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Pierre Tize Mha, Prashant Dhondapure, Mohammad Jahazi, Amèvi Tongne, Olivier Pantalé. Interpolation and Extrapolation Performance Measurement of Analytical and ANN-Based Flow Laws for Hot Deformation Behavior of Medium Carbon Steel. *Metals*, 2023, 13 (3: Special Issue Hot Deformation of Metal and Alloys), pp.633. 10.3390/met13030633 . hal-04062000

HAL Id: hal-04062000

<https://hal.science/hal-04062000>

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Interpolation and Extrapolation Performance Measurement of Analytical and ANN-Based Flow Laws for Hot Deformation Behavior of Medium Carbon Steel

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Abstract: In the present work, a critical analysis of the most commonly used analytical models and recently introduced ANN-based models is performed to evaluate their predictive accuracy within and outside the experimental interval used to generate them. The high temperature deformation behavior of a medium carbon steel was studied over a wide range of strains, strain rates, and temperatures using hot compression tests on a Gleeble-3800. The experimental flow curves were modeled using Johnson-Cook, Modified-Zerilli-Armstrong, Hansel-Spittel, Arrhenius, and PTM models, as well as an ANN model. The mean absolute relative error and root-mean-square error values were used to quantify the predictive accuracy of the models analyzed. The results indicated that the Johnson-Cook and Modified-Zerilli-Armstrong models had a significant error, while the Hansel-Spittel, PTM, and Arrhenius models were able to predict the behavior of this alloy. The ANN model showed excellent agreement between the predicted and experimental flow curves, with an error of less than 0.62%. To validate the performance, the ability to interpolate and extrapolate the experimental data was also tested. The Hansel-Spittel, PTM, and Arrhenius models showed good interpolation and extrapolation capabilities. However, the ANN model was the most powerful of all the models.

Keywords: Artificial Neural Network; Constitutive flow Law; Analytical flow law; Interpolation; Extrapolation; Gleeble

1. Introduction

Large size forged blocks made of medium carbon high-strength steels are extensively used in the automotive industry as dies for the production of bumpers and dashboards through the plastic injection process. The manufacturing process of the large blocks starts with ingot casting, followed by open die forging, and a quench and temper heat treatment process to achieve the desired mechanical properties [1–3]. In recent years, in order to respond to the market demand, larger size forgings have had to be produced. In parallel, more stringent conditions related to chemical homogeneity, hardness, grain size and mechanical properties from the surface to the core of the forged block have been required. Of the three manufacturing steps (casting, forging and heat treatment), forging is where the most important microstructural changes take place, and which greatly influences the final properties that can be achieved [4–6]. The open die forging process is fundamentally a hot compression process during which work strengthening effects, such as hardening (WH) and precipitation, take place concomitantly with softening phenomena such as recovery and recrystallization under static and/or dynamic thermomechanical loads [7,8]. It has also been reported that phase transformation can occur during deformation. The extent and intensity of the above phenomena strongly depend on three macroscopic quantities, namely, the strain ϵ , the strain rate $\dot{\epsilon}$ and the temperature T [9–12].

Considering the large size of blocks, a purely experimental approach, based on trial and error, can not be used by industry, and therefore, reliable predictive tools, such a

Citation: P. Tize Mha, P. Dhondapure, M. Jahazi, A. Tongne, O. Pantalé Interpolation and Extrapolation Performance Measurement of Analytical and ANN-Based Flow Laws for Hot Deformation Behavior of Medium Carbon Steel. *Metals* **2023**, *13*, 0. <https://doi.org/>

Received:

Revised:

Accepted:

Published:

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Finite Element analysis (FEA) codes, have been developed, and are commercially available. However, the prediction reliability of such analyses is a function of the accuracy of the material constitutive model, which describes the mutual interactions between the strain, the strain rate, and the temperature during deformation. As mentioned above, precipitation and phase changes are considered negligible as deformation takes place in the austenite at temperatures above the dissolution temperature of most carbides. Hence, most constitutive models depend on macroscopic parameters which influence the hardening and softening of the material. A large number of phenomenological, semi-empirical, or physical models [13–16] have been developed. Among these, the Johnson-Cook (JC) [17], Hansel-Spittel (HS) [18], and Zerilli-Armstrong (ZA) [19] models are the best known and most widely available in FEA codes. Despite their simplicity, each of them suffers from certain shortcomings: as reported by Jia et al. [20], the JC model suffers from a lack of non-coupling between the strain, the strain rate and the temperature, while the HS model is better adapted for higher strain rate conditions [18]. To circumvent these shortcomings, several modified model forms have been developed [18,21–26]. However, as reported by many authors [27–29], even after adjusting the constants, the high temperature flow behavior, particularly when dynamic recrystallization takes place, cannot be accurately predicted, and none of the models is able to accurately predict the flow behavior outside the experimental testing interval that was used to determine the model constants. Due to its more physics-based formulation, the ZA model and its modified form, MZA [30–34], and the Arrhenius-type hyperbolic sine constitutive models are preferred to the JC and HS models both for the prediction of the hot deformation behavior and microstructure analysis of the material [35–37]. The Arrhenius formulation has been revised repeatedly to achieve a more accurate determination of the activation energy for high temperature deformation [38,39]. To overcome the strong dependency of the models on specific alloy types, Tize Mha et al. [40] recently proposed a constitutive model, PTM, whose formulation is independent of the alloy type. This formulation is based on the MZA model and the polynomial functions of undefined order that are used during the identification. However, in the PTM model, the high order of the polynomial function (up-to 10 in some cases) can affect its accuracy.

The Artificial Neural Network (ANN) is an approach used to predict the flow stress behavior of materials without requiring a mathematical formulation of the flow law. It is therefore not necessary to postulate a mathematical expression to identify the parameters of the model. Since the behavior of materials is highly nonlinear at high temperatures, and depends on many factors which are also nonlinear, the evaluation of the flow stress behavior by an analytical model whose parameters are identified by a classical regression method is limited. Faced with these limitations, ANN models are of major interest because they are particularly suited to deal with complex and nonlinear relationships. Consequently, ANNs have been successfully applied to predict the flow stress behavior of materials under hot working conditions [41–44]. Although ANN models can predict well the material flow behavior, there is a problem with its implementation in finite element software, as reported by Pantalé et al. [16,45]. In fact, the implementation of a constitutive model in an FEA code requires the derivatives of the model with respect to strain, strain rate and temperature.

Although progress has been made in improving constitutive models to better predict the flow stress behavior of materials, problem still persist with the models' efficiency. Indeed, a model is considered appropriate for predicting the material's behavior if its predictions and experimental results correlate well. It is questionable whether a model that correctly describes the behavior of a material in a defined experimental window can be used to accurately predict its behavior for conditions different from those for which it was identified. In other words, the question is whether interpolation and extrapolation techniques can be used to extend the applicability of a constitutive equation for different processing windows.

The present work aims to address the preceding question using the recently introduced ANN model. To this end, initially, hot compression tests, simulating the open die forging process, are carried out and the flow curves generated are modeled using the above-

mentioned constitutive equations and an ANN model developed in this work. Then, the interpolation and extrapolation capabilities of each model are evaluated. The results are validated based on experimental work carried out herein and on data obtained from the literature.

2. Materials and Experiments

2.1. Experimental procedure

The material used in this study consists of a medium carbon steel whose chemical composition is given in Table 1. Cylindrical samples were machined with an initial diameter

Table 1. Chemical composition of medium carbon steel, Fe = balance

Element	C	Mn	Mo	Si	Ni	Cr	Cu
Wt %	0.30	0.89	0.52	0.34	0.68	1.86	0.17

of $d = 10$ mm and a height of $h_0 = 15$ mm. Hot compression tests were performed on a Gleeble-3800 thermomechanical simulator (see Figure 1), for 5 temperature levels, namely, 1050°C, 1100°C, 1150°C, 1200°C and 1250°C, with the 6 strain rates of 0.001 s⁻¹, 0.01 s⁻¹, 0.1 s⁻¹, 1 s⁻¹, 2.0 s⁻¹ and 5 s⁻¹. Thin tantalum sheets were used as lubricating material at



Figure 1. The Gleeble-3800 thermomechanical simulator system used for this study

the contact surface of the anvils and samples to minimize friction during testing. Figure 2 shows the inside of the Gleeble thermomechanical simulator with the specimen in place. We used 3 thermocouples soldered to the specimen to record the temperature history during the test and to ensure that the specimen was at the correct temperature prior to testing.

As shown in Figure 3, the samples were heated to a temperature of 1260°C with a heating rate of 2°C/s and held at this temperature for 5 min to eliminate thermal gradients. They were then cooled down with a rate of 1°C/s to the test temperature and then held at constant temperature for 1 min before deformation. During the compression phase, the temperature of the specimen is kept constant by the thermal control system of the machine. After compression, the specimen is quickly quenched to freeze its microstructure for later analysis. Figure 3 also shows the aspects of the specimens before and after the compression test: h_0 and r_0 are the height and radius before compression and h , r_m and r_t are the height, large radius and small radius of the specimen after compression, respectively.

The stress-strain curves are automatically exported from the Gleeble thermomechanical simulator system as the true stress and true strain according to the L-gauge, where the formula to get the curves is given by $\sigma = F/A$ and $\epsilon = \ln(1 + \Delta h/h_0)$ or C-gauge, having

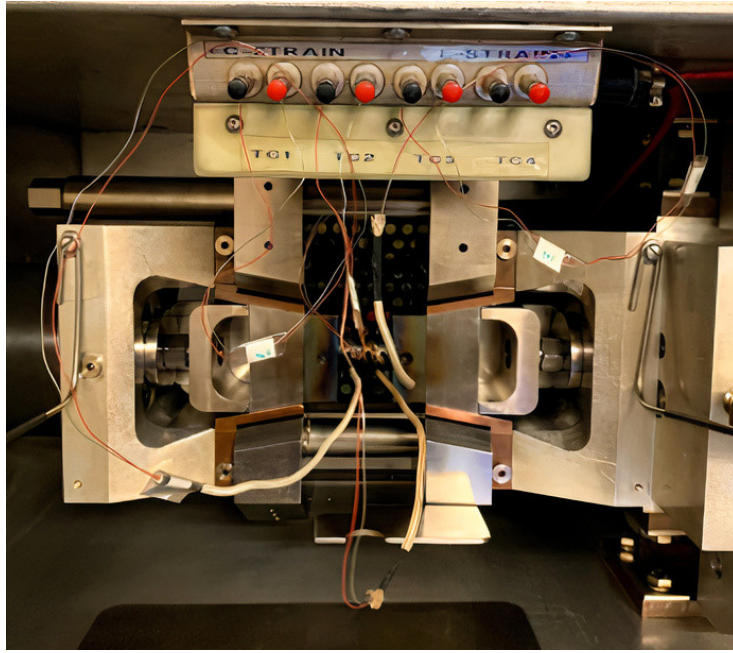


Figure 2. The inside of the Gleeble-3800 thermomechanical simulator with the specimen in place

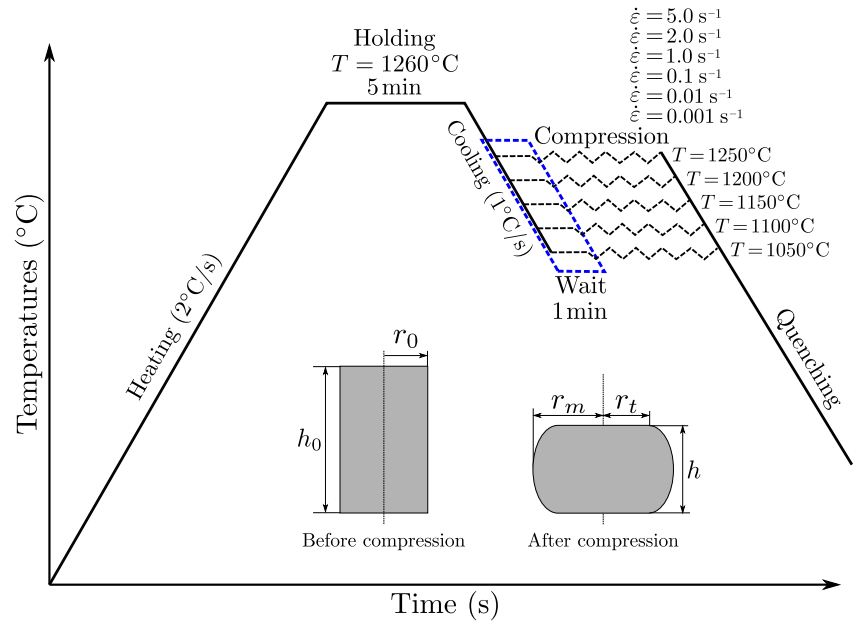


Figure 3. Schematic diagram of the experimental process on the Gleeble-3800 thermomechanical simulator

the following formulas: $\sigma = 4F/\pi(d + \Delta d)^2$ and $\epsilon = 2 \ln(d/(d + \Delta d))$ with $d = 2r_0$, and where F is the force as measured by the Gleeble load cell. As raw data contains noises, the `savgol_filter` method from the `scipy` library is used to remove noise and provide smoother data. To allow further use of the data in numerical simulations, elastic parts are removed.

