



Dataset generation for drone optimal placement using machine learning

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Dataset Generation for Drone Optimal Placement Using Machine Learning

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2022
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Unmanned aerial vehicle (UAV), or drone is increasingly becoming a promising tool in communication system. This report explains the generation details of a dataset which will be used to designing an algorithm for the optimal placement of UAVs in the drone-assisted vehicular network (DAVN) [1]. The goal is to improve the drones' communication and energy efficiency after our previous work [2]. The report is organized as followed: the first section is devoted to the delay analysis of the vehicle requests in the DAVN using queuing theory; the second part of the report models the energy consumption of the drones while the third section explains the simulation scenario and dataset features. The notations and terminologies used in this report are summarized in the last section.

1 End to End Delay Analysis

This section analyzes the vehicle-to-drone (V2D) communication delay with the help of queuing theory. In fact, the total V2D delay of a class- i request from a vehicle- j , denoted as $E[W_{i,j}]$, consists of three parts:

- V2D propagation delay, $W_{i,j}^{V2D}$,
- queuing delay at the drone side, $W_{i,j}^s$,
- and drone-to-vehicle (D2V) propagation delay, $W_{i,j}^{D2V}$.

Thus, we have

$$E[W_{i,j}] = E[W_{i,j}^{V2D}] + E[W_{i,j}^s] + E[W_{i,j}^{D2V}] \leq T_i, \forall j \in \mathbb{V}, \quad (1)$$

where \mathbb{V} is the set of vehicles in the scenario. The transmission delays at both the vehicle and the UAV side are neglected since these values are very small compared to the propagation delays. It should be noted that if a request of class- i is not responded by the UAV after a certain time, T_i , where T_i is the maximum waiting time for request class- i , the request will be regarded as expired and will not be processed by the UAV later

Fig. 1 illustrates the queuing model for a single-drone-assisted vehicular network. The V2D and D2V communications are represented by the blue and grey dashed lines. At the mean time, the queuing model at UAV side is also highlighted. In the following sections, we will tackle the three delays in detail.

1.1 V2D and D2V Propagation Delay

The V2D and D2V propagation delays of a class- i request are calculated by

$$\begin{aligned} E[W_{i,j}^{V2D}] &= \frac{d_{i,j}}{r_i}, \\ E[W_{i,j}^{D2V}] &= \frac{d'_{i,j}}{r_i}, \end{aligned} \quad (2)$$

where $d_{i,j}$ and $d'_{i,j}$ are the distances between the UAV and the vehicle- j from which the request is sent during the V2D and D2V communication, r_i is the achievable propagation rate.

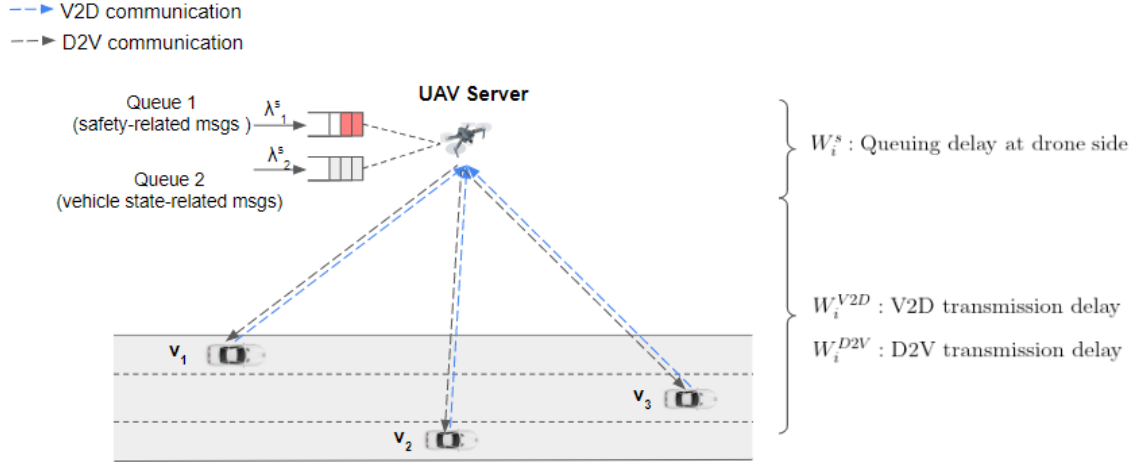


Figure 1: A simple DAVN with V2D and D2V communications.

1.2 Queuing Delay at the Drone's Side

Inside of this section, we relief the server index s and vehicle index j in the expressions for simplicity, since the delay should be the same for any vehicle in the scenario. However, outside this section, the two indexes are kept.

We consider two types of message that can be sent by the vehicles to the drone, one is safety-related message with high priority, the other is vehicle state-related message with low priority. When the vehicle faces a road risk, or an emergency vehicle, it will send a safety message to the corresponding drone. On the other hand, each vehicle update its state to the drone by periodically sending a vehicle state information containing vehicle's GPS position and kinematic parameters. It should be noticed that the length of safety-related messages varies and is assumed to follow an exponential distribution. On the contrary, the length of vehicle state information is a constant and never changes.

At the drone's side, the service of a low priority request (i.e. vehicle state information) can be interrupted by the arrival of a high priority request (i.e. safety message). That is to say, the queuing model is a priority queuing with preemption. In this case, the waiting times of the high priority requests, denoted as W_1 , are not affected by the low priority requests, and are only related to the arriving process and service process of requests of class-1. On the other hand, the waiting times of low priority requests, denoted as W_2 , are affected by high priority requests, with an additional waiting time due to the arrival and interruption from a high priority request.

