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#### CISS Princeton, 2021

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Why combustion?

**Statistics** 

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NO <sub>x</sub> reduction
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COMMERCIAL FUELS, SURROGATES, BIOFUELS
Gasoline
Diesel
Jet fuel
Biofuels

4/

5/

# Part 1 INTRODUCTION



La Guerre du Feu (Quest for Fire), Jean-Jacques Annaud, 1981

# Where is combustion?



# What is combustion? (1/2)

The **oxidation** of a fuel, ultimately leads to the formation of carbon dioxide, water, and heat in the case of organic fuels (e.g. hydrocarbons).

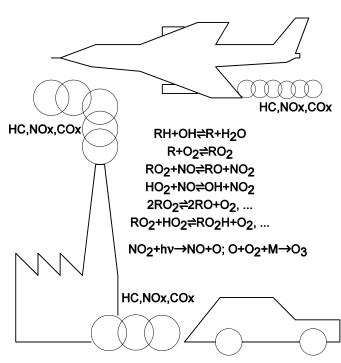
Other definition: an exothermic redox reaction between a fuel (reductant) and an oxidant (e.g., oxygen from air)

Incomplete combustion yields UHC and soot.



NOx resulting from nitrogen oxidation can also be released.





# What is combustion? (2/2)

Combustion involves chemical reactions, thermochemistry, kinetics, heat and mass transfer, radiation...

The overall/global chemical equation, e.g.  $2 H_2 + O_2 = 2 H_2O$ ,  $CH_4 + 2O_2 = CO_2 + 2H_2O$ , is a mass balance that does inform on the reaction pathways to products.

#### **Equivalence ratio and excess air:**

```
\phi= {[Fuel]/[O<sub>2</sub>]} / {[Fuel]/[O<sub>2</sub>]} at stoichiometry \lambda = 1/\phi
```

```
2 H_2 + O_2 = 2 H_2O \phi=1 and \lambda=1 (stoichiometric mix) 3 H_2 + O_2 = 2 H_2O + H_2 \phi>1 and \lambda<1 (fuel-rich, excess of fuel, some left over) 2 H_2 + 2 O_2 = 2 H_2O + O_2 \phi<1 and \lambda>1 (fuel-lean, excess of oxygen, some left over)
```

The combustion of methane involves a long sequence of elementary reactions (*initiation*, *propagation*, *branching*, *and termination*). They involve stable species and labile species (*atoms*, *radicals*). These reactions proceed with reaction rates ranging from slow (e.g., RH+O<sub>2</sub>) to very fast (R+R').

# Why combustion?

Transport accounts for ca. 20% of the total global primary energy consumed, ca. 23% of CO<sub>2</sub> emissions, ca. 7 billion tons of CO<sub>2</sub>.

> 99.9%Transport is powered by I.C. engines (land and marine) and air transport by GT.

G. Kalghatgi Applied Energy 225 (2018) 965–974

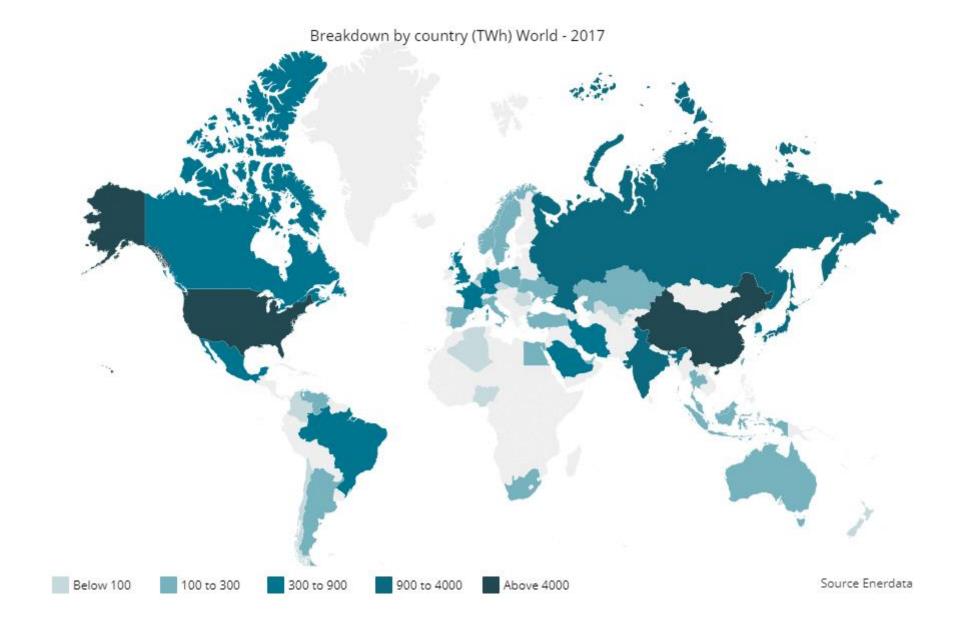
**Table 2**Fuel capacity and equivalent battery pack size for three different types of aircraft.

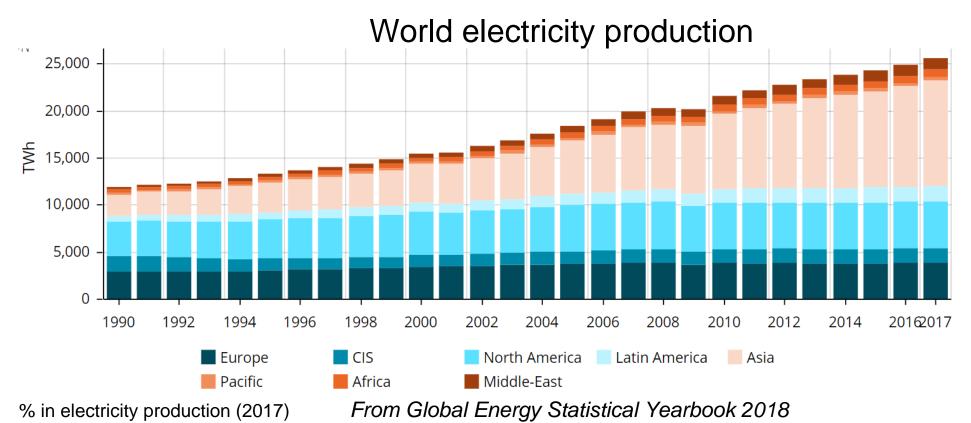
	Maximum Take-off Weight (MTOW), kg	Volume of fuel, liters	Weight of fuel <sup>a</sup> , kg	Energy content of fuel (ECF) <sup>a</sup> , MWh	Weight of battery pack with the same ECF <sup>b</sup> (WBP), kg	WBP/MTOW
Embraer 135 [41] Airbus A320 Neo [42]	20,000 76,000	5146 26,730	4168 21,651	51 266	284,831 1,479,506	14 19
Airbus A380-800 [43]	576,000	323,545	262,071	3223	17,908,216	31

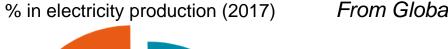
The global demand for transport energy is ca. 105 TWh of liquid fuel energy/day (38,325 TWh/year)

In 2016 the consumption of wind and solar energy together reached 1,292 TWh/year.

In 2016 the consumption of electricity reached almost 25,000 TWh/year.



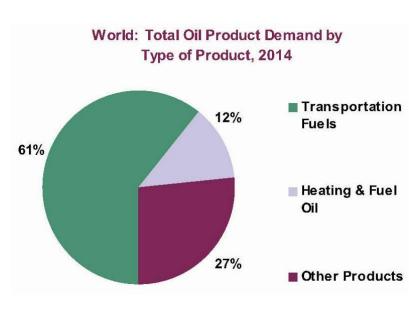




Non renewables

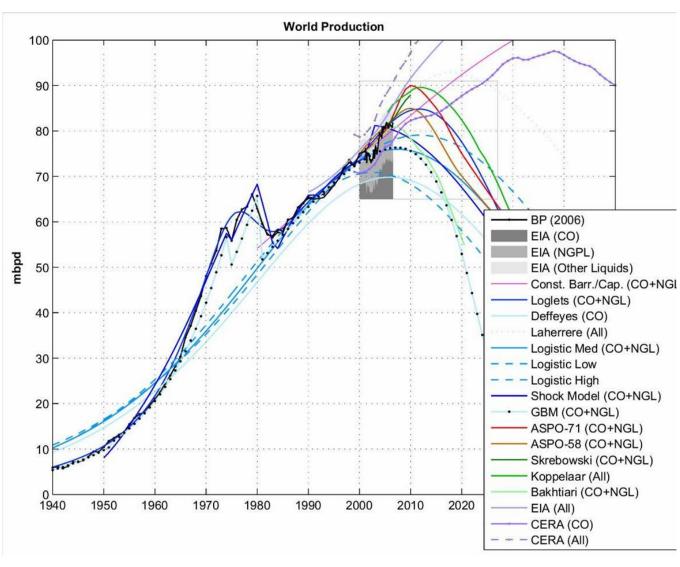
Renewables

#### What do we burn?

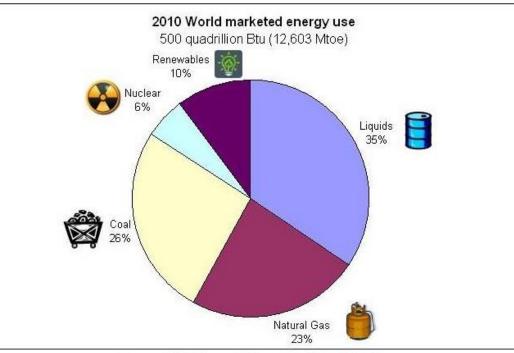


IEA, 2009

98% transport fuels are oil-derived



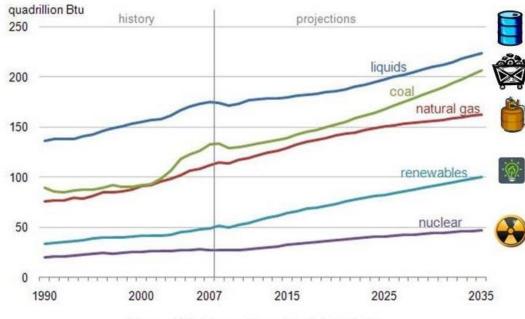
#### **Primary Energy Use**



Source: U.S. Energy Information Administration (Report #:DOE/EIA-0484(2010))

#### **Primary Energy Use**

Figure 2. World marketed energy use by fuel type



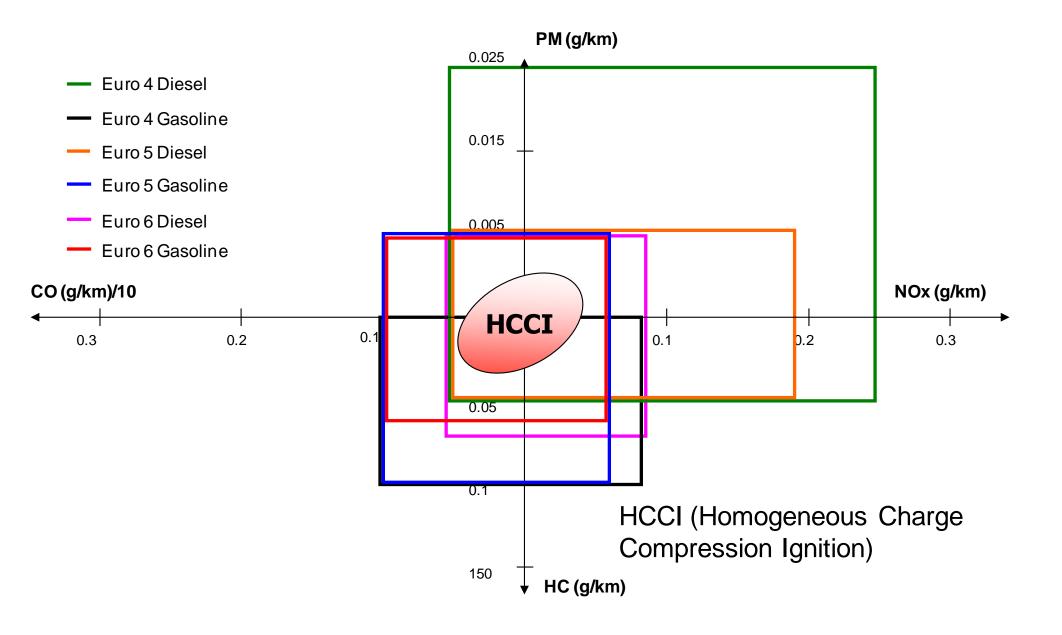
Source: U.S. Energy Information Administration (Report #:DOE/EIA-0484(2010))

# **Sustainability:**

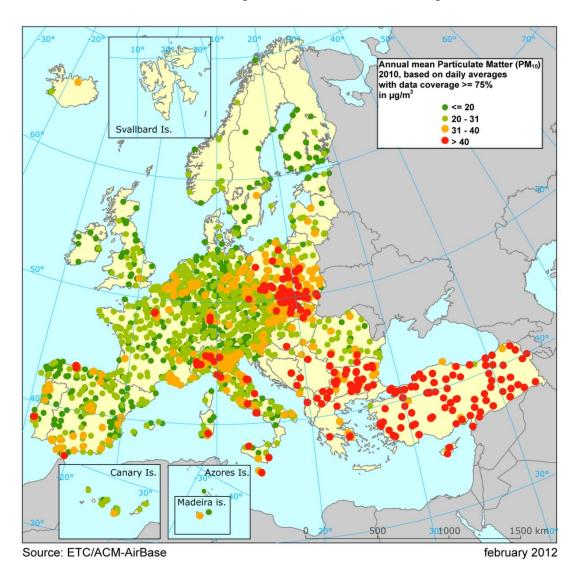
#### We need to burn cleaner

We need more efficient combustion (energy production)

# **EU** regulations



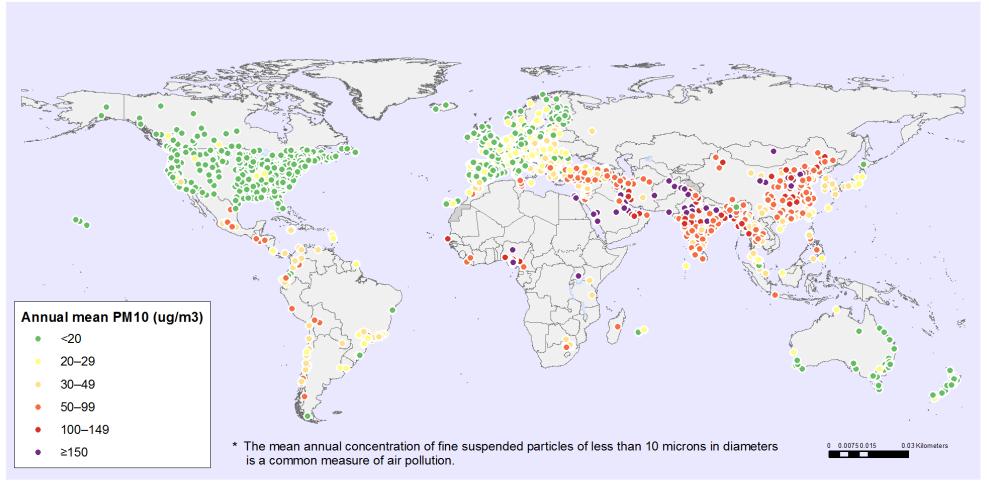
# **PM**<sub>10</sub> (<10 microns)



Sources of particulates: industry, agriculture, air and ground transportation (soot, tires, brakes), homes, wild fires, volcanoes, soil erosion and hurricanes/tornados, sea salts...

#### **PARTICULATES**

Concentration of particulate matter with an aerodynamic diameter of 10 µm or less (PM10) in nearly 3000 urban areas\*, 2008–2015



The boundaries and names shown and the designations used on this map do not imply the expression of any opinion whatsoever on the part of the World Health Organization concerning the legal status of any country, territory, city or area or of its authorities, or concerning the delimitation of its frontiers or boundaries. Dotted and dashed lines on maps represent approximate border lines for which there may not yet be full agreement.

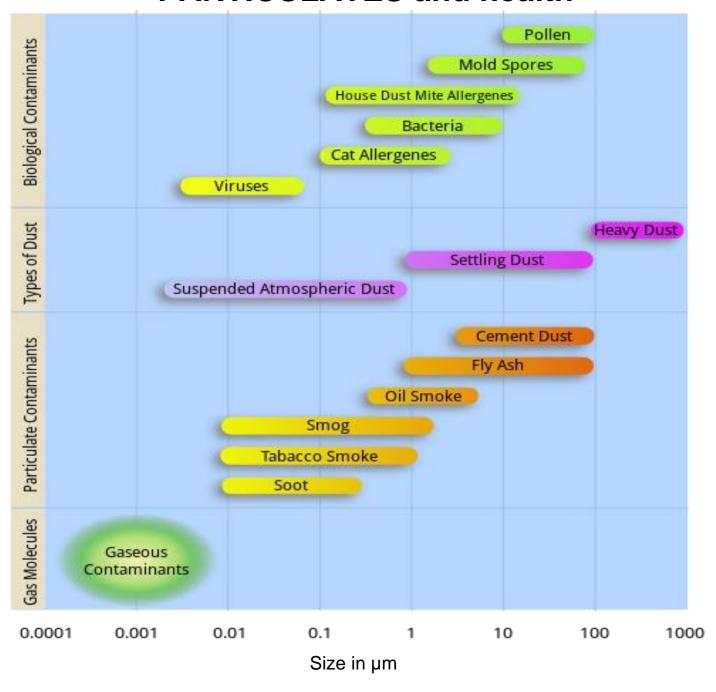
Data Source: World Health Organization Map Production: Information Evidence and Research (IER) World Health Organization



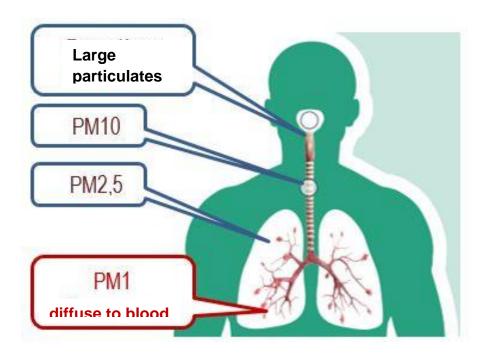
© WHO 2016. All rights reserved.

See also https://www.conserve-energy-future.com/causes-and-effects-of-particulate-matter.php

#### **PARTICULATES** and health



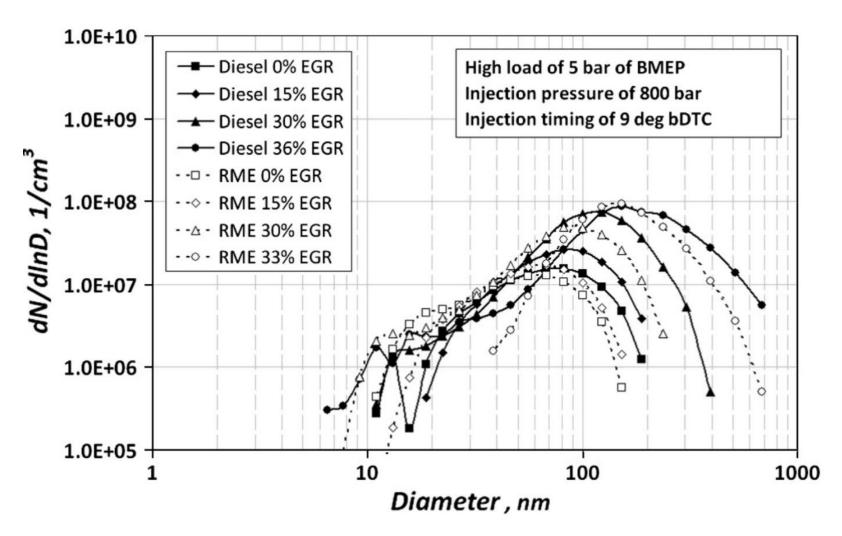
#### **PARTICULATES** and health



Source: produits.xpair.com

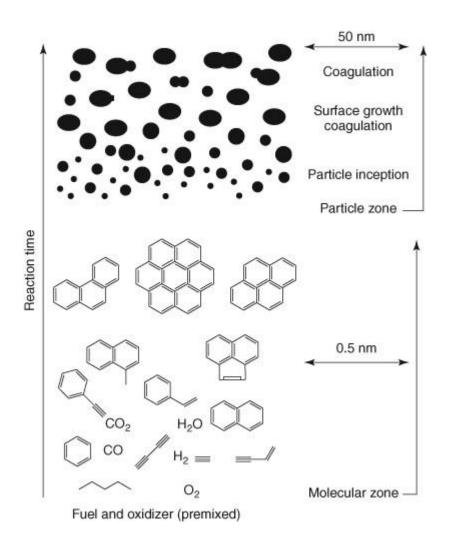
N.B. PM1 (< 1 µm or 1000 nm)

## **PARTICULATES** from i.c. engines



Source: Labecki et al., Fuel (2013) https://doi.org/10.1016/j.fuel.2013.05.013

## **PARTICULATES**

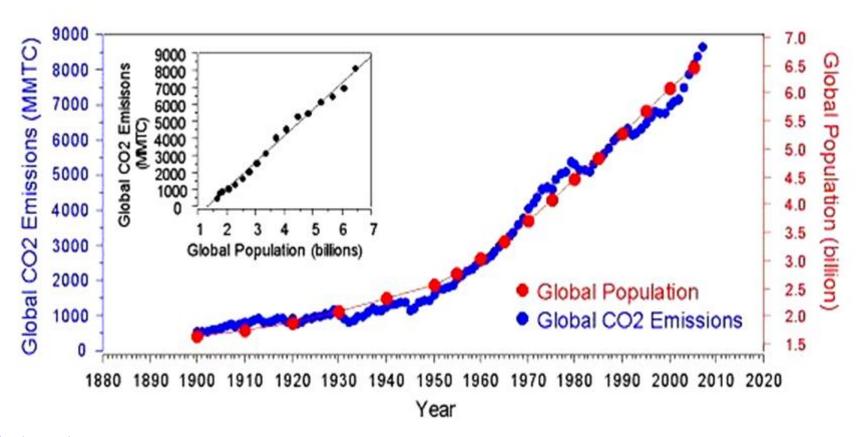


TEM image of soot particles generated by CAST

Source: <a href="http://www.sootgenerator.com">http://www.sootgenerator.com</a>

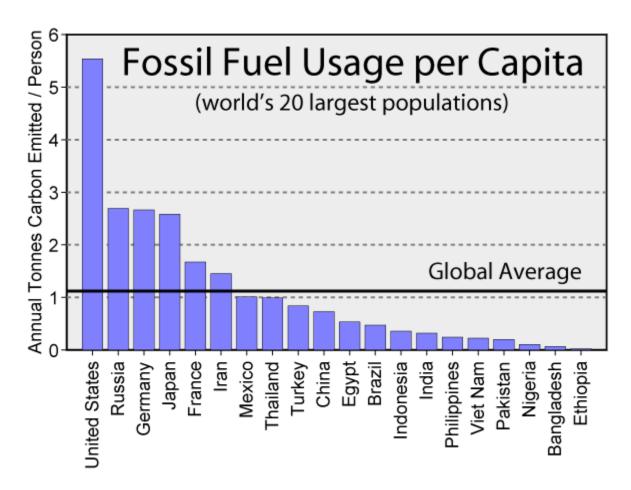
Source: H. Bockhorn (1986)

GHG: CO<sub>2</sub>

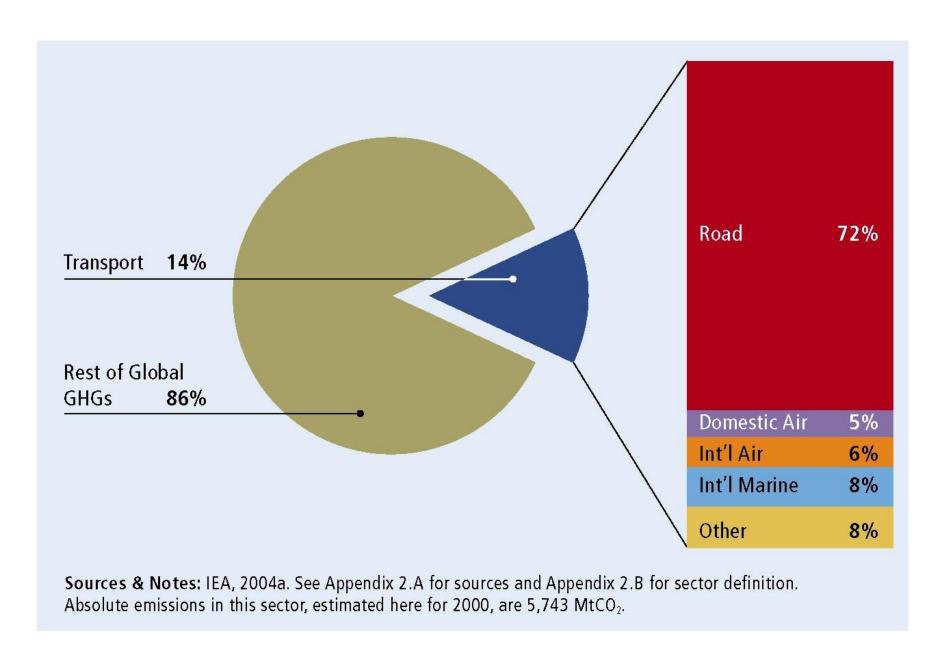


http://www.worldclimatereport.com

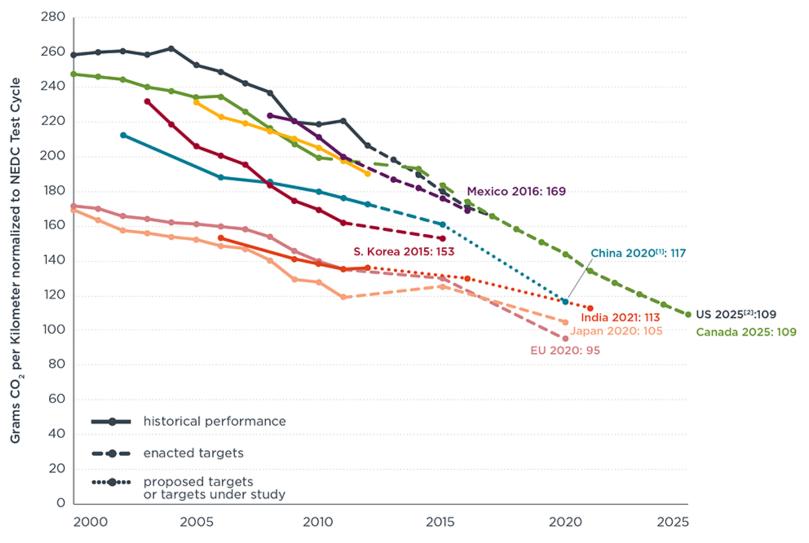
#### GHG: CO<sub>2</sub>



## **GHG**

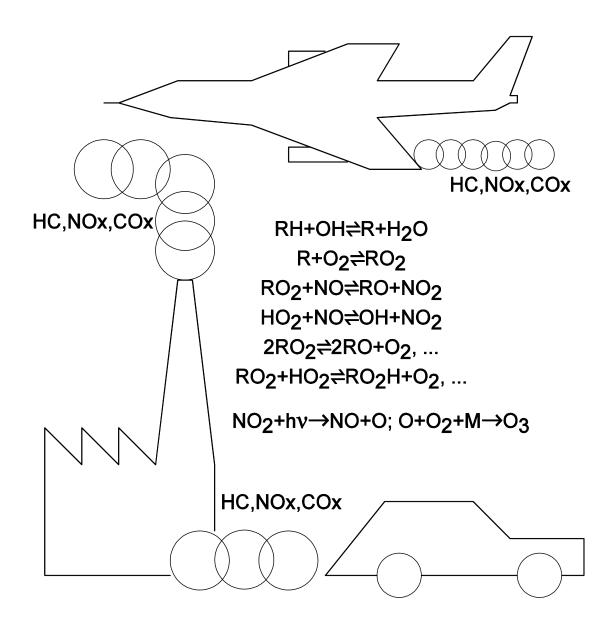


#### REDUCE CO<sub>2</sub> EMISSIONS



- [1] China's target reflects gasoline vehicles only. The target may be higher after new energy vehicles are considered.
- [2] US, Canada, and Mexico light-duty vehicles include light-commercial vehicles.
- [3] Supporting data can be found at: http://www.theicct.org/info-tools/global-passenger-vehicle-standards

# **Chemical Kinetics and Modeling**



# **Chemical Kinetics and Modeling**

(a)
$$C_{10}H_{16} \xrightarrow{-11} C_{10}H_{15} \xrightarrow{+O_2} C_{10}H_{15}O_2 \xrightarrow{RO_2} C_{10}H_{15}O_4C_{10}H_{15} \longrightarrow O_2 + 2C_{10}H_{15}O \xrightarrow{H-shift} C_{10}H_{14}OH$$

$$Limonene \qquad (R') \qquad (RO_2^{-1}) \qquad (R$$

# **Experimental data** ← Model

# Constrain the model by using

**Global parameters:** Ignition delays (initiation reactions, R+O₂) Burning velocities (H fluxes)

Detailed information: Species concentrations (~ all processes)

Initiations:  $RH \leftrightarrows R + H$ 

 $RH \leftrightarrows R' + R''$ 

 $RH + O_2 \leftrightarrows R + HO_2$ 

Propagations:  $RH + X \hookrightarrow R + HX$  (X= H, O, OH, HO<sub>2</sub>, CH<sub>3</sub>, HCO, ...)

Terminations:  $R + H \hookrightarrow RH$ 

 $R' + R" \hookrightarrow RH$ 

**☞** Different types of 'reactors': ST, PF, PSR, Flames (laminar premixed, opposed flow), RCM, engines

# Global vs. detailed chemistry

 $H_2 + \frac{1}{2} O_2 = H_2O$ : mass balance; misrepresents reaction pathways

Global Rate = A T<sup>n</sup> exp[-E / R T] [ Fuel ] [  $O_2$  ]  $\frac{1}{2}$ 

#### In reality, many more reactions:

*M*= collision partner, e.g. diluent

The value of k<sub>i</sub> indicates how fast the reaction can proceed

Such sets of reactions constitute a "chemical kinetic reaction mechanism"

#### **Kinetics**

$$A + B \rightarrow C + D$$

Rate = 
$$-d[A]/dt = k_{+}[A][B] = A_{+}T^{n} exp[-E/RT][A][B]$$

**Reverse reaction** 

$$C + D \rightarrow A + B$$

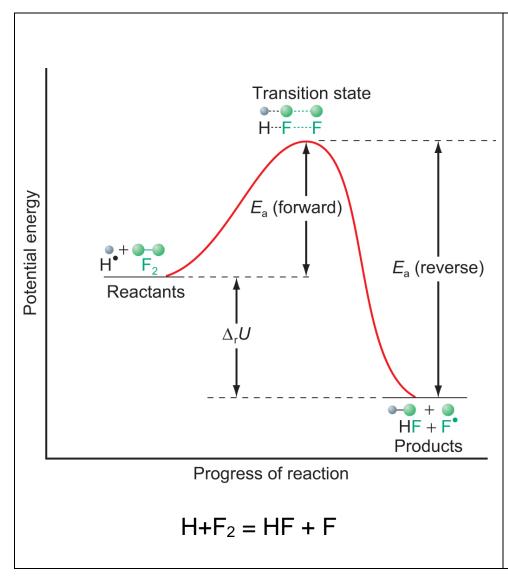
Rate = 
$$-d[C]/dt = k_{-}[C][D] = A_{-}T^{n'} \exp[-E'/RT][C][D]$$

#### Equilibrium constant computed from thermochemistry $K_{eq} = k_{+}/k_{-}$

k+ and/or k- are determined experimentally or computed

K can be obtained in tabulations (JANAF, NASA ...)

#### **Thermodynamics**



As H approaches F<sub>2</sub>, the F-F bond extends and electron density moves from that F-F bond into the newly forming F-H bond. This involves an increase in potential energy.

1<sup>st</sup> law: The energy U of an isolated system is constant

dU = dQ + dW; Q= heat absorbed by the system; w= work done on system

**2**<sup>nd</sup> **law**: Mechanical energy can be transferred completely into heat but heat cannot be transformed completely into mechanical energy

$$dS \ge dQ/T$$
;  $S = \text{entropy}$ 

$$dS = dQ_{rev}/T$$
 and  $dS > dQ_{irrev}/T$ 

**3<sup>rd</sup> law**: The entropy of a perfectly crystalline substance at 0°K is 0

$$S = 0$$
 at  $T = 0$ ° $K ( \lim_{T \to 0} (S) = 0 )$ 

# **Thermodynamics**

Gibbs energy: G = H - TS

At constant T,  $\triangle G = \triangle H - T \triangle S$ 

Equilibrium occurs at minimum G (at constant T, P)

Equilibrium constant:  $\Delta G^{\circ} = -RT \ln(K)$  (° refers to the standard state)

Heat capacities (p at constant pressure; v at constant volume):

$$C_v = (\partial U/\partial T)_v$$
  $C_p = (\partial U/\partial T)_p$   $C_p = C_v + R$  (ideal gas; universal gas constant 8.314 J/mol/K)

$$H(T_2) = H(T_1) + \int_{T_1}^{T_2} C_p dT$$

$$\Delta_r H(T_2) = \Delta_r H(T_1) + \int_{T_1}^{T_2} \Delta_r C_p dT$$

# **Thermodynamics**

#### Gibbs Energy (G) indicates the spontaneity of a reaction

G depends on Enthalpy and Entropy

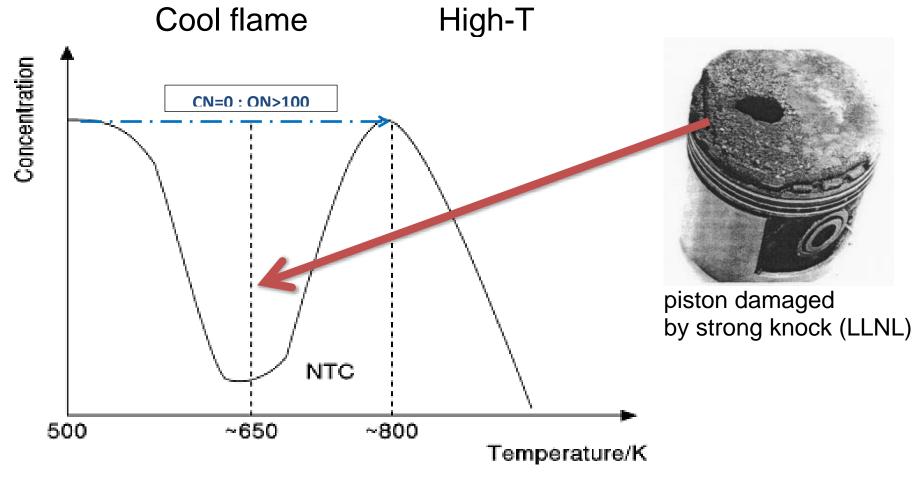
Entropy contribution increases as T increases: G = H - TS

 $\Delta_r G < 0$  for spontaneous reaction

 $\Delta_r G > 0$  for non-spontaneous reaction

## **Global fuels properties**

Cetane number, Octane number



Fuel concentration vs. temperature

S.I. engines: ON=100 for iso-octane ( $C_8H_{18}$ ) and ON=0 for n-heptane ( $C_7H_{16}$ )

C.I. engines: CN=100 for n-hexadecane ( $C_{16}H_{34}$ ) and CN=0 for 1-methylnaphtalene ( $C_{11}H_{10}$ )

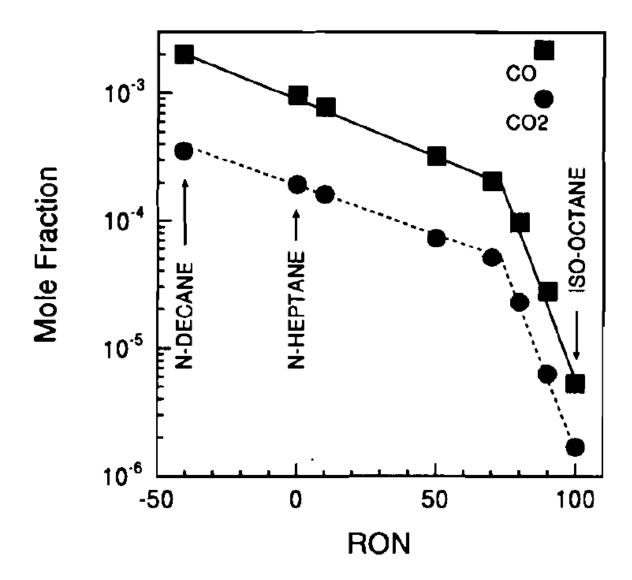
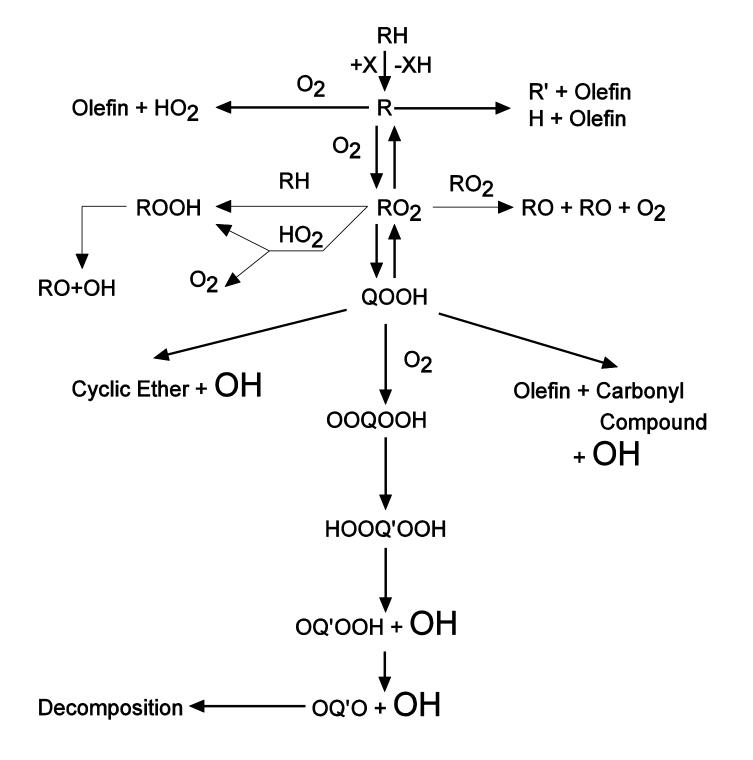


FIGURE 6 Carbon monoxide and carbon dioxide maximum mole fraction at low temperature, as a function of RON.

Dagaut et al., CST 103:1-6, 315-336 (1994).



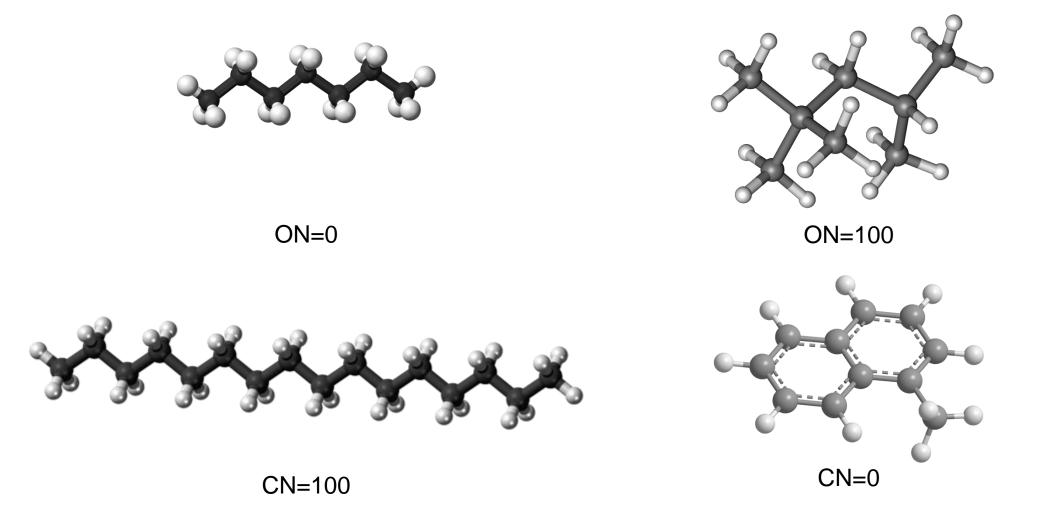
(a)
$$C_{10}H_{16} \xrightarrow{-H} C_{10}H_{15} \xrightarrow{+O_2} C_{10}H_{15}O_2 \xrightarrow{RO_2} C_{10}H_{15}O_4C_{10}H_{15} \xrightarrow{-P} O_2 + 2 C_{10}H_{15}O \xrightarrow{H-shift} C_{10}H_{14}OH$$

$$Limonene \qquad (R') \qquad (RO') \qquad (RO') \qquad (RO')H_{14}OH$$

$$C_{10}H_{14}OH \xrightarrow{+O_2} C_{10}H_{15}O_3 \xrightarrow{H-shift} C_{10}H_{15}O_5 \xrightarrow{H-shift} C_{10}H_{15}O_7 \xrightarrow{H-shift} C_{10}H_{1$$

$$(HOO)_4 POO^{\bullet} \xrightarrow{\text{H-shift}} (HOO)_4 P'OOH \xrightarrow{+O_2} (HOO)_5 P''OO^{\bullet} \xrightarrow{\text{H-shift}} (HOO)_5 P'''O$$

## **Structure-reactivity**



S.I. engines: ON=100 for iso-octane ( $C_8H_{18}$ ) and ON=0 for n-heptane ( $C_7H_{16}$ )

C.I. engines: CN=100 for n-hexadecane ( $C_{16}H_{34}$ ) and 0 for 1-methylnaphtalene ( $C_{11}H_{10}$ )

#### **Hydrocarbons** CN **Paraffins** 2-Méthylpentane 33 3-Méthylpentane 30 n-Heptane 56 2,2,4-Triméthylpentane 12 n-Décane 76 л-Dodécane 80 3-Éthyldécane 48 4,5-Diéthyloctane 20 2,3,4,5,6-Pentaméthylheptane 9 n-Tridécane 88 2,5-Diméthylundécane 58

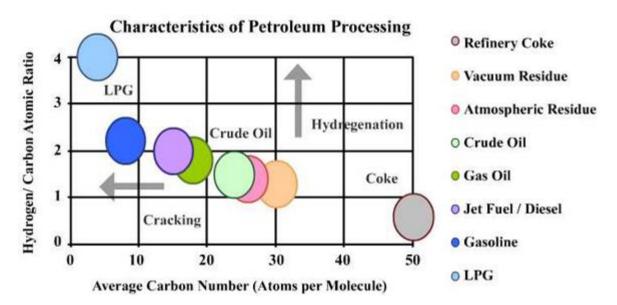
Hydrocarbons	CN	
n-Hexadécane	100	
Heptaméthylnonane	15	
5-Butyldodécane	45	
7,8-Diméthyltétradécane	40	
n-Heptadécane	105	
7-Butyltridécane	70	
n-Octadécane	110	
9-Méthylheptadécane	66	
8-Propylpentadécane	48	
7.8-Diéthyltétradécane	67	
5,6-Dibutyldécane	30	
n-Nonadécane	110	
n-Eicosadécane	110	
9,10-Diméthyloctadécane	59	
7-Hexylpentadécane	83	
2,9-Diméthyl-5,6-diisoamyldécane	48	
9,10-Dipropyloctadécane	47	
10,13-Diméthyldocosane	56	
9-Heptylheptadécane	87	
Oléfines		
Diisobutylène	10	
Tétradéc-1-ène	79	
Hexadéc-1-ène	88	
4-Butyldodéc-4-ène	45	
Tétraisobutylène	4	

Hydrocarbons	CN	
n-Nonylbenzène	50	
n-Octylxylène	20	
2-Phénylundécane	51	
2-Phénylundéc-2-ène	23	
2-Méthyl-2-(2-naphtyl)hexane	10	
n-Dodécylbenzène	68	
4-Phényidodécane	42	
2-n-Octylnaphtalène	18	
4-Méthyl-4-(2-naphtyl)heptane	9	
7-Phényltridécane	41	
n-Tétradécylbenzène	72	
2-Phényitétradécane	49	
3,6-Diméthyl-3-(2-naphtyl)octane	18	
5-Méthyl-5-(2-naphtyl)nonane	12	
2-Méthyl-2-(2-naphtyl)décane	18	
3-Éthyl-3-(2-naphtyl)nonane	13	
2-Méthyl-4-isobutyl-4-phénylundécane	38	
2-Méthyl-2-phénylpentadécane	39	
5-Butyl-5-phényltétradécane	58	
1,2,4-Triméthyl-5-hexadécylbenzène	42	
5-Phényleicosane	39	

Hydrocarbons	RON	MON	
Paraffins			
Méthane	> 100	110.0	
Éthane	> 100	104.0	
Propane	> 100	100,0	
n-Butane	95,0	92,0	
2-Méthylpropane	> 100	99.0	
n-Pentane	61.7	61,9	
2-Méthylbutane	92,3	90,3	
2.2-Dimethylpropane	85,5	80,2	
n-Hexane	24,8	26.0	
2-Méthylpentane	73.4	73,5	
3-Methylpentane	74,5	74,3	
2,2-Diméthylbutane	91,8	93,4	
2,3-Diméthylbutane	103,5	94,3	
n-Heptane	0,0	0,0	
2-Méthylhexane	42.4	46,4	
3-Méthythexane	52,0	55,0	
3-Éthylpentane	65,0	69,3	
2.2-Dimethylpentane	92,8	95,6	
2.3-Diméthylpentane	91,1	88,5	
2.4-Diméthylpentane	83.1	83,8	
3.3-Diméthylpentane	80.8	86,6	

Hydrocarbons	RON	MON
2,2,3-Triméthylpentane	108.7	99.9
2,2,4-Triméthylpentane	100,0	100,0
2,3,3-Triméthylpentane	106,1	99,4
2,3,4-Triméthylpentane	102,7	95,9
2-Méthyl-3-éthylpentane	87,3	88.1
3-Méthyl-3-éthylpentane	80,8	88.7
n-Nonane et n-alcanes supérieurs	< 0	< 0
Oléfines		
Éthylène	100.0	81.0
Propylène	102,0	85.0
But-1-ène		80,0
But-2-ène	100,0	83.0
Pent-1-ène	90,9	77.1
Pent-2-ène	98,0	80,0
2-Méthylbut-1-ène	102,5	81.9
2-Méthylbut-2-ène	97,3	84.7
Hex-t-ène	76,4	63.4
Hex-2-ène	92,7	80,8
Hex-3-ène	94,0	80,1
2-Méthylpent-L-ène	95.1	78,9
3-Méthylpent-I-ène	96,0	81,2
4-Méthylpent-1-ène	95.7	80,9
2-Méthylpent-2-éne	97,8	83.0
3-Méthylpent-2-ène	97,2	81,0
4-Méthylpent-2-ène	99,3	84,3
2-Éthylpent-1-ène	98,3	79,4
3,3-Dimétylbut-1-ène	111.7	93,5
2,3-Dimétylbut-2-ène	97.4	80,5
2,3-Dimétylbut-l-ène	101,3	82,8
Hept-1-ène	54,5	50,7
Hept-2-ène	73,4	68,8

## **Composition of Fuels**



- NG: methane + higher alkanes (ca. C<sub>8</sub>)
- LPG: region-dependent; C<sub>3</sub>–C<sub>4</sub> alkanes and alkenes
- Gasoline: C<sub>4</sub>–C<sub>12</sub> hydrocarbons. Mixture of paraffins (alkanes), olefins (alkenes), cycloalkanes (naphthenes), aromatics
- Kerosene (Jet A-1 fuel), standard AFQRJOS (Aviation Fuel Quality Requirements for Jointly Operated Systems):  $C_6$ – $C_{16}$  hydrocarbons. Mixture of paraffins (alkanes), cycloalkanes (naphthenes), aromatics and <2% alkenes.
- Diesel: C<sub>6</sub>–C<sub>28</sub> hydrocarbons. Mixture of paraffins (alkanes), olefins (alkenes), cycloalkanes (naphthenes), aromatics, naphteno-aromatics

## Composition of Fuels: Additives to replace Pb(C<sub>2</sub>H<sub>5</sub>)<sub>4</sub> and others

Table 1. Relative Antiknock Effectiveness of Various Compounds<sup>a</sup>

tetraethyllead	118	tetraethyltin	4
tetraphenyllead	73	triphenylarsine	1.6
iron pentacarbonyl	50	xylidine	1.6
nickel carbonyl	35	diphenylamine	1.5
diethyl telluride	27	<i>N</i> -methylaniline	1.4
triethylbismuth	24	dimethylcadmium	1.2
diethyl selenide	7	aniline	1.0
stannic chloride	4.1	ethanol	0.1

<sup>&</sup>lt;sup>a</sup> Vs aniline = 1 on a mole basis. From ref 1e, by permission of Springer-Verlag and Ethyl Corp.