2.2. Compression tests results

The set of flow stress σ^y versus strain ϵ curves obtained from compression tests performed on the Gleeble-3800 simulator for each test condition (6 strain rates and 5 temperatures) is presented in Figure 4. All data curves contain 700 equidistant strain values up-to $\epsilon = 0.7$. Therefore, there are 6 strain rates and 5 temperatures, and the database consists of 21,000 data points. For the identification of the parameters of all the analytical

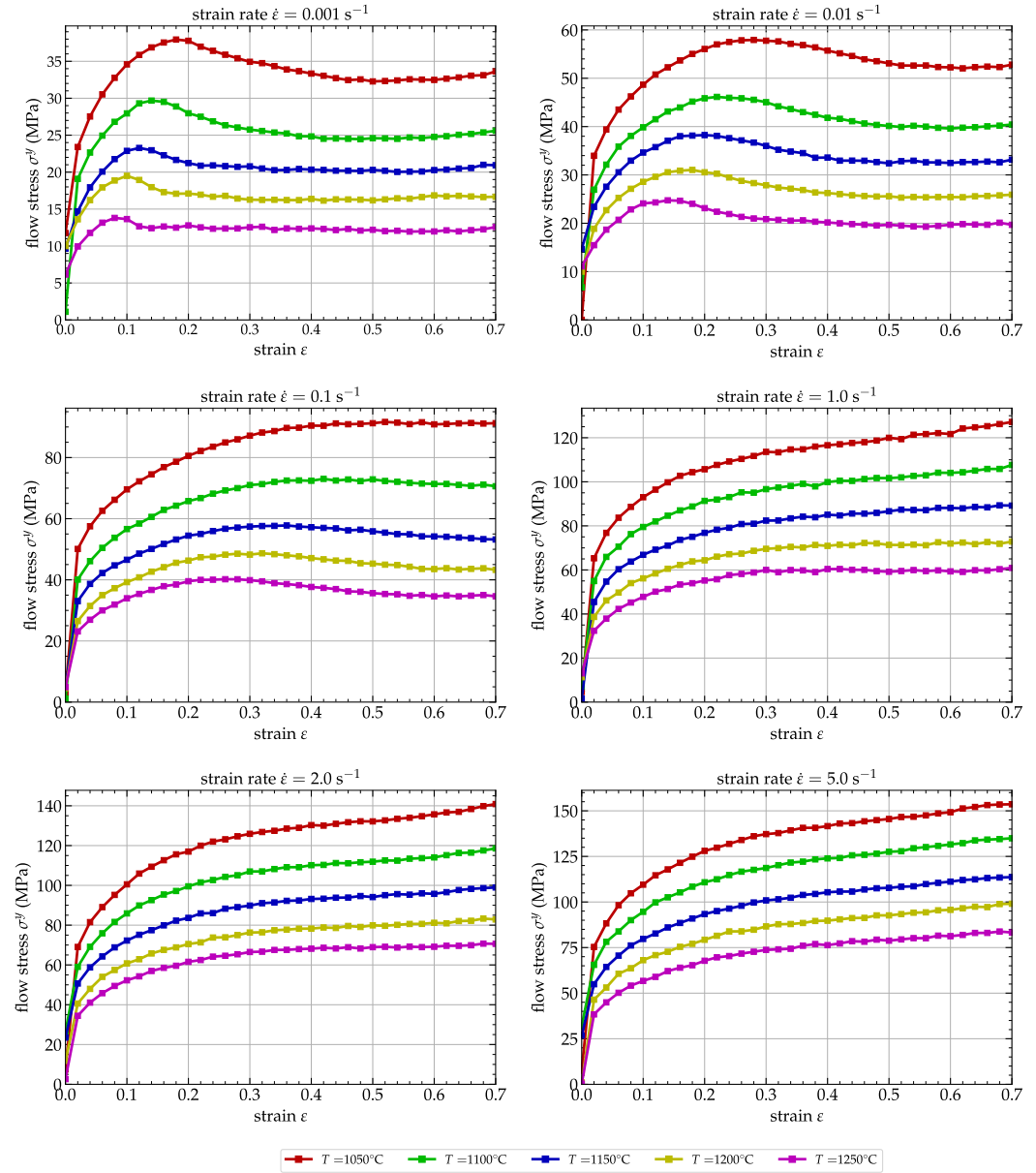
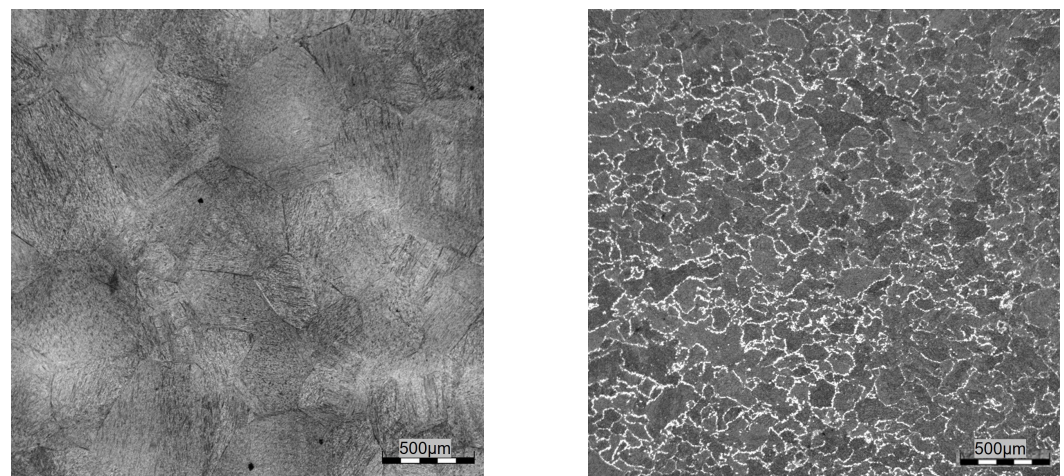


Figure 4. Stress-strain curves of medium carbon alloy extracted from the Gleeble device at various temperatures T and strain rates $\dot{\epsilon}$

models presented in this article, we restricted the database to 35 strain values between 0.02 and 0.7, with a step of 0.02 (this is illustrated in Figure 4, where the data used correspond to the dots on the graphs). The overall behavior of these curves shows that the flow stress σ^y increases with increasing strain rate $\dot{\epsilon}$, but decreases with increasing temperature T . It should be noted that the strain also influences the flow stress. Indeed, for the lowest strain rates $\dot{\epsilon}$, the flow stress σ^y increases with the strain ϵ until a value of about $\epsilon = 0.2$ to 0.3 , and then decreases to maintain a more or less constant value until the end of the test. For the highest strain rates (1 s^{-1} , 2 s^{-1} , and 5 s^{-1}), the flow stress increases throughout the test. The slight increase in stress at low strain rates, when the strain is large, has been reported to be due to friction between the sample and the anvil during the test [46]. The frictional effect is also visible when testing at low strain rates, as the effect of lubrication decreases over time, and friction thus increases. The increase of stress observed at the beginning of the deformation and up-to 0.1 is due to work hardening (WH). After 0.1 and up-to 0.2, the flow stress shows a continuous reduction with increasing stress until a peak or an inflection of the work-hardening rate. This behavior indicates that thermal softening becomes more

and more predominant until it exceeds WH. At this step, the stress curve shows three different patterns with the increasing strain: (i) gradual decrease to a steady state with DRV/DRX softening. This is the case for all deformation temperatures and strain rates between 0.001 and 0.1 s^{-1} , except for those at 1050°C and 1100°C ; (ii) higher stress levels without significant softening and work-hardening at 1050°C and 1100°C and a 0.1 s^{-1} strain rate; and (iii) continuous increase with significant work hardening (all deformation temperatures and strain rate of 1 s^{-1}). Therefore, it can be concluded that the softening due to DRX, characterized by a flow curve with a single peak followed by a steady-state flow, takes place at high temperatures and low strain rates. In contrast, at higher strain rates and lower temperatures, the higher work hardening rate slows down the rate of softening due to DRX, and therefore, both the peak stress and the onset of steady-state flow are shifted to higher strain levels. In fact, the drop observed in stress is because of DRX occurrence at all temperatures and strain rates of $0.001\text{--}5 \text{ s}^{-1}$.

Figure 5(a) shows the microstructure obtained when the sample is held at 1260°C for 5 minutes and rapidly cooled to room temperature. From this image, large grains can be observed. On the other hand, Figure 5(b) shows the microstructure after compression at 1150°C with a 0.001 s^{-1} strain rate. To obtain this, the sample is heated up to 1260°C , then held for 5 minutes and cooled down to 1150°C to be held at this temperature for 1 minute before compression. After compression, the sample is rapidly cooled to room temperature to preserve the microstructure. From this image, full recrystallization can be observed and is in correlation with Figure 4, where after the peak, a steady stress is observed that describes the end of the recrystallization.



(a) Before hot compression

(b) After hot compression

Figure 5. Optical micrographs of a medium carbon steel (a) before and (b) after hot compression at 1150°C with a 0.001 s^{-1} strain rate.

The material is in the single phase austenite state at all of the test temperatures examined. The data presented in Figure 4 will be utilized in the following section to determine the parameters of the 6 flow laws proposed and to determine which model most closely aligns with experimental results.

3. Identification of Constitutive Flow Laws Parameters

3.1. The Johnson-Cook model

The JC model, as mentioned above, is one of the most widely used analytical models because it can be applied to many materials under different conditions of strain ϵ , strain rate $\dot{\epsilon}$, and temperature T . However, the formulation of this model does not take into account the simultaneous effect of strain, strain rate and temperature. Indeed, it is formulated by describing the effect of each physical parameter (ϵ , $\dot{\epsilon}$ and T) separately as a factor in the mathematical expression of the model, hence its inability to describe the phenomenon

of softening induced by temperature. The equation that describes this model is given as follows [17]:

$$\sigma^y = (A + B\varepsilon^{pn}) \left[1 + C \ln\left(\frac{\dot{\varepsilon}}{\dot{\varepsilon}_0}\right) \right] \left[1 - \left(\frac{T - T_0}{T_m - T_0} \right)^m \right], \quad (1)$$

where σ^y is the flow stress, ε^p is the plastic strain, A is the initial elastic limit of the material, B is the strain hardening coefficient, n is the strain hardening exponent, C and m are the material constants that describe the strain rate hardening coefficient and the thermal softening coefficient, respectively. The other values are reference values: $\dot{\varepsilon}_0$ is the reference strain rate, T_m and thus T_0 are the melting temperature (1460°C in our case) and the reference temperature, respectively. For the determination of the parameters of the analytical models, the reference values for strain rate and temperature are $\dot{\varepsilon}_0 = 0.001 \text{ s}^{-1}$ and $T_0 = 1050^\circ\text{C}$. In our approach, the reference strain rate and reference temperature for identifying the JC model are the lowest values used during the test, however, sometimes these values do not always give the best results for the model.

The procedure used to determine the parameters of the Johnson-Cook law is in accordance with the one proposed by Zeng et al. [11]. This method allows sequentially obtaining the 5 parameters in the order A , B , n , C and m . Thus, according to the experimental data, the initial elastic limit of the material at the reference strain rate $\dot{\varepsilon}_0$ and the reference temperature T_0 is $A = 13.5143 \text{ MPa}$. For the determination of the constants B and n , from the results of the compression test at T_0 and $\dot{\varepsilon}_0$, these two constants can then be determined by considering only the first term $(A + B\varepsilon^{pn})$ in Equation (1). Thus, here, $B = 21.816 \text{ MPa}$ and $n = 0.0746$. Once the parameters A , B and n are known, the determination of C , considering only all the curves at $T = T_0$ then gives $C = 0.3404$. Finally, the last parameter m is identified from the curves at $\dot{\varepsilon} = \dot{\varepsilon}_0$ and from knowledge of the parameters A , B , C , n and gives $m = 0.7057$. All parameters of the Johnson-Cook model are reported in Table 2.

Table 2. Parameter values of the Johnson-Cook flow law for a medium carbon steel

A (MPa)	B (MPa)	n	C	m
13.5143	21.816	0.0746	0.3404	0.7057

The values predicted by the Johnson-Cook flow law (solid line) and the experimental values (dots) are compared in Figure 6. The JC model is unable to describe the evolution of the average flow stress for all strain and strain rate levels. The experimental flow stresses show a growth and then a decrease with strain, especially for low strain rates, while the JC model, by its formulation, only allows an increasing evolution of the flow stress σ^y as a function of the strain independently of the strain rate value. The discrepancy between the predicted and experimental values is large for low strains and sometimes acceptable for high strains. As expected, the mathematical formulation of the Johnson-Cook flow law is unable to represent the stress drop at low strain rates, with the JC model increasing only monotonically with the strain. Since most of the parameters are identified at low strain rates (A , B and n), this results in a very poor fit of the model to the experimental data.

