  and L has not been observed in Hopfinger & Toly (1976) and more recently Poulain-398 Zarcos et al. (2022) since in these experiments the measurement region is limited to a 399 distance of 35 cm of the grid position. In Hopfinger & Toly (1976) the grid oscillated at 400 mid depth in 80 cm height tank with a ten times faster oscillation frequency, leaving an 401 effective 35 cm distance from the grid, as also in Poulain-Zarcos et al. (2022). Further, a 402 ten times lower frequency was used to keep the turbulence weak compared to the vortex. 403 Figures 3(b,c) show indeed that at z > 30 cm the length scale L stagnates around a 404 value of  $\approx 7.5 \pm 1.5$  cm, corresponding to the mesh size, while  $\frac{u_{0,2D}}{fS^{3/2}} \approx 0.013$ , implying a 405 maximum Reynolds number of  $Re_t = u_{0,2D}L/\nu$  roughly between 120 and 240. 406

In Hopfinger & Toly (1976), the grid oscillated at the tank mid height, and the virtual origin (where L = 0) corresponded to the position  $z_{top} - (S/2 + 0.01m)$ , with  $z_{top}$  the maximum position of the grid, thus slightly below centre position of the grid. This is also taken as the reference position (z = 0) in figures 3(b,c). In the present experiments, the grid oscillates near the bottom, and the level with L = 0 corresponding to the bottom of the tank, showing the effects of a near boundary on the turbulence.

The turbulent intensity in the vicinity of the vortex changes as the vortex descends. To compare the experimental results with the numerical simulations, a consistent definition of the turbulent intensity is taken

$$I_t = \frac{(3/2)^{1/2} \overline{u}_{0,2D}}{U_{v0}},\tag{4.2}$$

that is expressed in terms of  $\overline{u}_{0,2D}$  which is determined by averaging  $u_{0,2D}$  over the path

Exp. Label	f (Hz)	S (cm)	$D_p$ (cm)	$U_{v0} \ ({\rm cm/s})$	$D_0 \ ({ m cm})$	$\overline{u}_{0,2D}~({\rm cm/s})$	$X_{max}/R_0$	$U_{v0}/\overline{u}_{0,2D}$
А	0.63	5.4	4	2.778	5.358	0.123	12.971	22.548
В	0.63	5.4	4	3.083	6.222	0.154	14.588	19.974
$\mathbf{C}$	0.65	5.4	4	4.128	5.915	0.168	13.665	24.548
D	0.65	7.4	4	1.553	6.250	0.173	4.599	8.960
$\mathbf{E}$	0.65	7.4	4	9.279	5.967	0.193	19.639	48.082
$\mathbf{F}$	0.65	7.4	4	9.346	6.603	0.219	18.541	42.652
G	0.65	7.4	4	7.546	6.029	0.193	16.862	39.024
Н	0.5	5.4	11	2.203	14.550	0.120	9.138	18.312
Ι	0.625	5.4	11	2.103	14.847	0.131	7.761	16.041
J	0.625	5.4	11	2.180	14.605	0.139	8.651	15.705
Κ	0.625	5.4	11	3.283	14.725	0.155	9.177	21.206
$\mathbf{L}$	0.65	10.4	11	3.235	17.485	0.311	4.494	10.395
Μ	0.67	10.4	11	1.408	14.732	0.252	1.212	5.583
Ν	0.50	10.4	11	1.232	14.337	0.189	2.379	6.510
Ο	0.55	10.4	11	1.320	11.827	0.212	4.264	6.219
Р	0.55	10.4	11	1.452	11.041	0.210	4.448	6.918
Q	0.60	10.4	11	0.528	14.094	0.226	1.117	2.335
Ř	0.60	10.4	11	0.264	13.525	0.226	0.484	1.170
$\mathbf{S}$	0.60	10.4	11	1.056	11.833	0.226	1.544	4.674
Т	0.65	7.4	11	2.219	14.537	0.198	6.034	11.206

Table 1. Data of the vortex in oscillating grid turbulence. The parameters f, S,  $D_p$  and  $D_0$  (which is  $\approx 2R_0$ ) are defined in figure 3(a);  $U_{v0}$  and  $\overline{u}_{0,2D}$  are the vortex and rms inplane turbulent velocities at t = 0, respectively, whereas  $X_{max}/R_0$  is the scaled maximum travelled distance by the vortex. The vortex starts in turbulence so that  $X_f = 0$ .

of the vortex. The prefactor  $(3/2)^{1/2}$  is used to convert the in-plane rms velocity to the rms of the three-dimensional velocity field.