1.2.1 Definitions and notations

We model the queuing delay of the safety message at the drones as a M/G/1 queue. The arriving process follows a Poisson process while the service time follows an exponential distribution. Moreover, the service times for different safety messages are independent and identically distributed. Note that for stability, it is required that the occupation rate $\rho_i = \lambda_i E[B_i]$ is less than one. The arrival process and service time distributions for safety-related messages and vehicle state-related

messages are shown in Table. 1

Table 1: The arrival process and service time distributions

	Arrival Process	Service Time Distribution
Safety Message	Poisson	Exponential
Vehicle State Information	Uniform	Constant

We define the following variables for the drone:

- ρ_i : the occupation rate of a class- i request;
- λ_i : the arrival rate of a class- i request;
- $E[B_i]$: the mean service time at the drone of a class- i request;
- $E[W_i]$: the mean waiting time in the queue of a class- i request;
- $E[R_i]$: the mean residual service time of a class- i request;
- $E[S_i]$: the mean sojourn time of a class- i request, note that $E[S_i] = E[B_i] + E[W_i]$;
- $E[L_i]$: the average number of requests of class i waiting in the queue.

Thus, for the total incoming traffic at the drone side, we have the following:

$$\lambda = \sum_{i=1}^n \lambda_i \quad (3)$$

$$E[B] = \sum_{i=1}^n \frac{\lambda_i}{\lambda} \cdot E[B_i] \quad (4)$$

$$\rho = \lambda \cdot E[B] \quad (5)$$

1.2.2 Mean waiting time of high priority requests

The average of W_1 can be expressed as followed:

$$E[W_1] = E[L_1]E[B_1] + \rho_1 E[R_1], \quad (6)$$

where L_1 denote the number of high priority request waiting in the queue.

According to Little's law we have

$$E[L_1] = \lambda_1 E[W_1]. \quad (7)$$

Combining the two equations yields

$$E[W_1] = \frac{\rho_1 E[R_1]}{1 - \rho_1}. \quad (8)$$

Since we have

$$E[R_1] = \frac{E[B_1^2]}{2E[B_1]}, \quad (9)$$

Equation (8) becomes

$$E[W_1] = \frac{\rho_1}{2(1-\rho_1)} \cdot \frac{E[B_1^2]}{E[B_1]}. \quad (10)$$

The sojourn time is then

$$E[S_1] = E[W_1] + E[B_1] = \frac{\rho_1}{2(1-\rho_1)} \cdot \frac{E[B_1^2]}{E[B_1]} + E[B_1]. \quad (11)$$

1.2.3 Mean waiting time of low priority requests

As explained before, the waiting time of low priority requests can be expressed as

$$E[W_2] = E[B_2] + E[W_+]$$

The customer has to first wait for the sum of the service times of all customers with the same or higher priority present in the queue plus the remaining service time of the customer in service. So

$$E[B_2] = \sum_{j=1}^2 E[L_j]E[B_j] + \sum_{j=1}^2 \rho_j E[R_j]. \quad (12)$$

On the other hand, the W_+ is related to all the higher priority requests arriving during its waiting time and service time. This leads to

$$E[W_+] = \lambda_1 E[W_1]E[B_1]. \quad (13)$$

Applying Little's law

$$E[L_2] = \lambda_2 E[W_2],$$

we have

$$E[W_2] = \frac{\sum_{j=1}^2 \rho_j E[R_j]}{(1 - (\rho_1 + \rho_2))(1 - \rho_1)}. \quad (14)$$

The mean sojourn time $E[S_2]$ of a class- i customer follows from $E[S_2] = E[W_2] + E[B_2]$, yeilding

$$E[S_2] = \frac{\sum_{j=1}^2 \rho_j E[R_j]}{(1 - (\rho_1 + \rho_2))(1 - \rho_1)} + E[B_2]. \quad (15)$$

Since we have

$$E[R_i] = \frac{E[B_i^2]}{2E[B_i]}, \quad (16)$$

Equation (17) finally becomes

$$E[S_2] = \frac{1}{(1 - (\rho_1 + \rho_2))(1 - \rho_1)} \cdot \sum_{j=1}^2 \rho_j \frac{E[B_i^2]}{2E[B_i]} + E[B_2]. \quad (17)$$

2 Energy Consumption Model

The energy consumed by the UAV- i to perform a complete data transfer is composed of two components: the communication energy that is used for the data transfer from the UAV to vehicles and other UAVs, E_i^c , and the propulsion energy of the UAV to adjust its location for data transfer, E_i^m , as in the following equations:

$$\begin{aligned} E_i &= E_i^c + E_i^m \\ &= \sum_{j \in \mathbb{V}} E_{i,j}^{D2V} + \sum_{m \in \mathbb{U}} E_{i,m}^{D2D} + E_i^m, \end{aligned} \quad (18)$$

where \mathbb{V} is the set of all vehicles in the communication range of the UAV- i , \mathbb{U} denotes the set of all UAVs. One can see that we consider both the drone-to-vehicle (D2V) and the drone-to-drone (D2D) communications. In the following sections, each of the components will be analysed.

2.1 Energy for communications

In order to model the communication energy consumption of our scenario, we first look at a simple scenario where the UAV- i is communicating with vehicle- j , as illustrated in Fig. 2. h_i is the height of the UAV- i , R is the communication range of the UAVs. $d_{i,j}^{euc}$ and $d_{i,j}^{hor}$ denote the Euclidean distance and the horizontal distance between the UAV- i and vehicle- j , respectively.

