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Fast linear solvers for incompressible CFD simulations with compatible discrete operator schemes

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Abstract

Finding a robust and efficient solver for (non-)symmetric systems that arise in incompressible Computational Fluid Dynamics (CFD) is of great interest to both academia and industry. We consider the Compatible Discrete Operator (CDO) discretization that has recently been devised for CFD simulations in the context of incompressible Stokes and Navier-Stokes flows. The discrete problems resulting from CDO schemes yield large saddle-point systems that require relevant numerical methods suitable to deal with large indefinite and poorly conditioned linear systems. In this paper, we focus on two segregated methods: the augmented Lagrangian Uzawa method and the generalized Golub-Kahan bidiagonalization, as well as a monolithic method based on an algebraic transformation by change of variables. We also employ algebraic multigrid (AMG) preconditioned Krylov solvers such as the Flexible Conjugate Gradient (FCG) method, and the Flexible Generalized Minimal Residual (FGMRES) method, to solve the linear systems. Using the CFD software code_saturne, we compare the numerical performance with respect to the choice of linear solvers and numerical strategies for the saddlepoint problem. In the numerical experiments, the AMG preconditioned Krylov methods show robustness in test cases of Stokes and Navier-Stokes problems.

Keywords: Algebraic multigrid method, Compatible discrete operator, Incompressible Navier–Stokes, Saddle–point problems

MSC Classification: 65F08, 65F10, 65M22, 76D05

1 Introduction

Problems in Computational Fluid Dynamics (CFD) arise in many academic and industrial fields, e.g., aerospace, petroleum and nuclear engineering. In this paper, we focus on the steady Stokes and Navier–Stokes equations in the case of incompressible flows. Let $\Omega \subset \mathbb{R}^d$, d = 2, 3 be an open bounded connected polytopal Lipschitz domain and $\partial\Omega$ be its boundary. The velocity is a vector-valued field denoted by \vec{u} and the pressure is a scalar-valued field denoted by p such that:

$$-\nu\Delta \vec{u} + \chi \left((\vec{u} \cdot \nabla) \vec{u} \right) + \nabla p = \vec{f} \qquad \text{in } \Omega, \qquad (1.1)$$

$$\nabla \cdot \vec{u} = 0 \qquad \qquad \text{in } \Omega. \tag{1.2}$$

Equation (1.1) refers to the conservation of the momentum and (1.2) to the incompressibility constraint, ensuring that the velocity field conserves mass. Here we assume a constant mass density. The parameter $\nu > 0$ denotes the fluid viscosity and \vec{f} is the volumetric forcing term. $-\nu\Delta\vec{u}$ is the viscous term and $(\vec{u}\cdot\nabla)\vec{u}$ is the convection term. The choice $\chi = 0$ corresponds to the Stokes equations whereas the choice $\chi = 1$ to the Navier–Stokes equations. Dirichlet boundary conditions are enforced on $\partial\Omega$. The pressure is uniquely defined by enforcing $\int_{\Omega} p = 0$. To manage the nonlinearity of the Navier–Stokes equation in the convection term, we introduce *Picard's iteration* yielding the following linearized Navier–Stokes equation (also known as *Oseen problem*) such that

$$-\nu\Delta \vec{u}^{(k)} + (\vec{u}^{(k-1)} \cdot \nabla)\vec{u}^{(k)} + \nabla p^{(k)} = \vec{f} \qquad \text{in } \Omega, \qquad (1.3)$$

$$\nabla \cdot \vec{u}^{(k)} = 0 \qquad \text{in } \Omega, \qquad (1.4)$$

for each iteration k, starting from an arbitrary initial guess $(\vec{u}^{(0)}, p^{(0)})$.

The discretization of equations (1.1) and (1.2) has been extensively studied in the literature. Depending on the choice of the velocity-pressure coupling (segregated or coupled), the definition and location of the degrees of freedom (DoF), the resulting linear system(s) can have quite different structure and different features. In what follows, we consider the Compatible Discrete Operator (CDO) schemes introduced in [1] for the spatial discretization. CDO schemes belong to a class of space discretization schemes called *mimetic*, *structure-preserving* or *compatible*. These schemes have been inspired by the seminal works of Bossavit [2] and of Hyman & Scovel [3]. They have shed a new light on the way to devise the discretization of partial differential equations (PDE) thanks to some concepts of differential geometry and algebraic topology. During the last two decades, several other discretization schemes belonging to this latter class have emerged, e.g., the Discrete Exterior Calculus (DEC) schemes [4], the Discrete Geometric Approach (DGA) [5], the Mimetic Spectral Element method [6], the Mimetic Finite Difference schemes [7], the Hybrid Mixed Mimetic (HHM) framework [8], the Finite Element Exterior Calculus (FEEC) schemes [9] or the De Rham complexes [10]. More recently, extensions to higher order discretizations have been devised in Virtual Element Methods (VEM) [11], Hybrid High Order (HHO) schemes [12] or the Discrete De Rham framework [13].

In this paper, we focus on CDO face-based schemes with a full velocity/pressure coupling. As explained in [14], this choice leads to a low-order approximation of the Navier–Stokes equations. The stability and the approximation properties are fulfilled on a wide range of meshes: from Cartesian meshes to polyhedral, non-matching and/or distorted meshes. The CDO face-based discretization is a stable method (cf. Section 2). The resulting linear system is a saddle-point problem formulated as $\mathcal{A}\mathbf{x} = \mathbf{b}$, where

$$\mathcal{A} = \begin{bmatrix} A & B^T \\ B & O \end{bmatrix}, \qquad \mathbf{x} = \begin{bmatrix} \mathbf{u} \\ \mathbf{p} \end{bmatrix} \text{ and } \mathbf{b} = \begin{bmatrix} \mathbf{f} \\ \mathbf{g} \end{bmatrix}.$$
(1.5)

Here, $A \in \mathbb{R}^{n \times n}$, $B \in \mathbb{R}^{m \times n}$, $O \in \mathbb{R}^{m \times m}$ is the null matrix, $\boldsymbol{u} \in \mathbb{R}^n$, $\boldsymbol{f} \in \mathbb{R}^n$, $\boldsymbol{p} \in \mathbb{R}^m$ and $\boldsymbol{g} \in \mathbb{R}^m$ for $n \ge m$. The blocks A, B^T , B and O are commonly referred to as the (1,1)-, (1,2)-, (2,1)- and (2,2)-blocks, respectively. In the case of the Stokes equations, Ais a symmetric positive definite (SPD) matrix corresponding to the viscous (diffusion) term, while B is associated with the divergence operator for the velocity field and B^T with the pressure gradient operator. In the case of the (linearized) Navier–Stokes equations, A is a non-symmetric matrix due to the addition of a convective term to the viscous term.

Different types of saddle-point problems occur in many applications of applied mathematics and engineering. To solve the corresponding linear systems, numerous techniques have been proposed and developed during the previous decades; please refer to the comprehensive introduction of Benzi, Golub and Liesen [15] on the resolution of saddle-point problems. In what follows, one splits these techniques to solve (1.5) into two categories: *segregated* techniques working iteratively on a subset of blocks and *monolithic* techniques working on the full system (all blocks at once). For these two categories, linear systems have to be solved either with a *direct* method or an iterative method.

Direct methods are robust with respect to the properties of the linear system (symmetry and conditioning for instance) but their computational cost in terms of CPU and mainly memory usage can become prohibitive on large scale systems. In this work, we consider parallel multi-frontal algorithms available in MUMPS [16] as direct solver.

Krylov subspace methods [17] are our choice of iterative methods: for instance, GMRES [18] on non-symmetric systems and CG [19] on symmetric positive definite systems. Krylov subspace methods can become very competitive solvers when combined with efficient preconditioning techniques. To employ variable preconditioning, i.e., the preconditioner can be modify at each iteration (e.g. inner-outer Krylov methods), flexible variants of those methods have been developed such as FGMRES [20], GCR [21, 22] or FCG [23]. For more details for non-symmetric problem solvers, we refer to [24, 25] and the references therein. The efficiency of the iterative solver relies mainly on the choice of the preconditioner. In this paper, one considers multilevel preconditioning technique based on algebraic multigrid (AMG) methods. These methods have been proven to work efficiently on large scale linear systems stemming from unstructured grids without needing geometric grid information; see [26] for a comparison of multigrid preconditioners. The main ingredients composing the design of AMG methods are the choice of the cycle (V-cycle for instance), that of the smoother (a symmetric Gauss-Seidel for instance) and the coarsening algorithm. There are two main classes of coarsening strategies: independent-set based AMG, such as *classical AMG* [27–30] and aggregation-based AMG, such as *smoothed aggregation AMG* [31] and *pairwise aggregation AMG* [22, 32]. Please refer to [33] for more references and details about AMG methods. More recently, Notay [34] then Bacq and Notay [35] investigated aggregation-based AMG for the Stokes problems and (linearized) Navier–Stokes problems, respectively, including a two-grid analysis.

In the manner of segregated techniques, one option is to utilize stationary iterations and Uzawa-like algorithms [15, 36], including the variant known as the Augmented-Lagrangian Uzawa (ALU) algorithm [37]. Another approach involves the Golub–Kahan Bidiagonalization (GKB) algorithm [38], which has recently emerged as a method for solving symmetric saddle-point systems.

Regarding monolithic techniques, apart from direct methods applied to the full system, other techniques vary in their approach to defining an efficient preconditioner for a Krylov subspace method. ILU type preconditioning or null space methods [15] are possible choices. An approximated block LU factorization has also been introduced in [39]. More generally, block preconditioning are among the most explored techniques of this category. In this case, the approximation of the Schur complement [40] (in our case, equal to $-BA^{-1}B^{T}$) is often required. In this situation, the quality of this approximation is a key ingredient to get an efficient solver. More specifically in the context of the incompressible Navier-Stokes problems Semi-Impicit Pressure Linked Equation (SIMPLE) type preconditioning [41] have been investigated. Other approaches, such as augmented Lagrangian preconditioning, which is applicable for high Reynolds number [42, 43] cases, and a vector penalty projection method [44] are other techniques of interest. Additionally, an algebraic transformation of the saddle-point system relying on a change of variables has been devised in [34] in the context of the Stokes problem. This transformation allows one to use multigrid methods on the (monolithic) transformed system.

Our main contribution is to present a first comparative study of algorithms used to solve saddle-point systems that arise from the CDO face-based discretization of the Stokes and Navier–Stokes equations. This discretization differs from the standard Finite Elements schemes since one performs a *static condensation* of the (1,1)-block (cf. Section 2). We focus on three algorithms: two segregated techniques, the *Augmented-Lagrangian Uzawa* (ALU) algorithm [37] and the *Golub–Kahan Bidiagonalization* (GKB) algorithm [38] and, one monolithic technique, the Notay's algebraic transformation [34]. To our knowledge, this is the first application of the CDO scheme with the GKB algorithm to solve the Stokes problem and with algebraic transformation for both the Stokes and Navier–Stokes problems. Additionally, several Krylov solvers preconditioned with different AMG strategies are compared either on the resolution of the (1,1)-block or on the resolution of the transformed saddle-point problem. These comparative studies are performed on the key ingredients that underpin the definition of an AMG, such as the type of cycle, the type of smoothers, and the coarsening strategy. Specifically, we consider an in-house implementation of the K-cycle algorithm

based on Notay's work [22, 45], BoomerAMG from the HYPRE library [46, 47], and GAMG/HMG from the PETSc library [48].

The remainder of the paper is structured as follows. After the introduction, we describe the main features and ingredients of the CDO face-based discretization in Section 2, along with the resulting saddle-point problem. In Sections 3 and 4, we provide detailed explanations of the different algorithms used in our comparative studies, as well as the various strategies and configurations of the multilevel algorithms used as a preconditioner for a Krylov solver. Numerical experiments to compare their numerical performance are reported in Section 5. At the end, conclusion and prospect for future improvements are presented in Section 6. We note that all the algorithms and discretizations described in this paper are freely available in the latest version of code_saturne¹.

