

Reconfiguration of microcontroller payload driver

Romain Boyer, Frédéric Camps

▶ To cite this version:

Romain Boyer, Frédéric Camps. Reconfiguration of microcontroller payload driver. Cubsat Symposium 2023, Dec 2023, Louvain, Belgium. 2023. hal-04330627

HAL Id: hal-04330627 https://hal.science/hal-04330627v1

Submitted on 8 Dec 2023

HAL is a multi-disciplinary open access archive for the deposit and dissemination of scientific research documents, whether they are published or not. The documents may come from teaching and research institutions in France or abroad, or from public or private research centers. L'archive ouverte pluridisciplinaire **HAL**, est destinée au dépôt et à la diffusion de documents scientifiques de niveau recherche, publiés ou non, émanant des établissements d'enseignement et de recherche français ou étrangers, des laboratoires publics ou privés.

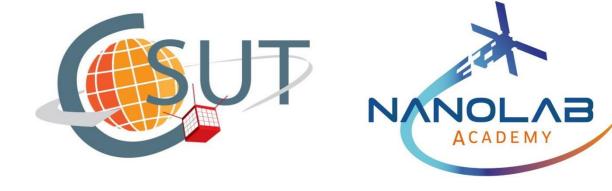












MPH

CUBESA CSUT

NIMPH, Nanosatellite to Investigate Microwave Photonics Hardware

NIMPH summary

The project:

NIMPH is an academic research project led by Volume 3U (300 cm³) the University Space Centre of Toulouse (CSUT). It involves numerous Master's

students from Paul Sabatier University (UPS) and the ENSEEIHT engineering school, mainly through internships, research projects and practical work, under the supervision of • Mission duration: 2 years professors and permanent staff from LAAS -
3 payloads: EDMon, RADMon, M2M CNRS.

Nanosatellite characteristics :

- Weight 2 kg
- UHF radio (430 MHz)
- Sun-synchronous low-Earth orbit
 - Power consumption: 7 W
- - C phase

EDMon (NIMPH Payload):

The NIMPH project's EDMon payload is designed to evaluate the influence of solar radiation and more generally space environment on the performance (gain & noise figure) of three Erbium doped fiber optic amplifiers (EDFA).

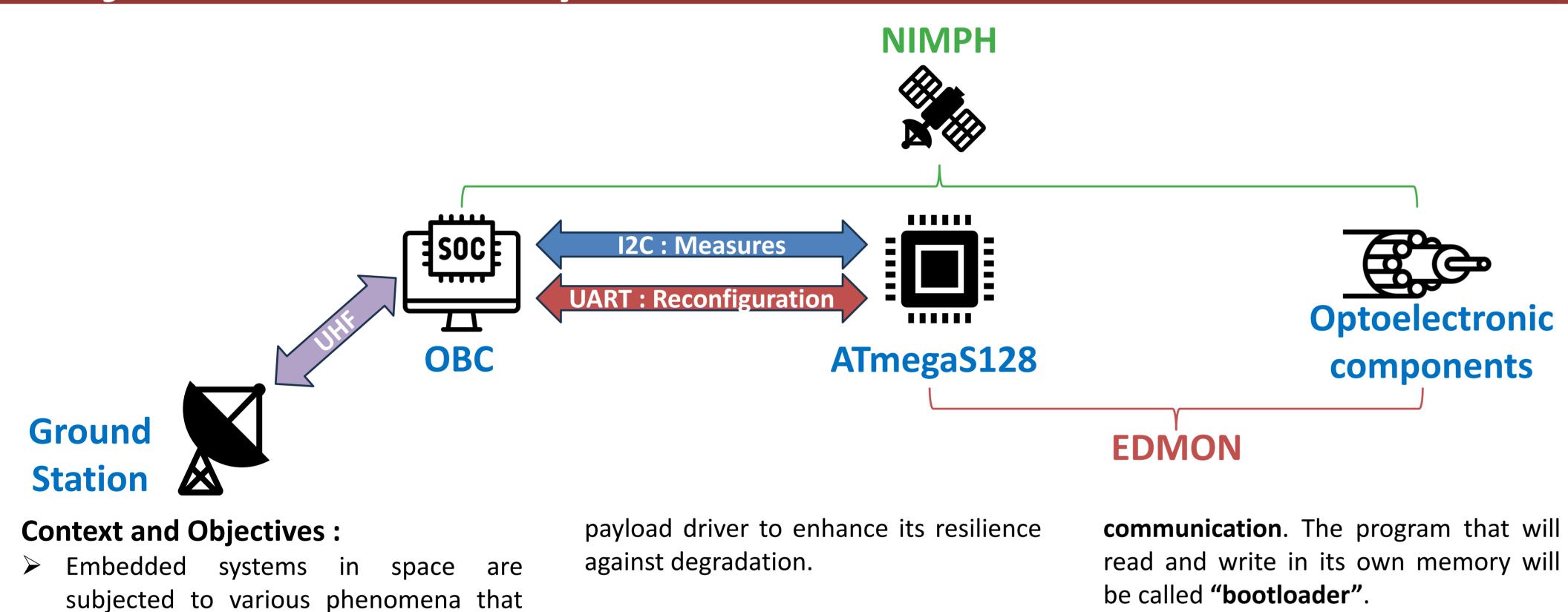
The main challenges are the measurement reliability and the miniaturization of the payload. A microcontroller ATmegaS128 drives electronics components.

