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## APPLICATION OF THE WAVELET TRANSFORM INVERSE ANALYSIS FOR MODAL IDENTIFICATION AND DAMAGE DETECTION

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**Abstract.** *The use of the Continuous Wavelet Transform method for modal identification and structural damage analysis is discussed in this paper. The wavelet transform method can be used to understand the responses of different structures under different loading conditions, i.e. after noise-embedded shock signals. The method allows the processed signal to be displayed in the time-frequency plane, making it possible to estimate modal characteristics, or study the effect of structural defects on the vibration behaviour of the structure. The paper summarises the important aspects of the method and proposes a novel formulation for structural assessment. Two example applications are presented, a four-degree-of-freedom linear system vibrating freely and a damaged masonry wall subjected to shear compression and ambient vibration tests. The examples show that the CWT method can be used to obtain valuable information for structural characterization.*

# 1 INTRODUCTION

The Wavelet Transform (WT) method appeared in mathematics more than a century ago, starting with Haar who discovered in 1909 an orthonormal basis consisting of step functions. The Haar basis construction is a precursor of what is known today as the multiresolution analysis [1] which refers to the expansion of a signal into components that can reproduce the original signal when added together. The principle of multiresolution spaces is to decompose a signal of finite energy  $\mathcal{L}^2(\mathbf{R})$  to two complementary spaces: (i) a space of approximation; and (ii) a detail space that contains the approximation error. Multi-resolution analysis can be seen as a way to zoom into, or out of, the signal without losing information. From the beginning of 1980s, under the impetus of several French researchers, especially Grossmann and Morlet [2], wavelet research in mathematics has grown steadily with significant contributions from many authors. Figure 1 provides a quick overview of the history of the WT method. Since then, the WT has been used in many areas of science and engineering. For example, is a recent reference published by A. Guillet and his co-workers, the method has been applied to physiological signals. In reference [14], the authors introduced time-log-frequency ratio distributions based on WT with analytic mother wavelets (the Grossmann wavelet belonging to the Morse wavelets family cf. [19]) that they applied to voice recordings. Two main trends exist in how the WT is computed and used: the Continuous Wavelet Transform (CWT) and the Discrete Wavelet Transform (DWT) method [3]. Our work will focus on the former.

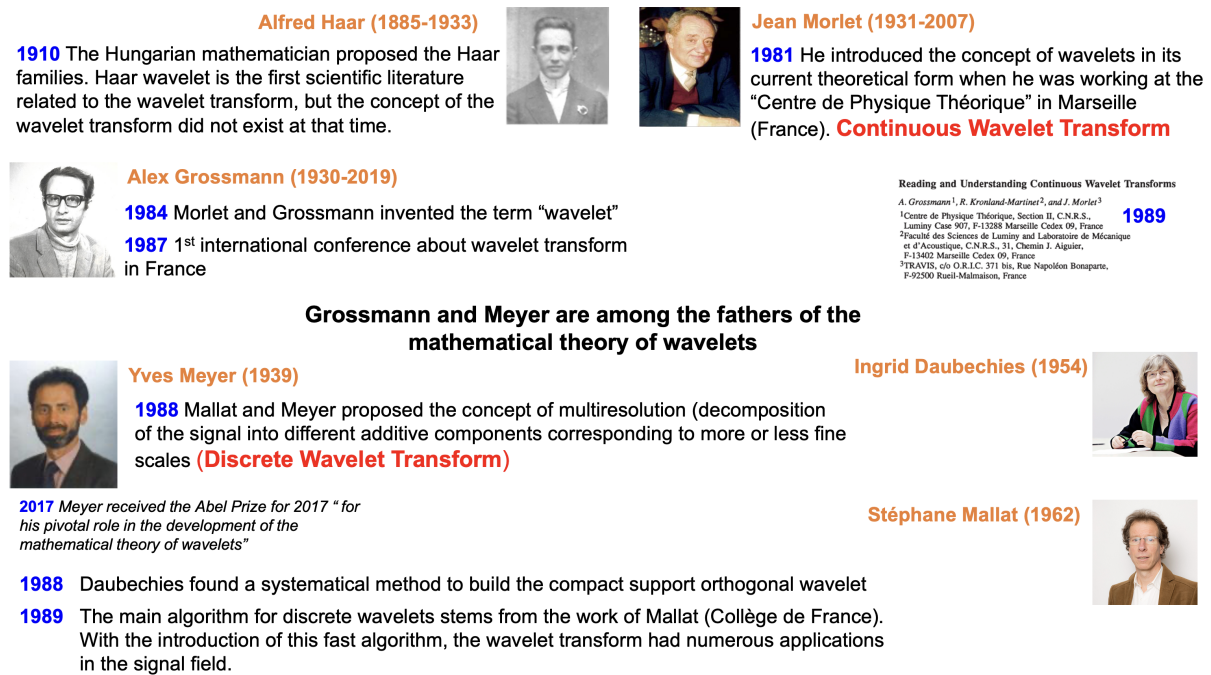


Figure 1: Short History of Wavelet Transform (WT).

The wavelet transform method, allows the time-frequency (TF) analysis of a signal. An interesting analogy between TF analysis and modern music notation is shown in Figure 2. Modern music notation gives an illustration of the time-frequency representation where the vertical position of the note-head within the staff indicates its pitch which can be modified by accidentals, associated with a frequency. For example, the frequency of A3 is set at 440 Hz. The time duration (note length, or note value) is indicated by the form of the note-head or with the addition of a note-stem plus beams or flags (cf. 2).

Among the large number of books on the wavelet transform method [4, 6, 3], few are dedicated to applications on civil engineering structures. The book by Chatterjee [7] describes several applications of wavelets to civil engineering problems and shows their importance, for example, in the analysis of non-stationarities in seismic ground motions, in the study of bridge vibrations caused by vehicle passage, or in the identification of structural damage. Furthermore, damaging events often result in short-time, non-stationary vibration characteristics that are difficult to analyse using classical modal approaches. Methods based on time-frequency representations, such as wavelet transform methods, are better suited to analysing signals generated by time-varying systems than those using the Fourier transform that are better suited to signals generated by time-invariant systems. This makes these time-frequency based methods a very interesting tool for structural damage detection.

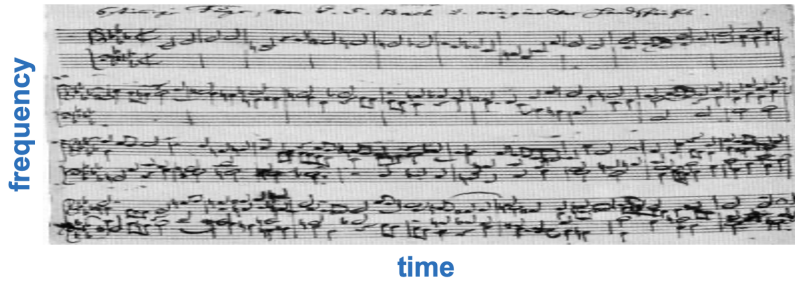


Figure 2: Writing of a musical piece: a famous illustration of time-frequency representation.

## 2 THE CONTINUOUS WAVELET TRANSFORM METHOD

The basic idea of the Continuous Wavelet Transform (CWT) method is to use a function  $\psi(t) \in L^1(\mathbb{R}) \cap L^2(\mathbb{R})$  called the “mother wavelet”<sup>1</sup>, which can generate an infinite continuous basis of functions, by varying the parameters  $a$  and  $b$ :

$$\psi_{a,b}(t) = (1/a)\psi[(t - b)/a], \quad (1)$$

where  $a$  is a scale parameter and  $b$  a temporal localization parameter.

There is a clear relationship between the scale parameter  $a$  and the frequency  $f$ . More specifically, a low value of  $a$  results in a compressed wavelet, allowing to capture details that change rapidly in the processed signal, thus resulting in a high frequency content. On the other hand, a large value of  $a$  results in a stretched wavelet, which captures slowly changing coarse features in the processed signal, resulting in a high frequency content. Thus, the ratio  $1/a$  can be assimilated to a frequency parameter  $f$ . There are several possible approaches that relate the scale  $a$  to the inverse of the frequency, typically denoted:  $a = f_\psi^*/f$ . In the references [8, 10],  $f_\psi^*$  is chosen to be equal to the peak frequency  $f_\psi^0$ , where  $f_\psi^0 = \arg \max(|\hat{\psi}|)$  and  $\hat{\psi}$  is the Fourier transform of the mother wavelet. Two other meaningful frequencies, classically found in literature, are the energy frequency:  $\tilde{f}_\psi^2$  and the time-varying instantaneous frequency of the

<sup>1</sup>A function  $\psi(t) \in L^1(\mathbb{R}) \cap L^2(\mathbb{R})$  can be admissible to a mother wavelet if it satisfies the admissibility condition:  $C_\psi = \int_{-\infty}^{+\infty} \frac{|\hat{\psi}(f)|^2}{|f|} df < \infty$  where  $\hat{\psi}(f) = \int_{-\infty}^{+\infty} \psi(t)e^{-2i\pi ft} dt$  is the Fourier transform of  $\psi(t)$ , which implies that  $\int_{-\infty}^{+\infty} \psi(t) dt = 0$ . cf. [4].

<sup>2</sup> $\tilde{f}_\psi = \frac{\int_{-\infty}^{+\infty} f |\hat{\psi}(f)|^2 df}{\int_{-\infty}^{+\infty} |\hat{\psi}(f)|^2 df}$ .

wavelet at its time center  $\check{f}_\psi(0)$ <sup>3</sup> (cf. [17]). These three definitions of  $f_\psi^*$  result to three different mappings between scale and frequency. The first will correctly give the frequency of a pure harmonic signal from the scale  $a$  at which its transform obtains a maximum. The second will correctly give the frequency of a pure cosine from the energy mean scale of the CWT  $\widetilde{a}_\psi(b)$ <sup>4</sup>. The third fixes the frequency to be the same as the phase progression of the transform at the location of an infinitesimally narrow impulse.

To better highlight the importance of the choice of  $f_\psi^0$ , consider the example of the CWT of a cosine  $u(t) = A_1 \cos(2\pi f_1 t + \theta_1)$ . The CWT can be written as:  $T_\psi[u](a, b) = \frac{1}{2} A_1 \widehat{\psi}(af_1) e^{i\theta_1}$  and its modulus  $|T_\psi[u](a, b)| = \frac{1}{2} A_1 |\widehat{\psi}(af_1)|$  is maximal when  $|\widehat{\psi}(af_1)|$  is maximal, so that  $af_1 = f_\psi^0$  and  $|T_\psi[u](a, b)| = A_1$ . In addition, the energy mean scale of the CWT  $\widetilde{a}_\psi(b)$  in the case of a pure cosine is equal to:  $\widetilde{a}_\psi(b) = \check{f}_\psi / f_1$ <sup>5</sup>.