(1e) Frey, F. W.; Shapiro, H. Top. Curr. Chem.1971, 16, 243–297. (f) Shapiro, H.; Frey, F. W. The Organic Compounds of Lead; Wiley-Interscience: New York, 1968

From: Organometallics 2003, 22, 25, 5154–5178 <a href="https://doi.org/10.1021/om030621b">https://doi.org/10.1021/om030621b</a>

## 3-Way catalyst to reduce (1) CO, (2) UHC, and (3) NOx (>1980) Pt/Rh/Pa

Reduction of nitrogen oxides to nitrogen  $(N_2)$ 

$$\bullet \mathrm{C} + 2 \mathrm{NO}_2 \ o \ \mathrm{CO}_2 + 2 \mathrm{NO}$$

$$\bullet \mathrm{CO} + \mathrm{NO} \, \to \, \mathrm{CO}_2 + \frac{1}{2} \mathrm{N}_2$$

$$ullet 2{
m CO} + {
m NO}_2 \ 
ightarrow \ 2{
m CO}_2 + rac{1}{2}{
m N}_2$$

$$\bullet \mathrm{H}_2 + \mathrm{NO} \, \rightarrow \, \mathrm{H}_2\mathrm{O} + \frac{1}{2}\mathrm{N}_2$$

Oxidation of carbon, hydrocarbons, and carbon monoxide to carbon dioxide

$$\bullet \mathrm{C} + \mathrm{O}_2 \ o \ \mathrm{CO}_2$$

$$\bullet \mathrm{CO} + \frac{1}{2}\mathrm{O}_2 \, o \, \mathrm{CO}_2$$

$$ullet a\,\mathrm{C}_x\mathrm{H}_y + b\,\mathrm{O}_2 \ o \ c\,\mathrm{CO}_2 + d\,\mathrm{H}_2\mathrm{O} \qquad a,\,b,\,c,\,d,\,x,\,y \in \mathbb{Z}$$

$$a,\,b,\,c,\,d,\,x,\,y\in\mathbb{Z}$$

D

#### **EtOH**

**Production:** 

Oxidation/hydratation of ethylene: C<sub>2</sub>H<sub>4</sub> + H<sub>2</sub>O → C<sub>2</sub>H<sub>5</sub>OH

Alcoholic fermentation of sugar (bio-ethanol):  $C_6H_{12}O_6 \rightarrow 2 C_2H_5OH+ 2 CO_2 + Heat$ 

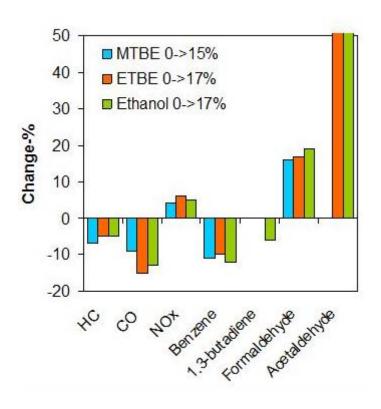
**RON: 108.6** 

#### **ETBE**

Production: isobutene + ethanol + catalyst => ethyl ter-butyl ether + H<sub>2</sub>O

**RON: 110-119** 

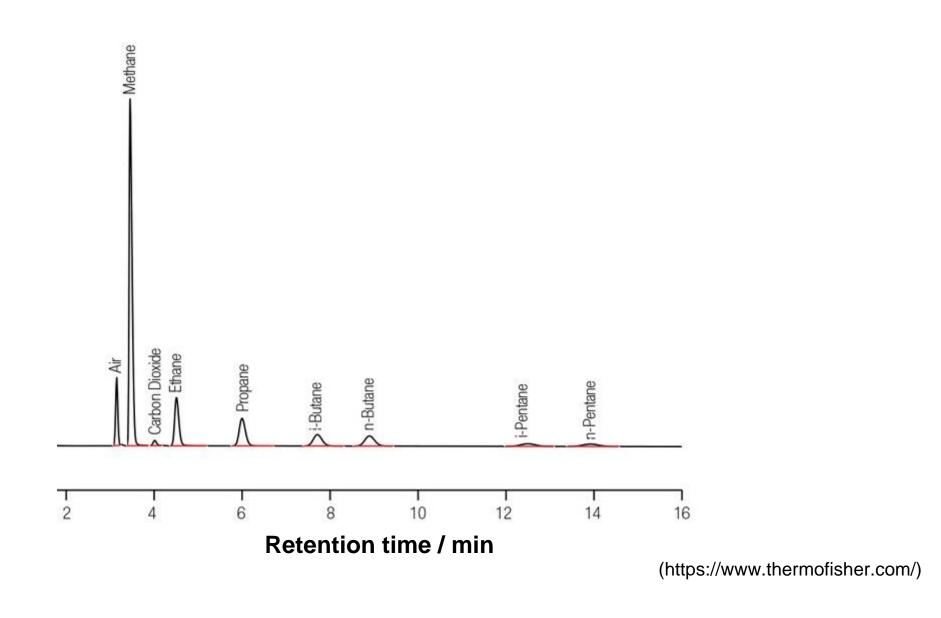
## **Composition of Fuels: Additives and Emissions**



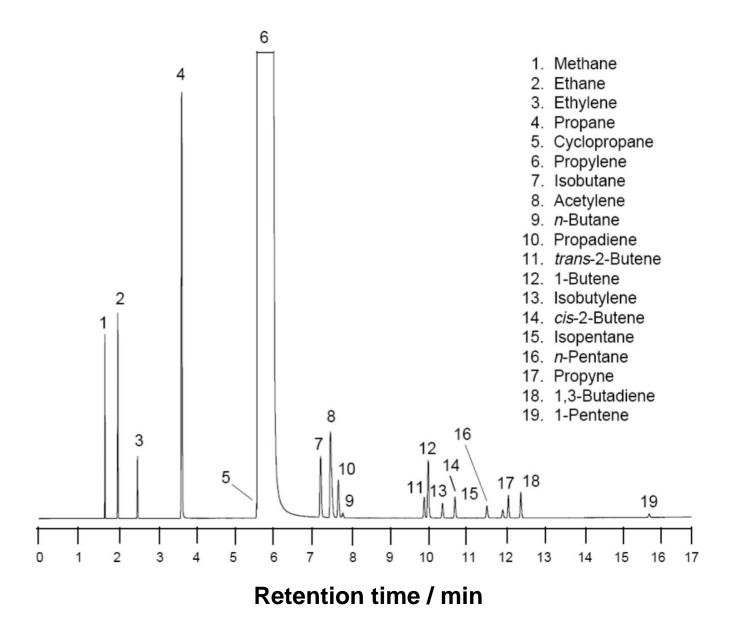
Effect of methyl ter-butyl ether, ethyl terbutyl ether, and ethanol on exhaust emissions. Change-% represents difference in emissions of blended fuels (15-17%) vs. non-oxygenated fuel.

From: Aakko-Saksa, P., Rantanen-Kolehmainen, L., Koponen, P., Engman, A. and Kihlman, J. (2011) Biogasoline options – Possibilities for achieving high bio-share and compatibility with conventional cars. SAE International Journal of Fuels and Lubricants, 4:298–317 (also SAE Technical Paper 2011-24-0111).

# GC analysis of a NG sample

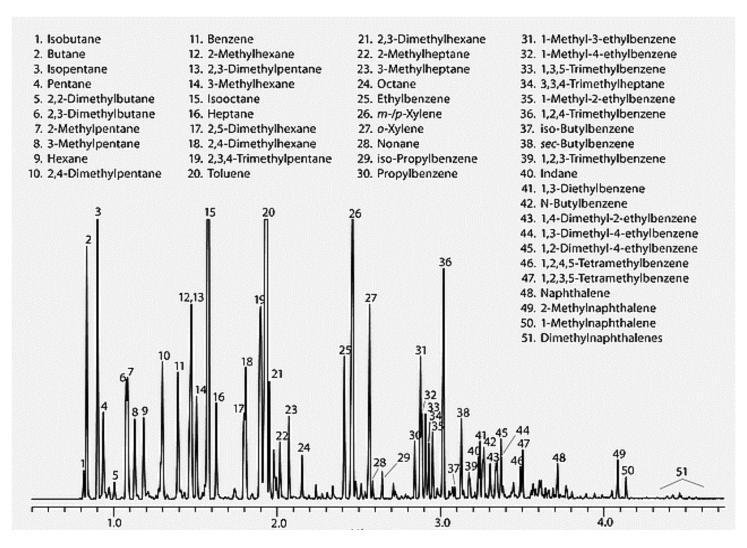


## GC analysis of a US LGP sample



(https://gassite.com/)

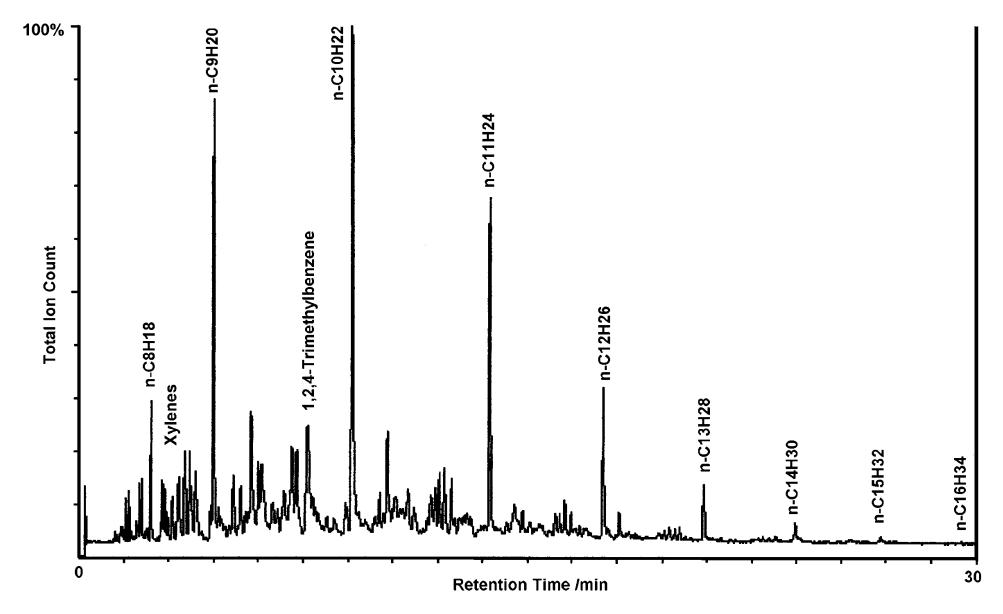
## GC analysis of a gasoline sample



Retention time / min

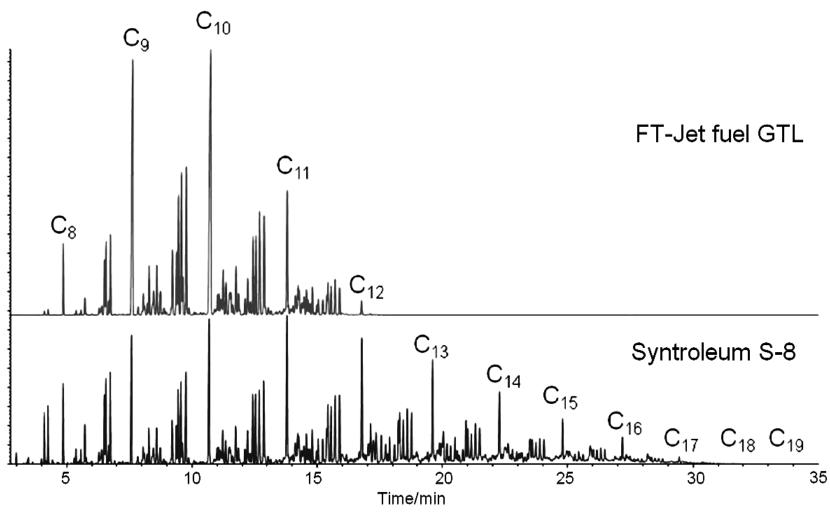
(https://www.sigmaaldrich.com)

# GC analysis of a Jet fuel sample



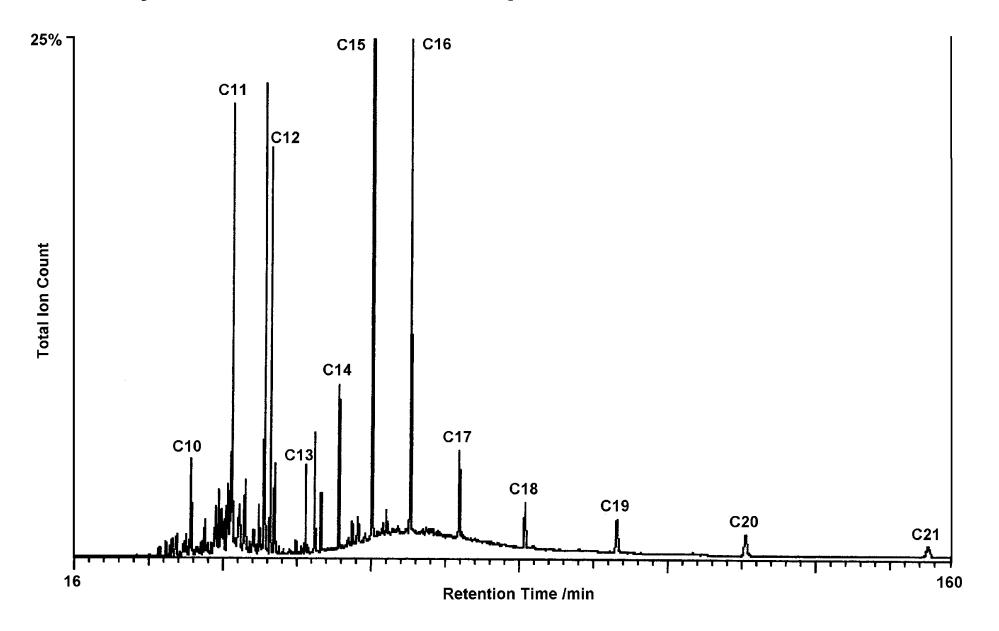
(Dagaut et al., CNRS)

# GC analysis of a GtL sample



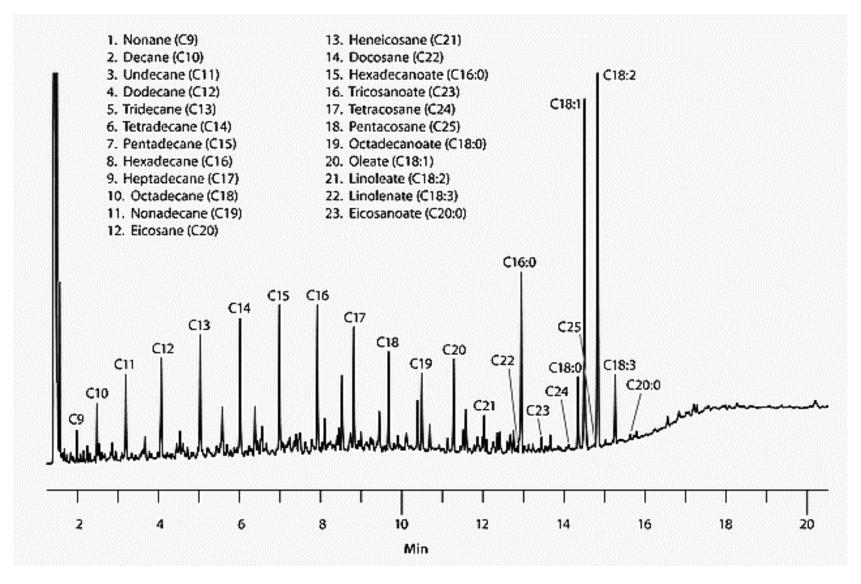
(Egolfopoulos et al., USC)

# GC analysis of a diesel fuel sample



(Dagaut et al., CNRS)

## GC analysis of a B20 diesel fuel sample



(https://www.sigmaaldrich.com)

## **Summary**

What is combustion? The oxidation of a fuel, ultimately leads to the formation of carbon dioxide, water, and heat in the case of organic fuels (e.g. hydrocarbons). Other definition: an exothermic redox reaction between a fuel (reductant) and an oxidant (e.g., oxygen from air). Incomplete combustion yields UHC and soot. NOx resulting from nitrogen oxidation can also be released

Why combustion? Transport accounts for ca. 20% of the total global primary energy consumed, ca. 23% of CO<sub>2</sub> emissions, ca. 7 billion tons of CO<sub>2</sub>, ca. from livestock farming. > 99.9%Transport is powered by I.C. engines (land and marine) and air transport by GT.

Chemical Kinetics and Modeling. Feed and feedback: Experimental data ↔ Model. Global parameters: Ignition delays (initiation reactions, R+O₂), Burning velocities (H fluxes) vs. **Detailed information**: Species concentrations (~ all processes). **Different types of 'reactors'**: ST, PF, PSR, Flames (laminar premixed, opposed flow), RCM, engines; provide complementary data.

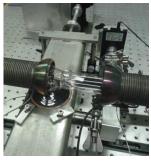
Global fuel properties: Cetane number, Octane number (S.I. engines: ON=100 for iso-octane and ON=0 for n-heptane; C.I. engines: CN=100 for n-hexadecane and CN=0 for 1-methylnaphtalene

Composition of Fuels. **NG**: methane + higher alkanes (ca.  $C_8$ ); **LPG**: region-dependent;  $C_3$ – $C_4$  alkanes and alkenes; **Gasoline**:  $C_4$ – $C_{12}$  hydrocarbons. Mixture of paraffins (alkanes), olefins (alkenes), cycloalkanes (naphthenes), aromatics; **Kerosene** (Jet A-1 fuel),  $C_6$ – $C_{16}$  hydrocarbons. Mixture of paraffins (alkanes), cycloalkanes (naphthenes), aromatics and <2% alkenes; **Diesel**:  $C_6$ – $C_{28}$  hydrocarbons. Mixture of paraffins (alkanes), olefins (alkenes), cycloalkanes (naphthenes), aromatics, naphteno-aromatics; **Additives**: EtOH, ETBE.

# Part 2

# EXPERIMENTAL TECHNIQUES FOR KINETIC MODELS ASSESSMENT











#### 1. Introduction

Chemical kinetic reaction mechanisms for combustion, either hand-written or automatically generated, rely on experimental data obtained over a large range of conditions.

However, combustion is a complex, generally exothermic, phenomenon involving strongly coupled chemical processes (reaction kinetics) and physical processes (diffusion and heat transfer). Thus, in order to better assess chemical kinetic reaction mechanisms, it is preferable to design experiments were the complexity of physical processes is minimized and the accuracy of the data is maximized.

This is the case for ideal reactors such as plug-flow reactors, perfectly stirred reactors, and shocktubes.

In practice, the experiments should be performed under conditions were ideal reactor models can be used, e.g., operating a JSR under highly diluted conditions, under near-isothermal conditions).

Indeed, such kinetic reaction mechanisms need to be validated through extensive comparison of modeling predictions and experimental results obtained under well-defined conditions. A wide range of experimental facilities can provide such data which are usually described as 'global' and 'detailed'. By combining data obtained from several techniques and conditions, one can check their consistency and use them to constrain chemical kinetic reaction mechanisms.

Global data include ignition delay times which can be obtained using shock-tubes, rapid compression machines, or plug-flow reactors, and laminar burning velocities or flame speeds determined using several types of experiments such as spherical flames in combustion vessels, Bunsen burners, stagnation-flow flames, counter-flow flames, or heat-flux burners. Ignition experiments are particularly useful for probing initiation and termination reactions and reactions of molecular oxygen with radicals whereas they are usually less useful for probing the kinetics of propagation reactions involving atoms and radicals.

#### Initiations reactions:

$$RH (+M) \rightleftharpoons R + H (+M), RH (+M) \rightleftharpoons R' + R'' (+M), and RH + O_2 \rightleftharpoons R + HO_2$$

### Propagation reactions:

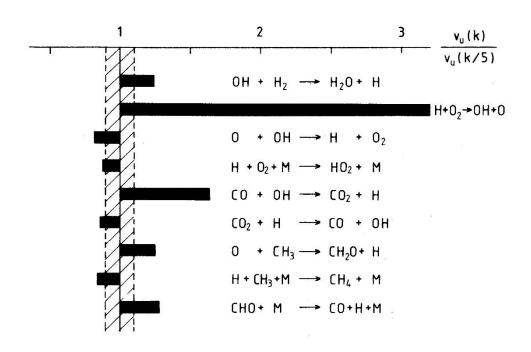
$$RH + X \rightleftharpoons R + HX$$
 (X= H, O, OH, HO<sub>2</sub>, CH<sub>3</sub>, HCO, ...) and radicals reactions, e.g.,

$$R + O_2 \rightleftharpoons R_{-H} + HO_2$$
,  $R \rightleftharpoons$  olefin + R',  $R \rightleftharpoons$  olefin + H

#### Termination reactions:

$$R + H (+M) \rightleftharpoons RH (+M)$$
 and  $R' + R'' (+M) \rightleftharpoons RH (+M)$ .

The paramount importance of H-atoms has been recognized long ago ( $_{\text{Tanford}, C., J. Chem. Phys., 1947. 15(7): p. 433-439.}$ ). Burning velocity experiments are very valuable for probing reactions involving H-atoms such as RH (+M)  $\rightleftharpoons$  R + H (+M) and R (+M)  $\rightleftharpoons$  product + H (+M). Burning velocities are also very sensitive to the main branching reaction in combustion, i.e.,  $\mathbf{H} + \mathbf{O_2} \rightleftharpoons \mathbf{OH} + \mathbf{O}$ 



Sensitivity of computed laminar burning velocity of a methane-air flame at 1 bar and  $T_u = 298 \text{ K}$  to reaction kinetics. From Warnatz, J., The structure of laminar alkane-, alkene-, and acetylene flames. Symposium (International) on Combustion, 18(1), p. 380, 1981.

**Detailed data** are mostly concentration profiles of stable and unstable chemical species observed during the oxidation and combustion of fuels. Many reactors in conjunction with sampling methods and analytical techniques have been used to acquire such data.

Analytical techniques are often used after gas sampling performed using a range of probes (e.g., low-pressure, cooled, molecular beam) or traps (cold trap, bubblers, traps containing absorbents). These probes should stop chemical reactions and transfer a chemical sample to appropriate analyzers without changing its composition. This assumption needs to be verified.

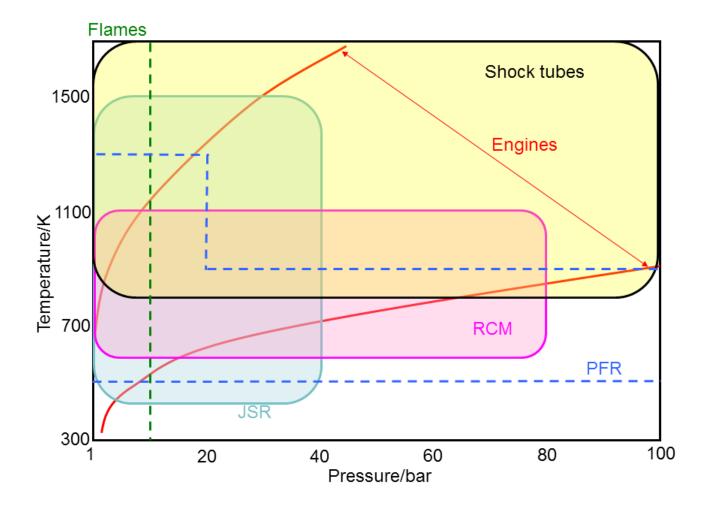
- Low-pressure probes reduce reactions rates by lowering molecular concentrations and temperature after gas expansion.
- Cooled probes reduce reaction rates which are exponentially temperature-dependent, according to the Arrhenius equation.
- Probes are responsible for disturbance of the reaction medium (flow, temperature) which can result in additional complications for interpreting the experimental results.

Many cool traps can be used to collect the condensable compounds at the temperature of the trap (water ice: 273 K;  $CO_2$  dry ice: 194.65 K; liquid nitrogen: 77.2 K). The use of liquid nitrogen traps oxygen ( $O_2$  boiling temperature = 90.2 K) and requires particular care to prevent hazards.  $\triangle$ 

Nowadays, the most **popular experimental techniques** used are flow reactors (jet-stirred reactors, tubular flow reactors), burner stabilized laminar flames (premixed low-pressure flames, opposed flow diffusion flames), and shock-tubes. These techniques by themselves are useful because they cover a wide range of conditions (temperature, pressure, equivalence ratio, initial concentrations, residence time, recirculation rate) allowing to probe the complexity of combustion chemical kinetics. But this is through their coupling to a large range of analytical techniques that one can acquire the data needed to validate detailed kinetic combustion models.

Among these **analytical techniques**, some are very popular whereas others are less frequently used: Gas chromatography (with thermal conductivity detector, flame ionization detector, mass spectrometry), molecular-beam mass spectrometry, Fourier transform infrared spectrometry are commonly used. They are commercially available, reliable, and easy to use.

Other **spectroscopy techniques** are also used in laboratory experiments. They are mostly used to measure radicals, atoms, and unstable molecular species in the UV or the infrared. Recent coupling of synchrotron-sourced photoionization with mass spectrometry allowed very detailed probing of oxidation and combustion processes. Several mass spectrometry techniques are used in laboratory experiments. They mostly differ by the use of different types of mass separation (time-of-flight, quadrupole, ion trap, Orbitrap®), and ionization mode (electronic, chemical, photonic). By combining the above-mentioned laboratory experiments, one can cover a very broad range of conditions relevant to practical applications such as internal combustion engines and gas turbines.



By combining shock-tubes and RCM experiments, one can probe fuels ignition under internal combustion engine conditions. The measurements of burning velocities and flame structures are limited to about 10 bar. Whereas individual reactor experiments have limited operating ranges, by combining them, one can provide detailed data over almost the entire range of pressure and temperature pertinent to I.C.engines and GT.

### 2. Shock-tubes and rapid compression machines

Shock-tubes and RCM are **batch reactors** which can provide both global and detailed combustion data, i.e, ignition delay times and speciation.

These techniques have been used for several decades. In 1890, Vieille started using compression driven shock tubes (Vieille, P., Comptes Rendus de l'Académie des Sciences, 1890. 111 p. 639-641)

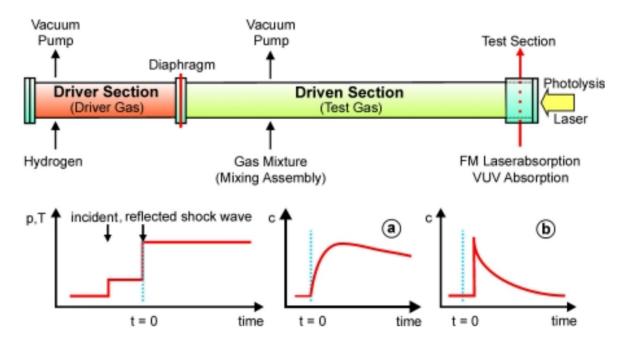
In 1906, Falk used a RCM to determine ignition temperatures (Falk, K.G., J. Am. Chem. Soc., 1906. 28 p. 1517).

Major improvements have been made over the years, allowing the acquisition of very useful global and detailed data for kinetic modelers (Hanson, R.K. and D.F. Davidson, PECS, 2014. 44: p. 103-114; Sung, C.J. and H.J. Curran, PECS, 2014. 44: p. 1-18; Goldsborough, S.S. et al., PECS, 2017. 63: p. 1-78).

## 2.1 Ignition data from RCM and ST.

RCM are limited to the investigation of relatively long ignition delays (5–100's ms) at moderate-T, c.a. 1000 K, and to P < 100 bar, shock-tubes can operate over a wider range of P (up to 100's bar) and to very high-T (1000's K) where ignition delays are rather short (ca. 1–100's ms).

#### **Shock-tube**



## (a) Pressure trace, (b) temperature trace, (c) spectroscopic trace

A shock tube is a several meters long tube with a driver (high-P) and a driven (low-P) section, separated by a diaphragm. The reacting mixture is introduced in the driven section. The driver section is filled with inert gas (He...). After diaphragm breaking, a shock wave forms and propagates downwards the tube at supersonic speed, heating and compressing the reacting mix gas within < 1 µs (incident SW). The SW reflects at the end wall and the preheated reacting mix is heated and compressed again (reflected SW).

https://www.friedrichs.phc.uni-kiel.de/en/research/shock-tube-frequency-modulation-spectroscopystossrohr

## **Ignition traces in ST**

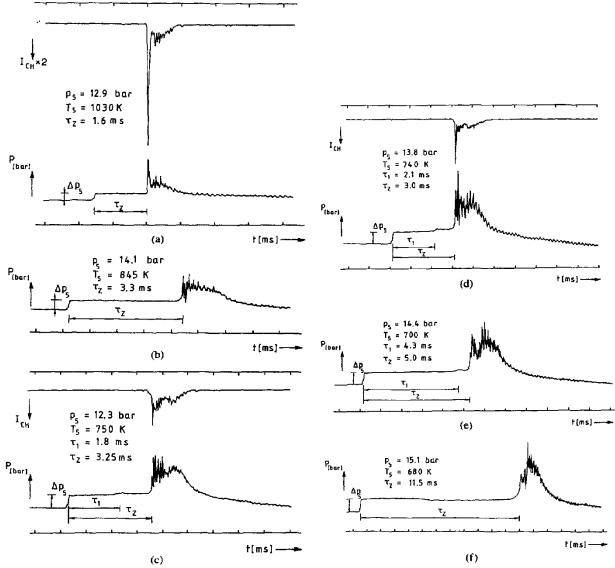
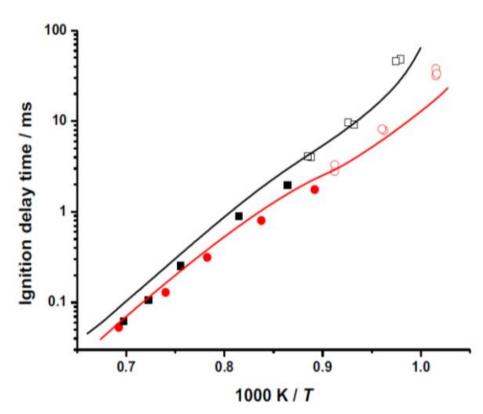


Fig. 1. Pressure and band emission traces of CH under changing influence parameters.

Ciezki and Adomeit, Comb. Flame 93 (1993) 421-433.



Experimental (symbols) and modeled (lines) ignition delay times for a  $\varphi$  = 0.5 NG/air mix measured using a RCM (open symbols) and a shock-tube (closed symbols) at 8–10 atm (black) and 19–20 atm (red). From Sung, C.J. and H.J. Curran, Using rapid compression machines for chemical kinetics studies. Progress in Energy and Combustion Science, 44: p. 10, 2014.

Modelers must be aware of a complication when trying to combine ignition data obtained in a shock-tube and a RCM. At first, they can look irreconcilable. In fact, it is necessary to consider facility-dependent effects before combining ignition delay times measured in shock-tubes and RCMs.

These have been described with great details in several publications and have been reviewed recently (Sung, C.J. and H.J. Curran, PECS, 2014. 44: p. 1-18).

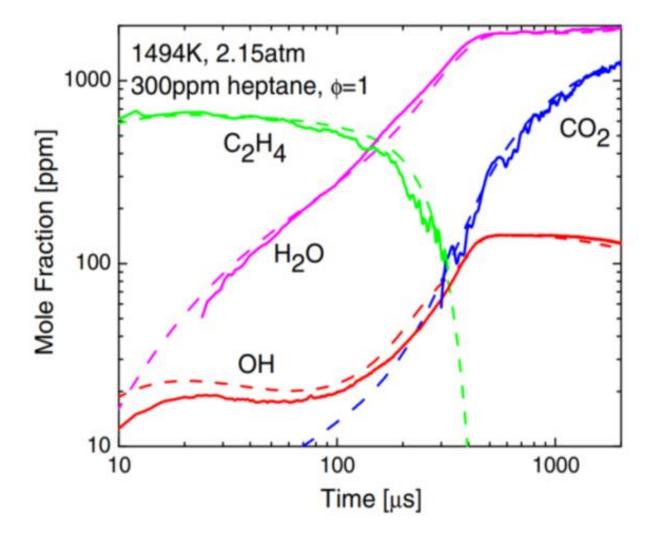
Up to now, a large set of data is available for the ignition of fuels ranging from hydrogen to practical fuels such as jet fuels or biodiesel (Dagaut, P., et al., CNF, 2014. 161(3): p. 835-847; Ramirez-Lancheros, H.P., et al., CNF, 2012. 159(3): p. 996-1008). These data have been extensively used to propose detailed and simple kinetic models.

### 2.2 Species measurements from ST and RCM.

Whereas speciation in shock-tubes has received much attention (studies concern both oxidation and pyrolysis), a more limited database is available from RCM experiments.

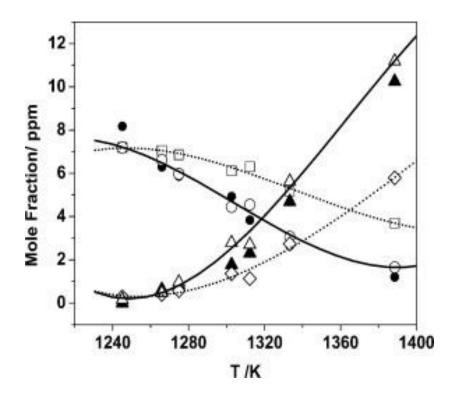
Several research groups have used shock tubes to measure species concentrations using spectroscopy (Hanson, R.K., PROCI, 2011. 33(1): p. 1-40; Roth, P., Forsch. Ing.-Eng. Res., 1980. 46(3): p. 93-102) and gas-chromatography (Tranter, R.S. et al., Rev. Sci. Instr., 2001. 72(7): p. 3046-3054; Tranter, R.S. et al., PCCP, 2002. 4(11): p. 2001-2010).

Hanson and co-workers have recently reported laser-absorption-based measurements in shock-tubes of time-histories of reactants, small-radicals, stable intermediates, and combustion products:



Time-history for stable and labile species measured during the oxidation of n-heptane in a shock-tube (continuous lines) are compared to kinetic modeling (dashed lines). From: Hanson, R.K., Applications of quantitative laser sensors to kinetics, propulsion and practical energy systems. Proceedings of the Combustion Institute, 33(1), p. 10, 2011.

Such data are particularly useful for improving kinetic reactions schemes. This is also true for data coming from single-pulse shock-tube experiments with gas-sampling and GC analyses (Sivaramakrishnan, R. et al., PROCI, 2005. 30(1): p. 1165-1173):



Toluene oxidation at  $\varphi = 1$  and 610 bar in a shock-tube. (•) Experimental data  $C_6H_5CH_3$ ; ( $\blacktriangle$ ) Expt. CO; ( $\Box$ ) KBG model  $C_6H_5CH_3$ ; ( $\diamondsuit$ ) KBG model CO; ( $\odot$ ) STB model  $C_6H_5CH_3$ ; ( $\vartriangle$ ) STB model CO; ( $\odot$ ) fit to KBG model predictions; and (—) fit to STB model predictions. From Sivaramakrishnan, R., R.S. Tranter, and K. Brezinsky, A high pressure model for the oxidation of toluene. Proceedings of the Combustion Institute, 30(1), p. 1169, 2005

More recent developments:

A miniature with high-repetition rate shock-tube was recently introduced by Tranter (*Tranter, R.S. and P.T. Lynch, Rev. Sci. Instr.*, 2013. 84(9): p. 094102) who used it to probe pyrolysis chemistry of dimethyl ether at high temperature (1400 –1700 K) and high pressure (3 –16 bar) with a tunable synchrotron-generated photoionization time-of-flight mass spectrometer (*Lynch, P.T. et al., Analytical Chemistry*, 2015. 87(4): p. 2345-2352). This new set-up opens up new horizons for chemical kinetics.

Data obtained with shock-tubes have been extensively used to propose detailed and simple kinetic models for the oxidation of fuels ranging from hydrogen to large hydrocarbons and practical fuels (gasoline and jet fuel, Zhu, Y. et al., in 53rd AIAA Aerospace Sciences Meeting. 2015; Li, Y., Ph.D., School of Chemistry. 2017, Nat. Univ. of Ireland: Galway; Javed, T. et al., CNF, 2017. 185(Sup. C): p. 152-159).

Species measurements in RCM through gas-sampling started in the 1960's (*Roblee, L.H.S., CNF, 1961.* 5(*Sup. C*): p. 229-234; *Martinengo, A. et al., Symp. (Int.I) Combust., 1965. 10(1): p. 323-330; Fish, A. et al., Proc. Royal Soc. London. A. Math. Phys. Sci., 1969. 313(1513): p. 261*). Several groups have performed such experiments for the ignition of hydrocarbons, alkyl nitrates, and oxygenated fuels. GC has been used in most of RCM experiments; exhaust gas analyzers for CO, CO<sub>2</sub>, NO<sub>x</sub>, and unburned hydrocarbons have also been used (*Ribaucour, M. et al. J. Chim. Phys. Phys.-Chim. Biol., 1992. 89(11-12): p. 2127-2152; Minetti, R. et al., CNF, 1994. 96(3): p. 201-211; Minetti, R. et al., CNF, 1995. 102(3): p. 298-309; Van Blarigan, P. et al., SAE Tech Pap 982484, 1998).* 

**Spectroscopic methods** in the UV and IR have also been used after Fish et al. (*Fish, A. et al., Proc. Royal Soc. London. A. Math. Phys. Sci., 1969. 313(1513): p. 261*). These data have been used to propose detailed and simple kinetic models for the oxidation of fuels (from H<sub>2</sub> to oxygenates and large HC (*Sung, C.J. and H.J. Curran, PECS, 2014. 44: p. 1-18*), but also served to identify the products of low-temperature oxidation of a range of fuels (*Minetti, R et al., CNF, 1994. 96(3): p. 201-211; Minetti, R. et al., CNF, 1995. 102(3): p. 298-309; Walton, S.M. et al., Fuel, 2011. 90(5): p. 1796-1804; <i>Karwat, D.M.A. et al., J. Phys.l Chem. A, 2011. 115(19): p. 4909-4921*).

### Advantages:

Can be run with very little fuel compared to flames and reactors experiments.

A wide range of operating conditions, in terms of P, T, and  $\phi$ , is covered by combining these techniques.

#### Limitations/weaknesses:

Batch reactor experiments are time consuming because they involve mixture preparation, pumping after each ignition experiment, replacement of the shock-tube diaphragm (needing disassembling / reassembling).

Also, pressure history must be well characterized to allow accurate kinetic modeling.

### 3. Flow reactors: Tubular Flow Reactors and Stirred Reactors

Flow reactors are particularly useful for measuring the concentration of reactants, intermediates species, and final products of fuels oxidation or pyrolysis or interaction of fuels with other species, e.g., NOx, SOx, CO<sub>2</sub>, H<sub>2</sub>O.

They usually operate at temperatures below 1500 K and pressure less than 50 bar.

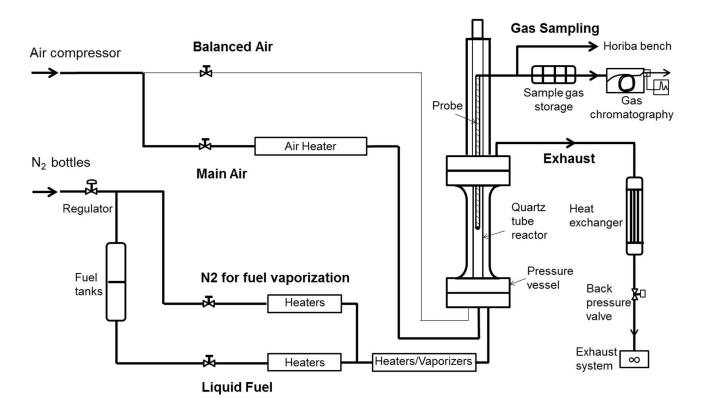
They are particularly useful for studying the low-T oxidation chemistry of fuels. In most of the experiments, high fuel dilution (100–1000's ppm) is used to avoid flame occurrence and large temperature gradients. Nevertheless, experiments are also performed with higher initial fuel concentrations (1–few mole %).

Whereas in **tubular flow reactors**, ideally called plug-flow reactors (PFRs), one can observe chemical reactions along the reactor axis; in jet-stirred reactors the chemical composition is ideally homogeneous. Flow reactors are usually heated by external ovens. Temperature measurements are of great importance for running accurate modeling. In tubular reactors, this means that measurements must be made along the reactors axis.

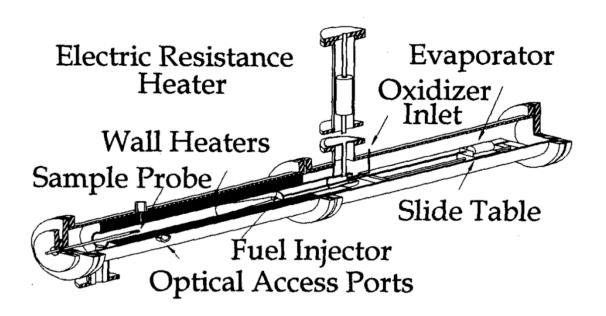
In **JSR**, temperature homogeneity is usually verified along the reactor main axis and measurements used as input in isothermal perfectly stirred reactor model. Compared to flame experiments, flow reactors are not limited to flammability limits. As shock-tubes, they allow studying fuel-lean oxidation to pyrolysis. Although this is not very common, tubular-flow reactors operating under plug-flow conditions can be used to determine ignition delays.

# 3.1 Species measurements.

Two types of flow reactors are mainly used in recent kinetic studies. **Tubular flow reactors** consist of a tube where reactants are injected and heated from the outside. The flow inside the tube can be laminar (*Rasmussen, C.L. et al., IJCK, 2008, 40(8): p. 454-480; Zhang, T.C. et al., J.Phys .Chem. A, 2008. 112(42): p. 10487-10494) or turbulent (Allen, M.T. et al., I.J.C.K., 1995. 27(9): p. 883-909; Kim, T.J. et al., Symp. (Int.) Combust., 1994. 25(1): p. 759-766; Zhewen, L. et al, Meas. Sci. Technol., 2017. 28(10): p. 105902):* 



Schematic of the Melbourne University high-pressure tubular flow reactor that operates up to 50 bar. From Zhewen, L., C. Julien, L. Nicolas, Y. Yi, and J.B. Michael, Measurement Science and Technology, 28(10), 105902, p. 3, 2017.



Schematic of the Princeton variable pressure tubular flow reactor that operates up to 20 atm and ca. 1200 K. From Kim, T.J., R.A. Yetter, and F.L. Dryer, Symposium (International) on Combustion, 25(1), p. 760, 1994.

Whereas most of the currently used PRFs use conventional analytical instruments (e.g., GC, GC-MS, FTIR) to probe the chemistry, molecular-beam mass spectrometry and tunable synchrotron VUV photoionization have been introduced recently (zhang, T.C. et al., J. Phys. Chem. A, 2008. 112(42): p. 10487-10494), opening new horizons for the understanding and validation of chemical kinetic reaction mechanisms.

Several **jet-stirred reactor** (JSR) geometries have been used (spherical, hemispherical, toroidal, near-conical), but the most popular design is a spherical reactor of less than 50 cm<sup>3</sup>. This technique potentially allows operation over a wide range of residence time (from few milliseconds to several seconds), depending on the reactor geometry (David, R. and D. Matras, Can. J. Chem. Eng., 1975. 53(3): p. 297-300). Temperature homogeneity is improved through preheating to a temperature close to the reactor

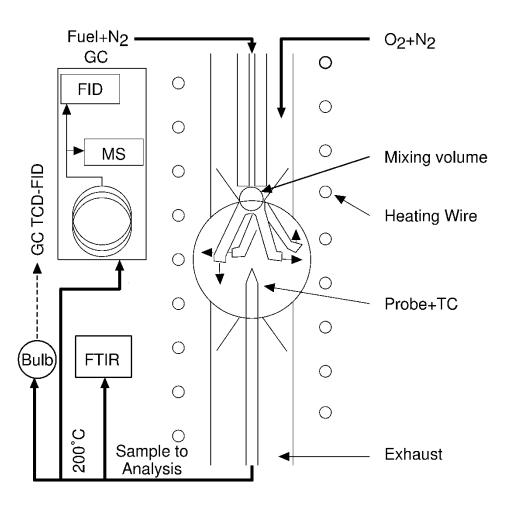
Operating temperature (Dagaut, P. et al., J. Phys. E-Sci. Instr., 1986. 19(3): p. 207-209; Rota, R. et al., Chem Eng Sci, 1994. 49(24A): p. 4211-4221).

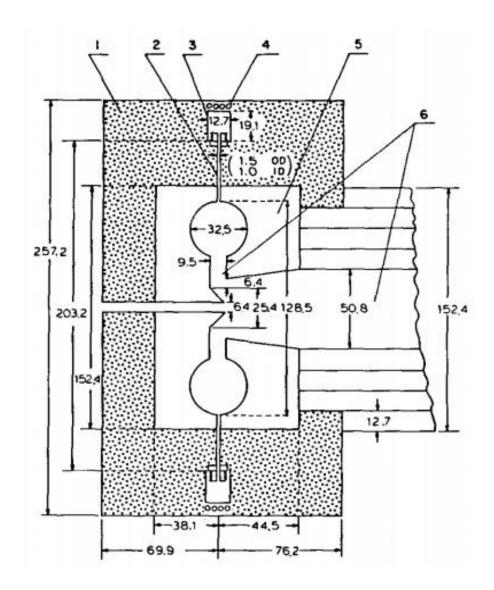
Composition homogeneity was shown to be easier to achieve.



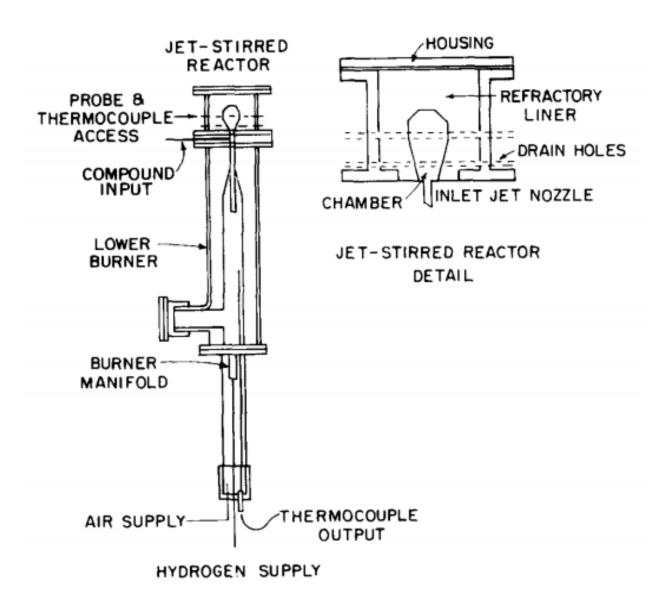
Picture of a fused-silica JSR used at CNRS Orléans. Stirring is provided by 4 injectors. With this reactor, one can operate from 40 ms to 3s.

# JSR set-up





Schematic of the MIT alumina toroidal jet-stirred reactor. Stirring is provided by 32 injectors. From Nenniger, J.E., A. Kridiotis, J. Chomiak, J.P. Longwell, and A.F. Sarofim, Characterization of a toroidal well stirred reactor. Symposium (International) on Combustion, 20(1), p. 474, 1985.

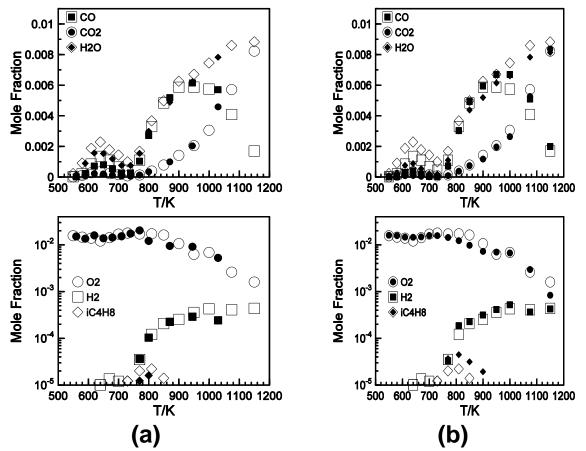


Schematic of the ceramic jet-stirred reactor developed at the University of Washington, Seattle. Stirring is provided by a single injector. From Westbrook, C.K., W.J. Pitz, M.M. Thornton, and P.C. Malte, Combustion and Flame, 72(1), p. 47, 1988.