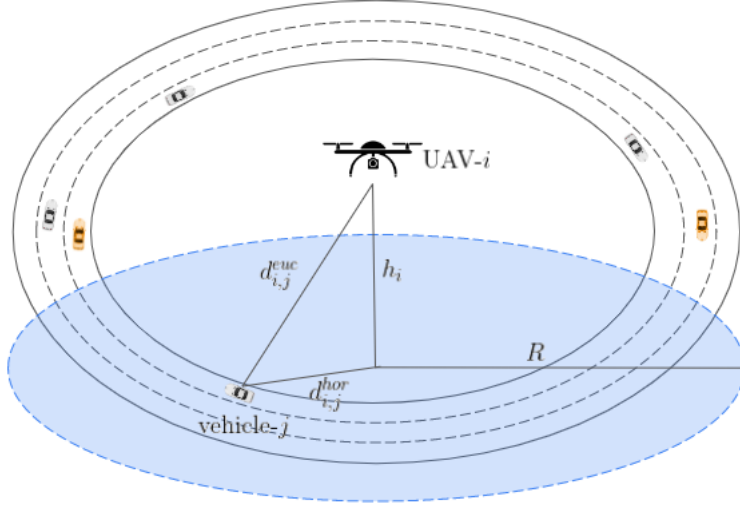


Figure 2: Scenario where the UAV- i is communicating with vehicle- j

Drone-to-vehicle path loss The D2V path loss is also known as air-to-ground path loss. In [3], the probability of having a line-of-sight link between the UAV- i and vehicle- j is formulated as followed:

$$P_{i,j}^{D2V}(LoS) = \frac{1}{1 + a \exp(-b(\theta_{i,j} - a))} \quad (19)$$

where a and b are environmental constant depending on rural or urban areas. θ is the elevation angle between UAV- i and vehicle- j , and it is equal to $\arctan(\frac{h_i}{d_{i,j}^{hor}})$. h_i is the height of UAV- i from ground level and $d_{i,j}^{hor}$ is the horizontal distant between the UAV- i and the vehicle- j .

Thus, the probability of non-line-of-sight loss is calculated as:

$$P_{i,j}^{D2V}(NLoS) = 1 - P_{i,j}^{D2V}(LoS) \quad (20)$$

Referring to [7], the path losses with LoS and NLoS links between UAV- i and vehicle- j can be written as:

$$PL_{i,j}^{D2V}(LoS) = 20 \log_{10} \left(\frac{4\pi f_c d_{i,j}^{euc}}{c} \right) + \eta_{LoS} \quad (21)$$

$$PL_{i,j}^{D2V}(NLoS) = 20 \log_{10} \left(\frac{4\pi f_c d_{i,j}^{euc}}{c} \right) + \eta_{NLoS}, \quad (22)$$

where η_{LoS} and η_{NLoS} are the mean additional losses for LoS and NLoS links, c is the speed of light, and $d_{i,j}^{euc} = \sqrt{h_i^2 + (d_{i,j}^{hor})^2}$ is the euclidean distance between the UAV- i and vehicle- j .

As a result, the average path loss of the D2V communication between UAV- i and vehicle- j can be computed as followed:

$$\begin{aligned} PL^{D2V}(i, j) &= P_{i,j}^{D2V}(LoS)PL_{i,j}^{D2V}(LoS) + P_{i,j}^{D2V}(NLoS)PL_{i,j}^{D2V}(NLoS) \\ &= 20 \frac{1}{1 + a \exp(-b(\theta_{i,j} - a))} \left[\log_{10} \left(\frac{4\pi f_c d_{i,j}^{euc}}{c} \right) + \eta_{LoS} \right] \\ &\quad + 20 (1 - P_{i,j}(LoS)) \left[\log_{10} \left(\frac{4\pi f_c d_{i,j}^{euc}}{c} \right) + \eta_{NLoS} \right] \\ &= \frac{\eta_{LoS} - \eta_{NLoS}}{1 + a \exp(-b(\theta_{i,j} - a))} \\ &\quad + 20 \log_{10} (d_{i,j}^{euc} \sec(\theta_{i,j})) + 20 \log_{10} \left(\frac{4\pi f_c}{c} \right) + \eta_{NLoS} \end{aligned}$$

And the channel gain is given by

$$G^{D2V}(i, j) = \frac{1}{PL^{D2V}(i, j)} \quad (23)$$

Consequently, the $\text{SNR}_{i,j}$ and the achievable data rate $C_{i,j}$ in bits per second (bps) of the D2V communication are presented in Equation. (24) and (25) [8, 9]:

$$\text{SNR}_{i,j}^{D2V} = \frac{p_i^{trans} G_{i,j}^{D2V}}{\sum_{n \in \mathbb{N}_{\text{int}}} p_n^{trans} G_{n,j}^{D2V} + N_0}, \quad (24)$$

where p_i^{trans} is the transmit power of UAV- i , \mathbb{N}_{int} is the set of possible interfering UAVs, N_0 is the noise power.

According to Shannon's theorem, the achievable rate of the D2V communication is:

$$C_{i,j}^{D2V} = B * \log_2 (1 + \text{SNR}_{i,j}^{D2V}), \quad (25)$$

where B is the bandwidth for the D2V communication. The energy consumption of the UAV transmitting the message is [10]:

$$E_{i,j}^{D2V} = \frac{S_i}{C_{i,j}^{D2V}} p_i^{trans}, \quad (26)$$

where S_i is the size of the message that UAV- i is going to send.