## 5. Numerical results: Vortex movement into a quiescent region

## 5.1. Vortex induced transport

To distinguish between the effect of the vortex on transport outside the vortex from the 421 processes that occur within the vortex, two groups of fluid particles were followed in time 422 - see figure 4. Drift is associated with differential movement of fluid and was quantified 423 using a distribution of particles in the plane z = 0 in the form of a structured grid. 424 The lateral displacement of the grid lines parallel to the direction of vortex propagation 425 gives an indication of blocking and source strength, while the distortion of the grid 426 lines perpendicular to the vortex propagation indicates Darwin drift. To characterise 427 the transport by the vortex 10 000 particles are seeded randomly within the vortex and 428 followed in time. The two groups of fluid particles are distinguished by colour (see figure 429 4(a)) with those initially inside the vortex are blue, and those initially outside the vortex 430 are red. 431

The drift profiles confirm the longitudinal displacement profiles reminiscent of those identified by Darwin and Maxwell. Drift profiles have been used to analyse the kinematic transport by dipolar vortices (Eames & Flor (1998)). The stretching of fluid elements passing close to the stagnation points is clear, leading to a tilting of fluid elements 435

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<sup>436</sup> along the *x*-axis. The lines of fluid particles parallel to the propagation direction are <sup>437</sup> laterally displaced, as a consequence of the volume flux deficit behind the vortex. The <sup>438</sup> lateral divergence is a consequence of the presence of the source flow due to the drag <sup>439</sup> on the vortex. The lateral displacement behind the vortex is approximately constant <sup>440</sup> over a period of these simulations. The volume flux behind the vortex changes when the <sup>441</sup> manner in which the vorticity is deposited in the wake is interrupted and this is observed <sup>442</sup> to occur when the vortex is undergoing an inviscid adjustment, corresponding to a time

beyond t = 40. Consequently the drag on the vortex is approximately constant up until 443 this time. In the external flow around the vortex, the accumulation and stretching of 444 fluid particles near the leading stagnation point highlights the front of the vortex. Their 445 motion confirms that the fluid enters the rear of the vortex and circulates the vortex 446 core. At a later time  $(t \sim 40)$ , the edge of the vortex is difficult to distinguish from 447 the side view due to three-dimensionality of the flow. The non-uniform radial particle 448 distribution within the vortex indicates the entrainment of ambient fluid which does not 449 contain particles. 450

#### 451

# 5.2. Velocity and vorticity field

Hill's spherical vortex is based on a steady inviscid model where the initial strength 452 of  $\omega_{\phi}$  varies linearly with distance from the centreline (see (3.5)). The most significant 453 process is the loss of vorticity due to the diffusive flux across the bounding streamline 454 which leads to vorticity being deposited behind the vortex. It is this loss which leads to a 455 faster reduction of the maximum vorticity in the vortex than vorticity annihilation. The 456 width of the vortical wake behind the vortex is  $\beta R_0$ , (where  $\beta \sim 1/3$ ) is smaller than for a 457 solid body because of the absence of flow separation that arises from the no-slip condition; 458 the vorticity strength on the surface of the solid body scales as  $U_0/R_0Re^{1/2}$ . The vortex 459 wake is much larger than a clean spherical bubble (Moore (1963)), even through the 460 vortical strength is comparable  $(U_0/R_0)$ , because of the kinematic constraint imposed by 461 the bubble surface. The width of the downstream vortex wake is largely determined by 462 cross-stream diffusion in a negatively strained flow, which tends to greatly widen the wake 463 (Hunt & Eames 2002). Viscous diffusion of oppositely signed vorticity on the centre-line 464 of the vortex leads to vorticity annihilation. This effect is far weaker than Maxworthy 465 identified because vortex compression leads to  $\omega_{\phi}$  tending to zero on the centreline. At a 466 later time of  $\tilde{t} = 40$ , the azimuthal instability has grown sufficiently large that the vortex 467 starts to become asymmetric. This instability is likely due to the mechanism explained 468 by Widnall (Widnall & Sullivan 1973) and appears for  $Re \gtrsim 2000$  (see Maxworthy 1977). 469