2 The Compatible Discrete Operator (CDO) framework

The discretization of the Stokes and Navier–Stokes equations under consideration rely on the CDO framework, as developed during Bonelle's PhD[49, 50]. It encompasses several discretizations according to the location of the degrees of freedom (DoF): vertex-based, edge-based, face-based and cell-based schemes.

In this work, we focus on the CDO face-based schemes for the discretization of the steady Stokes and Navier–Stokes equations. The velocity DoFs are hybrid in the sense that they are located in the cells and at the faces (see Figure 1; Left part). They are defined as the mean-value of the velocity vector-field in a cell and on a face for each component. The main ingredients underpinning the discretization (the velocity gradient reconstruction operator and the velocity divergence operator) are recalled and we refer to [14, 51] for more details. The resulting CDO face-based scheme is analyzed in [51] and references therein (stability, consistency, *a priori* error estimates, etc.). In particular, the stability results imply that \mathcal{A} , defined in (1.5), is invertible. Namely, the solvability conditions [15, Theorem 3.4]

$$\ker(H) \cap \ker(B) = \{\mathbf{0}\} \quad \text{and} \quad \ker(B^T) = \{\mathbf{0}\}, \tag{2.1}$$

where H is the symmetric part of A are fulfilled since A is non-singular and B is a full rank operator (*LBB inf-sup* condition).

Notation

Let C (resp. F) be the set of the mesh cells (resp. mesh faces). The number of elements in a set X is denoted by #X (e.g. #C is the number of mesh cells). The set of faces associated to a cell $c \in C$ is denoted by F_c . The set of faces is split into two subsets: the set of boundary faces $F^{\partial} := \{f \in F | f \subset \partial\Omega\}$ and the set of the interior faces $F^{\circ} := F \setminus F^{\partial}$. For a face f, |f| corresponds to its surface and \vec{x}_f to its barycenter. One chooses an arbitrary orientation \vec{n}_f to each face and one denotes $\vec{n}_{f,c}$ its outward

 $^{^{1} \}rm https://www.code-saturne.org$

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normal such that $\vec{n}_{f,c} = \iota_{f,c}\vec{n}_f$ with $\iota_{f,c} = \pm 1$ according to the arbitrary choice. For a cell c, |c| and \vec{x}_c denote its volume and barycenter, respectively.

The vector-valued piecewise constant polynomial space in a cell $c \in C$ (resp. on a face $f \in F$) is denoted by $\mathbb{P}_0^d(c)$ (resp. $\mathbb{P}_0^d(f)$) with d the dimension of Ω (here d = 3). The scalar-valued piecewise constant polynomial space in a cell $c \in C$ is denoted by $\mathbb{P}_0(c)$. The local space of velocity DoFs (i.e. associated to a cell $c \in C$) is denoted by $\mathcal{U}(c)$ and the local space of pressure DoFs by $\mathcal{P}(c)$ such that

$$\mathcal{U}(c) := \left(igwedge M_0^d(f)
ight) igwedge \mathbb{P}_0^d(c) \qquad ext{and} \qquad \mathcal{P}(c) := \mathbb{P}_0(c).$$

For $c \in C$, an element $\boldsymbol{u}_c \in \boldsymbol{\mathcal{U}}(c)$ is such that $\boldsymbol{u}_c := ((\vec{u}_f)_{f \in F_c}, \vec{u}_c) \in \mathbb{R}^{d(\#F_c+1)}$ and $p_c \in \mathcal{P}(c)$ is simply a constant value inside the cell $c. \boldsymbol{\mathcal{U}}(\Omega)$ denotes the global space of velocity DoFs and $\mathcal{P}(\Omega)$ the global space of pressure DoFs. These spaces are defined by

$$\boldsymbol{\mathcal{U}}(\Omega) := \left(\bigotimes_{f \in F} \mathbb{P}^d_0(f) \right) \times \left(\bigotimes_{c \in C} \mathbb{P}^d_0(c) \right) \text{ and } \mathcal{P}(\Omega) := \bigotimes_{c \in C} \mathbb{P}_0(c).$$

An element $\boldsymbol{u} \in \boldsymbol{\mathcal{U}}(\Omega)$ is such that $\boldsymbol{u} := ((\vec{u}_f)_{f \in F}, (\vec{u}_c)_{c \in C}) \in \mathbb{R}^{d(\#F+\#C)}$. An element $p \in \mathcal{P}(\Omega)$ is such that $p := (p_c)_{c \in C} \in \mathbb{R}^{\#C}$. The space of pressure DoFs is denoted by $\mathcal{P}^*(\Omega)$ when a zero mean-value is prescribed on the pressure DoFs. The space of velocity DoFs is denoted by $\boldsymbol{\mathcal{U}}^{\circ}(\Omega)$ when homogeneous Dirichlet boundary conditions are enforced on $\partial\Omega$. For the sake of clarity, we only focus on this kind of boundary conditions. These two spaces are defined as follows:

$$\mathcal{P}^*(\Omega) := \{ p \in \mathcal{P}(\Omega) \mid \sum_{c \in C} p_c | c | = 0 \} \text{ and } \mathcal{U}^{\circ}(\Omega) = \{ \boldsymbol{u} \in \mathcal{U}(\Omega) \mid \vec{u}_f = 0 \ \forall f \in F^{\partial} \}.$$

Main operators

The two main operators used to discretize the Stokes problem (1.1)- (1.2) are the velocity gradient reconstruction operator and the velocity divergence operator. The pressure gradient is simply obtained as the adjoint operator of the velocity divergence. For a cell $c \in C$, the local reconstruction operator for the velocity gradient is denoted by G_c and is defined by a piecewise constant tensor in each pyramid $\mathfrak{p}_{f,c}$ associated to a face $f \in F_c$ (see Figure 1). Let $\mathfrak{P}_{F_c} := {\mathfrak{p}_{f,c}}_{f \in F_c}$ be the set of all pyramids in the cell $c. G_c : \mathcal{U}(c) \mapsto \mathbb{P}_0^{d \times d}(\mathfrak{P}_{F_c})$ such that

$$\mathsf{G}_{c}(\boldsymbol{u}_{c})|_{\mathfrak{p}_{f,c}} = \mathsf{G}_{0,c}(\boldsymbol{u}_{c}) + \eta \frac{|f|}{|\mathfrak{p}_{f,c}|} \left((\vec{u}_{f} - \vec{u}_{c}) - \mathsf{G}_{0,c}(\boldsymbol{u}_{c})(\vec{x}_{f} - \vec{x}_{c}) \right) \otimes \vec{n}_{f,c}$$

where $\eta > 0$ is a scaling coefficient related to the stability of the reconstruction. Depending on the value of this coefficient, one can recover the Generalization of the Crouzeix–Raviart framework [52] with $\eta = 1$, the Discrete Geometric Approach [5] with $\eta = 1/d$ or the Hybrid Finite Volume method [8] with $\eta = 1/\sqrt{d}$.

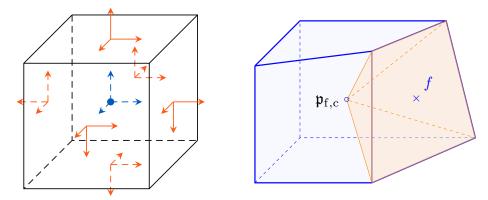


Fig. 1 Left: Example in a hexahedral cell of the locations of the velocity DoFs (three components on each face in orange and three components in the cell in blue), and the pressure DoF (the mean-value in the cell – the blue bullet). Right: Example of a pyramid of basis the face f. The velocity gradient is constant inside this volume.

 $\mathsf{G}_{0,c}(\boldsymbol{u}_c)$ is a piecewise constant tensor in the cell c corresponding to the \mathbb{P}_0 -consistent gradient reconstruction defined as

$$\mathsf{G}_{0,c}: \mathcal{U}(c) \mapsto \mathbb{P}_0^{d \times d}(c) \qquad \text{s.t.} \qquad \mathsf{G}_{0,c}(\boldsymbol{u}_c) := \frac{1}{|c|} \sum_{f \in F_c} |f|(\vec{u}_f - \vec{u}_c) \otimes \vec{n}_{f,c}. \tag{2.2}$$

The global velocity gradient reconstruction operator is simply defined by collecting the local velocity gradient reconstruction operators. The definition of the velocity divergence relies on the identity $\nabla \cdot (\vec{u}) = \text{trace}(\nabla \vec{u})$. For each cell $c \in C$, the local velocity divergence operator D_c is defined as

$$\mathsf{D}_{c}: \mathcal{U}(c) \mapsto \mathcal{P}(c) \qquad \text{s.t.} \qquad \mathsf{D}_{c}(\boldsymbol{u}_{c}) := \frac{1}{|c|} \sum_{f \in F_{c}} |f| \vec{u}_{f} \cdot \vec{n}_{f,c}. \tag{2.3}$$

In the case of the Navier–Stokes equations, one also introduces a convection operator, see [51] for more details.

Weak formulation of the discrete Stokes problem

The discrete weak formulation for the Stokes problem stated in (1.1)–(1.2) with $\chi = 0$ relies on the two previous operators. With homogeneous Dirichlet boundary conditions on the velocity field, this yields: find $(\boldsymbol{u}, \boldsymbol{p}) \in \boldsymbol{\mathcal{U}}^{\circ}(\Omega) \times \boldsymbol{\mathcal{P}}^{*}(\Omega)$ such that $\forall \boldsymbol{w} \in \boldsymbol{\mathcal{U}}^{\circ}(\Omega)$ and $\forall \boldsymbol{q} \in \boldsymbol{\mathcal{P}}^{*}(\Omega)$,

$$\sum_{c \in C} \int_{c} \nu \mathsf{G}_{c}(\boldsymbol{u}_{c}) : \mathsf{G}_{c}(\boldsymbol{w}_{c}) - \sum_{c \in C} \int_{c} \mathsf{D}_{c}(\boldsymbol{w}_{c}) \, \boldsymbol{p}_{c} = \sum_{c \in C} \int_{c} \vec{f} \cdot \vec{w}_{c}$$
(2.4)

$$-\sum_{c\in C}\int_{c}\mathsf{D}_{c}(\boldsymbol{u}_{c})\boldsymbol{q}_{c} = 0.$$
(2.5)

Algebraic viewpoint

Switching to the algebraic viewpoint, the different summands in equations (2.4)–(2.5) correspond to the different blocks of the local saddle-point systems $\hat{\mathcal{A}}_c \hat{\boldsymbol{x}}_c = \hat{\boldsymbol{b}}_c$ associated to each cell $c \in C$ with $\hat{\boldsymbol{x}}_c$ and $\hat{\boldsymbol{b}}_c$, two arrays restricted to all the face and cell DoFs associated to a cell. Elements which will be modified by the static condensation are written with a hat symbol. More specifically, the system corresponds to

$$\hat{\mathcal{A}}_c := \begin{bmatrix} \hat{A}(c) & B^T(c) \\ B(c) & 0 \end{bmatrix} \quad \text{and} \quad \hat{\boldsymbol{b}}_c := \begin{bmatrix} \hat{\boldsymbol{f}}(c) \\ g(c) \end{bmatrix}$$
(2.6)

where $\hat{A}(c)$ is a square matrix of size $d(\#F_c+1)$ associated to $\int_c \nu \mathsf{G}_c(\boldsymbol{u}_c) : \mathsf{G}_c(\boldsymbol{w}_c)$ and B(c) is a rectangular matrix of size $d \times d(\#F_c+1)$ associated to $\int_c \mathsf{D}_c(\boldsymbol{u}_c) \boldsymbol{q}_c$. The static condensation modifies the velocity block $\hat{A}(c)$ and its right-hand side $\hat{\boldsymbol{f}}(c)$. To detail this operation, one splits the system (2.6) to make appear the contribution stemming from the face and cell velocity DoFs.