These reactors have been used to provide useful data for modeling the pyrolysis and oxidation of a wide range of fuels:

hydrogen, ammonia, carbon monoxide, syngas, hydrocarbons, oxygenates, and complex fuels such as gasoline, jet-fuels, Diesel-fuels, synthetic fuels, and biodiesel.

An example of such results is given next for the oxidation of a **conventional jet A-1 and 2 synthetic jet-fuels** (Dagaut, P. et al., CNF, 2014. 161(3): p. 835-847).



Comparison of experimental data obtained from the JSR oxidation of (a) Jet A-1 (closed symbols) and GtL (open symbols) and (b) CtL (closed symbols) and GtL jet fuel (open symbols) at  $\varphi = 1.0$ , 10 bar, and a mean residence time of 1 s. From Dagaut, P., CNF, 2014, 161(3), p. 840.

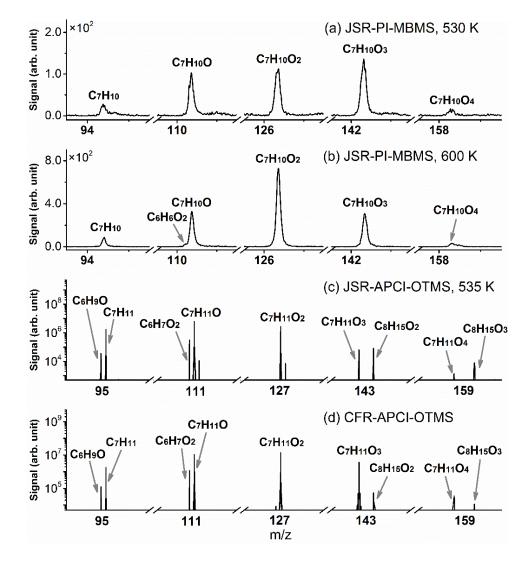
These data show differences in terms of reactivity and formation of intermediate products that can be explained through detailed kinetic modeling (Dagaut, P. et al., CNF, 2014. 161(3): p. 835-847; CST, 2014. 186(10-11): p. 1275-

1283; GT2015-42004 in ASME Turbo Expo 2015; CST 2016. 188(11-12): p. 1705-1718; PROCI, 2017. 36(1): p. 433-440).

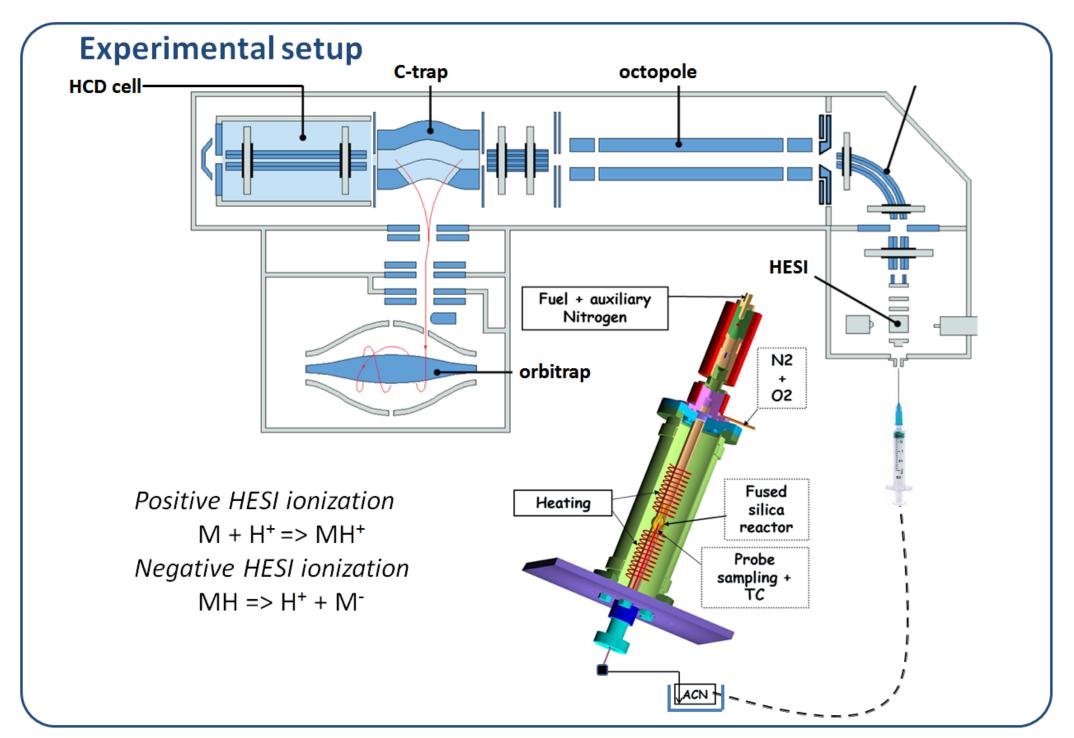
Whereas gas chromatography and FTIR spectrometry are usually used in conjunction with small sonic probes in JSRs experiments to probe the chemistry (Herbinet, O. and G. Dayma, in Cleaner Combustion: Developing Detailed Chemical Kinetic Models, 2013, Springer-Verlag, London), molecular-beam mass spectrometry and tunable synchrotron VUV photoionization have been introduced recently, allowing deeper investigations of combustion chemistry (Battin-Leclerc, F. et al., PROCI, 2011. 33(1): p. 325-331; Moshammer, K. et al., J. Phys. Chem. A, 2015. 119(28): p. 7361-7374; Wang, Z. et al., PNAS, 2017. 114(50): p. 13102-13107).

Recent results have been obtained through the combination of JSRs and high resolution mass spectrometry (Photoionization-MBMS and APCI-Orbitrap MS). They demonstrate that currently accepted reaction schemes for hydrocarbons oxidation are missing reaction pathways leading to the formation of highly oxygenated molecules.

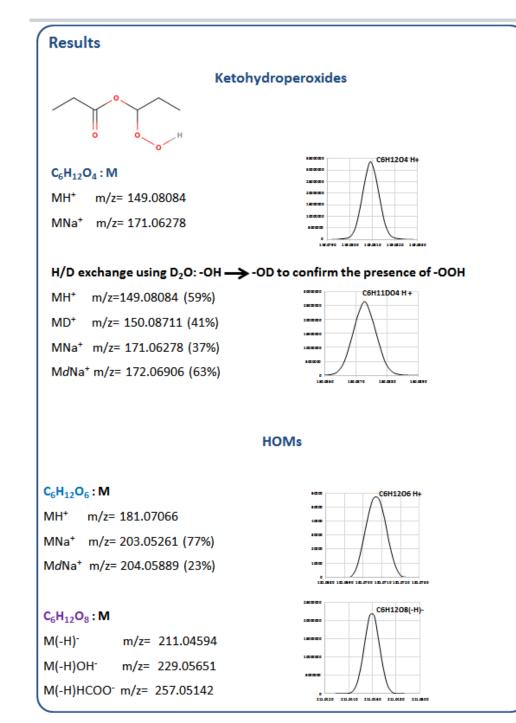
The inclusion of such reactions and products in kinetic scheme could influence significantly model predictions (*Wang, Z. et al., PNAS, 2017. 114(50): p. 13102-13107*).

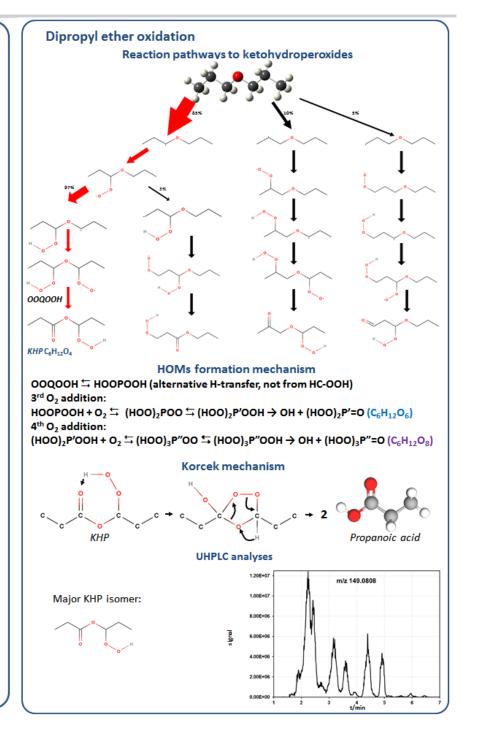


Mass spectra of intermediates with the molecular formula of  $C_7H_{10}O_x$  (x=0 –4). (a) and (b) are for JSR-1 PI-MBMS measurements at T= 530 K and 600 K, respectively. Photon energy is 9.6 eV. (c) is for JSR-2 APCI-OTMS measurements at 535 K. (d) is for CFR engine APCI-OTMS measurements. From Wang, Z.D. et al., Combustion and Flame, 187, Supporting information, p.S5, 2018.



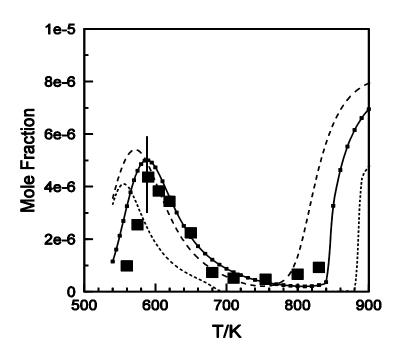
Dagaut et al., Mediteranean Combustion Symposium, 2019





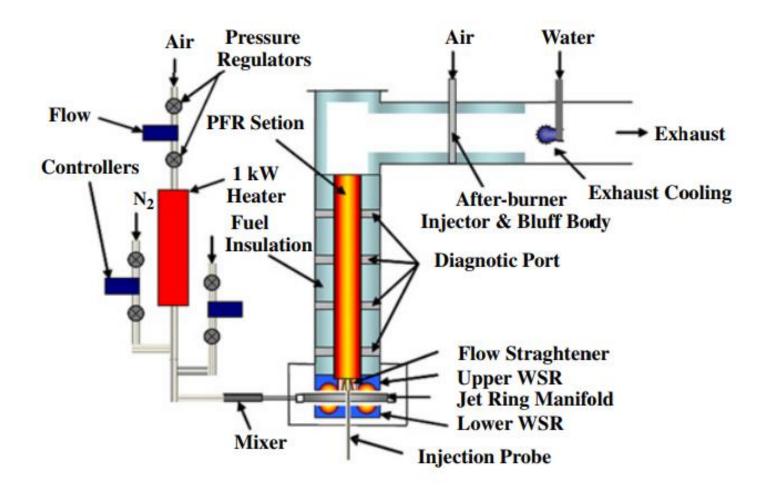
Dagaut et al., Mediteranean Combustion Symposium, 2019

Quantitative measurements using cavity ring-down spectroscopy (CRDS) in the near-IR of HO<sub>2</sub> and H<sub>2</sub>O<sub>2</sub> were reported recently (*Djehiche, M. et al., JACS, 2014. 136(47): p. 16689-16694; Le Tan, N.L. et al., Fuel, 2015. 158: p. 248-252*). The gas mixtures were sampled with a wide angle fused silica nozzle, the tip being located 5 mm inside the reactor. The CRDS cell was kept at low-P (0.3 to 10 mbar), while operating the JSR at 1 atm.



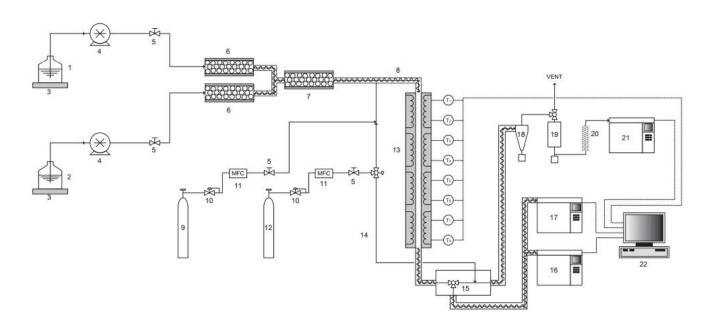
HO<sub>2</sub> concentration profile measured by CRDS during the oxidation of 5000ppm of dimethyl ether in a JSR at an equivalence ratio of 0.5 and a mean residence time of 1.5s. The data (symbols) are compared to simulations using three literature mechanisms. From Le Tan, N.L., M. Djehiche, C.D. Jain, P. Dagaut, and G. Dayma, Fuel, 158, p. 250, 2015.

JSR and PFR have been combined at MIT(Lam, F.W. et al., Symp. (Int.) Combust., 1989. 22(1): p. 323-332) to allow probing combustion chemistry over a wider range of residence times. The original design was further modified at NIST by Lenhert and Manzello



Schematic of the NIST **jet-stirred reactor/plug-flow reactor assembly** inspired from an earlier MIT design(Lam, F.W. et al., Symp. (Int.) Combust., 1989. 22(1): p. 323-332). From Lenhert, D.B. and S.L. Manzello, Proc. Combust. Inst., 32(1), p. 658, 2009.

Both PFRs and JSRs can be pressurized (Rasmussen, C.L. et al., IJCK, 2008. 40(8): p. 454-480; Allen, M.T. et al., IJCK, 1995. 27(9): p. 883-909; Dagaut, P. et al., J. Phys. E-Sci. Instr., 1986. 19(3): p. 207-209). Whereas fused-silica reactors are commonly used, some were built in metal (Lignola, P.G. and E. Reverchon, CST, 1988. 60(4-6): p. 319-333; Ciajolo, A. et al., CST, 1997. 123(n): p. 49-61; Harper, M.R. et al., CNF, 2011. 158(1): p. 16-41; Wada, T. et al., CTM, 2013. 17(5): p. 906-936) and refractory materials (e.g., ceramic or alumina) (Westbrook, C.K. et al., CNF, 1988. 72(1): p. 45-62; Bilbao, R. et al., Proc.. Ind. & Eng. Chem. Res., 1994. 33(11): p. 2846-2852). Whereas fused-silica is generally considered chemically inert in combustion studies, other materials such as metals have catalytic activity that cannot be ignored.



Schematic of the Ghent University Incoloy 800HT tubular flow reactor. From Harper, M.R., K.M. Van Geem, S.P. Pyl, G.B. Marin, and W.H. Green, Comprehensive reaction mechanism for n-butanol pyrolysis and combustion. Combustion and Flame, 158(1), p. 18, 2011.

# Flow reactors advantages:

Operating temperature range and the possibility to investigate pyrolysis to oxidation, whereas flame studies are much more limited.

Reactors are particularly useful for gaining insights into reaction products and intermediates through the use of advanced detection and/or quantification techniques.

Numerous analytical techniques are used after gas sampling achieved using a range of probes for stopping chemical reactions and transferring a chemical sample to the appropriate analyzers. Also, one should be aware of possible complications such as surface reactions.

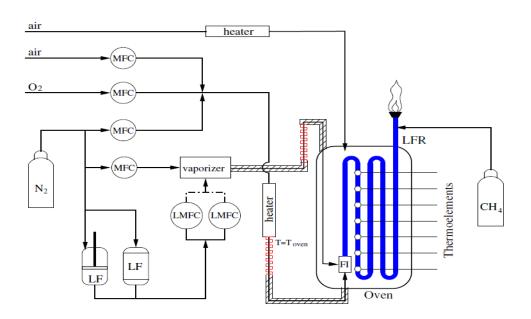
# Flow reactors disadvantages:

Can operate over limited temperature, pressure, and residence time ranges. This is due to material range of use and reachable flow rates.

Experiments **need much larger fuel quantities** compared to shock-tube and RCM experiments. The quantification of intermediate species by photoionization remains limited due to unknown photoionization efficiency difficult to compute using current theoretical methods.

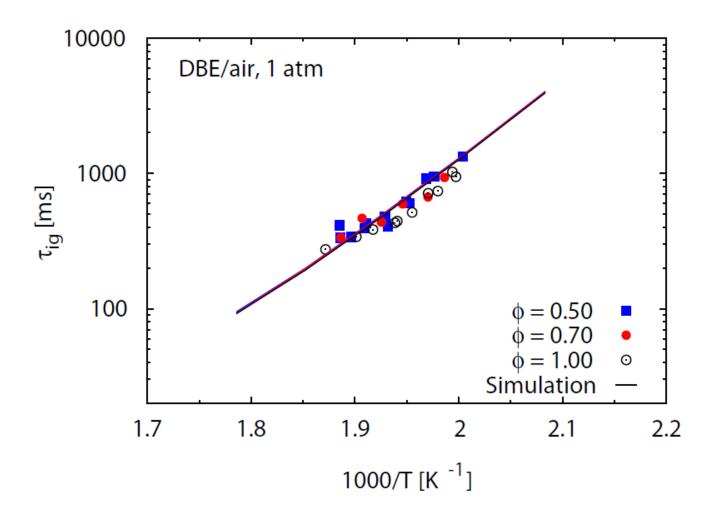
# 3.2 Ignition data from PFR

Ignition delays can also be determined using PFRs. Recently an experimental setup was designed for this purpose (*wada*, *T. et al.,. CTM*, *2013. 17(5): p. 906-936*). The 1st-stage ignition is observed as a temperature increase of a few degrees in the reactor. After the first-ignition, strong heat loss to the reactor wall reduces the temperature and stops chemical reactions. The 1st-stage ignition is determined based on the distance between fuel injection and the location of the first T-rise and the flow rate in the reactor.



Schematic of the Aachen University stainless steel laminar tubular flow reactor. From Cai, L.M., A. Sudholt, D.J. Lee, F.N. Egolfopoulos, H. Pitsch, C.K. Westbrook, and S.M. Sarathy, Combustion and Flame, 161(3), p. 802, 2014.

This set-up was successfully used for measuring first-stage ignition delays of biofuels:



Ignition delay times of dibutyl ether/air mixtures at 1 atm. From Cai, L.M., A. Sudholt, D.J. Lee, F.N. Egolfopoulos, H. Pitsch, C.K. Westbrook, and S.M. Sarathy. Combustion and Flame, 161(3), p. 802, 2014.

#### 4. Flames

Laminar flames are used to obtain both global (laminar burning velocity) and detailed (spatial speciation or flame structure) data usable for validating kinetic models.

Flame experiments are currently performed over a wide range of pressure, from ca.0.04 to 60 bar.

Burning velocities have been obtained from ca. 0.1 to 60 bar whereas flame structures are available up to ca. 10 bar.

Major improvements of the methods have been made over the years, allowing the acquisition of very valuable data for kinetic modelers over a very wide range of conditions and for many fuels (Ranzi, E. et al., PECS, 2012. 38(4): p. 468-501; Egolfopoulos, F.N. et al., PECS, 2014. 43: p. 36-67).

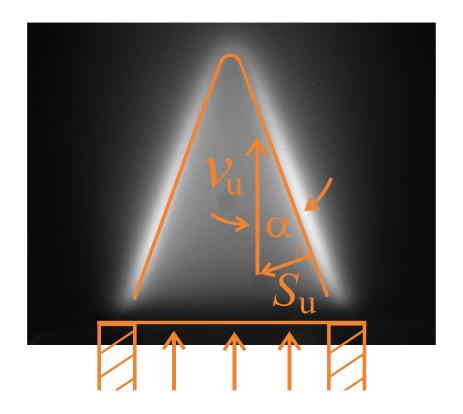
# **4.1 Burning velocities**

The laminar flame speed is defined as the propagation speed of a steady, laminar, one-dimensional, planar, stretch-free, and adiabatic flame.

It is an important fundamental property of a flammable mixture, being a measure of its reactivity, diffusivity, and exothermicity.

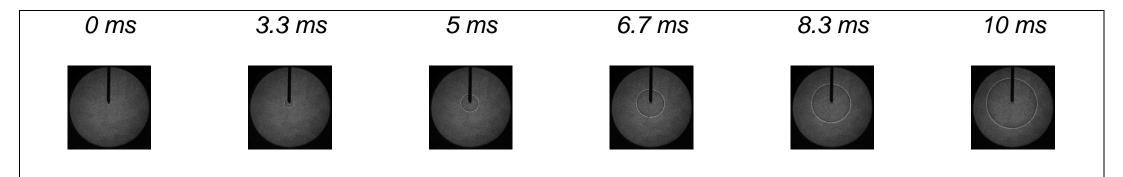
It constitutes an important validation target for kinetic models and a key parameter in turbulent combustion.

Burning velocity can be extracted from a range of experimental configurations, e.g., soap bubble method, flames in tubes, flat flame burner method, conical flames (Bunsen type), heat flux method, spherical flames in constant volume chamber, and stagnation flame/opposed-flow method:



Determination of the burning velocity  $S_u$  by applying the cone angle method ( $S_u=V_u \sin \alpha$ ). From Mzé Ahmed, A., P. Dagaut, K. Hadj-Ali, G. Dayma, T. Kick, J. Herbst, T. Kathrotia, M. Braun-Unkhoff, J. Herzler, C. Naumann, and U. Riedel, The Oxidation of a Coal-to-Liquid Synthetic Jet Fuel: Experimental and Chemical Kinetic Modeling Study. Energy & Fuels, 26(10), p. 6072, 2012.

Spherical flames in constant volume chamber:



Shadowgraphs of the temporal evolution of an ethyl propanoate/air flame front at P = 1 bar,  $T_u = 423$  K and  $\varphi = 0.9$ . The temporal increase of the flame radius is used to compute the stretched laminar burning velocity. The unstretched burning velocity is obtained after extrapolation to zero-stretch using proposed methods in the literature.

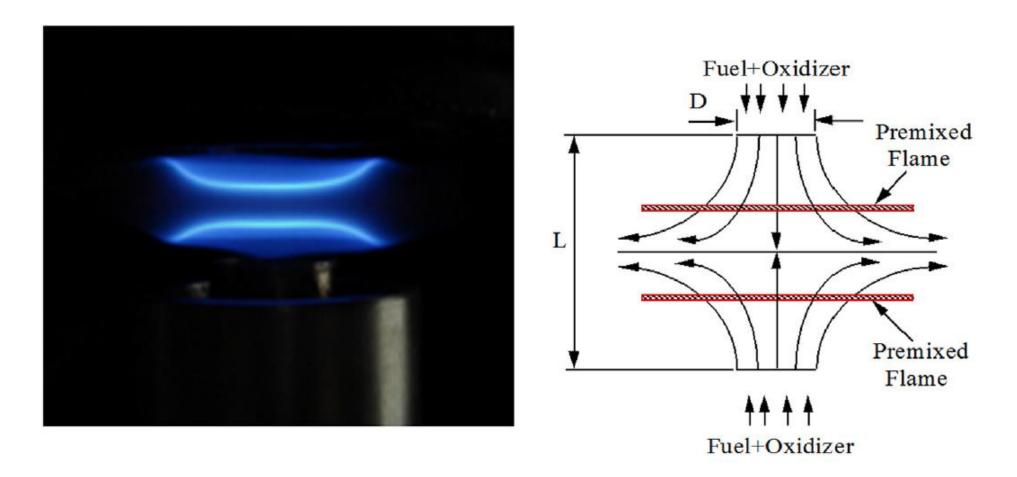
From Dayma, G., F. Halter, F. Foucher, C. Mounaim-Rousselle, and P. Dagaut, Laminar Burning Velocities of C(4)-C(7) Ethyl Esters in a Spherical Combustion Chamber: Experimental and Detailed Kinetic Modeling. Energy & Fuels, 26(11), p. 6670, 2012.

Nowadays spherical flames in constant volume chamber and stagnation flame/opposed-flow method are the most widely used.

They have been reviewed recently (Ranzi et al. PECS, 2012. 38(4): p. 468-501 and Egolfopoulos et al. PECS, 2014. 43: p. 36-67). High pressure and temperature conditions are hardly reachable using Bunsen flames, counter-flow flames or heat flux burner. Most of the results reported at elevated pressures were obtained with spherical expanding flames. One limitation of this method comes from the fact that the spherical flame surface is changing during propagation inducing stretch effects which must be accounted for using extrapolation methods.

Until the work of Wu and Law (*Symp. (Int.) Combust., 1985. 20(1): p. 1941-1949*), undetermined stretch effects led to lots of scatter in measurements.

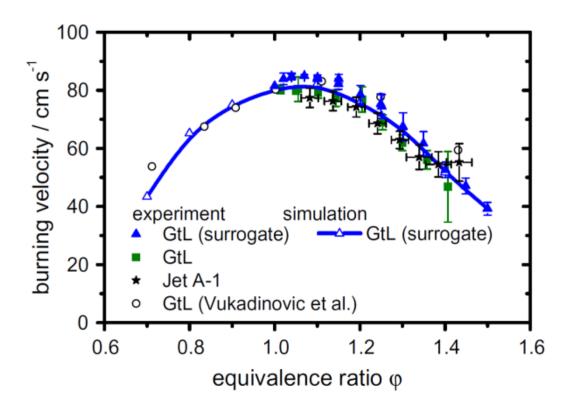
Significant reduction of uncertainty on flame speed measurements has resulted from stretch correction, as outlined by Law (AIAA Journal, 2012. 50(1): p. 19-36) for methane-air flames for which the ±25 cm/s scatter got reduced to ca. 2 cm/s recently by considering the non-linear nature of stretch on burning velocity (Kelley, A.P. and C.K. Law, CNF, 2009. 156(9): p. 1844-1851; Halter, F. et al., CNF, 2010. 157(10): p. 1825-1832). With such low uncertainties, burning velocities are very valuable for kinetic models assessment.



Picture of twin stagnation flames (left) and schematic view (right).

From Egolfopoulos, F.N., N. Hansen, Y. Ju, K. Kohse-Hoinghaus, C.K. Law, and F. Qi, Advances and challenges in laminar flame experiments and implications for combustion chemistry. Progress in Energy and Combustion Science, 43, p. 49, 2014.

Burning velocities for simple to complex fuels have been published. An example of such results is given here for the combustion of synthetic jet-fuels.



Comparison of measured (symbols) and predicted laminar burning velocities of synthetic and conventional jet-fuel-air mixtures at Tu = 473 K and p = 1 bar.

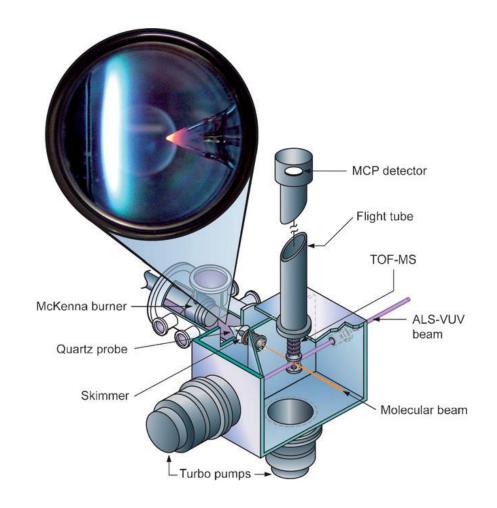
From Dagaut, P., F. Karsenty, G. Dayma, P. Diévart, K. Hadj-Ali, A. Mzé-Ahmed, M. Braun-Unkhoff, J. Herzler, T. Kathrotia, T. Kick, C. Naumann, U. Riedel, and L. Thomas, Experimental and detailed kinetic model for the oxidation of a Gas to Liquid (GtL) jet fuel. Combustion and Flame, 161(3), p. 846, 2014.

# 4.2 Species measurements.

The measurement of flames structure has a long history (*Eltenton, G.C., J. Chem. Phys., 1947. 15(7): p. 455-481; Fristrom, R.M. and A.A. Westenberg, Flame Structure. 1st Ed. 1965: McGraw-Hill. 424*). Nowadays, flame structures mostly come from two methods: low-pressure premixed flat flames and stagnation flames.

- These techniques have been reviewed recently (Egolfopoulos, F.N. et al., PECS, 2014. 43: p. 36-67).
- Samples are extracted from the flame using a probe and sent to analyzers (gas chromatography, mass spectrometry).
- Molecular beam-mass spectrometry has been used extensively.

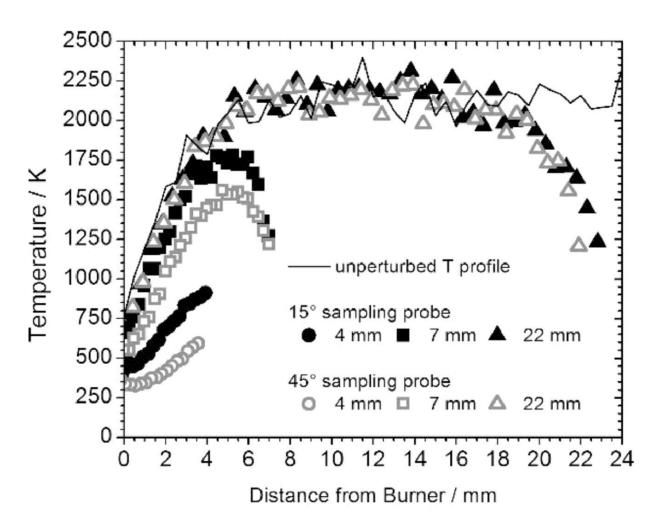
More recently, photoionization by synchrotron-sourced vacuum-ultraviolet radiation was employed, generating a large body of kinetic data unreachable by other techniques (*Egolfopoulos, F.N. et al., PECS, 2014. 43:* p. 36-67; Qi, F. et al., Rev. Sci. Instr., 2006. 77(8): p. 84101; Qi, F., PROCI, 2013. 34(1): p. 33-63; Hansen, N. et al., PECS, 2009. 35(2): p. 168-191; Cool, T.A. et al., J. Chem. Phys., 2003. 119(16): p. 8356-8365; Rev. Sci. Instr., 2005. 76(9); Westmoreland, P.R. et al., Comb. Expl. Shock Waves, 2006. 42(6): p. 672-677):



Schematic of a low-pressure McKenna burner experimental set-up. Gases from the flame are sampled through a fused-silica probe (picture) into a time-of-flight mass spectrometer where chemicals are photo-ionized by synchrotron-generated vacuum-ultraviolet radiation.

From Taatjes, C.A. et al. Physical Chemistry Chemical Physics, 10(1), p. 22, 2008.

Fused-silica probe can cause significant **perturbations** to the flame, making difficult to model and interpret the experiments, as demonstrated in a recent study by Hansen et al. (*CNF*, 2017. 181: p. 214-224, PCI 2019, 37, p.1401).

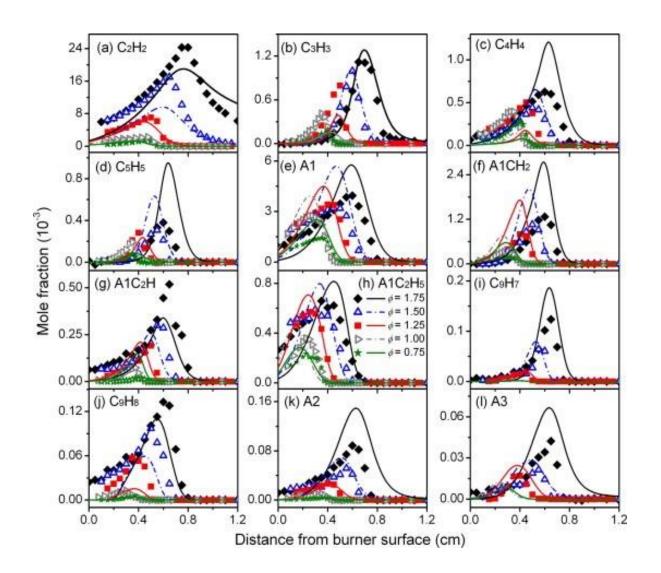


Temperature along the centerline for 3 different sampling positions near 4, 7, and 22 mm for 2 different cone designs. <a href="https://doi.org/10.1016/j.proci.2018.05.034">https://doi.org/10.1016/j.proci.2018.05.034</a>

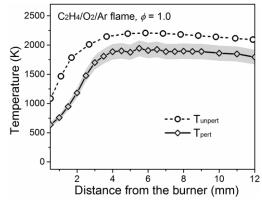
Also, the use of a thermocouple for measuring temperature profiles in the flame can alter the flow fields and temperature profiles (*Skovorodko, P.A. et al., CTM, 2013. 17(1): p. 1-24; CNF, 2012. 159(3): p. 1009-1015*), although these effects are small compared to sampling probe perturbations.

Nevertheless, a wide range of fuels have been studied in flames.

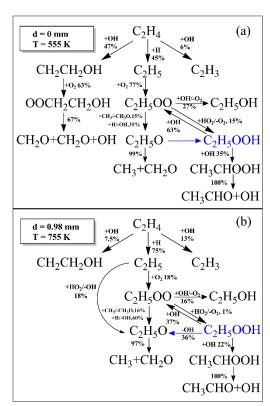
Examples of concentration profiles of C<sub>2</sub>–C<sub>14</sub> species measured in low-pressure premixed flames of toluene next.

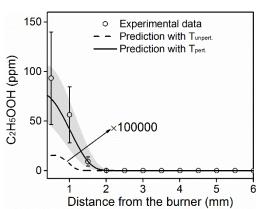


Experimental (symbols) and simulated (lines) mole fraction profiles of acetylene ( $C_2H_2$ ), propargyl ( $C_3H_3$ ), vinylacetylene ( $C_4H_4$ ), cyclopentadienyl ( $C_5H_5$ ), benzene (A1), benzyl (A1C $H_2$ ), phenylacetylene (A1C $H_2$ ), ethylbenzene (A1C $H_3$ ), indenyl ( $H_4$ ), indene ( $H_5$ ), indene ( $H_5$ ), indene ( $H_6$ ), naphthalene (A2) and phenanthrene (A3) in the premixed flames of toluene at five equivalence ratios (0.75 to 1.75). The data were obtained by **MB-MS** with photoionization by synchrotron-sourced vacuum-UV radiation. From Yuan, W.et al.,CNF 162(1), p. 36, 2015.



Temperature profiles in the present **ethylene/O<sub>2</sub>/Ar flame**. Open diamonds and open circles represent the  $T_{pert}$  and  $T_{unpert}$  profiles, respectively. Shadows represent the scaled  $T_{pert}$  profiles considering the uncertainties of maximum  $T_{pert}$  values.





Measured (symbol) and predicted (lines) mole fraction profiles of  $C_2H_5OOH$  in the present **ethylene/O<sub>2</sub>/Ar flame**. The solid and dashed lines represent the predicted results with the  $T_{pert}$  and  $T_{unpert}$  profiles, respectively. Shadows represent the predicted results considering the uncertainties of  $T_{pert}$ .

ROP analysis with the Hashemi model\* by using the  $T_{pert}$  profile at (a) d = 0 mm (T = 555 K) and (b) d = 0.98 mm (T = 755 K).

From Xiaoyuan Zhang et al. Comb. Flame 204 (2019) 260–267

<sup>\*</sup>H. Hashemi, J.G. Jacobsen, C.T. Rasmussen, J.M. Christensen, P. Glarborg, S. Gersen, M. van Essen, H.B. Levinsky, S.J. Klippenstein, High-pressure oxidation of ethane, Combust. Flame 182 (2017) 150-166.

Whereas time-of-flight mass spectrometry with photoionization by synchrotron-generated vacuumultraviolet radiation are very useful for detecting intermediate species, the differentiation between isomers can be difficult when photoionization energies are too close.

Dias et al. (CST, 2004. 176(9): p. 1419-1435) have introduced a useful method consisting of a conventional EI-MBMS setup where a portion of the sample is sent to a GC-MS through a capillary, allowing separation of isomers of stable products that could not be differentiated based on their ionization energies or mass.

Other workers also combined EI-MBMS measurements with GC-MS measurements to get better characterization of isomers (Bourgeois, N. et al., PROCI, 2017. 36(1): p. 383-391).

#### 5. Some conclusions and perspectives

The most common experiments for kinetic mechanism assessment have been presented. It was shown that shock-tube and RCM are very useful for determining fuel ignition properties but also to measure chemical products. Recent advances in CFD modeling of RCM (Bourgeois, N. et al., PROCI, 2017. 36(1): p. 383-391; CNF, 2018. 189: p. 225-239) are expected to facilitate the use of RCM ignition data for kinetic model validation. Tubular flow reactors and jet-stirred reactors are commonly used. Their coupling with advanced analytical techniques is able to provide unique data for kinetic models assessment. However, current limitations due to unknown photoionization efficiency for many intermediates must be addressed, possibly through the use of advanced theoretical methods. Flames can also provide valuable data in terms of burning velocities and speciation, although, limiting perturbations by conventional large sampling probes remains a major challenge for future work.

Momenclature: APCI-OTMS: Atmospheric pressure chemical ionization-Orbitrap® mass spectrometry; CFD: Computational fluid dynamics; CFR: Cooperative Fuel Research; CRDS: Cavity ring-down spectroscopy; CtL: Coal-to-liquid; EI-MBMS: Electron ionization molecular beam-mass spectrometry; FTIR: Fourier-transform infrared; GC: Gas chromatography; GC-MS: Gas chromatography-mass spectrometry; GtL: Gas-to-liquid; IR: Infrared; JSR: Jet-stirred reactor; PFR: Plug-flow reactor; PI-MBMS: Photoionization molecular beam-mass spectrometry; PSR: Perfectly-stirred reactor; RCM: Rapid compression machine; ST: Shock-tube; T<sub>u</sub>: temperature of fresh gas; UV: Ultraviolet; VUV: Vacuum ultraviolet; φ: equivalence ratio.

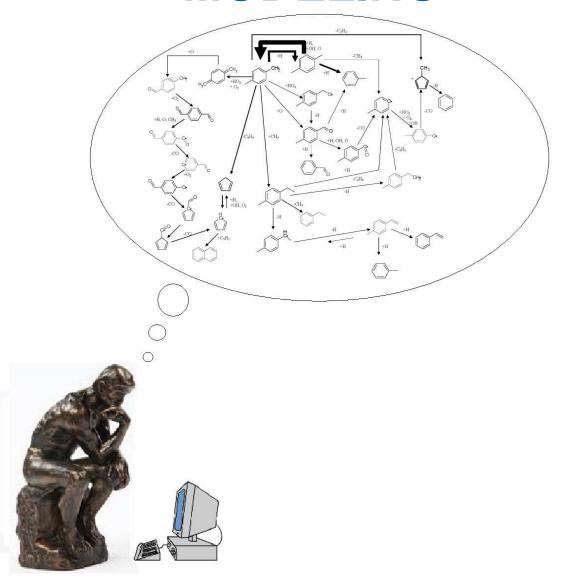
#### **Summary**

Shock-tubes and rapid compression machines. Shock-tube and RCM are very useful for determining fuel ignition properties but also to measure chemical products.

Flow reactors: Tubular Flow Reactors and Stirred Reactors. Commonly used. Their coupling with classical (GC, MS, FTIR) and advanced analytical techniques is able to provide unique data for kinetic models' assessment.

Flames. They provide valuable data in terms of burning velocity and speciation. Beware of probe perturbations.

# Part 3 MODELING



#### **Modeling: General information**

Need accurate kinetics, thermochemistry, and transport data

Use inputs from theory and measurements and also estimations by analogy, tabulations

Need accurate data that are used to constrain the model

## Modeling

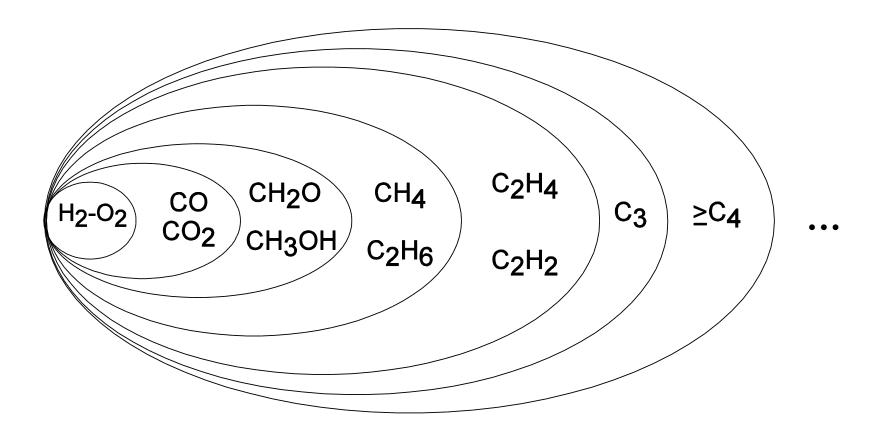
Chemkin computer package.

Kinetic reaction mechanism with modified Arrhenius equation,  $k=A T^b \exp(-E/RT)$ ; k(P,T).

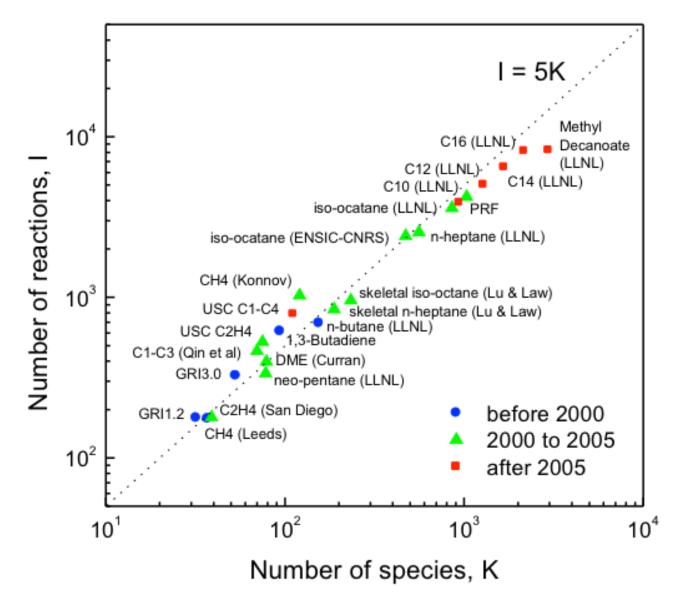
Reaction mechanism with strong hierarchical structure.

The core-mechanism is  $H_2/O_2$  (H, O, OH, HO<sub>2</sub>, H<sub>2</sub>O<sub>2</sub>, O<sub>2</sub>, O<sub>3</sub>, H<sub>2</sub>).

## Modeling: Hierarchical structure of chemical kinetic schemes



## Modeling: Size of chemical kinetic schemes



from T.F. Lu, C.K. Law, PECS 35 (2009) 192-215

#### **Modeling: reaction scheme**

#### REACTIONS k=A\*T\*\*n\*exp(-E/RT)

```
A/cc,mole,s n E/cal/mol Ref
     H+M=H2+
H+
                        7.310E+17 -1.00
                                          0.0
                                               !(BAULCH 76)
                   М
    O + M = O2 +
                        1.140E+17 -1.00
                                          0.0
                                               !(BAULCH 76)
0+
                   М
    H+ M = OH+
                        6.200E+16 -0.60
                                               !(DIXON-LEWIS 81)
                   М
                                          0.0
0+
H2+ O2 = OH+
                  OH
                                               !(MILLER 77)
                        1.700E+13 0.00 47780.0
0+
    H2
         = OH+
                  Н
                        3.870E+04 2.70 6260.0
                                               !GRI
H+
    O2
         = OH+
                        4.400E+14 -0.12 16812.0
                                               !Nicolle 2004
    O2+ M = HO2+
                                               !(WARNATZ 84)
H+
                    М
                        8.000E+17 -0.80
                                          0.0
    OH+ M = H2O+
                    М
                        8.615E+21 -2.00
                                          0.0
                                               !(BAULCH 76)
H+
H2+ OH = H2O+
                   Н
                        2.161E+08 1.51 3430.0
                                               !(MICHAEL 88)
H2O+
       O = OH+
                        1.500E+10 1.14 17260.0
                                               !(WARNATZ 84)
                   OH
HO2+
      OH = H2O+
                    O2
                        2.890E+13 0.00
                                       -497.0
                                               !(KEYSER 88)
HO2+
       O = OH+
                   O2
                                       -400.0
                                               !(JPL 87-41)
                        1.810E+13 0.00
H+ HO2 = H2+
                  02
                        4.280E+13 0.00 1411.0
                                               !(94BAU/COB)
H+ HO2 = OH+
                   OH
                                        874.0
                                               !(94BAU/COB)
                        1.690E+14 0.00
H+ HO2
        = H2O+
                   0
                        3.010E+13 0.00 1721.0
                                               !(BAULCH 92)
HO2+ HO2 = H2O2+
                     O2 4.075E+02 3.32 1979.0
                                               !(HIPPLER 90)
OH + OH (+M) = H2O2 (+M)
                        7.224E+13 -0.37
                                          0.0
                                               !(94BAU/COB)
H2O2+ OH = HO2+ H2O 5.800E+14 0.00 9557.0
                                               !(92HIP/TRO)
H2O2+ H = HO2+
                   H2
                        1.700E+12 0.00
                                       3750.0
                                               !(BAULCH 72)
        H = H2O+
H2O2+
                   OH
                        1.000E+13 0.00
                                       3590.0
                                               !(WARNATZ 84)
        O = HO2+
                    OH
                        2.800E+13 0.00
                                               !(ALBERS 71)
H2O2+
                                       6400.0
```

#### **Modeling: thermochemistry**

```
COMPOSITION
                       ELEMENT
                                                       Phase LOWER-T HIGHER-T MID-T
                                          00
                                    10
                                                 00
                                                             300.00
                                                                                     1000.00
H
                                                       0G
                                                                         5000.00
 0.25000000E+01 0.00000000E+00 0.0000000E+00 0.0000000E+00 0.0000000E+00
 0.25471600E+05-0.46000000E+00 0.25000000E+01 0.00000000E+00 0.0000000E+00
 0.00000000E+00 0.00000000E+00 0.25471600E+05-0.46000000E+00
H2
                                    20
                                          00
                                                 00
                                                       0G
                                                             300.00
                                                                         5000.00
                                                                                     1000.00
                                                                                                      1
 0.29914200E+01 0.70006000E-03-0.56340000E-07-0.92300000E-11 0.15800000E-14
-0.83500000E+03-0.13550000E+01 0.32981200E+01 0.82494000E-03-0.81430000E-06
                                                                                                      3
-0.94750000E-10 0.41349000E-12-0.10125000E+04-0.32940000E+01
                                    10
                                          00
                                                 00
                                                       0G
                                                             300.00
                                                                         5000.00
                                                                                                      1
                                                                                     1000.00
0
 0.25420600E+01-0.27550000E-04-0.31000000E-08 0.45500000E-11-0.44000000E-15
 0.29230800E+05 0.49200000E+01 0.29464300E+01-0.16381700E-02 0.24210300E-05
-0.16028400E-08 0.38907000E-12 0.29147600E+05 0.29640000E+01
02
                                          00
                                                 00
                                                       0G
                                                             300.00
                                                                         5000.00
                                    20
                                                                                     1000.00
 0.36975800E+01 0.61352000E-03-0.12588000E-06 0.17750000E-10-0.11400000E-14
                                                                                                      2
-0.12339000E+04 0.31890000E+01 0.32129400E+01 0.11274900E-02-0.57562000E-06
                                                                                                      3
 0.13138800E-08-0.87686000E-12-0.10052000E+04 0.60350000E+01
                                                                                                      4
(a_{1,k}...a_{7,k}) to calculate thermodynamics over the range 1000 - 5000 K and (a_{8,k}...a_{14,k}) over the range 300 - 1000 K.
                  \frac{C_{p,k}}{R} = a_{1,k} + a_{2,k}T + a_{3,k}T^2 + a_{4,k}T^3 + a_{5,k}T^4 + \frac{H_k^0}{RT} = a_{1,k} + a_{2,k}\frac{T}{2} + a_{3,k}\frac{T^2}{2} + a_{4,k}\frac{T^3}{4} + a_{5,k}\frac{T^4}{5} + a_{6,k}\frac{1}{T}
                  \frac{S_k^0}{R} = a_{1,k} \ln T + a_{2,k} T + a_{3,k} \frac{T^2}{2} + a_{4,k} \frac{T^3}{2} + a_{5,k} \frac{T^4}{4} + a_{7,k}
```

$$G^{0} = H^{0} - TS^{0} , \quad \Delta G^{0} = G^{0}_{prod} - G^{0}_{react} = -RT \ln(K_{p}) , \quad K_{c} = \frac{k_{for}}{k_{rev}} = K_{p} \left(\frac{P}{RT}\right)^{\sum V_{prod} - \sum V_{react}}$$

## **Modeling: transport**

SPECIES	STRUCTURE	L-J POTENTIAL WELL	L-J COLLISION DIAM.	DIPOLE MOMENT	POLARIZABILITY	ROTATIONAL RELAX COLL NBR
		ε/k	σ	μ	α	Z rot
0	0	80.000	2.750	0.000	0.000	0.000
02	1	107.400	3.458	0.000	1.600	3.800
OH	1	80.000	2.750	0.000	0.000	0.000
H20	2	572.400	2.605	1.844	0.000	4.000
H202	2	107.400	3.458	0.000	0.000	3.800

**Structure: 0= atom; 1= linear; 2= non-linear** 

Very good source for the transport properties and their estimates in R. C. Reid, R. C., J. M., Prausnitz, B. E., Poling *The properties of Gases and liquids*, 4th ed, McGraw-Hill, New York, 1987.

