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Enrique Flores-Montoya, Pierre-Alexandre Masset, Thierry Schuller, Laurent Selle. Speed-up drivers for H 2 -enriched flames in Porous Media Burners. Proceedings of the Combustion Institute, 2024, 40 (1-4), pp.105666. 10.1016/j.proci.2024.105666 . hal-04775337

HAL Id: hal-04775337 https://hal.science/hal-04775337v1

Submitted on 12 Nov 2024

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Speed-up drivers for H₂-enriched flames in Porous Media Burners

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Abstract

An experimental study on the influence of porosity and hydrogen enrichment on the stabilization of premixed CH_4 -Air flames in Porous Media Burners (PMBs) is presented. Flame stabilization is analyzed via direct flame front tracking, which is made possible by a novel experimental apparatus. The use of additive manufacturing for computer-generated topologies allows making optically-accessible PMBs featuring see-through directions. This methodology also enables topology tailoring which is here exploited to study the influence of porosity on burner's performance. Flame front tracking reveals a different stabilization trend in highly H₂-enriched flames. A comparison with a theoretical model is used to remove the effect of preheating and focus on other fuel properties. This suggests a flame-speed enhancement mechanism driven by Lewis number effects in Le < 1 mixtures. Together with recent 3D Direct Numerical Simulations, these results provide evidence that preferential diffusion effects are key in the stabilization of flames in PMBs. These phenomena, not considered in state-of-the art 1D-Volume Averaged Models, remain crucial for the design of efficient PMB using hydrogen as a fuel.

Keywords: Flame Front Tracking; Additive Manufacturing; Hydrogen; Optical Access; Stabilization Mechanisms

Information for Colloquium Chairs and Cochairs, Editors, and Reviewers

1) Novelty and Significance Statement

The novelty of this research is the development of a pioneering experimental setup that provides direct flame front visualization in Porous Media Burners (PMBs). As a result, the flame front can be tracked, enabling a detailed analysis of flame stabilization mechanisms in CH_4 -Air and H_2 -enriched flames.

It is significant because it reveals the existence of stabilization mechanisms in H_2 -enriched flames that are essentially different from those of CH_4 -Air flames and are not taken into account in current models for combustion in inert porous media. An appropriate characterization of H_2 flames in PMBs will require the incorporation of these mechanisms.

2) Author Contributions

- E. F. M. Investigation, Software, Data curation, Visualization, Formal analysis, Writing Original Draft
- P. A. M. Conceptualization, Formal analysis, Writing Review & Editing
- T. S. Conceptualization, Formal analysis, Writing Review & Editing, Supervision, Funding acquisition
- L. S. Conceptualization, Formal analysis, Writing Review & Editing, Supervision, Funding acquisition

3) Authors' Preference and Justification for Mode of Presentation at the Symposium

The authors prefer **PPP** presentation at the Symposium, for the following reasons:

- Novel experimental setup: optically-accessible PMBs with see-through directions benefit visually from a poster's physical form.
- Complex background information: the paper relies on significant background information and requires a broader contextualization (experimental setup, theoretical model...).
- One-to-one level discussion will help the understanding and arise enriching discussions.

1 1. Introduction

The storage of excess renewable energy in the form 2 of hydrogen is a viable path for the reduction of CO_2 з emissions. The subsequent use of H₂ as a fuel re-4 quires addressing specific challenges such as flame 5 stabilization and NOx mitigation. In the present 6 work, heat-recirculating burners are considered for 7 their ability to address these issues. In such burners, 8 energy from the burnt gases is transported towards 9 the fresh reactants to preheat them. Thanks to pre-10 heating, flame stabilization is enhanced and ultra-lean 11 12 combustion is attainable. This enables low flame temperatures thus abating NO_x emissions. Porous Media 13 Burners (PMB) are heat-recirculating devices where 14 a flame is stabilized within the cavities of a porous 15 matrix. Owing to their outstanding properties [?], 16 PMBs are well suited to achieve a low-NOx carbon-17 free flame via lean hydrogen combustion. Surpris-18 ingly, few works have addressed the combustion of 19 pure hydrogen-air in PMBs [? ?]. Other studies 20 21 have considered the effects of hydrogen enrichment on canonical fuels such as methane and natural gas [? 22 ?] and on alternative energy vectors such as ammo-23 nia [??]. Hydrogen addition was found to have a re-24 markable impact on burner operation, shifting the sta-25 bility map towards lower equivalence ratios [?]. The 26 aforementioned studies focus on the evolution of the 27 burner operating range and report changes in the tem-28 perature profiles and in pollutants emissions. How-29 ever, they do not specifically address the stabilization 30 mechanisms of these flames and how they are affected 31 by hydrogen blending. Recent DNS have shown that 32 preferential diffusion effects play a key role in the sta-33 bilization of H₂ flames in PMBs [??]. This effect is 34 not specifically taken into account in 1D models such 35 as those used for PMBs [?]. 36

