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Contrail aging simulation of a supersonic aircraft

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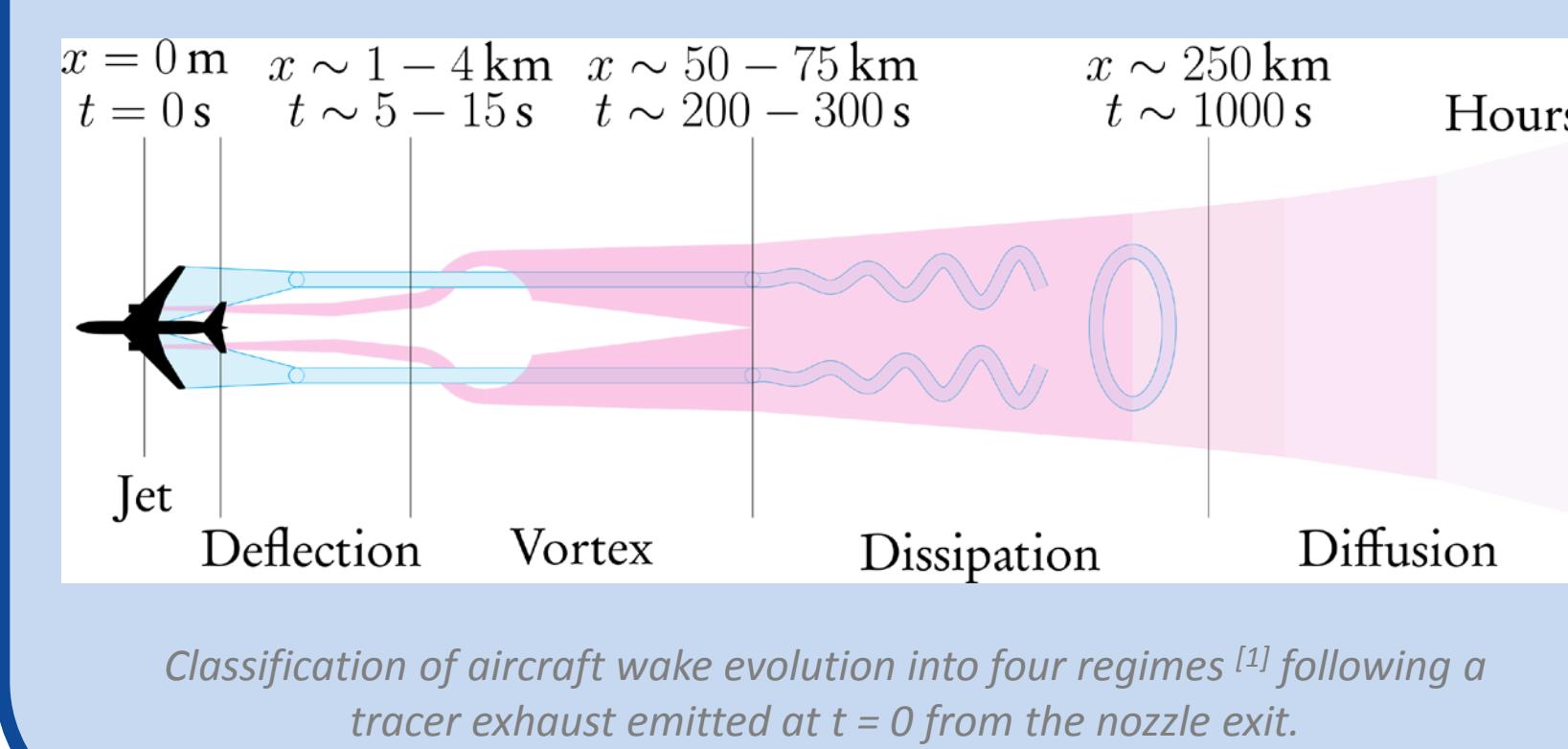
Context

SENECA project ("noiSe and EmissioNs of supErsoniC Aircraft") aims at:

- developing supersonic aircrafts;
- investigating the impact of specific supersonic engine technologies on the aircraft performance and noise;
- quantifying a range of climate impacts of supersonic aviation.