$$\hat{\mathcal{A}}_{c} := \begin{bmatrix} \frac{\hat{A}_{FF}(c) | A_{FC}(c) | B_{F}^{T}(c)}{A_{CF}(c) | A_{CC}(c) | 0} \\ \hline \mathbf{B}_{F}(c) | 0 | 0 \end{bmatrix} \quad \text{and} \quad \hat{\mathbf{b}}_{c} := \begin{bmatrix} \frac{\hat{\mathbf{f}}_{F}(c)}{\hat{\mathbf{f}}_{C}(c)} \\ \hline \mathbf{g}(c) \end{bmatrix}$$
(2.7)

Since the block $A_{CC}(c)$ is a diagonal square matrix of size d, the static condensation technique [53] allows one to reduce easily the size of the local system by removing the cell DoFs related to the velocity as follows:

$$\mathcal{A}_{c} := \begin{bmatrix} A_{c} & B_{F}^{T}(c) \\ \hline B_{F}(c) & 0 \end{bmatrix} \quad \text{and} \quad \mathbf{b}_{c} := \begin{bmatrix} \mathbf{f}(c) \\ \hline \mathbf{g}(c) \end{bmatrix}$$
(2.8)

where the velocity block $A_c := \hat{A}_{FF}(c) - A_{FC}(c) \cdot A_{CC}^{-1}(c) \cdot A_{CF}(c)$ is a square matrix of size $d\#F_c$ and $\mathbf{f}(c) := \hat{\mathbf{f}}_F(c) - A_{FC}(c) \cdot A_{CC}^{-1}(c) \mathbf{f}_C(c)$. The resulting system (1.5) with n = d#F and m = #C stems from the cell-wise assembly process of the local systems detailed in (2.8).

3 Some strategies to solve saddle-point problems

Here we introduce three algorithms for solving saddle-point problems obtained in stable discretizations of the Stokes or Navier–Stokes equations. In general, a direct solver is a robust and precise option to solve the system (1.5), especially when the (1,1)-block is non-symmetric and ill-conditioned. Most direct solvers rely on factorizing a matrix which requires a high computational complexity and memory storage. For 2D problems, direct solvers are an efficient option, but for a large sparse matrix as in typical 3D problems or a dense matrix, this makes direct solvers prohibitively expensive in terms of computation time and/or memory usage. Hence, we focus on iterative methods for solving saddle-point systems.

For the special case of the Stokes problem, i.e. $\chi = 0$ in (1.1)-(1.2), the system (1.5) is symmetric. We first introduce the Craig's variant of the Golub-Kahan bidiagonalization (GKB) for symmetric saddle-point systems. It shows a good performance in terms of iteration count and solution time, however, it is only applicable for symmetric matrices and thus its usage in **code_saturne** is limited. As explained earlier, we solve the Navier–Stokes problem with Picard (fixed-point) iterations, where each iteration corresponds to an Oseen problem (1.3)- (1.4). The saddle-point matrix is non-symmetric due to the presence of the advective field defined as $\vec{u}^{(k-1)}$. The second algorithm that we introduce is the Augmented Lagrangian Uzawa (ALU) method suited for symmetric or non-symmetric indefinite matrices. In the practical cases of this paper and with some well-chosen parameter, it solves the problem in an acceptably small number of iterations. As a third alternative, we use an algorithm that we call *Notay's algebraic transformation*. It can be applied to symmetric or non-symmetric problems and the obtained transformed matrices can be solved with an efficient AMG solver or with a Krylov subspace method preconditioned with an AMG.

3.1 Golub-Kahan bidiagonalization

We briefly summarize the generalized GKB method that has been introduced in [38]. The algorithm solves symmetric saddle-point systems, which implies that the GKB method can only be applied to the Stokes equations in our context. As a first step, we need to transform the saddle-point system to obtain a zero vector in the upper part of the right-hand side vector. We furthermore use a common regularization technique [54], known as the augmented Lagrangian approach. Let therefore $\gamma \geq 0$ be a scaling factor and $W \in \mathbb{R}^{m \times m}$ be an SPD matrix. By recalling the solvability condition (2.1), we can replace (1.5) by the equivalent augmented system $\mathcal{A}_{\gamma} \boldsymbol{x}_{\gamma}^* = \boldsymbol{b}_{\gamma}^*$ with

$$\mathcal{A}_{\gamma} = \begin{bmatrix} A_{\gamma} & B^{T} \\ B & O \end{bmatrix}, \qquad \mathbf{x}_{\gamma}^{*} = \begin{bmatrix} \mathbf{u}_{\gamma}^{*} \\ \mathbf{p} \end{bmatrix}, \qquad \mathbf{b}_{\gamma}^{*} = \begin{bmatrix} 0 \\ \mathbf{g}_{\gamma}^{*} \end{bmatrix}, \qquad (3.1)$$

where $A_{\gamma} = A + \gamma B^T W^{-1} B$, $\boldsymbol{u}_{\gamma}^* = \boldsymbol{u} - A_{\gamma}^{-1} (\boldsymbol{f} + \gamma B^T W^{-1} \boldsymbol{g})$ and $\boldsymbol{g}_{\gamma}^* = \boldsymbol{g} - B A_{\gamma}^{-1} \boldsymbol{f}$. After the regularization, in a similar way with Lanczos process, bidiagonalization leads us to derive the following GKB formulation: find $V \in \mathbb{R}^{n \times n}$, $Q \in \mathbb{R}^{m \times m}$ and $D \in \mathbb{R}^{n \times m}$ such that

$$\begin{cases} B^{T}Q = A_{\gamma}V \begin{bmatrix} D\\ O \end{bmatrix}, & V^{T}A_{\gamma}V = I_{n}, \\ BV = WQ \begin{bmatrix} D^{T} & O \end{bmatrix}, & Q^{T}WQ = I_{m}, \end{cases}$$
(3.2)

with ${\cal D}$ being an upper bidiagonal matrix

$$D = \begin{bmatrix} \alpha_1 & \beta_1 & 0 & \cdots & 0 \\ 0 & \alpha_2 & \beta_2 & \cdots & 0 \\ \vdots & \ddots & \ddots & \ddots & \vdots \\ 0 & \cdots & 0 & \alpha_{n-1} & \beta_{n-1} \\ 0 & \cdots & 0 & 0 & \alpha_n \end{bmatrix}$$

with entries α_i and β_i computed successively as given in Algorithm 1. By multiplying the augmented system $\mathcal{A}_{\gamma} \boldsymbol{x}_{\gamma}^* = \boldsymbol{b}_{\gamma}^*$ with the block-diagonal matrix with diagonal elements V^T and Q^T from the left, and using the change of variables $\boldsymbol{u}_{\gamma}^* = V\boldsymbol{y}$ and $\boldsymbol{p} = Q\boldsymbol{z}$, we obtain with (3.2)

$$\begin{bmatrix} I_n & D \\ D^T & O \end{bmatrix} \begin{bmatrix} \boldsymbol{y} \\ \boldsymbol{z} \end{bmatrix} = \begin{bmatrix} \boldsymbol{0} \\ Q^T \boldsymbol{g}_{\gamma}^* \end{bmatrix}.$$
 (3.3)

Hence, once we derive V and Q, we can also get the solution of the augmented system by solving (3.3). Craig's variant algorithm [38, Algorithm 3.1] used to solve the augmented system is presented in Algorithm 1. In each iteration of the algorithm, we compute column by column the matrices V and Q and obtain the next update of the solution $\boldsymbol{u}_{\gamma}^{*}$ (and thus \boldsymbol{u}) and \boldsymbol{p} from the system (3.3). This algorithm is a three-term recurrence, hence it is not necessary to store all the basis vectors of V and Q, but only the previous ones are needed. Please note that in each Golub-Kahan iteration, we have to solve a linear system defined by the matrix A_{γ} . We call this solution step the *inner iteration*. Depending on the size of the system, a direct solver or another iterative solver may be applied.

In Algorithm 1, the stopping criterion CHECK() is yet undefined. In the following we will use a lower bound estimate of the energy error as in [38, 55]. The A_{γ} orthogonality of V implies

$$\|\boldsymbol{e}^{(k)}\|_{A_{\gamma}}^{2} = \sum_{j=k+1}^{n} \zeta_{j}^{2} > \xi_{k,l}^{2} := \sum_{j=k+1}^{k+l+1} \zeta_{j}^{2},$$

with $e^{(k)} = u_{\gamma}^* - u_{\gamma}^{(k)}$ being the error, ζ_j defined in Algorithm 1 and l > 1 an integer. The quantity $\xi_{k,l}$ is a lower bound for the error at step k-l. To obtain a lower bound estimate for $e^{(k)}$, the algorithm thus needs to run l more iterations. With a stopping tolerance $\epsilon < 1$, the stopping criterion is then defined as

if
$$\xi/\bar{\xi} \leq \epsilon$$
, then *convergence=true*, where $\xi = \sum_{j=k-l+1}^{k} \zeta_j^2$ and $\bar{\xi}^2 = \sum_{j=1}^{k} \zeta_j^2$.

We set l = 5 as default in our experiments. For more details on bounds of $\|\boldsymbol{e}^{(k)}\|_{A_{\gamma}}^2$, we refer to [38, 55].

Algorithm 1 Craig's variant algorithm.

Input: A_{γ} , B, W, b_{γ}^* , maxit. Output: $\boldsymbol{u}_{\gamma}^{*}, \boldsymbol{p}$. 1: $\beta_1 = \| \boldsymbol{b}_{\gamma}^{*'} \|_{W^{-1}}; \boldsymbol{q}_1 = W^{-1} \boldsymbol{b}_{\gamma}^{*} / \beta_1; \boldsymbol{r} = A_{\gamma}^{-1} B^T \boldsymbol{q}_1; \alpha_1 = \| \boldsymbol{r} \|_{A_{\gamma}}; \boldsymbol{v}_1 = \boldsymbol{r} / \alpha_1.$ 2: $\zeta_1 = \beta_1 / \alpha_1$; $\boldsymbol{d}_1 = \boldsymbol{q}_1 / \alpha_1$; $\boldsymbol{u}_{\gamma}^{(1)} = \zeta_1 \boldsymbol{v}_1$; $\boldsymbol{p}^{(1)} = -\zeta_1 \boldsymbol{d}_1$. 3: k = 0; convergence = false. 4: while convergence = false & k < maxit dok = k + 1.5: $\boldsymbol{s} = W^{-1} (B\boldsymbol{v}_k - \alpha_k W \boldsymbol{q}_k).$ 6: $\begin{aligned} s &= w \quad (Bv_k - \alpha_k w q_k), \\ \beta_{k+1} &= \| \boldsymbol{s} \|_W; \ \boldsymbol{q}_{k+1} = \boldsymbol{s} / \beta_{k+1}, \\ \boldsymbol{r} &= A_{\gamma}^{-1} (B^T \boldsymbol{q}_{k+1} - \beta_{k+1} A_{\gamma} \boldsymbol{v}_k); \ \alpha_{k+1} &= \| \boldsymbol{r} \|_{A_{\gamma}}; \ \boldsymbol{v}_{k+1} = \boldsymbol{r} / \alpha_{k+1}, \\ \zeta_{k+1} &= -(\beta_{k+1} / \alpha_{k+1}) \zeta_k; \ \boldsymbol{d}_{k+1} &= (\boldsymbol{q}_{k+1} - \beta_{k+1} \boldsymbol{d}_k) / \alpha_{k+1}, \\ \boldsymbol{u}_{\gamma}^{(k+1)} &= \boldsymbol{u}_{\gamma}^{(k)} + \zeta_{k+1} \boldsymbol{v}_{k+1}; \ \boldsymbol{p}^{(k+1)} = \boldsymbol{p}^{(k)} - \zeta_{k+1} \boldsymbol{d}_{k+1}. \end{aligned}$ 7: 8: 9: 10: $convergence \leftarrow CHECK()$ 11: 12: end while 13: $\boldsymbol{u}^*_{\gamma} \leftarrow \boldsymbol{u}^{(k+1)}_{\gamma}; \, \boldsymbol{p} \leftarrow \boldsymbol{p}^{(k+1)}.$ 14: return u^*_{γ} and p