Drone-to-drone path loss The D2D path loss is also known as air-to-air path loss. Contrary to the D2V scenario, the D2D path loss is dominated by the free-space LoS propagation. Thus the D2D LoS path loss between UAV- n and UAV- m is given as the same as the D2V communication, as described in Equation. (27):

$$PL_{n,m}^{D2D} = 20 \log_{10} \left(\frac{4\pi f_c d_{n,m}^{euc}}{c} \right) + \eta_{LoS}^{D2D}, \quad (27)$$

where η_{LoS}^{D2D} is the mean additional loss of the D2D communication link, c is the speed of light, and $d_{n,m}^{euc} = \sqrt{(x_n - x_m)^2 + (y_n - y_m)^2}$ is the euclidean distance between UAV- n and UAV- m . Consequently, the channel gain $G_{n,m}^{D2D}$, $\text{SNR}_{n,m}^{D2D}$, achievable data rate $C_{n,m}^{D2D}$ and energy consumption $E_{n,m}^{D2D}$ for transmitting a message of size S_n are calculated as followed:

$$G_{n,m}^{D2D} = \frac{1}{PL_{n,m}^{D2D}} \quad (28)$$

$$\text{SNR}_{n,m}^{D2D} = \frac{p_n G_{n,m}^{D2D}}{\sum_{i \in \mathbb{N}_{\text{int}}} p_i G_{i,j}^{D2D} + N_0} \quad (29)$$

$$C_{n,m}^{D2D} = B * \log_2 (1 + \text{SNR}_{n,m}^{D2D}) \quad (30)$$

$$E_{n,m}^{D2D} = \frac{S_n}{C_{n,m}^{D2D}} p_n^{trans} \quad (31)$$

2.2 Energy for mobility

According to [5, 6], the motion power model P_i of UAV- i of speed V_i is represented as Equation. (32). Thus, the energy consumption to move from location $M(x_1, y_1, z_1)$ to location $M'(x_2, y_2, z_2)$ can be calculated by Equation. (33):

$$P_i = P_0 \left(1 + \frac{3V_i^2}{U_{\text{tip}}^2} \right) + P_1 \left(\sqrt{1 + \frac{V_i^4}{4v_0^4}} - \frac{V_i^2}{2v_0^2} \right)^{1/2} + \frac{1}{2} d_0 \rho s A V_i^3, \quad (32)$$

$$E_i(M, M') = \frac{d_{MM'}}{V_i} P_i = \frac{\sqrt{(x_1 - x_2)^2 + (y_1 - y_2)^2 + (z_1 - z_2)^2}}{V_i} P_i \quad (33)$$

where P_0 and P_i are calculated by the following equations as formulated in [17]:

$$P_0 = \frac{\delta}{8} \rho s A \Omega^3 R^3 \quad (34)$$

$$P_1 = (1 + k) \frac{W^{3/2}}{\sqrt{2\rho A}}. \quad (35)$$

U_{tip} , v_0 , d_0 , s , ρ and A are constant parameters related to UAV's dynamic properties and air density.

3 Dataset Generation

3.1 Simulation Setup

We consider a 4-km three-lane highway where 4 UAVs hover over the highway, as illustrated in Fig. 3. The UAVs move according to the elliptical trajectories as detailed in section 3.2. The maximum horizontal speed of UAVs is 30 km/h, the maximum vertical speed is 10 km/h, the limited flying height is between 100 m - 150 m [12, 13].

On the other hand, vehicle trajectories are retrieved by the TraCI interface of Simulation of Urban MObility (SUMO) [20, 21], a free and open source traffic simulator. Vehicles move according to the Krauss mobility model and LC2013 lane change model. The maximum allowed velocity is 100 km/h. For a more authentic scenario, some vehicles are set to be “aggressive” with impolite behaviors such as low intention to cooperate with others, stay in the leftmost lane for a long time and exceed the speed limit, represented by the yellow cars. Moreover, some vehicles are ambulances that have higher priority to pass and lead the in front vehicles to initiate lane change, represented by the red cars [2].

3.2 UAV Trajectory

3.2.1 Horizontal trajectory for x and y

We adopt elliptical trajectory for the four UAVs as illustrated in Fig. 5. The UAVs move according to the elliptic curve. The initial mathematical expression is shown in Equation. (36):

$$\frac{x^2}{a^2} + \frac{y^2}{b^2} = 1, \quad (36)$$

where $2a$ is the width of the ellipse and $2b$ is the height of the ellipse.

In our simulation scenario, the four ellipses are distributed on a circle centered on $(0, 0)$ with a radius of 637 meters, representing the circular highway, as shown in Fig. 4. Consequently, $2a = 637 \times 2 = 1274$, $2b = 637$. The horizontal speed and vertical speed of the drones are limited to 30km/h ($\approx 8.33\text{m/s}$) and 10km/h ($\approx 2.78\text{m/s}$). The traffic state is updated every 0.4 second. Thus, x and y positions are functions of simulation step, s , where v_{hor} and v_{vrt} are the horizontal