Using a mother wavelet  $\psi$ , the CWT method is then used to decompose a function  $u(t)$  into the time-frequency domain as follows:

$$T_\psi[u](a, b) = \frac{1}{a} \int_{-\infty}^{+\infty} u(t) \overline{\psi}\left(\frac{t-b}{a}\right) dt, \quad (2)$$

where  $\overline{\psi}$  is the conjugate of  $\psi$ .

Regarding the regularizing term  $1/a$  in the definition of Eq. 2. This term is appropriate when the magnitude of the modulus wavelet transform is wished to reflect the amplitude of the analysed signal  $u(t)$ . It is generally more useful to describe time-localized signals by their amplitude; hence, Eq. 2 uses the  $1/a$  normalization which is known as the  $L_1$  norm, it is used in Carmona *et al.* [4], preferred by Argoul [12, 13], or Lilly[15]. Instead, the factor  $1/\sqrt{a}$  can guarantee that the wavelet maintains constant energy. This is known as the  $L_2$  norm of  $T_\psi$  and is more appropriate when one wishes that the modulus-squared wavelet transform reflects the energy of the analyzed signal.

The definition of CWT of Eq. 2 can be rewritten in the frequency domain:

$$T_\psi[u](a, b) = \int_{-\infty}^{+\infty} \widehat{u}(f) \widehat{\psi}(af) e^{2i\pi fb} df, \quad (3)$$

where  $\widehat{u}(f) = \int_{-\infty}^{+\infty} u(t) e^{-2i\pi ft} dt$  is the Fourier transform of  $u(t)$  and  $\widehat{\psi}$  is the mother wavelet in the frequency domain. Eq. 3 allows for an easy numerical computation by means of the fast Fourier transform (FFT) algorithm.

## 2.1 Selection of mother wavelet

Apart from the widely-used Morlet wavelet, which is only approximately analytic, various analytic wavelets have been proposed, including the Cauchy-Paul, the derivative of the Gaussian wavelet, the lognormal or log Gabor, the Shannon and the Bessel, among others. Olhede and Walden [16] and Lilly and Olhede [17] have showed that all known analytic wavelets can be grouped together in a much larger family, first introduced by Daubechies and Paul [18], whose properties were studied in detail [19]. This broad “superfamily” of wavelets is known as the

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<sup>3</sup>  $\check{f}_\psi(t) = \frac{1}{2\pi} \frac{d}{dt} \Im \{ \ln \psi(t) \} = \frac{1}{2\pi} \frac{d}{dt} (\arg \{ \psi(t) \})$ .  
<sup>4</sup>  $\widetilde{a}_\psi(b) = \frac{\int_{-\infty}^{+\infty} a |T_\psi[u](a, b)|^2 da}{\int_{-\infty}^{+\infty} |T_\psi[u](a, b)|^2 da}$ .  
<sup>5</sup>  $\widetilde{a}_\psi(b) = \frac{\int_{-\infty}^{+\infty} a |T_\psi[u](a, b)|^2 da}{\int_{-\infty}^{+\infty} |T_\psi[u](a, b)|^2 da} = \frac{\frac{1}{4} A_1^2 \int_0^{+\infty} a |\widehat{\psi}(af_1)|^2 da}{\frac{1}{4} A_1^2 \int_0^{+\infty} |\widehat{\psi}(af_1)|^2 da} = \frac{1}{f_1} \frac{\int_0^{+\infty} f |\widehat{\psi}(f)|^2 df}{\int_0^{+\infty} |\widehat{\psi}(f)|^2 df} = \frac{\check{f}_\psi}{f_1}$ .

generalized Morse wavelets. Within that family, one can objectively say which wavelet choice is the “best” for the problem at hand. The generalized Morse wavelets  $\psi_{\beta,\gamma}$  depend on two parameters  $\beta$  and  $\gamma$  which control the shape of the wavelet. The general expression is as follows:

$$\widehat{\psi}_{\beta,\gamma}(f) = c_{\beta,\gamma} f^\beta e^{-(2\pi f)^\gamma} H(f) \quad (4)$$

where  $c_{\beta,\gamma}$  is a normalization constant and  $H(f)$  is the unit step function.

Among the broad family of generalised Morse wavelets, the Cauchy-Paul mother wavelet  $\psi_n(t)$  has been preferred in our work. This is a complex single parameter mother wavelet where  $n$  is an integer that controls the shape of the wavelet. With respect to Eq. 4, the Cauchy-Paul wavelet belongs to the Morse family when  $\beta = n$  and  $\gamma = 1$ . The definition with respect to parameter  $n$ , becomes:

$$\widehat{\psi}_n(f) = c_n f^n e^{-2\pi f} H(f) \quad (5)$$

where  $c_n$  is a normalization constant. An asymmetry in the frequency domain of the Cauchy-Paul wavelet leads to distinct values for the frequencies previously introduced in order to define the scale parameter  $a = f_\psi^*/f$ , i.e.:

$$f_{\psi_n}^0 = \frac{n}{2\pi}, \quad \tilde{f}_\psi = \frac{n + \frac{1}{2}}{2\pi}, \quad \check{f}_{\psi_n} = \frac{n + 1}{2\pi}, \quad (6)$$

There are several ways to define the  $c_n$  parameter. In older papers [11, 12, 13] we preferred the  $L^\infty$  norm in the time domain:  $\|\psi_n\|_\infty = 1$ . In this paper we have chosen the norm:  $\|\widehat{\psi}_n\|_\infty = \max |\widehat{\psi}_n(f)| = 2$ , which leads to  $c_n = 2(2\pi e/n)^n$ . Therefore, the peak frequency is:  $f_{\psi_n}^0 = n/2\pi$ , for which the maximum value is:  $\widehat{\psi}_n(f_{\psi_n}^0) = 2$ . This is the same choice as in Lilly [15], since it provides a direct equality between  $|T_\psi[u](t, a(t))|$  and the amplitude of  $u$ .

There are several reasons for choosing an analytic mother wavelet, and in particular the Cauchy-Paul wavelet. The first is that an analytic mother wavelet which is a complex function whose spectrum contains only positive frequencies, leading to:  $\widehat{\psi}_n(f) = 0 \forall f < 0$ . As an analytic mother wavelet, it only responds to non-negative frequencies in the signal being analysed and it produces a transform whose modulus is less oscillatory than in the case of a real mother wavelet. This property is a real advantage for detecting and tracking the instantaneous frequencies contained in the signal. Eq. 3 is simplified as follows:

$$T_\psi[u](a, b) = \int_0^{+\infty} \widehat{u}(f) \widehat{\psi}(af) e^{2i\pi fb} df. \quad (7)$$

The second reason concerns the phase retrieval problem which is related to the reconstruction of a function from its scalogram, that is, from the modulus of its wavelet transform. Mallat and Waldspurger [20] mathematically proved that the reconstruction of the analyzed function from the modulus of its Cauchy-Paul wavelet transform is unique up to a global phase. They have also showed that the reconstruction operator is continuous but non uniformly continuous. The authors specify that the proofs are special to Cauchy-Paul wavelets and cannot be extended to generic wavelets because they make intensive use of the link between Cauchy wavelets and holomorphic functions.

## 2.2 Resolution in the time-frequency domain

For the successful CWT decomposition, it is essential to control the *time-frequency resolution* of the signal wavelet transform. This is necessary in order to correctly interpret the CWT plot and also to take into account *edge effects* [13].

Referring to the conventional frequency analysis of constant- $Q$  filters, Le and Argoul [13] introduced a parameter  $Q_m$  called “quality factor”. This parameter can be used in order to control the time-frequency resolution of the CWT. Introducing the duration  $\Delta t$  and the frequency bandwidth  $\Delta f$  of the wavelet transform and its relationship with the the duration  $\Delta t_\psi$  and the frequency bandwidth  $\Delta f_\psi$  of the mother wavelet are as follows:

$$\Delta t = a \Delta t_\psi \quad \Delta f = \frac{\Delta f_\psi}{a}, \quad (8)$$

where  $\Delta t_\psi$  and  $\Delta f_\psi$  are stated in terms of root mean squares for  $L_2$  norm which are equivalent to the standard deviation in statistics:

$$\begin{aligned} \Delta t_\psi &= \frac{1}{\|\psi\|_2^2} \int_{-\infty}^{+\infty} (t - t_\psi)^2 |\psi(t)| dt \\ \Delta f_\psi &= \frac{1}{\|\widehat{\psi}\|_2^2} \int_{-\infty}^{+\infty} (f - f_\psi)^2 |\widehat{\psi}(f)|^2 df, \end{aligned} \quad (9)$$

where  $t_\psi$  and  $f_\psi$  are the centre of  $\psi$  and  $\widehat{\psi}$ , respectively:

$$t_\psi = \frac{1}{\|\psi\|_2^2} \int_{-\infty}^{+\infty} t |\psi(t)|^2 dt \quad f_\psi = \frac{1}{\|\widehat{\psi}\|_2^2} \int_{-\infty}^{+\infty} f |\widehat{\psi}(f)|^2 df. \quad (10)$$

It is noteworthy to introduce the CWT uncertainty  $\mu$ , equal to the mother wavelet uncertainty  $\mu_\psi$ :

$$\mu = \Delta t \Delta f = \Delta t_\psi \Delta f_\psi = \mu_\psi \quad (11)$$

Using the relationship between scale and frequency previously discussed ( $a = f_\psi^*/f$ ), Eq. 8 becomes:

$$\Delta t = \frac{f_\psi^* \Delta t_\psi}{f} \propto \frac{1}{f} \quad \Delta f = \frac{\Delta f_\psi}{f_\psi^*} f \propto f, \quad (12)$$

This  $Q_m$  factor introduced in [13] is defined as the ratio of the mean value  $\tilde{f}$  (for  $L_2$  norm) previously defined over two times its standard deviation  $\Delta f$  of the wavelet transform, i.e.:

$$Q_m = \frac{\tilde{f}}{2\Delta f} = \frac{\frac{\tilde{f}_\psi}{a}}{2\frac{\Delta f_\psi}{a}} = \frac{\tilde{f}_\psi}{2\Delta f_\psi}. \quad (13)$$

Notice from Eq. 13 that the resolution parameter  $Q_m$  is independent of the frequency parameter  $a$ . From the formulae above, it is easy to see that:

$$\Delta t = 2 Q_m \mu_\psi \frac{f_\psi^*}{\tilde{f}_\psi} \frac{1}{f} \quad \Delta f = \frac{1}{2 Q_m} \frac{\tilde{f}_\psi}{f_\psi^*} f. \quad (14)$$

where the ratio  $\frac{\tilde{f}_\psi}{f_\psi^*}$  is close to 1 and equal to 1 if  $f_\psi^* = \tilde{f}_\psi$ ; for example for Cauchy wavelet it depends on the value of  $n$  (Eq. 6).