Objectives

- Model and characterize the contrails due to supersonic aircraft during cruise
- Provide data to climatologists to calculate their global impact



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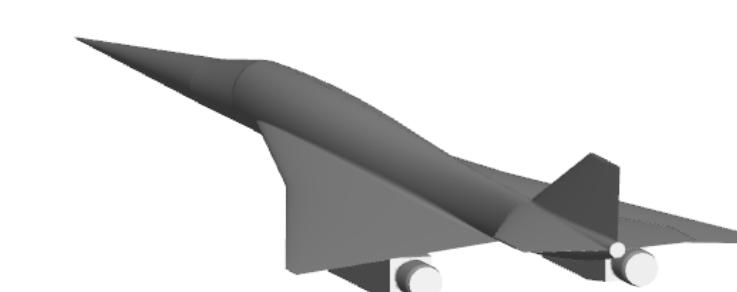
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Jet regime modeling

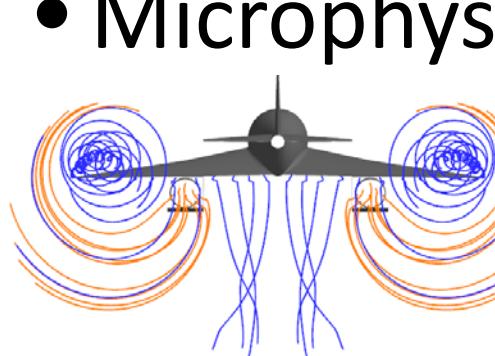
CFD RANS Modeling

- CEDRE/CHARME aerodynamic modeling
- K- ω turbulent model (SST Menter)



CAD of E19 Aeolus bizjet
(Cranfield University)