The accuracy and predictive ability of the models are usually assessed through certain coefficients such as the Mean Absolute Relative Error (E_{MAR}) defined by Equation (2):

$$E_{\text{MAR}}(\%) = \frac{1}{N} \sum_{i=1}^N \left| \frac{\sigma_i^p - \sigma_i^e}{\sigma_i^e} \right| \times 100, \quad (2)$$

and the Root Mean Square Error (E_{RMS}) defined by Equation (3):

$$E_{\text{RMS}}(\text{MPa}) = \sqrt{\frac{1}{N} \sum_{i=1}^N (\sigma_i^p - \sigma_i^e)^2}, \quad (3)$$

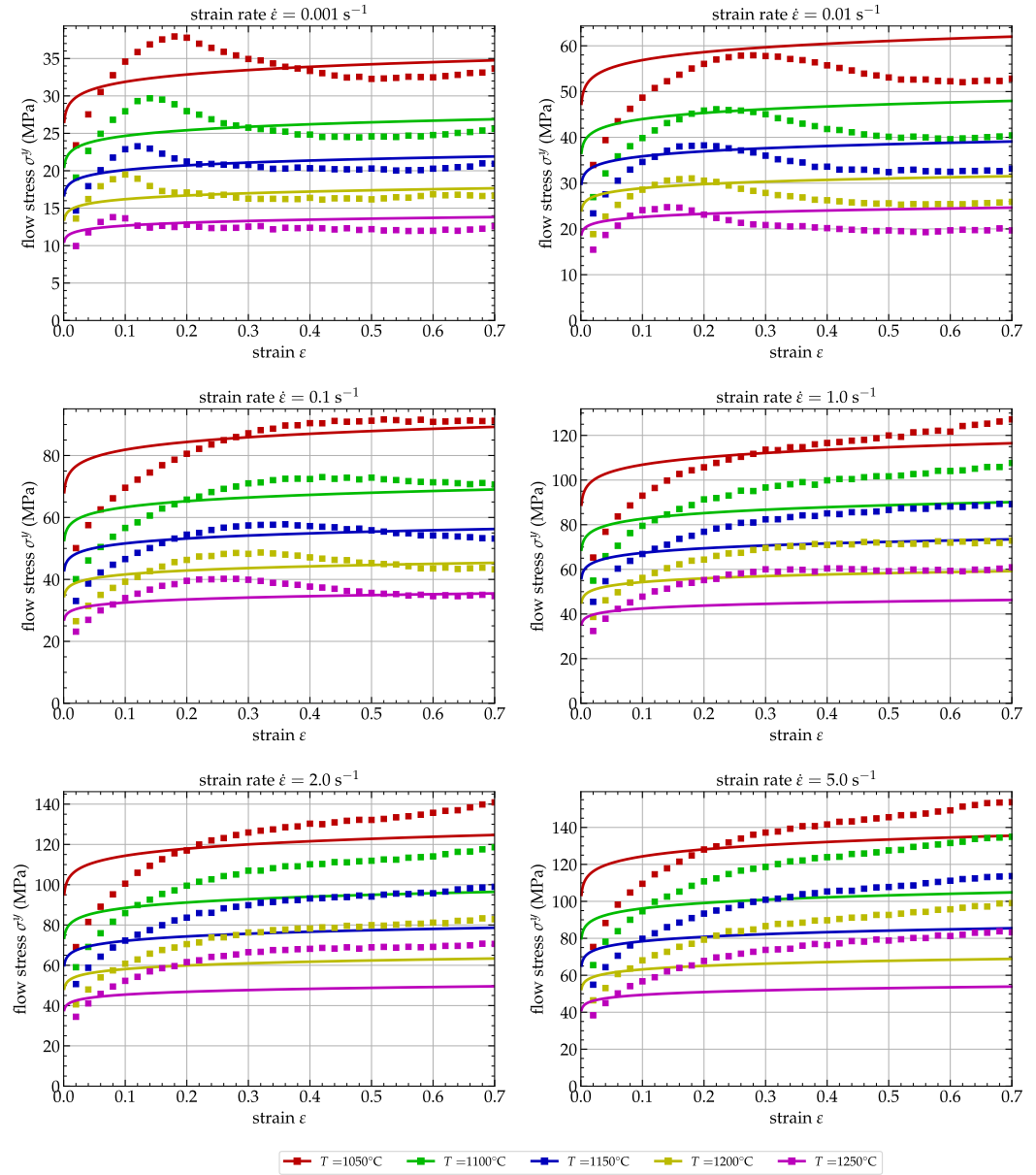


Figure 6. Comparison between the experimental (dots) and predicted (lines) flow stresses σ^y by the Johnson-Cook model

where σ_i^e is the experimental value, σ_i^p is the value predicted using the given model of the flow stress σ^y and N is the total number of data points used to compute those coefficients (in our case $N = 21\,000$). For the JC model, $E_{\text{MAR}} = 14.05\%$ and $E_{\text{RMS}} = 12.00\text{ MPa}$. As reported by Phaniraj [47], the correlation coefficient (R) is not always an accurate measure to evaluate the reliability of the constitutive law especially in the case of a highly non-linear functions because it only shows the correlation of the model with respect to the data and not its accuracy. A good (high) value of R (close to 1) does not necessarily mean a good prediction of the model, but simply establishes a good linearity correlation between the experiment and the prediction, we therefore avoided its use in our analysis.

3.2. The Modified Zerilli-Armstrong model

The MZA model, which is the modified form of the ZA model, like the JC model presented earlier, is one of the most widely used models implemented in many FEA codes such as the Abaqus software. The difference between the MZA model and JC model is related to the consideration of the three physical parameters to describe the reality observed

from experiments. In the JC model, the parameters are considered separately, while in MZA, they are considered simultaneously, and for that formulation, MZA is preferred to the JC model [48]. However, the original form of the ZA model has some limitations due to the fact that it is considered as a two terms function (thermal and athermal functions), and to improve the formulation, Samantaray et al. [49] proposed a modified form given by the following equation:

$$\sigma^y = (C_1 + C_2 \varepsilon^n) \exp \left[-(C_3 + C_4 \varepsilon^p)(T - T_0) + (C_5 + C_6(T - T_0)) \ln \left(\frac{\dot{\varepsilon}}{\dot{\varepsilon}_0} \right) \right], \quad (4)$$

where the 7 coefficients C_i and n are the parameters of the model to be identified for a given material. To get the parameters of the MZA model, we apply the method proposed by [49], and the parameters are summarized in Table 3, while their predictions are plotted in Figure 7. It can be seen from this figure that both the MZA and JC models are not able to faithfully

Table 3. Parameter constants of Modified Zerilli-Armstrong model

C_1 (MPa)	C_2 (MPa)	C_3	C_4	C_5	C_6	n
13.5143	21.2591	4.7902×10^{-3}	1.4895×10^{-4}	0.1389	1.495×10^{-4}	0.0621

reproduce the experimental data, especially at low strain rates, but they are slightly better at higher strain rates. The deviation between the predicted values and the experimental values is large because this model has a problem correctly describing the softening in its formulation. For the MZA model, $E_{MAR} = 21.20$ % and $E_{RMS} = 19.57$ MPa, showing an overall worse performance than the JC model, as reported in [40]. In addition, it can be observed that at high strain rates the MZA is not good either, even if there is no softening effect, and this can be explained by the fact that this model is formulated for quasi-static phenomena.

3.3. The Hansel-Spittel model

The Hansel-Spittel model [50] is one of the least known models in terms of integration in FEA codes, although its parameters can more easily be determined than those of the JC or MZA models. Programming a simple identification script is sufficient to identify its parameters for a given material. The equation of the HS model is given by the following relation:

$$\sigma^y = A e^{m_1 T} \varepsilon^{m_2} \dot{\varepsilon}^{m_3} e^{(m_4/\varepsilon)} (1 + \varepsilon)^{m_5 T} e^{m_6 \varepsilon} \dot{\varepsilon}^{m_7 T} T^{m_8}, \quad (5)$$

where again, σ^y is the flow stress, ε is the strain, $\dot{\varepsilon}$ is the strain rate and T is the temperature, as proposed earlier. The coefficients A and m_i are the 9 parameters of the model to be identified. However, this model has some shortcomings, notably related to the fact that its accuracy varies according to the number of parameters taken into account during the identification. For its identification, several authors restrict its expression to a reduced number of only 5 or 6 m_i parameters, by forcing a zero value for the other parameters [51–53].

In the present study, the best results were obtained by taking the model defined by only the first 7 m_i terms of the equation (5), so that $m_8 = 0$. From the experimental data obtained during the compression tests, an identification procedure based on the use of the LMFIT optimizer [54] allowed to compute the parameters reported in Table 4.

A comparison of the values predicted by the HS model and the experimental values is presented in Figure 8, where the dots represent the experimental values and the solid lines are the values predicted by the Hansel-Spittel flow law. For the HS model, $E_{MAR} = 7.75$ % and $E_{RMS} = 3.80$ MPa. It appears that both this model and the previous ones do not adequately predict the experimental one, and the difference is relatively significant for all strain rates below 1 s^{-1} . This shows that this model is not appropriate for the characterization of

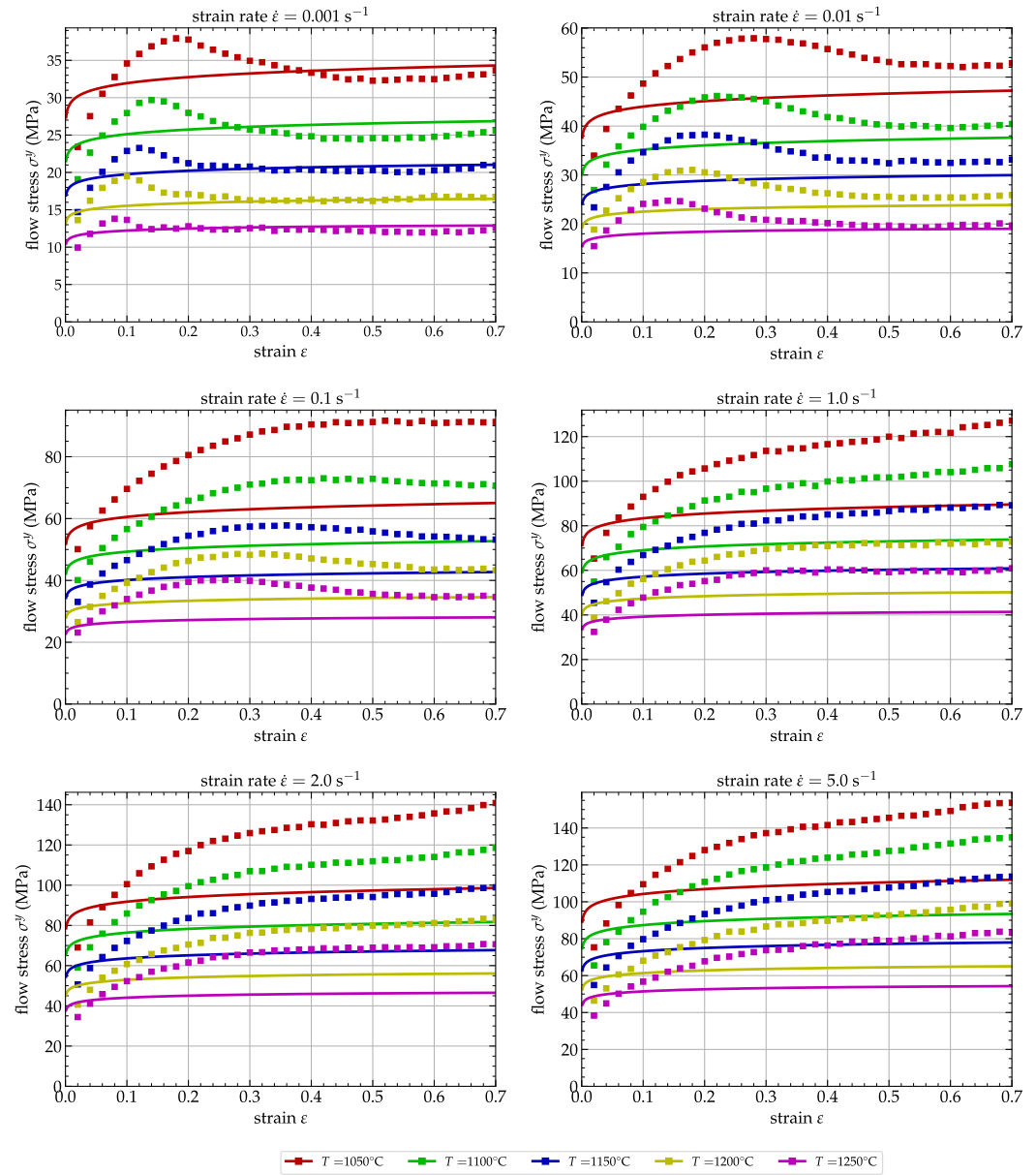


Figure 7. Comparison between the experimental (dots) and predicted (lines) flow stresses σ^y by the Modified Zerilli-Armstrong model

Table 4. Parameter values of the Hansel-Spittel flow law for the medium carbon steel

A	m_1	m_2	m_3	m_4
5.954×10^3	-3.3576×10^{-3}	0.2641	-0.0868	2.2688×10^{-4}
	m_5	m_6	m_7	m_8
	-4.2163×10^{-4}	-0.0561	2.264×10^{-4}	0

this alloy, particularly because of the strong non-linear behavior observed for low strain rate values. The DRX phenomenon cannot be reproduced by this type of model.