Thus, a small  $Q_m$  value increases the resolution in the time axis ( $\Delta t$  small), while a large value increases the frequency resolution ( $\Delta f$  small).

Properly adjusting the  $Q_m$  parameter is particularly important in the case of two close Eigenmodes which allows to separate them. In reference [13], it was shown that the value of  $Q_m$  can be chosen within the limits below:

$$Q_m^{(min)} = c_f \frac{f_j}{2Df_j} \leq Q_m \leq Q_m^{(max)} = \frac{\pi L f_j}{c_t}, \quad (15)$$

where  $L$  is the length of the signal,  $c_t, c_f$  are constants that are used to identify edge effects in the time and the frequency domain, respectively,  $f_j$  is the Eigenfrequency of interest and  $Df_j$  is its distance from the closest Eigenfrequency. Note that typically  $c_t$  and  $c_f$  receive values between 3 and 5.

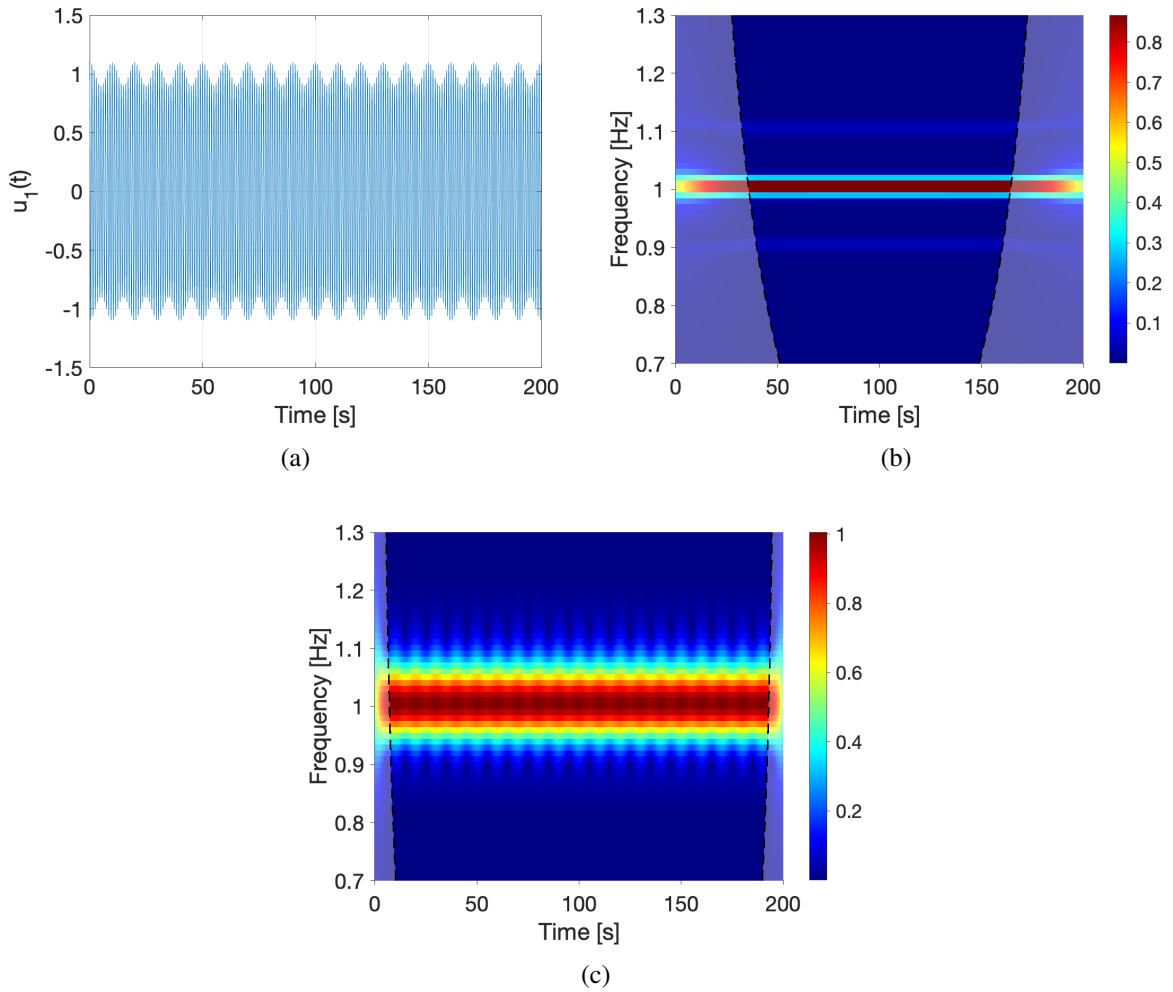


Figure 3: Wavelet analysis on an amplitude-modulated signal  $u_1(t) = A(t) \cos(2\pi t)$  with  $A(t) = 1 + 0.2 \cos(2\pi t/10)$ : (a) the signal  $u_1(t)$  in the time domain, (b) scalogram computed with  $Q_m = 75$  and (c) scalogram computed with  $Q_m = 15$ .

The importance of the resolution parameter can be understood from the two examples of Figures 3 and 4. The figures present two examples of signals, with their time-frequency scalogram plots.



The first example concerns the amplitude modulated signal  $u_1(t) = A(t) \cos(2\pi t)$ , where  $A(t) = 1 + 0.2 \cos(2\pi t/10)$  (Figure 3a). The expression for  $u_1(t)$  can be misleading. By developing the equation using trigonometric formulae, it can be shown that  $u_1(t)$  is the sum of three cosines:  $u_1(t) = \cos(2\pi t) + 0.1 \cos(\frac{11}{10} 2\pi t) + 0.1 \cos(\frac{9}{10} 2\pi t)$  which can be identified in the scalogram plot of Figure 3b) as three horizontal ridges. Due to the amplitude dominance of the first term at 1Hz, the other two frequencies at 9/10 and 11/10Hz cannot be seen unless the quality factor  $Q_m$  is chosen “correctly”. According to Eq. 15, the bounds are:  $Q_m^{(min)} = 25$  and  $Q_m^{(max)} = 125.6$  respectively, with  $c_t = c_f = 5$ ,  $f_j = 1Hz$ ,  $\Delta f = 1 / 10$  Hz and  $L = 200s$ . If we had chosen a quality factor lower than  $Q_m^{(min)} = 25$ , giving precedence to time resolution over frequency resolution, the scalogram plot would have shown oscillations at the edges, as shown in Figure 3c).

The second example (Figure 4) is that of an acoustic signal  $u_2(t)$  measured when the passage of a semi-trailer on a joint of pavement in very good condition on a motorway bridge and plotted in Fig. 4a). The frequency content of this signal is mainly between 0 Hz and 2500 Hz (see. Fig. 4b)). Zooming in on the amplitude spectral density of  $u_2(t)$ , which is the square root of its classical Pseudo Spectral Density (PSD) over this frequency range, we see three main peaks around the frequencies 250, 600 and 900Hz. Table 1, shows the limits of the  $Q_m$  factor:  $Q_m^{(min)}$  and  $Q_m^{(max)}$ , for the three frequency values obtained using Eq. (15). Two scalograms of this signal are then plotted for two different values of the quality factor in Figures 4c) and 4d):  $Q_m = Q_m^{(min)} = 2$  and  $Q_m = Q_m^{(max)} = 125$ , respectively, corresponding to the limits:  $Q_m^{(min)} = 2$  and  $Q_m^{(max)} = 125$  for the relevant frequency:  $f_j = 250Hz$  given in Table 1.

A comparison of these two scalograms provides an obvious interpretation. The lower the value of  $Q_m$ , the more visible the time effect is, with the vertical lines corresponding to the times that the 5 axles of the semi-trailer passed over the pavement joint.

Table 1: Values of the limits of the quality factor  $Q_m^{(min)}$  and  $Q_m^{(max)}$  from Eq. (15) corresponding to the three frequency peaks in the FT of the signal generated by the passage of a semi-trailer over the joint.

Relevant frequency		
$f_j$ (Hz)	$Q_m^{(min)}$	$Q_m^{(max)}$
250	2	125
600	5	301
900	7.5	452

Another point of interest, also related to  $Q_m$  is the *edge-effect* problem. The problem is attributed to the finite length of the signal, the discretization of the measured data record and to the nature of the CWT. Edge effects cannot be removed and there will always be a domain in the time-frequency plane that should be neglected due to this problem. Edge effects are seen with dashed lines in Figs. 3 and 4; the effect of the quality parameter  $Q_m$  on edge effects is evident. As the  $Q_m$  value increases, the edge effects become more significant, thus the useful time interval for modal identification is reduced but the decoupling of the modes is more effective.