CHEMKIN details: R. J. Kee, J. Warnatz, M. E. Coltrin, and J. A. Miller, *A FORTRAN computer code package for the evaluation of gas-phase, multicomponent transport properties*, Sandia Report 86-8246.

#### **Example: OH + dimethyl ether**

OH radicals produced by photolysis of H<sub>2</sub>O and monitored by fluorescence at 310 nm

At low radical concentrations, the OH fluorescence is directly proportional to the OH concentration and the first-order rate expression can be integrated to obtain

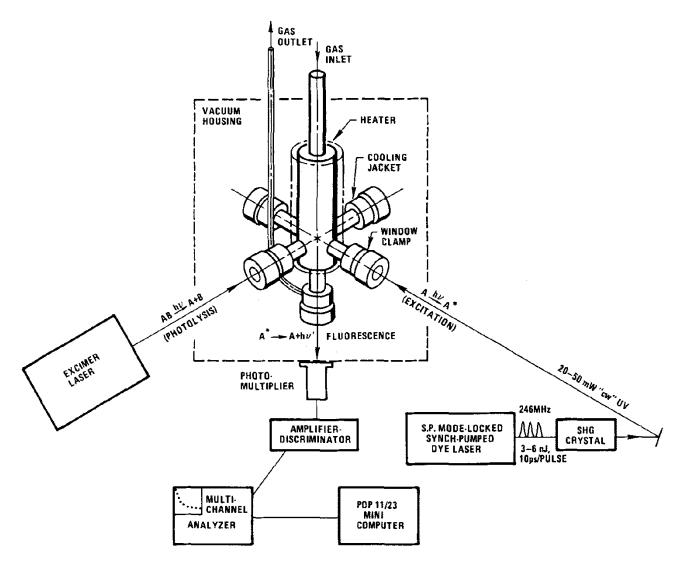
$$F_t = F_{t0} \exp\{-k^{1st}(t - t_o)\} = F_{t0} \exp\{-(k_o + k_r[R]) (t - t_o)\}$$

where  $F_t$ , and  $F_{t0}$  are the OH radical fluorescence intensities at times t and  $t_0$ , respectively,  $k^{1st}$  is the **total** first-order decay rate,

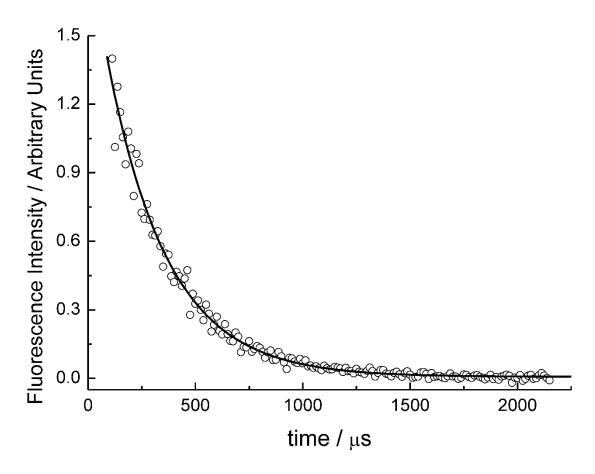
k<sub>0</sub> is the first-order rate constant for OH removal in the absence of reactant (attributed to diffusion out of the viewing zone and reaction with possible impurities in the diluent gas)

k<sub>r</sub>, is the bimolecular rate constant for the reaction of OH with the reactant, R.

#### Rate constants measurements: experimental set-up



Tully et al., Twentieth Symposium (International) on Combustion/The Combustion Institute, 1984/pp. 715-721



Values of k<sup>1st</sup> are determined for various reactant concentrations by non-linear least-squares exponential analysis of the experimental OH fluorescence decay curves and ranged from 20-2000 s<sup>-1</sup>.

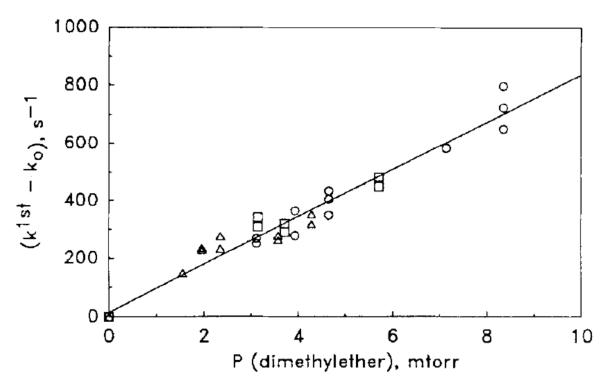


Figure 1. Plot of  $(k^{1st} - k_0)$  vs. dimethylether concentration at 296 K,  $(\triangle)$  25 torr,  $(\Box)$  37.5 torr,  $(\bigcirc)$  50 torr. The line represents a linear least-squares analysis. Wallington, Liu, Dagaut, Kurylo, Int. J. Chem. Kinet. 20 (1988) 41-49

$$k^{1st} - k_0 = f([DME]); slope => k_r$$

Experiments are repeated at several temperatures to obtain the variation of the rate constant versus temperature.

**Example: OH+neopentane** 

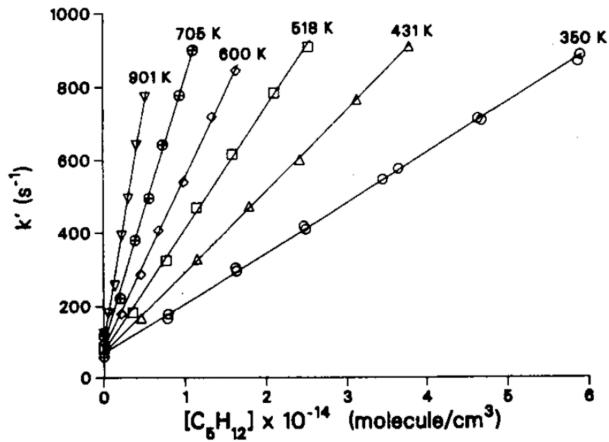


Fig. 2. First-order decay constant k' as a function of  $[C_5H_{12}]$  at various experimental temperatures.

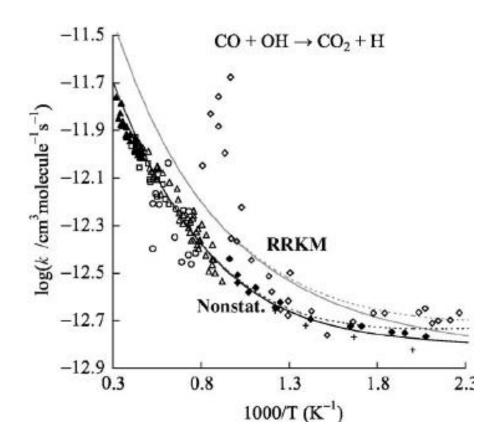
Tully et al., Twentieth Symposium (International) on Combustion/The Combustion Institute, 1984/pp. 715-721

## Modeling: Temperature dependencies of elementary reactions

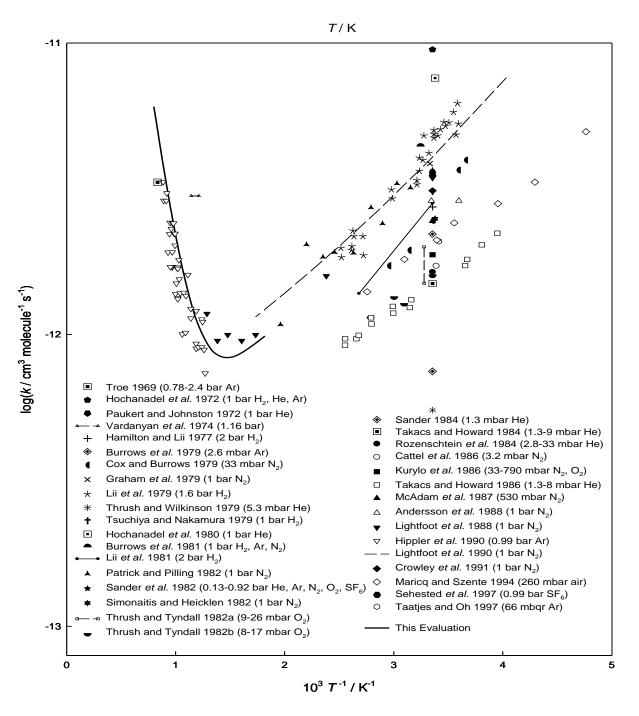
In 1889, Svante Arrhenius proposed the Arrhenius equation from direct observations of the plots of

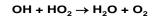
rate constants vs. temperature:  $k=A \exp(-Ea/RT)$ 

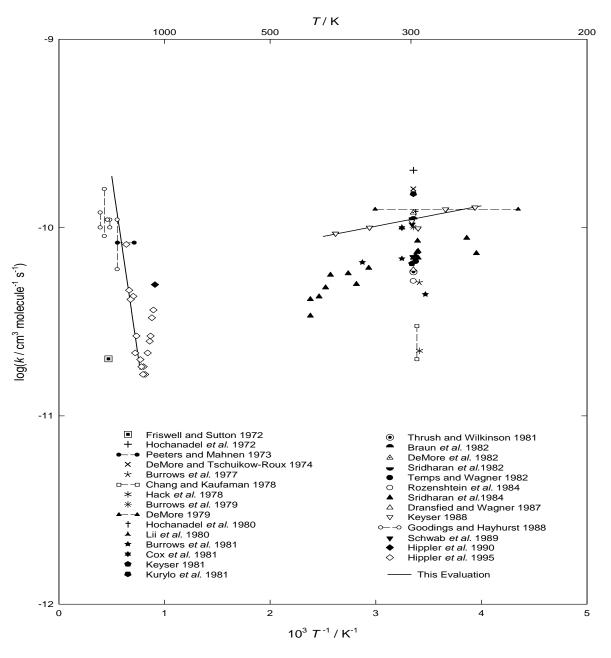
Later, modified Arrhenius expression:  $k=A T^n \exp(-Ea/RT)$ 



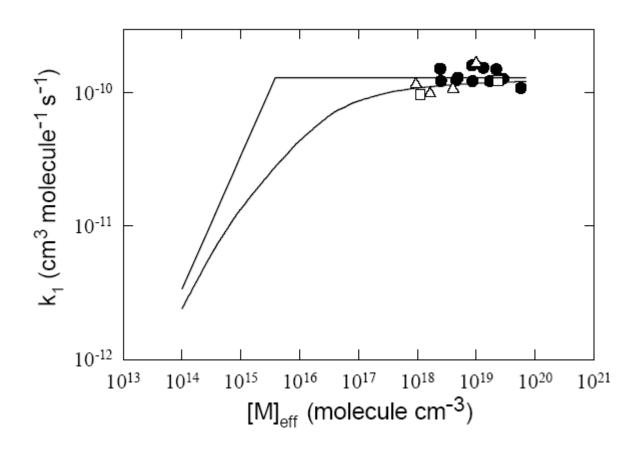
#### $HO_2 + HO_2 \rightarrow H_2O_2 + O_2$

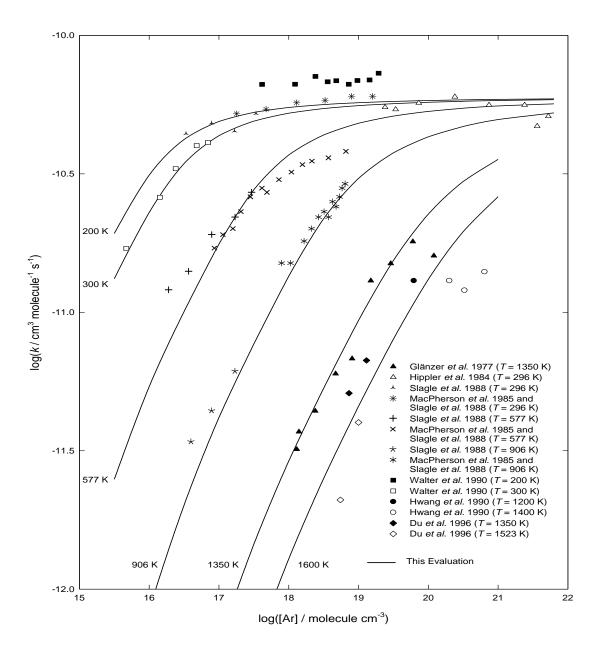






## **Modeling: Pressure dependencies**





## Modeling: Pressure dependencies

Lindemann-Hinshelwood (1/2)

Assume every collision leads to stabilization (M: collision partner)

$$A + M \longrightarrow A^* + M \qquad k_1(T)$$

$$A^* + M \rightarrow A + M \qquad k_2(T)$$

$$A^* \longrightarrow product \qquad k_3(T)$$

The quasi-steady state approximation (QSSA) for A\*: d[A\*]/dt=0

**Steady state for [A\*]**: 
$$[A^*] = k_1 [A] [M] / (k_2 [M] + k_3)$$

$$Rate = k_3 [A^*] = k_3 k_1 [A] [M]/(k_2[M] + k_3) = k_{uni} [A]$$

**High Pressure limit** ([M]  $\rightarrow \infty$ ,  $k_2$  [M]  $>> k_3$ ):

$$Rate= k_3 [A^*] = k_3 k_1 [A] \frac{[M]}{(k_2[M] + k_3)} = k_{uni} [A]$$

$$Rate = k_1 k_3 [A] / k_2 = k_{uni} [A] = k_{uni} = k_1 k_3 / k_2$$

#### **Modeling: Pressure dependencies**

Lindemann-Hinshelwood (2/2)

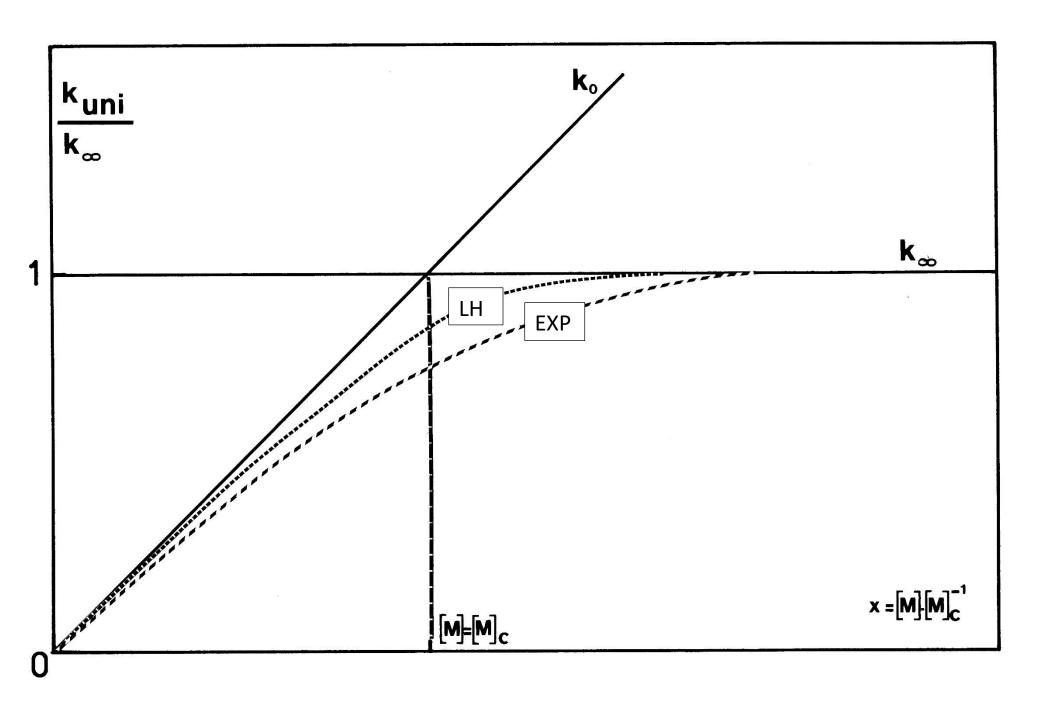
Steady state for [A\*]: [A\*] = 
$$k_1$$
 [A] [M] / ( $k_2$  [M] +  $k_3$ )

Rate=  $k_3$  [A\*] =  $k_3$   $k_1$  [A] [M]/( $k_2$ [M] +  $k_3$ ) =  $k_{uni}$  [A]

Low Pressure limit ([M]  $\rightarrow$  0,  $k_2$  [M]  $<< k_3$ ):

$$Rate= k_3 [A^*] = k_3 k_1 [A] [M]/(k_2[M] + k_3) = k_{uni} [A]$$

$$Rate= k_1 [A] [M] = k_0 [A]; => k_{uni} = k_1 [M]$$



Troe fitting: improved fit (LH too far from exp.)

$$k(T,p) = \frac{k_0[M]k^{\infty}}{k^{\infty} + k_0[M]}F$$

$$\log_{10} F = \frac{\log_{10} F_{cent}}{1 + \left[\frac{\log_{10} (p^*) + c}{N - d(\log_{10} (p^*) + c)}\right]^2}$$

$$d = 0.14$$

$$p^* = k_0[M]/k^{\infty}$$

$$c = -0.4 - 0.67 \log_{10} F_{cent}$$

$$N = 0.75 - 1.27 \log_{10} F_{cent}$$

$$F_{cent} = (1 - a) \exp(-T/T^{***}) + a \exp(-T/T^{*}) + \exp(-T/T^{**}) + \exp(-T/T^{**})$$

#### Troe Formalism in CHEMKIN format

$$F_{cent} = (1 - a) \exp(-T/T^{***}) + a \exp(-T/T^{*}) + \exp(-T/T^{**})$$

CHEMKIN uses 3- or 4-Troe parameters (in the order: a, T\*\*\*, T\*, T\*\*)

### Example:

OH + OH (+M) = H2O2 (+M) 7.224E+13 -0.37 0.0 !(94BAU/COB) High-Pres rate cst

LOW / 2.211E+19 -0.76 0.0/ low-P rate cst

TROE / 0.5 1.0E+08 1.0e-06/ a, T\*\*\*, T\* (T\*\* not used here)

## **Modeling: Kinetic analyses**

#### 1-Reaction pathways

How reactions proceed?

How reactants and intermediates are consumed?

How products are formed?

## ROP(Product1):

reaction rate (R1)/(sum of reaction rates yielding Product1)

## ROC(Product1):

reaction rate (R1)/(sum of reaction rates consuming Product1)

Net rate of production= (total rate of production) – (total rate of consumption)

#### What are the important routes for NO-reduction?

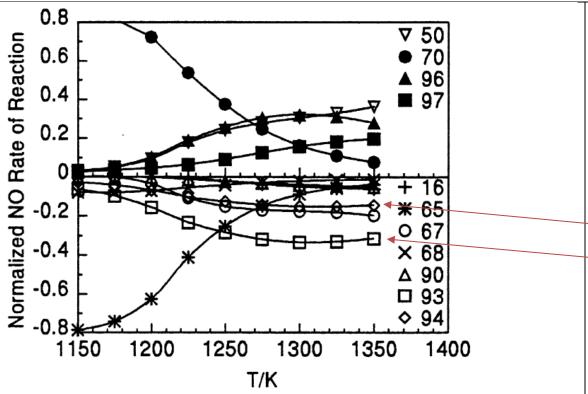


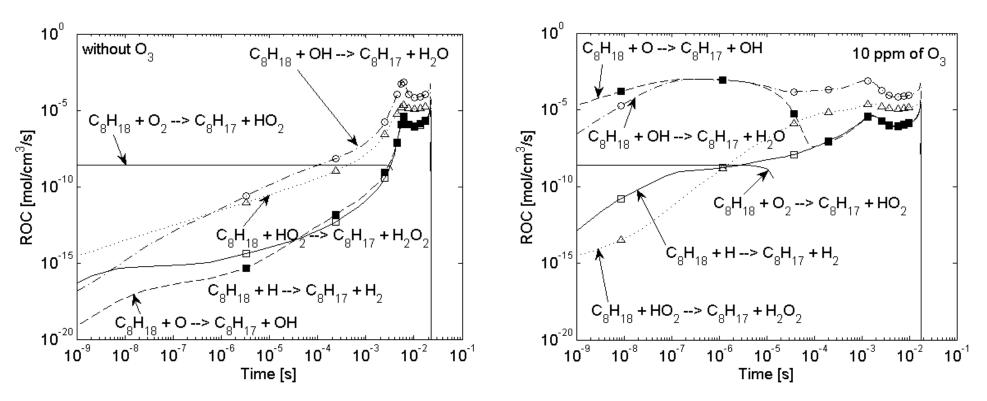
Fig. 3. The influence of temperature on the main reaction paths involved in the reduction of NO by propane at 1 atm ( $\varphi = 1.25$ ;  $\tau = 0.12$  s; 1000 ppm of NO; 2930 ppm of propane). Reactions: NH + NO  $\rightleftharpoons$  N<sub>2</sub>O + H (16); HNO + H  $\rightleftharpoons$  NO + H<sub>2</sub> (50); NO + HO<sub>2</sub>  $\rightleftharpoons$  NO<sub>2</sub> + OH (65); NO + H + M  $\rightleftharpoons$  HNO + M (67); NO + HCO  $\rightleftharpoons$  HNO + CO (68); NO<sub>2</sub> + H  $\rightleftharpoons$  NO + OH (70); CH<sub>2</sub> + NO  $\rightleftharpoons$  HCN + OH (90); HCCO + NO  $\rightleftharpoons$  HCNO + CO (93); HCCO + NO  $\rightleftharpoons$  HCNO + CO<sub>2</sub> (94); HCNO + O  $\rightleftharpoons$  NO + HCO (96) HCNO + OH  $\rightleftharpoons$  NO + CH<sub>2</sub>O (97).

Two rxns of HCCO with NO

(94, ◊) not counter-balanced to reform NO

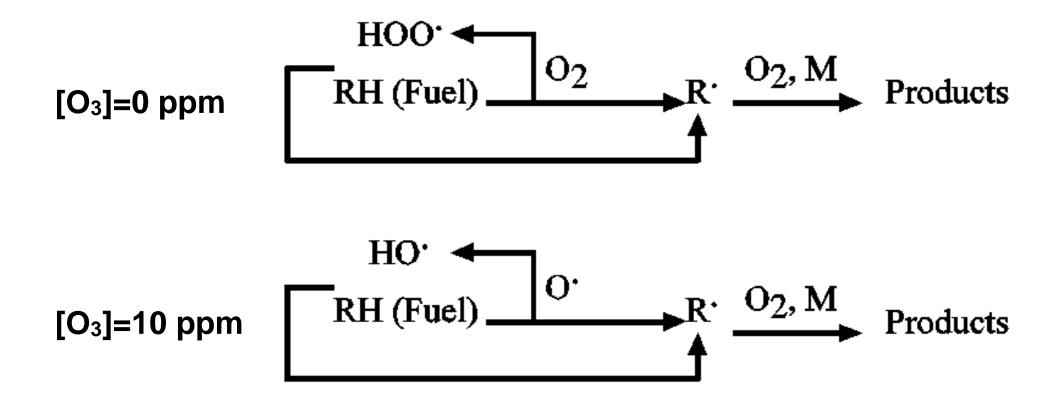
(93, □) counter-balanced by (96) to reform NO

## How PRF100 reactions pathways are modified by ozone injection? [O<sub>3</sub>]=0 [O<sub>3</sub>]=10 ppm



Reaction pathway analysis from rates of consumption (at the bottom) for isooctane (PRF100) at initial temperature of 800 K, initial pressure of 50 bar and equivalence ratio of 0.3. From Masurier et al. Energy Fuels 2013, 27, 5495-5505.

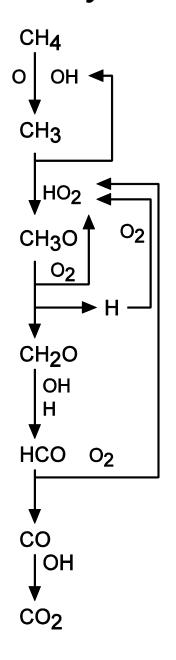
## How reactions pathways are modified by ozone injection?

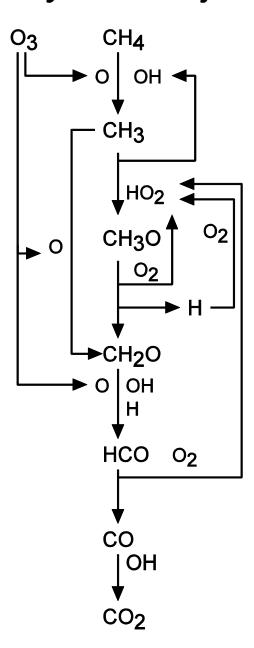


Early reaction paths involved in neat and ozone-seeded fuel oxidation.

From Masurier et al. Energy Fuels 2013, 27, 5495-5505.

## How reactions pathways are modified by ozone injection?





## 2-Brute force method and 1st order sensitivity analyses

What is the impact of a variation of a given parameter (e.g., A-factor,  $\Delta H_f$ ) on the model predictions?

What reactions influence the prediction of the formation/consumption of the product 1?

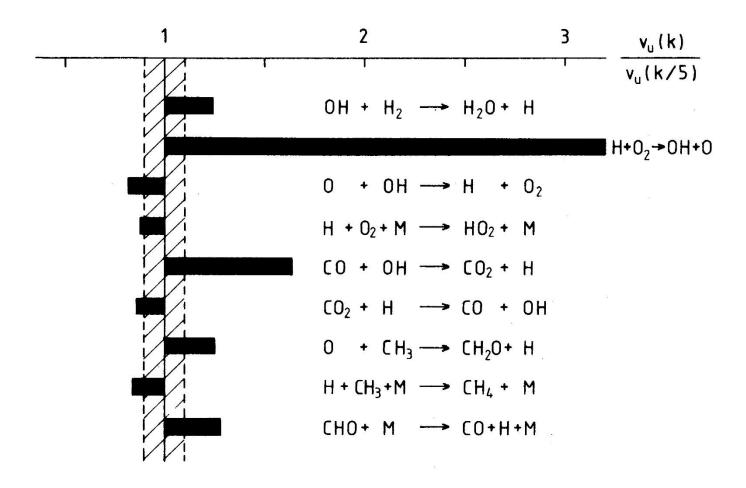
```
Initial k => [product1]<sub>0</sub>

k*\epsilon => [product1]_+

k/\epsilon => [product1]_-
```

 $S = \{[product1]_{ini} - [product1]_{mod}\} / [product1]_{ini}; e.g., [product1]_{mod} = conc. after k_j x 5:$ 

## Brute force method sensitivity analysis (k/5)



Sensitivity of computed laminar burning velocity of a methane-air flame at 1 bar and  $T_u$  = 298 K to reaction kinetics. From Warnatz, J., The structure of laminar alkane-, alkene-, and acetylene flames. Symposium (International) on Combustion, 18(1), p. 380, 1981.

## 2-Brute force method and 1st order sensitivity analyses

What is the impact of a variation of a given parameter (e.g., A-factor,  $\Delta H_f$ ) on the model predictions?

What reactions influence the prediction of the formation/consumption of the product 1?

```
Initial k => [product1]<sub>0</sub>

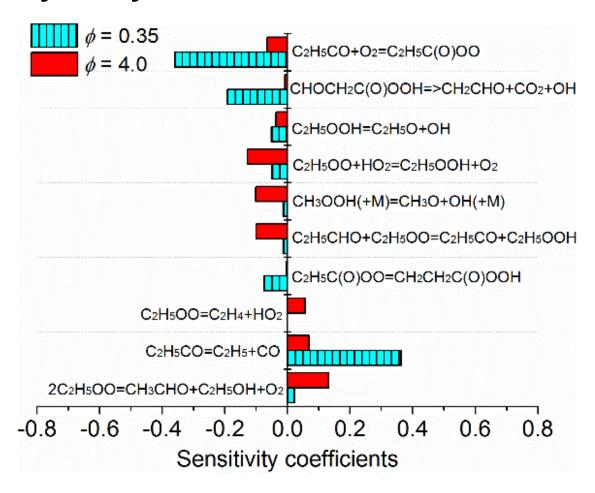
k*\epsilon => [product1]_+

k/\epsilon => [product1]_-
```

```
S = [product1]_{ini}/ [product1]_{mod}; e.g., [product1]_{mod} = conc. \ after \ k_j / 5 s = \partial n_i / \partial \pi_i s' = (\partial n_i / n_i) / (\partial \pi_i / \pi_i)
```

where  $n_i$  is the response of the model and  $\pi$  is a model parameter (A-factor,  $\Delta H_f$ ), e.g.,  $s_{i,j}=(\partial c_i/c_i)$  /  $(\partial A_j/A_j)$  for conc. of species in reaction j:

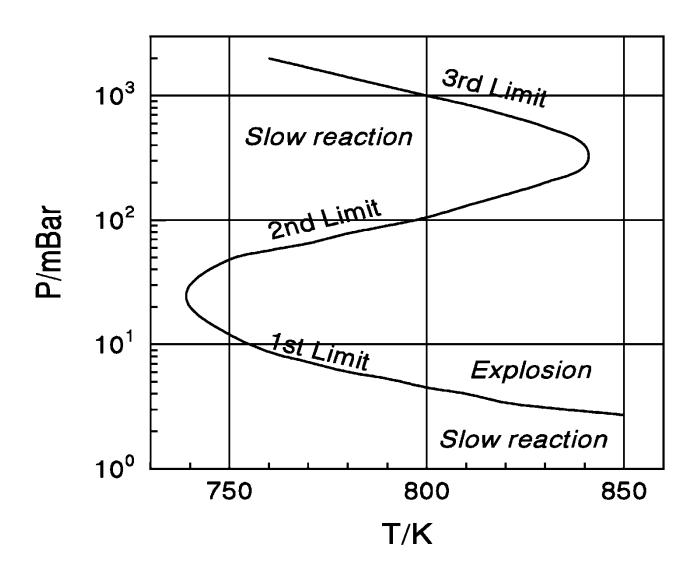
#### 1st order sensitivity analysis



Sensitivity analysis of the present model at  $\varphi = 0.35$  (575 K, 1 atm) and  $\varphi = 4.0$  (625 K, 1 atm) in JSR oxidation of propanal. From New insights into propanal oxidation at low temperatures: Experimental and kinetic modeling study. X. Zhang et al., Proc. Combust. Inst (2019)

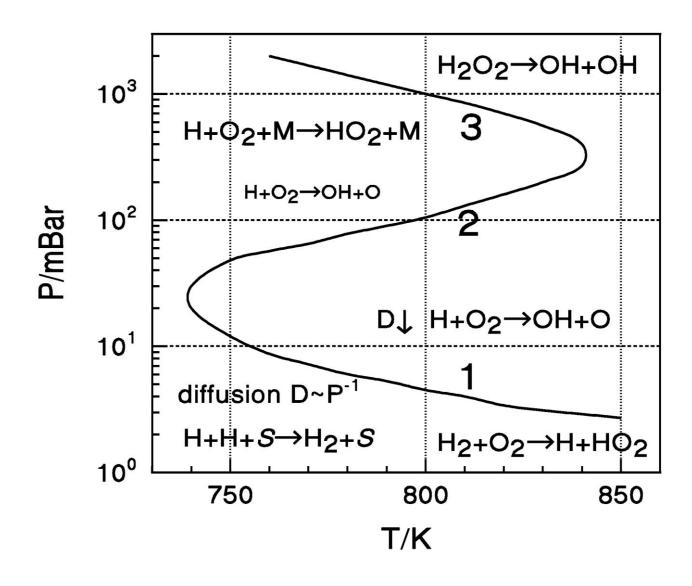
## Modeling: Pressure/Temperature dependencies and reaction pathways

## Explosion Limits of a $\Phi$ =1 H<sub>2</sub>-O<sub>2</sub> Mixture



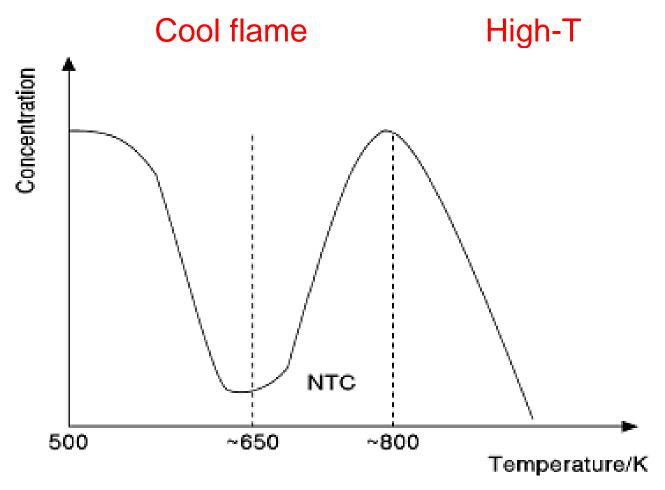
#### **Modeling: Pressure dependencies**

# Explosion Limits of a $\Phi$ =1 H<sub>2</sub>-O<sub>2</sub> Mixture



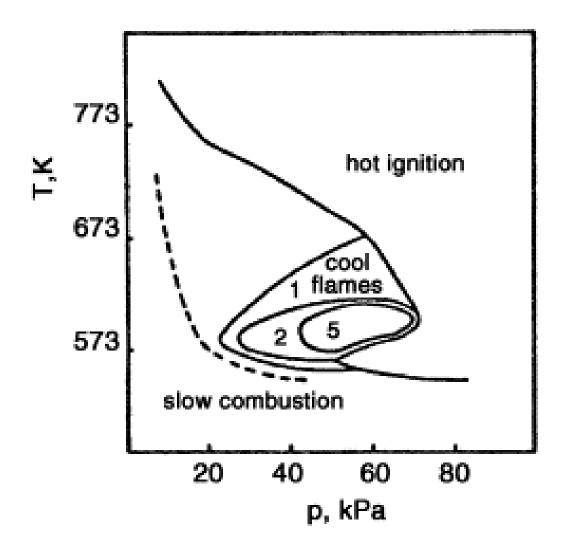
REACTIONS k=A x T <sup>n</sup> x exp(-E/RT)	A/cc,mole,s n	E/cal/mo	ol Ref.
H+ $H+$ $M=$ $H2+$ $M$	7.310E+17 -1.0	0.0	!(BAULCH 76)
O+ O+ M = O2+ M	1.140E+17 -1.0	0.0	!(BAULCH 76)
O+ H+ M = OH+ M	6.200E+16 -0.6	0.0	!(DIXON-LEWIS 81)
H2+ O2 = OH+ OH	1.700E+13 0.0	47780.0	!(MILLER 77)
O+ H2 = OH+ H	3.870E+04 2.7	6260.0	!GRI
H+ O2 = OH+ O	4.400E+14 -0.12	16812.0	!Nicolle 2004
H+ O2+ M = HO2+ M	8.000E+17 -0.8	0.0	!(WARNATZ 84)
H+ OH+ M = H2O+ M	8.615E+21 -2.0	0.0	!(BAULCH 76)
H2+OH=H2O+H	2.161E+08 1.51	3430.0	!(MICHAEL 88)
H2O+ O = OH+ OH	1.500E+10 1.14	17260.0	!(WARNATZ 84)
HO2+ OH = H2O+ O2	2.890E+13 0.0	-497.0	!(KEYSER 88)
HO2+ O = OH+ O2	1.810E+13 0.0	-400.0	!(JPL 87-41)
H+ HO2 = H2+ O2	4.280E+13 0.0	1411.0	!(94BAU/COB)
H+ HO2 = OH+ OH	1.690E+14 0.0	874.0	!(94BAU/COB)
H+ HO2 = H2O+ O	3.010E+13 0.0	1721.0	!(BAULCH 92)
HO2+ HO2 = H2O2+ O2	4.075E+02 3.32	1979.0	!(HIPPLER 90)
OH + OH (+M)= H2O2 (+M)	7.224E+13 -0.37	0.0	!(94BAU/COB)
H2O2+ OH = HO2+ H2O	5.800E+14 0.0	9557.0	!(92HIP/TRO)
H2O2+ H = HO2+ H2	1.700E+12 0.0	3750.0	!(BAULCH 72)
H2O2+ H = H2O+ OH	1.000E+13 0.0	3590.0	!(WARNATZ 84)
H2O2+ O = HO2+ OH	2.800E+13 0.0	6400.0	!(ALBERS 71)

# **Modeling: Hydrocarbons oxidation**

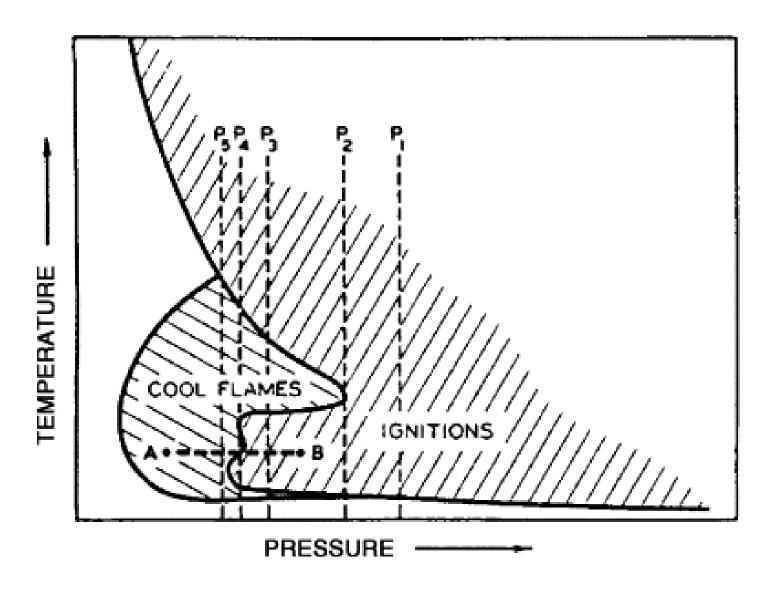


Fuel concentration vs. temperature

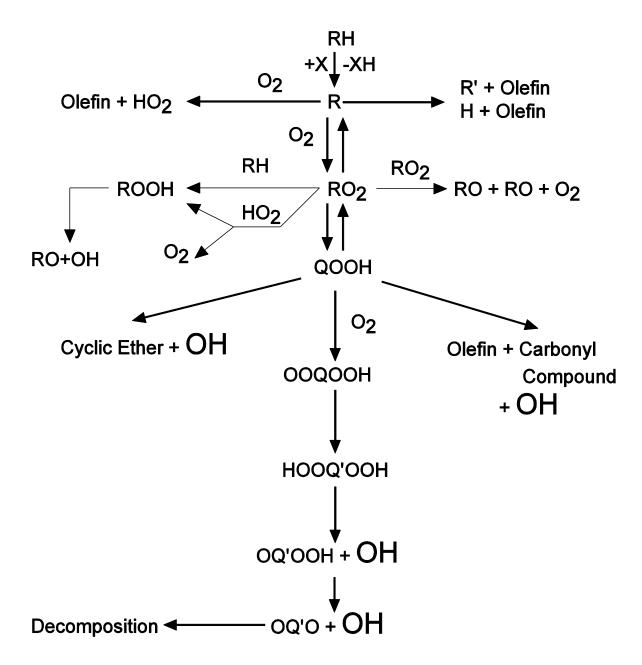
## Modeling: Multiple cool flames

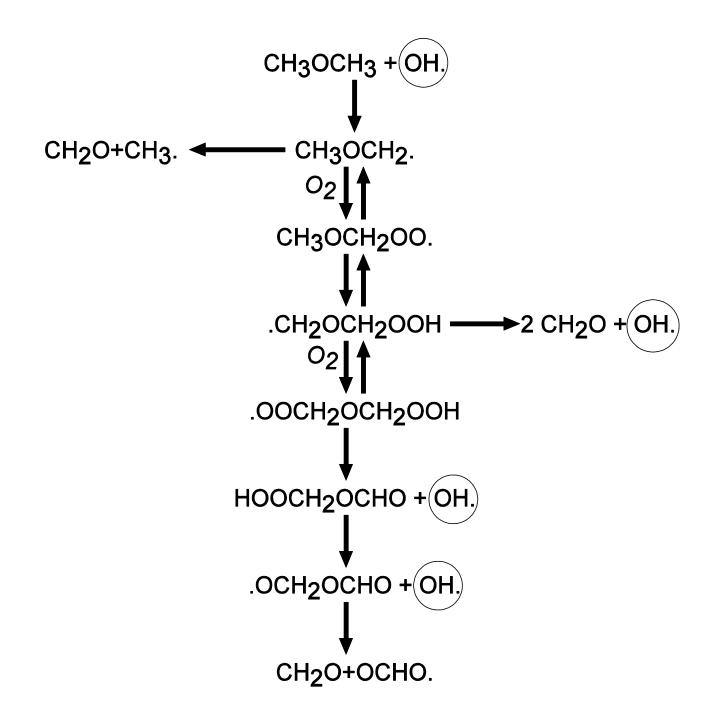


Ignition diagram of a propane/oxygen (1:1) mixture. The numbers refer, to the number of cool flames occurring in the respective region. From P.G. Lignola, E. Reverchon, Prog. Energy Combust. Sci., 13 (1987), p. 75



Ignition diagram for fuel concentration within the flammable range. Moving from A to B can yield to strong ignition





Branching reactions: multiplication of the number of active species

#### Low-T

$$R + O_2 \rightarrow RO_2$$
;  $RO_2 \rightarrow QOOH \rightarrow O_2QOOH \rightarrow 3$  radicals

#### **Medium-T**

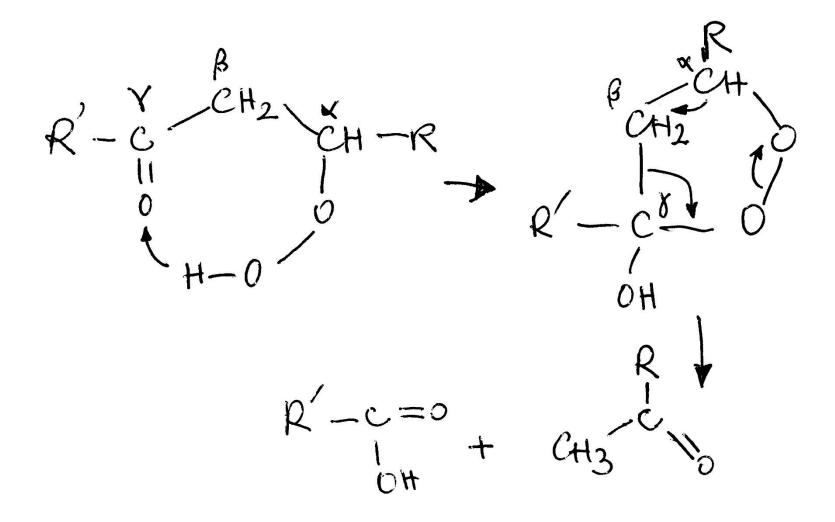
$$H + O_2 + M \rightarrow HO_2 + M$$
;  $RH + HO_2 \rightarrow R + H_2O_2$ ;  $H_2O_2 + M \rightarrow 2 OH + M$   
 $HO_2 + HO_2 \rightarrow H_2O_2$ 

#### High-T

$$H + O_2 \rightarrow OH + O$$

#### Korcek mechanism

γ-Ketohydroperoxides decompose to a carbonyl and a carboxylic acid



#### **Waddington mechanism**

#### More reaction pathways at low-T

Example: di-n-propyl ether oxidation

$$R + O_2 \leftrightarrows RO_2 \leftrightarrows QOOH$$

OOQOOH 

HOOPOOH (alternative H-transfer, not from HC-OOH)

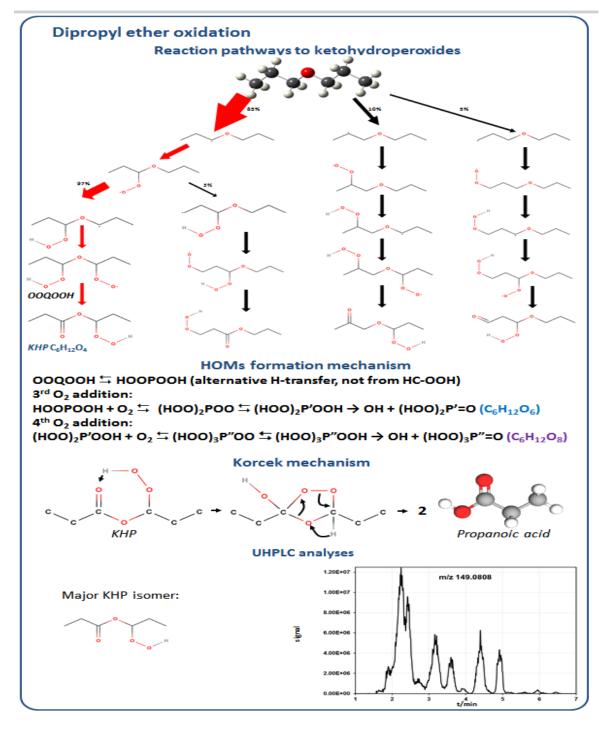
3<sup>rd</sup> O<sub>2</sub> addition:

$$\mathsf{HOOPOOH} + \mathsf{O}_2 \leftrightarrows (\mathsf{HOO})_2 \mathsf{POO} \leftrightarrows (\mathsf{HOO})_2 \mathsf{P'OOH} \to \mathsf{OH} + (\mathsf{HOO})_2 \mathsf{P'=O} (\mathsf{C}_6 \mathsf{H}_{12} \mathsf{O}_6)$$

4<sup>th</sup> O<sub>2</sub> addition:

$$(HOO)_2P'OOH + O_2 \leftrightarrows (HOO)_3P"OO \leftrightarrows (HOO)_3P"OOH \rightarrow OH + (HOO)_3P"=O (C_6H_{12}O_8)$$

5<sup>th</sup> addition ...