The planar streamline pattern, in the frame moving with the vortex, is shown in figure 5(a), along with the distribution of the vorticity component that is normal to the page. The vortex center and velocity are determined from the fluid particles initially released inside the vortex. The permanent cross-stream displacement of streamlines originating upstream of the vortex, seen in figure 5(a), are indicative of a source flow, or equivalently, a drag on the vortex.

As the vortex propagates, the vortex core changes shape and material is shed into its wake. Cross-stream slices of the scalar distribution are shown in figures 5(b)(i-v). Material is lost from the vortex through a diffusive flux at the side of the vortex (see figure 5a) with a trail of vorticity and tracer in the wake. The diffusivity of the tracer is set the same as that of vorticity. The overall growth of the three-dimensional effects are affected by both, mesh resolution and mesh shape.



Figure 4. (a) Time sequence showing the displacement and transport of fluid particles by a Hill's spherical vortex moving with initial Reynolds number  $Re_0 = 2000$ . The red particles were initially placed in a grid arrangement with planes that lie perpendicular and parallel to the initial velocity of the vortex; the green particles were placed randomly within the vortex. The panels are shown for  $\tilde{t} = 5, 10, 20, 30$  and 40 where  $\tilde{t} = U_{v0}t/R_0$ . The vortex is represented as an iso-surface of the second-invariant of the velocity gradient tensor,  $II = \frac{1}{2}(||\boldsymbol{\Omega}||^2 - ||\boldsymbol{S}||^2)$ , where  $\boldsymbol{\Omega} = (\nabla \boldsymbol{u} - (\nabla \boldsymbol{u})^T)/2$  and  $\boldsymbol{S} = (\nabla \boldsymbol{u} + (\nabla \boldsymbol{u})^T)/2$ (Hunt *et al.* (1988)). The iso-surface  $II = 0.02 \text{ s}^{-2}$  is plotted. (b) The projected view of vortex shown iso-contour of  $II = 0.02 \text{ s}^{-2}$  and  $|\boldsymbol{\omega}| = 0.1 \text{ s}^{-1}$ .

#### 5.3. Speed, shape and impulse

The vortex position was calculated from the mean and the maximum distance moved 483 by marked internal fluid particles and both gave substantially the same result, as shown 484 in figure 6(a). Figure 6(a) shows a comparison with the model and  $\alpha = 0.012$ . This agrees 485 for short time during which the three-dimensional deformation of the vortex ring is still 486 small, since from figure 6(c),  $C_Q = Q_w/\pi R_0^2 U_{v0} \sim 0.10$ , giving  $\alpha \sim 0.0125$ . At around 487  $\tilde{t} \sim 40$ , the azimuthal instability mentioned above starts to grow and the simple drag 488 model is not applicable. The vortex decreases faster than anticipated apparently due to 489 the rapid increase in its radius. 490



Figure 5. (a) The distribution of vorticity  $\omega_z$  in the plane z = 0, is shown for  $\tilde{t} = 5, 10, 20, 30$  and 40. The streamlines are plotted in the the frame moving with the vortex, where the vortex speed is determined from the average Lagrangian velocity. (b) The distribution of scalar, C, initially set as C = 1 within the vortex, is indicated for times  $\tilde{t} = 5, 10, 20, 30$  and 40.