In reference [14], based on a study on simulated digital signals, the authors concluded that the choice of  $Q_m$  is a compromise between two objectives, (i) to discriminate close frequencies (in which case larger values of  $Q_m$  are preferred), (ii) to adopt a correct temporal resolution

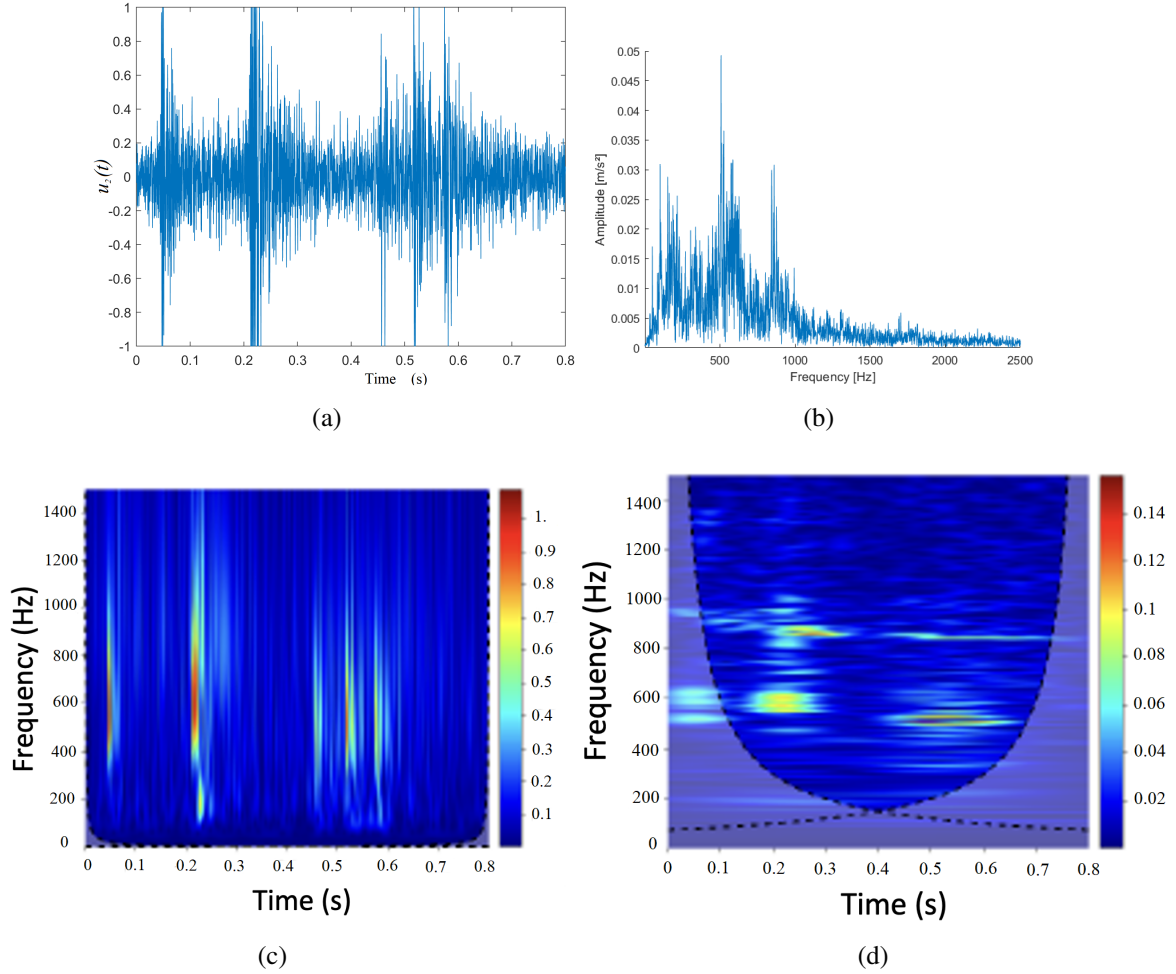


Figure 4: Acoustic response  $u_2(t)$  of a joint of pavement on a motorway bridge measured when a semi-trailer passes by: (a) time response, the amplitude is normalised between  $[-1, 1]$ , note that on this recording, the passage of a lorry saturates the microphone, (b) amplitude spectral density of  $u_2(t)$  over the range  $[0, 2500]$ Hz, (c) Scalogram of  $u_2(t)$  computed with  $Q_m = Q_m^{(min)} = 2$  for  $f_j = 250$ Hz and (d) Scalogram of  $u_2(t)$  computed with  $Q_m = Q_m^{(max)} = 125$  for  $f_j = 250$ Hz.

for the detection of abrupt frequency changes (in which case smaller values of  $Q_m$  are more effective).

In conclusion, the computation of CWT of Eq. 7 can be performed for  $f = f_\psi^0 / a = n / (2\pi a)$  within a chosen frequency band and with an adapted time-frequency resolution  $Q_m$ . This remark led Rouby *et al.* [21] to introduce an adapted CWT:  $T_\psi[u](f_\psi^0 / f, b)$  with a time-frequency resolution varying for each frequency, leading to a uniform resolution in the whole time-frequency plane.

### 3 ASYMPTOTIC SIGNALS - RIDGE DEFINITION AND EXTRACTION

#### 3.1 One component signal

Several authors [4, 5, 22, 23] have worked on the use of CWT for so-called asymptotic signals. The typical example is the chirp model [4, 5] which is a pseudo-periodic signal modulated in frequency around a carrier frequency and also modulated in amplitude by an envelope whose

variations are slow compared with the oscillations of the phase [23]. The CWT results that are obtained in the context of asymptotic signals are very useful in particular for modal analysis [13]. The first step is to uniquely associate to any real signal  $u(t)$  a canonical amplitude  $A^{(u)}(t)$ , assumed to be positive, and a phase  $\theta^{(u)}(t)$  assumed to be increasing, and satisfying:  $u(t) = A^{(u)}(t) \cos(\theta^{(u)}(t))$ .  $A^{(u)}(t)$  and  $\theta^{(u)}(t)$  can be determined with the aid of the Hilbert transform  $\mathcal{H}[u]$  which is defined in the Fourier domain as:  $\overline{\mathcal{H}[u]}(f) = -i \operatorname{sgn}(f) \widehat{u}(f)$  and allows the definition of the analytical signal  $Z_u(t)$  associated to  $u(t)$  such as:

$$Z_u(t) = u(t) + i\mathcal{H}[u](t) = A^{(u)}(t) e^{i\theta^{(u)}(t)} \quad (16)$$

Several authors have showed that the CWT of an asymptotic signal will tend to “concentrate” in the neighborhood of a curve  $a_1(t)$  called “ridge” that consists of an aggregation of points called ridge points. The ridge points are commonly obtained either from the CWT modulus of the signal, or from its phase, and are called *amplitude ridge points* and *phase ridge points*, respectively [24].

In the time-scale map, a ridge can be defined (see reference [23]) from its canonical phase  $\theta^{(u)}$  by:

$$a_1(t) = \frac{2\pi f_\psi^*}{\dot{\theta}^{(u)}(t)}, \quad (17)$$

where  $f_\psi^*$  appears in the chosen mapping between scale and frequency previously detailed. The restriction of the CWT to the ridge  $a_1(b)$ , is called the “skeleton” of the wavelet transform. From the skeleton, it is possible to reproduce the signal, or more precisely, what is associated with the analytical signal  $Z_u(b)$ , while it behaves like the product of  $Z_u(b)$  by a multiplicative factor entirely characterised by the mother wavelet and the ridge  $a_1(b)$ :

$$T_\psi[u](a_1(b), b) = \frac{1}{2} \overline{\psi} \left( a_1(b) \dot{\theta}^{(u)}(b) \right) Z_u(b) = \frac{1}{2} \overline{\psi} (2\pi f_\psi^*) Z_u(b). \quad (18)$$

The process of estimating the ridge from the absolute value and/or from the phase information of the CWT of the signal is called “ridge extraction”. Different techniques for extracting ridges exist [4] and can be classified into two categories: the “differential” and the “global” methods [13].

Differential methods rely on local properties of the CWT of the signal  $u(t)$ , they are verified theoretically on the ridge curve and they are based on the partial differential equations of the CWT. The differential method used here is based on the modulus of  $T_\psi[u](a, b)$ , which is maximum at  $b$  in the vicinity of the ridge, and therefore verifies a cancellation of its partial derivative at  $a$ . This definition is given in [24] and its implementation has the advantage that it is particularly simple and stable since it is a simple search for maxima.

The global methods, introduced in reference [25], are based on the search for curves that maximize the energy of the CWT while maintaining a certain regularity of the solution. When the considered frequency and amplitude modulated signal is embedded in noise and near the ridges, the contribution of the signal is much larger than that of the noise, while the wavelet transform of the noise spreads in the whole time-frequency plane. Several algorithms for global ridge extraction are detailed in the book by Carmona *et al.* [4] and are discussed with reference to their robustness to noise.

Once the ridge extraction method has been chosen and the ridge has been determined, the analytical signal  $Z_u(b)$  can be obtained. Its real and imaginary parts give the signal and its Hilbert transform, respectively. The final goal of ridge extraction is to get an estimate of  $\dot{\theta}_u(t)$  from Eq. 17 and then of  $Z_u(b)$  feeding it back into Eq. 18.

### 3.2 Multicomponent signals

The analytical expression of the structural responses of linear systems is well-known, even in the case of non-proportionally damped systems [26]. The aim of this section is to characterise the behaviour of structures from multi-channel dynamic signals obtained from measurements made by a set of  $N$  sensors, typically accelerometers. The set of displacement measurements at these sensor points is grouped into the vector:  $\mathbf{u} = [u_1, u_2, \dots, u_N]^T$ . Note that this notation is generic and can also be used when the signal is an acceleration. The modal decomposition expresses the state equation as a linear combination of the various modes of the system. Thanks to the modal decomposition approach for linear systems, every signal can be expanded as a linear combination of the different modes of the system, e.g.  $M$  components, each corresponding to a different Eigenmode of the structure [26]. The CWT of each component of  $\mathbf{u}$  is also grouped in the vector  $\mathbf{T}_\psi[\mathbf{u}]$ , as follows:

$$\mathbf{T}_\psi[\mathbf{u}](a, t) = [T_\psi[u_1](a, t), T_\psi[u_2](a, t), \dots, T_\psi[u_N](a, t)]^T \quad (19)$$

Therefore, the displacement  $u_k(t)$ , taking into account the  $M$  modes, can be obtained as:

$$u_k(t) = \Re \sum_{l=1}^M \left\{ A_l^{(u_k)}(t) e^{i\theta_l^{(u_k)}(t)} \Phi_k^{(l)} \right\}, \quad (20)$$

where  $A_l^{(u_k)}(t) e^{i\theta_l^{(u_k)}(t)}$  is the analytical modal participation factor of the  $l$ -th complex mode  $\Phi^{(l)}$  ( $\Phi_k^{(l)}$  being its  $k$ -th component) to the structural response  $\mathbf{u}_k(t)$ , while its real part are assumed to be asymptotic<sup>6</sup>.