Dagaut et al., MCS 2019

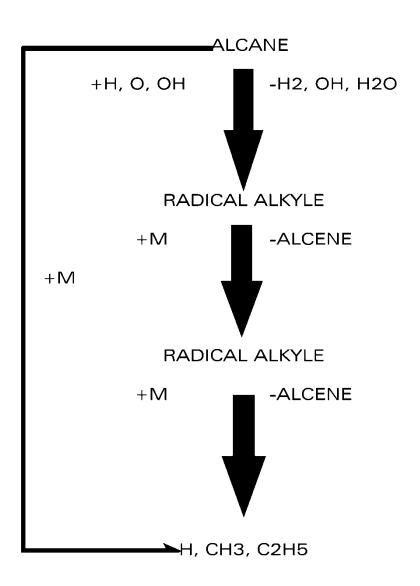
# Reaction pathways to highly oxygenated products considered in atmospheric chemistry (a) and recently extended reaction pathways in combustion (b)

ly extended reaction pathways in combustion (b)
(a)
$$C_{10}H_{16} \xrightarrow{-H} C_{10}H_{15} \xrightarrow{+O_2} C_{10}H_{15}O_2 \xrightarrow{RO_2} C_{10}H_{15}O_4C_{10}H_{15} \xrightarrow{O_2} C_2 + 2 C_{10}H_{15}O \xrightarrow{H-shift} C_{10}H_{14}OH$$

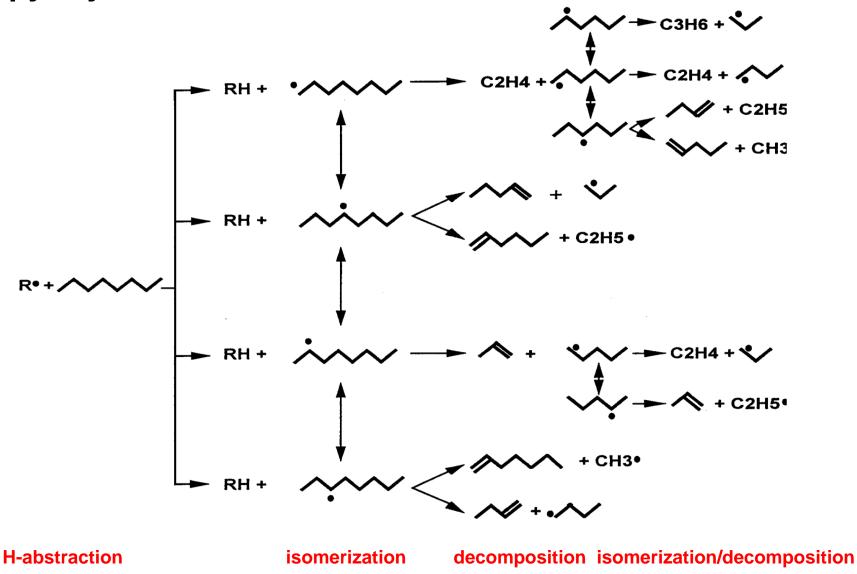
$$Limonene \qquad (R') \qquad (RO') \qquad (RO') \qquad (RO') \qquad (RO')H_{14}OH$$

$$C_{10}H_{14}OH \xrightarrow{+O_2} C_{10}H_{15}O_3 \xrightarrow{H-shift} C_{10}H_{15}O_5 \xrightarrow{H-shift} C_{10}H_{15}O_7 \xrightarrow{H-shift} C_{10}H_{15}O_7 \xrightarrow{H-shift} C_{10}H_{15}O_9 \xrightarrow{H-shift} C_{10}H_{15}O_{11}OO' (ROO)_2Q'OH)OO' (ROO)_2Q'OH)OO' (ROO)_2Q'OH)OO' (ROO)_2Q'OH)OO' (ROO)_2Q'OH)OO' (ROO)_2Q'OH)OO' (ROO)_2Q'OH)OO' (ROO)_2Q'OO' \xrightarrow{H-shift} C_{10}H_{14}O_1 \xrightarrow{O_2OH} HOOQ'OH)OO (ROO)_2Q'OO' \xrightarrow{H-shift} HOOQ'OOH \xrightarrow{+O_2} C_{10}H_{14}O_1 \xrightarrow{O_2OH} HOOQ'OOH \xrightarrow{H-shift} HOOQ'OOH \xrightarrow{+O_2} C_{10}H_{14}O_1 \xrightarrow{O_2OH} HOOQ'OO' (HOO)_2Q'OO' \xrightarrow{H-shift} HOOQ'OOH \xrightarrow{+O_2} C_{10}H_{14}O_1 \xrightarrow{O_2OH} HOOQ'OOH \xrightarrow{H-shift} HOOQ'OOH \xrightarrow{+O_2} C_{10}H_{14}O_1 \xrightarrow{O_2OH} HOOQ'OO' (HOO)_2Q'OO' \xrightarrow{H-shift} (HOO)_2$$

# **Pyrolysis and high-T oxidation**



#### n-Octane pyrolysis



In R': radical position #1 => ethylene

In R': radical position #2 => propene

In R': radical position #3 => butene, heptene

In R': radical position #4 => pentene, hexene

# **Bonds dissociation energies**

Bond A	Donal .	Bond-dissociation energy at 298 K			
Bond <b>♦</b>	Bond <b>♦</b>	(kcal/mol) ♦	(kJ/mol) ♦	(eV/Bond) ♦	
I—I	lodine	36	151	1.57	
Br–Br	Bromine	46	192	1.99	
CI-CI	Chlorine	58	242	2.51	
O=CH <sub>2</sub>	Formaldehyde	179	748	7.75	
N≡N	Nitrogen	226	945	9.79	
О-Н	in α-tocopherol (an antioxidant)	77	323	3.35	
О-Н	in methanol	105	440	4.56	
O=CO	Carbon dioxide	127	532	5.51	
C-CI	in CH <sub>3</sub> Cl	83.7	350	3.63	
C-C	in typical alkane	83–90	347–377	3.60-3.90	
H–H	Hydrogen	104	436	4.52	
0=0	Oxygen	119	498	5.15	
C≣O	Carbon monoxide	257	1077	11.16	
H-F	Hydrogen fluoride	136	569	5.90	
О-Н	in water	119	497	5.15	

# **Bonds dissociation energies**

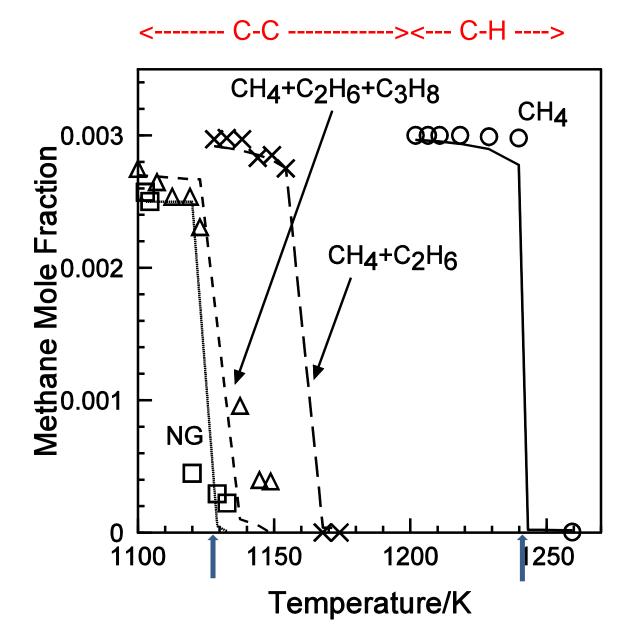
Daniel A	Bond <b>≑</b>	Bond-dissociation energy at 298 K		
Bond <b>♦</b>		(kcal/mol) ◆	(kJ/mol) ♦	(eV/Bond) <b>≑</b>
H <sub>3</sub> C-H	Methyl C-H bond	105	439	4.550
C <sub>2</sub> H <sub>5</sub> –H	Ethyl C-H bond	101	423	4.384
(CH <sub>3</sub> ) <sub>2</sub> CH-H	Isopropyl C–H bond	99	414	4.293
(CH <sub>3</sub> ) <sub>3</sub> C-H	t-Butyl C-H bond	96.5	404	4.187
(CH <sub>3</sub> ) <sub>2</sub> NCH <sub>2</sub> -H	C–H bond α to amine	91	381	3.949
(CH <sub>2</sub> ) <sub>3</sub> OCH-H	C–H bond α to ether	92	385	3.990
CH <sub>3</sub> C(=0)CH <sub>2</sub> - H	C–H bond α to ketone	96	402	4.163
CH <sub>2</sub> CH-H	Vinyl C-H bond	111	464	4.809
НСС-Н	Acetylenic C–H bond	133	556	5.763
C <sub>6</sub> H <sub>5</sub> —H	Phenyl C-H bond	113	473	4.902
CH <sub>2</sub> CHCH <sub>2</sub> -H	Allylic C-H bond	89	372	3.856
C <sub>6</sub> H <sub>5</sub> CH <sub>2</sub> -H	Benzylic C-H bond	90	377	3.907
H <sub>3</sub> C-CH <sub>3</sub>	Alkane C-C bond	83–90	347–377	3.60-3.90
H <sub>2</sub> C=CH <sub>2</sub>	Alkene C=C bond	~170	~710	~7.4
НС≣СН	Alkyne C≡C triple bond	~230	~960	~10.0

# Bonds dissociation energies vs. kinetic parameters

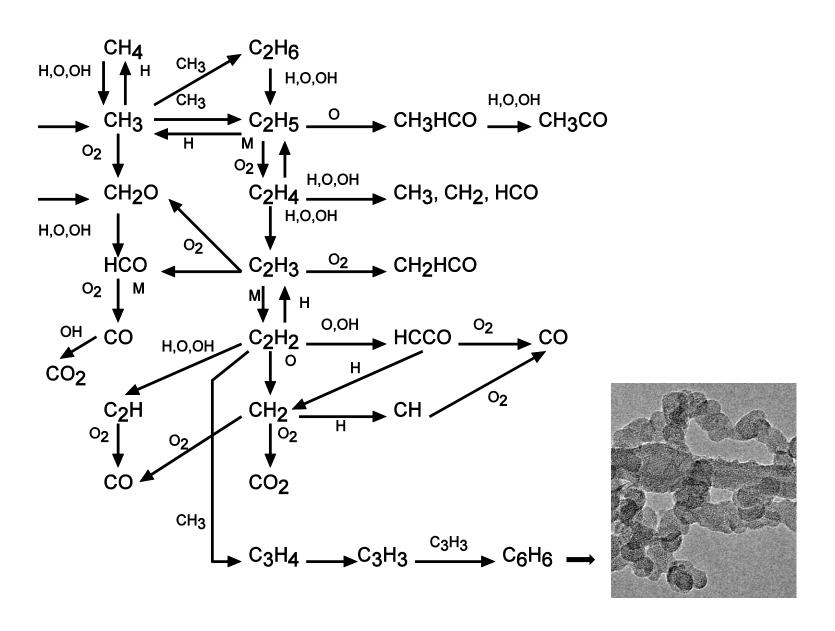
Reaction	Α	n	E/cal/mol	bond, BDE
CH4 = CH3 + H	1.168E+33	-5.43	108732.0	C-H, 105 kcal/mol
C2H6 = C2H5 + H	6.684E+33	-5.48	105330.0	C-H, 101 kcal/mol
C3H8 = C2H5 + CH3	1.698E+44	-1.77	103004.0	C-C, 88 kcal/mol

# Single-fuel vs. multi-fuel components





Oxidation of methane and NG-mixtures in a JSR at 1 atm and 140ms.



#### **Summary**

Modeling: General information. Need accurate kinetics, thermochemistry, and transport data. Use inputs from theory and measurements and also estimations by analogy, tabulations. Need accurate data that are used to constrain the model. Reaction mechanism has a strong hierarchical structure. The core-mechanism is H<sub>2</sub>/O<sub>2</sub> (H, O, OH, HO<sub>2</sub>, H<sub>2</sub>O<sub>2</sub>, O<sub>3</sub>, O<sub>3</sub>, H<sub>2</sub>).

Temperature dependencies of elementary reactions. Modified Arrhenius expression: k=A T<sup>n</sup> exp(-Ea/RT)

Pressure dependencies: Lindemann-Hinshelwood, Troe.

Kinetic analyses: ROP, ROC

Sensitivity analyses: Probe how the model responds to variations of the kinetic paramers

Pressure/Temperature dependencies and reaction pathways: cool flames, high-T oxidation (e.g.,  $R+O_2 \rightarrow RO_2$  vs.  $R-H+HO_2$ )

Oxidation at low-T. More complex than generally considered. Combustion chemistry vs. tropospheric chemistry.

Pyrolysis and high-T oxidation

Single-fuel vs. multi-fuel components. The most reactive components drive the overall oxidation of the complex fuel (e.g., NG vs. methane).

# Part 4 POLLUTANTS



#### 1-NOx formation

#### 1-1-Thermal-NO (Zel'dovich, 1946)

$$N_2+O \rightarrow NO+N$$
 (75.5kcal/mole) 
$$N+O_2 \rightarrow NO+O$$
 
$$N+OH \rightarrow NO+H$$

Global rate (NO formation) =  $[N_2] \times [O_2] \exp(-133000/RT)$ 

# 1-2-Prompt-NO (Fenimore, 1979)

$$CH+N_2 \rightarrow (HCN+N) NCN + H$$
  $CH_2+N_2 \rightarrow HCN+NH$   $C+N_2 \rightarrow CN+N$ 

Followed by:

$$HCN+X \rightarrow CN+HX$$
 $NCN+O \rightarrow CN+NO$ 
 $NCN+OH \rightarrow HCN+NO$ 
 $NCN+H \rightarrow HCN+N$ 
 $NCN+O_2 \rightarrow NO+NCO$ 
 $NCN+O_2 \rightarrow NO+CNO$ 

# 1-3-N<sub>2</sub>O (Malte and Pratt, 1974)

$$N_2 + O(+M) = N_2O(+M)$$

$$N_2O+H=N_2+OH$$

$$N_2O+O=NO+NO$$

$$N_2O+O=N_2+O_2$$

$$N_2O+OH=N_2+HO_2$$

# 1-4-NNH (Bozzelli, Dean, *IJCK* 1995)

 $N_2+H=NNH$ 

 $NNH+H=N_2+H_2$ 

 $NNH+O=N_2O+H$ 

 $NNH+O=N_2+OH$ 

NNH+O=NH+NO

 $NNH+OH=N_2+H_2O$ 

 $NNH+O_2=N_2+HO_2$ 

 $NNH+O_2=N_2+H+O_2$ 

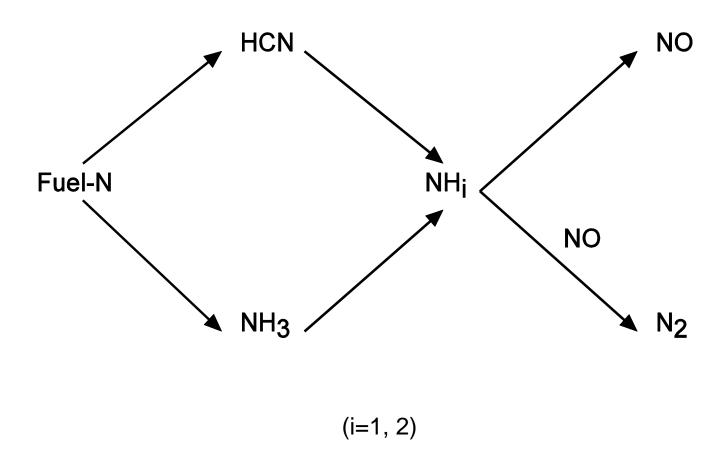
 $NNH+NH=N_2+NH_2$ 

 $NNH+NH_2=N_2+NH_3$ 

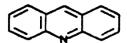
NNH+NO=N<sub>2</sub>+HNO

#### 1-5-Fuel-NO

Formation of HCN and NH<sub>3</sub> by pyrolysis of amines, pyridinic compounds or pyrroles followed by oxidation of HCN or NH<sub>3</sub> to NO and N<sub>2</sub>O



#### Pyridinic-type

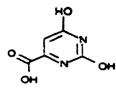


ecridine or 2,3,5,6-dibenzopyridine

nicotinic acid or 3-pyridinecarboxylic acid

# 2

1,2-bis (4-pyridy!)-

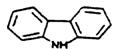


orotic acid or 6-uracilcarboxylic acid

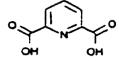
3-pyridal or 3-hydroxypyridine

citrazinic acid or 2,6-dihydroxy-4-pyridinecarboxylic acid

#### Pyrrole-type



carbazole or dibenzopyrrole



dipicolinic acid or 2,6pyridinedicarboxylic acid

antipyrine or 2,3dimethyl-1-phenyl-5-pyrazolone

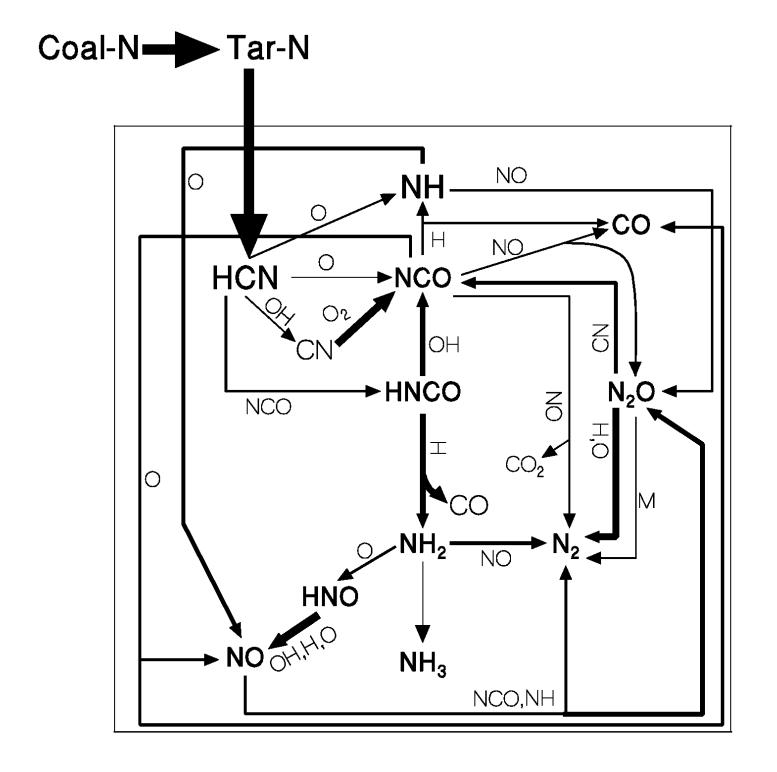
2-pyrrolecarboxylic acid

OL-pyroglutamic acid or DL-5-pyrrolidone-2-carboxylic acid

#### Amino-type

biuret or N-carbamoyurea

meglumine or N-methyl-D-glucamine



#### 2-NOx reduction

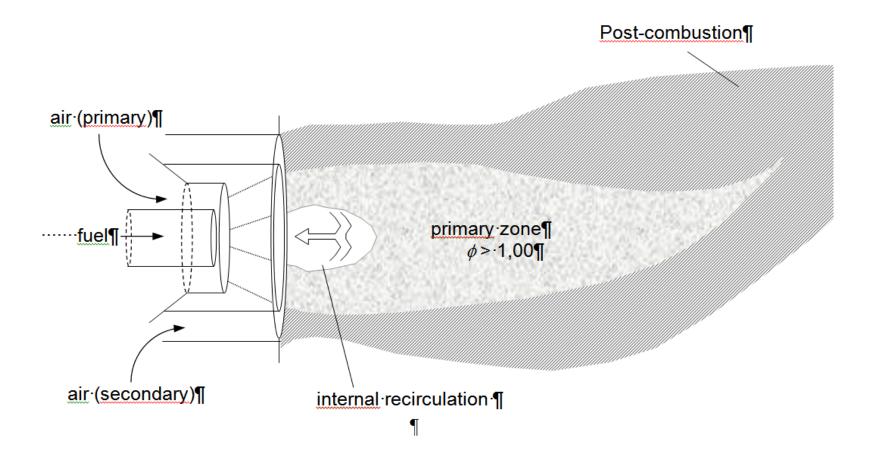
## 2-1-Reduction through combustion modifications

#### 2-1-1-Optimization of burner parameters – low-NOx burners

Burner parameter optimization techniques and lowNOx burners are used to limit NO production during combustion. These burners are specially designed to control the mixing of air and fuel to create more or less turbulent flames stabilized by internal recirculation zones. The temperature of the flame is lowered, thus limiting the production of thermal-NO. This type of burner works as a dual internal staging of fuel and combustion air:

The fuel burns with primary air (70-90%) under fuel-rich conditions. Secondary air (10-30%) is injected over the main combustion zone and completes the oxidation of the fuel. This increases the volume of the flame which decreases the flame temperature and thus the production of thermal-NO.

#### Low-NOx · burner¶



#### 2-1-2-Flue gas recirculation (FGR, EGR)

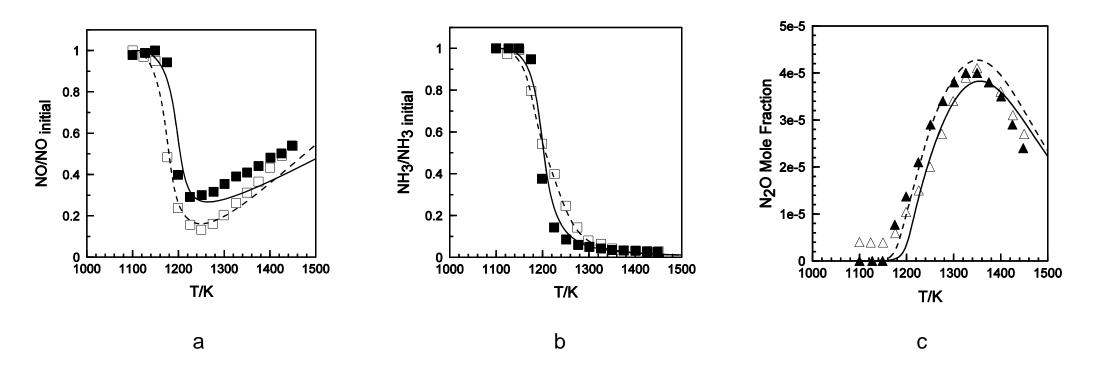
The recirculation of the fumes inside the oven or burner allows a dilution of the flame and therefore a sharp decrease in temperature. Generally, 20 to 30% of the flue gases recirculate and are mixed with the combustion air. The stoichiometry is not modified since the concentration of oxygen in the fumes is negligible. The efficiency is relatively low (<20%) because the contribution of thermal NO does not dominate in installations burning coal.

#### 2-1-3-Fuel staging

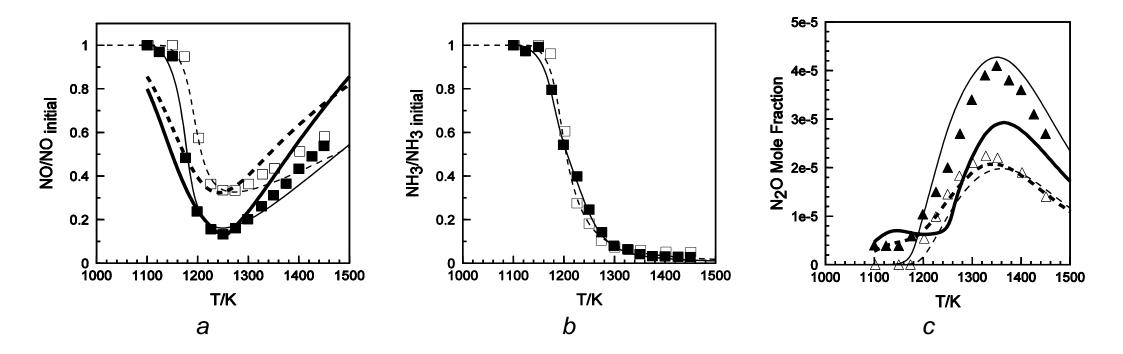
Staging of the fuel allows alternation between a fuel-rich zone and a fuel-lean zone which limits the temperature of the flame, improves the distribution of oxygen, and limits NOx formation.

#### 2-2-Chemical reduction of NOx

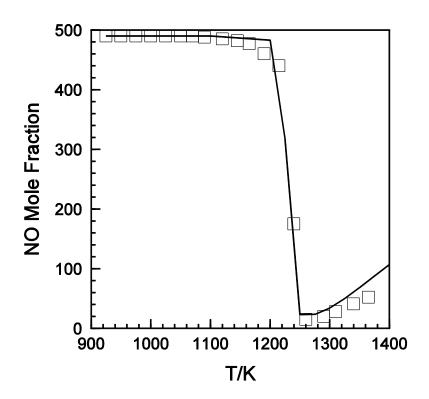
#### 2-2-1-NOx reduction by selective non-catalytic reduction SNCR (Lyon, 1974)



Effect of NO initial concentration on its removal by ammonia in lean conditions ( $\Phi$ =0.1). The initial conditions were: 1000 ppm of NH<sub>3</sub>, 12500 ppm of O<sub>2</sub>, residence time=100 ms, 500 ppm of NO (open symbols and dashed lines) or 1000 ppm of NO (closed symbols and solid line). The data (symbols) are compared to the modeling results (lines).

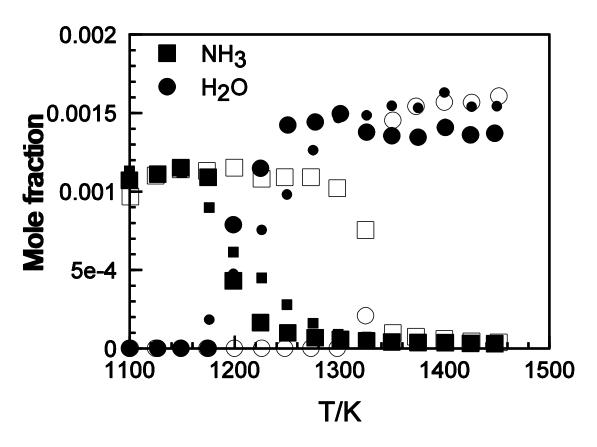


Effect of NH<sub>3</sub> initial concentration on NO reduction and N<sub>2</sub>O formation in lean conditions ( $\Phi$ =0.1). The initial conditions were: residence time=100 ms, 500 ppm of NO, (i) 500 ppm of NH<sub>3</sub> and 6250 ppm of O<sub>2</sub> (open symbols and dashed lines), (ii) 1000 ppm of NH<sub>3</sub> and 12500ppm of O<sub>2</sub> (closed symbols and solid line). The data (symbols) are compared to the modeling results using the present kinetic reaction mechanism (thin lines) and that of literature (thick lines).



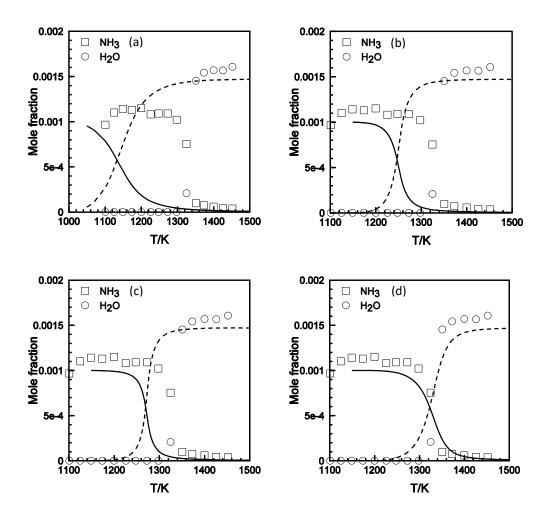
The **reduction of NO by ammonia** in a plug flow reactor: comparison between the experimental results of Kasuya et al. [F. Kasuya, P. Glarborg, J.E. Johnsson, K. Dam-Johansen, *Chem. Eng. Sci.* 50 (1995) 1455.] (symbols) and this modeling (line). The initial conditions were: 1000 ppm of NH<sub>3</sub>, residence time=(88 K/T) s, 500 ppm of NO,4% O<sub>2</sub>, 5% H<sub>2</sub>O, balance N<sub>2</sub>.

#### Ammonia oxidation boosted by NO:



Impact of the initial concentration of NO on NH<sub>3</sub> conversion. Experimental results obtained in a JSR at 1 bar, 1000 ppm NH<sub>3</sub>,  $\tau$ =100 ms,  $\phi$ =0.1, 0 ppm (open symbols), 500 ppm (small black symbols), and 1000 ppm (large black symbols) of NO.

P. Dagaut, CST(2019) <a href="https://doi.org/10.1080/00102202.2019.1678380">https://doi.org/10.1080/00102202.2019.1678380</a>



Data (symbols) and computed (lines) results for NH<sub>3</sub> oxidation in a JSR: 1000 ppm of NH<sub>3</sub>,  $\tau$ =100ms;  $\phi$ =0.1. Models: (a), (b), (c), and (d).

- [a] A.A. Konnov, Combust. Flame 156 (11) (2009) 2093-2105.
- [b] Y. Song, H. Hashemi, J.M. Christensen, C. Zou, P. Marshall, P. Glarborg, Fuel 181 (2016) 358-365.
- [c] J. Otomo, M. Koshi, T. Mitsumori, H. Iwasaki, K. Yamada, Int. J. Hydrogen Energy 43 (5) (2018) 3004-3014.
- [d] P. Dagaut, P. Glarborg, M.U. Alzueta, Prog. Energy Combust. Sci. 34 (1) (2008) 1-46.

### Ammonia oxidation boosted by NO:

Reaction pathway analyses were performed to delineate the mechanism responsible for the mutual sensitization of ammonia and nitric oxide. The computations shows that it occurs via several reaction pathways leading to OH production, which is the main species involved in ammonia oxidation. In the present conditions HO<sub>2</sub> is mainly produced via:

$$NNH+O_2 \rightarrow N_2+HO_2$$

and

$$H+O_2+M \rightarrow HO_2+M$$
.

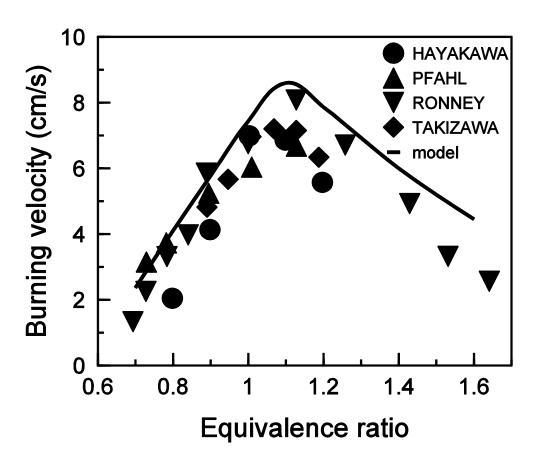
The production of OH results from a sequence of reaction including

$$NH_2+NO \rightarrow NNH+OH$$

$$NNH \rightarrow N_2\text{+}H$$

$$H+O_2 \rightarrow OH+O$$

$$NO+HO_2 \rightarrow NO_2 + OH$$
.

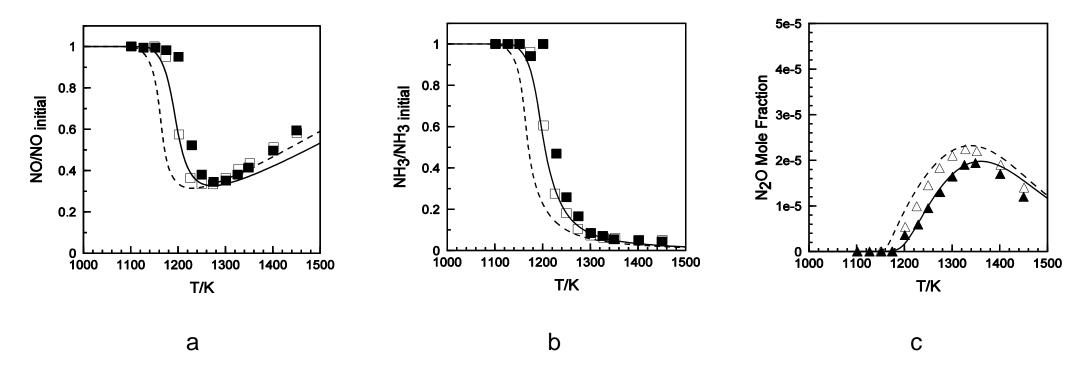


Computed (lines) and literature experimental (symbols) results for NH<sub>3</sub>-air flames at 1 atm.

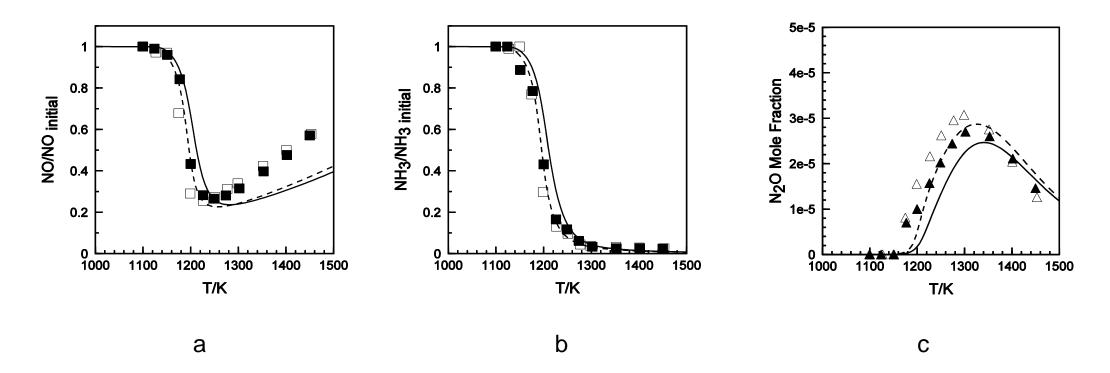
The kinetics of the reactions  $NH_2 + H \rightarrow NH + H_2$  and  $HNO + H \rightarrow NO + H_2$  were updated (Otomo et al., 2018) to better simulate burning velocities of ammonia in air.

Dagaut, CST (2019) https://doi.org/10.1080/00102202.2019.1678380

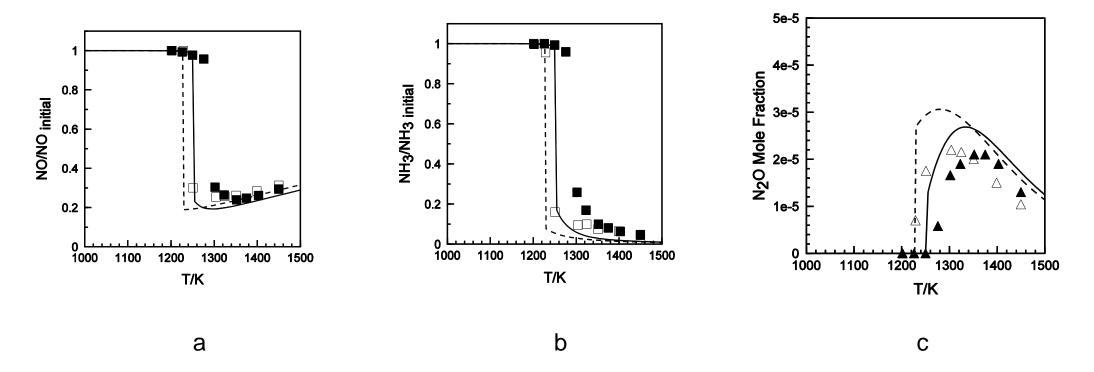
### Perturbation by sulfur dioxide



Effect of  $SO_2$  initial concentration on NO removal by ammonia in lean conditions ( $\Phi$ =0.1). The initial conditions were: 500 ppm of NH<sub>3</sub>, 6250 ppm of O<sub>2</sub>, residence time=100 ms, 500 ppm of NO (open symbols and dashed lines) and 1000 ppm of  $SO_2$  (closed symbols and solid line). The data (symbols) are compared to the modeling results (lines).



Effect of SO<sub>2</sub> initial concentration on NO removal by ammonia in lean conditions ( $\Phi$ =0.1). The initial conditions were: 1000 ppm of NH<sub>3</sub>, 12500 ppm of O<sub>2</sub>, residence time=200 ms, 1000 ppm of NO (open symbols and dashed lines) and 1000 ppm of SO<sub>2</sub> (closed symbols and solid line). The data (symbols) are compared to the modeling results (lines).



Effect of  $SO_2$  initial concentration on NO removal by ammonia in fuel-rich conditions ( $\Phi$ =2). The initial conditions were: 1000 ppm of NH<sub>3</sub>, 625 ppm of O<sub>2</sub>, 200 ms, 1000 ppm of NO (open symbols and dashed lines) and 1000 ppm of  $SO_2$  (closed symbols and solid line). The data (symbols) are compared to the modeling results (lines).

#### NH<sub>2</sub> production:

$$NH_3+OH => NH_2 + H_2O, R(NH_2)=0.863$$
 (149)

$$NH_3+O => NH_2 + OH, R(NH_2)=0.124$$
 (150)

NH<sub>2</sub> reacts with NO via (161) and (162),

$$NH_2 + NO => N_2 + H_2O, R(NO) = -0.544$$
 (161)

$$NH_2 + NO => NNH + OH, R(NO)=-0.322$$
 (162)

**OH** radicals are produced via

$$NH_2 + NO => NNH + OH, R(OH)=0.41$$
 (162)

$$H + O_2 => OH + O, R(OH)=0.187$$
 (-74)

$$NO+ HO_2 => NO_2 + OH, R(OH)=0.157$$
 (99)

$$NH_3 + O => NH_2 + OH, R(OH) = 0.142$$
 (149)

O-atoms are produced by reaction (-74),

$$H+O_2 => OH + O, R(O)=0.997$$
 (-74)

SO<sub>2</sub> contributes moderately to the removal of O-atoms through reaction (7):

$$SO_2 + O(+M) => SO_3(+M), R(O) = -0.03$$
 (7)

The model indicates that SO<sub>2</sub> reacts mostly through 3 reactions:

$$SO_2 + O(+M) => SO_3(+M), R(SO_2) = -0.173$$
 (7)

$$H + SO_2 + M => HOSO + M, R(SO_2) = -0.204$$
 (57)

$$SO_2 + NH_2 => NH_2SO_2, R(SO_2) = -0.43$$
 (72)

HOSO formed in reaction (57) recycles SO<sub>2</sub> via reaction (13):

$$HOSO + O_2 => HO_2 + SO_2$$
, **R(HOSO)=-0.999** (13)

The sequence of reactions (13) + (57)

$$HOSO + O_2 => HO_2 + SO_2$$
 (13)

$$H + SO_2 + M => HOSO + M \tag{57}$$

is equivalent to  $H + O_2 + M = HO_2 + M$ 

=> **reduction of the radical pool** since the fraction of H atoms reacting in (57) will not produce OH and O via reaction (-74) and OH via reaction (100), NO<sub>2</sub> + H => NO +OH.

Thus, under such conditions, introducing 1000 ppm of SO<sub>2</sub> reduces the rate of production of O by a factor of 1.8 and that of OH by a factor of 1.75. Since O and OH are the major agents of oxidation of NH<sub>3</sub>, via reactions (149) and (150), the rate of ammonia oxidation is reduced by a factor of 1.7, resulting in the reduction of the rate of NH<sub>2</sub> production by a factor of 1.7 and in a reduction of 42% of NO-consumption rate.

# **Summary**

Under fuel-lean conditions, the addition of SO<sub>2</sub> inhibits the SNCR process via:

 $H + SO_2 + M = HOSO + M$  followed by  $HOSO + O_2 = HO_2 + SO_2$  equivalent to the equation:

$$H + O_2 + M = HO_2 + M$$
.

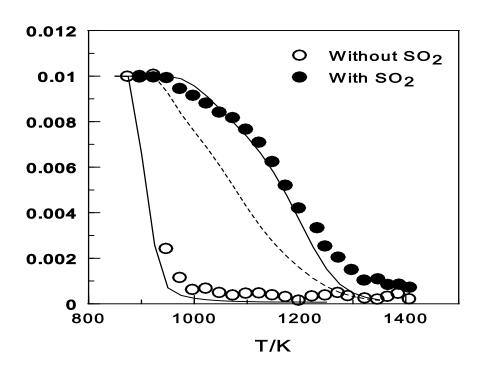
Under fuel rich conditions, the addition of SO<sub>2</sub> inhibits the process via:

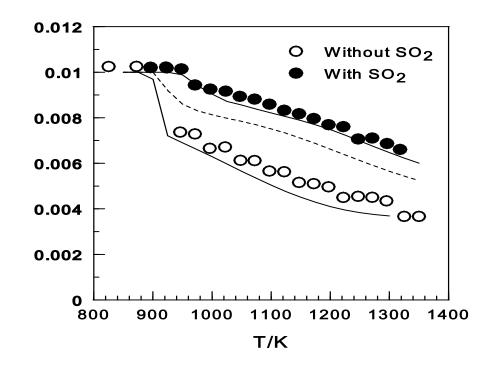
$$H + SO_2 + M = HOSO + M$$
 followed  $HOSO + H = H_2 + SO_2$ 

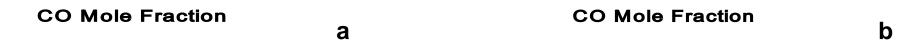
and via  $H + SO_2 + M = HOSO + M$  followed by  $HOSO + O_2 = HO_2 + SO_2$ .

SO<sub>2</sub> does not reduce the efficiency of the thermal de-NOx process but shifts the optimal temperature to higher values.

#### Further inhibiting effects of SO<sub>2</sub>

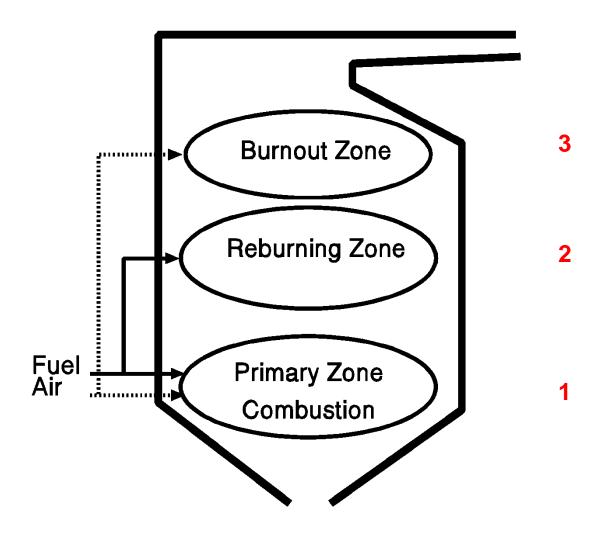




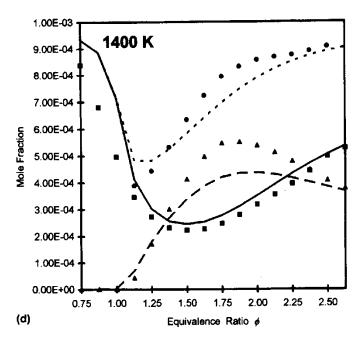


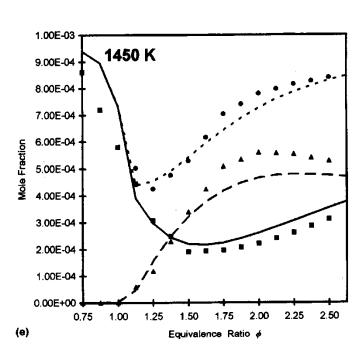
The **effect of SO<sub>2</sub> on the oxidation of a CO/H<sub>2</sub> mixture** in a plug-flow reactor. Initial conditions: (a) CO = 1.0%,  $H_2 = 1.0\%$ ,  $O_2 = 1.0\%$ ,  $H_2 = 2.0\%$ , balance  $N_2$ , without and with  $SO_2 = 1.2\%$ , residence time is 192.7/T or 192.3/T; (b) CO = 1.0%,  $H_2 = 1.0\%$ ,  $O_2 = 0.5\%$ ,  $H_2 = 2.0\%$ , balance  $N_2$ , without and with  $SO_2 = 0.3\%$ , residence time is 192.7/T or 192.3/T. Inhibition is due to  $H+SO_2+M=HOSO+M$  followed by  $HOSO+H=H_2+SO_2$ . From Dagaut et al., Int J Chem Kinet 35: 564–575, 2003.

### 2-2-2-NOx reduction by reburning)



- (1) Thermal-NO production in near-stoichiometric conditions;
- (2) fuel-rich zone, NO + HC  $\rightarrow$  N<sub>2</sub>, HCNO<sub>x</sub>;
- (3) excess-air, HCNOx oxidation → NO





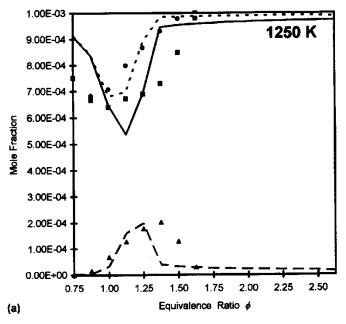


FIGURE 11 The effect of equivalence ratio on the reburning of NO by a  $CH_4/C_2H_6$  (10:1) mix at 1 atm ( $\tau = 0.12$  s; 1000 ppm of NO; 7272 ppm of  $CH_4$ ; 728 ppm of  $C_2H_6$ ). Comparison between experimental data (symbols) and modeling (lines): NO,  $\blacksquare$ —; HCN,  $\blacktriangle$ ---; TFN,  $\bullet$ ...; (a) 1250 K; (b) 1300 K; (c) 1350 K; (d) 1400 K; (e) 1450 K.

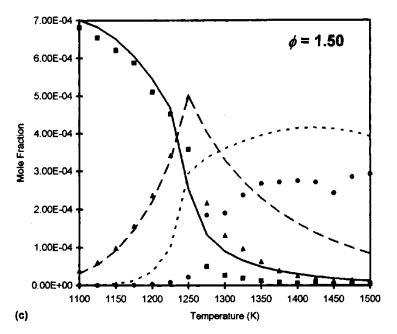


FIGURE 9 (Continued).

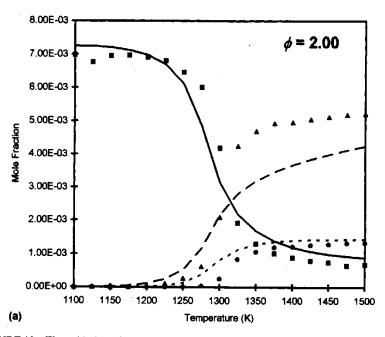
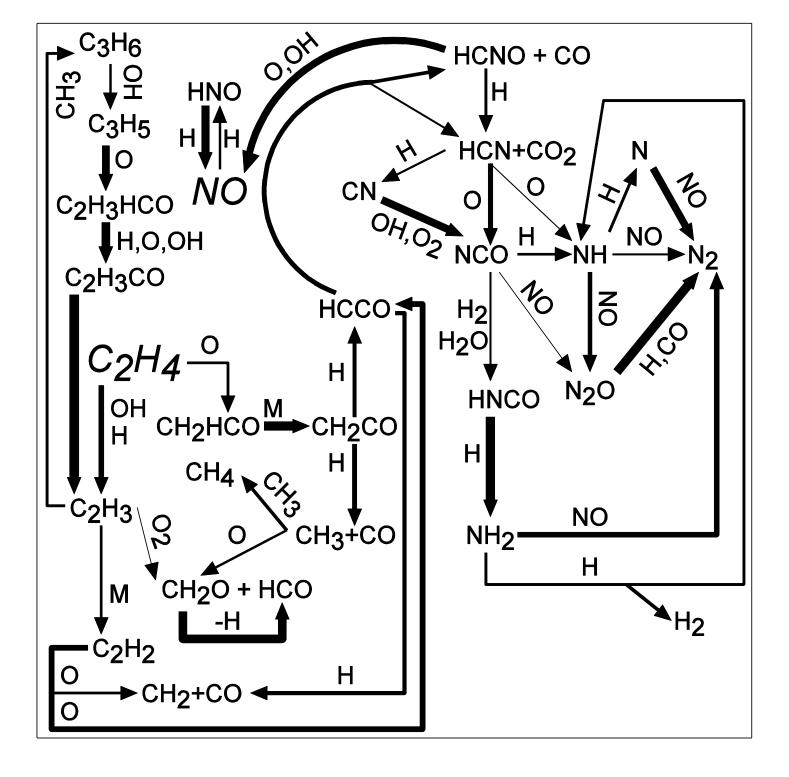
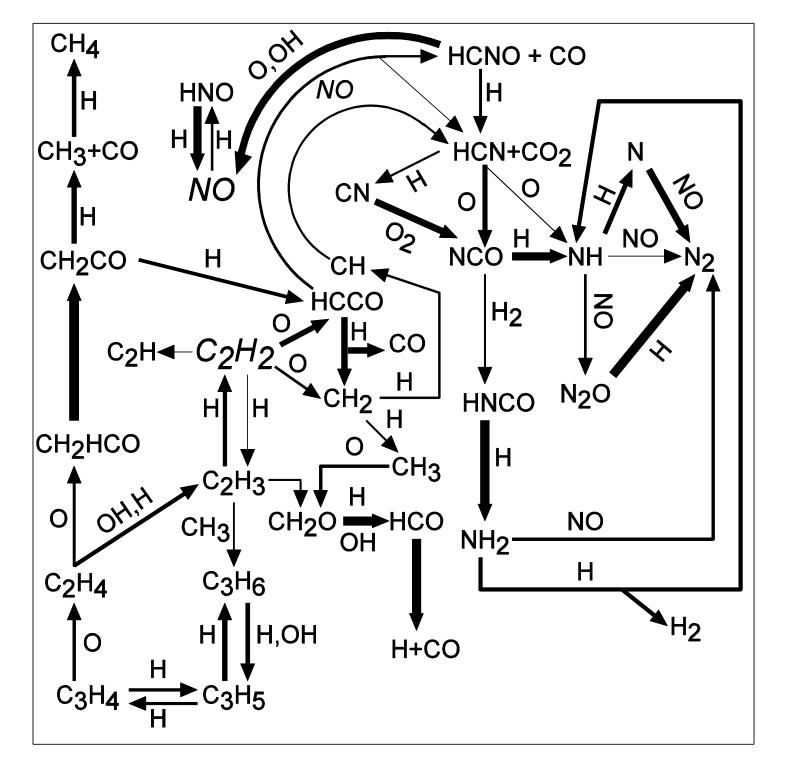
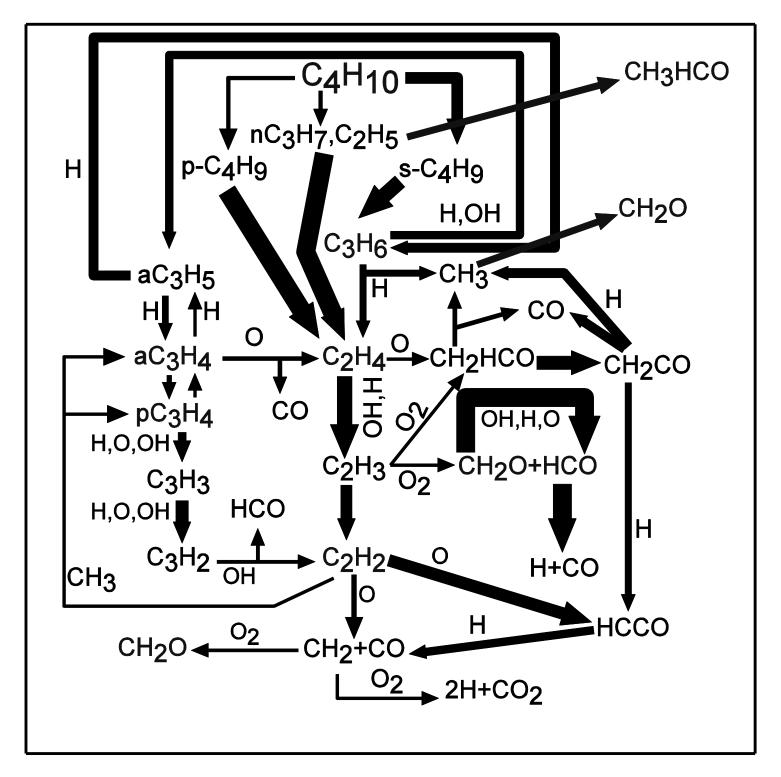
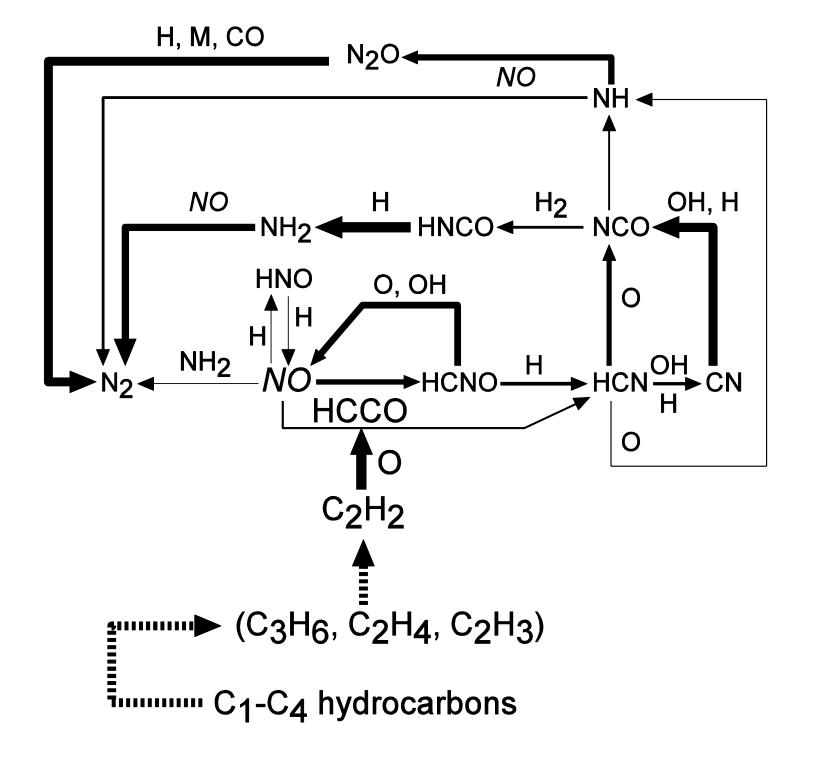


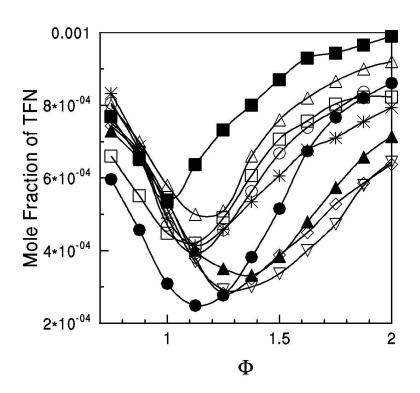
FIGURE 10 The oxidation of a  $CH_4/C_2H_6$  (10:1) mix in a JSR at latm ( $\tau = 0.12$  s; 7272 ppm of  $CH_4$ ; 728 ppm of  $C_2H_6$ ; 8546 ppm of  $O_2$ ;  $\phi = 2$ ). Comparison between experimental data (symbols) and modeling (lines); (a)  $CH_4$ ,  $\blacksquare$ ;  $CO_2$ ,  $\bullet$ ; (b)  $O_2$ ,  $\blacksquare$ ;  $H_2$ ,  $\triangle$ ; (c)  $C_2H_6$ ,  $\blacksquare$ ;  $C_2H_4$ ,  $\triangle$ ;  $C_2H_2$ ,  $\bullet$ .