We will conclude with a remark about the choice of the matrix W. The matrix Wcan generally be any SPD matrix, and it can play two important roles. In the first case, we can relax the condition on the definiteness of A, such that it may only be symmetric positive semi-definite. With (2.1), the (1,1)-block A_{γ} is then SPD and the GKB algorithm can be applied. If A is however already SPD, this manipulation is not the purpose of the augmentation. The goal is then to obtain a linear system that is easier to solve, which translates into an improvement in convergence. For many matrices, the simple choice of W being the identity is enough [54]. In [55], the authors show that for the augmentation parameter γ being big enough, the algorithm converges in only a few iterations and mesh-independent convergence can be achieved. We emphasize, however, that while the iteration count for the outer Golub-Kahan method decreases, the matrix A_{γ} becomes more and more ill-conditioned. When an inner iterative solver is used, the number of inner iterations thus increases. It is important to find a good balance, and when the (1,1)-block exhibits favorable properties for certain solvers, it might be more efficient in terms of computation time to discard the augmentation. For the Stokes equation, the symmetric (1,1)-block corresponds to the discretization of the Laplace operator, for which efficient multigrid solvers exist. After experimentation, (see, e.g., [56, 57]) we decided not to use the augmentation of the (1,1)-block in Section 5 and set the parameter $\gamma = 0$. The transformation to obtain a zero vector in the upper right-hand side is still to be kept by using the matrix A in the transformation.

3.2 Augmented Lagrangian Uzawa method

The Augmented Lagrangian–Uzawa (ALU) method is an efficient variant of the classical Uzawa method [36]. As explained in [15, Section 8.2], it has the same solution as the saddle-point system (1.5). As before, let $\gamma \geq 0$ be a scaling factor for the

augmentation term $B^T W^{-1}B$. The ALU method corresponds to the Uzawa method applied to the system $\mathcal{A}_{\gamma} \boldsymbol{x} = \boldsymbol{b}_{\gamma}$ where \mathcal{A}_{γ} is defined in (3.1) and $\boldsymbol{b}_{\gamma} = [\boldsymbol{f}_{\gamma}, \boldsymbol{g}]^T$ with $\boldsymbol{f}_{\gamma} = \boldsymbol{f} + \gamma B^T W^{-1} \boldsymbol{g}$. In code_saturne, an incremental formulation of the ALU method as detailed in Algorithm 2 is considered. The previous discussion about the choice of the matrix W remains also true for ALU. We choose a diagonal matrix with entries equal to the volume of each mesh cell for W. An initial guess on the pressure (resp. velocity) field \boldsymbol{p}_0 (resp. \boldsymbol{u}_0) is needed in this method. It is either a null array in the steady case or the latest known pressure (resp. velocity) field in an unsteady case. The efficiency and convergence of the ALU method relies on the value of the parameter γ . As mentioned in [15], convergence holds for $\gamma \in (0, 2/\rho)$ where ρ is the largest eigenvalue of the Schur complement $-BA_{\gamma}^{-1}B^T$. A too large value for the parameter γ induces an ill-conditioned system which slows down the convergence of the inner linear system defined by A_{γ} . Once again, a trade-off between the iteration count and the cost of the inner resolution has to be found. In the experiments in Section 5, we choose $\gamma = 100$.

Algorithm 2 Augmented Lagrangian–Uzawa algorithm (incremental form). Input: A, B, W, $\boldsymbol{f}, \boldsymbol{g}, \boldsymbol{u}_0, \boldsymbol{p}_0, \gamma, \epsilon, k_{\text{max}}$. Output: u, p. 1: Initialise $A_{\gamma} = A + \gamma B^T W^{-1} B$; $\boldsymbol{f}_{\gamma} = \boldsymbol{f} + \gamma B^T W^{-1} \boldsymbol{g}$ and k = 1. 2: Velocity solve: $A_{\gamma} \boldsymbol{u}^{(1)} = \boldsymbol{f}_{\gamma} - B^T \boldsymbol{p}_0; \, \boldsymbol{\delta} \boldsymbol{u}^{(1)} = \boldsymbol{u}^{(1)} - \boldsymbol{u}_0.$ 3: Pressure update: $\delta p^{(1)} = \gamma W^{-1} (B u^{(1)} - g); p^{(1)} = p_0 + \delta p^{(1)}.$ while $\| \boldsymbol{\delta u}^{(k)} \| < \epsilon \| \boldsymbol{\delta u}^{(1)} \|$ & $k < k_{\max}$ do 4: Solve the velocity increment: $A_{\gamma} \delta u^{(k+1)} = -B^T \delta p^{(k)}$ 5: Velocity update: $\boldsymbol{u}^{(k+1)} = \boldsymbol{\delta} \boldsymbol{u}^{(k+1)} + \boldsymbol{u}^{(k)}$. 6: Pressure update: $\delta \boldsymbol{p}^{(k+1)} = \gamma W^{-1} (B \boldsymbol{u}^{(k+1)} - \boldsymbol{g}); \boldsymbol{p}^{(k+1)} = \boldsymbol{p}^{(k)} + \delta \boldsymbol{p}^{(k+1)}$. 7: k = k + 1.8: 9: end while 10: $\boldsymbol{u} \leftarrow \boldsymbol{u}^{(k)}$ and $\boldsymbol{p} \leftarrow \boldsymbol{p}^{(k)}$. 11: return u and p.

3.3 Notay's algebraic transformation

The following approach of algebraically transforming the linear system in (1.5) by a change of variables was initially introduced by Y. Notay in [34]. To keep our presentation self-contained, we briefly present the main idea. Notay's right-hand side transformation introduces the change of variables

$$\begin{pmatrix} \boldsymbol{u} \\ \boldsymbol{p} \end{pmatrix} = \begin{pmatrix} I & -\alpha D_A^{-1} B^T \\ O & I \end{pmatrix} \begin{pmatrix} \tilde{\boldsymbol{u}} \\ \tilde{\boldsymbol{p}} \end{pmatrix}$$
(3.4)

which, when combined with a change of sign of the divergence constraint, yields the system $\tilde{A}\tilde{x} = \tilde{b}$ where

$$\tilde{\mathcal{A}} := \begin{pmatrix} A & (I - \alpha A D_A^{-1}) B^T \\ -B & \alpha B D_A^{-1} B^T \end{pmatrix}, \qquad \tilde{\boldsymbol{x}} = \begin{bmatrix} \tilde{\boldsymbol{u}} \\ \tilde{\boldsymbol{p}} \end{bmatrix} \text{ and } \tilde{\boldsymbol{b}} = \begin{bmatrix} \boldsymbol{f} \\ -\boldsymbol{g} \end{bmatrix}$$
(3.5)

with a positive constant α and D_A being the diagonal matrix built from A. Hereafter, we assume $\alpha = 1$. Other transformations, e.g. the two-sided variant introduced in [58], are not considered in this work since they appear less efficient [34].

When we consider the vector v of the null space of A, we can decompose it with respect to the velocity and the pressure. It follows

$$\tilde{\mathcal{A}}\boldsymbol{v} = \boldsymbol{0} \Longleftrightarrow \begin{cases} A\boldsymbol{v}_u + (I - \alpha A D_A^{-1}) B^T \boldsymbol{v}_p = \boldsymbol{0} \\ -B\boldsymbol{v}_u + \alpha B D_A^{-1} B^T \boldsymbol{v}_p = \boldsymbol{0} \end{cases},$$

and since A is invertible, we have $\boldsymbol{v}_u = -A^{-1}(I - \alpha A D_A^{-1})B^T \boldsymbol{v}_p$. By substitution of \boldsymbol{v}_u into $-B\boldsymbol{v}_u + \alpha B D_A^{-1} B^T \boldsymbol{v}_p = \boldsymbol{0}$, we can derive $BA^{-1}B^T \boldsymbol{v}_p = \boldsymbol{0}$ so that

$$\boldsymbol{v}_p^T B A^{-1} B^T \boldsymbol{v}_p = 0 \Longleftrightarrow \boldsymbol{v}_p \in \ker(BA^{-1}B^T) = \ker(B^T).$$

By recalling the fact that the CDO discretization satisfies the solvability conditions (2.1) (e.g. $\ker(B^T) = \mathbf{0}$), it implies that the kernel of $\tilde{\mathcal{A}}$ is trivial, therefore the transformed matrix is invertible.

The (2,2)-block $BD_A^{-1}B^T$ of (3.5) is the product of a discrete divergence with a discrete gradient which is similar to a discrete Laplacian [58]. If we replace D_A^{-1} with A^{-1} , so that the (2,2)-block is the exact Schur complement of A, \tilde{A} becomes a lower triangular block system which can be easily solved. However, the computation of A^{-1} in a large problem is often too expensive so that we may want to use an approximation of the Schur complement, instead. We also refer to [59] for sparse approximations of the Schur complement.

4 Algebraic multigrid preconditioning for Krylov solvers

Multigrid methods are among the most relevant methods to solve large-scale systems. They can be used as a solver or preconditioner. In this paper, multigrid methods are always used as preconditioners. Multigrid methods have shown to be efficient preconditioners in different settings, in particular for scalar diffusion and convection-diffusion problems [22, 30, 60–62]. Let us recall that the suboptimal performance of plain aggregation AMG schemes for diffusion operators is well known and documented in the literature [30] which has motivated the research on approaches like smoothed aggregration and the K-cycle, for instance.

The methods considered in this article expose a considerable number of options to the user. So, these AMG methods can be extensively tuned. Nevertheless, an exhaustive search of the combinatorial parameter space is beyond the scope of our current

work. Their performance on a given system depends on a certain number of parameters, such as the choice of the smoother, the number of smoother iterations, the coarsening operator (pairwise or N-times pairwise for the aggregration schemes or different aggressive coarsening techniques for the C/F schemes) and the number of levels, the interpolation strategy between grids, etc. Finding an appropriate or even optimal set of coarsening parameters is non-trivial. For example, one has to strike a balance between coarsening factors and computational costs. A too strong coarsening factor (which translates into a smaller number of levels and in the case of plain aggregation AMG in general also implies lower operator complexities) tends to increase the number of iterations to solve linear systems while its memory requirement is reduced. On the other hand, a too small coarsening factor requires a larger number of levels to reduce the size of the coarsest system below a given threshold. In general, an AMG setting leads to a coarsening factor (τ) and an operator complexity (C) such that

$$2 \le \tau \le 4$$
 and $1.3 \le \mathcal{C} \le 2$,

where τ and C are defined by

$$\tau = \frac{1}{L} \sum_{\ell=1}^{L} \frac{n_{\ell}}{n_{\ell-1}} \quad \text{and} \quad \mathcal{C} = \frac{1}{nnz(A_L)} \sum_{\ell=0}^{L} nnz(A_{\ell}),$$

on a hierarchy of matrices A_{ℓ} , $0 \leq \ell \leq L$, A_0 being the matrix on the coarsest and A_L the matrix on the finest level with n_{ℓ} and $nnz(A_{\ell})$ denoting the number of unknowns and the number of nonzero elements of A_{ℓ} on each level, respectively. Following the recent work of Lin, Shadid and Tsuji [63], we have included Krylov smoothers in our comparison.