The vector  $\phi^{(l)}$  is the complex  $l$ -th mode, which we have chosen to normalise as  $(\phi^{(l)})^T \phi^{(l)} = 1$ , based on the generalisation of a criterion usually used for real modal deformations:  $\|\phi^{(l)}\|^2 = 1$  (cf. Carpine [10]). In the case of free responses, the dynamic signals contain the vibrations of each mode of the structure, associated with an exponentially damped sinusoidal component. Thus, Eq. (20) becomes:

$$u_k(t) = \Re \sum_{l=1}^M \left\{ A_l^{(u_k)}(t) e^{i\theta_l^{(u_k)}(t)} \Phi_k^{(l)} \right\} = \Re \sum_{l=1}^M \left\{ Z_l^{(u_k)} e^{i\lambda_l t} \Phi_k^{(l)} \right\}, \quad (21)$$

where  $Z_l^{(u_k)}$  is a complex constant,  $\lambda_l = 2\pi i f_l \sqrt{1 - \xi_l^2} - \xi_l 2\pi f_l$  is the  $l$ -th pole and  $f_l, \xi_l$  are the Eigenfrequency and the modal damping ratio associated with mode  $l$ , respectively.

With the aid of the above form, which is a sum of asymptotic amplitude and frequency modulated components, in the case of a single asymptotic amplitude and frequency modulated signal, the absolute value of its CWT tends to concentrate near the ‘‘ridges’’ of the transform [4]. In the time-frequency plane, the ridge is a well-defined region, but most importantly, the wavelet transform acts as a ‘‘regularizing’’ filter that concentrates the information that is carried within the signal and hence allows to characterize the instantaneous frequencies.

Based on the discussion above, the linearity of the CWT and a good choice of the mother wavelet  $\psi$  can allow to separate these different components and extract the ridge for each of them. There are several approaches for detecting multiple ridges, the final choice depends on their interaction, or their independence. When these ridges do not interact and are located at

<sup>6</sup>the real part of  $u_k$  is equal to  $A_l^{(u_k)}(t) \cos(\theta_l^{(u_k)}(t))$

distinct regions of the time-frequency plane, a frequently encountered case in the analysis of dynamic signals for which each instantaneous frequency remains in the vicinity of a horizontal straight line, the methods previously mentioned can be implemented as in [13, 27].

As previously discussed, the ridges can then be deduced using Eq. 17. A common problem is when there are two close Eigenfrequencies,  $f_j$  and  $f_k$ . The possibility of extracting the ridges is guaranteed if the following condition is satisfied:

$$\xi_j \leq \frac{\sqrt{2} |f_j - f_k|}{c_f f_j}, \quad (22)$$

where  $c_f$  is a constant related to edge effects [28],  $\xi_j$  is the modal damping ratio of the  $j$ -th mode and the  $j$  and  $k$  indexes correspond to two neighboring modes  $j$  and  $k$ , respectively. If the condition of Eq. 22 is not met, the extraction of accurate ridges is not guaranteed and may even not be possible since the modes are too close and they cannot be separated in the case that they are heavily damped. This tells us that two close modes cannot be well separated from each other if they are damped too much. Figure 5 plots the  $\xi_j$  values as function of  $f_j$  and  $\Delta f = |f_j - f_k|$  for  $c_f = 3$ . If the condition of Eq. 22 is met, the extraction of accurate ridges is guaranteed, provided that the quality factor  $Q_m$  is appropriately chosen, as already discussed in Eq. 13 and Section 2.1.

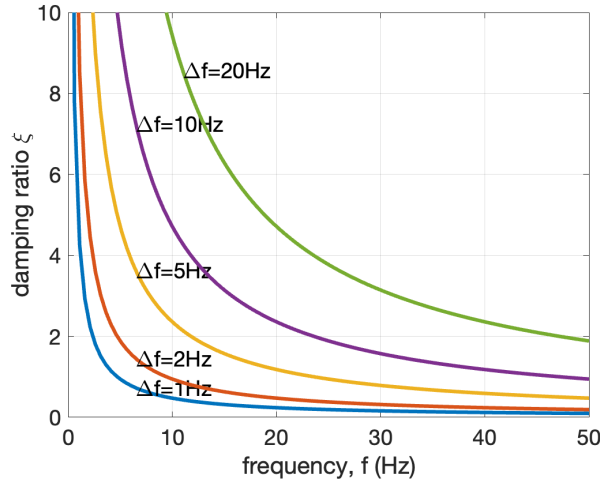


Figure 5: Relationship between damping ratio and frequency for ridge identification according to Eq. 22 ( $c_f = 3$ ).

As already discussed, at each measurement point  $u_k$ ,  $k \in [1, N]$ , a set of ridges  $a_k^{(l)}(t)$  can be extracted from the time-frequency plot, for the  $l$ -th mode excited by the shock, where  $l = 1, \dots, M$ . For the extraction of ridges, differential methods based on a local analysis of the extrema of the CWT modulus are here preferred. Thus the ridges, or the instantaneous frequencies  $a_l^{(k)}(t)$ , are extracted by the computation of local maxima of  $|T_\psi[u_k](a, t)|$  as function of time  $t$  and for the  $l$ -th mode excited by the shock:

$$a_l^{(k)}(t) = \arg_{a} \max |T_\psi[u_k](a, t)|. \quad (23)$$

So, for a mode  $l$ , a set of  $k = 1, \dots, M$  ridges  $a_l^{(k)}(t)$  is obtained, and a procedure to retain only one ridge for the instantaneous frequency must be then made, in [13], for example assuming the

average of the  $k$  signals:

$$a_l^{\text{mean}}(t) = \frac{1}{N} \sum_{k=1}^N a_l^k(t) \quad (24)$$

An alternative way of obtaining a single ridge  $a_\Sigma^{(l)}(t)$  for mode  $l$  instead of several ridges  $a_l^{(k)}(t)$  (one for each measurement point) was recently been proposed by the authors in [8]. This new procedure is based on the computation of the Averaged Continuous Wavelet Transform (ACWT)  $\widetilde{T}_\psi[\mathbf{u}](a, t)$ , which combines the CWTs of each measurement point  $u_k$  ( $k \in [1, N]$ ), as follows:

$$\widetilde{T}_\psi[\mathbf{u}](a, b) = \sum_{k=1}^N [T_\psi[u_k](a, b)]^2. \quad (25)$$

Eq. (25) is suitable for ridge extraction in the case of free responses of systems whose Eigenvectors are real or with a negligible imaginary part, which is the case of weakly damped systems [26] and practically refers to all civil engineering structures. In fact, the use of the squares of the transforms makes it possible to orient the contributions of the useful signal in each measurement channel according to the same orientation in the complex plane, while those of noise remain a priori randomly distributed. Therefore, Eq. (25) allows to obtain a single ridge  $a_\Sigma^{(l)}(t)$  for mode  $l$  as follows:

$$a_\Sigma^{(l)}(t) = \underset{a}{\text{arg max}} \left| \left[ \widetilde{T}_\psi[\mathbf{u}](a, t) \right] \right|^{\frac{1}{2}} = \underset{a}{\text{arg max}} \left| \sum_{k=1}^N [T_\psi[u_k](a, t)]^2 \right|^{\frac{1}{2}}. \quad (26)$$

Finally, for both definitions of the ridges, i.e.  $a_l^{\text{mean}}(t)$  (Eq. 23), or  $a_\Sigma^{(l)}(t)$  (Eq. 26), a procedure has been proposed in [8] to smooth the result of the maxima obtained for each time  $t$ ; for two neighbouring points in frequency of a maximum to the right and to the left make it possible to perform a parabolic interpolation from which we obtain the coordinates of a new ridge in the time scale plane for the mode  $l$  under consideration  $(t, \check{a}_k^{(l)}(t))$ , or  $(t, \check{a}_\Sigma^{(l)}(t))$ . A procedure for chaining the discrete points of the time-frequency plane to transform them into ridges is finally performed (cf. [4]). The set of maxima of the absolute value of the CWT along the ridges present in the signal forms the skeleton of the CWT of the signal. According to the definition chosen for a ridge  $l$  in eq. (23) or in eq. (26), the absolute value of the CWT along each ridge  $T_\psi[u](\check{a}_k^{(l)}(t), t)$  or the ACWT along the single ridge  $\widetilde{T}_\psi[\mathbf{u}](\check{a}_\Sigma^{(l)}(t), t)$  is preferred.

#### 4 SYSTEM DYNAMIC CHARACTERIZATION

The modal parameters of a system under transient vibrations can be identified by extracting the ridges and the skeleton of the CWT time-frequency representation. Once the ridges of the CWT have been extracted, the instantaneous frequencies, the modal damping ratio and the modal shapes can be estimated. If the system is purely linear, the shape of the ridges associated with the Eigenfrequencies will be a straight horizontal line. Furthermore, the damping ratio of the modes of interest can be estimated from the exponential decrease in amplitude associated with these ridges, and finally the modal shapes can be obtained from the relative amplitudes and phase shifts between the channels corresponding to the different sensors.

Therefore, the extraction of the ‘‘ridges’’ is a critical aspect for the successful application of the CWT for modal identification in the case of transient structural responses. Among the authors who have used the CWT for modal identification from transient structural responses, we

can cite two references published in 1997: Staszewski [29] and Ruzzene *et al.* [30]. Staszewski [29] proposed several CWT-based methods for estimating damping ratios and applied them to simulated multi Degree-of-freedom (DoF) systems. Ruzzene *et al.* [30] showed that the CWT analysis of the free response of a system allows the estimation of its natural frequencies and viscous damping ratios. A more complete procedure, which also gives access to frequencies and modal shapes, can be found in Lardies & Gouttebroze [31]. In Le & Argoul [13], the authors propose a more precise and complete method where the choice of the mother wavelet, its quality factor and the management of the edge effects of the TOC are studied in depth. The subsequent article by Erlicher & Argoul [27] discusses the use of this procedure in the case of systems with non-proportional damping, and therefore in the presence of complex deformations.

For amplitude and phase modulated signals of the form:  $u(t) = A(t) \cos(\phi(t))$ , the restriction of the wavelet transform to its ridge behaves mainly as the associated complex signal of  $u(t)$ :  $A(t) \exp[i\phi(t)]$ . This representation also allows the reconstruction of such original signals in non-significant noise situations [4]. If the system behaviour is close to be linear, from the CWT (or the ACWT) of its transient responses, the extracted ridges are similar to horizontal lines and the associated skeleton has an exponential decrease [13]. The logarithm  $\log |\widetilde{T}_\psi[\mathbf{u}](\check{a}_k^{(l)}(t), t)|$  for the CWT, or  $\log |\widetilde{T}_\psi[\mathbf{u}](\check{a}_\Sigma^{(l)}(t), t)|$  for the ACWT can be then deduced and the calculation of the slope of the “straight lines” for each mode  $l$  can be performed in order to estimate the corresponding modal damping ratio of the  $l$ -th mode. The slope allows to get an estimate to the near sign of the product  $2\pi f_l \xi_l$  that is the reciprocal of the time constant characterizing the exponential decay of the  $l$ -th mode.