Despite the intrinsic complexities associated to H_2 , 37 even with conventional fuels the modeling of porous 38 media combustion is challenging [?]. So far, the 39 use of opaque reticulated foams in experimental stud-40 ies has hindered direct optical access to the interior 41 of the porous matrix. Recently, [?] achieved a 3D 42 reconstruction of the temperature field in both phases 43 44 of a PMB by combining X-ray μ CT and infrared thermometry. In [?], CH^{*} chemiluminiscence was used 45 to study the local structure of CH₄-Air flames sta-46 47 bilized in a two dimensional array of cylinders with a staggered configuration. Despite these and previ-48 ous remarkable works [? ? ?], the use of non-49 intrusive techniques and laser diagnostics in the liter-50 ature is scarce. Moreover, the geometrical parameters 51 of the porous structure are known to have a direct im-52 pact on the combustion properties [? ?]. However, 53 the influence of the matrix porosity, ϵ , is rarely as-54 sessed in experimental studies. Generally, variations 55 56 of ϵ are small and in most cases are a side effect of the utilization of different foams rather than a controlled 57 parameter. Recently, several works have considered 58 computer defined geometries built via Additive Man-59 ufacturing (AM) [? ? ?]. This methodology allows 60

for topology tailoring and has been applied to enhance the burner operating range via spatial gradation of the topological parameters [?]. Here, this approach is used to produce PMBs with see-through directions. As a result, optical access to the interior of the matrix is granted and direct probing of the flame front is enabled. Computer designed geometries allow the independent variation of pore size, d_p and void fraction, ϵ . Here, we address the influence of porosity on the burner performance by comparing two PMBs with the same pore size and different porosities, $\epsilon = 0.5$ and $\epsilon = 0.8$. The influence of hydrogen enrichment on the burner operating domain and on flame stabilization is also analyzed. The tracking of the flame position as a function of the operating conditions allows us to compare the stabilization of CH₄-Air flames and H₂enriched flames. For these flames, our results indicate stabilization mechanism other than heat recirculation, which are not accounted for in present 1D models.

2. Experimental setup & procedure

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A sketch of the experimental setup is presented in Figure ??. Hydrogen, methane and air mass flow rates, denoted by $\dot{m}_{\rm H_2}$, $\dot{m}_{\rm CH_4}$ and $\dot{m}_{\rm Air}$, are adjusted using Bronkhorst ELFLOW and Vöetglin Redy smart mass flow controllers. Premixed reactants go through a convergent section and a sintered steel plate to homogenize the flow. To mitigate the thermal drift, a cooling circuit sets the wall temperature at the straight duct section to $T_w = 293$ K. During operation, the temperature of the sintered steel plate increases due to the radiative heat exchange with the PMB base. Consequently, the incoming flow is mildly preheated. Two K-type thermocouples with different bead sizes are used to measure the reactants' inlet temperature T_{in} right before they enter the porous matrix. Following [?], these two temperature measurements are combined to override the effect of radiation on the thermal balance of the thermocouple. After this correction, the uncertainty in the inlet temperature measurement is $\delta T_{in} \simeq 10 \text{ K.}^1$ This value is later used to compute the one-dimensional laminar flame properties for each operating point. For optical access, the cylindrical porous matrix is placed inside a quartz tube with inner diameter D = 50 mm and supported by an internal quartz with a smaller diameter. The burner height is L = 30 mm and its width is adjusted to obtain a porous-to-quartz gap smaller than the pore size at room temperature. This enables assembling and prevent breakage by thermal expansion during operation. The x axis is oriented in the streamwise direction and the origin is at the burner inlet. Throughout this paper, this coordinate is often expressed in nondimensional form by normaliz-

¹This value was obtained applying a standard procedure for uncertainty quantification: $\delta f(x_i) = (\sum (\partial x_i f \cdot \delta x_i)^2)^{1/2}$. We retained the contributions of the temperature measurements, the mass flow rate and the bead emissivity.