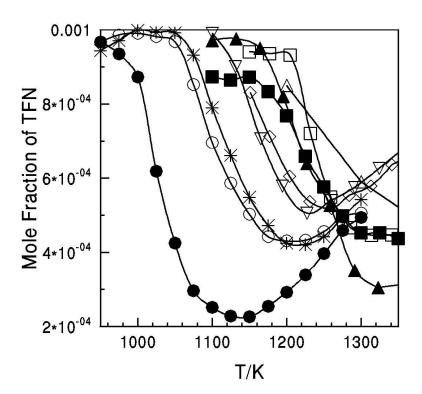






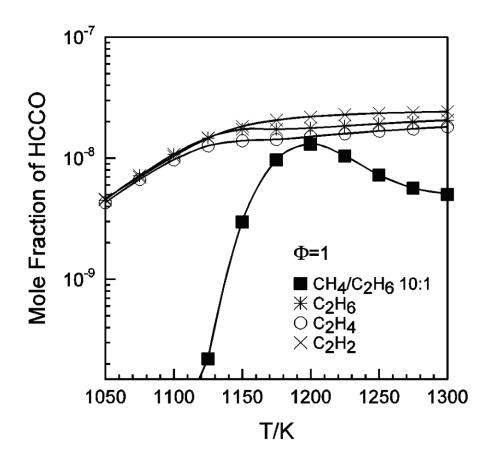


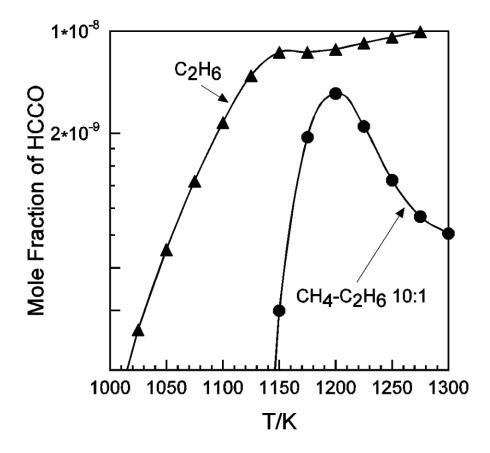
TFN (NO+HCN) vs eq. ratio; 1300 K, 1000ppm NO, 8800 ppm C, 0.12s. Reburn fuels: NG-blend ■; ethane \*; ethylene ○, acetylene ●; NG △; propene □; propane ◊; n-butane Δ; i-butane ▲.



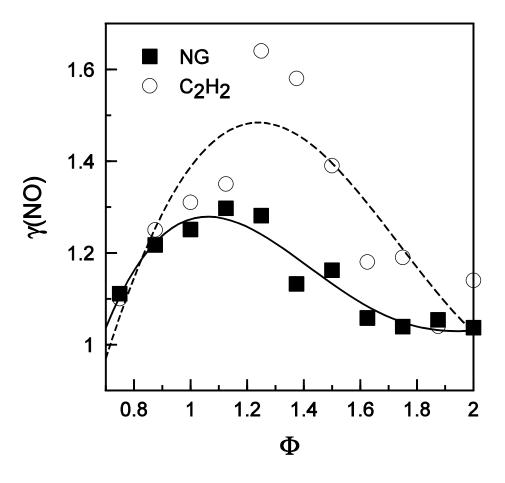
TFN (NO+HCN) vs. T for 8 reburn fuels: NG-blend ■; ethane \*; ethylene ○, acetylene ●; NG △; propene □; propane ◊; n-butane Δ; i-butane ▲. Stoichiometric mix, 1000ppm NO, 8800 ppm C, t: 0.12-0.16s

# **HCCO** production





# Inhibiting effect of SO<sub>2</sub> on NO-reburning using 2 reburn fuels at 1300K



$$\gamma(NO) = X(NO)_{w. SO2} / X(NO)_{w/o SO2}$$

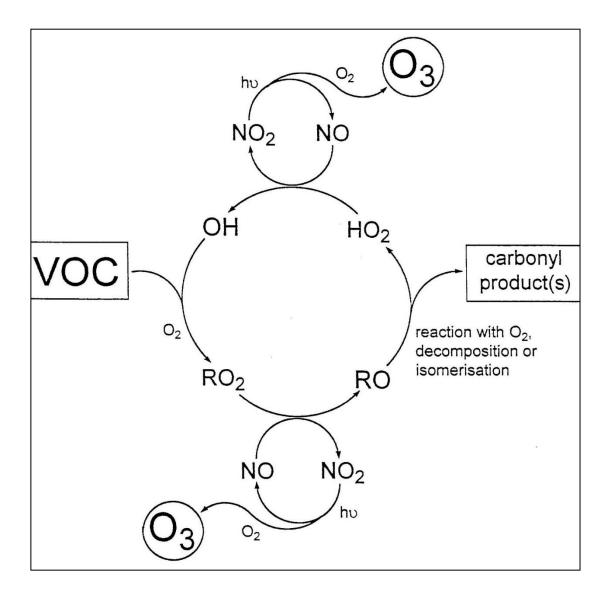
The residual of NO increases in presence of sulfur dioxide

#### **3-UHCs and Soot**

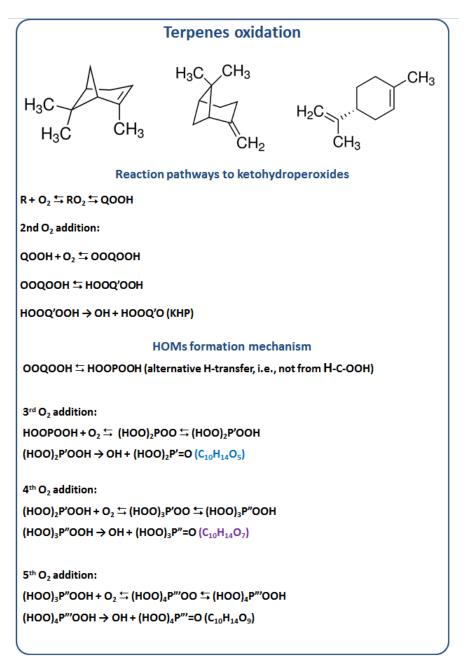
#### Organic compounds in the troposphere

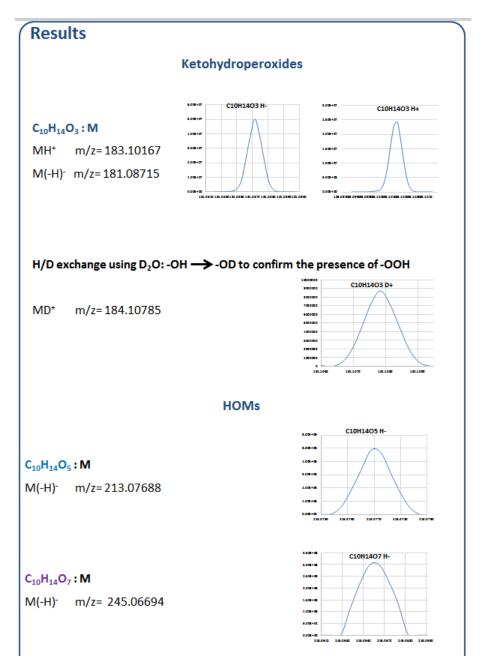
Class	Compound	Formula	Typical Source	Sink	Concentration Range
Alkanes	Methane	CH <sub>4</sub>	Microbial processes, natural gas	ОН	1.7 ppm
	Ethane	$C_2H_6$	Motor vehicles	ОН	0-100 ppb
	Hexane	$C_6H_{14}$	Motor vehicles	OH	0–30 ppb
Alkenes	Ethene	$C_2H_4$	Motor vehicles, microbial processes	OH, O <sub>3</sub>	0–100 ppb
	Propene	$C_3H_6$	Motor vehicles	$OH, O_3$	0-50 ppb
	Isoprene	$C_5H_8$	Vegetation	$OH, O_3$	0.2–30 ppb
Alkynes	Acetylene	$C_2H_2$	Motor vehicles	ОН	0–100 ppb
Aromatics	Benzene	$C_6H_6$	Motor vehicles	OH	••
	Toluene	$C_7H_8$	Motor vehicles	OH	
Aldehydes	Formaldeyde	НСНО	Motor vehicles	$h\nu$ , OH	
	Acetaldehyde	CH₃CHO	Motor vehicles	$h\nu$ , OH	
	Acrolein	CH <sub>2</sub> CHCHO			
Ketones	Acetone	CH <sub>3</sub> C(O)CH <sub>3</sub>		$h\nu$ , OH	0–10 ppb
Acids	Formic acid	НСООН		Rain	11
	Acetic acid	CH₃COOH		Rain	
Alcohols	Methanol	CH <sub>3</sub> OH		ОН	

# Oxidation of organic compounds in the troposphere

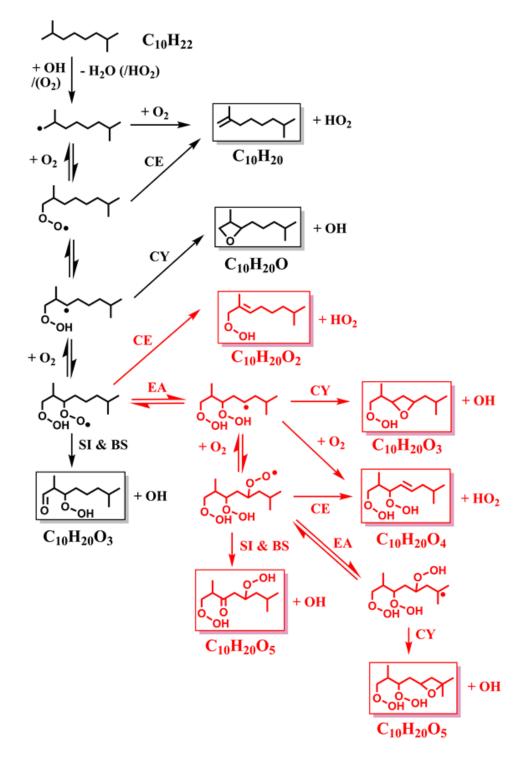


### Terpenes to HOMs and Secondary Organic Aerosols (SOAs)



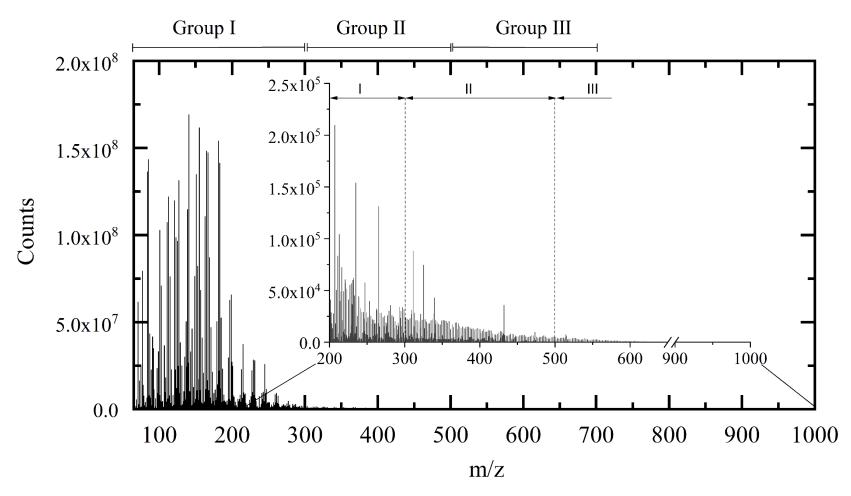


From Belhadj et al. ICCK, 2019



Additional pathways to HOMs by Wang et al. PNAS (2017)

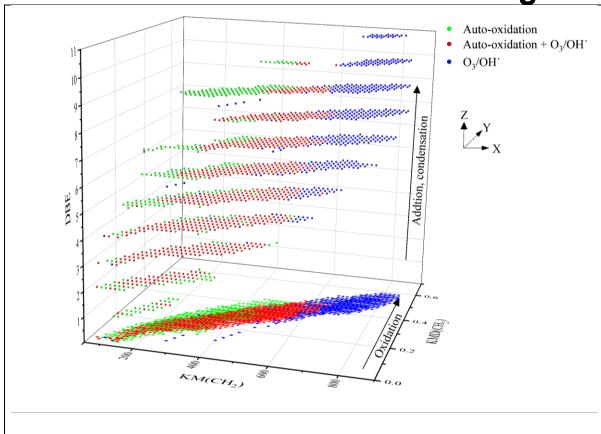
#### **HOMs observed in a JSR**



**Limonene oxidation** sample from a JSR analyzed by FIA and HESI (negative mode). Group I corresponds to compounds resulting from multiple oxidation reactions including fragmentation and condensation.

Groups II and III correspond to higher molecular masses, resulting from addition and condensation reactions. R. Benoit, N. Belhadj, M. Lailliau, P. Dagaut, Atm. Chem. Phys., 2021 <a href="https://doi.org/10.5194/acp-2020-1070">https://doi.org/10.5194/acp-2020-1070</a>

Limonene oxidation: Kendrick diagram



Kendrick's mass analysis (Kendrick, 1963) allows representing in two dimensions and in a new reference frame, a complex mass spectrum of an organic mixture. This reference frame is based on a mass defect calculated from structural units (CH<sub>2</sub>, O, CHO, ...). In a Kendrick representation, the homologous series (constructed by the repeated addition of structural units CH<sub>2</sub>, O, CHO, ...) are aligned on the same horizontal line. The mass defect is calculated by the difference between the Kendrick mass and the nominal mass. If CH<sub>2</sub> is chosen as the structural unit, in Kendrick's plots, the x-axis represents the Kendrick Mass:

$$KM(CH_2) = observed \ mass * \frac{nominal \ mass \ of \ CH_2}{exact \ mass \ of \ CH_2}$$

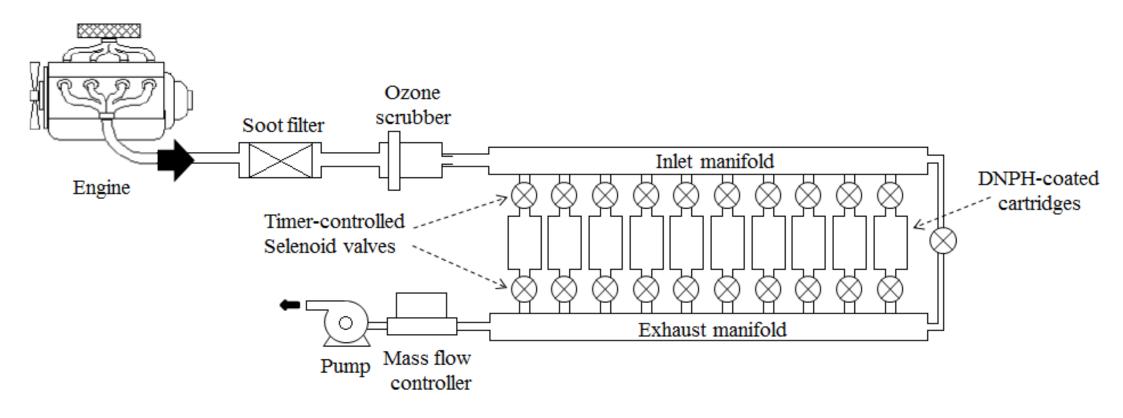
The y-axis represents the Kendrick Mass Defect:

$$KMD(CH_2) = nominal\ mass - Kendrick\ mass\ (CH_2)$$

The **number of double bond equivalent** (DBE) represents the sum of the number of unsaturation and ring present in a compound.

All the chemical products, resulting from limonene oxidation by ozonolysis/photooxidation and autoxidation gathered in the form of a Kendrick diagram correlated to the DBE: • new chemical products from autoxidation experiments in JSR; • products common to the 3 modes of oxidation; • chemical products with molecular formula not observed in JSR.

https://doi.org/10.5194/acp-2020-1070



Engine and gas sampling system. DNPH+carbonyl; HPLC with UV detection @360nm.

From Dagaut et al., J. Eng. Gas Turbines Power 141, 031028-1 (2019)

#### **Diesel engine conditions**

Nbr of cylinders 4

Cycle 4

Cylinder (cm<sup>3</sup>) 1460.74

Vol. Ratio 15.21

injector Continental SA.

Type of injection Direct Common Rail

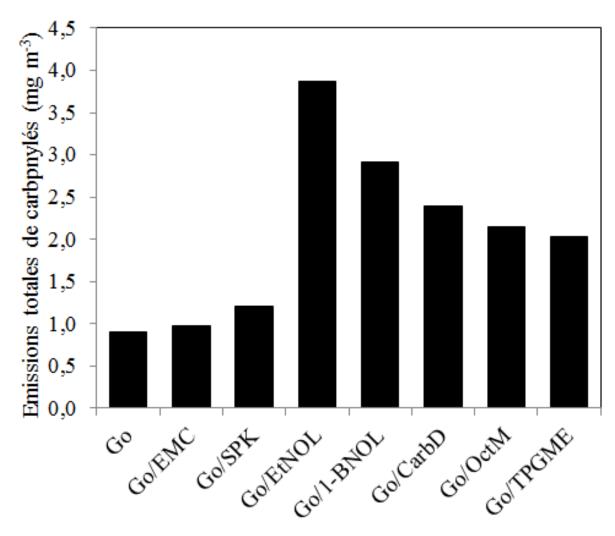
Nbr of injectors 4

Nbr of injection 3 per cycle

Post-treatment no

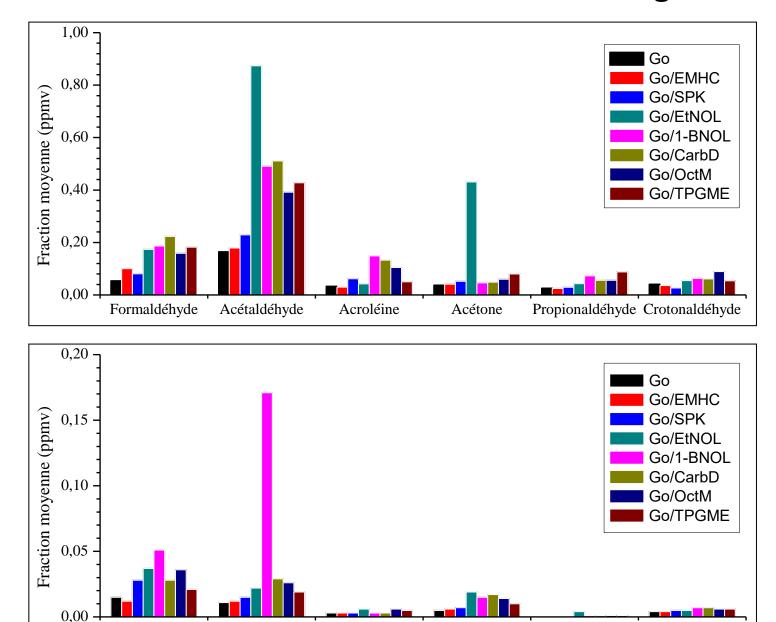
#### Additives used (mix Diesel/additive 90/10 vol.)

Mix DCN	Additive	Chemical class	Formula	Density	M.W.
				(g/mL @ 25°C)	(g/mol)
55,34	none				
46.32	EtNOL	Alcool	C <sub>2</sub> H <sub>6</sub> O	0.789	46.07
49.71	1-BNOL	Alcool	C <sub>4</sub> H <sub>10</sub> O	0.810	74.12
48.96	CarbD*	ester of carbonate	C <sub>5</sub> H <sub>10</sub> O <sub>3</sub>	0.975	118.13
54.22	OctM	methyl ester	C <sub>9</sub> H <sub>18</sub> O <sub>2</sub>	0.877	158.24
55.56	ЕМНС	Mixed methylesters	C <sub>17.92</sub> H <sub>33</sub> O <sub>2</sub>	0.883#	280
54.75	TPGME	ether	C <sub>10</sub> H <sub>22</sub> O <sub>4</sub>	0.963	206.28
52.74	SPK	Mixed paraffins <sup>‡</sup>	C <sub>11.03</sub> H <sub>23.37</sub>	0.761#	156



Global emission of carbonyl compounds at I.C.E. exhaust





Shahla, Ph.D., 2015

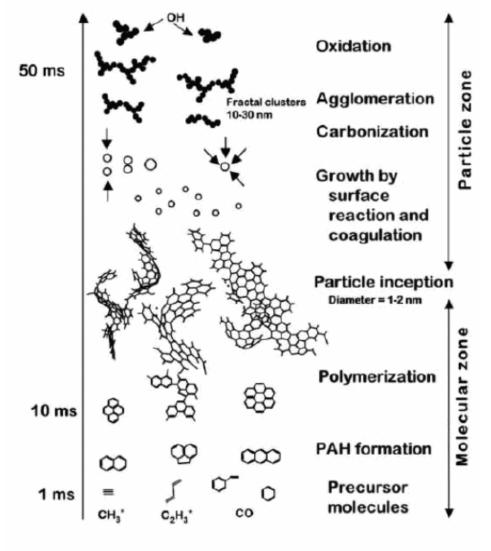
Meth.+2-butan. n-butyraldéhyde Benzaldéhyde

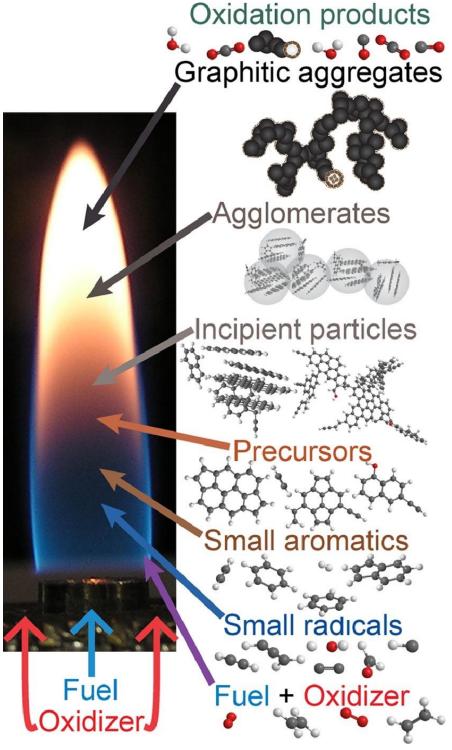
Valéraldéhyde p-Tolualdéhyde

Hexanal

#### **Soot and PAHs**

H.A.Michelsen, Proc. Combust. Inst. 36, 717-735 (2016)





## PAHs formation via the HACA mechanism (Frenklach and Wang)

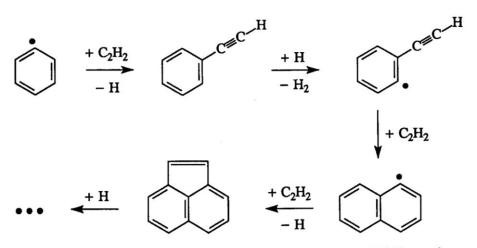


Fig. 10.2. H-abstraction-C<sub>2</sub>H<sub>2</sub>-addition reaction pathway of PAH growth

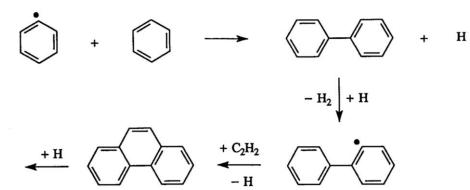


Fig. 10.3. PAH growth initiated by aromatic "combination"

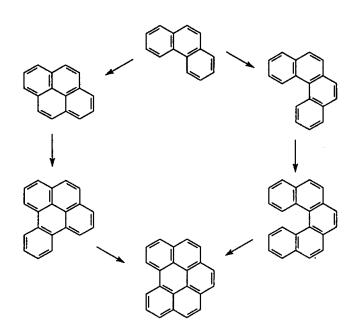


Fig. 10.4. Comparison of two pathways of PAH growth

The main kinetic features of PAH growth after a certain PAH size,  $i_0$ , are revealed by considering an analytical solution with the smallest set of reactions that represent the principal elements of the HACA sequence [10.24]. This minimal reaction set is given as

$$A_i + H \rightleftharpoons A_i \bullet + H_2 \tag{10.3}$$

$$A_i \bullet + C_2 H_2 \rightleftharpoons A_i C_2 H_2 \bullet \tag{10.4}$$

$$A_i C_2 H_2 \bullet + C_2 H_2 \to A_{i+1} + H$$
 (10.5)

### **Benzene formation**

Fuels: acetylene (HC≡CH) propene (CH<sub>3</sub>-CH=CH<sub>2</sub>), propyne (HC≡C-CH<sub>3</sub>), allene (H<sub>2</sub>C=C=CH<sub>2</sub>) 1,3-butadiene (H<sub>2</sub>C=CH-CH=CH<sub>2</sub>)

Conditions: JSR, 1 atm, 900-1300 K

Benzene formation from acetylene, allene and propyne proceeds through a **C**<sub>3</sub> **channel** involving the recombination of propargyl radicals:

$$C_3H_3 + C_3H_3 \rightarrow C_6$$
 intermediates  $\rightarrow$  Benzene (1)

In the case of 1,3-butadiene, the formation of benzene is driven by 2 competitive routes, a ( $C_2+C_4$ ) route and the  $C_3$  route (1):

 $1,3-C_4H_6+C_2H_3 \rightarrow C_6$  intermediate  $\rightarrow$ Benzene (2)

 $1,3-C_4H_5 + C_2H_2 \rightarrow C_6$  intermediate  $\rightarrow$ Benzene (3)

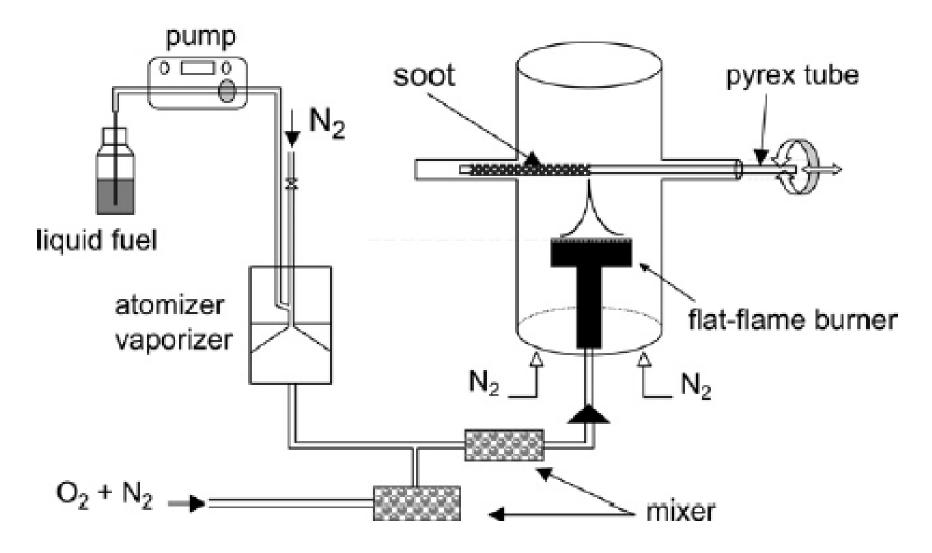
According to our computations, the formation of 1,3-C<sub>4</sub>H<sub>5</sub> results from the intermediate formation of 1,3-cyclopentadiene (1,3-CPD):

$$aC_3H_5 + C_2H_2 \rightarrow 1,3-CPD \rightarrow C_5H_5 \rightarrow 1,3-C_4H_5$$
 (5)

For propene, the early formation of benzene involves a **C**<sub>3</sub> **route**: the recombination of allyl radicals, formed by H-atom abstraction from propene, producing 1,5-hexadiene

$$aC_3H_5+aC_3H_5 \rightarrow 1,5-C_6H_{10} \rightarrow C_6H_9 \rightarrow cyclo-C_6H_9 \rightarrow 1,3-cyclohexadiene \rightarrow C_6H_7 \rightarrow C_6H_6$$

From P. Dagaut and M. Cathonnet. A comparative study of the kinetics of benzene formation from unsaturated C<sub>2</sub> to C<sub>4</sub> hydrocarbons, Combust. Flame, 113, 620 (1998).



**Experimental set-up, premixed sooting flame.** 

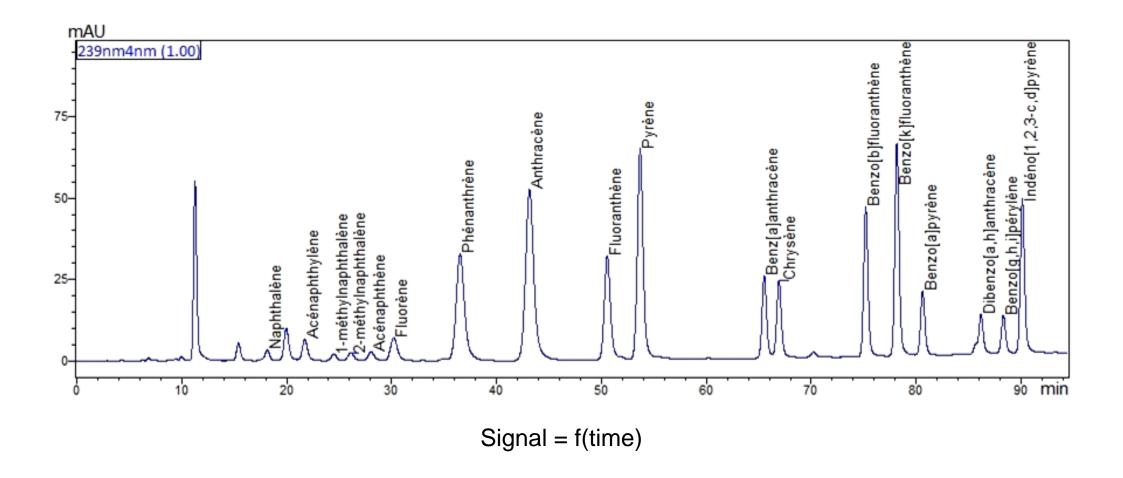
From Dagaut et al., J. Eng. Gas Turbines Power 141, 031028-1 (2019)

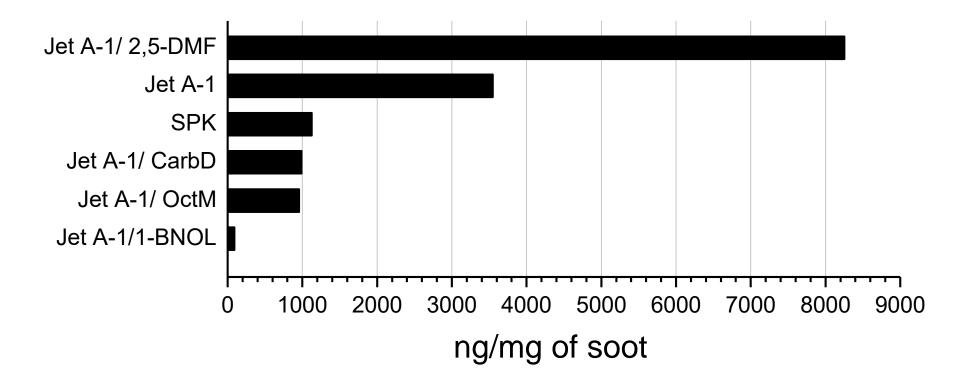
# **Experimental conditions**

Fuel	Formula	Fuel flow rate (cm <sup>3</sup> . s <sup>-1</sup> )	Air flow rate (cm <sup>3</sup> . s <sup>-1</sup> )		E.R.
			N <sub>2</sub>	<b>O</b> <sub>2</sub>	
Jet A-1	C <sub>11</sub> H <sub>22</sub>	1.58	35	11.7	2.23
Jet A-1/1-BNOL*	$C_{9.6}H_{19.6}O_{0.2}$	1.87	35	12.0	2.24
Jet A-1/CarbD*	$C_{9.8}H_{19.6}O_{0.6}$	1.80	35	11.7	2.28
Jet A-1/OctM*	$C_{10.6}H_{21.2}O_{0.4}$	1.68	35	11.3	2.28
Jet A-1/2,5-DMF*	$C_{10}H_{19.2}O_{0.2}$	1.70	35	11.2	2.23

<sup>\*</sup> Jet A-1/additif 80:20 v/v

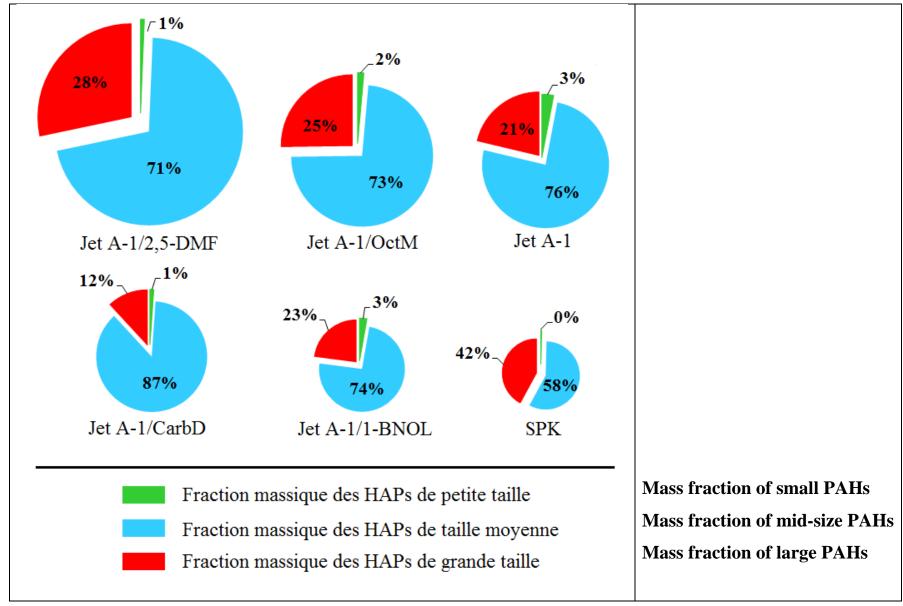
# **HPLC Chromatogram showing 18 HAPs after extraction**







Concentration of 18 HAPs on soot particles



Contribution of ≠ classes of PAHs to total amount of PAHs on soot (Shahla, 2015)

## Global toxicity of soot samples\*

Fuel	Equivalent toxicity(TEQ)	Variation to Jet A-1 %  -99	
Jet A-1/1-BNOL	1,294		
Jet A-1/CarbD	10,834	-94	
Jet A-1/OctM	83,976	-57	
SPK	115,904	-40	
Jet A-1	193,574	0	
Jet A-1/2,5-DMF	574,136	+197	

From Shahla (2015)

$$TEQ = \left(\sum_{i=HAP} C_i \times TEF_i\right) \cdot f$$

<sup>\*</sup>Nisbet et Lagoy (*Regulatory Toxicology and Pharmacology, vol. 16, pp. 290-300, 1992*) defined a global equivalent toxicity:

# Effect of trace species on ignition: NOx, ozone

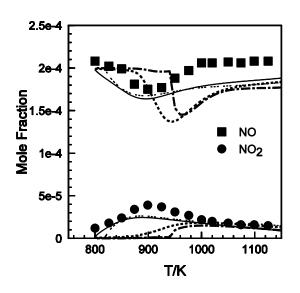
#### 1-NOx-HC interactions

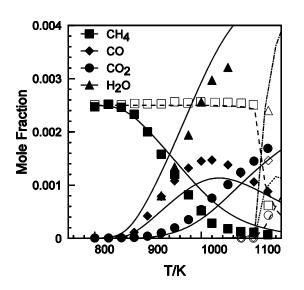
The mutual sensitization of the oxidation of methane and NO proceeds through the NO to NO<sub>2</sub> conversion by HO<sub>2</sub> and CH<sub>3</sub>O<sub>2</sub>.

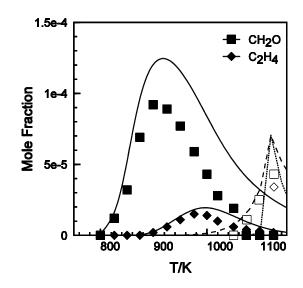
At 1-10 atm, the conversion of NO to NO<sub>2</sub> by CH<sub>3</sub>O<sub>2</sub> is more important at low temperatures (800 K) than at higher temperatures (850-900 K) where the reaction of NO with HO<sub>2</sub> dominates the production of NO<sub>2</sub>.

The NO to NO<sub>2</sub> conversion is enhanced by the production of HO<sub>2</sub> and CH<sub>3</sub>O<sub>2</sub> radicals from the oxidation of the fuel. The production of OH resulting from the oxidation of NO promotes the oxidation of the fuel: NO + HO<sub>2</sub> => OH+ NO<sub>2</sub> is followed by OH + CH<sub>4</sub> => CH<sub>3</sub>. At low temperature, the reaction further proceeds via CH<sub>3</sub> + O<sub>2</sub> => CH<sub>3</sub>O<sub>2</sub>; CH<sub>3</sub>O<sub>2</sub> + NO => CH<sub>3</sub>O + NO<sub>2</sub>. At higher temperature, the production of CH<sub>3</sub>O involves NO<sub>2</sub>: CH<sub>3</sub> + NO<sub>2</sub> => CH<sub>3</sub>O.

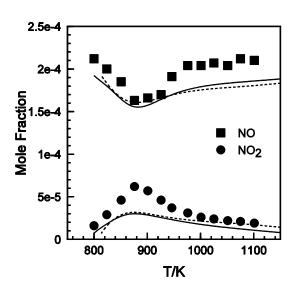
The sequence of reactions:  $CH_3O \Rightarrow CH_2O + H$ ;  $CH_2O + OH \Rightarrow HCO$ ;  $HCO + O_2 \Rightarrow HO_2$  and  $H + O_2 \Rightarrow HO_2$ .  $=> CH_2O + H$ ;  $CH_2O + OH \Rightarrow HCO$ ;  $HCO + O_2 \Rightarrow HO_2$  and  $H + O_2 \Rightarrow HO_2$ .

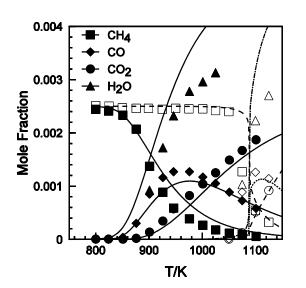


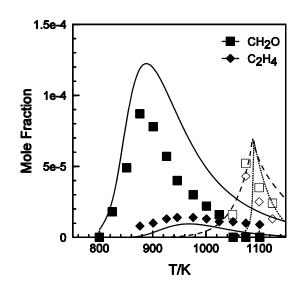




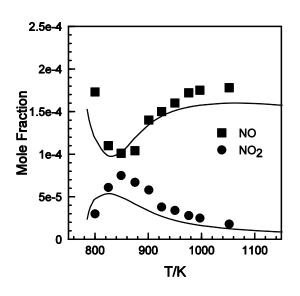
The mutual sensitization of the oxidation of methane and NO in a JSR at 1 atm: Effect of the introduction of 200ppm of NO on the oxidation of methane in fuel-lean conditions ( $\varphi$ =0.1, 2500 ppm of CH<sub>4</sub>, 50000 ppm of O<sub>2</sub>, t=120 ms). (a): The dashed-dotted line represents the results obtained with the mechanism and thermochemical data of [Hori 2002]. The results obtained with the mechanism and thermochemical data of [Hori 1998] are presented as dashed lines, those using [Faravelli 2003] as a dotted line (... ...), the results of the proposed model are presented as full lines. In (b) and (c):The filled symbols and the continuous lines refer respectively to the data and the simulations (proposed scheme) with NO added; the open symbols and dotted lines refer respectively to the data and simulations (proposed scheme) without NO.

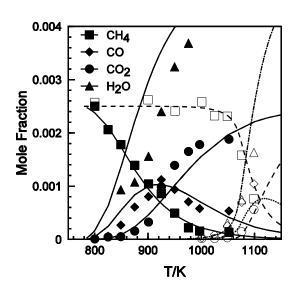


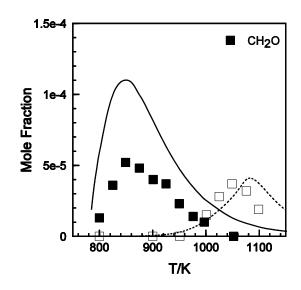




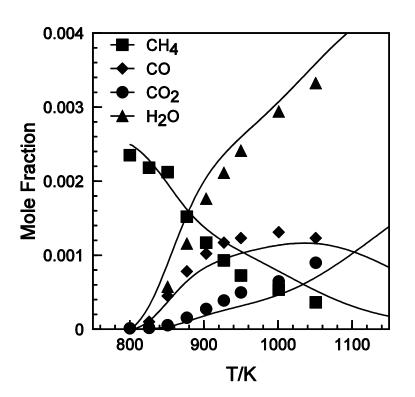
The mutual sensitization of the oxidation of methane and NO in a JSR at 1 atm: Effect of the introduction of **200ppm of NO** on the oxidation of methane in fuel-lean conditions ( $\varphi$ =0.1, 2500 ppm of CH<sub>4</sub>, 50000 ppm of O<sub>2</sub>, t=240 ms). (a)The experimental results (symbols) are compared to the computations (dashed lines using the model of [Faravelli 2003], continuous line for this work). In (b) and (c):The filled symbols and the continuous lines refer respectively to the data and simulations with NO added; the open symbols and dotted lines refer respectively to the data and simulations without NO.

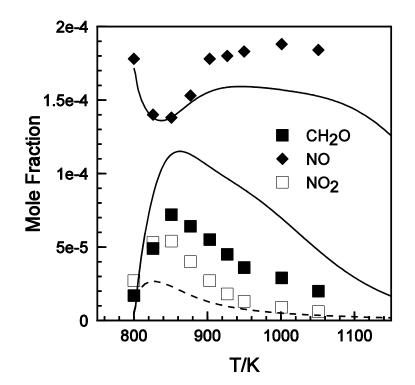




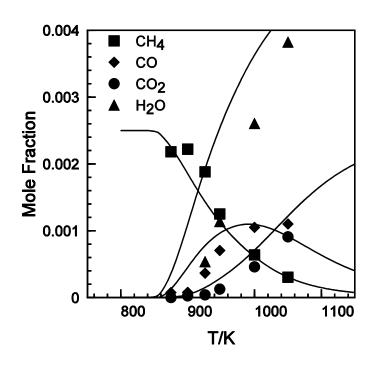


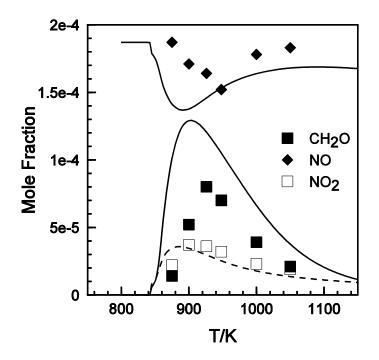
The mutual sensitization of the oxidation of methane and NO in a JSR at 10 atm: Effect of the introduction of **200ppm of NO** on the oxidation of methane in fuel-lean conditions ( $\varphi$ =0.5, 2500 ppm of CH<sub>4</sub>, 10000 ppm of O<sub>2</sub>, t=1000 ms). (a)The NO<sub>x</sub> experimental results are compared to the computations. (b) and (c):The filled symbols and the continuous lines refer respectively to the data and simulations with NO added; the open symbols and dotted lines refer respectively to the data and simulations without NO.



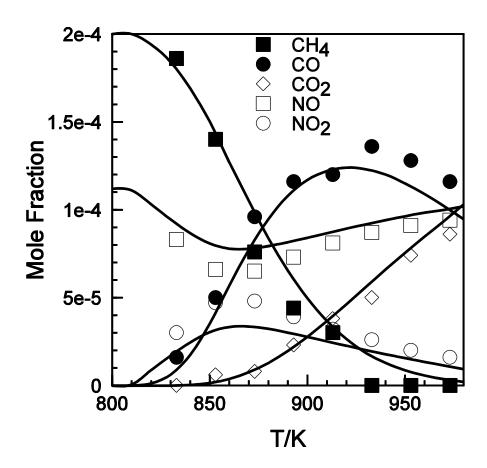


The mutual sensitization of the oxidation of methane and NO in a JSR at 10 atm (200 ppm of NO,  $\varphi$ =1, 2500 ppm of CH<sub>4</sub>, 5000 ppm of O<sub>2</sub>, t=1000 ms). Comparison between modeling (lines) and experiments (symbols).