The calculation of the Eigenshapes requires a set of measurements grouped in the vector:  $\mathbf{u} = [u_1, u_2, \dots, u_N]^T$ . The CWT of each component of  $\mathbf{u}$  along the smoothed ridge  $\check{a}_\Sigma^{(l)}$  of mode  $l$  are also grouped in the vector  $\mathbf{T}_\psi[\mathbf{u}]$ , as follows:

$$\mathbf{T}_\psi[\mathbf{u}](\check{a}_\Sigma^{(l)}(t), t) = [T_\psi[u_1](\check{a}_\Sigma^{(l)}(t), t), T_\psi[u_2](\check{a}_\Sigma^{(l)}(t), t), \dots, T_\psi[u_N](\check{a}_\Sigma^{(l)}(t), t)]^T \quad (27)$$

Depending on the choice made for the definition of the ridge, i.e. either Eq. 23, or Eq. 26, the instantaneous complex modal shapes  $\phi^{(l)}(t)$  can be derived from the relative amplitude and phase of the CWT calculated along each ridge. There are several ways to normalise the modal vector. One way is to choose the unit amplitude for the measurement point  $u_{\max}$ , where the max index corresponds to the measurement point where the modal amplitude is greatest [13]. The  $k$ -th component  $\phi_k^{(l)}(t)$  of the “instantaneous” complex mode can be expressed as follows:

$$\phi_k^{(l)}(t) = \frac{T_\psi[u_k](\check{a}_k^{(l)}(t), t)}{T_\psi[u_{\max}](\check{a}_k^{(l)}(t), t)}. \quad (28)$$

As already discussed above, in this work we prefer the following scaling condition:  $(\phi^{(l)}(t))^T \phi^{(l)}(t) = 1$  for instantaneous mode shape  $\phi^{(l)}(t)$  [8, 10]. This definition results to:

$$\phi_k^{(l)}(t) = \pm \frac{T_\psi[u_k](\check{a}_\Sigma^{(l)}(t), t)}{[T_\psi[\mathbf{u}](\check{a}_\Sigma^{(l)}(t), t))^T T_\psi[\mathbf{u}](\check{a}_\Sigma^{(l)}(t), t)]^{1/2}}, \quad (29)$$

where the sign follows the continuity of  $\phi_k^{(l)}$  over time. The amplitude of mode  $l$  is then equal to:

$$A_l(t) = |T_\psi[\mathbf{u}](\check{a}_\Sigma^{(l)}(t), t)^T T_\psi[\mathbf{u}](\check{a}_\Sigma^{(l)}(t), t)|^{1/2} \quad (30)$$

To obtain a “constant” mode, especially in the case of linear behaviour, the mean value over time, denoted  $\bar{\phi}$ , can be calculated for each component  $k$  of the  $l$ -th mode:

$$\bar{\phi}_k^{(l)} = \frac{1}{t_f - t_i} \int_{t_i}^{t_f} \phi_k^{(l)}(t) dt. \quad (31)$$

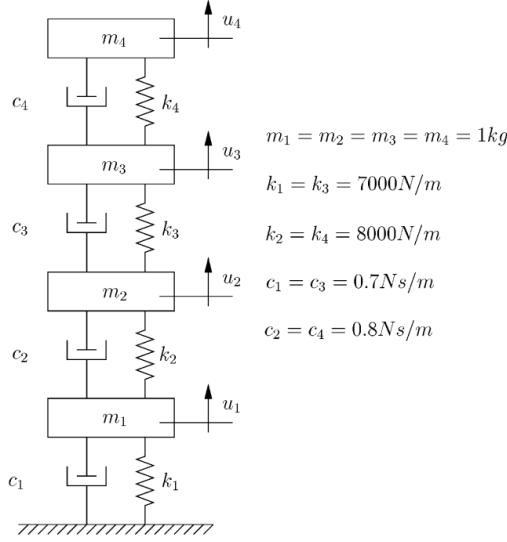


Figure 6: Characteristics of the four DoF system [13].

## 5 Applications

### 5.1 Numerical example: Transient response of a four DoF structure

The first example is the four DoF mass spring-damper model originally studied by Le and Argoul [13] and revisited here. The characteristics of the system are shown in Figure 6. The structure is displaced from original position by imposing initial displacements:  $u_1 = 1.00\text{m}$ ,  $u_2 = 0.75\text{m}$ ,  $u_3 = 0.50\text{m}$  and  $u_4 = 0.25\text{m}$  and zero velocities ( $\ddot{u}_1 = \ddot{u}_2 = \ddot{u}_3 = \ddot{u}_4 = 0$ ). The system is left to freely oscillate after the initial displacements. The sample of duration  $L = 5$  sec is taken over  $M = 1024$  points (sampling period  $T = 0.0049$  sec). The wavelet analysis is performed using the Cauchy wavelet, while for the edge effect  $c_t = c_f = 3$  is assumed.

Figure 7 shows the displacement response histories of the four DoF system. Furthermore, Figure 8 shows the modulus of the squared ACWT for three  $Q_m$  values, i.e.  $Q_m = 5, 15$  and  $30$ . The large  $Q_m$  value ( $Q_m = 30$ ) allows to clearly see the Eigenfrequencies, but the edge effects become more prominent, since the value  $Q_m = 30$  exceeds the maximum value  $Q_m^{(max)}$  according to Eq. 15 (see also Table 2).

Figure 9a shows the squared ACWT for  $Q_m = 20$  and the four ridges identified, i.e.  $a_{\Sigma}^{(i)}(t)$ ,  $i = 1, 2, 3, 4$ . Due to the linear response, the ridges are straight, while due to the large  $Q_m$  value, the ridge identification of the first mode is restricted to a narrow region by the edge effects. Moreover, Figure 9b shows the plots of  $\widetilde{T}_{\psi}[\mathbf{u}](a, b)$ , i.e. of the squared ACWT ridges versus time and the corresponding ridges  $\alpha_{\Sigma}^{(l)}(t)$ . The slope of each curve provides the modal damping ratio. Notice that, since the ACWT is squared (see Eq. 25), the slope should be divided by 2. Finally, Figure 10 shows the four averaged Eigenmodes  $\bar{\phi}$  obtained after applying Eq. 29



and Eq. 31. The integration limits  $[t_f, t_i]$  of Eq. 31 are determined by the edge effects for the frequency of interest.

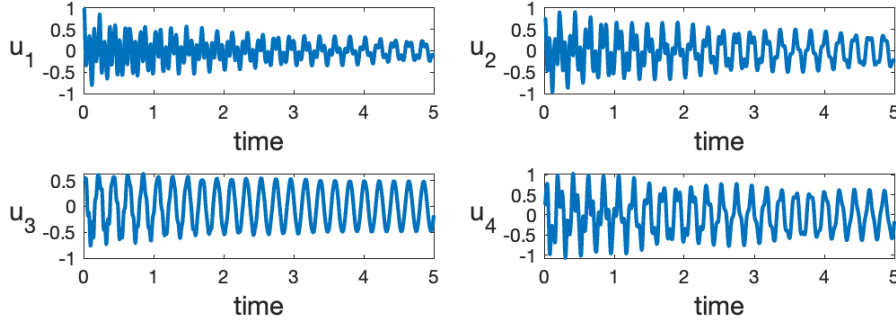


Figure 7: Response histories for the four-DoF system.

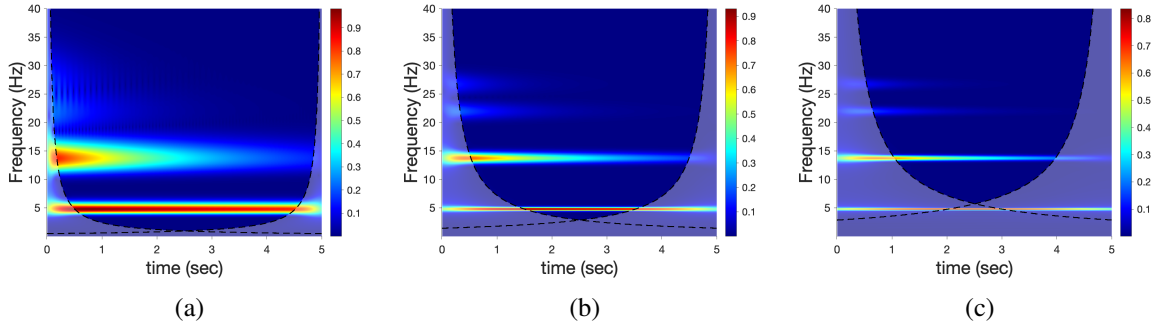


Figure 8: Scalogram  $|\widetilde{T}_\psi[\mathbf{u}](a, b)|$  computed for: (a)  $Q_m = 5$ , (b)  $Q_m = 15$ , (c)  $Q_m = 30$ .

Table 2 summarizes the results of this first example. The first three columns provide the minimum and maximum values of the  $Q_m$  parameter obtained with Eq. 15. Since the problem is simulated, the exact Eigenfrequencies and the modal damping ratio values have been exactly calculated. Table 2 compares the exact values with those obtained with the aid of the proposed ACWT approach, proving its accuracy and efficiency.

Table 2: Identified modal characteristics using the ACWT method and compared with analytical characteristics.