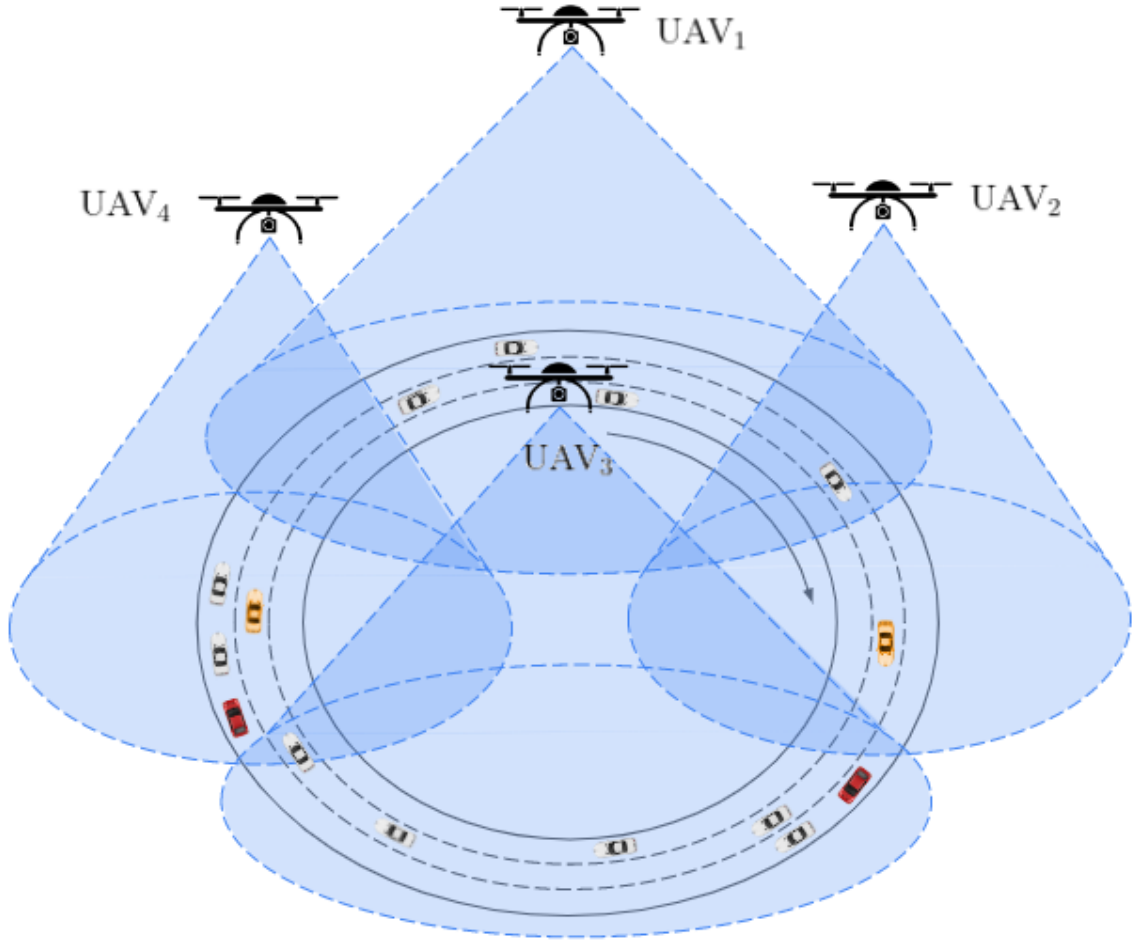


Figure 3: Simulation scenario for dataset generation. Grey cars are ordinary cars, yellow cars are aggressive cars and red cars are emergency cars. The four blue ellipses represent the communication range of each UAV

speed and vertical speed in meter per second:

$$x = v_{hor} \times 0.4s$$

$$y = v_{vrt} \times 0.4s$$

It should be noted that the speed of the four drones are set to 5, 10, 20, 30 km/h. On the other hand, the center coordinates of the four UAVs are

$$(Cx_1, Cy_1) = (0, 637)$$

$$(Cx_2, Cy_2) = (637, 0)$$

$$(Cx_3, Cy_3) = (0, -637)$$

$$(Cx_4, Cy_4) = (-637, 0)$$

Finally, the trajectories with expression in the form of

$$\frac{(x - Cx_i)^2}{a^2} + \frac{(y - Cy_i)^2}{b^2} = 1$$

for the four UAVs can be represented as in the following equations:

$$\begin{aligned} \frac{x^2}{a^2} + \frac{(y - 637)^2}{b^2} &= 1 \\ \frac{(x - 637)^2}{a^2} + \frac{y^2}{b^2} &= 1 \\ \frac{x^2}{a^2} + \frac{(y + 637)^2}{b^2} &= 1 \\ \frac{(x + 637)^2}{a^2} + \frac{y^2}{b^2} &= 1 \end{aligned}$$

3.2.2 Vertical trajectory for z

We adopt a random walk trajectory to determine $z_i, i \in [1, 4]$. It should be noticed that at each step, the UAV moves up or down according to the predefined probability array $p = (p_1, 1 - p_1)$, where p_1 is the probability to moving up, and p_2 is the probability to moving down. In the simulation, we set $p = (0.5, 0.5)$. The random walk algorithm is detailed in Algorithm. 1.

3.2.3 Final trajectories

Examples of the generated trajectories of the UAVs are shown in Fig. 6.

3.3 Dataset parameters

The dataset will be generated through a branch of simulations with different vehicular density ρ . The samples (i.e. the vehicles and drones kinematic parameters) are retrieved, computed and