To explore the adjustment from an axisymmetric vortex ring to a ring with a series 491 of modes, the shape of the vortex was analysed using information from the Lagrangian 492 particles. Projecting the particles onto a normal plane enabled the outer edge of the 493 vortex  $R_m(\theta)$  to be determined (see figure 6(b) insert), from which the average  $(\overline{R}_m)$ , 494 maximum and minimum  $(max(R_m), min(R_m))$  are determined. We would speculate, 495 based on Maxworthy's (1977) observations, that the vortex will ultimately break down. 496 The volume flux  $Q_w$ , downstream of the vortex, is evaluated from (2.8b) as a function 497 of distance from the vortex centre, and shown in figure 6(c) for four times. During the 498 initial movement of the vortex, the diffusive flux of vorticity into the wake tends to occur 499 at a constant rate, giving rise to a constant volume flux except beyond the starting point 500



Figure 6. (a) The vortex position,  $X_v$ , determined from Lagrangian particle information, is plotted as a function of time (see legends for symbols). (b) Characterisation of the cross-sectional shape of the vortex (determined from the maximum radial distance  $R_m(\theta)$ mean and minimum of  $R_m$  of Lagrangian particles from the centreline). (c) Volume flux  $Q_w$ , determined from (2.8b) as a function of distance behind the vortex, for different times (see insert); (d) Scaled Lagrangian and Eulerian measures of momentum  $(M_x)$  and impulse  $(I_x)$  as functions of time.

where the flow is irrotational. The volume flux behind the vortex starts to decrease at around  $\tilde{t} = 20$ , as the inviscid adjustment causes the size to increase and the vortex entrainment becomes pronounced. This is also evident in figure 5(b) where the tracer concentration in the near wake is also diminished. 501

# 6. Vortex moving into a region of turbulence

## 6.1. Numerical results

Figure 7(a) shows the movement of the initially spherical vortex, for different back-507 ground levels of turbulence ( $I_t = 0.1, 0.2$  and 0.5). The vortex is visualised by placing fluid 508 particles within the vortex and following their evolution in time. The vortex is initially 509 compact in the non-turbulent region and then quickly loses mass through the wake as 510 it moves into the region of turbulence. With increasing strength of external turbulence, 511 the rate at which the vortex loses mass increases. Its head remains initially coherent and 512 while the tagged fluid elements tend to be dispersed in the longitudinal direction, the 513 size of the head of the vortex increases more slowly. At a later stage, the growth of the 514 vortex tends to be dominated by passive dispersion. The presence of turbulence leads 515 to the vortex tending to slow down with an increased sensitivity to being deflected by 516



Figure 7. (a) Lagrangian transport of fluid particles by a vortex moving into a turbulent flow for contrasting levels of ambient turbulence (i)  $I_t = 0.1$ , (ii) 0.2 and (iii) 0.5; in each case  $L/R_0 = 4$ , at times  $\tilde{t} = 5, 10, 20, 30, 40$  and 50. A red line is plotted along the center of the domain, starting from the initial vortex position, to give perspective to the images. The position of the vortex is shown in (b,i) and the normalised velocity plotted against the vortex radius in (b,ii). A horizontal line is plotted at  $X_f/R_0 = 5.0$ . The decay of the vortex velocity is shown in (c,i) for contrasting levels of turbulent intensity. The Lagrangian measures of vortex for (c,i) momentum, (c,ii) impulse, (c,iii) vortex force and (c,iv) diffusive flux of impulse are shown as functions of time.

the ambient turbulence. The mutual distance between the vortices increases while they deform, and flatten the more they enter in the turbulent region. With higher levels of turbulence, the vortex is blocked over a shorter distance and deforms faster, so that the penetration distance decreases. These effects become more pronounced with the intensity of the turbulence.

Figure 7(a) shows the vortex position as a function of time and indicates that over the 522 period of the tests, the vortex slows down faster as it penetrates into the turbulent region 523 (figure 7(c,i)). At the higher values of turbulent intensity, the vortex is largely arrested 524 and is observed to penetrate a finite distance into the turbulence. The impulse of the 525 vortex decreases due to the loss of vorticity but it has a very specific tendency to increase 526 as it interacts with the interface between the turbulent and non-turbulent region. This 527 blocking effect is similar to a vortex impacting a wall which causes the vortex to flatten, 528 vorticity to increase via stretching, leading to an increase in the vortex impulse (see 529 Verzicco *et al.* 1995). This is evident in figure 7(c,i) for the highest external turbulence 530  $(I_t = 0.5)$ . The change of vortex impulse due to vorticity diffusion is negative except for 531 short periods with  $I_t = 0.5$ , when detrained fluid elements pick up vorticity from the 532 ambient turbulence. 533