Reconfiguration of microcontroller payload driver

LAAS-CNRS, Université de Toulouse, CNRS, Toulouse, France

Frédéric Camps, Romain Boyer, Cyril Campanella / Contact fcamps@laas.fr, rboyer@laas.fr

Objectives and System Architecture:



Architecture :

- **Payload EDMon NIMPH:** Major payload of the nanosatellite NIMPH with optoelectronic components. To control them, an ATmegaS128 microcontroller is integrated into one of the two electronic cards of the payload.
- **On-Board Computer (OBC):** The on-board computer is a Soc Z7030 ("Ninano" electronic card developed by CNES and STEEL). It is responsible for controlling every single part of the satellite, including the microcontroller from the EDMon payload thanks to a spatial hypervisor XNG (FentISS). An I2C serial bus is used between them to transmit data and issue commands (set DAC, get ADC...). A UART serial bus is used to reconfigure EDMon microcontroller (transmit binary the programs, reconfiguration orders...).

damage memory components, such as radiation, wear and tear of electronic degradation of components, and electronic cards due to severe temperature cycling. Therefore, there is a need to reprogram a nanosatellite

may disrupt system operations and > To mitigate potential memory damage

and prevent program failure, the ATmegaS128 must include a specific program to modify the FLASH memory (128 kB) and reload a new mission software (MSW). The main objective is the EDMON make payload to reconfigurable UART via

In order to enhance system resilience and address potential failure modes within the failure tree, such as optical, electronic, mission or reconfiguration issues, the bootloader is also used as part of the objective and system architecture.

Ground Station SCC: A ground station SCC from U-Space / CNES is used to send Telecommands (TC) for reconfiguration that includes EDMon flight programs within the data fields. These Telecommands are received by the OBC before starting a reconfiguration of the EDMon controller.

Memory Architecture:

FLASH Memory Architecture:

- ATmegaS128's FLASH memory (128kB) is divided in two sections: Application section (blue, 120kB) and Bootloader section (orange, 8kB).
- Application Section hold main program, so then flight measurement program
- Bootloader Section hold **bootloader** program which is executed after a hardware reset.

Application Section Subdivision:

Application section is divided in six slots of 20kB.