3.4. The Arrhenius model

The Arrhenius type model [55] is one of the most used models in the framework of material forming, especially when it comes to studying the material microstructure. The model takes into account the physical phenomena describing the behavior of the material

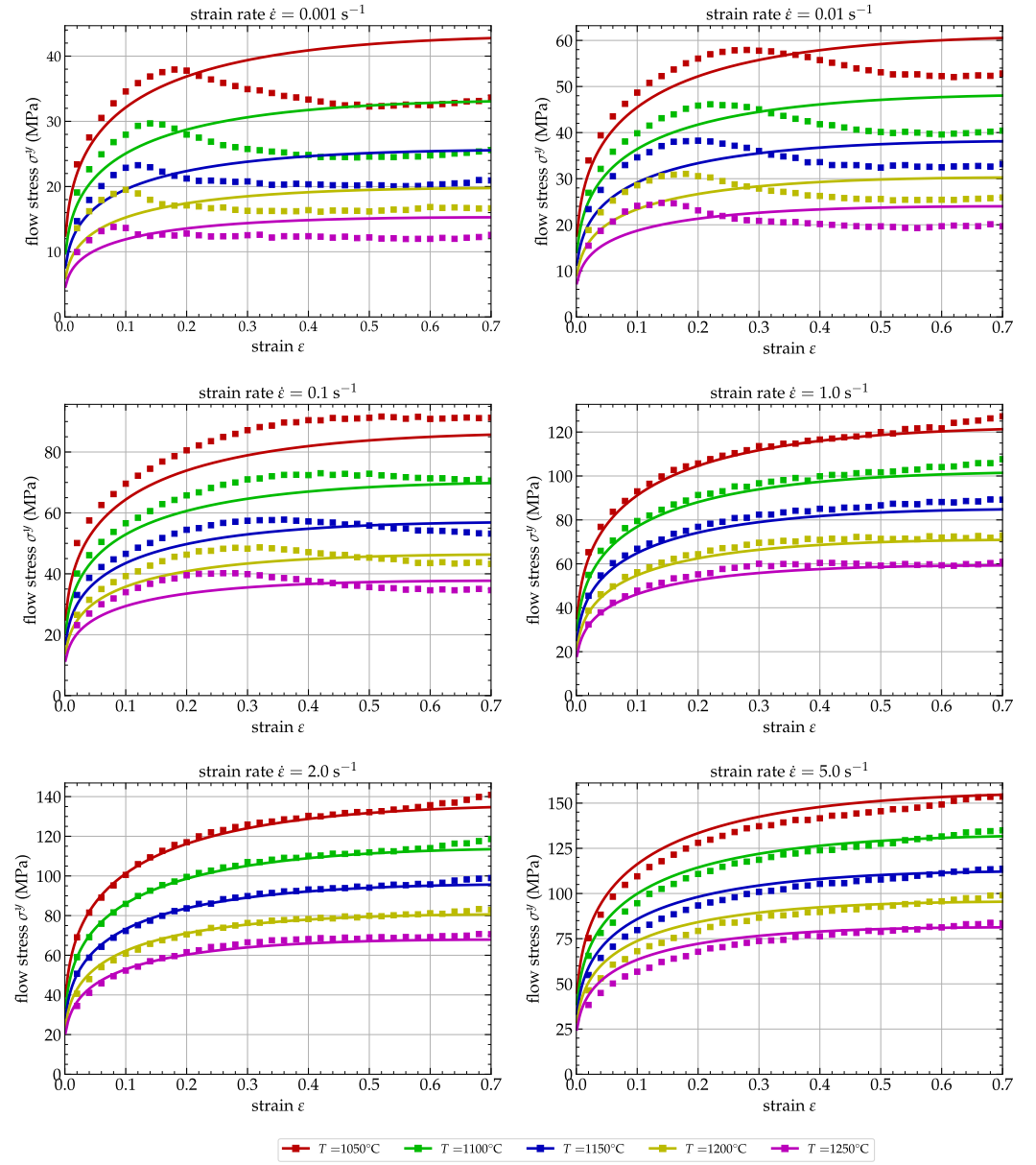


Figure 8. Comparison between the experimental (dots) and predicted (lines) flow stresses σ^y by the Hansel-Spittel model

in the formulation of the relationships between the stress σ^y , the strain ϵ , the strain rate $\dot{\epsilon}$ and the temperature T expressed as power law, exponential law and hyperbolic sine. This makes it easier to describe the softening phenomenon observed in the material due to increasing temperature. The following equations describe the Arrhenius model:

$$\dot{\epsilon} = \begin{cases} A_1 \sigma^{y^{n_1}} \exp\left(-\frac{Q}{RT}\right) & \alpha \sigma^y < 0.8 \\ A_2 \exp(\beta \sigma^y) \exp\left(-\frac{Q}{RT}\right) & \alpha \sigma^y > 1.2 \\ A_3 [\sinh(\alpha \sigma^y)]^{n_2} \exp\left(-\frac{Q}{RT}\right) & \text{for all } \sigma^y \end{cases} \quad (6)$$

with:

$$Z = \dot{\epsilon} \exp\left(\frac{Q}{RT}\right), \quad (7)$$

where Z is the Zener-Hollomon parameter [56], $\dot{\epsilon}$ is the strain rate (s^{-1}), Q is the apparent activation energy (J mol^{-1}), R is the universal gas constant ($8.314 \text{ J mol}^{-1}\text{K}^{-1}$), T is the absolute temperature (K), σ^y is the flow stress (MPa) for a given strain, strain rate and temperature and $A_1, A_2, A_3, n_1, n_2, \alpha$ and $\beta = \alpha n_1$ are dependent on the material. The corresponding values are independent of the temperature and are obtained from the stress-strain curves at different strain rates and temperatures by the regression method. Combining Equations (6) and (7) allows to express the flow stress σ^y as a function of the Z parameter:

$$\sigma^y = \frac{1}{\alpha} \ln \left\{ \left(\frac{Z}{A} \right)^{1/n} + \left[1 + \left(\frac{Z}{A} \right)^{2/n} \right]^{1/2} \right\} \quad (8)$$

To obtain the constitutive equation, all the parameters A , Q , α and n must be determined for a given material. The strain has a significant non-linear influence on the behavior of the material by the strain hardening and softening mechanisms at high values of deformation. A strain dependent factor must therefore be taken into account in the Arrhenius model, which leads to the definition of the modified Arrhenius model for which the A , Q , α and n parameters are expressed as a function of the strain ϵ by means of polynomial functions of degree m (varying from 1 to 9) of the form:

$$A(\epsilon) = \exp \left[\ln A_0 + \ln A_1 \epsilon + \ln A_2 \epsilon^2 + \ln A_3 \epsilon^3 + \dots + \ln A_m \epsilon^m \right] \quad (9)$$

$$Q(\epsilon) = Q_0 + Q_1 \epsilon + Q_2 \epsilon^2 + Q_3 \epsilon^3 + \dots + Q_m \epsilon^m \quad (10)$$

$$\alpha(\epsilon) = \alpha_0 + \alpha_1 \epsilon + \alpha_2 \epsilon^2 + \alpha_3 \epsilon^3 + \dots + \alpha_m \epsilon^m \quad (11)$$

$$n(\epsilon) = n_0 + n_1 \epsilon + n_2 \epsilon^2 + n_3 \epsilon^3 + \dots + n_m \epsilon^m \quad (12)$$

The determination of the order m of the polynomials defining Equations (9-12) depends on the ability of the model to represent the non-linear dependence of the stress on strain and its generalization. The values $\ln A_i$, α_i , n_i , and Q_i ($i = 0, 1, 2, 3, \dots, m$) are the coefficients of the polynomials used to determine using a regression method. Setting $m = 9$ gives the best results, and the corresponding parameters are reported in Table 5. This modified form of

Table 5. Parameter values of the Arrhenius flow law for the medium carbon steel

α_i	$Q_i (\times 10^{-6})$	$\ln A_i$	n_i
$\alpha_0 = 0.0407$	$Q_0 = 0.467$	$\ln A_0 = 35.8092$	$n_0 = 4.8217$
$\alpha_1 = -0.5167$	$Q_1 = -0.6517$	$\ln A_1 = -58.822$	$n_1 = 3.2814$
$\alpha_2 = 6.3912$	$Q_2 = 7.6084$	$\ln A_2 = 740.3303$	$n_2 = 71.5963$
$\alpha_3 = -47.3364$	$Q_3 = -48.016$	$\ln A_3 = -5.0493 \times 10^3$	$n_3 = -1.9562 \times 10^3$
$\alpha_4 = 220.0014$	$Q_4 = 66.795$	$\ln A_4 = 1.1305 \times 10^4$	$n_4 = 1.4461 \times 10^4$
$\alpha_5 = -654.4553$	$Q_5 = 468.8898$	$\ln A_5 = 2.022 \times 10^4$	$n_5 = -5.431 \times 10^4$
$\alpha_6 = 1.2421 \times 10^3$	$Q_6 = -2.3032 \times 10^3$	$\ln A_6 = -1.5387 \times 10^5$	$n_6 = 1.1761 \times 10^5$
$\alpha_7 = -1.4523 \times 10^3$	$Q_7 = 4.3707 \times 10^3$	$\ln A_7 = 3.1798 \times 10^5$	$n_7 = -1.4882 \times 10^5$
$\alpha_8 = 952.0619$	$Q_8 = -3.9394 \times 10^3$	$\ln A_8 = -2.9725 \times 10^5$	$n_8 = 1.0239 \times 10^5$
$\alpha_9 = -267.4994$	$Q_9 = 1.397 \times 10^3$	$\ln A_9 = 1.0759 \times 10^5$	$n_9 = -2.9621 \times 10^4$

the Arrhenius behavior law allows an accurate and reliable prediction over a wide range of temperatures and strain rates. Equation (13) is finally used to compute the flow stress σ^y from the strain ϵ , the strain rate $\dot{\epsilon}$ and the temperature T :

$$\sigma^y = \frac{1}{\alpha(\epsilon)} \ln \left\{ \left(\frac{\dot{\epsilon} \exp \left(\frac{Q(\epsilon)}{RT} \right)}{A(\epsilon)} \right)^{\frac{1}{n(\epsilon)}} + \left[1 + \left(\frac{\dot{\epsilon} \exp \left(\frac{Q(\epsilon)}{RT} \right)}{A(\epsilon)} \right)^{\frac{2}{n(\epsilon)}} \right]^{\frac{1}{2}} \right\} \quad (13)$$

Figure 9 shows a comparison of the values predicted by the Arrhenius model and the experimental values. The difference between the experimental and predicted values is small. However for the strain rate $\dot{\epsilon} = 0.01 \text{ s}^{-1}$ and for the two low temperature values, the AR model is unable to predict the softening. For the AR model, $E_{\text{MAR}} = 3.56 \%$ and $E_{\text{RMS}} = 2.18 \text{ MPa}$.