Mode	$\Delta f$ (Hz)	$Q_m^{min}$	$Q_m^{max}$	Eigenfrequency $f_i$ (Hz)		Damping ratio $\xi_i$ (%)	
				exact	identified	exact	identified
1	4.7397	1.50	24.9	4.7397	4.7430	0.15	0.17
2	8.8503	2.3	71.4	13.5900	13.6915	0.43	0.42
3	7.9135	4.08	112.9	21.5035	21.9165	0.68	0.66
4	4.4379	8.8	136.3	25.9414	26.6851	0.81	0.79

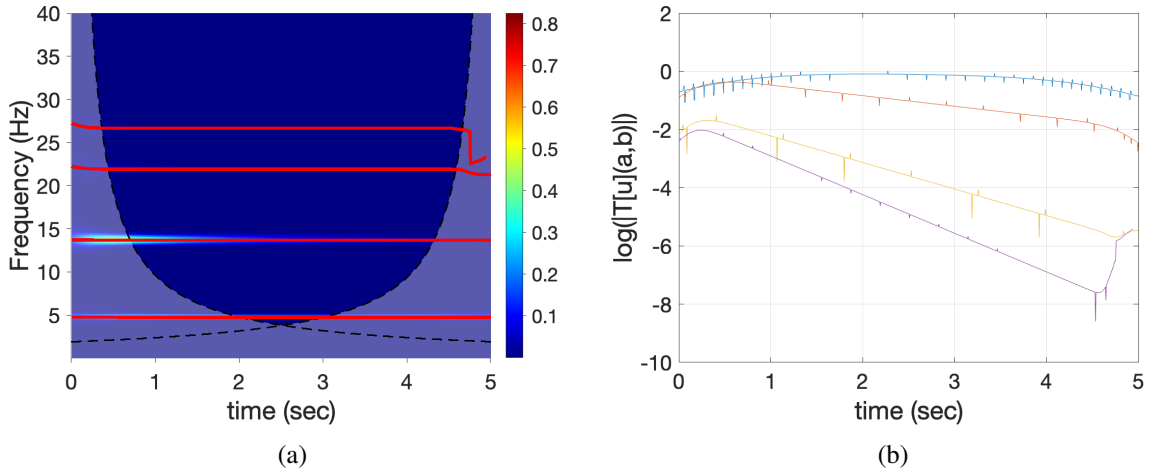


Figure 9: (a) Scalogram  $\left| \widetilde{T}_\psi[\mathbf{u}](a, b) \right|$  (computed with:  $Q_m = 20$ ) and ridges identified, (b) damping calculation from the slope of  $\log$  of the ridge. Note that since the ACWT transform is the sum of transforms that are in the power of two (Eq. 25), the slope of  $\log \left( \left| \widetilde{T}_\psi[\mathbf{u}](a, b) \right| \right)$  has to be divided by 2. From top to bottom:  $f_i = 4.7, 13.7, 21.9, 26.6$  Hz.

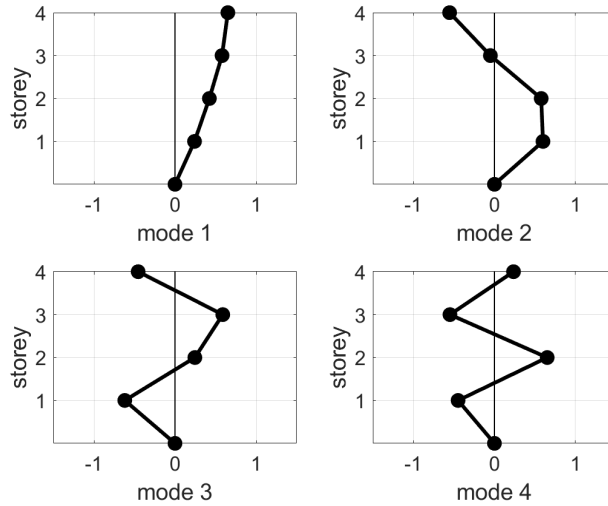


Figure 10: The four identified Eigenmodes obtained using Eq. 29.

## 5.2 “Damaged” masonry wall under dynamic excitation

The second example focuses on vibration data that were acquired during an experimental campaign carried out on masonry wall specimens. The tests were performed in LabSCo laboratory of IUAV University, Venice [32]. These data were used to apply the CWT method for the dynamic identification through transient responses embedded into noise responses.

The experimental campaign was carried out on UnReinforced Masonry (URM) panels subjected to Shear-Compression (SC) and ambient-vibrations tests using a system of nine accelerometers (Fig. 11). The test specimens are ordinary URM walls with Flemish bond pattern as masonry arrangement. The unit dimensions of a brick are:  $250 \times 120 \times 55$ mm and each wall is 150cm tall, 130cm width and 25cm thick, with 1cm for the thickness of mortar joints.

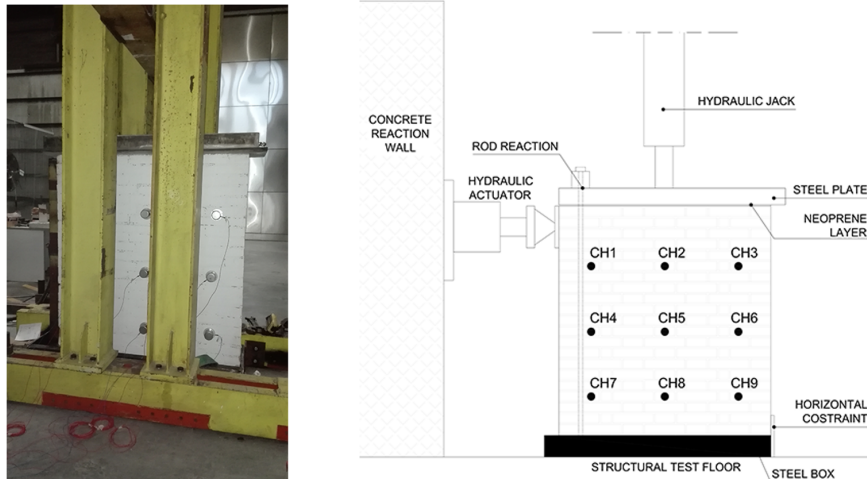


Figure 11: Masonry wall test. Location of the nine accelerometric sensors.

Among the types of URM panels tested by the Italian team, including the undamaged walls already analysed by the wavelet transform method in [8, 33], only those with the fired brick masonry walls have been processed here and are compared with the undamaged masonry wall at the loading step, immediately after the appearance of the diagonal crack, shown in Figs. 12a and 12b for the undamaged and damaged wall, respectively. A core of bricks that were exposed to high temperature were put in the wall arrangement, aiming to reproduce discontinuities and irregularities of the wall cross-section as they can be found in existing walls and buildings. In particular, 12 brick units were subjected to a cycle of exposition to high temperature ( $600^{\circ}\text{C}$ ) for 40 minutes and they were cooled at ambient temperature as shown in Figure 12b. The final modified panel can be seen as a “damaged” panel where the so-called “damage” is controlled by the number of fired bricks, their location and their firing temperature.

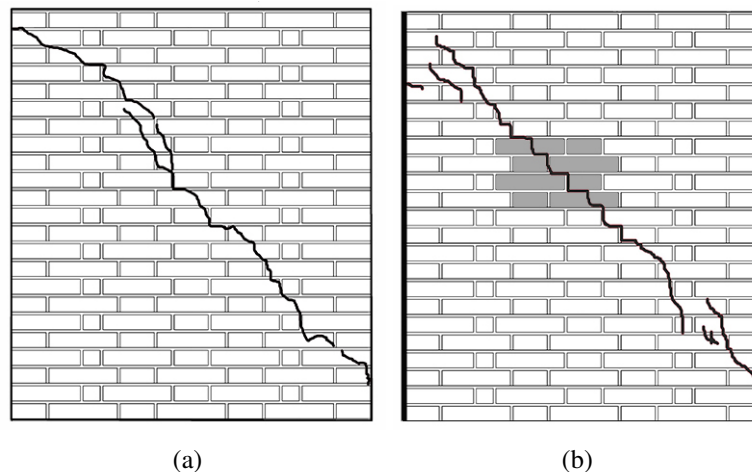


Figure 12: Photo of the wall at the loading step, immediately after the appearance of the diagonal crack, (a) “Undamaged” masonry wall, and (b) “Damaged” wall with the group of 12 fired bricks (6 for each side) drawn in hatched lines.

The signals acquired, contain some discontinuities in the waveform that can be assumed as transient responses, i.e. change in the steady-state of the system. These responses are probably

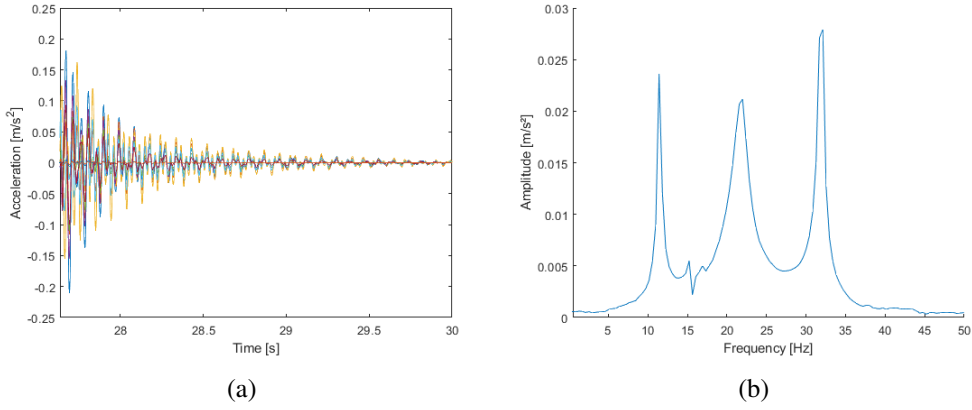


Figure 13: “Damaged” masonry wall - After impact responses : (a) The nine “transient” acceleration signals data set used for the processing, and (b) Averaged Fourier Transform (Eq.32).

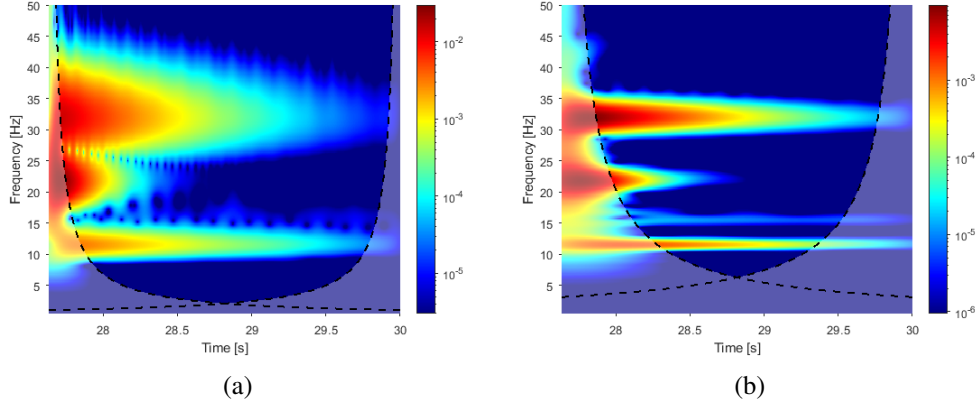


Figure 14: “Damaged” masonry wall testing. Scalograms of the ACWT for the nine acceleration signals. Effect of the quality factor: (a)  $Q_m = 5.42$ , and (b)  $Q_m = 18.06$ .

due to the testing environment triggered by other testing machines. The set of a “transient” acceleration signals  $\ddot{u}_k(t)$  data set  $k \in [1, 9]$  plotted in Figure 13a, has been processed using the CWT method. An Averaged Fourier Transform (AFT)  $\tilde{\ddot{u}}(f)$  combining the Fourier transform of each of the  $N = 9$  channel acceleration signals is introduced, defined by the square root of the sum of the squared FT of each individual acceleration signal by:

$$\tilde{\ddot{u}}(f) = \left( \sum_{k=1}^N \widehat{\ddot{u}}_k^2(f) \right)^{\frac{1}{2}}, \quad (32)$$

and its absolute value is plotted in Figure 13(b). The effect of the quality factor is illustrated in Figure 14, where the absolute value of the square  $(\widehat{T}_\psi[\ddot{u}](a, b))^2$  of the ACWT given in Eq. (25) is plotted in the time-frequency domain for two different quality factor values  $Q_m^{(1)} = 5.42$  and  $Q_m^{(2)} = 18.06$ . For the first three modes, Eigenfrequencies  $f_i$  and modal amplitudes  $A_i$  are then estimated by the classical peak picking method applied to the AFT, the corresponding modal damping ratios are obtained with the classical half power technique (cf. [34, 35]) and listed in Table 3.