ing with the burner length, $\xi = x/L$. Additive Man-114 ufacturing (AM) is used to fabricate a PMB made of 115 316L stainless steel. This alloy exhibits a notable re-116 sistance to high temperature corrosion and its fusion 117 temperature is about 1650 K. In our PMBs, the fluid-118 solid interphase is described by a Diamond-Triply Pe-119 riodic Minimal Surface (Diamond-TPMS) defined by 120 the implicit equation: 121

$$s(kx)s(ky)s(kz) + s(kx)c(ky)c(kz) +$$
(1)
$$c(kx)s(ky)c(kz) + c(kx)c(ky)s(kz) = t$$

In Eq. (??), $s(\cdot)$ and $c(\cdot)$ stand for the sine and co-122 158 sine functions while $k = 2\pi/\Lambda$ and t denote the lat-123 159 tice wavenumber and threshold respectively. These 124 parameters, k and t, can be modulated to adjust the 125 161 porosity, ϵ , and the pore size, d_p , of the structure. 162 126 127 When computer-defined geometries are used, an ex-163 act representation of the topology is available through 164 128 a CAD. Therefore, an accurate evaluation of the topo-129 165 logical parameters of the burner can be performed 130 166 thanks to several off-the-shelf tools such as PORESPY 131 167 and OPENPNM. These parameters, including the solid 132 and gas phase tortuosities, τ_s and τ_q , and the inter-133 169 phase volumetric surface, S_v , are gathered in Tab. ??. 134 170

Table 1: Topological parameters of the PMBs.

a_p [mm]	ϵ	$[m^{-1}]$	$ au_g$	$ au_s$	
2.5	0.5	659	1.58	1.58	
2.5	0.8	729	1.19	2.52	

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The resulting geometries offer various visual path-179 136 ways in certain directions. These see-through di-180 137 rections provide optical access to the interior of the 138 181 porous matrix and hence direct visualization of the 182 139



Fig. 2: CH* chemiluminiscence with TC lens (A) versus conventional imaging (B).

flame front. In addition, they allow the insertion of thermocouples from the sides to measure the temperature in the burner. Temperature probes of type N with a sheath diameter of 0.5 mm are used to reconstruct the temperature profile along the symmetry axis of the burner. Inside the burner, thermocouples are subjected to a significant radiative flux from the solid matrix. This alters their thermal balance causing their temperature to lie somewhere between the solid and the gas temperature. A Lavision Imager SC-MOS camera is aligned with one of the horizontal seethrough directions of the porous matrix and used to visualize the flame. To maximize the signal intensity, the exposure time is set to 2000 ms. Due to perspective distortion, if a conventional optics is used, only the visual pathway aligned with the camera direction is completely cleared. To circumvent this issue, the camera is fitted with an Opto Engineering Telecentric lens TC4M-64. Telecentric (TC) lenses remove perspective distortion providing an orthogonal projection of the objects. Thanks to TC lens, we obtain a cleared optical access for all the visual pathways aligned with the exploration see-through direction. The image resolution is 2560×2160 px and with the TC optics the typical scale is about 23 μ m per pixel. Because of the elevated temperatures reached during operation, the porous matrix radiates in the low visible range. To isolate the flame front region, the camera mounts an Edmund Optics bandpass filter with a central wavelength of 430 nm and a bandwidth of 10 nm. This allows collecting the line-of-sight integrated emission of the of the CH* radical. Figure ?? shows a comparison between flame visualization with a conventional imaging system and with the present optical setup.

The burner operating conditions are defined by the global equivalence ratio, ϕ , the load expressed in terms of thermal power, $P = \sum_f \dot{m}_f Q_f$, with $f = H_2$, CH₄, and the hydrogen power fraction, $\alpha_P = \dot{m}_{\rm H_2} Q_{\rm H_2} / P$. Here, Q_f denotes the fuel Lower Heating Value. The mean absolute uncertainty in the equivalence ratio is $\delta \phi = 0.008$ and the relative uncertainties in δP and $\delta \alpha_P$ are smaller than 3% and 4%, respectively. In this work, the focus is on