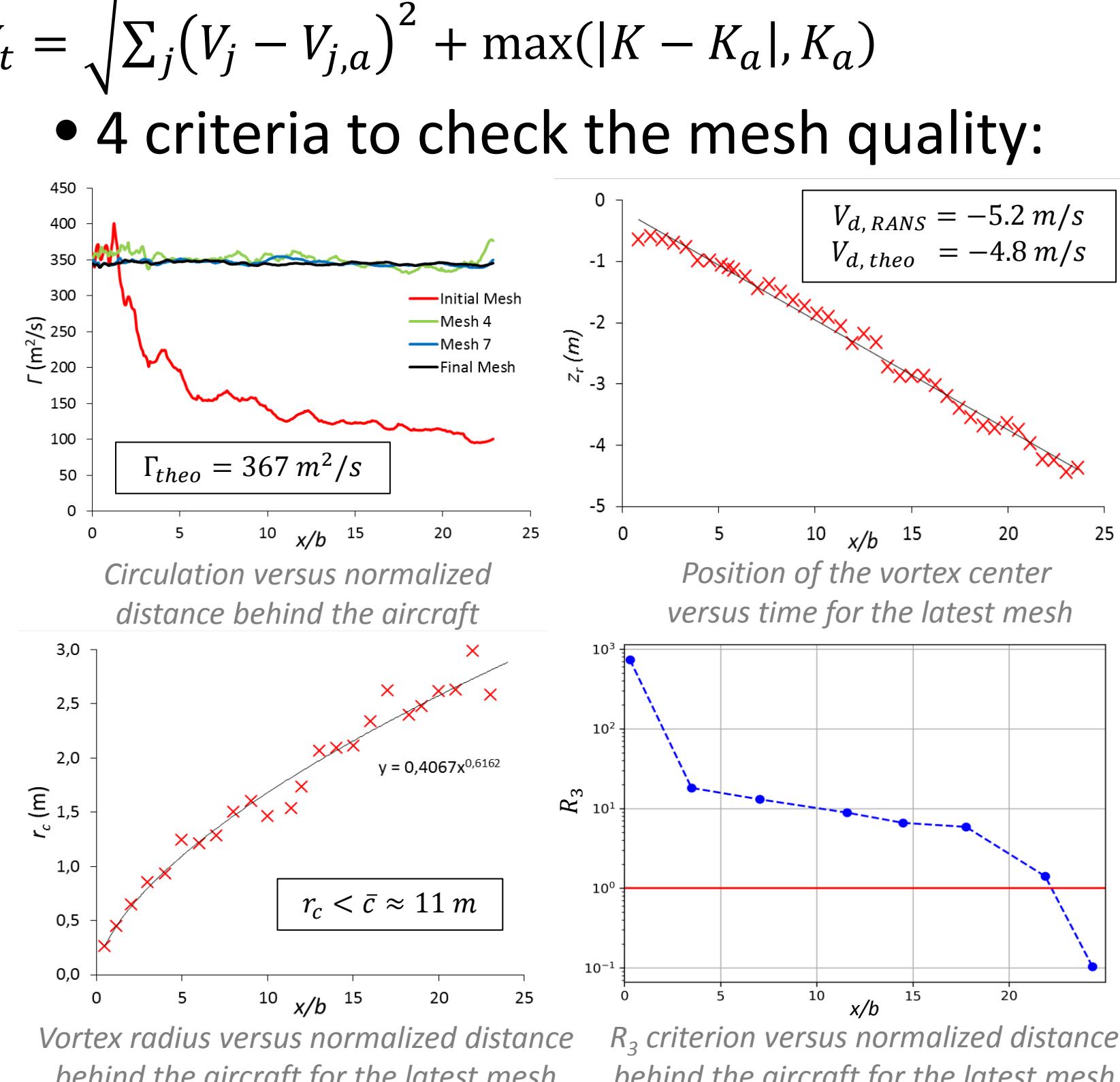
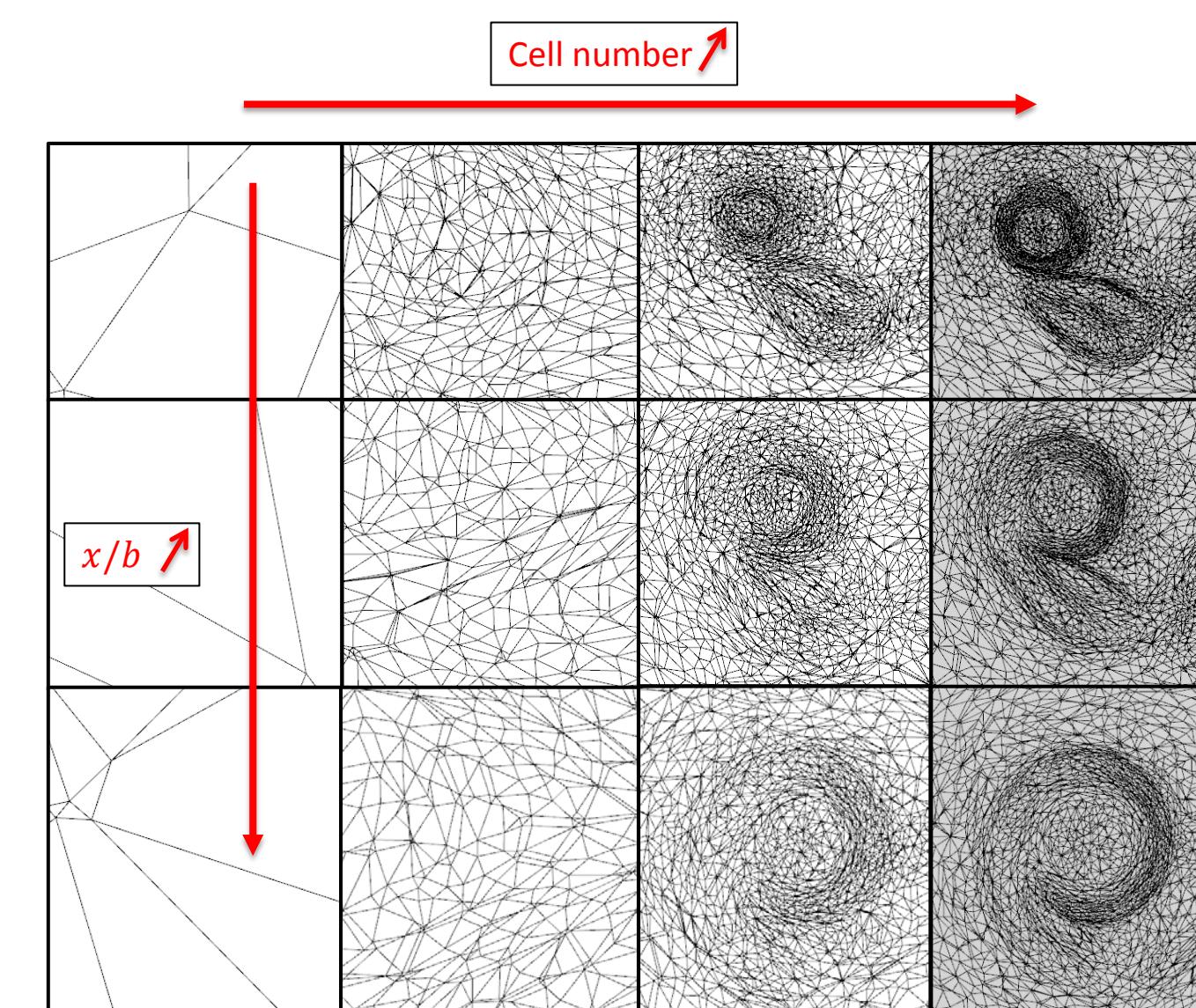
- Gas-phase chemistry^[2]
- Microphysics model^[3]