Table 3: “Damaged” masonry wall testing - After impact signals. - Identified modal characteristics (Eigenfrequency  $f_i(Hz)$ , modal amplitude  $A_i (m/s^2)$  and modal damping ratio  $\xi_i (\%)$ ) for the first three modes) by classical identification techniques (peak picking and half power) from the AFT versus the ACWT identification technique (indication of the quality factor  $Q_m$ ).

Mode	estimated with classical techniques from AFT			estimated with ACWT techniques			
	$f_i(Hz)$	$A_i (m/s^2)$	$\xi_i (\%)$	$f_i(Hz)$	$A_i (m/s^2)$	$\xi_i (\%)$	$Q_m$
1st	11.41	0.024	1.98	11.54	-	0.96	18.06
2nd	21.97	0.0212	4.25	22.12	-	3.07	18.18
3rd	32.11	0.0279	1.31	31.79	-	0.73	40.17

The ridges corresponding to the first three modes are plotted in Figure 15(a), and are similar to horizontal straight lines showing that the mechanical behaviour of the wall remains “linearized” during its response to the impact. Finally, the average value along each ridge allows to estimate the three first Eigenfrequencies whose values are given in Table 3. The two first modal characteristics have then been compared to those obtained with the undamaged wall during a similar step of loading after diagonal cracking. The calculation of the slope of the three “straight lines” is then performed and the modal damping ratios  $\xi_i$  for  $i \in [1, 3]$ , of the three first modes are deduced and given in Table 3. The values obtained are of the same order of magnitude but slightly lower than those obtained by the half-power method applied to the AFT given in Eq. (32). The three complex “constant” modal shapes are then computed from Eq. (31). The imaginary part of the three first complex modal shapes being quite small, thus only their real part is plotted in Figures 16(a), (b) and (c) for the undamaged wall and in Figures 17(a), (b) and (c) for the damaged wall, respectively. In both cases “undamaged” and “damaged”, the shape of these three identified modes is similar and appears to correspond to the first bending mode for the first mode identified, the first torsion mode for the second mode and a combination of the two for the third. The identified modal Eigenfrequency  $f_i$  and modal damping ratio  $\xi_i$  for the first three modes ( $1 \leq i \leq 3$ ), between the “undamaged” and “damaged” wall, estimated by ACWT techniques are given and compared in Table 4. It can be seen that the modal characteristics identified for the second mode (torsional type) for the “undamaged” wall are very different from those for the “damaged” wall (lower eigenfrequency and higher modal damping ratio).

Table 4: Comparison of identified modal characteristics (Eigenfrequency  $f_i(Hz)$ , and modal damping ratio  $\xi_i (\%)$ ) for the first three modes, between the “undamaged” and “damaged” wall, estimated by ACWT techniques with indication of the corresponding quality factor  $Q_m$ .

Mode	“undamaged” wall at horizontal load of 224.2kN			“damaged” wall at horizontal load of 186.79kN		
	$f_i(Hz)$	$\xi_i (\%)$	$Q_m$	$f_i(Hz)$	$\xi_i (\%)$	$Q_m$
1st	11.01	0.90	18.94	11.54	0.96	18.06
2nd	28.18	0.83	18.33	22.12	3.07	18.18
3rd	34.11	0.74	63.45	31.79	0.73	40.17

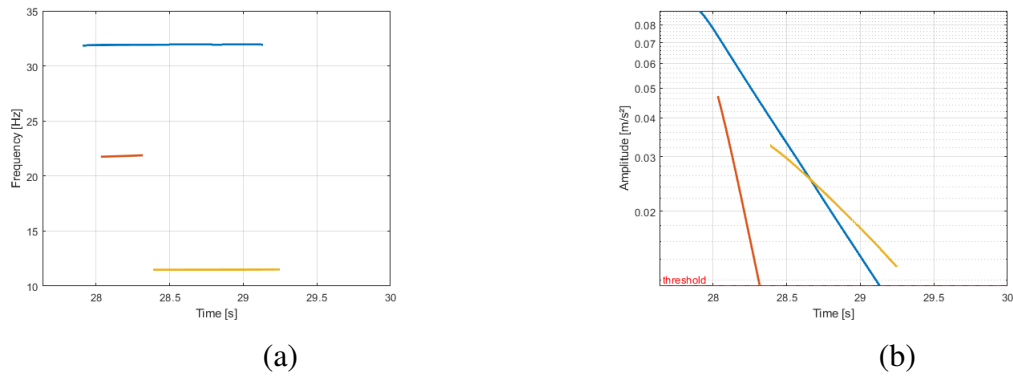


Figure 15: “Damaged” masonry wall testing - After impact signals. Extraction of ridges and skeletons from the ACWT: (a) instantaneous eigen frequencies (ridges), and (b) Logarithm of the absolute value of the ACWT along each ridge. For both figures, yellow curves correspond to the first mode, red curves to the second mode and blue curves to the third mode.

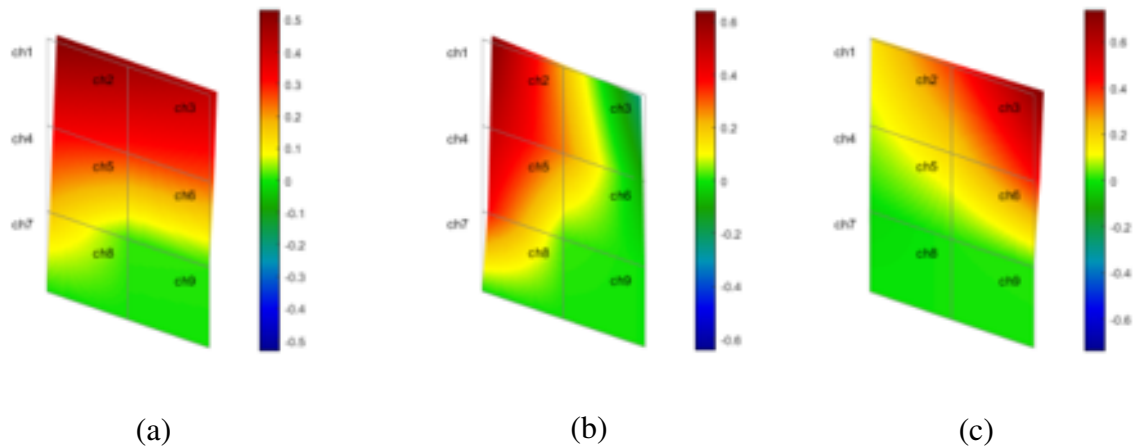


Figure 16: “Undamaged” masonry wall testing. Modes shapes: (a) First mode ( $f_1 = 11.01 \text{ Hz}$ ), (b) Second mode ( $f_2 = 28.18 \text{ Hz}$ ), and (c) Third mode ( $f_3 = 34.11 \text{ Hz}$ ).

## 6 CONCLUSIONS

The wavelet analysis method allows the processed signal to be represented in the time-frequency domain, thus facilitating access to the estimation of modal characteristics. Applied to transient signals, it allows the detection of spurious modes generated by noise or redundant DoF of the model, and the identification of the presence of close modes and their associated modal parameters. Two application examples are presented, showing that inverse CWT analysis can provide valuable information for structural modal identification and for evaluating the effect of damage on the modal characteristics of the structure under study. The next step is to apply it to damage assessment, as demonstrated by first studies to be published soon.

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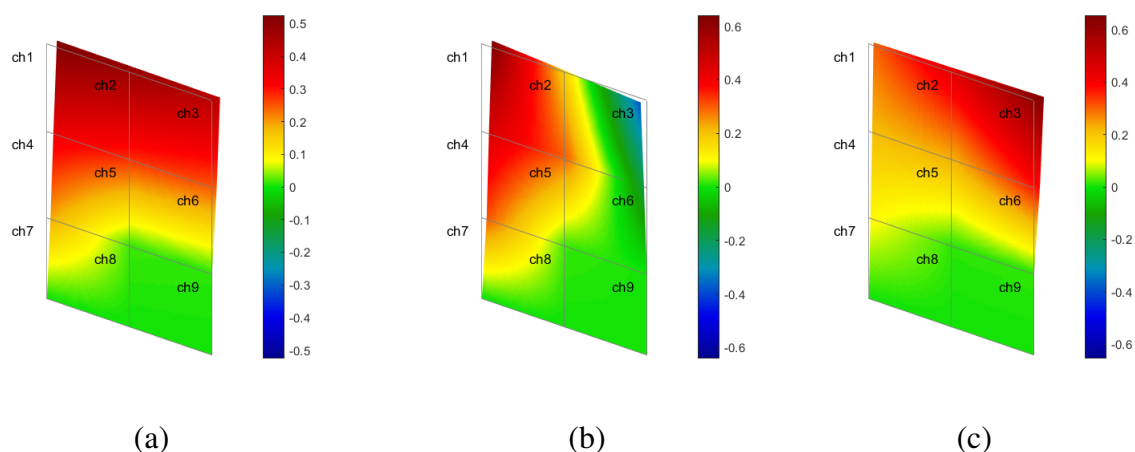


Figure 17: Damaged masonry wall testing. Modes shapes: (a) First mode ( $f_1 = 11.54 Hz$ ), (b) Second mode ( $f_2 = 22.12 Hz$ ), and (c) Third mode ( $f_3 = 31.79 Hz$ ).

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