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the submerged combustion regime, where the flame is 183 stabilized inside the PMB and there is an intense cou-184 pling between combustion and heat transfer. How-185 ever, submerged flames cannot be stabilized directly 186 from cold start, a transient starting process is required 187 to achieve this regime. The starting of PMBs is an 188 overlooked topic in the literature. Generally, flames 189 are first stabilized on top of the burner in the sur-190 face combustion regime. When multi-staged config-191 urations are used, the transition from surface to sub-192 merged combustion is straightforward. Usually, ce-193 ramic foams with a small pore size are used for the up-194 195 stream stages. This way, the upstream porous matrix acts as flame arrestor, quenching the flame and ensur-196 ing stability during the process. However, in single-197 staged configurations the transition from surface to 198 submerged combustion is challenging. In CH₄-Air 199 flames, the progressive reduction of the bulk to flame 200 speed ratio often causes Flame Repetitive Extinction 201 and Ignition (FREI) and is likely to result in flash-202 back. Ceramic porous matrices feature a low resis-203 204 tance to thermal shocks [? ?]. As a result, the burner operating conditions must be varied extremely 205 slowly to prevent breakage. This imposes long tran-206 sient times, further increasing the complexity of the 207 starting process. Here, we propose a time-efficient 208 method to achieve this transition that is made possi-209 ble by the use of metallic burners. The procedure is 210 detailed down below. 211

First, a CH₄-Air flame is stabilized on top of the 212 213 burner in the surface combustion regime. Typically, the burner is started with a 0.8 kW CH₄-Air flame 214 at $\phi = 0.6$. Then, to achieve submerged combustion 215 conditions, a small fraction of hydrogen is temporar-216 217 ily added so as to pull the flame inside of the porous matrix. For $\epsilon = 0.8$ matrices, an increase of the hy-244 218 drogen content to $\alpha_P = 20$ % generally suffices to 245 219 trigger the transition. The flame front is tracked live 246 220 as it performs a controlled flashback across the porous 247 221 matrix (see Fig. ??). In the sequence of Fig. ?? snap-222 248 shots are labeled with the elapsed time from the start 249 223 of the transient phase. Note that a relatively fast tran-224 250 sition can be achieved without matrix breakage. The 225 251 enhanced thermal shock resistance has proven to be a 226 252 major advantage of metallic PMBs when compared to 253 227 their ceramic counterparts. Their improved durabil-228 254 ity largely simplifies burner operation by reducing the 255 229 risk of fracture and hence the laboratory time. Contin-230 uous surveillance is needed during the transient phase 257 231 in order to prevent a flashback once the flame is close 258 232 to the inlet. When the flame reaches the bottom of the 259 233 burner, hydrogen content, α_P , power, P, and equiva-234 260 lence ratio, ϕ are re-adjusted and set to a known stable 235 261 operating point. The system is then allowed to reach 262 236 thermal equilibrium. 237 263

The exploration of the burner operating range is 264 performed by varying the mixture equivalence ratio, 265 ϕ , at constant power, P, and hydrogen content, α_P . 266 Close to flashback and blowoff limits, the equivalence 267 ratio is varied in steps of $\Delta \phi = 0.01$. A step of equivalence ratio of $\Delta \phi = 0.02$ is used elsewhere. For 269



Fig. 3: *Controlled flashback* transient. (A) Ignition with CH_4 -Air at $\phi = 0.6$. (B) Hydrogen addition to trigger transition to submerged combustion. (C) Removal of hydrogen. (D) Steady state in thermal equilibrium is reached.



Fig. 4: Procedure to compute the axial profile of CH^{*} and the flame position ξ_f .

CH₄-Air flames, the burner operation at high equivalence ratios was limited by overheating rather than by flashback. The thermal limit of the material is the major drawback of metallic PMBs. When the operating point of the burner is changed, temperature measurements are used to assess thermal equilibrium. Convergence is assumed when the maximum temperature variation over two minutes is less than 1 K for all thermocouples. For each experimental point, thermocouple and mass flow controller readings are recorded and averaged for two minutes. In parallel, an image of the flame inside the burner is taken. From images, the transversely averaged $CH^{*}(x)$ profile is computed, which is assumed to be proportional to the heat release rate and it is used to compute the flame position inside the PMB. This allows tracking the flame as a function of the operating conditions. The procedure to compute the flame position from images is schematized in Fig. ??. The images are first masked using a backlight mask. This also removes direct reflections in the solid matrix. The axial profile is obtained by computing the surface integral of CH^{*} intensity at each row of pores. The resolution of this raw profile is given by the spacing between the pores. It is equal to half the period of the lattice, $\Lambda/2$, and defines the uncertainty in the estimation of the flame