5 Numerical experiments

In this section, we present numerical results for the solution of the saddle-point linear system (1.5) arising from a CDO face-based discretization with the algorithms described previously. We will use the GKB method for the solution of the Stokes equations and ALU for the Navier–Stokes equations and will then compare both of them to the Notay's algebraic transformation method.

The test problems were run with the open source CFD software code_saturne, which in addition to in-house implementations of several Krylov solvers and preconditioners, provides interfaces to external solvers, such as MUMPS [16], Hypre [46, 47, 64] and PETSc [65] (e.g., please see Appendix A for the usage). In the following experiments, we employ Hypre through the PETSc interface. In our comparison, we tested different AMG approaches:

- (in-house) K-cycle of code_saturne inspired by the algorithm described in [22]. It combines therefore multiple passes pairwise aggregation coarsening with Krylov-acceleration on the intermediate levels.
- BoomerAMG of the Hypre library with the classical C/F (Ruge-Stüben) or the HMIS coarsening [29, 30, 66] with/without aggressive coarsening

- GAMG of the PETSc library [48] with plain and smoothed aggregation schemes [31]
- HMG of the PETSc library which is a recent hybrid AMG toolbox [67]

More precisely, HMG allows us to choose any available PETSc solver/preconditioner as smoother and as coarse solver while using Hypre BoomerAMG, GAMG, or other multilevel methods for the construction of coarser level matrices and interpolation. For example, in our test cases, we combine the context of Hypre BoomerAMG for coarsening and the context of GAMG for smoothing. We refer to [65] for more details of the usage of HMG. Furthermore, we provide Appendix B for the example of PETSc options to use AMG preconditioning in our numerical experiments.

code_saturne is not able to perform sparse matrix-matrix multiplications and computations with non-square matrices up to now. These operations have to be done in an external environment. We use the PETSc library for this purpose and thus only consider linear solvers available through PETSc when using Notay's algebraic transformation.

Table 1 sums up the main characteristics of the AMG methods used in the computations.

	K-cycle	GAMG	BoomerAMG
Applied through	code_saturne	PETSc	PETSc
Type of cycle	K-cycle	V-cycle	V-cycle
Type of coarsening	pairwise	smoothed	classical or HMIS
	aggregation [22]	aggregation [31]	coarsening [66]
aggreg. limit / threshold	mesh dep	endent, but factors satisfyin	g $\tau \approx 4$ or $C \approx 1.3$
max. levels		10 (i.e. $L \le 10$)	
		Smoothers	
Pre-/post-smoothing	Defau	lt: one step each; otherwise s	stated explicitly
symm. Gauss-Seidel	yes	yes	yes
Krylov smoother	no	GMRES(10)	GMRES(10) (through HMG)
Stop. criterion Kryl. sm.	N/A	max 10 iter / tol = 10^{-3}	$\max 10 \text{ iter } / \text{ tol} = 10^{-3}$

 Table 1 Settings for the algebraic multigrid preconditioners.

General setting for outer solvers

We compare and discuss combinations of solvers and preconditioners for the three methods GKB, ALU and Notay's algebraic transformation. The choice of settings for the GKB and ALU method with respect to the inner and outer solvers and a symmetric or non-symmetric inner system can be found in Table 2. For the non-symmetric matrix in the algebraic transformation, we use FGMRES. We choose 10^{-8} as the stopping tolerance and set the maximum number of iterations to 10 000. As preconditioners for this system, as well as for the inner systems in the ALU and GKB methods, we use algebraic multigrid methods. The setups and different combinations are given in Table 1. We want to emphasize that the choice of parameters, i.e. to define a threshold, for the construction of aggregation or prolongation is a significantly important matter. In particular, in case of GAMG for Stokes systems, this parameter must not be low

and to provide relevant coarser levels, we found that $C \approx 1.3$. Further particular configurations are discussed during the description of the numerical experiments.

Configuration setup for the test cases

We finish this section with giving the configurations for the test cases:

- Hardware: Intel(R) Core(TM) i9-10885H CPU @ 2.40GHz, 32GB
- The version of code_saturne: code_saturne 7.2.2
- The version of PETSc: 3.15.0
- The version of HYPRE: 2.20
- The version of MUMPS: 5.3.5
- No parallel computing (neither MPI nor openMP), i.e. sequential run.

Outer tolerance Augment. parameter γ	$\begin{array}{c} {\rm GKB} \\ 10^{-8} \\ 0 \end{array}$	$\begin{array}{c} \mathrm{ALU} \\ \mathrm{10^{-7}} \\ \mathrm{100} \end{array}$
Inner solver sett	ings	
Inner tolerance	10^{-8}	10^{-7}
symm. inner solver	FCG	-
non-symm. inner solver	-	FGMRES
non-symm. inner solver Preconditionner (PETSc)	- Hypr	FGMRES e, GAMG

Table 2 Settings for the GKB and ALU algorithms.

5.1 Taylor-Green Vortex problem (3D Stokes problem)

The first test case is an adaptation of the Taylor-Green Vortex (TGV) problem with an analytic solution consisting of the product of sine and cosine functions. Its analytic solution satisfies the steady Stokes equations, i.e. the equations (1.1) and (1.2) with $\chi = 0$ and $\nu = 1$. The analytic solution is given by

$$\vec{u} = \begin{pmatrix} 0.5\sin(2\pi x)\cos(2\pi y)\cos(2\pi z)\\ 0.5\cos(2\pi x)\sin(2\pi y)\cos(2\pi z)\\ -\cos(2\pi x)\cos(2\pi y)\sin(2\pi z) \end{pmatrix} \text{ and } p = \sin(2\pi x)\sin(2\pi y)\sin(2\pi z),$$

where $(x, y, z) \in [0, 1]^3$ and \vec{f} is given to fulfil (1.1). The solution is presented graphically in Figure 2.

For the discretization, we will use two types of meshes. The first one is Cartesian and the second one is tetrahedral. In the abbreviations used in the following, we denote the Cartesian meshes by H and the tetrahedral ones by T. These letters are followed by the number of elements. The resulting numbers of degrees of freedom for the velocity and pressure components are given in Table 3. Since this 3D TGV problem leads to a symmetric matrix, we may use the GKB algorithm. We will compare its performance to Notay's algebraic transformation approach.

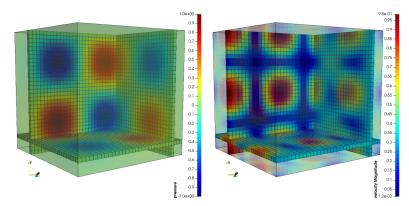


Fig. 2 3D TGV problem: pressure field (left) and velocity field (right).

Table 3 3D TGV problem: the degrees of freedomwith respect to meshes.

	Degrees of freedom						
Mesh	Velocity	Pressure	Total unknowns				
H32	304128	32768	336 896				
H64	2396160	262144	2658304				
T5	188361	30480	218841				
T6	374964	61052	436016				

5.1.1 Solution by GKB

The GKB algorithm is a segregated method for matrices of type (1.5) with a symmetric (1,1)-block and we need, as explained earlier, an inner iterative solver for the (1,1)-block. Here, we choose a preconditioned flexible conjugate gradient method. In general, we can employ the GKB algorithm via PETSc or directly in code_saturne, but in the following experiments we choose the in-house implementation. Since the system of inner iterations is symmetric, we are able to use GAMG with smoothed aggregation. Table 4 illustrates the elapsed times and the numbers of iterations, where each column corresponds to the respective preconditioner of FCG. The numbers of iterations are sufficiently good for those three preconditioners. However, in terms of setup time, the in-house K-cycle aggregation step shows a much better performance than PETSc. To summarise the result of Table 4, we have

Setup time: K-cycle < Hypre BoomerAMG < GAMG Solution time: GAMG < K-cycle < Hypre BoomerAMG Total time for H32: K-cycle < GAMG < Hypre BoomerAMG Total time for H64: GAMG < K-cycle < Hypre BoomerAMG The number of iterations: Hypre BoomerAMG < GAMG < K-cycle

These results show that a higher setup time might be recovered by a faster solution process, and also that a low number of iterations might not necessarily mean that the method performs best in practice. We furthermore see that the performance of the solvers is mesh-independent. The winner in the total time to solution is GAMG for H64, whereas for H32, it is the in-house K-cycle (although, admittedly, not by far).

GKB Strategy Mesh H32H64GAMG Preconditioner GAMG K-cycle Hypre K-cycle Hypre 7.9725.62.843.77 Setup time (s) 0.9631.8Solution time (s) 18.121.615.9173.6187.4140.5Total time (s) 19.124.419.7181.6213.0172.3# iterations 13/21/2837/21/1475/21/10513/23/299 5/22/1107/23/161 (inner/outer/total)

Table 4 3D Stokes: performance of numerical methods for Stokes system with GKB and AMGpreconditioned FCG as inner solver.

Remark 5.1 In our numerical experiments, the GKB algorithm is used without augmentation, i.e., we set $\gamma = 0$. On the other hand, the ALU algorithm requires a rather large augmentation parameter to show a good performance even in the symmetric case. The downside is the resulting, more ill-conditioned matrix A_{γ} in the inner solution step. As pointed out in [51], due to its favorable performance caused mainly by an efficient inner solution, the GKB algorithm is preferred to the ALU algorithm in Stokes problems. Therefore, we do not provide any further discussion for the ALU method in the symmetric case.

5.1.2 Solution by algebraic transformation

As second approach, we apply Notay's algebraic transformation and solve the resulting system with FGMRES preconditioned by multigrid. For BoomerAMG, we keep the settings of Table 1, whereas we follow the official PETSc manual (please see [65, Section Algebraic Multigrid (AMG) Preconditiners]) for recommended settings of GAMG parameters for non-symmetric problems (e.g., see Appendix B.2). Results are presented in Table 5. Although Hypre requires a higher time for setup, its overall solution time is smaller than the one of GAMG. This effect is especially visible for the mesh H64.

For comparing these results to the GKB method, the measure of numbers of iterations can not be used, since the transformed matrix is of another size than \mathcal{A}_{γ} and thus the cost of one iteration is different. Comparing the time to solution in Tables 4 and 5, we observe that Hypre on the transformed system is faster than the any of the tested multigrid preconditioners in the GKB formulation.

We have carried out further test cases to study the influence of different cycletypes for the multigrid method. As seen in Table 6, using multigrid cycles other than V(1,1) in BoomerAMG does not show any improvement in terms of iteration numbers and solution times, since the interpolation operator of BoomerAMG is good enough. We observe that increasing the number of relaxation steps in GAMG reduces iteration

Strategy	Algebraic transformation					
Mesh	H32		H64			
Preconditioner	Hypre	GAMG	Hypre	GAMG		
Setup time (s) Solution time (s) Total time to solution # iterations	6.344 12.243 18.587 36	$\begin{array}{c} 4.029 \\ 16.589 \\ 20.618 \\ 64 \end{array}$	54.707 105.760 160.467 36	32.965 315.444 348.409 168		

Table 5 3D Stokes: algebraic transformation with AMGpreconditioned FGMRES.

counts, see Table 6. For the non-symmetric system under consideration, BoomerAMG outperforms GAMG. We have empirically tried to figure out optimal settings for the usage of GAMG, but no choice worked satisfactorily. Thus, hereafter, only Hypre BoomerAMG is used in combination with symmetric Gauss-Seidel smoothing for the preconditioning step in the algebraic transformation strategy.

Table 6 3D Stokes: algebraic transformation with respect to multigrid cycles on H32.