Streamlines behind the E19 bizjet colored by origin: orange for the engine exit and blue for wing

Mesh adaptation procedure (INRIA Tool)

- Variable used to refine the mesh: $V_t = \sqrt{\sum_j (V_j - V_{j,a})^2} + \max(|K - K_{al}|, K_a)$
- Mesh adaptation illustration^[4]



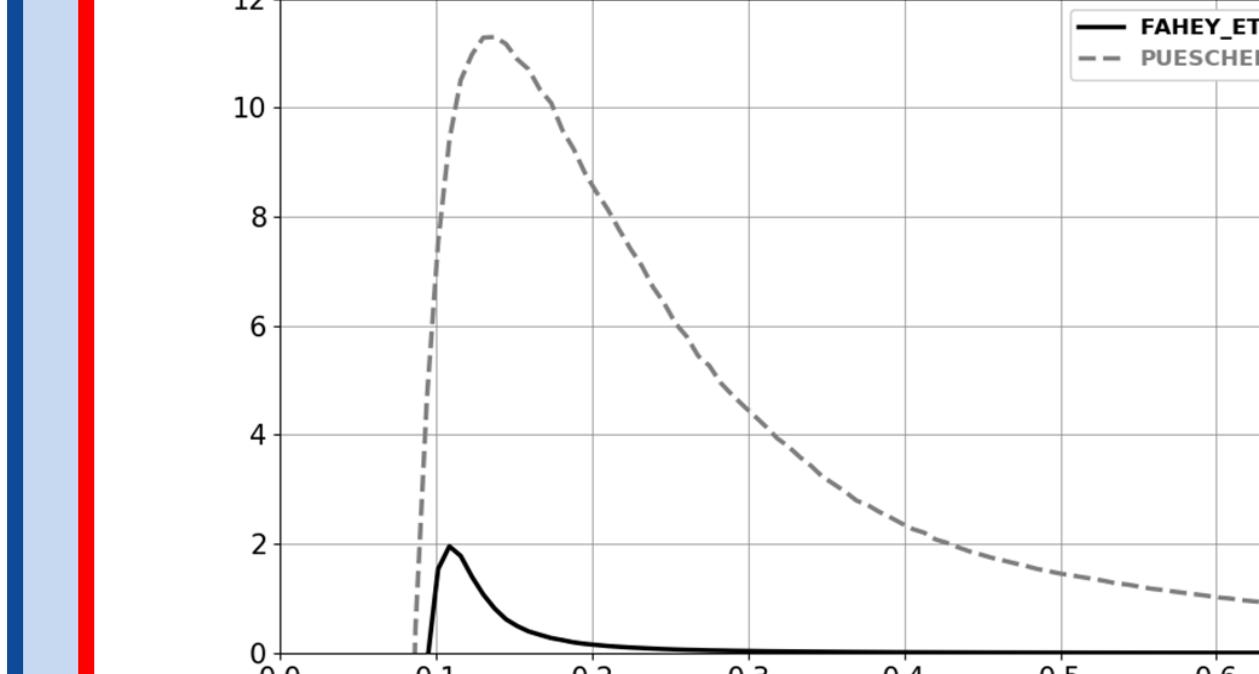
Parametric study

- Tropical conditions in lower stratosphere: $P_a = 9750 \text{ Pa}$, $T_a = 196 \text{ K}$
- Ice SuperSaturated Region (ISSR): $RH_{ice} = 100 \%$, $X_{H_2O} = 8 \text{ ppmv}$