position, δx_f . The experimental profiles are then fit-270 ted by a beta-prime distribution for subgrid interpo-271 lation. The position of the flame, x_f , is taken as the 272 location of the function maximum. For each operat-273 ing point, the measured inlet temperature, T_{in} and the 274 mixture composition defined by α_P and ϕ are used 275 to compute the properties of the corresponding pre-276 mixed laminar flame: the laminar burning velocity, 277 S_{L_0} , the adiabatic flame temperature, T_{ad} , the flame 278 thickness, $\delta_{T_0} = (T_{ad} - T_{in}) / \max(\partial T / \partial x)$, the inlet gas density, ρ_{in} , and the Zeldovich number, Ze. 279 280 More specifically, the Zeldovich number is computed 281 as, $\text{Ze} = E(T_{ad} - T_{in})/RT_{ad}^2$, where R is the uni-282 versal gas constant and E is the activation energy, cal-283 culated from: 284

$$E = -2R\partial [\ln(\rho_{in}S_L)]/\partial(1/T_{ad})$$
(2)

The derivative in Eq. (??) is computed from two one-285 dimensional flames by performing small variations 286 in the N₂ molar fraction while keeping the equiva-287 288 lence ratio and inlet temperature constant. This calculations are performed using CANTERA and the de-289 tailed kinetic scheme UCSD (57 species and 268 re-290 actions) with mixture-averaged transport coefficients. 291 Finally, temperatures are normalized as $\theta = (T - T)^2$ 292 $T_{in})/(T_{ad}-T_{in}).$ 293

294 3. Results & Discussion

Figure ?? depicts three 1 kW CH₄-Air flames with 295 distinct equivalence ratios stabilized in the $\epsilon = 0.8$ 296 PMB. Post-treated images are presented in the left 324 297 column, where the interpolated flame position is in-298 325 dicated with a white dotted line. On the right, the 326 299 experimental profiles of CH* and their beta-prime fit 327 300 are displayed in red dotted and black dashed lines, 328 301 respectively. Additionally, solid blue lines represent 329 302 the normalized temperature profiles acquired via the 303 330 thermocouples. Near the flashback limit, flames sta-304 bilize close to the burner inlet, as illustrated in the 332 305 lower plot of Fig. ??. As the equivalence ratio is de-333 306 creased, flames gradually shift downstream, stabiliz-334 307 ing at higher $\xi = x/L$ values. At the blowoff limit, 335 308 top plot of Fig. ??, flames stabilize approximately in 336 309 the middle of the burner. Note that the flame front is 337 310 reasonably flat across the transverse direction, which 338 311 suggests that a one-dimensional model is a reasonable 339 312 first approximation. 340 313

314 3.1. Influence of porosity

The influence of porosity on the PMB performance 344 315 345 is now examined for CH₄-Air mixtures. Figure ?? 316 shows the experimental points in the $P - \phi$ map for ³⁴⁶ 317 the two burners considered. The blowoff and high- 347 318 temperature limits are marked with dashed and dash-348 319 dot lines, respectively. The shaded region outlines 349 320 the estimated boundaries of the burner's operating do-350 321 main. Dashed areas in the upper right corners denote 322 351 unexplored regions due to thermal constraints of the 352 323



Fig. 5: A) Line-of-sight integrated CH* signal for three 1 kW CH₄-Air flames at different equivalence ratios. B) Profiles of CH* and normalized temperature, θ , as a function of the axial position, $\xi = x/L$.

material. In these zones, the elevated temperatures reached inside the PMB can lead to severe deformation of the porous structure. Figure ?? shows that the stability domain of the burner is directly affected by ϵ . When the porosity lowered from $\epsilon = 0.8$ to $\epsilon = 0.5$, the operating domain is shifted towards higher equivalence ratios. This can be explained by an analysis of the kinematic balance between the flow velocity, u_B , and the local flame speed, S_L . When the porosity is reduced, the effective burner section is reduced by the same factor and velocities inside the porous matrix increase for an imposed mass flow rate. Therefore, in order to ensure a kinematic balance between the flow and flame velocities, higher values of S_L are required. The local flame speed can be enhanced either by increasing the recirculation efficiency, η_{rec} , or by operating at larger equivalence ratios, ϕ . Ceteris paribus, recirculation efficiency decreases with porosity [?]. However, porosity variations also drive changes in other parameters such as the specific exchange surface, S_v , which directly scales the heat exchange term, h_v . Thus, predicting its influence is not straightforward in practice.

In PMBs, heat recirculation enables increasing the mass consumption rate per flame unit surface. This increase is quantified by the flame speed-up, defined as the ratio between the bulk velocity in the PMB, computed as $u_B = \dot{m}/(\rho_{in}A\epsilon)$, and the laminar burning velocity at inlet conditions, $S_{L_0}(\phi, T_{in})$.