Iterative solver	FGMRES						
Preconditioner	Hypr	e Boomer	AMG	GAMG			
Multigrid cycle	V(1,1)	V(2,2)	V(3,3)	V(1,1)	V(2,2)	V(3,3)	
Total time (s) # iterations	$11.540 \\ 26$	$14.911 \\ 28$	$ \begin{array}{r} 18.832 \\ 30 \end{array} $	17.317 123	18.174 77	$19.407 \\ 59$	

In a second trial, we solved the system arising in the change of variable approach on the tetrahedral meshes T5 and T6 with Hyper BoomerAMG. Table 7 indicates that the algebraic transformation of the matrix works also well for tetrahedral meshes. While increasing the number of smoothing steps does not improve iteration numbers on the Cartesian grid (Table 6), more pre- and post-smoothing steps do reduce the number of iterations for tetrahedral meshes, see Table 7, since the tetrahedral meshes are more complex. In terms of the elapsed time, V(3,3) is however slower than V(1,1).

Table 7 3D Stokes: V(1,1) and V(3,3) on tetrahedral meshes with Hypre BoomerAMG.

Iterative solver	FGMRES					
Mesh	Г	`5	T6			
Multigrid cycle	V(1,1)	V(3,3)	V(1,1)	V(3,3)		
Total time (s) $\#$ iterations	$21.608 \\ 37$	27.648 26	$58.962 \\ 37$	71.328 27		

Stationary and relatively basic algorithms such as the (weighted) Jacobi or Gauss-Seidel iterations are frequently used as smoothers in multigrid cycles. However, according to [63, 68] Krylov methods can be relevant as smoothers in multigrid settings as well.

In the following, we report on FGMRES preconditioned by AMG (here, GAMG) with Krylov-smoothing for the non-symmetric problems. For the Krylov smoother, we allow up to 10 GMRES iterations combined with an inner tolerance of 10^{-3} , which stops when either one or the other criterion is reached. We may now think in a nested way and precondition the smoother itself, e.g., the Gauss-Seidel method as preconditioner for the smoother GMRES(10). At the coarsest level, we solve the system with a direct solver. Table 8 shows that Krylov smoothing reduces in general the number of iterations. However, comparing these results to Table 5, the elapsed time is greater than for Gauss-Seidel smoothing. Furthermore, we see a dependence of the solver performances on the meshes, i.e., the iteration numbers are unstable.

 Table 8
 3D Stokes: GAMG with Krylov smoothers

 FGMRES[30](GAMG(GMRES(10)+X)).

Mesh	Smoother	#it	Time (s)
H32	GMRES	29	175.226
пэ2	GMRES+SGS	3	39.534
H64	GMRES	48	2426.144
п04	GMRES+SGS	29	2980.572
T6	GMRES	84	715.284
10	GMRES+SGS	11	187.401

Although Krylov smoothing cannot decrease the elapsed time in GAMG, it may lead to smaller iteration numbers for combinations of meshes and solvers. The HMG context of PETSc allows us to combine Krylov smoothers with classical AMG coarsening in Hypre. In Table 9, it is shown that HMG with symmetric Gauss-Seidel smoothing shows similar numerical results as Hypre BoomerAMG. On the other hand, using Krylov smoothers sometimes reduces the number of iterations, whereas more time is spent in the solver. Similarly, as for Krylov smoothing in GAMG, the performances depend on the mesh and the experiments failed for fine meshes. Thus, we do not observe any advantage from using HMG with Krylov smoothing in Stokes systems for our particular test cases.

5.1.3 Summary for the Stokes test case

Our study compared the GKB and algebraic transformation methods in the Stokes test case. We found that AMG preconditioned GMRES solvers were robust in solving the inner system of the GKB method. The three distinct coarsening strategies yielded comparable results and are independent of mesh sizes. In the algebraic transformation, only the classical AMG via Hypre BoomerAMG showed robustness. We attempted to improve the numerical performance of GAMG by employing Krylov smoothers, but this approach required much more computational cost and the results

Table 9 3D Stokes: HMG preconditioning FGMRES[30](HMG), Boomer classical coarsening with max it=200 (if # it \geq 200, the test fails).

Mesh	Smoother	# it	Time (s)
H32	SGS GMRES+SGS	$\frac{28}{135}$	$13.483 \\ 338.316$
H64	SGS	29	119.388
	GMRES+SGS	H	Failed
Т6	SGS	32	52.344
	GMRES+SGS	I	Failed

depended on the meshes. Furthermore, with the proper use of AMG preconditioning, the algebraic transformation approach was slightly faster than the GKB method. Therefore, we recommend using the algebraic transformation with $\alpha = 1$ in combination with BoomerAMG and the symmetric Gauss-Seidel smoother in practice, as it is a monolithic method that yields better numerical performance.

5.2 Burggraf problem (2D Navier–Stokes)

As second test case, we consider the steady incompressible Navier–Stokes problem given by equations (1.1) and (1.2) with $\chi = 1$ and $\nu = 1/Re$. To resolve the nonlinearity, we apply Picard's iteration and solve the Oseen's problem (1.3)-(1.4). As for the Stokes problem, we are interested in solving the saddle-point systems arising from a CDO discretization scheme. Due to the convection terms of the velocity fields, the (1,1)-block becomes non-symmetric. The test problem in the following is a 2D Burggraf flow [69], which is a 2D analytic polynomial solution to the Navier–Stokes equation. The exact solution is given by

$$\begin{split} \vec{u} &= \begin{pmatrix} 16x^2(x^2 - 2x + 1)y(2y^2 - 1) \\ -16x(2x^2 - 3x + 1)y^2(y^2 - 1) \end{pmatrix}, \\ p &= \frac{8}{Re} \Big(24x^3(0.2x^2 - 0.5x + 1/3) + 4x(2x^2 - 3x + 1)(12y^2 - 2) + (24x - 12)(y^2(y^2 - 1)) \Big) \\ &+ 64 \Big(0.5(x^2(x^2 - 2x + 1))^2 (24y^3(y^2 - 1) - 2y(12y^2 - 2)(2y^2 - 1)) \\ &- 2y^3(y^2 - 1)(2y^2 - 1)(x^2(x^2 - 2x + 1)(12x^2 - 12x + 2) - 4x^2(2x^2 - 3x + 1)^2)) \Big) \end{split}$$

For more details about the exact solution, we refer to [51, 69]. As a default parameter, we assume the Reynolds number to be Re = 100. A centered convection scheme is used. Furthermore the relative (resp. absolute) stopping criterion for the Picard iteration is set to 10^{-6} (resp. 10^{-12}). The maximal number of Picard iterations is equal to 50.

The solution is graphically presented in Figure 3. For the 2D Burggraf model, we consider the Cartesian meshes H64, H128, H256 and H512. The numbers of degrees of freedom for \boldsymbol{u} and \boldsymbol{p} on these meshes along with the discrete L_2 error norms for the velocity at cells, $E_2(\boldsymbol{u})$, and for the pressure, $E_2(\boldsymbol{p})$, are gathered in Table 10. The expected second order of convergence rate is reached for the velocity. One recovers also a second order convergence rate for the pressure which is higher than the expected

first order. This super-convergence stems from the smoothness of the solution and the usage of uniform Cartesian meshes. These results also validate the values of the stopping criteria for the Picard iterations and the iterative solvers.

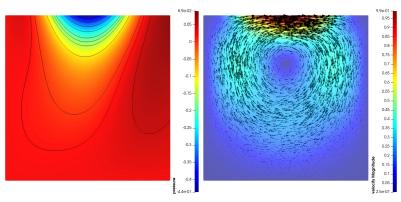


Fig. 3 2D Burggraf flow with Re = 100: pressure field (left) and velocity field (right).

Table 10 2D Burggraf problem at Re = 100: Errors norms and convergence rates for the velocity and the pressure fields at cells along with the number of degrees of freedom.

	Degrees of freedom				Err	ors	
Mesh	Velocity	Pressure	Total unknowns	$E_2(\boldsymbol{u})$	$\operatorname{rate}(\boldsymbol{u})$	$E_2(\boldsymbol{p})$	$rate(\boldsymbol{p})$
H64	49536	4096	53632	4.40e-3	-	3.60e-3	-
H128	197376	16384	213760	1.10e-3	2.0	9.06e-4	2.0
H256	787968	65536	853504	2.76e-4	2.0	2.28e-4	2.0
H512	3148800	262144	3410944	6.89e-5	2.0	5.72e-5	2.0

Due to the non-symmetry of the (1,1)-block, the GKB method is no longer applicable but see also [70]. Notably, the algebraic transform approach remains a viable option in this scenario. For comparison, we also employ the ALU method, as described in Section 3.2.

5.2.1 Solution by ALU method

The ALU method is set with a relative (resp. absolute) tolerance equal to 10^{-6} (resp. 10^{-12}) and a maximal number of iteration equal to 50. Many settings have been tested in order to keep the best one between the default settings and the tested ones. Here is the list of the strategies used to solve the augmented inner linear system associated to A_{γ} :

(MUMPS) LU factorization performed with the MUMPS library. An analysis by block relying on the Approximate Minimal Degree (AMD) [71] is used.

(HMG) FGMRES with right preconditioning, a restart after 25 iterations and a relative tolerance set to 10^{-6} . Upper triangular block preconditioning is used (relying on the fieldsplit feature of PETSc) and one applies one V-cycle of HMG on each diagonal block. 1 iteration of SGS is applied as down and up smoother.

(GAMG) Same as above but HMG preconditioner is replaced by GAMG. GAMG relies on the smooth aggregation (2 steps)

(BAMG) Same as above but HMG preconditioner is replaced by boomerAMG. BoomerAMG is set with a *strong threshold* equal to 0.30 and uses 2 levels of aggressive coarsening. Moreover, the SGS smoother is replaced by a forward Gauss-Seidel for down smoothers and by a backward Gauss-Seidel for up smoothers.

The main parameter associated to this method is the scaling coefficient related to the augmentation term. Figure 4 illustrates a sensitivity study on the scaling coefficient γ with respect to the CPU time, the number of iterations of the ALU method and the total number of iterations associated to the inner linear system. The optimal value of γ is different between a direct and an iterative inner solver but the number of iterations of the ALU method is very close. $\gamma \in [10, 10^4]$ delivers optimal performances for a direct solver, while $\gamma \in [0.1, 0.5]$ delivers the best performances for an iterative solver. For an iterative inner solver, the optimal value corresponds to a trade-off between the number of ALU iterations needed at each Picard iteration. These results show the robustness of a direct approach and the strong influence of γ on the conditioning of the A_{γ} matrix.

We then collect the detailed results related to the ALU method for the best candidate in each strategy in Table 11. A comparative study of the performance of the different strategies with respect to the mesh refinement is plotted in Figure 5. The strategy ALU(100).MUMPS is the most efficient tested strategy for this 2D case. The strategy ALU(0.2).HMG and ALU(0.2).BAMG are *h*-independent. One can expect from the slopes of the plot CPU time vs. number of velocity DoFs that the ALU(0.2).HMG strategy will outperform the ALU(100).MUMPS strategy after 1 or 2 additional mesh refinements.

5.2.2 Solution by algebraic transformation

A first set of computations with the algebraic transformation is done using a LU factorization with MUMPS through the PETSc library. This reference is compared to iterative strategies relying on:

(HMG) FGMRES with right preconditioning, a restart after 60 iterations and with a relative (resp. absolute) tolerance set to 10^{-8} (resp. 10^{-14}). An upper triangular block preconditioning is used (relying on the fieldsplit feature of PETSc) and one applies one V-cycle from the HMG toolbox using 1 step of a local forward Gauss-Seidel as down smoother and 1 step of local backward Gauss-Seidel as up smoother.

(GAMG) Same settings as above but HMG is replaced by GAMG relying on the smooth aggregation (default settings) and 1 iteration of SGS as smoother.