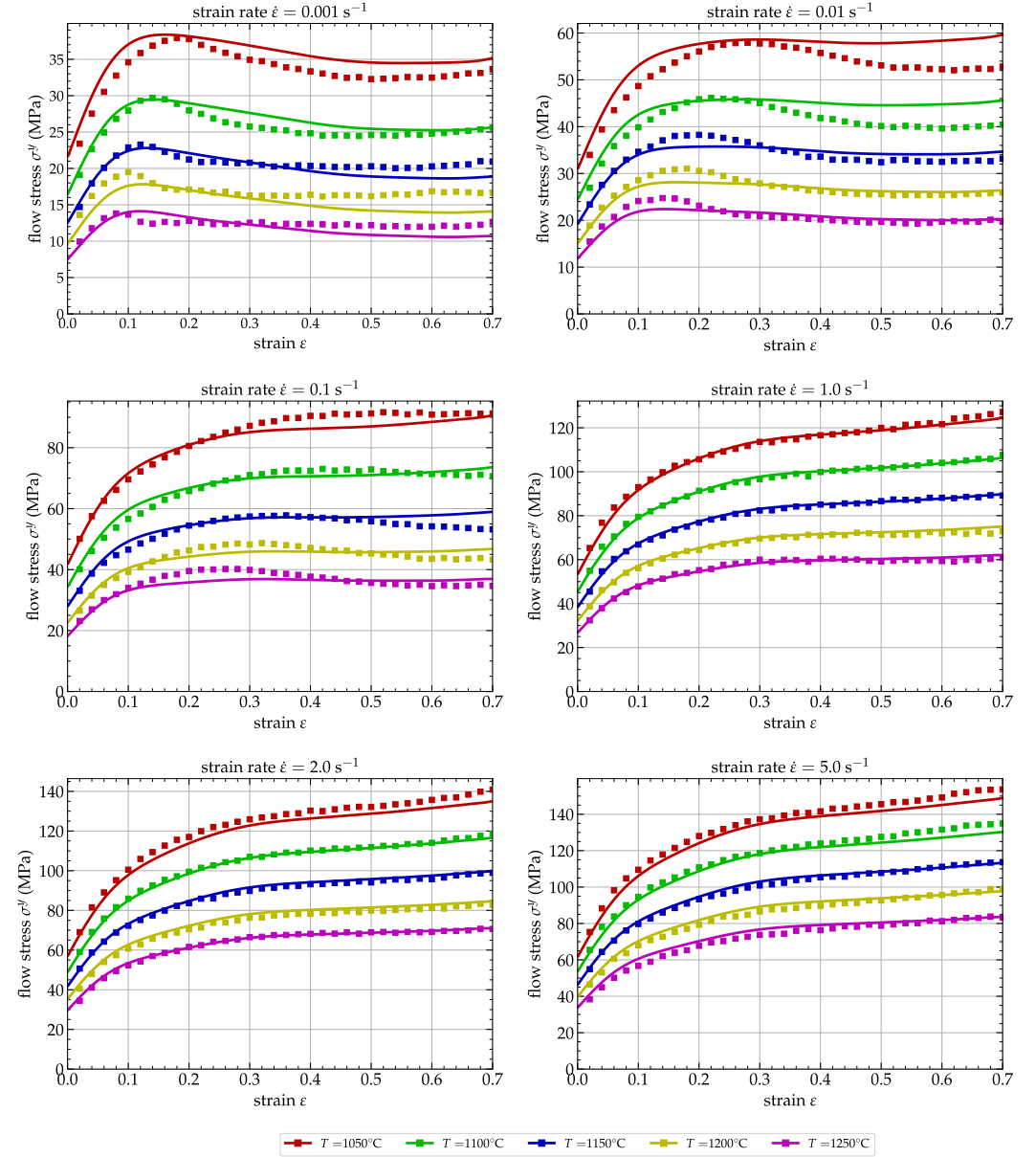


Figure 9. Comparison between the experimental (dots) and predicted (lines) flow stresses σ^y by the Arrhenius model

3.5. The PTM model

The PTM model [40] is a generalized formulation of the MZA model presented in Section 3.2. When establishing its formulation, the main shortcomings of the MZA model were taken into account, in order to render the PTM model flexible for any type of material studied because it removes the need for a limited number of parameters as in the MZA model. Its construction is based on the use of polynomial functions, as is the case in the AR model. Thus, the physical parameters' dependent intrinsic functions of the PTM model, allow adjusting the model according to the degree selected for each of the 4 constituent

polynomials, which provides a good fit for each function. The equation describing the PTM model is therefore given by :

$$\sigma^y = \left(\sum_{i=0}^q A_i \varepsilon^{p_i} \right) \exp \left[\left(\sum_{j=0}^r B_j \varepsilon^{p_j} \right) (T - T_0) + \left(\sum_{k=0}^s \left(\sum_{l=0}^t C_k^l \varepsilon^{p_l} \right) (T - T_0)^k \right) \ln \left(\frac{\dot{\varepsilon}}{\dot{\varepsilon}_0} \right) \right] \quad (14)$$

where A_i, B_j, C_k^l are the parameters (Table 6) of the model to be identified using the procedure proposed in [40]. Quantities q, r, s and t define the degree of the polynomials used to describe the behavior of the material. The larger these quantities, the greater the number of parameters needing to be identified for the PTM model. The determination of the parameters of this model is done thanks to the LMFIT Python library [54] (for more details about this model we refer the reader to our previous work [40]). Thus, all the parameters of this model are calculated with $q = 5, r = 5, s = 1$ and $t = 5$.

Figure 10 presents a comparison of values predicted by the PTM model and experimental values. The PTM model is suitable for describing the flow behavior of medium carbon steel, but for the two strain rates $\dot{\varepsilon} = 0.01 \text{ s}^{-1}$ and $\dot{\varepsilon} = 0.1 \text{ s}^{-1}$, the prediction is not very good. The deviation between the predicted values and the experimental values for other strain rates is relatively low. For the PTM model, $E_{\text{MAR}} = 4.79 \%$ and $E_{\text{RMS}} = 4.59 \text{ MPa}$, which is an overall worse performance than the AR model.

Table 6. Parameter values of the PTM flow law for the P20 steel

A_i	B_i	C_0^i	C_1^i
$A_0 = 16.8529$	$B_0 = -3.5418 \times 10^{-3}$	$C_0^0 = 0.1608$	$C_1^0 = -1.9037 \times 10^{-5}$
$A_1 = 340.6451$	$B_1 = -0.0132$	$C_0^1 = -0.6202$	$C_1^1 = 2.67 \times 10^{-3}$
$A_2 = -1.9594 \times 10^3$	$B_2 = -4.4888 \times 10^{-3}$	$C_0^2 = 4.9516$	$C_1^2 = -3.5788 \times 10^{-3}$
$A_3 = 4.836 \times 10^3$	$B_3 = 0.2218$	$C_0^3 = -13.1694$	$C_1^3 = -0.0222$
$A_4 = -5.5176 \times 10^3$	$B_4 = -0.4988$	$C_0^4 = 15.25$	$C_1^4 = 0.0609$
$A_5 = 2.4058 \times 10^3$	$B_5 = 0.3211$	$C_0^5 = -6.587$	$C_1^5 = -0.0413$

3.6. The Artificial Neural Network model

Because of their predictive capacity and adaptability, Artificial Neural Networks (ANNs) are increasingly widely used today in many scientific fields. Their operation is based on a training process during which the principle of minimizing the error between the model's output and the training data allows the adjustment of the model's parameters, as in any machine training process.

Neural networks generally have two uses: classification and regression. The first is the ability to classify data into different groups, and for example, to distinguish between images of cats and dogs. The second corresponds to the universal approximation capacity of these neural networks, which is of interest to us herein, and thus, to the ability, after training, to predict the flow stress σ^y values according to the input data, akin to what is done by the above identified analytical models. The main difference is that this approximation is not linked to a fixed mathematical formulation (as in JC, MZA, AR, HS and PTM analytical models), but is only dependent on the data used for training, the number of layers, the number of neurons per layer, and the activation functions associated with the neurons of the network. A feed-forward ANN, as used in our application, contains an input layer, an output layer and a number of hidden layers (2 in our case). Each layer of neurons is connected to the one before it and the one after it by weighted connections. Thus, all the neurons of the k layer are connected to all the neurons of the $(k - 1)$ layer as shown in Figure 11.

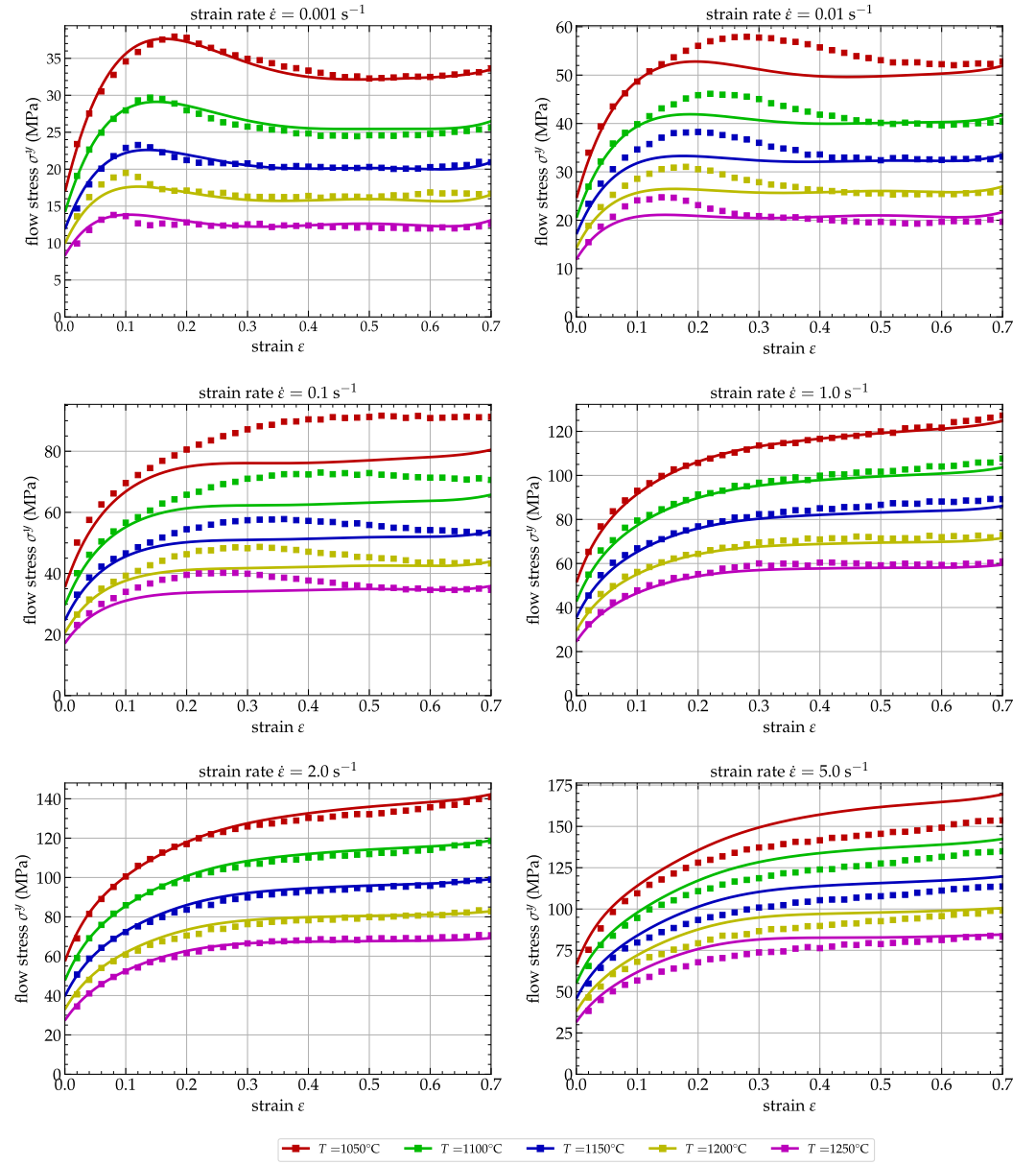


Figure 10. Comparison between the experimental (dots) and predicted (lines) flow stresses σ^y by the PTM model

Any hidden layer k , containing n neurons, takes a weighted sum of the outputs \vec{y} of the immediately preceding layer $(k - 1)$, containing m neurons, given by the following equation:

$$y_i^{(k)} = \sum_{j=1}^m w_{ij}^{(k)} \hat{y}_j^{(k-1)} + b_i^{(k)}, \quad (15)$$

where $y_i^{(k)}$ is the entry of the i^{th} neuron of layer k , $\hat{y}_j^{(k-1)}$ is the output of the j^{th} neuron of layer $(k - 1)$, $w_{ij}^{(k)}$ is the associated weight parameter between the i^{th} neuron of layer k and the j^{th} neuron of layer $(k - 1)$ and $b_i^{(k)}$ is the associated bias of the i^{th} neuron of layer k . Those weights w_{ij} and bias b_i , for each layer, are the training parameters of the ANN which we have to adjust during the training procedure described in Pantalé et al. [16,45]. For the proposed model, we selected the Sigmoid activation function, so that, each neuron

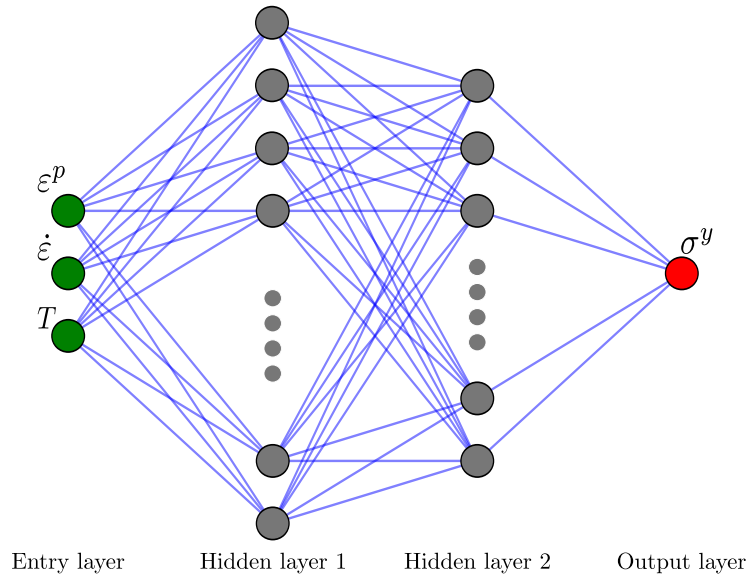


Figure 11. Two hidden layers Artificial Neural Network architecture

in the hidden layer k provides an output value \hat{y} from the input value y of the same neuron defined by Equation (15) according to the following equation:

$$\hat{y} = \frac{1}{1 + e^{-y}} \quad (16)$$

No activation function is used for the output neuron of the ANN as usually done in regression ANN.