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Fig. 6: Operating domain spanned by power, P, and equivalence ratio, ϕ , for CH₄-Air flames in PMBs with different porosities $\epsilon = 0.5$ and $\epsilon = 0.8$.



 $\xi_f = x_f/L$, at various powers for two porosities, ϵ .

Here, \dot{m} and $A = \pi D^2/4$ denote the total mass flow 412 353 rate and the burner section, respectively. In finite 413 354 PMB, the speed-up is a function of the position of the 355 414 flame in the burner [?]. Our novel experimental setup 356 415 357 allows measuring the evolution of the speed-up with 416 the flame position, x_f , and comparing it to theoretical 417 358 predictions. Figure ?? displays the results for CH₄- 418 359 Air flames with different powers for the two burners 419 360 under investigation. For $u_B/S_{L_0} \simeq 1$, flames stabi- 420 361 lize in the vicinity of the burner inlet, $\xi = 0$. Due to 421 362 heat losses, some of these flames can feature a speed-422 363 up smaller than unity. Unfortunately, the overheating 423 364 limit of the burner was here reached before flashback 365 424 occurrence and thus the u_B/S_{L_0} at flashback is not 425 366 available for $ext{CH}_4 ext{-Air flames.}$ When u_B/S_{L_0} is in- 426 367 creased, the flame is progressively pushed towards the 427 368 369 burner outlet. u_B/S_{L_0} is maximum around $\xi_f \simeq 0.5$ and a further increase in the bulk to laminar flame 429 370 speed ratio leads to blowoff. At the blowoff limit, the 430 371 flame speed-up rises to approximately $u_B/S_{L_0} \simeq 7$ 372 431 in both burners. For the limited power range reported 432 373

here, the burner load does not affect the flame stabi-374 lization in terms of $u_B/S_{L_0}(\xi_f)$. Overall, ϵ is found 375 not to alter the speed-up at blowoff but it has an im-376 pact on the shape of the $u_B/S_{L_0}(\xi_f)$ curve. In pe-377 riodic structures, local anchoring effects can result 378 in discrete jumps of the flame front position. In the 379 $\epsilon = 0.5$ PMB, the discretization of the flame po-380 381 sition, ξ_f , is more relevant and results in a different shape of the speed-up stabilization curve. This 382 explains the larger spreading of experimental points 383 around $\xi_f \simeq 0.2$ in the $\epsilon = 0.5$ burner. 384

385 3.2. Influence of H₂-enrichment

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The influence of hydrogen enrichment on flame stabilization is now discussed. The operating domain of the $\epsilon = 0.8$ burner spanned by Power, P, and equivalence ratio, ϕ is displayed in Fig ??a. The blowoff, flashback and high temperature limits are indicated with dotted, solid and dashed lines, respectively. For the sake of clarity, no shadowed regions are included here. Different markers are used for each hydrogen power fraction, α_P . Note that the $\alpha_P = 0\%$ points in Fig. ??a are those of Fig. ?? with $\epsilon = 0.8$. Empty markers denote operating points where the laminar burning velocity S_{L_0} is lower than $0.5 \,\mathrm{cm} \,\mathrm{s}^{-1}$. For those operating points, combustion in the absence of heat recirculation is virtually impossible. Safety considerations make of combustion beyond flammability limit a major asset of PMBs for practical applications. Hydrogen content has a first order impact on the operating equivalence ratios. As shown in Fig. ??a, the operating domain of the burner is shifted towards lower equivalence ratios for increasing values of α_P . As a rule of the thumb, the operating equivalence ratio ϕ is reduced by $\Delta \phi = 0.1$ every time the hydrogen power fraction is doubled. In the $\epsilon = 0.5$ PMB (not shown here), operating domains are slightly shifted towards higher equivalence ratios but the overall behavior is the same. In Fig. ??b, the speed-up u_B/S_{L_0} is plotted against the flame normalized position for different powers and hydrogen contents, α_P . The addition of hydrogen has a direct impact on the flame stabilization. A H2-enrichment of $\alpha_p = 20 \%$ results in an upward displacement of the stabilization curve, but the overall behavior remains roughly the same. A larger increase in the hydrogen content to $\alpha_P = 40 \%$ drives a substantial change in the stabilization trend of the flame. For increasing u_B/S_{L_0} ratios, highly H₂-enriched flames remain attached to the burner inlet, anchoring to the first row of pores at roughly $\xi_f \simeq 0.05$. Then, when a critical speed-up is reached, these flames are blown off the burner. As the H₂ content is increased, the stabilization curve $u_B/S_{L_0}(\xi_f)$ becomes very stiff. This is a significantly different behavior than that usually predicted by Volume Averaged Models (VAMs) and theory and it was also observed in 3D DNS [? ?]. Note that the mass consumption rate per flame unit surface can rise to nearly $u_B/S_{L_0} \simeq 140$ for $\alpha_P = 80$ %. The $u_B/S_{L_0}(\xi_f)$ stabilization curves