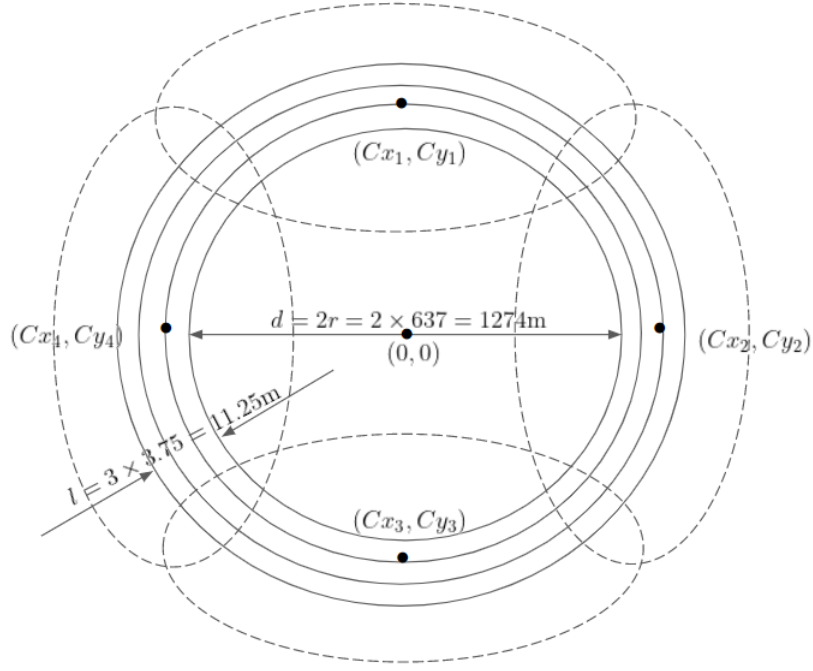


Figure 4: Details of the ellipses

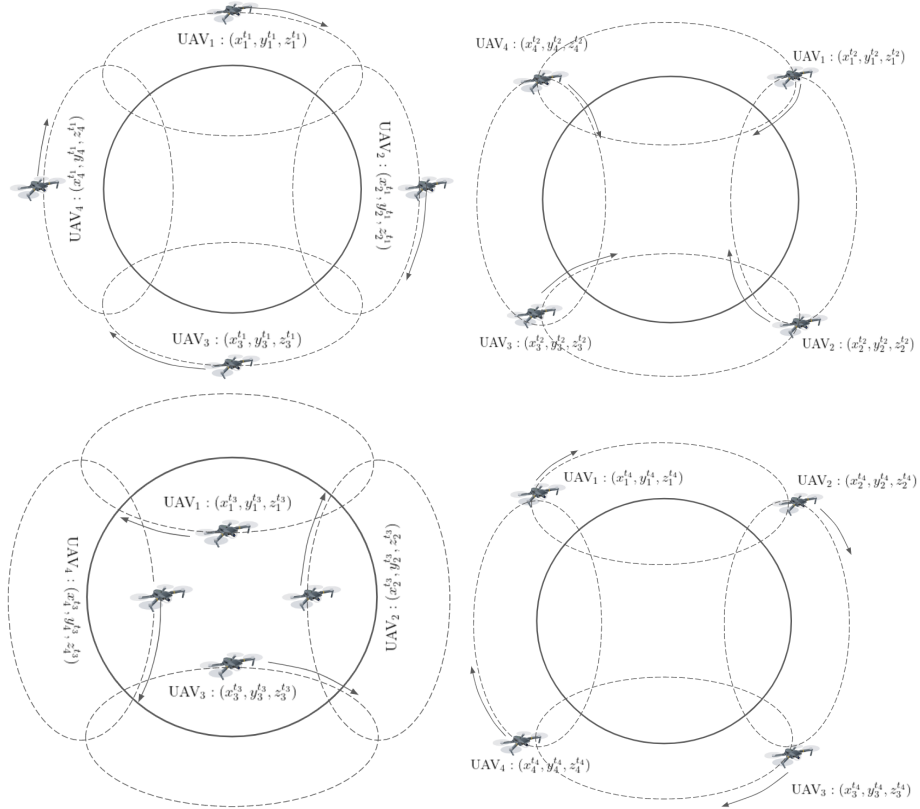
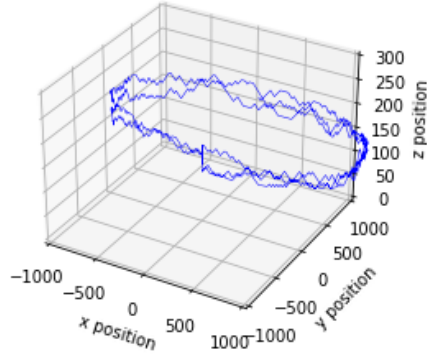


Figure 5: Examples of the generated trajectories of UAVs at time t_1, t_2, t_3 and t_4

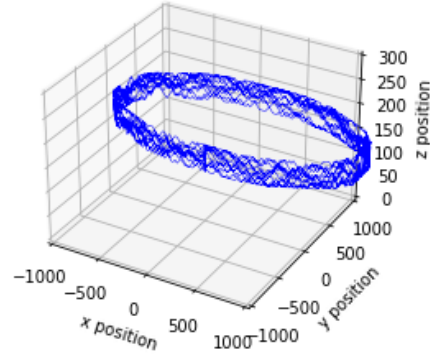
Algorithm 1 Random Walk Algorithm for Vertical Trajectory Generation

- 1: **Input:** S is the number of simulation steps, Z is the vertical action range, p and $1 - p$ are the probability to go up and down at each step, Δz is the vertical step length
 - 2: Initialize vertical trajectory as an all-zero array z of length S
 - 3: **for** $i \leftarrow 0$ to S **do**
 - 4: Randomly choose a vertical direction *up* or *down* according to p
 - 5: **if** the chosen direction is *up* (or *down*) **then**
 - 6: **if** $z[i] + \Delta z$ (or $z[i] - \Delta z$) $\in Z$ **then**
 - 7: Perform the movement
 - 8: **end if**
 - 9: **end if**
 - 10: **end for**
-

