

14th European Adhesion Conference (EURADH) 7th World Congress on Adhesion and Related Phenomena (WCARP) \*

10 - 13 September 2023 - Garmisch-Partenkirchen, Germany



# Visco-hyperelastic modeling of aeronautical sealant in pure bonded and hybrid bolted/bonded joints

Presented by: Co-Authors: Minh Nhat TO Éric PAROISSIEN Frédéric LACHAUD Valérie NASSIET Bouchra HASSOUNE Maëlenn AUFRAY Hélène WELEMANE - ICA, LGP, CIRIMAT (FR)

- ICA, Toulouse (FR)
- ICA, Toulouse (FR)
- LGP, Tarbes (FR)
- LGP, Tarbes (FR)
- CIRIMAT, Toulouse (FR)
- LGP, Tarbes (FR)

INSTITUT CLÉMENT ADER

Minh Nhat TO, nhat.to@isae-supaero.fr, +33 (0)7.89.96.78.18

## Content

- 1. Introduction
- 2. Methodology
- 3. Visco-hyperelastic parameters determination
- 4. Assessment of joint behavior
- **5.** Conclusion & Perspectives

## I. Introduction

## PhD context



## I. Introduction

## Problematic

- Hybrid bolted/bonded joints offer many advantages over pure bolted or bonded joints.
- In aerospace industry, **bolted-sealed joints** with very **small stiffness sealant** are widely used.
- ✓ With recent developments, sealant can play a part in the mechanical function, contributing in relieving stress thanks to its viscoelastic nature
- $\Rightarrow$  How can the load transferred by the sealant in a bolted-sealed joint be determined?
- Linear elasticity has difficulty representing rubber-like sealant large time-dependent deformations.
- ⇒ How can the complex visco-hyperelastic sealant behavior be accounted ?



Relative adhesive shear modulus

## I. Introduction

## Objective

• Propose a methodology to evaluate load transferred by the sealant based on numerical and experimental tests

• Present the process of determining viscohyperelastic parameters of sealant and their applications on pure bonded and bolted-sealed joints









#### - Sealant: **PR 1782 C 2**

is a two-part product based on a liquid polysulfide polymer polymerized with manganese dioxide. The putty takes 12 hours for curing under ambient conditions. [LJF-1]

According to thermomechanical analysis (TMA), the shear modulus of sealant in the rubbery/elastomeric state is constant over the experimental temperature range: [18; 24°C].

- Substrates: Aluminum 2024-T3
- Bolt EN3115-3-5 [AEC-1] & Nut ASNA2531-3 (Steel alloys)







Mechanical characterization of sealant

#### Uniaxial tensile (UAT) test



## Pure shear (**PS**) test





#### Rectangular specimen and PS test setting-up

Dumbbell specimen and UAT test setting-up

[ISO-1 ISO 37:2005(E)] [MEU-1 Meunier, L. (2007)]

## Test speed selection



Uniaxial Tensile Test



Pure Shear Test

- Elastomeric behavior is very sensitive to test speed: same local strain rate for both tests
- Monitoring by Digital Image Correlation (DIC)
  - For a local strain rate of 0.065 s<sup>-1</sup>
    - UAT at 250 mm/min
    - PS at 80 mm/min

## III. Visco-hyperelastic parameters determination

Mechanical characterization of sealant



**Relaxation test:** in the first phase, displacement increases with time until the elongation ratio reaches 2, then deformation is kept constant and the machine continues to capture the reaction force of the specimen.

## III. Visco-hyperelastic parameters determination

Hyperelastic parameters

<u>Idea</u>: For each hyperelastic model, the engineering stress equations are determined as a function of the hyperelastic coefficients and elongation  $\lambda$  by taking the differential of the strain energy density **W** while using the assumptions specific to each test.



✓ The Mooney-Rivlin model with 5 coefficients is selected for this research.

**Viscoelastic parameters** 

#### Generalized Maxwell model with 3 Maxwell arms





Maxwell arm	E <sub>i</sub> [MPa]	т <sub>i</sub> [S]	<b>g</b> i
i = 1	0.0620	1	0.0725
i = 2	0.1382	10	0.1615
i = 3	0.1472	100	0.1722



## Joint tests



The test speed is **0.5 mm/min** for both the pure bonded joint and the bolted-sealed joint tests.

The relationship between applied force and local displacement at the center point of the middle substrate in the overlap zone is evaluated.





## **FE models**



Pure bonded joint:

2D plane strain and 3D models for the comparison

- $\Rightarrow$  Simulation tests are carried out using the 2D plane strain model to save calculation time.
- <u>Bolted-sealed joint:</u> 3D model

## **FE models**

■ Substrates & bolt:

#### Boundary conditions:



#### Applied displacement: 0,15 mm

- □ Evaluate joint behavior under low loads corresponding to fatigue state.
- □ Chosen according to the mesh resolution, sealant material law and convergence capability.

Material	Material properties	
AI 2024-T3	E = 72 GPa, v = 0.33	
Steel alloys	E = 21 GPa, v = 0.33	
Contact "bolt-hole"	μ = 0.3	

#### Sealant:

**Materials:** 

VHE	Visco-hyperelastic: 5-coeff Mooney-Rivlin + Prony series
HE	Pure hyperelastic : 5-coeff Mooney-Rivlin



#### <mark>15</mark>/18





- Greater influence of the sealant for small thickness of sealant layer
- VHE represents better the joint behavior than HE

#### Assessment of load transferred by the sealant

• Bolt load transfer rate  $(\tau)$ : is defined by the ratio of the load supported by the bolt to the total load applied.