Fig. 8: A) Operating domain of the $\epsilon = 0.8$ burner for different hydrogen power fractions, α_P , in the $P - \phi$ space. B) Flame speed-up, u_b/S_{L_0} , versus normalized flame position, $\xi_f = x_f/L$, for different powers and hydrogen contents.

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433 display a similar trend in the $\epsilon = 0.5$ burner.

476 Hydrogen flames feature larger Zeldovich num-434 bers and are thus more sensitive to preheating than 477 435 478 methane flames. However, in H₂-enriched flames, the 436 different stabilization trends observed in Fig. ??b sug-437 480 gest the existence of flame speed enhancement mech-438 anisms other than preheating. To dig deeper into this 481 439 482 question we compare the flame mass consumption 440 483 rate observed in experiments, u_B , to the flame speed-441 484 up predicted by the asymptotic model of [?], S_{L_P} . 442 In this theoretical model, heat recirculation is the sole 485 443 flame-stabilization mechanism and mixture sensitive-486 444 ness to preheating is accounted for via the Zeldovich 445 488 number. Hence, variations in u_B/S_{L_P} with the hy-446 drogen content hint towards the existence of other 489 447 stabilization mechanisms. To compute S_{L_P} we im-490 448 491 449 pose the flame position observed in experiments, ξ_f 492 and evaluate the theoretical model. Model param-450 eters are computed as follows: topology dependent 493 451 parameters are extracted from Tab. ??; gas constants 494 452 such as density, ρ_{in} , specific heat, c_{p_g} , thermal con-495 453 454 ductivity, λ_q , and viscosity, μ , are evaluated at the 455 fresh gases temperature, T_{in} , and computed using the UCSD kinetic scheme; the interphase heat transfer 498 456 coefficient is $h_v = S_v \lambda_g \text{Nu}/d_p$, where the Nusselt 499 457 500 number Nu is calculated using the correlation pre-458 sented in [?] and the pore-based Reynolds and 501 459 Prandtl numbers are defined as, $Re = \rho u_B d_p / \mu$ and 502 460 $Pr = c_{pg} \mu / \lambda_g$, respectively. A solid thermal con-503 461 ductivity of $\lambda_s = 13 \text{ W m}^{-1} \text{ K}^{-1}$ is assumed and ra-504 462 diative thermal losses at the burner ends are taken into 463 account. In this asymptotic model [?], different ex-464 507 pressions can be used to evaluate the flame speed ra-465 508 tio as a function of heat recirculation $S_{L_P}/S_{L_0}(\eta_{\text{rec}})$. 466 509 Here, the speed-up correlation developed by [?] is 467 510 chosen. 468 511

Results are plotted in Fig. ?? as a function of the 469 470 hydrogen molar fraction in the fuel, χ_{H_2} , for both 513 burners. For each hydrogen content, u_B/S_{L_P} is aver-471 514 aged over the whole dataset and the vertical lines cor-472 515 respond to the maximum and minimum values. For 473 516 474 CH₄-Air flames, the u_B/S_{L_P} ratio should be equal