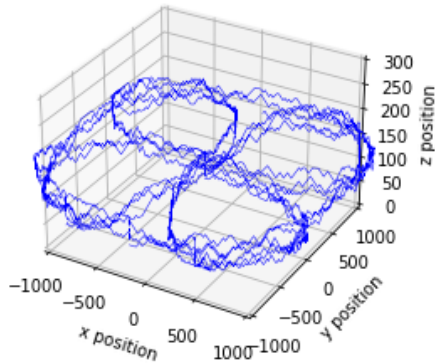
Trajectory of the UAV_1 with 1000 steps



Trajectory of the UAV_1 with 5000 steps



Elliptical Trajectories of the UAVs



Elliptical Trajectories of the UAVs

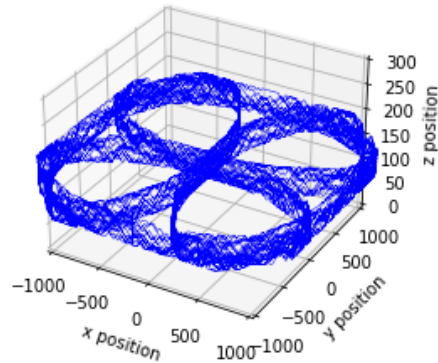


Figure 6: Elliptical trajectories of UAVs

stored every 0.4s. The parameters used for simulation are shown in Table. 2. Each sample consists of the observations of each of the 4 UAVs, as well as the overall collision rate on the highway, as represented by the following:

$$o = \{o_1, o_2, o_3, o_4, \bar{r}\},$$

where o_1, o_2, o_3 , and o_4 are the observations of UAV₁, UAV₂, UAV₃ and UAV₄, respectively. \bar{r} is the total collision number on the highway.

An observation of UAV- i at step t is expressed as

$$o_i[t] = \{x_i[t], y_i[t], z_i[t], \theta_i[t], \rho_i, \bar{W}_i, \bar{E}_i, \bar{t}_{ri}, \bar{t}_{bi}\}, \quad (37)$$

where $x_i[t]$ is the longitudinal position of UAV- i at step t , $y_i[t]$ denotes the lateral position at step t , $z_i[t]$ is the height at step t , $\theta_i[t]$ is the heading direction at step t ($\theta_i[t] = 1$ means moving up, $\theta_i[t] = -1$ means moving down), ρ_i is the number of vehiculars in the communication range of UAV- i , \bar{W}_i the mean waiting time of the vehicle requests in the communication range of UAV- i , \bar{E}_i the average energy consumption of UAV- i , \bar{t}_{ri} and \bar{t}_{bi} the mean risky time and mean blocking time of the vehicles in the communication range of the UAV- i . The algorithm for generating the dataset is presented in Algorithm. 2

Algorithm 2 Algorithm for dataset generation

```

1: Input:  $S$  is the number of simulation steps,  $\mathbb{V}$  is the set of vehicles on the road
2: Initialize
3: for  $t \leftarrow 0$  to  $S$  do
4:   for every vehicle  $j, j \in \mathbb{V}$  do
5:     Determine the corresponding UAV of the vehicle according to the vehicle's GPS position
     and UAV's coverage, denote the corresponding UAV as UAV- $i$ 
6:     Compute  $W_i^j, E_i^j$  according to Equation. (1) and Equation. (18)
7:     Retrieve  $t_{ri}^j, t_{bi}^j$  from vehicle state information
8:     Store  $W_i^j, E_i^j, t_{ri}^j$  and  $t_{bi}^j$  in the UAV- $i$ 's buffer
9:   end for
10:  for every UAV  $i, i \in [1, 4]$  do
11:    Compute the average  $\bar{W}_i, \bar{E}_i, \bar{t}_{ri}, \bar{t}_{bi}$ 
12:    Store current position  $x_i[t], y_i[t], z_i[t]$ , current heading direction  $\theta_i[t]$ , current vehicular
    density  $\rho_i$ , and  $\bar{W}_i, \bar{E}_i, \bar{t}_{ri}, \bar{t}_{bi}$  as current observation of UAV- $i$ 
13:  end for
14:  Store  $\bar{r}$ , number of collisions happen at the current step
15: end for

```

4 Notations and Terminologies

The notations and terminologies used in this report are summarized in Table 2.

The drones are equipped with SoC semiconductors: Snapdragon 821 with Quad-core up to 2.15GHz.

The mean service time for the two types of messages are calculated as followed:

- The length of safety message is exponentially distributed with parameter $\lambda'_1 = \frac{2.15\text{GHz} \times 4}{512 \times 8\text{bits}} \approx$

2.1×10^6 . Thus, $E[B_1] = \frac{1}{\lambda_1} = 4.763 \times 10^{-7}$ s, $E[B_1^2] = \frac{1}{(\lambda_1)^2} \approx 2.269 \times 10^{-13}$.

- On the other hand, the length of vehicle state information is a constant and equals to 32 bytes. Thus, $E[B_2] = \frac{32 \times 8}{2.15 \text{GHz} \times 4} = 2.977 \times 10^{-8}$ s, $E[B_2^2] = 0$.