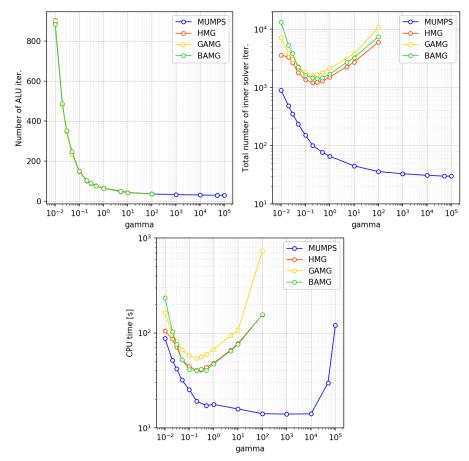


Fig. 4 2D Burggraf flow test case at Re = 100 with the mesh H128: Sensitivity of the scaling coefficient γ in the ALU method with respect to the number of (cumulated) ALU iterations (top left), to the total number of iterations (top right) and to the CPU time (bottom).

We first perform a sensitivity analysis to the tolerance criterion for the mesh H128 since the stopping criterion holds on the transformed system. The results are collected in the Table 12. We observe that the same level of accuracy on the error norms is obtained with a stronger tolerance compared to the ALU method. The pressure field and the approximation of the velocity divergence are the two quantities where this loss of accuracy appears first. To circumvent this issue, the relative tolerance on the FGMRES is set to 10^{-8} (10^{-14} for the absolute one). The mean number of iterations of the FGMRES is also higher than with the ALU method so that a sensitivity to the number of iterations before restarting the FMGRES has been done. A value equal to 60 is a good trade-off between the efficiency and the memory usage.

To further reduce the cost of this approach, we investigate also the influence of the scaling parameter α used in the change variable (3.4). The results of these tests collected in Figure 6 indicates that $\alpha = 10$ is a good value.

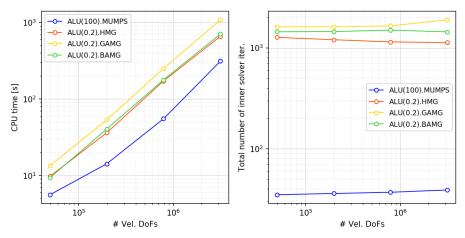


Fig. 5 2D Burggraf flow test case at Re = 100 using the ALU method: Evolution of the CPU time with respect to the mesh refinement (left) and with respect to the total number of iterations of the inner solver (right) for the different strategies used as inner solver.

Mesh	strategy		# iterat	Time [s]		
		Picard	ALU	Total inner	setup	solve
	ALU(100).MUMPS	13	36	36	11.0	14.2
H128	ALU(0.2).HMG	13	102	1204	12.9	36.1
11120	ALU(0.2).GAMG	13	104	1618	13.8	53.2
	ALU(0.2).BAMG	13	104	1451	13.2	63.4
	ALU(100).MUMPS	13	37	37	41.9	55.1
H256	ALU(0.2).HMG	13	103	1146	56.2	171.7
11200	ALU(0.2).GAMG	13	106	1652	61.9	250.6
	ALU(0.2).BAMG	13	106	1499	54.8	271.9
	ALU(100).MUMPS	13	39	39	252.2	310.7
H512	ALU(0.2).HMG	13	108	1128	226.0	658.2
11012	ALU(0.2).GAMG	15	122	1895	265.1	1081.6
	ALU(0.2).BAMG	13	110	1 441	229.9	1071.4

Table 11 2D Burggraf test case at Re = 100: Results for the different strategies relying on the ALU(γ) method.

Table 12 2D Burggraf test case at Re = 100: Sensitivity of the relative tolerance for the algebraic transformation with the GAMG strategy, $\alpha = 10$ and the mesh H128.

relative tolerance	# iterations		Time [s]	Error	norms	$ div(\boldsymbol{u}) _{\infty}$
	Picard	Total inner	solve	$E_2(\boldsymbol{u})$	$E_2(\boldsymbol{p})$	
10^{-6}	13	299	22.7	1.51e-3	1.84e-3	4.0e-3
10^{-7}	13	415	25.9	1.10e-3	8.80e-4	3.4e-4
10^{-8}	13	529	28.3	1.10e-3	9.04e-4	2.5e-5
10^{-9}	13	633	32.1	1.10e-3	9.06e-4	2.8e-6
10^{-10}	13	737	34.7	1.10e-3	9.06e-4	2.5e-7

25

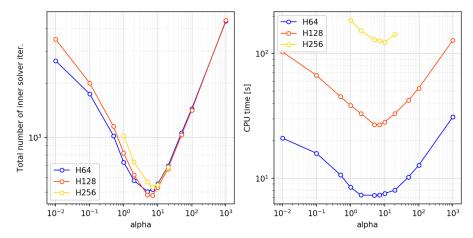


Fig. 6 2D Burggraf flow test case at Re = 100 using the algebraic transformation and GAMG as preconditioner: Evolution of the total number of iterations with respect to the scaling parameter α for the mesh H64, H128 and H256 (left); Evolution of the CPU time with respect to the scaling parameter α for the mesh H64, H128 and H256 (right).

The detailed results for the algebraic transformation are gathered in Table 13 and the evolution of the CPU time and that of the total number of iterations are displayed in Figure 7. One observes that the performance of the algebraic transformation when refining the mesh deteriorates slightly with MUMPS and GAMG. The strategy relying on HMG is not *h*-robust. GAMG exhibits much better performances than HMG.

Mesh	Mesh Solver strategy		# iterations		Time [s]		Error norm	
		Picard	Total inner	setup	solve	$E_2(\boldsymbol{u})$	$E_2(\boldsymbol{p})$	
	Notay(1).MUMPS	13	13	24.1	25.1	1.10e-3	9.06e-4	
H128	Notay(10).HMG	13	1660	9.3	53.7	1.10e-3	9.06e-4	
	Notay(10).GAMG	13	529	9.9	29.7	1.10e-3	9.04e-4	
	Notay(1).MUMPS	13	13	119.6	123.1	2.76e-4	2.28e-4	
H256	Notay(10).HMG	13	3620	42.0	418.6	2.76e-4	2.28e-4	
	Notay(10).GAMG	13	545	43.5	128.8	2.76e-4	2.29e-4	
	Notay(1).MUMPS	13	13	655.8	673.0	6.89e-5	5.72e-5	
H512	Notay(10).HMG	13	6637	169.8	2936.8	6.89e-5	5.77e-5	
	Notay(10).GAMG	13	660	168.1	568.5	7.38e-5	6.40e-5	

Table 13 2D Burggraf test case at Re = 100: Results for the different strategies relying on the algebraic transformation. Each strategy is denoted by Notay(α).XXX where XXX is a shortcut of the strategy used to solve the transformed system.

Remark 5.2 There is no influence of the value of the α parameter when using a direct solver. Results with BoomerAMG are not detailed with the algebraic transformation since the different tested settings deliver poor performances when $\alpha = 1$ and lack of robustness when

 $\alpha>1.$ For instance, the CPU time to solve the system with BoomerAMG is equal to 952.9s for the mesh H256.

 $Remark\ 5.3$ Krylov smoothers like GMRES/ILU(0) have also been tested but yield poor performances too.

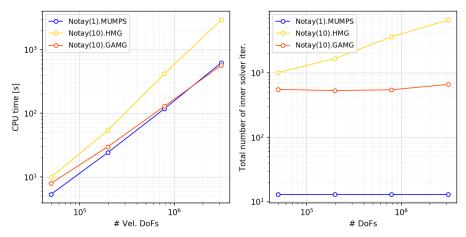


Fig. 7 2D Burggraf flow test case at Re = 100 using the algebraic transformation: Evolution of the CPU time with respect to the mesh refinement (left) and with respect to the total number of iterations (right) for the different strategies.

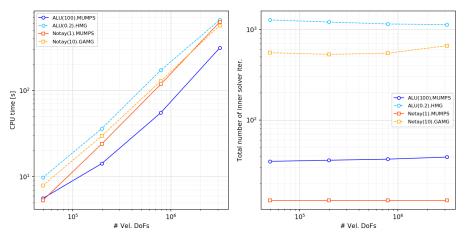


Fig. 8 Comparison of the performances of the best strategies used to solve the 2D Burggraf flow test case at Re = 100: Evolution of the CPU time with respect to the mesh refinement (left) and with respect to the total number of iterations (right).

The algebraic transformation is less efficient than the ALU method since the ALU method with MUMPS is the most efficient tested strategy for this test case. However, focusing on iterative approaches, the algebraic approach with GAMG is faster than the best strategy with an ALU method (HMG); see Figure 8.

5.2.3 Reynolds numbers

Lastly, we investigate how the variation of the Reynolds number influences the performance of the solver. As the Reynolds number increases, the nonlinear term in the Navier–Stokes equation becomes increasingly important, which makes the system of equations more difficult to solve. In particular, the (1,1)-block A becomes dominated by the convective term and more and more ill-conditioned.

Results are gathered in Table 14 for the ALU method and in Table 15 for the algebraic transformation. The main conclusion for all the tested strategies of resolution is that only a direct solver is robust with respect to Re. Moreover, the ALU(γ).MUMPS strategy appears as the most efficient tested strategy. Indeed, the CPU per Picard iteration remains stable with a direct approach. In the case of ALU, it is even more efficient at higher Re but the value of γ should be adapted at low Re to recover the same level of efficiency. For instance, the CPU time to solve the problem goes down to 4.4s using $\gamma = 1000$ at Re = 1. The algebraic transformation at Re = 1000 fails to converge with the settings which is tuned for Re = 100.

Table 14 2D Burggraf test case: Efficiency with respect to Re on the mesh H128 for the ALU method. The column *mean* corresponds to the mean number of iterations needed to reach the tolerance at each ALU iteration.

Strategy	Re	# iterations		ſ	Time [s]		Error norm		
50100085	100	Picard	ALU	mean	total	solve	solve/Picard	$E_2(\boldsymbol{u})$	$E_2(\boldsymbol{p})$
ALU(100)	1	4	16	1	16	7.6	1.9	1.97e-4	6.50e-4
	10	6	18	1	18	8.7	1.4	2.14e-4	7.08e-4
MUMPS	100	13	36	1	36	14.2	1.1	1.10e-3	9.06e-4
	1000	25	83	1	83	24.9	1.0	1.76e-2	1.50e-2
ALU(0.2)	1	4	685	5	3841	97.6	24.4	1.97e-4	6.50e-4
	10	6	137	7	1031	27.9	4.6	2.14e-4	7.08e-4
HMG	100	13	102	11	1204	36.1	2.8	1.10e-3	9.04e-4
	1000	24	681	66	45531	1129.9	47.1	1.76e-2	1.50e-2

Remark 5.4 The value of $\gamma = 0.2$ is optimal only for Re = 100. For instance, the CPU time needed to solve the problem with ALU(20).HMG at Re = 1 is equal to 13.0s and 18.1s with ALU(2).HMG at Re = 10.

Strategy	Re	# iterations		Time [s]		Error norm		
	10		mean	total	solve	solve/Picard	$E_2(\boldsymbol{u})$	$E_2(\boldsymbol{p})$
	1	4	1	4	7.9	2.0	1.97e-4	6.50e-4
Neteral MUMDO	10	6	1	6	12.1	2.0	2.14e-4	7.08e-4
Notay(1).MUMPS	100	13	1	13	25.1	1.9	1.10e-3	9.06e-4
	1000	25	1	25	50.1	2.0	1.76e-2	1.50e-2
	1	4	50	200	10.0	3.9	1.97e-4	6.50e-4
N-+(10) CAMC	10	6	43	258	14.5	3.5	2.14e-4	7.08e-4
Notay(10).GAMG	100	13	40	529	29.7	2.3	1.10e-3	9.04e-4
	1000	Failed	-	-	-	-	-	-

Table 15 2D Burggraf test case: Efficiency with respect to Re on the mesh H128 for the algebraic transformation.