Figure 8 shows the turbulent field and vortex, as isocontours of Q, for different time 534 and turbulent intensity. The colour in the images shows the (normalised) concentration 535 of the scalar tracer and gives an indication of the material that was originally within 536 the vortex. The decaying turbulence leads to an increase in the integral lengthscale and 537 decrease in the rms velocity, which is seen by the increased space between vortical tubes. 538 The interface between the turbulent and non-turbulent region is deflected by the initial 539 penetration into the turbulence, and eventually disperses. The sequences show that the 540 coherence of the vortex is quickly lost as the strength of the external vorticity increases. 541 The size of the vortex, evident from the highest concentration of passive scalar, grows in 542 time. For the lowest level for turbulent intensity  $(I_t = 0.1)$ , the shed material is aligned 543 to the direction of motion of the vortex, with the wake vorticity quickly dispersed by the 544 external turbulence. 545

The penetration of the vortex into homogeneous turbulence is thus accompanied by a 546 reduction in the vortex speed associated with the loss of wake material and vortex growth 547 in size. The vortex speed  $U_v$  is determined from the Lagrangian measure of momentum 548  $(M_x^L/V_0)$ . For  $I_t = 0.1$ , the reduction of speed leads to an approximately linear decrease of 549 velocity with distance with a rate faster than in the absence of turbulence (figure 9(a)). 550 Both, the ambient turbulence and the vortex speed, decrease in time but at different 551 rates; initially the ambient turbulence decreases faster than the vortex speed, leading to 552  $U_v/u_0$  increasing, but since the loss of mass from the vortex is continuous, the vortex 553 ultimately slows down faster than the rate at which the turbulence decays. 554

#### 6.2. Experimental results

Typical visualisations of the evolution of the vortex moving in turbulence generated 556 by the oscillating grid near the bottom of the tank are shown in figure 10(a-f) for the 557 vortex generated with the 11 cm (inner) diameter tube. During an initial stage, the vortex 558 develops and grows as a result of entrainment (see figure 10(a)) showing relatively little 559 influence of the ambient turbulence for the large vortex. Small turbulent vortices in the 560 ambient are advected around the spherical vortex, and either strained in the wake of the 561 spherical vortex, or entrained into the spherical vortex leading to its intensification (see 562 figures 10(b,c)). During this latter process, the spherical vortex moves outwards and then 563 inwards causing oscillations in vortex radius. Inversely, by conservation of momentum, 564 the propagation velocity decreases and increases, respectively. With distance smaller but 565



Figure 8. A cross-section of the flow structure is shown for (a)  $I_t = 0.1$ , (b) 0.2 and (c) 0.5. The iso-volume contours for the second invariant of the velocity gradient tensor Q are shown, with values  $QR_0^2/U_{v0}^2$  shown greater than 0.025, 0.0125 and 0.0005 respectively for (a), (b) and (c). The time instances correspond to those in figure 7(a). The colour field corresponds to the scalar field C and varies from cyan to yellow (from C = 0 to maximum C). The reference red line is the centreline of the simulations. The lateral span of the box indicates the vortex start and the end of the computational domain, which is smaller than the computational domain.

intenser turbulent vortices are confronted, leading to the breakup of the sharp vortex front into smaller vortices (see figures 10(d-e)). When the vortex approaches the top position of the grid (at about z = 20 cm), the vortex is eventually destroyed by the turbulent jets generated by the oscillating grid (see figure 10(f)). Although the same sequence was observed for the smaller vortices (generated with the 4 cm diameter tube), the turbulent vortices are now comparable to the size of the grid, which deviates, intensifies or weakens



Figure 9. The vortex speed,  $U_v$  (3.15), determined from the numerical simulations, is shown as a function of the distance the vortex has moved. In (a), the contrasting influence of turbulent intensity is shown. In (b), the ratio of the  $U_v/u_e$  is shown as a function of vortex position, where  $u_e$  is determined from (3.13).