⇒ Sealant material identification allows for bolted-sealed joint behavior modeling and the determination of load transferred by the sealant.

## **V. Conclusion & Perspectives**

- Visco-hyperelastic coefficients can be determined from several test data (UAT and PS) using the presented optimization method.
- Only PS experimental data would be needed to predict the force transfer distributions in double-lap joints.
- □ The more complex the sealant material model, the lower the convergence capability of the numerical simulation.

- Increasing the load transferred by sealant could help to improve the fatigue life of the bolted-sealed joint
- An idea for increasing the fatigue strength of an bolted-sealed joint by reinforcing the sealant formulation will be developed.







14<sup>TH</sup> EURADH & 7<sup>TH</sup> WCARP 2023

# THANK YOU Any questions?

Contact: Minh Nhat TO, nhat.to@isae-supaero.fr, +33 (0)7.89.96.78.18



## REFERENCES

STITUT CLÉMENT ADER

**[AEC-1]** AECMA, prEN6115-P5 (Aerospace Series, Bolts), European Association of Aerospace Industries (2007)

[ATR-1] A. Atre, W. S. Johnson, Analysis of the effects of interference and sealant on riveted lap joints, Journal of Aircraft (2007), vol. 44, No. 2

**[AUS-1]** P.E. Austrell, Modeling of Elasticity and Damping for Filled Elastomers, Report TVSM 1009 (1997).

**[BON-1]** L. Boni et A. Lanciotti, Fatigue behavior of double lap riveted joints assembled with and without interfay sealant, Fatigue & Fracture of Engineering Materials and Structures 34 (2010) 60-71

[BRI-1] L.C. Brinson, T.S. Gates, Comprehensive Composite Materials, 2000

**[CAB-1]** M. Cabello et al, A non-linear hyperelastic foundation beam theory model for double cantilever beam tests with thick flexible adhesive, International Journal of Solids and Structures (2015),

**[DEC-1]** C. Dechwayukul et al, Analysis of the effects of thin sealant layers in aircraft structural joints, AIAA journal (2003), vol. 41, No. 11

**[HOA-1]** C.T. Hoang-Ngoc, E. Paroissien, Simulation of single-lap bonded and hybrid (bolted/bonded) joints with flexible adhesive, International Journal of Adhesion & Adhesives 30 (2010) 117–129

**[ISO-1]** ISO 37:2005(E): Rubber, vulcanized or thermoplastic: Determination of tensile stressstrain properties

**[KAM-1]** N. Kamnerdtong, The shear properties of a polymer sealant and analyses of the distorsion and fatigue of sealed countersunk riveted lap joints, PhD thesis (2001)

**[KEL-1]** G. Kelly. Quasi-static strength and fatigue life of hybrid (bonded/bolted) composite single-lap joints. Journal of Composite Structures 2006; 72:119–29.

[LJF-1] Le Joint Français, PR 1782 C Mastic d'étanchéité réservoirs structuraux interposition faible densité, Édition 10/2011

**[MEU-1]** L. Meunier et al. Caractérisation du comportement mécanique d'élastomères silicones chargés et non chargés. 18ème Congrès Français de Mécanique, Aug 2007, Grenoble, France. hal-01978979

Toulouse

**[MOO-1]** M.A. Mooney. A theory of large elastic deformation. Journal of Applied Physics. 1940; 11(9):582-592.

http://dx.doi.org/10.1063/1.1712836.

**[MOR-1]** D.C. Moreira, L.C.S. Nunes, Comparison of simple and pure shear for an incompressible isotropic hyperelastic material under large deformation, Polymer Testing 32 (2013) 240–248

[NIU-1] Michael .C.Y. Niu. Airframe Structural Design

**[OGD-1]** R.W. Ogden. Large deformation isotropic elasticity on the correlation of theory and experiment for the incompressible rubber-like solids. Proceedings of the Royal Society of London. Series A, Mathematical and Physical Sciences. 1972; 326(1567):565584.

[PAR-1] E. Paroissien, F. Lachaud et al. Simplified stress analysis of hybrid (bolted/bonded) joints, International Journal of Adhesion and Adhesives 77 (2017) 183–197

**[PAR-2]** Paroissien, E. Modélisation simplifiée des transferts d'effort dans les assemblages boulonnés et/ou collés, Mémoire d'Habilitation à Diriger des Recherches, Université de Toulouse III Paul Sabatier (FR). (2020)

**[RIV-1]** R.S. Rivlin. Large elastic deformations of isotropic materials VII. Experiments on the deformation of rubber, Philosophical Transitions of the Royal Society of London A. (1951) 243(865), pp. 251-288.

[ROS-1] P.L. Rosendahl, From Bulk to Structural Failure: Fracture of Hyperelastic Materials (2020)

**[SAS-1]** M. Sasso, G. Palmieri, G. Chiappini, D. Amodio, Characterization of hyperelastic rubberlike materials by biaxial and uniaxial stretching tests based on optical methods, Polymer Testing 27 (2008) 995–1004

**[SHA-1]** M. Shahzad et al. Mechanical Characterization and FE Modelling of a Hyperelastic Material, Materials Research. 2015; 18(5): 918-924.

[WIL-1] M. L. Williams, Structural analysis of viscoelastic materials, AIAA Journal (1964) vol 2: 785-808