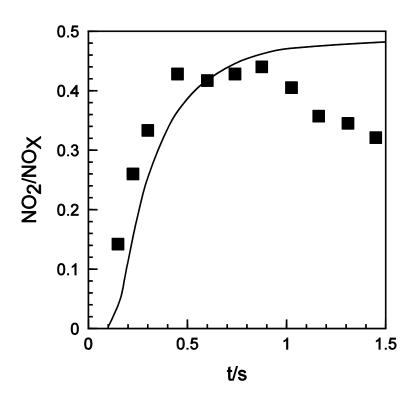


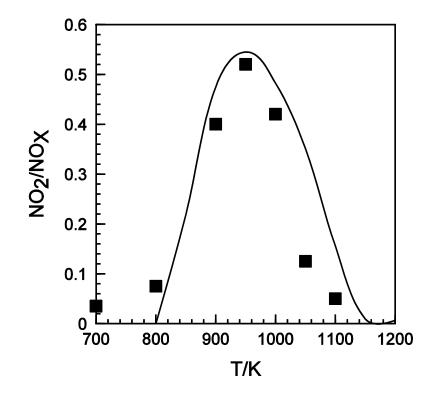


The mutual sensitization of the oxidation of methane and NO in a JSR at 10 atm (200 ppm of NO,  $\varphi$ =0.5, 2500 ppm of CH<sub>4</sub>, 10000 ppm of O<sub>2</sub>, t=240 ms). Comparison between modeling (lines) and experiments (symbols).

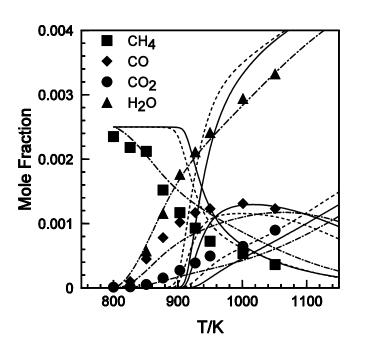


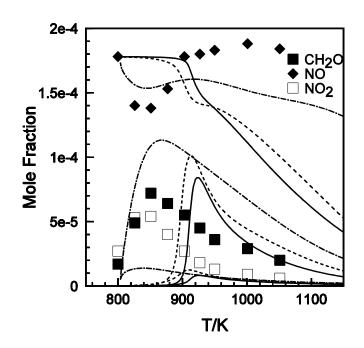
The mutual sensitization of the oxidation of methane and NO in a tubular flow reactor at 1 atm (112 ppm of NO, 200 ppm of CH<sub>4</sub>, 5% of O<sub>2</sub>, t=2.8 s). Comparison between modeling (lines) and experiments (symbols).



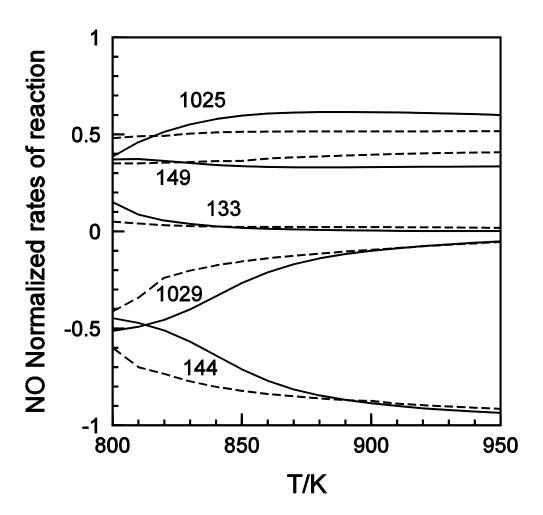


Comparison between this modeling (lines) and experimental data (symbols) obtained in a **tubular flow reactor** at 1000 K [Hori 1998] (initial conditions: 20 ppm of NO and 50 ppm of methane in air).



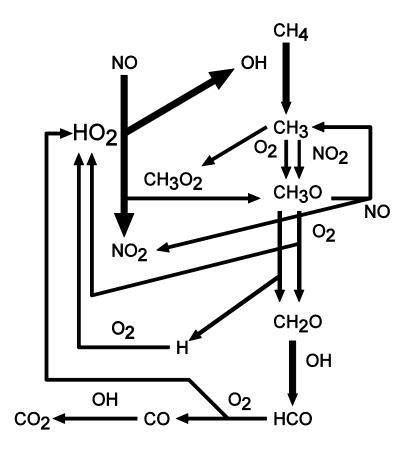


The mutual sensitization of the oxidation of methane and NO in a JSR at 10 atm (200ppm of NO,  $\varphi$ =1, 2500 ppm of CH<sub>4</sub>, 5000 ppm of O<sub>2</sub>, t=1000 ms). Comparison between **several modeling** results using the mechanism and thermochemical data of [Hori 1998] (continuous line), [Hori 2002] (dashed lines), [Faravelli 2003] (dash-dot line) and experiments (symbols).

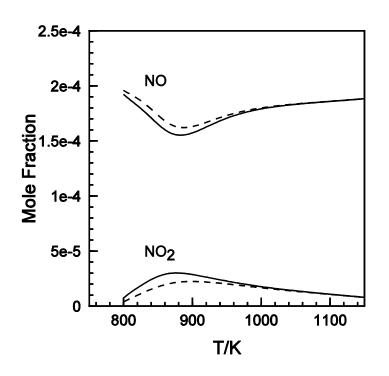


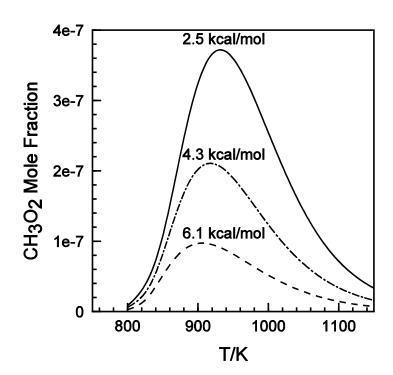
Normalized rates of reaction of NO at 1 atm (continuous lines) and 10 atm (dotted lines).

HNO+ NO2 = NO + HONO (133); NO + HO<sub>2</sub> = NO<sub>2</sub> + OH (144); NO<sub>2</sub> + H = NO + OH (149); CH3+NO2=CH3O+NO (1025); CH3O2+NO=CH3O+NO2 (1029)



Schematic representation of the reaction paths involved in the mutual sensitization of the oxidation of methane and NO.

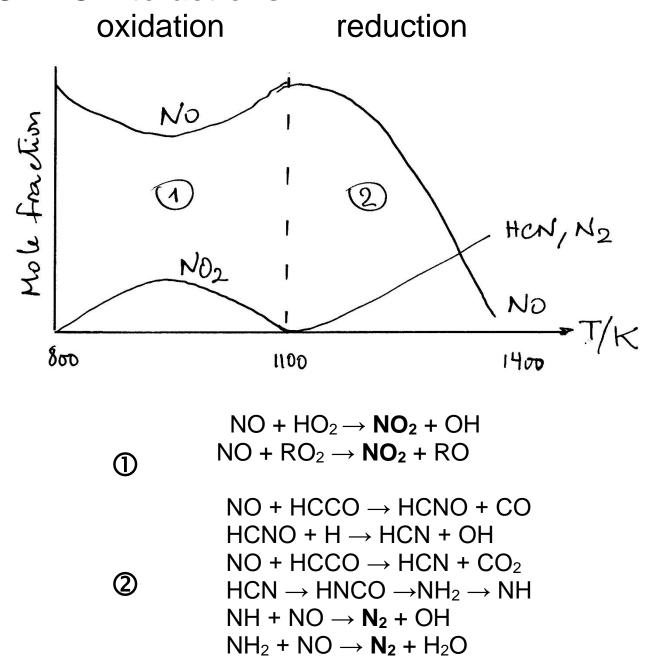




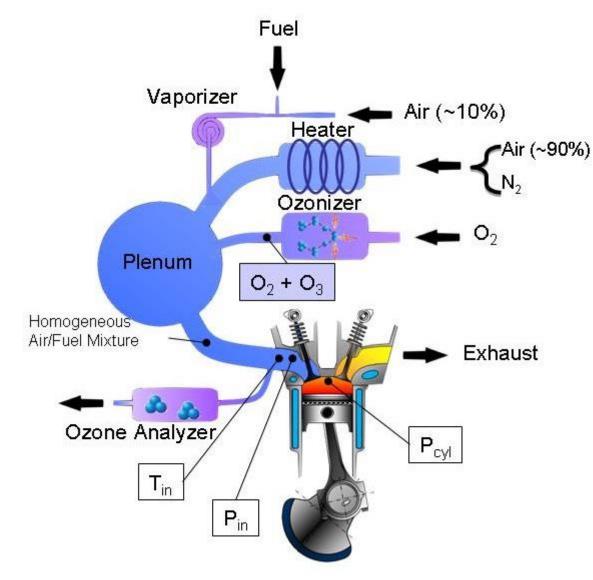
Sensitivity of the computations to the heat of formation of the methylperoxy radical (initial conditions: 2500 ppm of methane, 50000 ppm of oxygen, 200 ppm of nitric oxide,  $\varphi$ =0.1, 240 ms). The upper value of  $\Delta H^{\circ}_{298}$  (CH<sub>3</sub>O<sub>2</sub>) used was 6.1 kcal/mole (dashed lines) and the lower value was 2.5 kcal/mole (continuous lines).

M. Hori, N. Matsunaga, N.M. Marinov, J.W. Pitz, C.K. Westbrook, Proc. Combust. Inst. 27 (1998) 389-396. M. Hori, Y. Koshiishi, N. Matsunaga, P. Glaude, N. Marinov, Proc. Combust. Inst. 29 (2002) 2219-2226. T. Faravelli, A. Frassoldati, E. Ranzi, Combust. Flame 132 (2003) 188-207.

# **Overview of NOx-HC interactions**



# HCCl control via Sensitization by ozone $(O_3 \rightarrow O_2 + O)$



#### **Engine Characteristics**

Bore	85 mm	
Stroke	88 mm	
Displaced Volume	499 cc	
Connecting Rod	145 mm	
Compression Ratio	16:1	

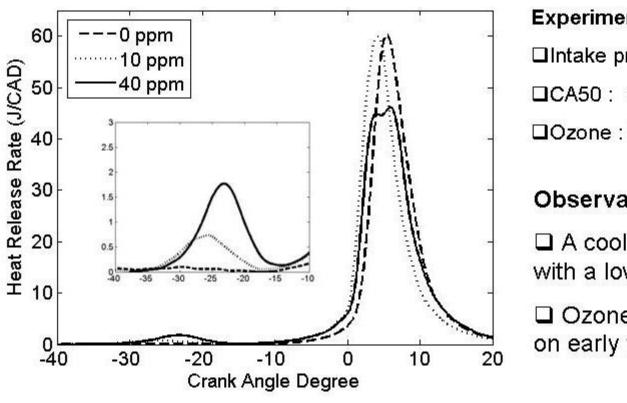
All the results presented were conducted for :

- □Constant rotation speed : 1500 rpm
- □Constant equivalence ratio 0.3

Masurier et al., ICE2013

# Results: HCCl control via Sensitization by ozone ( $O_3 \rightarrow O_2 + O$ )

# Heat release rates analysis at low temperature with ozone seeding



#### **Experimental conditions:**

□Intake pressure: 1.3 bar

□CA50: ~5 CAD

□Ozone: 0, 10 and 40 ppm

#### Observations:

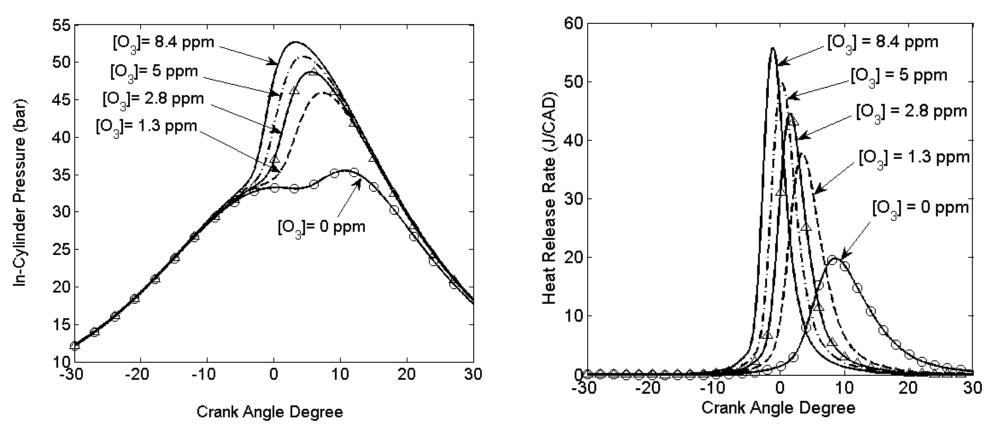
- ☐ A cool flame occurs with a low HRR.
- Ozone mainly acts on early fuel oxidation

Masurier et al.. ICE2013

 $O_2$  + fuel  $\rightarrow$   $HO_2$  +R followed by  $HO_2$  + fuel  $\rightarrow$   $H_2O_2$  +R  $W/O O_3$ : (slow)

with O<sub>3</sub>: O + fuel  $\rightarrow$  OH +R followed by OH + fuel  $\rightarrow$  H<sub>2</sub>O +R (FAST)

# Results: HCCl control via Sensitization by ozone ( $O_3 \rightarrow O_2 + O$ ) In-cylinder pressures and Heat release rates

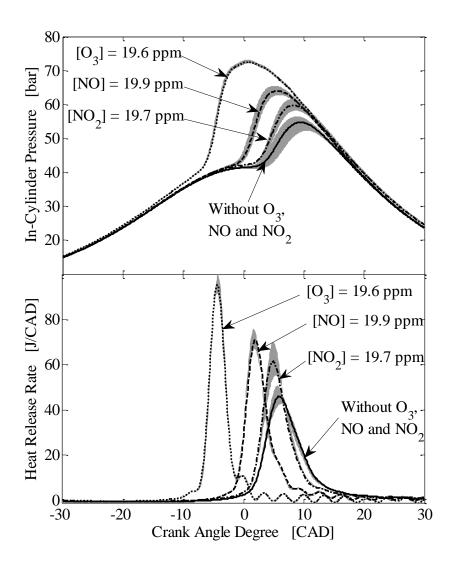


Masurier et al., ICE2013

w/o O<sub>3</sub>:  $O_2$  + fuel  $\rightarrow HO_2$  +R followed by  $HO_2$  + fuel  $\rightarrow H_2O_2$  +R (slow)

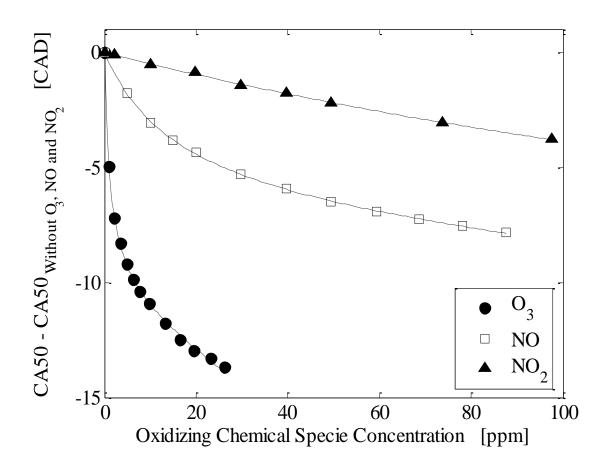
with O<sub>3</sub>:  $O + fuel \rightarrow OH + R$  followed by  $OH + fuel \rightarrow H_2O + R$  (FAST)

# Results: HCCI control via Sensitization by Ozone, NO, and NO<sub>2</sub>



In-cylinder pressure and heat release rate traces without any species and with 20 ppm of each species separately injected. *Masurier et al., SIC 35/ PROCI 2015* 

# HCCI control via Sensitization by Ozone, NO, and NO<sub>2</sub>



Shift of the CA50 as a function of the three species when they are separately injected. (**CA50** is the crank angle where **50** % **of the fuel has burned**) *Masurier et al., SIC 35/ PROCI 2015* 

Effect on CA50: O<sub>3</sub>>>NO>NO<sub>2</sub>

# HCCI control via Sensitization by Ozone, NO, and NO<sub>2</sub>

# Simple computations to understand the process

• Ozone mainly decomposes into oxygen molecules (O<sub>2</sub>) and O-atoms, *FAST*. Then, the fuel reacts directly with O-atoms to yield OH radicals and rapid oxidation of the fuel ensues:  $C_8H_{18}+O\rightarrow C_8H_{17}+OH$  (a) followed by  $C_8H_{18}+OH\rightarrow C_8H_{17}+H_2O$  (b).



NO is mostly consumed by reaction with HO₂, resulting in the initial oxidation of the fuel via C<sub>8</sub>H<sub>18</sub>+O<sub>2</sub>→C<sub>8</sub>H<sub>17</sub>+HO₂, *SLOW*,
 OH radicals are produced via NO+HO₂→NO₂+OH, *FAST*.
 Subsequently, rapid fuel consumption can take place via (b) due to OH production. Consequently, as nitric oxide requires an HO₂ radical to yield an OH radical, this explains the lower effect of NO on ignition delays

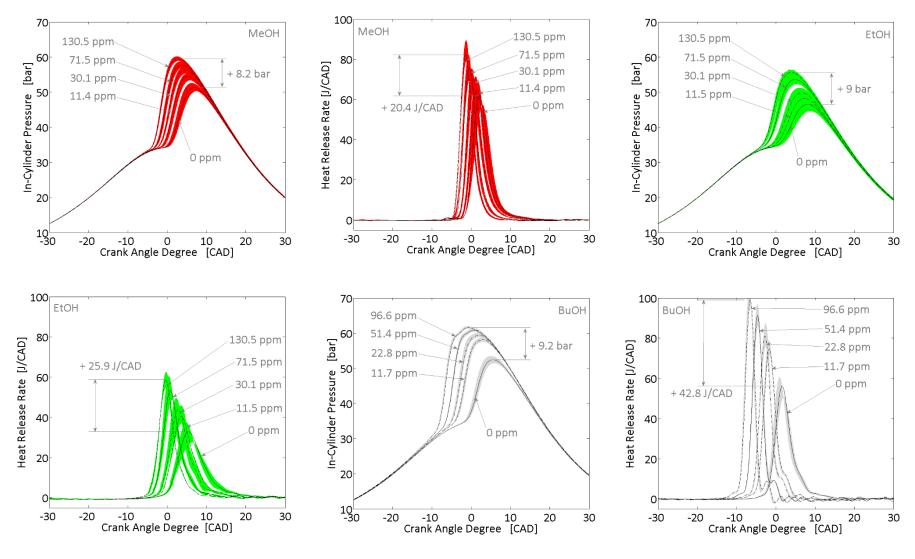
compared to ozone.



• **Nitrogen dioxide** addition: OH production results from the following reaction system:  $CH_3+NO_2\rightarrow CH_3O+NO$ ;  $NO_2+HO_2\rightarrow HONO+O_2$ ;  $HONO+M\rightarrow NO+OH+M$ ; and  $NO+HO_2\rightarrow NO_2+OH$ . As nitrogen dioxide presents intermediate reactions before OH production, its **effect on ignition delays is the lowest** of the 3 additives considered.



# **HCCI** control via Sensitization by Ozone and NOx



In-cylinder pressure traces and heat release rate traces for **alcohols** as a function of the ozone input. Black curves correspond to the average of 100 cycles recorded and areas represent the variation over 100 cycles. *Masurier et al., Appl. Energ. 2016* 

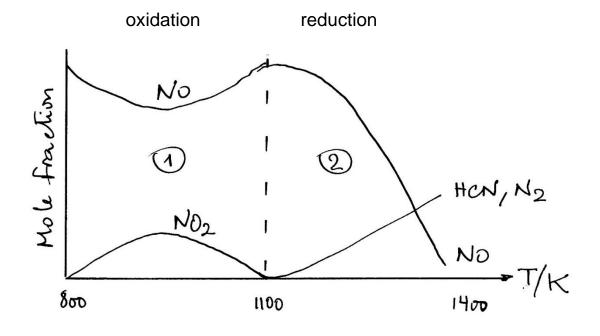
# **Summary**

NO<sub>x</sub> formation: Zeldovich, Prompt-NO, N<sub>2</sub>O, NNH, Fuel-NO

NO<sub>x</sub> reduction: SNCR, Reburning

**UHC** and soot

Effect of trace species on ignition: NOx, ozone. Enhanced oxidation rate by 'traces' of oxidants



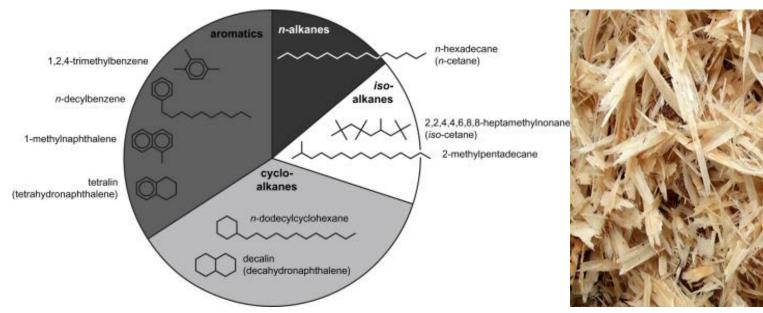
- $\begin{array}{ccc}
  NO + HO_2 \rightarrow NO_2 + OH \\
  O & NO + RO_2 \rightarrow NO_2 + RO
  \end{array}$ 
  - NO + HCCO → HCNO + CO
  - HCNO + H → HCN + OH

 $HCN \rightarrow HNCO \rightarrow NH_2 \rightarrow NH$ 

 $\mathsf{NH} + \mathsf{NO} \to N_2 + \mathsf{OH}$ 

 $NH_2 + NO \rightarrow N_2 + H_2O$ 

# Part 5 COMMERCIAL FUELS, SURROGATES, BIOFUELS





(from W.J.Pitz and C.J.Mueller)

# MODELING USING SURROGATES/MODEL-FUELS

Surrogate model fuels\*are used for the kinetic modeling to simplify the problem

**DCN**, **fuel composition** in terms of **chemical classes** and hydrocarbons concentrations, **H/C ratio**, and the **availability of valid chemical kinetic oxidation sub-models** are used to select the components of the model fuels.

**DCN** is a parameter related to fuel ignition

<sup>\*</sup>S. Dooley et al., Combust. Flame 157 (12) (2010) 2333-2339.

F.L. Dryer, Proc. Combust. Inst. 35 (2015) 117-144.

A. Agosta et al., Exp. Thermal Fluid Sci. 28 (7) (2004) 701-708.

# Modeling using surrogates/model-fuels

The **fuel composition** impacts the relative formation of products and intermediates.

The **fuel composition** impacts radical pool and cross-reactions

The **H/C ratio** is a parameter influencing soot formation.

Threshold sooting index (TSI) is a parameter related to soot tendency

Molecular weight is a parameter related to fuel diffusivity

# Validation of this approach needs extensive testing

# 5.1 Gasoline

Gasoline is constituted by several hundreds of components: it is not feasible to incorporate of all them in a kinetic model.

Therefore, surrogate model fuels are used to describe gasoline behavior.

In this example, 4 hydrocarbons of dominant gasoline chemical classes were chosen to represent a commercial gasoline:

iso-octane for iso-paraffins,

toluene for aromatics,

1-hexene for olefins,

ETBE for oxygenated additives.

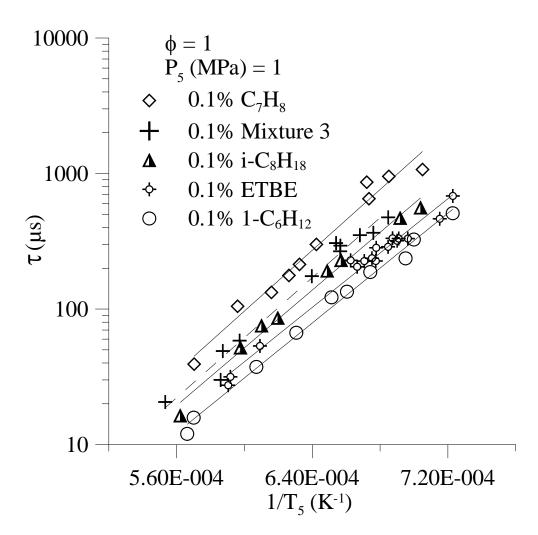
Mole fraction composition of the different surrogate gasoline mixtures

	Iso-octane	Toluene	1-hexene	ETBE
Mixture 1	50	35	15	0
Mixture 2	47.5	33.25	14.25	5
Mixture 3	45	31.5	13.5	10

Reactions of interaction between different fuel fragments during the oxidation of surrogate mixtures.

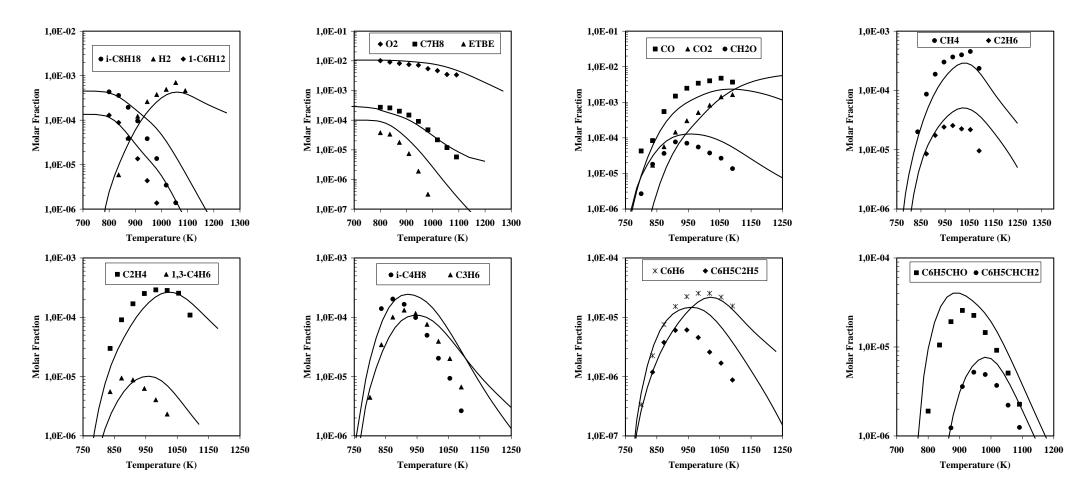
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1-C_6H_{12} + i-C_4H_7 = 1.3-C_6H_{11} + i-C_4H_8
1-C_6H_{12} + i-C_4H_7 = 1.4-C_6H_{11} + i-C_4H_8
1-C_6H_{12} + i-C_4H_7 = 1.5-C_6H_{11} + i-C_4H_8
1-C_6H_{12} + i-C_4H_7 = 1.6-C_6H_{11} + i-C_4H_8
1-C_6H_{12} + dimethyl 4.4-penthyl 3-ene = 1.3-C<sub>6</sub>H<sub>11</sub> + dimethyl 4.4-pentene
1-C_6H_{12} + dimethyl 4,4-penthyl 1-ene 2 = 1,4-C<sub>6</sub>H<sub>11</sub> + dimethyl 4,4-pentene
1-C_6H_{12} + dimethyl 2,4-penthyl 1-ene 2 = 1,5-C<sub>6</sub>H<sub>11</sub> + dimethyl 2,4-pentene2
1-C_6H_{12} + C_4H_5O = 1.3-C_6H_{11} + OC_4H_6
1-C_6H_{12} + C_4H_5O = 1.4-C_6H_{11} + OC_4H_6
1-C_6H_{12} + C_4H_5O = 1.5-C_6H_{11} + OC_4H_6
1-C_6H_{12} + C_4H_5O = 1.6-C_6H_{11} + OC_4H_6
i-C_8H_{18} + 1.3-C_6H_{11} = 1-C_6H_{12} + trimethyl 2.2.4-penthyl
i-C_8H_{18} + 1.3-C_6H_{11} = 1-C_6H_{12} + trimethyl 2.2.4-penthyl-4
i-C_8H_{18} + 1.3-C_6H_{11} = 1-C_6H_{12} + trimethyl 2.2.4-penthyl-3
i-C_8H_{18} + n-C_3H_7 = C_3H_8 + trimethyl 2,2,4-penthyl
i-C_8H_{18} + n-C_3H_7 = C_3H_8 + trimethyl 2.2.4-penthyl-4
i-C_8H_{18} + n-C_3H_7 = C_3H_8 + trimethyl 2,2,4-penthyl-3
i-C_8H_{18} + n-C_3H_7 = C_3H_8 + trimethyl 2.2.4-penthyl-3
i-C_8H_{18} + n-C_3H_7 = C_3H_8 + trimethyl 2,4,4-penthyl
1.3-C_6H_{11} + C_7H_8 = 1-C_6H_{12} + C_6H_5CH_2
1.4 - C_6H_{11} + C_7H_8 = 1 - C_6H_{12} + C_6H_5CH_2
1.5 - C_6 H_{11} + C_7 H_8 = 1 - C_6 H_{12} + C_6 H_5 C H_2
1.6 - C_6 H_{11} + C_7 H_8 = 1 - C_6 H_{12} + C_6 H_5 C H_2
1.3-C_6H_{11} + C_6H_5CH_2 = C_7H_8 + 1.2-C_6H_{10}
1.4 - C_6H_{11} + C_6H_5CH_2 = C_7H_8 + 1.3 - C_6H_{10}
1.6 - C_6 H_{11} + C_6 H_5 C H_2 = C_7 H_8 + 1.6 - C_6 H_{10}
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M. Yahyaouiet al., Proc. Combust. Inst. 31, 385-391 (2007)



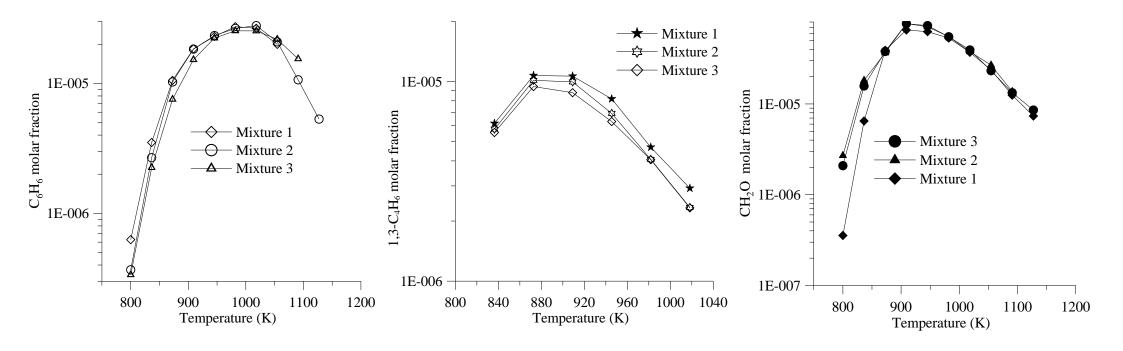
Comparison between neat hydrocarbons ignition delay times and mixture 3 (45% iso-octane, 31.5 toluene, 13.5% 1-hexene, 10% ETBE).

M. Yahyaouiet al., Proc. Combust. Inst. 31, 385–391 (2007)



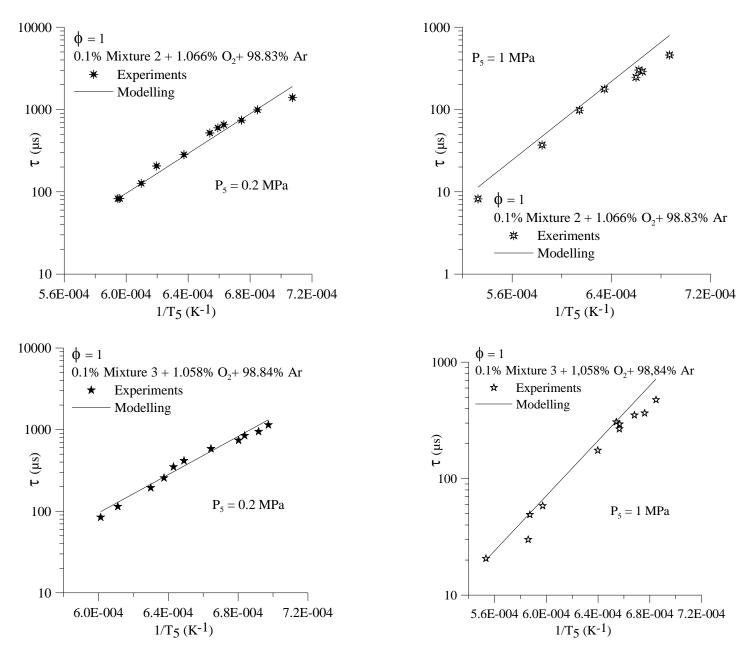
Comparison between experimental and computed concentration profiles in JSR for the oxidation of the Mixture 3 (45% iso-octane, 31.5 toluene, 13.5% 1-hexene, 10% ETBE)

M. Yahyaouiet al., Proc. Combust. Inst. 31, 385-391 (2007)



Experimental mole fractions of benzene, 1,3-butadiene and formaldehyde obtained from the oxidation of different initial fuel composition versus temperature in a JSR. (45% iso-octane, 31.5 toluene, 13.5% 1-hexene, 10% ETBE). Mix 1: 0% ETBE; Mix 2: 5% ETBE; Mix 3: 10% ETBE.

M. Yahyaouiet al., Proc. Combust. Inst. 31, 385-391 (2007)

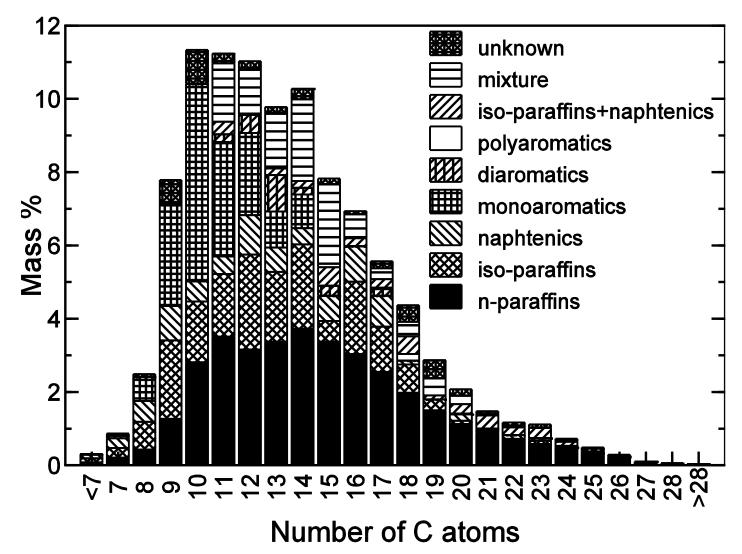


Comparison between experimental (symbols) and computed (lines) ignition delays of the Mixture 2 and 3 in a shock tube M. Yahyaouiet al., Proc. Combust. Inst. 31, 385–391 (2007)

### 5.2 Diesel

Example of Diesel oxidation study: The major components of the diesel fuel studied were n-paraffins (36.6% by weight), i-paraffins (14.8% w), cycloalkanes (31.4% w) and aromatic hydrocarbons (17.3% w) including mono- and poly-aromatic hydrocarbons.

The global formula for this diesel fuel was determined to be C15.5H30.



The reaction mechanism consisted of 2755 reversible reactions involving 377 species.

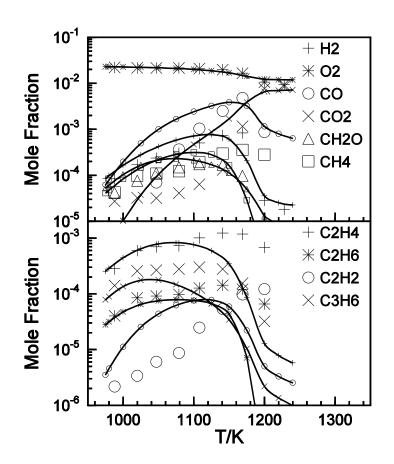
The rates of reaction were computed from the kinetic reaction mechanism and the rate constants calculated at the experimental temperature. The rate constants for the reverse reactions were computed from the forward rate constants and the appropriate equilibrium constants.

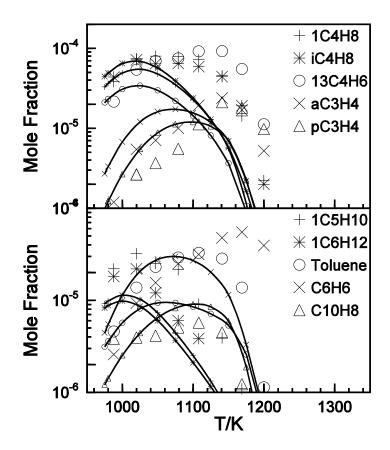
The pressure dependencies of P-dependent reactions were taken into account and updated.

The oxidation mechanism for the diesel fuel was obtained by merging the individual oxidation mechanisms previously validated for the oxidation of n-hexadecane, iso-octane, n-propylcyclohexane, n-propylbenzene, and 1-methylnaphtalene.

Few 'coupling reactions' were included whereas no specific kinetic adjustments were made to better fit pressure dependences. As in previous work from this group, the proposed kinetic mechanism has a strong hierarchical structure.

The **model-fuel had 4 constituents**: n-hexadecane (36.1% by weight, 23.5% vol.), n-propylcyclohexane (23.1%w, 26.9% vol.), n-propylbenzene (18.7% w, 22.9% vol.), iso-octane (14.7% w, 19% vol.), and 1-methylnaphthalene (7.4%w, 7.7% vol.).

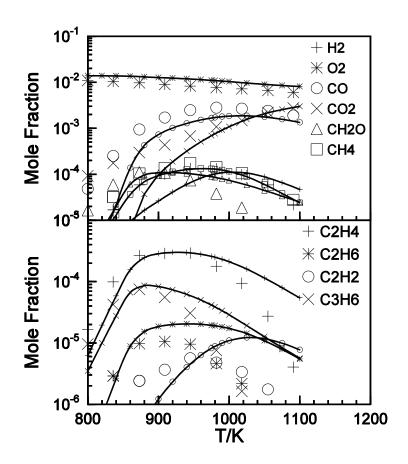


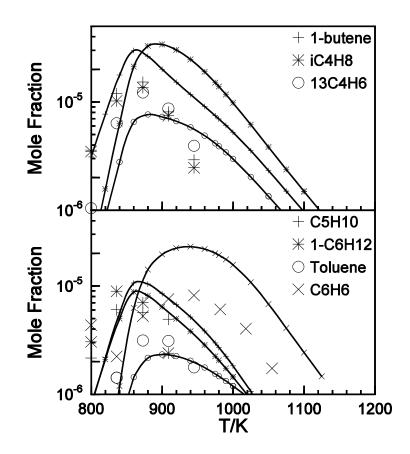


Synthetic diesel fuel oxidation in a JSR at 1 atm and  $\varphi$  = 0.5. The initial conditions were:

 $C_{15.5}H_{30},\ 0.03\%\ ;\ O_2,\ 2.30\%\ ;\ N_2,\ 97.60\%\ ;\ \tau{=}0.1s.$ 

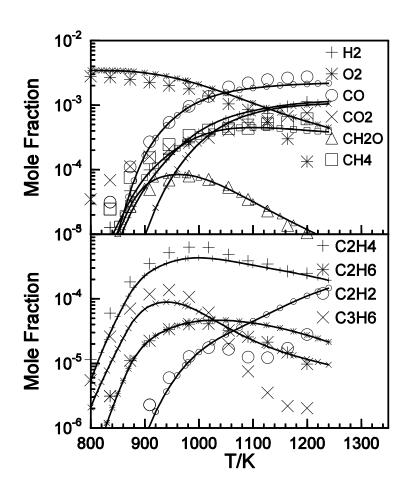
The experimental data (symbols) are compared to the computations (lines and small symbols).

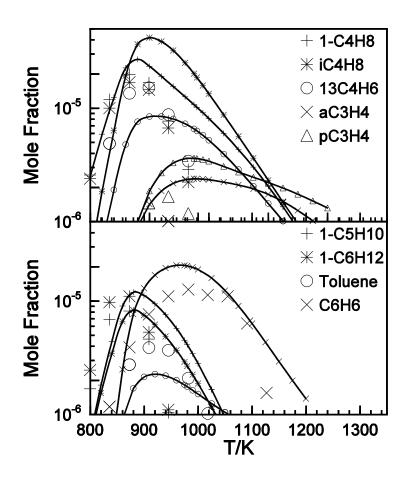




Synthetic diesel fuel oxidation in a JSR at 10 atm and  $\varphi$  =0.5.

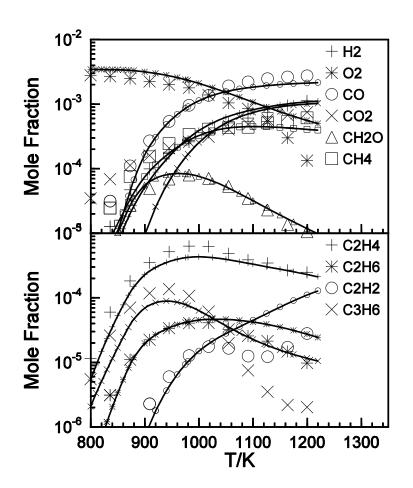
The initial conditions were:  $C_{15.5}H_{30}$ , 0.05%;  $O_2$ , 1.38%;  $N_2$ , 98.57%;  $\tau$ =0.5s.

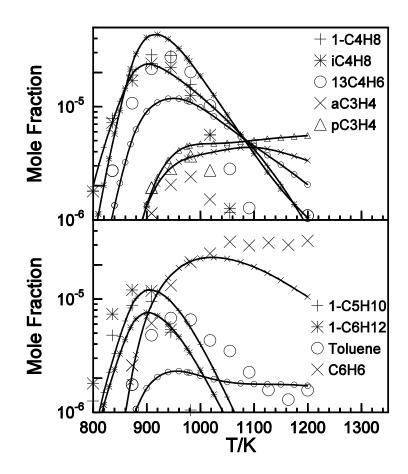




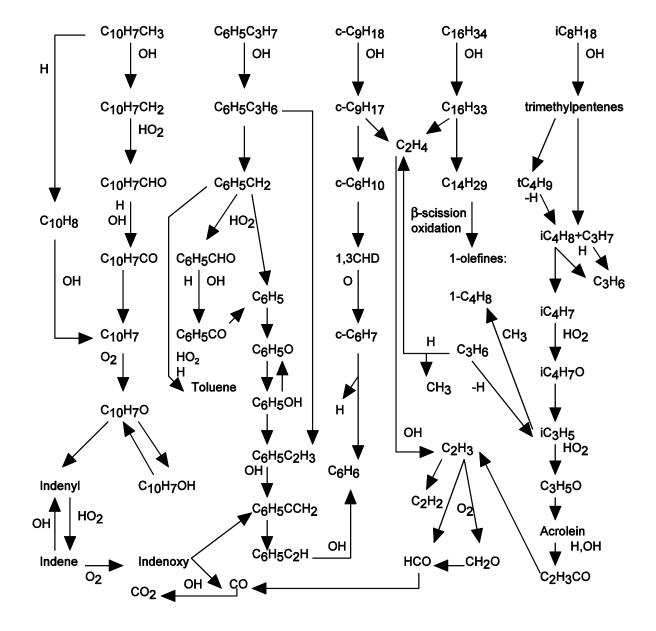
Synthetic diesel fuel oxidation in a JSR at 10 atm and  $\phi$  =1.0.

The initial conditions were:  $C_{15.5}H_{30}$ , 0.05%;  $O_2$ , 0.69%;  $N_2$ , 99.26%;  $\tau$ =0.5s.





Synthetic diesel fuel oxidation in a JSR at 10 atm and  $\phi$  =2. The initial conditions were:  $C_{15.5}H_{30}$ , 0.05%;  $O_2$ , 0.345%;  $N_2$ , 99.6%;  $\tau$ =0.5s.

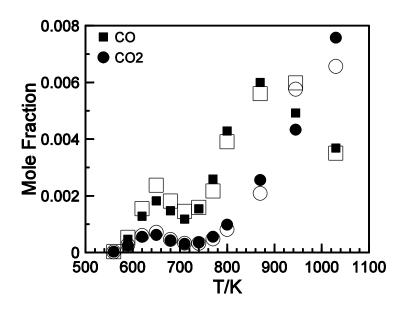


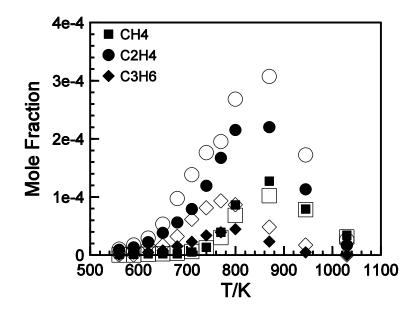
Main reaction paths during the oxidation of the model fuel in a JSR at 10 atm, 1000 K, and  $\phi$ =1.0. Initial conditions: C<sub>15.5</sub>H<sub>30</sub>, 0.05%; O<sub>2</sub>, 0.69%; N<sub>2</sub>, 99.26%;  $\tau$ =0.5s.

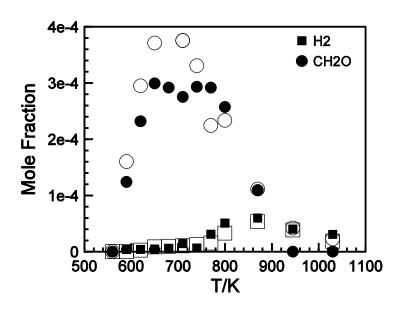
## 2<sup>nd</sup> example of Diesel oxidation study:

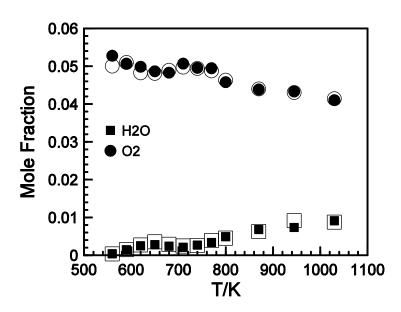
A surrogate Diesel fuel called the IDEA fuel, consisting of 70% n-decane and 30% 1-methyl naphthalene was formulated previously as part of the Integrated Development on Engine Action (IDEA) program. This fuel mixture matches both the physicochemical properties and combustion behavior of a conventional Diesel fuel. The IDEA fuel has properties similar to those of a conventional Diesel fuel, i.e. it has a normal density of 798 kg/m3 at 20°C, a CN of ca. 53, and hydrogen-to-carbon ratio of 1.8.

The kinetic oxidation mechanisms of large n-paraffins and aromatics have been developed separately in several fundamental studies and merged to simulate the oxidation of surrogate gasoline, kerosene, and Diesel fuels. A long carbon chain n-paraffin compound is highly suitable for representing the paraffinic fraction of a Diesel fuel because of the high concentration of these chemicals in this kind of fuel. On the other hand, aromatic hydrocarbons play an important role in soot formation reactions and must be used in Diesel surrogate mixtures. They also contribute to the reduction of the cool-flame oxidation of long chain n-alkanes.