Emission soot parameters

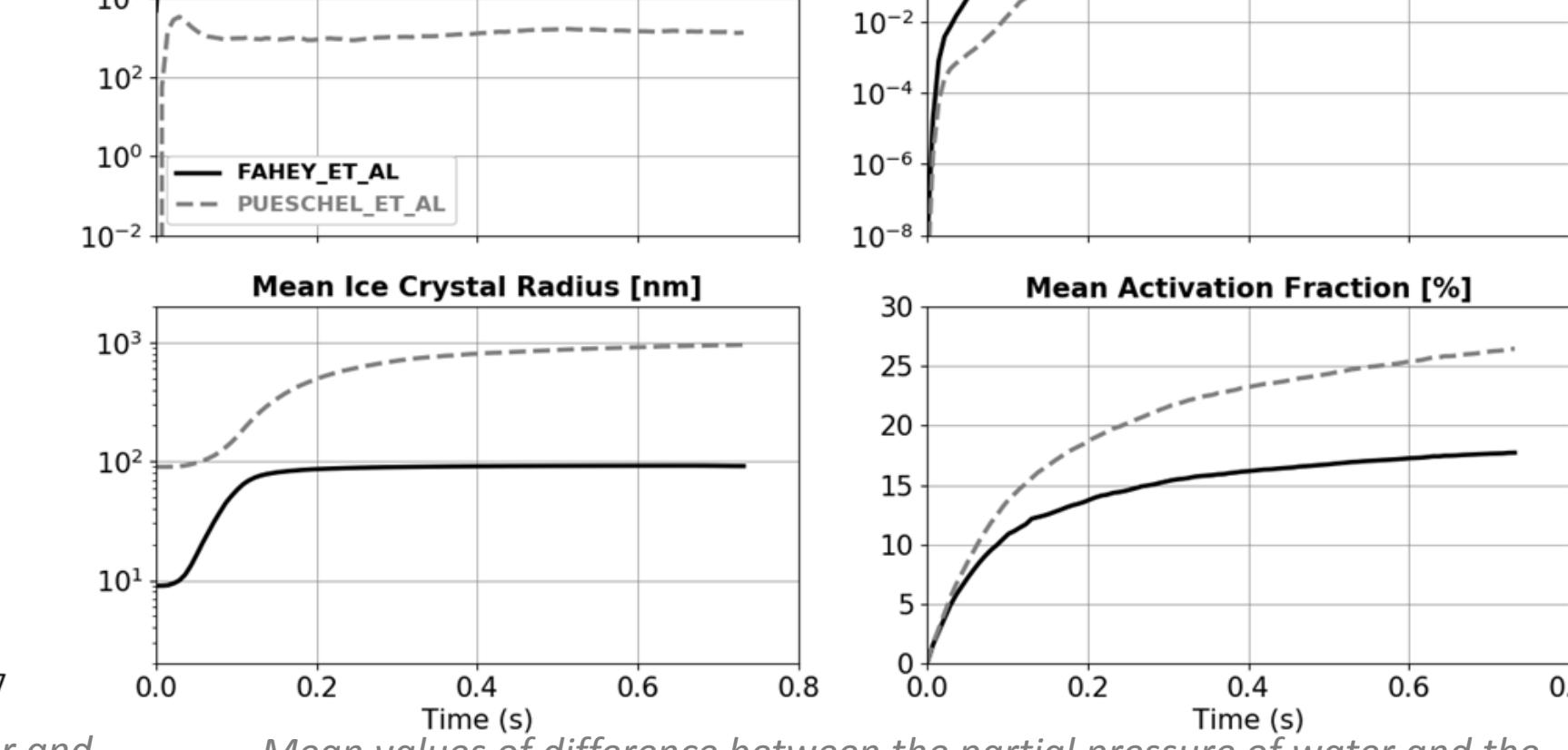
Case	r_s [nm]	ρ [kg/m ³]	N_s [#/m ³]
FAHEY ET AL. ^[5]	9	900	$4.76 \cdot 10^{14}$
PUESCHEL ET AL. ^[6]	90	1600	$2.68 \cdot 10^{11}$

P_H2O - P_{sat} sol (Pa)



Mean values of difference between the partial pressure of water and the saturation pressure over ice versus physical time behind the aircraft

Mean Ice Crystal Number [/cm³]



Mean values of difference between the partial pressure of water and the saturation pressure over ice versus physical time behind the aircraft

Vortex and Dissipation regime modeling

CFD LES Methodology^[7]

- CEDRE/CHARME aerodynamic modeling
- Smagorinsky model
- Microphysics model^[3]
- 120 s contrail aging simulated
- Characteristic vortex descent time:

$$t_{\text{descent}} = \frac{B_0}{V_{\text{descent}}} = 2.1 \text{ s}$$

- LES initialization in 5 steps:
 - Slice extraction from RANS simulation
 - LES mesh creation
 - Interpolation RANS \Rightarrow LES
 - Adding fluctuations
 - Adding atmospheric stratification

CFD LES Simulation