After some tests of different types of network architectures, and in accordance with previous works, a network structure with two hidden layers including 15 neurons for the first hidden layer and 7 neurons for the second layer give the best compromise between prediction, training time and model compactness. From a global architecture point of view, the input layer is composed of 3 neurons (ε^p , $\dot{\varepsilon}$, T) and the output layer is composed of a single neuron corresponding to the σ^y flow stress. This architecture leads to a global model with 180 parameters to be identified (60 for the first layer, 112 for the second layer and 8 for the output layer).

The Python program used for training the neural network was developed using the specialized Python library, Tensorflow [57]. The Adaptive Moment Estimation (ADAM) optimizer [58] was used for the training phase. The training data were those from the tests presented in Section 2.2, and were composed of 21,000 quadruplets of (ε^p , $\dot{\varepsilon}$, T , σ^y) values. The training was performed on the basis of 5,000 epochs of the experimental data set. 40 minutes of training on a Dell XPS-13 7390 laptop running Ubuntu 22.04 LTS 64 bits with 16 GB of Ram and an Intel 4-core i7-10510U processor allowed to obtain the converged parameters of the ANN model. Figure 12 shows the evolution of the training error defined by the \log_{10} of the E_{RMS} during the training phase. As can be seen in this figure, after 5,000 epochs, we can consider that we have reached a stationary state of the model training, and that it is useless to continue with the training phase.

Once the training phase is over, the trained model can be used to predict the behavior of the medium carbon alloy as a function of the input data similarly to what is done with analytical models. One can either use the model directly by providing it with new input data, or retrieve the 180 parameters identified during the training and inject them into a mathematical model based on Equations (15) and (16), which can be implemented in any language (e.g., in FORTRAN for use on the Abaqus Explicit FEA code), as proposed in Pantalé et al. [16,45]. For compactness, the parameters of the ANN model and the complete procedure to compute the flow stress σ^y from the input data are provided in the Appendix.

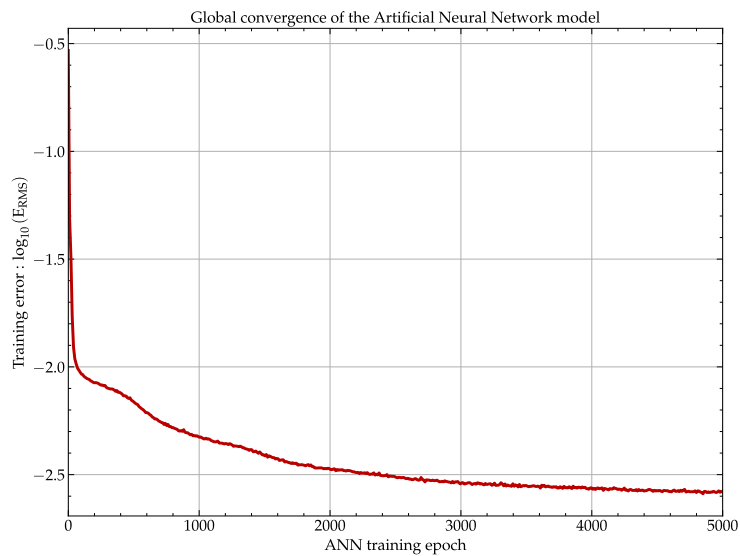


Figure 12. Convergence of the ANN model during the training phase

As for the above considered analytical models, Figure 13 shows a comparison between the flow stresses predicted by the ANN model and the data measured during the hot compression test. The experimental data and the ANN prediction correlate very well over the entire range of data and the predicted data can track the hardening and softening regions of the hot deformed material well. For the ANN model, $E_{MAR} = 0.62\%$ and $E_{RMS} = 0.38\text{ MPa}$ which is excellent. This model can be used to simulate the hot deformation of this type of alloy with much greater reliability to the actual material behavior than the analytical models presented in the above sections.

3.7. Comparison of analytical and ANN models

A summary of the coefficients for evaluating the high-temperature flow stress prediction capability of the medium carbon alloy for all models presented in this work is reported in Table 7. From the Table 7, we can see that the ANN model has a much better predictive

Table 7. Accuracy coefficients for all the analyzed models

Coefficients	JC	MZA	HS	AR	PTM	ANN
$E_{MAR}(\%)$	14.05	21.20	7.75	3.56	4.79	0.62
$E_{RMS}(\text{MPa})$	12.00	19.57	3.80	2.18	4.59	0.38

capacity than all the analytical models presented in the above sections. Globally, the values of E_{MAR} and E_{RMS} are 6 times lower than with the best of the analytical models, i.e., the Arrhenius model, quoted as reference in the context of the hot forming of alloys [59].

The ANN, Arrhenius and PTM models are the only models that take into account softening with the deformation of medium carbon at low strain rate, unlike the other three models, which only present an increase in the flow stress with the strain, irrespective of the strain rate and temperature, hence their poor performance in predicting the behavior of this material, and more particularly, at low strain rates. The parameters reported in Table 7 and the correlations that can be seen in Figures 6 to 13 allow concluding that the ANN model is the most efficient of all the models presented when it comes to describing the behavior of the medium carbon alloy for high-temperature deformation applications.

4. Interpolation and Extrapolation Capability of Models

In order to better compare the performances of the above analysed models (the 5 analytical models and the ANN based model), in this section, we present the ability of each

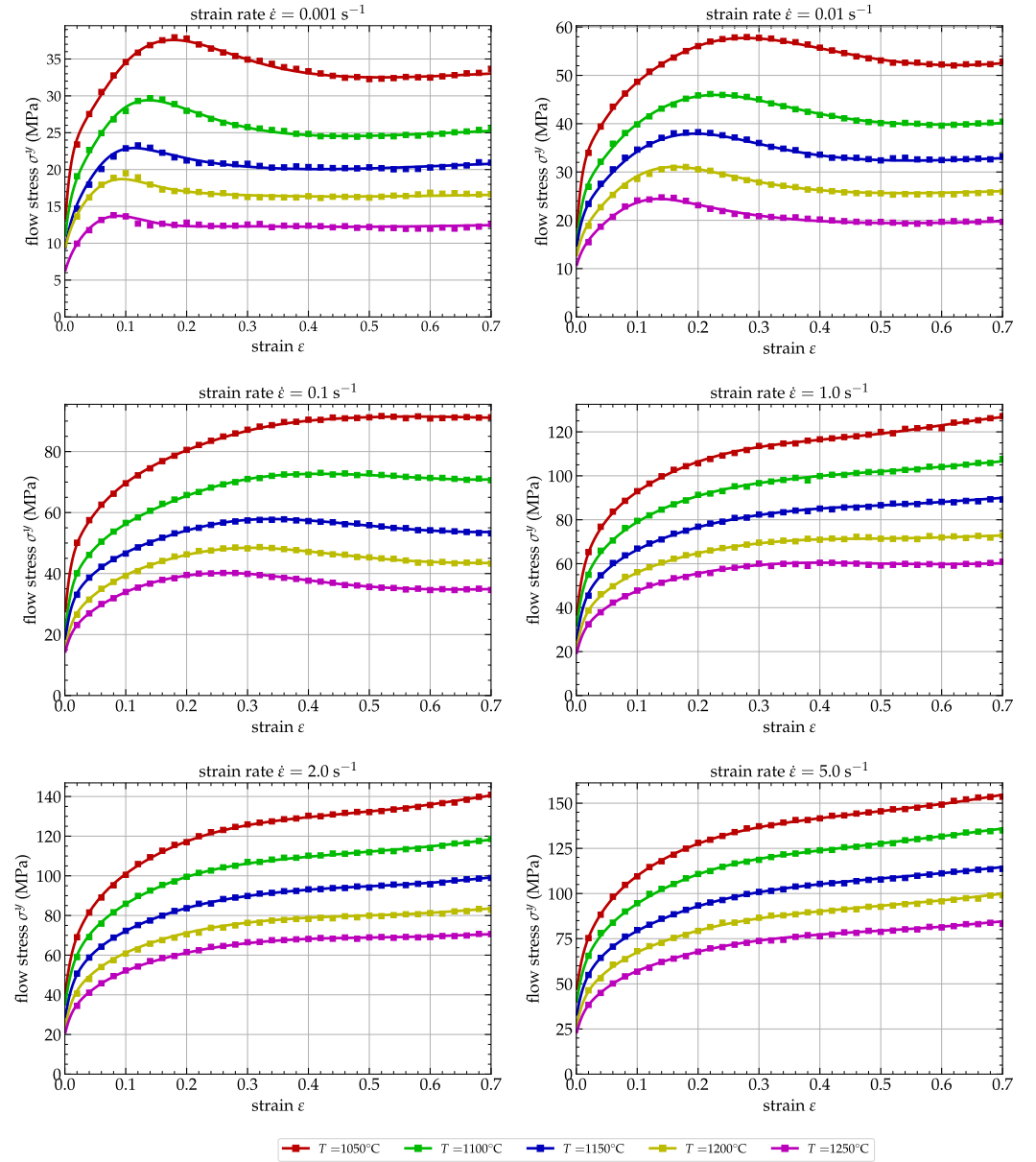


Figure 13. Comparison between the experimental (dots) and predicted (lines) flow stresses σ^y by the 3-15-7-1-sigmoid ANN model

of these models to interpolate and extrapolate the results as a function of the strain rate $\dot{\epsilon}$. The identification of the above analysed models is based on a set of experimental data corresponding to 6 strain rates, 5 temperatures and strains ranging from 0 to 0.7. To test the training capacity and reliability of these models, we propose here to perform the training of the models on only 5 strain rates by voluntarily omitting the strain rate $\dot{\epsilon} = 1 \text{ s}^{-1}$ or $\dot{\epsilon} = 5 \text{ s}^{-1}$.

Thus, one of the omitted strain rates ($\dot{\epsilon} = 1 \text{ s}^{-1}$) is within the range of strain rates for model identification (0.001 s^{-1} to 5 s^{-1}), and therefore, we can test the ability of the models to interpolate the results from the other 5 strain rates and be able to quantify any deviation from experimental values. On the other hand, since the omitted strain rate ($\dot{\epsilon} = 5 \text{ s}^{-1}$) is outside the range of strain rates for the model identification (0.001 s^{-1} to 2 s^{-1}), we will test the ability of the models to extrapolate results and quantify deviations with the experimental values.

4.1. Interpolation validation

For interpolation validation, the chosen omitted strain rate is $\dot{\epsilon} = 1 \text{ s}^{-1}$ and those used for identification (or training for the ANN) are the 5 others, i.e., $\dot{\epsilon} = [0.001, 0.01, 0.1, 2, 5] \text{ s}^{-1}$. All models were re-identified from the same experimental data on a data set corresponding to 5 strain rates and 5 temperatures.