to one but the obtained values are about 0.7. This means that speed-ups measured in the experiment are smaller than those predicted by the asymptotic model. These differences can be attributed to the numerous simplifications made in the derivation of the asymptotic model. Strong assumptions such as constant transport coefficients and zero radial heat losses hinder quantitative predictions. However, these modeling limitations are expected to affect all flames rather equally, regardless of the fuel composition. Despite the fact that the model does not include radial heat losses, it does account for in/outlet losses by radiation. Switching them on/off only shifts the curves in Fig. ?? downwards by 2 - 7% but does not affect the trends. Thus, comparing u_B/S_{L_P} ratios for different hydrogen contents the error should remain constant. In other words, if preheating is the sole mechanism responsible for flame speed-up, the u_B/S_{L_P} ratio should be independent of the mixture composition. Figure ?? reveals that the ratio between the observed and the predicted speed-up increases with the hydrogen molar fraction. Despite the significant dispersion in the results, the trend is clear and this supports the idea that a mechanism different from preheating affects flame speed-enhancement in H₂enriched flames. Note that in Fig. ??, the u_B/S_{L_P} ratio is also mildly affected by the porosity ϵ . This is because its influence on the speed-up is not quantitatively captured by the model. Changing the porosity affects other topological parameters that have a strong impact on h_v . Thus, the influence of porosity on simulations is, to a certain extent, encapsulated in h_v . The evaluation of the asymptotic model with different off-the-shelf correlations for the Nusselt number affects the predicted speed-ups yet it does not alter the conclusions drawn from Fig. ??. For the sake of comparison, Fig. ?? shows the theoretical speed-up curves for 1kW CH₄-Air flames in the two burners with different porosity ϵ . The comparison between Figs. ?? and ?? confirms the overall qualitative agreement despite the over-prediction by the model. It has been checked that the kinetic mechanism is not at the ori-







Fig. 10: Influence of ϵ on the burner speed-up in CH₄-Air flames. 573

gin of these discrepancies, which are very likely due 576 517 to the underestimation of heat losses in the theoretical 577 518 model. 519

For non-unity Lewis-number fuels such as H₂, 520 578 preferential diffusion effects are known to play a ma-521 jor role in flame stabilization [?]. Recent 3D-DNS [? 522 ?] revealed that the formation of locally enriched 523 580 pockets caused by flame-front curvature has a strong 581 524 impact on flame stabilization in PMBs. Therein, 525 582 it was shown that hydrogen flames can accommo-526 date larger mass flow rates by stretching and increas-583 527 ing their surface without moving downstream. Here, 528 flame front tracking has revealed a different stabiliza-529 tion trend in highly H_2 -enriched flames. Then, the 585 530 influence of Zeldovich number has been removed via 586 531 normalization by the theoretical flame speed S_{L_P} . 587 532 Variations in the u_B/S_{L_p} ratio with hydrogen con- 588 533 tent suggest the existence of other flame speed en- 589 534 hancement mechanisms different from preheating in 590 535 536 H₂-enriched flames. These results reinforce the claim 591 that Lewis number effects must be taken into account 592 537 for an accurate evaluation of the flame speed-up. 593 538

4. Conclusion 539

In this experimental study, direct flame visualiza-540 tion in a PMB has been achieved by means of an ex-541 perimental setup where computer-generated topolo-542 gies are built via additive manufacturing to produce 543 optically accessible burners. Flame-front tracking has 544

been used to study the influence of porosity and H2enrichment on flame stabilization in PMBs. The analyzed porosity variation was found to alter the operating equivalence ratios, ϕ , but did not affect the blow-off limit in terms of speed-up, $u_B/S_{L_0}(\xi_f)$. A comparison with the theoretical model presented in [?] reveals that quantifying the influence of porosity is challenging. This is because theoretical predictions are very sensitive to the modeling of the heat exchange term, h_v , which is in turn influenced by porosity variations. Moreover, H₂ enrichment is found to shift the operating domain towards lower equivalence ratios and to alter the flame stabilization. Qualitative differences between the stabilization of CH₄-Air and H2-enriched flames suggest the existence of different stabilization mechanisms for non-unity Lewisnumber fuels. Comparison with a theoretical model allows separating the sensitivity to preheating and brings to light other stabilization mechanisms. Variations of the u_B/S_{L_P} ratio with the hydrogen content point out the existence of other flame speed enhancement mechanisms in H2-enriched flames. The present experimental results are consistent with recent observations made in 3D-DNS of premixed H2-Air flames that could stabilize and accommodate large flow rates while being anchored at the inlet. Therein, preferential diffusion effects were shown to play a major role on flame stabilization in PMBs. The present experiments provide further evidence that these phenomena need to be modeled if an accurate description of the burning rate is sought. Future work must address these modeling tasks and seek pore-level diagnostics to unveil the details of flame stabilization in PMBs.

Declaration of competing interest

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

Acknowledgments

The PhD of Enrique Flores Montoya is funded by BULANE and the Occitanie Region (Défi Clé Hydrogène Vert). Vincent Baylac (CIRIMAT) and Sebastien Cazin (IMFT) are acknowledged for their technical support with the manufacturing of the PMBs and with the optical diagnostics, respectively. The financial support of the European Research Council under the European Union's Horizon 2020 research and innovation program Grant Agreement 832248, SCIROCCO is also acknowledged.

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