Figure 10. Typical vortex evolution in quiescent ambient fluid (top), and with ambient turbulence generated by an oscillating grid near the base of the tank (bottom) at (a) t = 3.5, (b) 10.8 (c) 17.2 (d) 24.7 (e) 30.9 and (f) 35.3s after the vortex generation. The internal diameter of the cylinder is 11 cm (see experiment Q in Table 1, and same parameters for top images without turbulence). The frames are 50 cm wide, and the position of the grid is represented in (f) by the black area near the bottom.

the initial vortex during its development stage increasing the variance of the speed and <sup>572</sup> size. <sup>573</sup>

The position of the vortex centres,  $X_v$ , are determined by manual segmentation of <sup>574</sup> images, such as figure 10, and also from the vorticity extremes obtained from PIV <sup>575</sup>



Figure 11. Results from the experimental study of (a) the vortex velocity  $U_v$  and (b) the normalised vortex velocity  $U_v/u_e$ , as a function of distance from vortex starting for contrasting values  $I_{ve}$ .

measurements. Following the dye that is captured in the vortex centres, the vortex 576 velocity was then calculated by differentiating  $X_v$  with time. The turbulence intensity at 577 the vortex centre is obtained by interpolating the  $u_0$  relation shown in figure 3(b). The 578 initial position of the vortex,  $X_{fe}$ , is considered as the moment the vortex was formed. 579 The vortex radius,  $R_0$ , is measured shortly after injection as half the distance between the 580 two vortex centres when it has reached its mature state, i.e. it has a circular shape and 581 slightly increases in speed. In case this is not possible because the vortex was perturbed 582 by ambient vortices, its maximum velocity shortly after injection has been taken. 583

Figure 11 shows the evolution of the vortex velocity with penetration distance. The 584 oscillations in  $U_v/U_{v0}$  and  $U_v/u_e$  are due to the small turbulent vortex structures that are 585 advected to the rear of the spherical vortex where they are entrained. Subsequently, these 586 small scale vortices are swept internally to the front of the vortex. During this entrainment 587 process the spherical vortex slightly expands in size, and by conservation of impulse, 588 decreases in speed (figure 11(a)). Figure 11(b) shows the relative strength of the vortex 589 to the local turbulent intensity which decreases rapidly with distance; by comparing 590 figure 9(b) and figure 11(b), the vortex appears to be arrested when  $U_v/u_e \sim 2-5$ . 591

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## 6.3. Comparison of numerical, experimental and integral models

The numerical model is based on decaying, homogeneous turbulence and contrasts with the box-turbulence in the experimental study which is statistically steady but spatially varying. The integral model (2.19)-(2.21) provides a means of understanding these general influences of the vortex dynamics.

Figure 12 shows the results of the integral model with the turbulent intensity (3.13) being defined at a distance  $x > X_f$  (for  $X_f = 5R_0$ ), consistent with the numerical model of a spherical vortex moving through a quiescent region prior to penetrating the decaying turbulence. The predicted trend for the velocity and position (figure 12a,c), as a function of time, show quite similar trends to the numerical model, in particular the abrupt change in the vortex speed as it penetrates into the turbulent region. The reduction of the vortex speed is a consequence of the loss of vortical fluid (reducing impulse) and entrainment



Figure 12. Predictions for vortex dynamics based on integral model (2.19)–(2.20) for (a) radius, (b) velocity and (c) position with time, and (d) speed with vortex position, where the turbulence decays according to (3.13) for n = -0.7.

that causes the vortex size to grow. As is evident from figures 7(a) and 10, the increase in the vortex radius is less than 50% during the rapid decrease in vortex speed. The later stage of the spherical vortex dynamics, where  $U_v/u_e < \lambda_c$ , is characterised by the vortex essentially stopping and spreading rapidly in a diffusive manner, a process that is not captured by the integral model. The critical ratio  $\lambda_c$  can be approximately determined from figure 9(b) to be  $\lambda_c \sim 3$ , while the experimental results (figure 11) suggest  $\lambda_c \sim 5$ .