Figure 14 shows a comparison, for strain rate $\dot{\epsilon} = 1 \text{ s}^{-1}$, of the flow stresses σ^y calculated by the models (as a line) and the experimental results (as dots) for the 5 analytical models and the neural network. Table 8 shows the E_{MAR} and E_{RMS} deviations between the

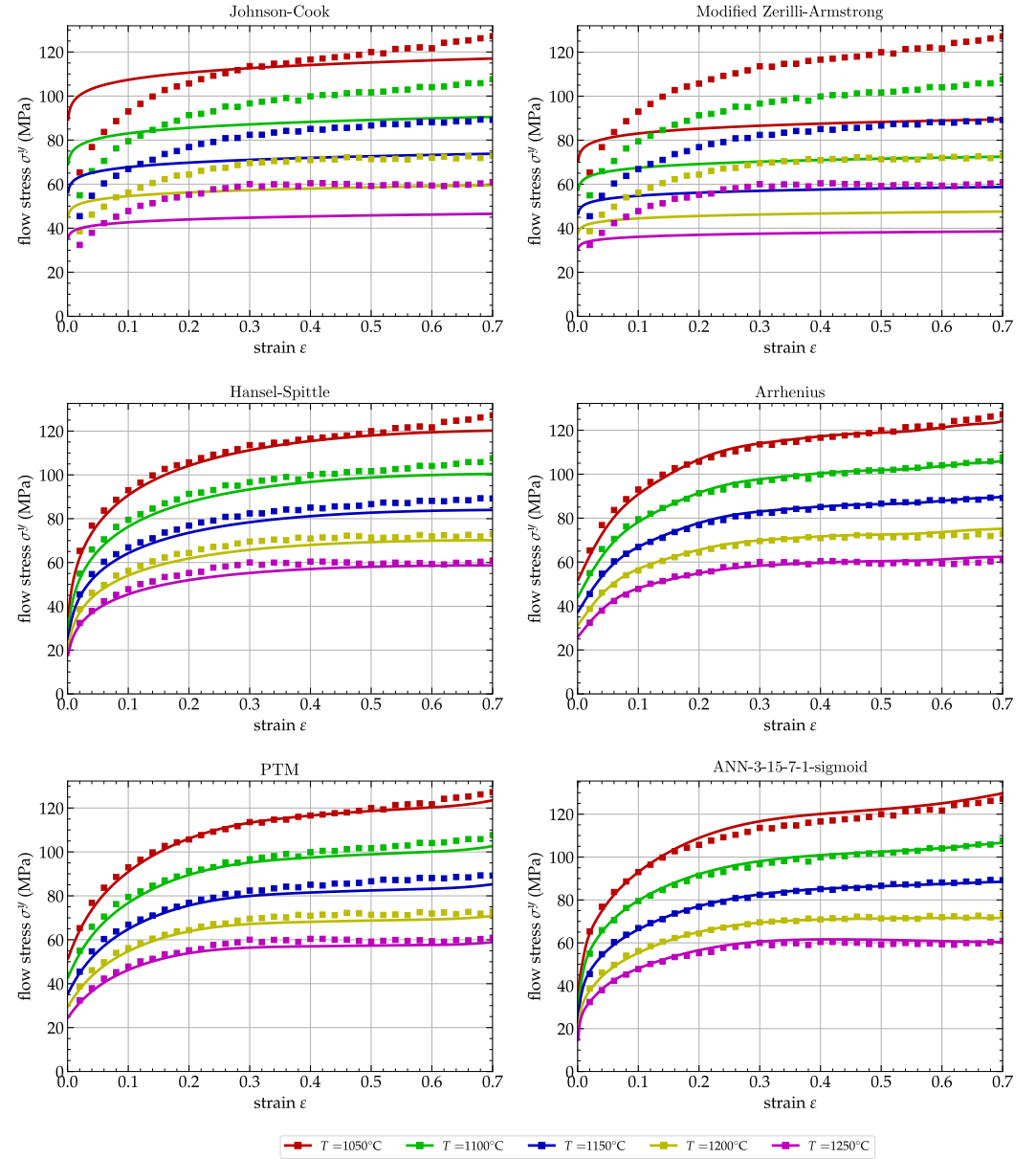


Figure 14. Comparison between the experimental (dots) and predicted (lines) flow stresses σ^y for $\dot{\epsilon} = 1 \text{ s}^{-1}$

different models and the experimental data calculated either for the 5 identified strain rates (lines referred as id. $\dot{\epsilon}$), the 6 strain rates (lines referred as all $\dot{\epsilon}$) or only on the strain rate $\dot{\epsilon} = 1 \text{ s}^{-1}$.

In Figure 14, it can be seen that the first two models (JC and MZA), presented in this study do not have the ability to reproduce the behavior of the material for the strain rate $\dot{\epsilon} = 1 \text{ s}^{-1}$. Overall, the JC model gives better results than the MZA model for $\dot{\epsilon} = 1 \text{ s}^{-1}$,

Table 8. Accuracy coefficients of interpolation for all models with $\dot{\epsilon} = 1 \text{ s}^{-1}$

Strain rate	Coefficients	JC	MZA	HS	AR	PTM	ANN
id. $\dot{\epsilon}$	$E_{\text{MAR}}(\%)$	13.79	20.22	8.57	3.96	5.10	0.70
	$E_{\text{RMS}}(\text{MPa})$	11.83	19.00	3.97	2.29	4.73	0.38
$\dot{\epsilon} = 1 \text{ s}^{-1}$	$E_{\text{MAR}}(\%)$	14.87	27.79	3.74	1.46	3.16	2.47
	$E_{\text{RMS}}(\text{MPa})$	12.02	23.67	3.10	1.46	2.65	2.77
all $\dot{\epsilon}$	$E_{\text{MAR}}(\%)$	14.42	28.90	7.55	3.45	4.90	0.96
	$E_{\text{RMS}}(\text{MPa})$	11.86	19.86	3.84	2.17	4.45	1.18

which is reflected in Table 8 by a lower value of E_{MAR} and E_{RMS} for JC than for MZA. Nevertheless, these values are higher than 10% for the JC model and 20% for the MZA model, which reflects the poor ability of these models to correctly model the behavior of this material. This finding is in agreement with the previous findings of Sections 3.1 and 3.2, which showed the inability of these models to take into account the softening of the behavior visible at low strain rates and low temperatures.

From a general appearance point of view, the other 4 models, HS, AR, PTM and ANN, give globally similar results, with a higher reliability for the AR and ANN models compared to the other two models. Table 8 shows a quantitative comparison on E_{MAR} and E_{RMS} for three different cases: on the 5 identified strain rates, on only the strain rate $\dot{\epsilon} = 1 \text{ s}^{-1}$ and on all 6 strain rates for these 6 models. It appears from this table that while the two models, AR and ANN, give equivalent (and excellent) results concerning the values of E_{MAR} and E_{RMS} for the strain rate $\dot{\epsilon} = 1 \text{ s}^{-1}$, the ANN model gives a globally better result across the entire strain rate spectrum, with values of $E_{\text{MAR}} = 0.96\%$ and $E_{\text{RMS}} = 1.18 \text{ MPa}$ respectively; that is to say, values that are approximately 2 to 3 times lower for the ANN model than for the AR model.

This shows the superior reliability of the ANN model over to the 5 analytical models presented in this study both in terms of the interpolation capability of the model and of the overall behavior.

4.2. Extrapolation validation

For validation of the models' ability to extrapolate, the omitted strain rate chosen is $\dot{\epsilon} = 5 \text{ s}^{-1}$ and those used for identification are $\dot{\epsilon} = [0.001, 0.01, 0.1, 1, 2] \text{ s}^{-1}$. The strain rate omitted in this analysis therefore has the highest value, which restricts the training domain.

In this new identification configuration, Figure 15 shows a comparison, for strain rate $\dot{\epsilon} = 5 \text{ s}^{-1}$, of the flow stresses σ^y computed by the models (as a line) and the experimental results (as dots) for the 5 analytical models and the neural network. Table 9 shows the E_{MAR} and E_{RMS} deviations between the different models and the experimental data calculated either for the 5 identified strain rates (lines referred as id. $\dot{\epsilon}$), the 6 strain rates (lines referred as all $\dot{\epsilon}$) or only on the strain rate $\dot{\epsilon} = 5 \text{ s}^{-1}$.

Table 9. Accuracy coefficients of extrapolation for all models with $\dot{\epsilon} = 5 \text{ s}^{-1}$

Strain rate	Coefficients	JC	MZA	HS	AR	PTM	ANN
id. $\dot{\epsilon}$	$E_{\text{MAR}}(\%)$	13.09	19.09	7.30	4.03	4.34	0.61
	$E_{\text{RMS}}(\text{MPa})$	9.86	16.26	3.36	2.32	3.63	0.32
$\dot{\epsilon} = 5 \text{ s}^{-1}$	$E_{\text{MAR}}(\%)$	20.16	24.27	7.73	3.53	11.46	3.87
	$E_{\text{RMS}}(\text{MPa})$	20.73	25.95	7.83	4.02	12.91	5.84
all $\dot{\epsilon}$	$E_{\text{MAR}}(\%)$	15.12	25.87	7.12	3.86	5.34	1.09
	$E_{\text{RMS}}(\text{MPa})$	12.36	18.24	4.43	2.68	6.23	2.40

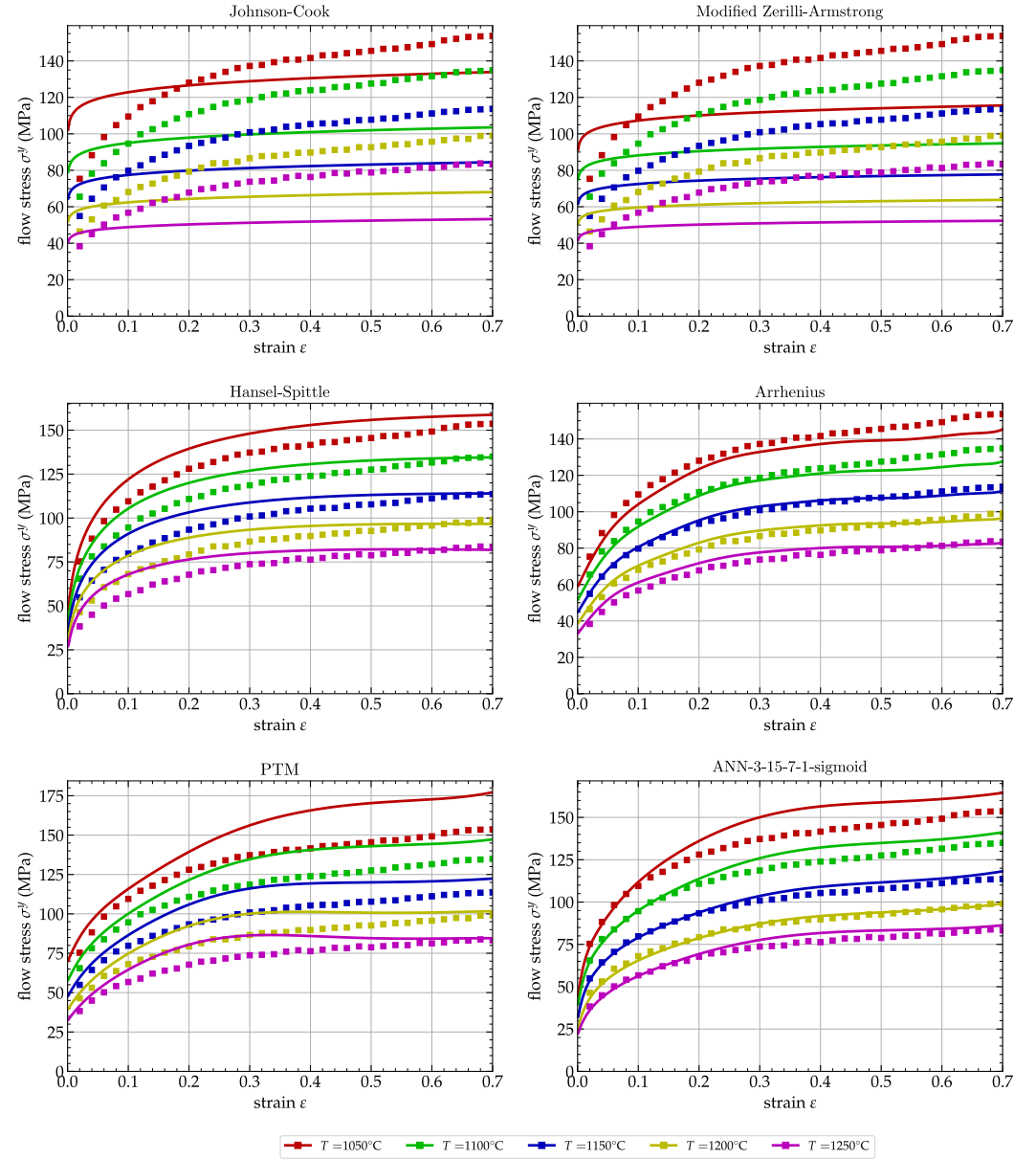


Figure 15. Comparison between the experimental (dots) and predicted (lines) flow stresses σ^f for $\dot{\epsilon} = 5 \text{ s}^{-1}$

As presented in the previous section regarding the interpolation capability of the models, in Figure 15 it can be seen that the first two models, JC and MZA, in this study again do not have the ability to correctly reproduce the material behavior for the strain rate $\dot{\epsilon} = 5 \text{ s}^{-1}$. The JC model performs better than the MZA model for $\dot{\epsilon} = 5 \text{ s}^{-1}$, which is reflected in Table 9 by a lower value of E_{MAR} and E_{RMS} for JC than for MZA. Nevertheless, these values are too large for proper use. Once again, it is the inability of these models to take into account the softening of the behavior at low strain rates and low temperatures that is at the origin of these values.