5.3 Lid driven cavity benchmark (2D Navier–Stokes)

We now consider the classical "Lid driven cavity" validation benchmark for the steady incompressible Navier–Stokes equations. This benchmark corresponds to a flow inside a unit square cavity induced by the movement of the top horizontal wall. A homogeneous Dirichlet boundary condition on the velocity is enforced on the two vertical boundaries and on the bottom horizontal boundary. A unit velocity field aligned with the horizontal axis is enforced on the top horizontal boundary. The value of the viscosity is set so that $Re = 1/\nu$. Contrary to the previous Burggraf test case, there is no analytical solution and the solution presents singularities on the velocity field at the two top corners which makes this test case more challenging. Among the broad literature available, we focus on the configuration at Re = 100 and refer to the work of Ghia et al. [72].

Cartesian meshes are used. They are slightly different to those used in the Burggraf test case in order to avoid interpolating the cell velocity vector used to plot the horizontal and vertical profiles gathered in Figure 9. Postprocessings of the velocity field are depicted in Figure 10. Results are very close to those obtained in [72].

We collect a synthesis of the performance results for two strategies (MUMPS and HMG) of the ALU method in Table 16 and for two strategies (MUMPS and GAMG) of the algebraic transformation in Table 17. The observations made in the Burggraf problem remain valid for the Lid driven cavity problem.

5.4 Summary for the Navier–Stokes test cases

As in the Stokes problem, we could solve the linearized Navier–Stokes operator by AMG preconditioned FGMRES with the algebraic transformation strategy. For the fixed Reynolds number of 100 and the moderately sized matrices used, the ALU method with an inner direct solver outperformed other methods. However, these results may not remain valid as the system size increases beyond a critical point, making it impractical to use a direct solver. In such cases, iterative solvers become necessary. Thanks to a tuning of γ in the ALU method or of α in the algebraic transformation, iterative solvers can reduce the gap with the ALU.MUMPS strategy. Encouraged

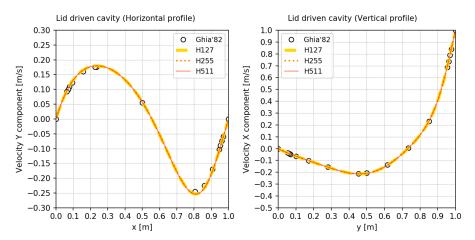


Fig. 9 2D Lid driven cavity flow at Re = 100: Vertical component of the velocity field along an horizontal profile through the cavity center (left) and horizontal component of the velocity along a vertical profile through the cavity center (right).

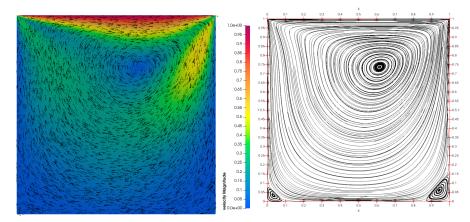


Fig. 10 2D Lid driven cavity flow at Re = 100: Magnitude of the velocity field (left) and streamlines of the velocity field (right).

Table 16 Lid driven cavity problem at $Re = 100$ solved by the ALU(γ)
method: Evolution of the performance with respect to the mesh refinement for
the two best tested strategies.

Mesh	strategy		# itera	tions	Time [s]	
		Picard	ALU	Total inner	setup	solve
H127	ALU(100).MUMPS ALU(0.2).HMG	11 12	31 80	$\begin{array}{c} 31 \\ 1069 \end{array}$	$7.1 \\ 13.4$	$9.8 \\ 41.5$
H255	ALU(100).MUMPS ALU(0.2).HMG	11 12	$31 \\ 87$	$\begin{array}{c} 31 \\ 1058 \end{array}$	$28.0 \\ 54.1$	$38.6 \\ 161.1$
H511	ALU(100).MUMPS ALU(0.2).HMG	11 12	32 88	32 1 028	$132.8 \\ 218.4$	$176.9 \\ 641.4$

30

Mesh	strategy	# i	terations	Time [s]	
		Picard	Total inner	setup	solve
H127	Notay(1).MUMPS Notay(10).GAMG	11 11	11 533	$\begin{array}{c} 19.7\\ 9.0\end{array}$	$20.1 \\ 28.4$
H255	Notay(1).MUMPS Notay(10).GAMG	11 11	$\begin{array}{c} 11 \\ 569 \end{array}$	$93.8 \\ 34.6$	95.6 117.3
H511	Notay(1).MUMPS Notay(10).GAMG	11 11	11 689	$481.9 \\ 142.2$	$490.6 \\ 563.1$

Table 17 Lid driven cavity problem at Re = 100 solved by the Notay(α) algebraic transformation: Evolution of the performance with respect to the mesh refinement for the two best tested strategies.

by the found best choice, we also compared its numerical performance for varying Reynolds numbers. While ALU.MUMPS is robust with respect to the Reynolds number, this is not the case for the iterative strategies ALU.HMG or Notay.GAMG. A tuning specific to the Reynods number can improve the performance of ALU.HMG at low Reynolds. However, if a high Reynolds number is imposed, our tested methods show poor results or fail. Therefore, our recommendation is valid with low Reynolds numbers and further investigation is required for the case of higher ones.

6 Conclusion and discussion

This paper addresses the steady-state incompressible Stokes and Navier–Stokes problems using CDO face-based discretization schemes. We compare three different numerical approaches, the Augmented Lagrangian–Uzawa (ALU), the Golub–Kahan Bidiagonalization (GKB) and the algebraic transformation, to solve saddle-point problems. Overall, the algebraic transformation exhibits the benefits of the monolithic scheme by the absence of additional outer iterations and shows better numerical performance when using iterative solvers.

In the Stokes case, the GKB strategy is found to be robust when using any AMG preconditioned GMRES solver, whereas the algebraic transformation requires Hypre BoomerAMG for robustness. For the Navier–Stokes problem, the ALU method with a direct solver shows reasonable performances on a broad range of Reynolds numbers. This is not the case for ALU with iterative solvers due to the ill-conditioned inner systems imposed by the augmented terms. However, the algebraic transformation remains well-suited to GAMG preconditioned Krylov solvers for low Reynolds number problems.

Statements and Declarations

Conflict of interest. The authors declare that they have no conflict of interest.

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Author contributions.

- Y.Jang: Writing- Original draft preparation, Writing Review & Editing, Investigation, Software, Visualization.
- J.Bonelle: Writing Review & Editing, Methodology, Software, Resources.
- C.Kruse: Writing Review & Editing, Methodology.
- F.Hülsemann: Writing Review & Editing, Conceptualization.
- U.Rüde: Writing Review & Editing, Supervision.

Data availability. The data that support the findings of this study are available from the corresponding author, Y. Jang, upon reasonable request. The CFD tool code_saturne is freely available and we refer to Appendix and an official document in https://www.code-saturne.org.

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Appendix A code_saturne options

Here, we provide parameter options to use a saddle-point problems solvers and K-cycle AMG in the context of code_saturne. Note that for the K-cycle, we rely on the

Option	Value	Result
	"gkb" or "gkb_saturne"	GKB method via code_saturne in house
CS NSKEY SLES STRATEGY	"gkb_petsc"	GKB method via PETSc
CS_NSKEY_SLES_SIRATEGY	"uzawa_al"	ALU method
	"notay"	Algebraic transformation
CS_NSKEY_GD_SCALE_COEF	"\	Set an augmented parameter $\gamma \ge 0$
CS_NSKEY_NL_ALGO_RTOL	"tol"	Set the relative tolerance for
		Picard's iterations to $tol > 0$
CS_NSKEY_MAX_NL_ALGO_ITER	"max_it"	Set the maximal number of
		Picard's iterations to $max_{it} \in \mathbb{N}$
CS_NSKEY_IL_ALGO_RTOL	"tol"	Set the relative tolerance for
		outer iterations to $tol > 0$
CS_NSKEY_MAX_IL_ALGO_ITER	"max_it"	Set the maximal number of
		outer iterations to $\mathtt{max_it} \in \mathbb{N}$
CC FOREY COLVED FAMILY	"cs"	Use a linear solver from code_saturne in house
CS_EQKEY_SOLVER_FAMILY	"petsc"	Use a linear solver from PETSc library
	"fcg"	FCG solver
CS_EQKEY_ITSOL	"fgmres"	FGMRES solver
	"mumps"	Direct solver (through MUMPS or PETSc)
CS_EQKEY_ITSOL_EPS	"tol"	Set the tolerance for
		iterative solvers to $tol > 0$
CS_EQKEY_ITSOL_MAX_ITER	"max_it"	Set the maximal number of
		iterative solvers to $\mathtt{max_it} \in \mathbb{N}$

implementation in code_saturne. Then,

cs_multigrid_set_solver_options() and cs_multigrid_set_coarsening_options()

allow us to set multigrid parameters for smoothing and coarsening, respectively. For more details of the usage of $code_saturne$, we refer to the official document²

Appendix B PETSc options

In this appendix, we present the example sets of PETSc options that we used for our numerical tests.

B.1 Hypre BoomerAMG

Option	Value	Result	
-pc_type	"hypre"	HypreBoomerAMG	
-pc_hypre_type	"boomeramg"	as a preconditioner	
-pc_hypre_boomeramg_coarsen_type	"Falgout" ("HMIS" or "PMIS")	The classical coarsening (with lower complexity)	
-pc_hypre_boomeramg_agg_nl	"0"	No aggressive coarsening	
-pc_hypre_boomeramg_relax_type_down -pc_hypre_boomeramg_relax_type_up	"symmetric-SOR/Jacobi"	SGS smoothing	

 $^{2} https://www.code-saturne.org/documentation/7.1/doxygen/src/index.html$

For more details of PETSc options and the full list of options, we refer to the main Hypre website³ and MOOSE website⁴.

B.2 GAMG

Option	Value	Result
-pc_type	"gamg"	GAMG
-pc_gamg_type	"agg"	as a preconditioner
-pc_gamg_agg_nsmooths	"1"	One smoothing step to use
		with smoothed aggregation
-pc_gamg_sym_graph	"true"(if a matrix is non-symmetric)	Symmetrize the graph
-mg_levels_ksp_type	"richardson"	SGS smoothing
-mg_levels_pc_type	"sor"	565 smoothing

Note that GAMG is basically built with the smoothed aggregation scheme so that it is more applicable for symmetric systems. In practice, by turning on -pc_gamg_sym_graph, it is required to symmetrize the graph for non-symmetric systems. Moreover, to consider unsmoothed aggregation coarsening, we should set -pc_gamg_agg_nsmooths=0.

With GAMG, it is possible to use Krylov smoothers. For example, we can employ the preconditioned GMRES for each smoothing step as following.

Option	Value	Result
-mg_levels_ksp_type	"richardson"	
-mg_levels_pc_type	"bjacobi"	
-mg_coarse_pc_type	"tfs"	Krylov smoothing by GMRES
-mg_levels_pc_bjacobi_blocks	"1"	
-mg_levels_sub_ksp_type	"gmres"	
-mg_levels_sub_ksp_max_it	"10"	Maximum 10 GMRES smoothing steps
-mg_levels_sub_ksp_rtol	"1e-3"	Set a tolerance of the GMRES smoother to 10^{-3}
-mg_levels_sub_pc_type	"sor"	SGS preconditioning for the GMRES smoother
-mg_levels_sub_pc_type	"ilu"	ILU(0) preconditioning for the GMRES smoother
-mg_levels_sub_pc_factor_levels	"0"	TLO(0) preconditioning for the GMRES smoother

B.3 HMG

Using HMG, we can combine the coarsening schemes from Hypre BoomerAMG and the smoothing operators from GAMG in PETSc as following.

Option	Value	Result
-pc_type -hmg_inner_pc_type	"hmg" "hypre"	HMG preconditioner with the classical coarsening

To set up smoothing process, it follows the same way of GAMG.

 $[\]label{eq:approx} ^{3} https://hypre.readthedocs.io/en/latest/index.html \\ ^{4} https://mooseframework.inl.gov/releases/moose/v1.0.0/application_development/hypre.html$

⁴⁰