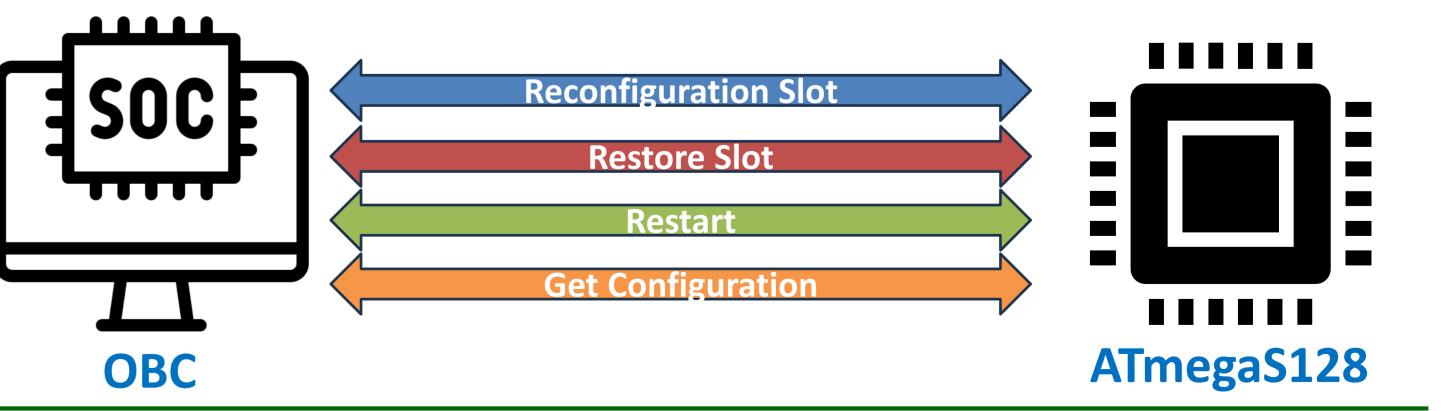
<u>Address :</u>		<u>Size :</u>	
0x00000	Slot 0		
0x04FFF	Running Slot	20kB	
0x05000	Slot 1		
0x09FFF	Backup Slot	20kB	
0x0A000	Slot 2		
0x0EFFF	Backup Slot	20kB	
0x0F000	Slot 3		
0x13FFF	Backup Slot	20kB	
0x14000	Slot 4		
0x18FFF	Launch MSW Slot	20kB	
0x19000	Slot 5		
0x1DFFF	Configuration Slot	20kB	
0x1E000	Bootloader program	8kB	
0x1FFFF			
<u>Legend :</u>	Application Section		
	Bootloader Section		

Communication protocol:

Reconfiguration Commands:

On-Board Computer has four commands available in order to proceed to a reconfiguration function. Therefore, these commands are integrated in the bootloader program:

- **Reconfiguration Slot**: Ask for write a program in a specific slot number. This slot number is asked by ATmegaS128 before data transfer.
- **Restore Slot**: Ask for copy a specific slot number into the running slot (slot number 0).
- **Restart**: Ask for a restart of the running program. This command simply exit bootloader program and jump to application section (address 0x00000).
- **Get configuration**: Ask for the information in the configuration slot.



- The first slot is in the **running slot** (where the current running program is located). Due to interrupt table location, only this slot can be run.
- Slot 1 to 3 are **backup slots** where backup programs are stored.
- Slot 4 is the **security slot** where the flight mission software program is stored. If an error occurred on UART bus, this slot is restored in running slot number 0.
- Finally, slot 5 is a **configuration slot** which keep the actual configuration of memory as number of pages in each slot, ID and software versioning, bootloader versioning ...

Communication protocol and Safety:

Communication protocol is essential to ensure resiliency in reconfiguration. Here are some examples of communication messages between OBC and ATmegaS128.

- Robust design of this UART system is essential for the mission and durability of the • nanosatellite. Effective management of Hamming distance ensures increased resilience to space radiation, thereby ensuring the success of this space mission.
- To secure heavy data transfer, cyclic redundancy code (CRC) is applied to data. The • algorithm used is **CRC16-CCITT FALSE** (ESA ECSS-E-ST-70-41C, Annex B.1). The same CRC is used in PUS protocol (Ground Station – NIMPH communications). This ensure resiliency of the written bootloader.