Table 2: MAIN NOTATIONS AND TERMINOLOGIES

Section	Parameter	Meaning	Value	Reference
V2D and D2V Propagation Delay	$E[W_{i,j}^{D2V}]$	The D2V propagation delays	$\frac{d_{i,j}}{r_i}$	
	$E[W_{i,j}^{V2D}]$	The V2D propagation delays	$\frac{d_{i,j}}{r_i}$	
	$d_{i,j}$ and $d'_{i,j}$	The distances between the UAV and the vehicle- j from which the request is sent during the V2D and D2V communication	$\sqrt{(x_i - x_j)^2 - (y_i - y_j)^2}$	
	$r_{i,j}$	The achievable propagation rate of the communication lnk	$B * \log_2 (1 + \text{SNR}_{i,j})$	
Queuing Delay at the Drone's Side	λ_i	The arrival rate of a class- i request	$\lambda_1 = 20\%$ of vehicle number $\lambda_2 = 2.5$	[18]
	μ	UAV service rate	250	[14]
	$E[B_i]$	The mean service time at the drone of a class- i request	$E[B_1] = 4.763 \times 10^{-7}$, $E[B_1^2] = 2.269 \times 10^{-13}$ $E[B_2] = 2.977 \times 10^{-8}$, $E[B_2^2] = 0$	[15]
	$E[W_i]$	The mean waiting time in the queue of a class- i request		
	$E[R_i]$	The mean residual service time of a class- i request		
	$E[S_i]$	The mean sojourn time of a class- i request	$E[S_i] = E[B_i] + E[W_i]$	
	$E[L_i]$	The average number of requests of class i waiting in the queue		
	$E[W_{i,j}]$	The total V2D delay of a class- i request from a vehicle- j		
	$W_{i,j}^{V2D}$	V2D propagation delay		
	$W_{i,j}^{D2V}$	Drone-to-vehicle (D2V) propagation delay		
	$W_{i,j}^s$	Queuing delay at the drone side		
	T_i	The maximum waiting time for request class- i	0.2 s	[16]
	a and b	Environmental constant depending on rural or urban area	14.39 0.13	[8]
	h_i	The height of UAV- i from ground level	[100,150]	[12]

Table 2 continued from previous page

Section	Parameter	Meaning	Value	Reference
Energy for D2V Communi- cations	$\theta_{i,j}$	Elevation angle between UAV- i and vehicle- j	$\arctan(\frac{h_i}{d_{i,j}^{hor}})$	
	$d_{i,j}^{hor}$	The horizontal distance between the UAV- i and the vehicle- j	$\sqrt{(x_1 - x_2)^2 + (y_1 - y_2)^2}$	
	$d_{i,j}^{euc}$	The euclidean distance between the UAV- i and vehicle- j	$\sqrt{h_i^2 + (d_{i,j}^{hor})^2}$	
	$P_{i,j}(LoS)$	The probability of line-of-sight link	$P_{i,j}(LoS) = \frac{1}{1+a \exp(-b(\theta_{i,j}-a))}$	
	$P_{i,j}(NLoS)$	The probability of non-line-of-sight link	$1 - P_{i,j}(LoS)$	
	$PL_{i,j}(LoS)$	Path loss with LoS link	$20 \log \left(\frac{4\pi f_c d_{i,j}^{euc}}{c} \right) + \eta_{LoS}$	
	$PL_{i,j}(NLoS)$	Path loss with NLoS link	$20 \log \left(\frac{4\pi f_c d_{i,j}^{euc}}{c} \right) + \eta_{NLoS}$	
	f_c	Transmit frequency for uplink and downlink of the D2V communication	2.4GHz	[12]
	η_{LoS} and η_{NLoS}	Additional losses for LoS and NLoS links	1 dB 20 dB	[8]
	c	The speed of light	3×10^8 m/s	
	$G(i,j)$	The channel gain of the communication link between the UAV- i and vehicle- j	$G(i,j) = \frac{1}{PL(i,j)}$	
	$SNR_{i,j}$	The signal-to-noise ratio	$\frac{p_i G_{i,j}}{\sum_{n \in \mathbb{N}_{int}} p_n G_{n,j} + N_0}$	
	$C_{i,j}$	The achievable data rate in bits per second (bps)	$B \cdot \log_2 (1 + SNR_{i,j})$	
	p_i^{trans}	The transmit power of UAV- i	280 mW	[12]
	N_0	The noise power	-174 dB/Hz	[19]
Energy for D2D Communi- cations	$PL_{n,m}^{D2D}$	The D2D LoS path loss between UAV- n and UAV- m	$20 \log \left(\frac{4\pi f_c d_{n,m}^{euc}}{c} \right) + \eta_{LoS}^{D2D}$	
	η_{LoS}^{D2D}	The mean additional loss of the D2D communication link		
	$d_{n,m}^{euc}$	The euclidean distance between UAV- n and UAV- m	$\sqrt{(x_n - x_m)^2 + (y_n - y_m)^2}$	
	$G_{n,m}^{D2D}$	The channel gain	$\frac{1}{PL_{n,m}^{D2D}}$	
	$SNR_{n,m}^{D2D}$	The signal-to-noise ratio	$\frac{p_n G_{n,m}^{D2D}}{\sum_{i \in \mathbb{N}_{int}} p_i G_{i,j}^{D2D} + N_0}$	
	$C_{n,m}^{D2D}$	Achievable data rate	$B \cdot \log_2 (1 + SNR_{n,m}^{D2D})$	
	$E_{n,m}^{D2D}$	Energy consumption for transmitting a message of size S_n	$\frac{S_n}{C_{n,m}^{D2D}} p_n^{trans}$	

Table 2 continued from previous page

Section	Parameter	Meaning	Value	Reference
	S_n	The transmitted message size	512 bytes	[12]
Energy for Mobility	V_i	Horizontal speed of UAV- i Vertical speed of UAV- i	30 km/h 10 km/h	
	P_0	Power constant representing the blade profile	84.14 N	[17]
	P_1	Power constant representing the induced power levels in hovering status	88.63 N	[17]
	U_{tip}	The tip speed of the rotor blade	120 m/s	[17]
	v_0	The mean rotor induced velocity in hover	4.03	[17]
	d_0	The fuselage drag ratio	0.6	[17]
	s	Rotor solidity	0.05	[17]
	ρ	Air density	1.225 kg/m ³	[17]
	A	Rotor disc area	0.503 m ²	[17]

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