The other 4 models, HS, AR, PTM and ANN, give better results with higher reliability for the AR and ANN models than for the other two models. Table 9 shows a comparison over all strain rates and over $\dot{\epsilon} = 5 \text{ s}^{-1}$ for these 6 models. The HS model performs worse in extrapolation than the AR and ANN models, while the PTM model is relegated to the last position in this ranking with values of E_{MAR} and E_{RMS} greater than 10% for the strain rate $\dot{\epsilon} = 5 \text{ s}^{-1}$. The two models, AR and ANN, give the best results for the values of E_{MAR} and E_{RMS} for the strain rate $\dot{\epsilon} = 5 \text{ s}^{-1}$. This time, the AR model performs a little

better on the $\dot{\epsilon} = 5 \text{ s}^{-1}$ strain rate as compared to the ANN model, but the latter gives a globally better result across the entire strain rate spectrum, with values of $E_{\text{MAR}} = 1.09\%$ and $E_{\text{RMS}} = 2.40 \text{ MPa}$ respectively.

We can therefore conclude from this part of the study that the AR model is the best performing of the 5 analytical models presented, which is in agreement with the fact that it is widely used for thermomechanical processing, but the ANN model approach has advantages over the AR model approach in that, overall, the ANN model is more faithful to experimental data than is the AR model, for all strain rates.

5. Conclusions

Experimental tests were performed on a Gleeble thermomechanical simulator for a modified medium carbon alloy to investigate the applicability and predictive accuracy of 5 analytical models and an Artificial Neural Network model over a range of strains ($0.0 - 0.7$), strain rates ($0.001 \text{ s}^{-1} - 5 \text{ s}^{-1}$) and temperatures ($1050^\circ\text{C} - 1250^\circ\text{C}$). The analytical models selected for this study were the Johnson-Cook (JC) model [17], the Modified Zerilli-Armstrong (MZA) model [49], the Hansel-Spittle (HS) model [50], the Arrhenius (AR) model [55] and the PTM model [40]. The ANN model selected is the one introduced by Pantalé et al. [16]. An analysis of the data from the Gleeble trials and a comparison of the 6 models proposed in this study against the experiments led to the following conclusions.

From an experimental point of view, it appears from the tests carried out that the flow stress σ^y increases with a decrease of the temperature T and an increase of the strain rate $\dot{\epsilon}$ due to the competitive appearance of the dynamic softening and work hardening mechanisms. The dynamic recrystallization (DRX) phenomenon, introduced through the difference between the maximum and permanent strains, shows a partially complete microstructure evolution. Thus, at high strain rates, it is difficult to visualize the DRX phenomenon on the flow curves due to the sensitivity of this phenomenon to the strain rate. A study focused on an in-depth analysis of the microstructure of this steel alloy and its impact on mechanical properties is currently underway.

Five analytical models and an Artificial Neural Network-based model were identified for this alloy. Among the analytical models, the JC and MZA models proved inadequate to reproduce the behavior of this material, while the HS, PTM and AR models showed their capabilities, presenting an acceptable E_{MAR} error (from 3.5% to 7.7%). The AR model (with a 3.56% error) proved superior to the other two, thus justifying its use in thermomechanical processes. The ANN model is largely more accurate than the analytical models in predicting the flow stress σ^y of medium carbon, with a $E_{\text{MAR}} = 0.62\%$.

To test the performance of each proposed model, a study was conducted to evaluate the interpolation and extrapolation capability of the developed models. In the case of interpolation, the HS, PTM and AR models correlated well with the experiment, but the ANN model was superior, with an error factor 5 times lower than the AR model. For data extrapolation, the HS, PTM and AR models again correlated well with the experiment, but the ANN model once again performed better globally.

Identifying the parameters of an ANN model from experimental data requires more time than identifying the parameters of analytical models (about 40 minutes on a working laptop), but as shown by Pantalé et al. [16,45], implementing an ANN model in a computational code is straightforward.

Author Contributions: Conceptualization, P.T.M. and O.P.; methodology, O.P.; software, P.T.M. and O.P.; validation, O.P.; formal analysis, O.P.; investigation, P.D.; resources, P.D. and M.J.; data curation, P.T.M. and P.D.; writing—original draft preparation, P.T.M.; writing—review and editing, O.P.; visualization, O.P.; supervision, M.J., A.T. and O.P.; project administration, M.J.; funding acquisition, M.J. All authors have read and agreed to the published version of the manuscript.

Funding: This work was supported by the Natural Sciences and Engineering Research Council of Canada (NSERC) in the framework of a Collaborative Research and Development project (CRD) [Grant number 5364418].

Data Availability Statement: The raw/processed data required to reproduce these findings cannot be shared at this time due to privacy and ethical concerns.

Acknowledgments: The authors acknowledge Mr. Jean-Benoit Morin, Director of Metallurgy and Quality from Finkl Steel-Sorel, Dr. Abdelhalim Loucif from R&D department of Finkl Steel-Sorel, Ecole de Technologie Supérieure, and Ecole Nationale d'Ingenieurs de Tarbes, France, for providing technical data, materials, and testing facilities.

Conflicts of Interest: The authors declare no conflict of interest.

Abbreviations

The following abbreviations are used in this manuscript:

ANN	Artificial Neural Network
AR	Arrhenius
CPU	Central Processing Unit
DRV	Dynamic Recovery
DRX	Dynamic Recrystallization
FEA	Finite Element Analysis
HS	Hansel-Spittel
JC	Johnson-Cook
MZA	Modified Zerilli-Armstrong
WH	Work Hardening
ZA	Zerilli-Armstrong

Appendix A

We report here after the computing process and the 180 coefficients of the Artificial Neural Network ANN-3-15-7-1-sigmoid model used in Section 3.6. In order to use this model, we describe here after the details of the procedure to compute the flow stress σ^y from the input variables ε^p , $\dot{\varepsilon}$ and T . This process can be decomposed into 3 phases:

- We first have to normalize the input values of the ANN x_i within the range $[0, 1]$ to avoid an ill-conditioned system as presented by many other authors in the literature [42,60]. Therefore, the three components of the input vector \vec{x} are obtained from the plastic strain ε^p , the plastic strain rate $\dot{\varepsilon}$ and the temperature T using the following expressions:

$$\vec{x} = \begin{cases} x_1 = \frac{\varepsilon^p - [\varepsilon^p]_{min}}{[\varepsilon^p]_{max} - [\varepsilon^p]_{min}} \\ x_2 = \frac{\ln(\dot{\varepsilon}/\dot{\varepsilon}_0) - [\ln(\dot{\varepsilon}/\dot{\varepsilon}_0)]_{min}}{[\ln(\dot{\varepsilon}/\dot{\varepsilon}_0)]_{max} - [\ln(\dot{\varepsilon}/\dot{\varepsilon}_0)]_{min}} \\ x_3 = \frac{T - [T]_{min}}{[T]_{max} - [T]_{min}} \end{cases} \quad (A1)$$

where $[]_{min}$ and $[]_{max}$ are the boundaries of the range of the corresponding field: $\varepsilon^p \in [0.0, 0.7]$, $\dot{\varepsilon} \in [0.001 \text{ s}^{-1}, 5.0 \text{ s}^{-1}]$, $T \in [1050 \text{ °C}, 1250 \text{ °C}]$ and $\sigma \in [1.311 \text{ MPa}, 153.739 \text{ MPa}]$. The reference strain rate is $\dot{\varepsilon}_0 = 0.001 \text{ s}^{-1}$.

- Then, we compute the output s of the ANN from the input vector \vec{x} using the following three equations:

$$\vec{y}_1 = \left[1 + \exp \left(-w_1 \cdot \vec{x} - \vec{b}_1 \right) \right]^{-1} \quad (A2)$$

$$\vec{y}_2 = \left[1 + \exp \left(-w_2 \cdot \vec{y}_1 - \vec{b}_2 \right) \right]^{-1} \quad (A3)$$

$$s = \vec{w}^T \cdot \vec{y}_2 + b \quad (A4)$$

- Finally, the flow stress σ^y can be obtained from the output s of the ANN using the following equation:

$$\sigma^y = ([\sigma]_{max} - [\sigma]_{min})s + [\sigma]_{min} \quad (A5)$$

Conforming to the computing process proposed by Equations (A1-A5), we report hereafter the 180 coefficients of the ANN-3-15-7-1-sigmoid model used in Section 3.6. The weight matrix for the first hidden layer w_1 is a 15×3 matrix:

$$w_1 = \begin{bmatrix} 2.2206 & -3.7555 & -6.7246 \\ -4.8598 & 5.7431 & -5.8538 \\ 2.3099 & 3.3325 & -5.1795 \\ 2.0475 & 0.8006 & -1.4259 \\ 8.8358 & -6.0362 & 0.8226 \\ -1.2613 & -0.9274 & -2.3725 \\ -0.3561 & 6.5032 & -10.7573 \\ -11.7226 & -2.0455 & 1.2248 \\ 3.1066 & 26.5580 & 18.6540 \\ -0.5150 & -5.6922 & 1.0104 \\ -6.4755 & 8.4888 & -2.4459 \\ -1.8791 & -0.5380 & 2.2295 \\ -6.0206 & 1.2776 & 0.2169 \\ 0.2619 & -4.7974 & -1.1282 \\ -27.1456 & -0.5327 & -0.5303 \end{bmatrix}$$

The biases of the first hidden layer \vec{b}_1 is a 15-component vector:

$$\vec{b}_1 = \begin{bmatrix} 5.9164 \\ -1.9074 \\ -2.6837 \\ -1.0954 \\ -0.9509 \\ 3.1682 \\ -4.3964 \\ 0.6343 \\ -5.0676 \\ 2.0228 \\ 1.1267 \\ -0.9671 \\ -0.6263 \\ 2.3110 \\ -0.3037 \end{bmatrix}$$

The weight matrix for the second hidden layer w_2 is a 7×15 matrix:

$$w_2^T = \begin{bmatrix} -0.5783 & 1.2724 & 0.5747 & 0.6449 & -4.2203 & -0.2380 & 0.2591 \\ -0.4852 & 5.2807 & 0.8888 & -8.2324 & -2.0075 & -0.6474 & -1.0787 \\ 4.4499 & -0.0137 & 0.1657 & 0.3198 & 4.9765 & -1.2503 & 0.8219 \\ 1.7571 & 0.7730 & 0.0208 & -1.3316 & -0.8945 & -0.7284 & -0.1831 \\ -1.0866 & 0.1330 & -0.8615 & -0.1283 & 0.2218 & -0.1772 & -2.7458 \\ -0.3925 & 1.3994 & 0.0630 & -1.8397 & -1.1047 & -1.9839 & 0.5767 \\ -0.2121 & 0.9977 & 1.2028 & -9.6525 & 0.5520 & 0.1062 & -0.0409 \\ -1.1518 & -1.6402 & -4.1501 & -1.0759 & 0.4749 & -2.8350 & 0.9225 \\ -1.1453 & -0.2173 & -0.1382 & 0.8264 & -0.5125 & 0.1882 & -0.8654 \\ -4.3301 & -0.3711 & -7.4305 & 3.5926 & -9.6217 & -1.2375 & 1.6171 \\ 2.3907 & -1.0085 & -0.8828 & -1.1891 & 0.9947 & 1.1178 & -1.0953 \\ -1.5955 & 1.6313 & 0.4916 & 0.1906 & -1.9216 & -1.4140 & 1.3827 \\ -1.6985 & 1.4277 & -3.4462 & -8.3777 & -1.2132 & -1.2158 & 2.8512 \\ -1.9954 & -2.0159 & -8.3455 & -0.5205 & 0.2942 & -1.3337 & 0.2026 \\ 4.3040 & -0.7164 & -1.0859 & 3.4294 & -23.8003 & 12.5859 & 7.3721 \end{bmatrix}$$

The biases of the second hidden layer \vec{b}_2 is a 7-component vector:

$$\vec{b}_2 = \begin{bmatrix} 0.7534 \\ 0.9473 \\ 0.6055 \\ -0.7793 \\ -1.0305 \\ -1.5779 \\ -0.1471 \end{bmatrix}$$

The weight vector for the output layer \vec{w} is a 7-component vector:

$$\vec{w} = \begin{bmatrix} 0.1920 \\ 0.3406 \\ 0.3839 \\ -0.2880 \\ 1.2047 \\ -1.4126 \\ -0.2215 \end{bmatrix}$$

The bias of the output layer b is a scalar:

$$b = 0.1178$$

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