Figure 13 shows a compendium of the penetration distance of a vortex versus the <sup>610</sup> inverse strength of the turbulence, for the experiments, numerical simulations and integral <sup>611</sup> model. The legend lists the specific configuration for the integral model, with the start of <sup>612</sup> the turbulent region and exponent for the turbulence growth listed. Three configurations <sup>613</sup> are listed, specifically decaying turbulence (n = -0.7), sustained turbulence (n = 0) and <sup>614</sup> growing turbulence (n = 2), with the later chosen to mimic the scenario of the vortex <sup>615</sup> moving into progressively stronger turbulence that occurs in the experiments. <sup>616</sup>

The curves for decaying turbulence show the vortex penetration distance at t =617  $50R_0/U_{v0}$ , while the remaining curves describe the ultimate penetration distance. The 618 numerical simulations were repeated with the vortex starting in turbulence  $(X_f/R_0 < 0)$ 619 to contrast the case of a vortex moving into turbulence  $(X_f/R_0 = 5)$ . Starting in 620 turbulence meant that its influence on the vortex dynamics was rapid, reducing the 621 penetration distance, giving a consistent trend similar to the integral model, with the 622 distance varying approximately with  $I_t^{-1}$ . The initial turbulence also affected the vortex 623 stability 624

For turbulence whose intensity is sustained, the integral model recovers the asymptotic result (2.26), where the penetration distance scales with  $I_t^{-1}$ . This approximately linear 625



Figure 13. The variation of the penetration distance into turbulence  $(X_{max} - X_f)$  is plotted against the inverse of turbulent intensity,  $I_t$ , with the results determined by experiments, computations and the integral model. For the experimental study,  $I_t$  is defined by (4.2). The results from the integral model are plotted as full curves and the decay law and turbulent front as listed in the legend. The numerical results for the case of a vortex starting in turbulence ( $\circ$ ) or moving into turbulence ( $\bullet$ ). The experimental results are distinguished between weak and strong vortices,  $\blacktriangle$  and  $\triangle$ , respectively and plotted as a function of the inverse. The asymptotic expression (2.26), for strong turbulence, is compared against the integral model for sustained turbulence.

trend is also evident in the experimental results. There is scatter in measurements of 627 the penetration distance for intense turbulence (high  $I_t$ ) and which is expected when 628 the spherical vortex starts to interact with coherent structures shed from the oscillating 629 bars. The integral model is shown for n = 2 where the vortex moves into a region of 630 turbulence whose strength progressively increases in time. Although this does not match 631 the spatial dependence seen in figure 3, it does highlight the significant reduction of the 632 penetration distance (compared to decaying and forced turbulence) with a reported trend 633 quite similar to the experimental observations. 634

### 635 7. Conclusion

This paper provides a detailed theoretical and experimental study of the arrestment of a spherical vortex penetrating a region of turbulence. The complexity of the new elements required a combination of different techniques to unravel the key physics.

An integral perspective of vortex dynamics in quiescent flow is able to reconcile the far field dipolar nature of the vortex flow with the near field source flow generated by entrainment and impulse loss. This result also holds for vortices moving in turbulence because the turbulence is generally characterised by a negligible total impulse. The vortex model of Maxworthy (1977) is extended to account for the effect of external turbulence in terms of increased entrainment and reconciles the results from the numerical and 644 experimental studies. A modest level of background turbulence causes the vortex to slow 645 down more rapidly and grow faster, leading to a finite penetration distance that scales 646 with  $I_t^{-1}$ . Ultimately the vortex speed becomes comparable to the external turbulence 647 after which the vortex stops and rapidly spreads as it is dispersed by the ambient 648 turbulence. The evolution the vortex has been assessed by tracking a large number 649 of fluid particles. This Lagrangian technique turns out to be a fruitful technique for 650 understanding vortex-turbulence interactions in the future. The consistent feature of the 651 experimental and theoretical model is the small growth rate of the vortex but a significant 652 increase in the loss of vortical fluid at the rear. 653

The paper has examined the impact of external turbulence on the vortex from an 654 integral perspective, specifically analyzing the vortex's movement and determining the 655 penetration distance. The distortion and modification of external turbulence are complex 656 phenomena. Disentangling these general flow processes will be the focus of our future 657 research efforts. 658

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# **Declaration of interests**

The authors report no conflict of interest.

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