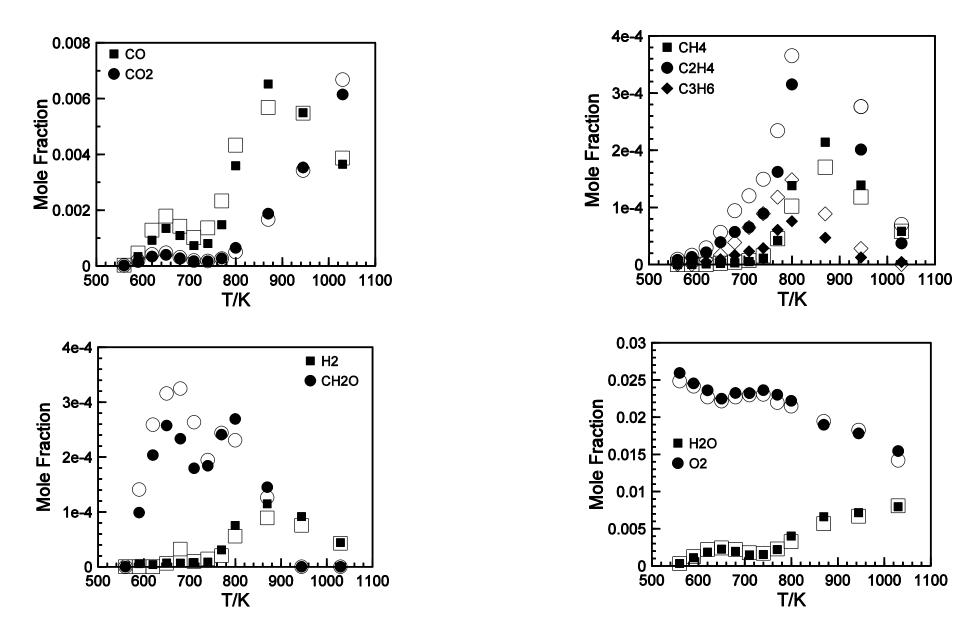




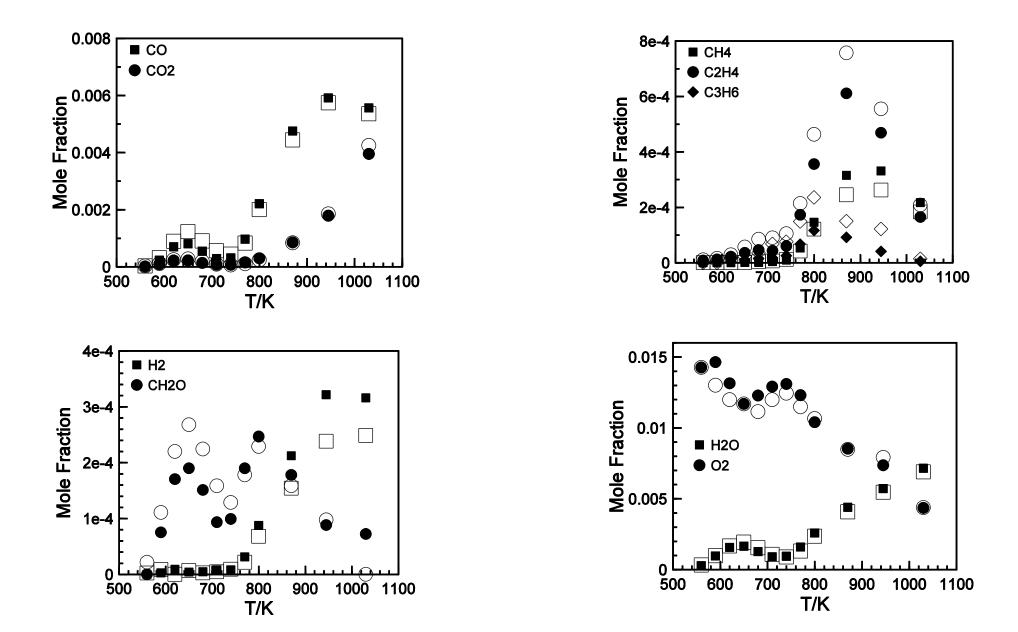




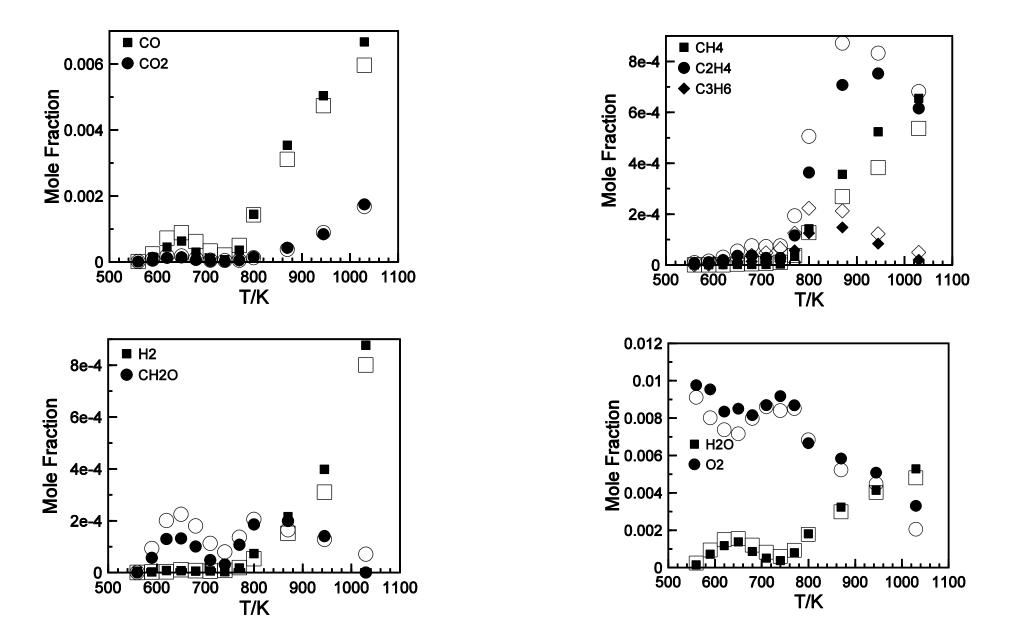
Experimental species concentration profiles from the oxidation of the conventional (filled symbols) and IDEA surrogate (empty symbols) Diesel fuels in a JSR at 10 atm,  $\phi$ =0.25 and  $\tau$ = 1s.



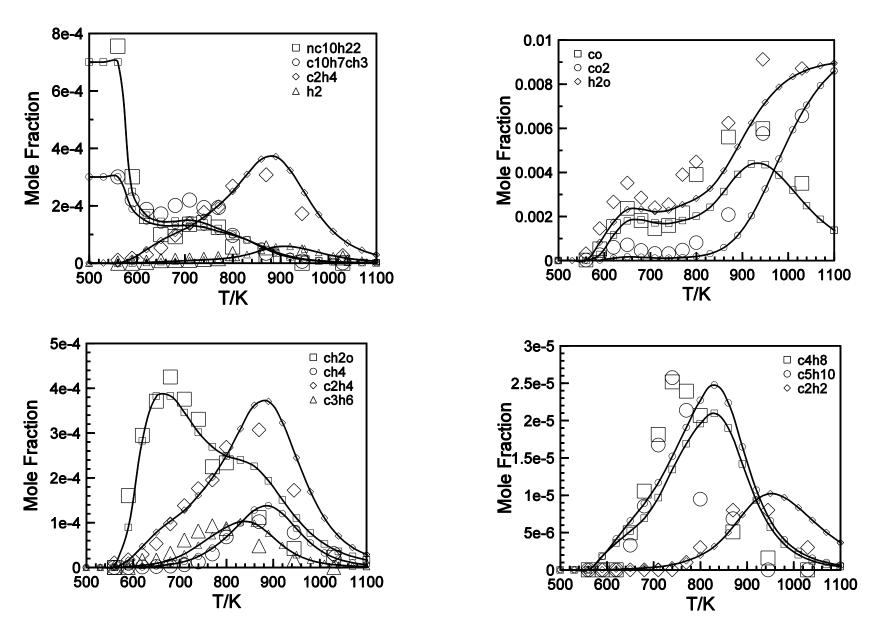
Experimental species concentration profiles from the oxidation of the conventional (filled symbols) and IDEA surrogate (empty symbols) Diesel fuels in a JSR at 10 atm,  $\varphi$ = 0.5 and  $\tau$ = 1s.



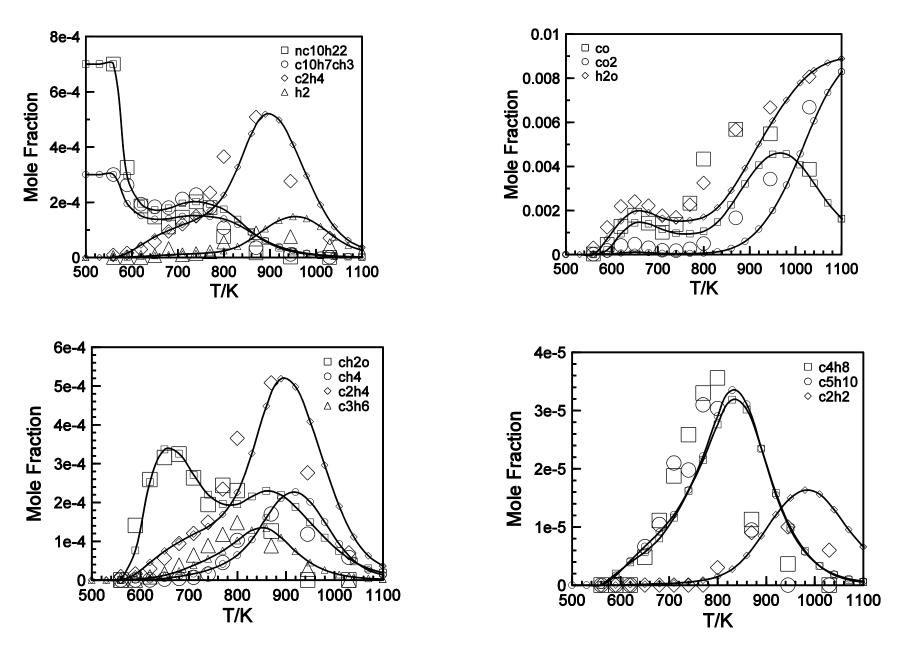
Experimental species concentration profiles from the oxidation of the conventional (filled symbols) and IDEA surrogate (empty symbols) Diesel fuels in a JSR at 10 atm,  $\varphi$ = 1 and  $\tau$ = 1s.



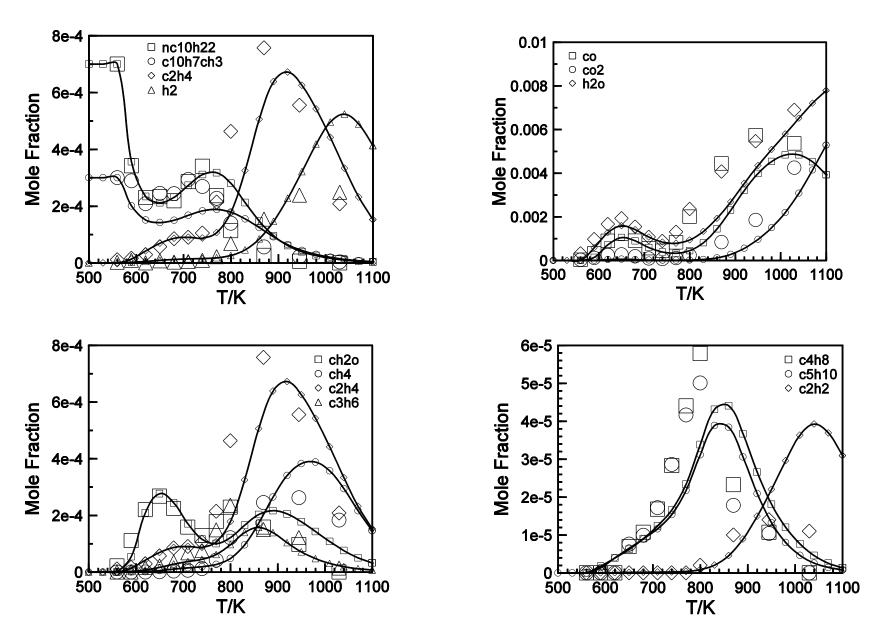
Experimental species concentration profiles from the oxidation of the conventional (filled symbols) and IDEA surrogate (empty symbols) Diesel fuels in a JSR at 10 atm,  $\varphi$ = 1.5 and  $\tau$ = 1s.



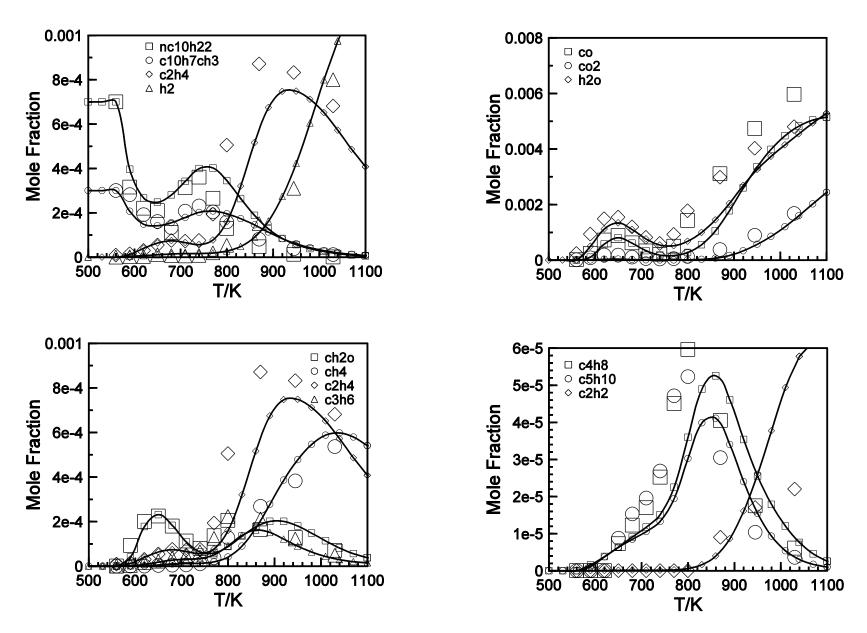
The IDEA surrogate Diesel fuel oxidation in a JSR at 10 atm,  $\tau$ = 1s and  $\phi$ = 0.25. The initial mole fractions were: 1 -Methylnaphthalene, 0.03%; n-Decane, 0.07%; O<sub>2</sub>, 5.96%; N<sub>2</sub>, 93.94%. The experimental data (filled symbols) are compared to the computations (lines with empty symbols). H. P. Ramirez L et al., Energy & Fuels **24**(3) 1668–1676 (2010)



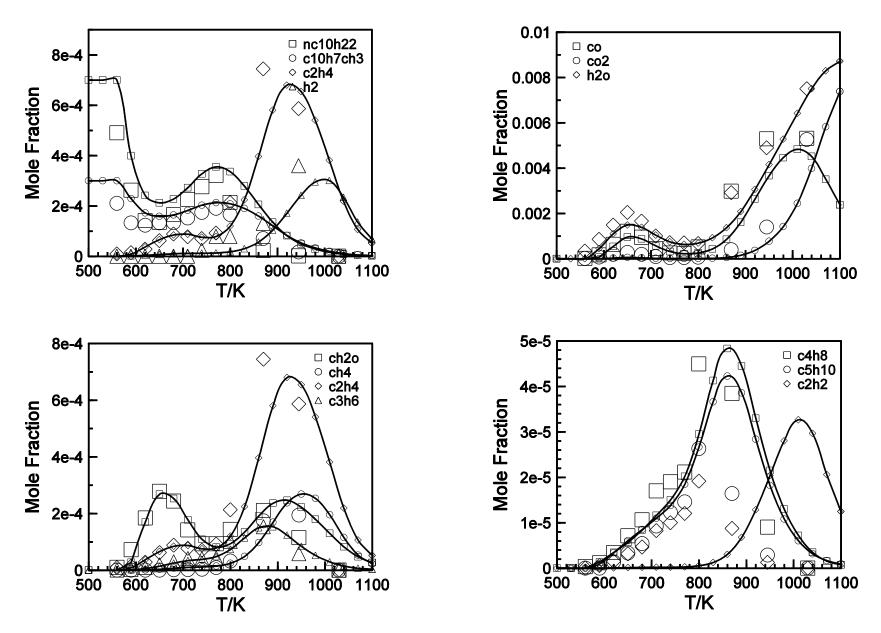
The IDEA surrogate Diesel fuel oxidation in a JSR at 10 atm,  $\tau$ = 1s and  $\phi$ = 0.5. The experimental data (filled symbols) are compared to the computations (lines with empty symbols).



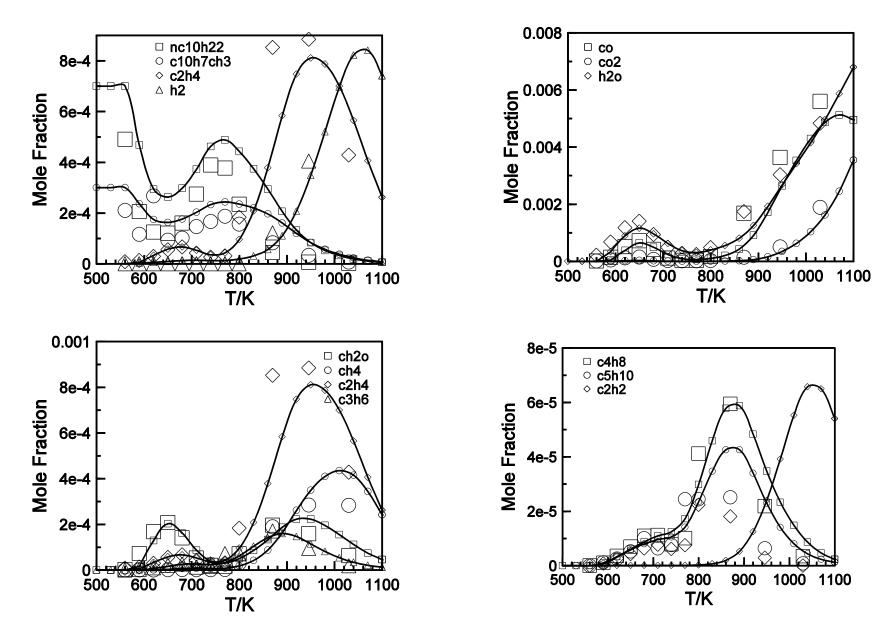
The IDEA surrogate Diesel fuel oxidation in a JSR at 10 atm,  $\tau$ = 1s and  $\phi$ = 1.0. The experimental data (filled symbols) are compared to the computations (lines with empty symbols).



The IDEA surrogate Diesel fuel oxidation in a JSR at 10 atm,  $\tau$ = 1s and  $\phi$ = 1.5. The experimental data (filled symbols) are compared to the computations (lines with empty symbols).



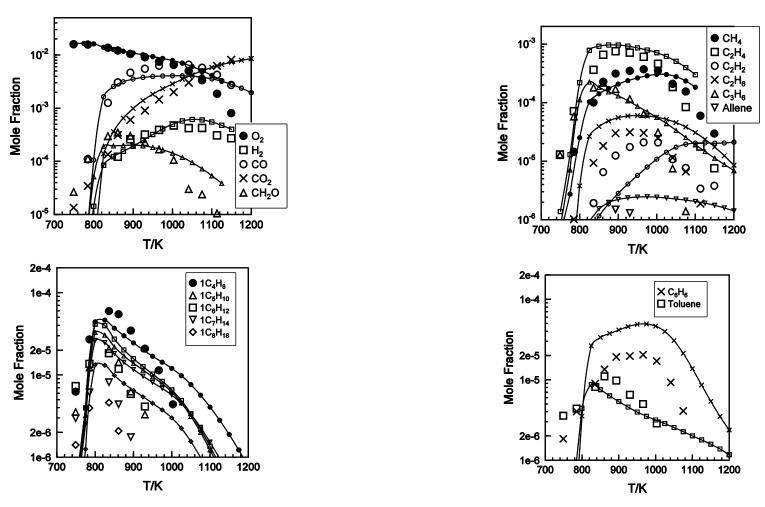
The IDEA surrogate Diesel fuel oxidation in a JSR at 6 atm,  $\tau$ = 0.6s and  $\phi$ = 0.5. The experimental data (filled symbols) are compared to the computations (lines with empty symbols).



The IDEA surrogate Diesel fuel oxidation in a JSR at 6 atm,  $\tau$ = 0.6s and  $\phi$ = 1.0. The experimental data (filled symbols) are compared to the computations (lines with empty symbols).

### 5.3 Jet fuels

Example of early jet fuel oxidation study:

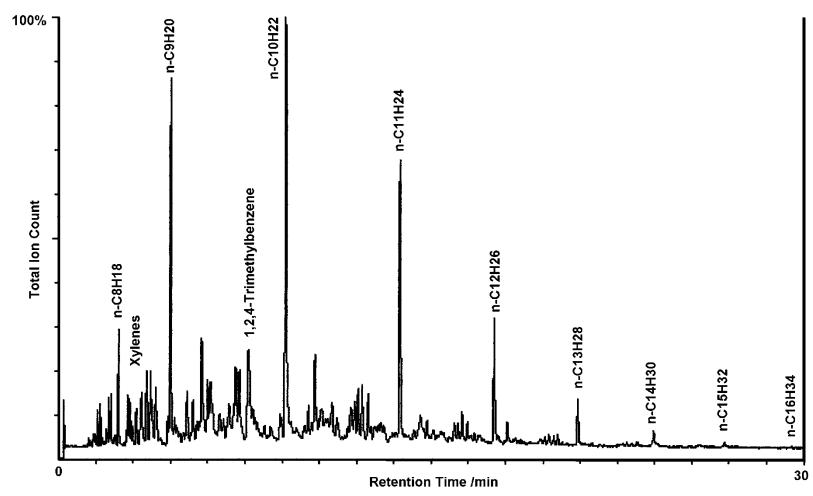


Oxidation of kerosene in a JSR at **10 atm** and t=0.5 s (initial conditions: 1000 ppmv of kerosene TR0, 16500 ppmv of O<sub>2</sub>, diluent nitrogen). Model fuel: n-decane/ n-propylbenzene/ n-propyl-cyclohexane (74% / 14% / 11% mole).

Dagaut & Cathonnet, PECS 32, 48-92, 2006

### Introduction

Kerosene (Jet A, Jet A1, JP-8, TR0) is a complex mixture of alkanes (50-65% vol.), mono- and polyaromatics (10-20% vol.) and cycloalkanes or naphtenes (mono- and polycyclic, 20-30% vol.) widely used in aircraft engines.



GC/MS analysis of a kerosene TR0 sample showing the importance of n-alkanes.

The compounds identified in kerosene at the highest levels of concentration are n-alkanes.

The average chemical formula for kerosene (Jet A, Jet A-1, TR0, JP-8) differs from on source to another:

C<sub>12</sub>H<sub>23</sub> in Gracia-Salcedo, C.M., Brabbs, T.A., and McBride, B.J., 1988, NASA Tech. Memorandum 101475,

C<sub>11</sub>H<sub>21</sub> in Edwards, T., and Maurice, L.Q., 2001, J. Propulsion and Power, **17**, 461-466,

C<sub>11.6</sub>H<sub>22</sub> in Martel, C.R., 1988, AFWAL/POSF Report, July 15, 1988

C<sub>11</sub>H<sub>22</sub> in Guéret, C., 1989, Thesis, University of Orléans (in French).

C<sub>11</sub>H<sub>23</sub> in Nguyen, H.L., and Ying, S.J., 1990, AIAA-90-2439.

For this study, the adopted formula was  $C_{11}H_{22}$ .

Due to the complexity of the composition of this fuel, it is necessary to use a surrogate model fuel for simulating its oxidation.

Under high-pressure JSR conditions, the detailed kinetic modeling of kerosene oxidation was initially performed using n-decane as a model-fuel, since n-decane and kerosene showed very similar oxidation rates under JSR and premixed flame conditions as reported in:

Dagaut et al., *Proc. Combust. Inst.*, **25**, pp 919-926, 1994.

Dagaut et al., *J. Chim. Phys. Phys.-Chim. Biol.* **92**, pp 47-76, 1995.

Cathonnet et al. RTO Meeting Proc. 14, pp 1-9, 1999.

Douté et al. *Combust. Sci. and Technol.* **106**, pp 327-344, 1995.

n-Decane is an acceptable model-fuel for kerosene oxidation as far as modeling the formation of aromatics is not a major issue since the oxidation of n-decane yields much less aromatics that kerosene.

Therefore, more complex model fuels are necessary to model the formation of aromatics from the oxidation of kerosene as demonstrated in the literature:

Mawid et al., 2002, AIAA 2002-3876.

Dagaut 2002, Phys. Chem. Chem. Phys., 4, 2079-2094.

Mawid et al. 2003, AIAA 2003-4938.

Mawid et al. 2004, AIAA 2004-4207.

Surrogate model fuels consisting of n-decane and mixtures of n-decane with simple aromatic hydrocarbons and cycloalkanes are tested here, mainly under JSR conditions.

The detailed kinetic reaction mechanisms for the pure components of the surrogate model fuel had first to be validated before merging the sub-schemes (Ristori et al. 2001, Combust. Sci. and Technol., **65**, pp 197-228; Dagaut et al. 2002, Fuel, **81**, pp 173-184) to yield a kerosene kinetic reaction mechanism

#### The study includes:

New experimental results obtained for the oxidation of kerosene in a JSR, over a wide range of equivalence ratio (0.5 to 2), and temperatures in the range 900-1300 K.

The oxidation of n-decane under JSR conditions shock-tube conditions premixed flame conditions,

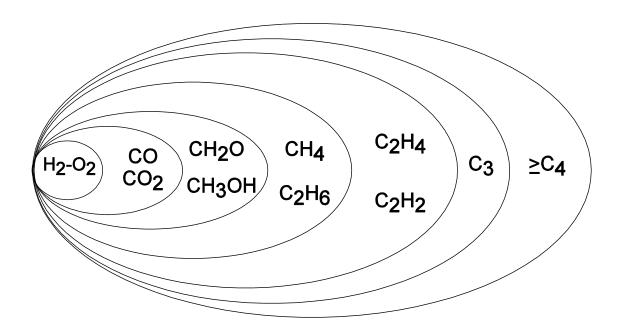
## Modeling

For simulating the oxidation of n-decane and kerosene in premixed flames, we used the Premix computer code.

For simulating the ignition delays of kerosene-air mixtures, we used the SENKIN code.

For the JSR computations, we used the PSR computer code.

The reaction rates are computed from the kinetic reaction mechanism and the rate constants of the elementary reactions calculated at the experimental temperature, using the modified Arrhenius equation.



The reaction mechanism used in this study has a strong hierarchical structure.

The reaction mechanism is based on the comprehensive commercial fuel oxidation mechanism developed earlier (Dagaut 2002, Phys. Chem. Chem. Phys., **4**, 2079-2094) where the rate expressions of pressure dependent reactions have been updated.

The reaction mechanism used here consisted of 209 species and 1673 reversible reactions.

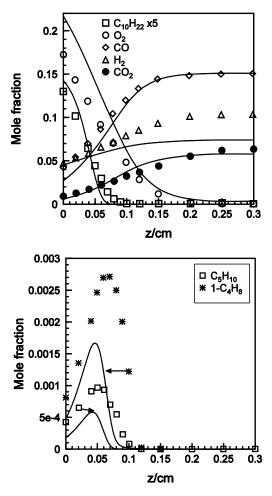
The rate constants for reverse reactions were computed from the corresponding forward rate constants and the appropriate equilibrium constants,

$$K_c = k_{forward} / k_{reverse}$$

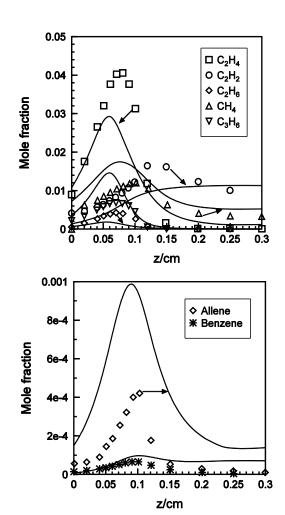
calculated using thermochemical data.

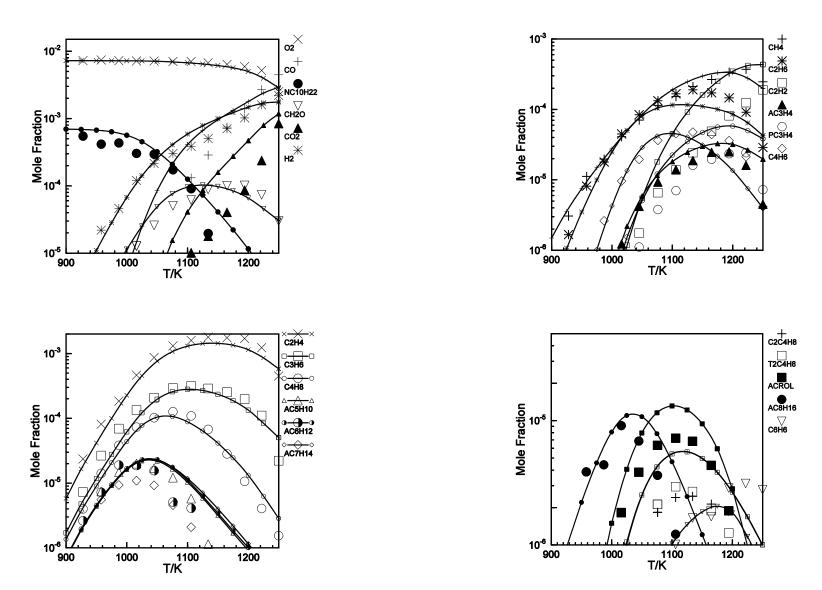
#### Results: n-decane

The kinetic model was tested against the atmospheric pressure n-decane premixed flame data of Douté et al. to verify the validity of the proposed kinetic scheme in flame conditions. The experimental temperature profile reported by the authors was used in the computations.



Dagaut & Cathonnet, PECS 32, 48-92, 2006





Oxidation of n-decane in a JSR: the experimental results consisted of the mole fractions of reactants, stable intermediates and final products measured at fixed residence time, as a function of T (example: 700 ppmv of n-decane, 7230 ppmv of  $O_2$ , in  $O_2$ , in  $O_2$ , in  $O_3$ , 1 atm).

Dagaut & Cathonnet, PECS 32, 48-92, 2006

#### Results: Kerosene

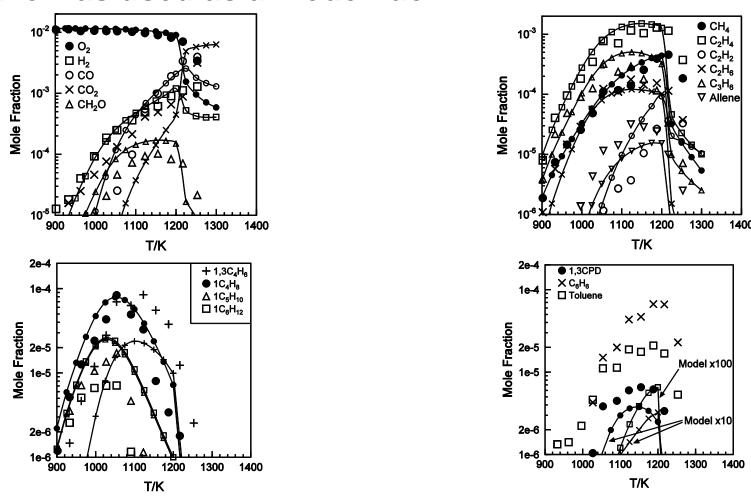
For the oxidation of kerosene in a JSR, the experimental results consisted of the mole fractions of the reactants, stable intermediates and final products measured at fixed residence time, as a function of temperature.

They are compared to PSR simulations.

To test the effect of the model fuel composition on the computations, we modeled the oxidation of a stoichiometric mixture kerosene/O<sub>2</sub>/N<sub>2</sub> using **four different model-fuels**:

- (1) n-decane
- (2) n-decane/n-propylbenzene (74% / 26% mole) mixture
- (3) n-decane/n-propylcyclohexane (74% / 26% mole) mixture
- (4) n-decane/ n-propylbenzene/ n-propyl-cyclohexane (74% / 14% / 11% mole) mixture

#### n-Decane was used as a model fuel:

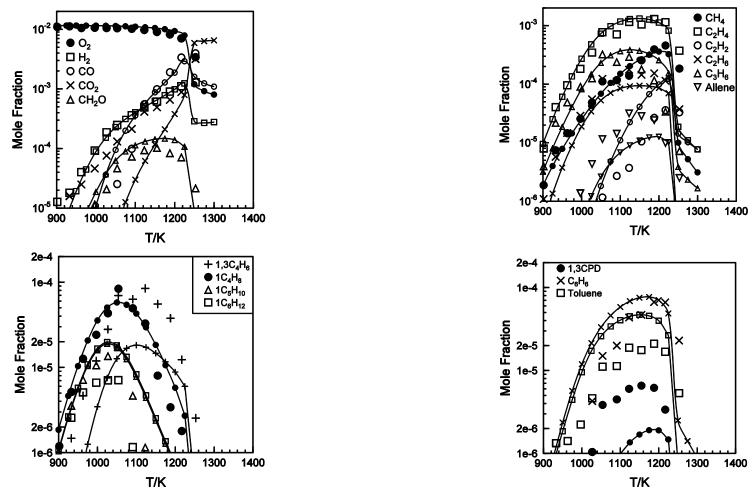


Kerosene oxidation of in a JSR (700 ppmv of kerosene, 11550 ppmv of O<sub>2</sub>, N<sub>2</sub>; 0.07 s, 1 atm).

- 1,3-Cyclopentadiene, benzene, and toluene are strongly underestimated!
- (1) These results confirm the similitude between n-decane and kerosene kinetics of oxidation
- (2) The inclusion of non-paraffin components in the model fuel is necessary to simulate the formation of aromatics from kerosene oxidation

Dagaut & Cathonnet, PECS 32, 48-92, 2006

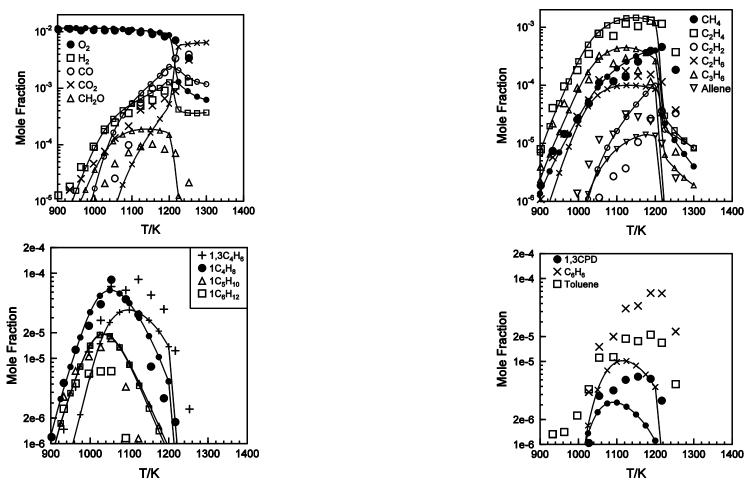
## n-decane/n-propylbenzene (74% / 26% mole) mixture as model fuel



Kerosene oxidation in a JSR (700 ppmv of kerosene, 11550 ppmv of O<sub>2</sub>, N<sub>2</sub>; 0.07 s, 1 atm).

A good agreement between the data and the modeling results for most of the species but 1,3-cyclopentadiene, benzene, and toluene: benzene and toluene are overestimated Thus the inclusion of cycloalkanes in the kerosene model fuel is necessary Dagaut & Cathonnet, *PECS* 32, 48-92, 2006

## n-decane/n-propylcyclohexane (74% / 26% mole) mixture as model fuel

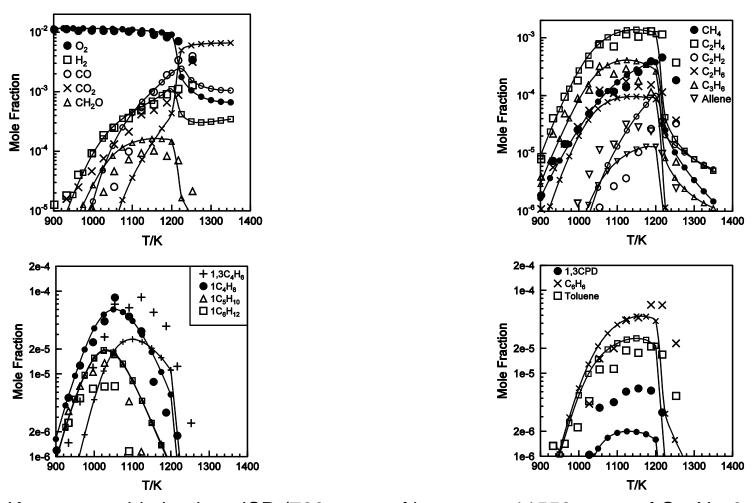


Kerosene oxidation in a JSR (700 ppmv of kerosene, 11550 ppmv of O<sub>2</sub>, N<sub>2</sub>; 0.07 s, 1 atm).

A good agreement between the data and the modeling for most of the species but benzene, and toluene which are strongly underestimated.

Expected: The oxidation of n-propylcyclohexane yields little benzene and toluene. Dagaut & Cathonnet, *PECS* **32**, 48-92, 2006

## n-decane/ n-propylbenzene/ n-propyl-cyclohexane (74% / 14% / 11% mole) as model fuel:



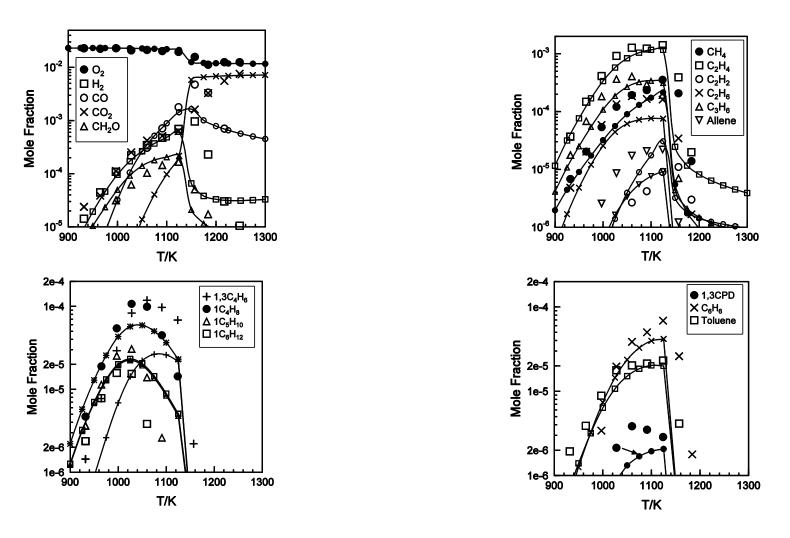
Kerosene oxidation in a JSR (700 ppmv of kerosene, 11550 ppmv of O<sub>2</sub>, N<sub>2</sub>; 0.07 s, 1 atm).

This mixture was more representative of the composition of kerosene: A good agreement between the data and the computational results for most of the species, including simple aromatics (benzene, toluene). Dagaut & Cathonnet, *PECS* 32, 48-92, 2006

n-decane/ n-propylbenzene/ n-propyl-cyclohexane (74% / 14% / 11% mole)

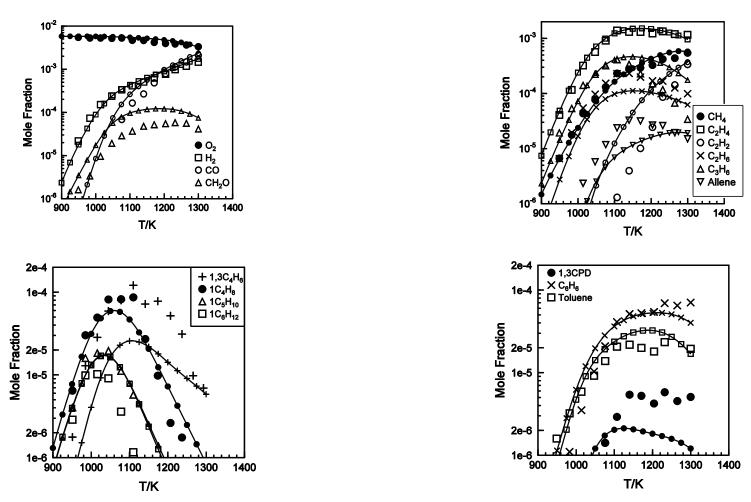
was selected for modeling the oxidation of kerosene in other experiments

# n-decane/ n-propylbenzene/ n-propyl-cyclohexane (74% / 14% / 11% mole) as model fuel:

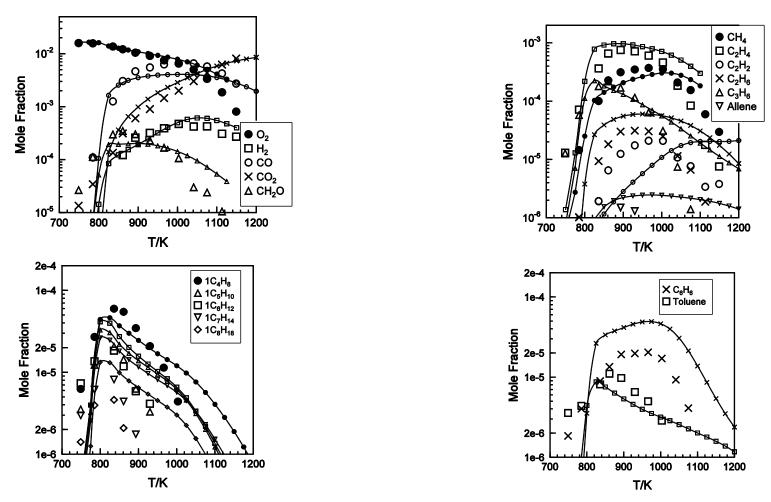


Kerosene oxidation (fuel lean) in a JSR (700 ppmv of kerosene, 23100 ppmv of O<sub>2</sub>, N<sub>2</sub>; 0.07 s, 1 atm). Dagaut & Cathonnet, *PECS* **32**, 48-92, 2006

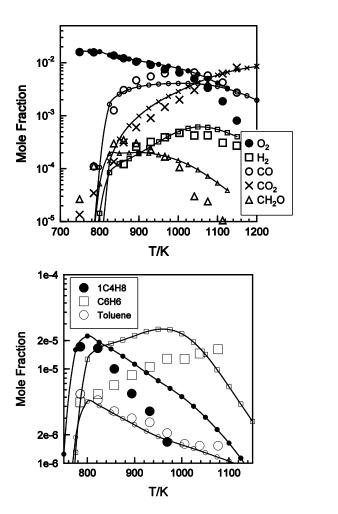
# n-decane/ n-propylbenzene/ n-propyl-cyclohexane (74% / 14% / 11% mole) as model fuel:

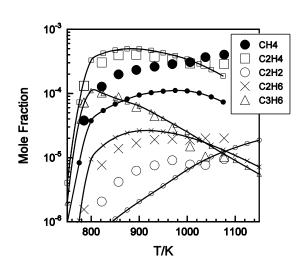


Kerosene oxidation (fuel rich) in a JSR (700 ppmv of kerosene, 5775 ppmv of  $O_2$ ,  $N_2$ ; 0.07 s, 1 atm). Dagaut & Cathonnet, *PECS* **32**, 48-92, 2006



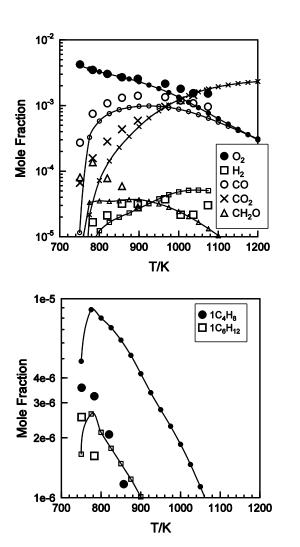
Oxidation of kerosene in a JSR at **10 atm** and t=0.5 s (initial conditions: 1000 ppmv of kerosene TR0, 16500 ppmv of O<sub>2</sub>, diluent nitrogen)
Dagaut & Cathonnet, *PECS* **32**, 48-92, 2006

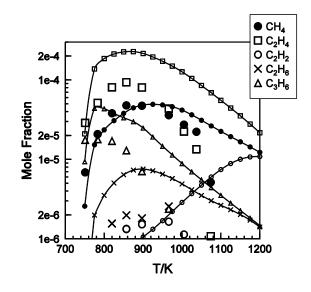




Oxidation of kerosene in a JSR (500 ppmv of kerosene, 8250 ppmv of oxygen, nitrogen diluent; 1.0 s, **20 atm**).

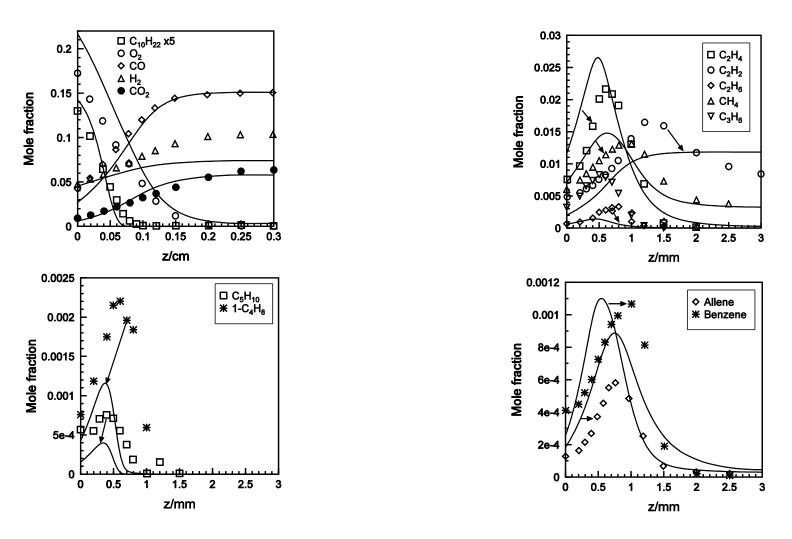
Dagaut & Cathonnet, PECS 32, 48-92, 2006





Oxidation of kerosene in a JSR at **40 atm** and t=2.0 s (initial conditions: 250 ppmv of kerosene TR0, 4125 ppmv of O<sub>2</sub>, diluent nitrogen)

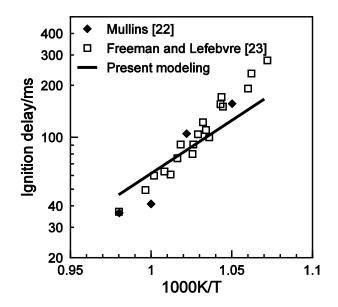
Dagaut & Cathonnet, *PECS* **32**, 48-92, 2006



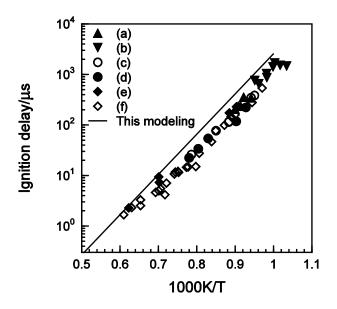
The oxidation of kerosene in premixed flame (Douté et al.) conditions:1 atm, 0.010739794 g/cm<sup>2</sup>/s, initial mole fractions: 0.0319 of kerosene, 0.28643 of oxygen.

Dagaut & Cathonnet, *PECS* **32**, 48-92, 2006

The ignition delays of few kerosene-air mixtures at atmospheric pressure have been reported before; some of them have been used in several previous modeling efforts showing reasonable agreement with these data.



Ignition delay of kerosene/air mixtures at 1 atm



Ignition delay of kerosene/air mixtures at **20 atm** Data: Dean et al. 20<sup>th</sup> ICDERS (2005); Starikovskii et al. (2003); Davidson and Hanson, 6th Int. Conf. on Chemical Kinetics, Gaithersburg, MD (2005).

Dagaut & Cathonnet, PECS 32, 48-92, 2006

A kinetic analysis of the reaction paths during the oxidation of the kerosene model-fuel at 10 atm, under stoichiometric conditions indicated that the **overall oxidation of the fuel is mostly driven by n-decane oxidation.** 

According to the model, at 900 K, the early stages of the fuel oxidation involve the oxidation of n-decane, n-propylbenzene, and n-propylcyclohexane.

Hydroxyl radicals are the main species involved in the oxidation of the fuel mixture. The oxidation of n-decane is responsible for the production of these radicals via a complex reaction scheme that can be summarized as follows:

```
n-C_{10}H_{22} => 3-C_{10}H_{21}, 4-C_{10}H_{21}, and 5-C_{10}H_{21}
n-C_{10}H_{22} => 1-C_8H_{17}, 4-C_8H_{17}, 2-C_8H_{17}, and 3-C_8H_{17}
```

The decyl and octyl radicals isomerize and decompose. Their decomposition yields 1-butyl and 1-propyl radicals that in turn decompose.

The further reactions in turn yield OH radicals:

```
1-C_4H_9+M => C_2H_5 + C_2H_4 + M;

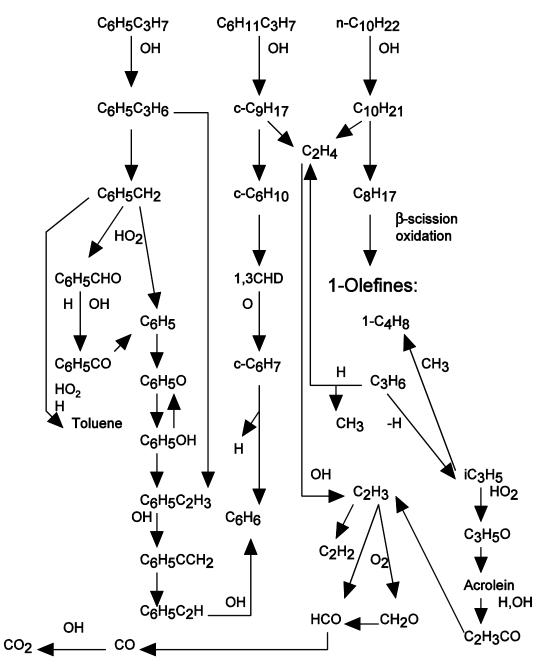
1-C_3H_7+M => CH_3 + C_2H_4 + M;

C_2H_5 + O_2 => C_2H_4 + HO_2;

2 HO_2 => H_2O_2 + O_2;

H_2O_2 + M => OH + OH + M;

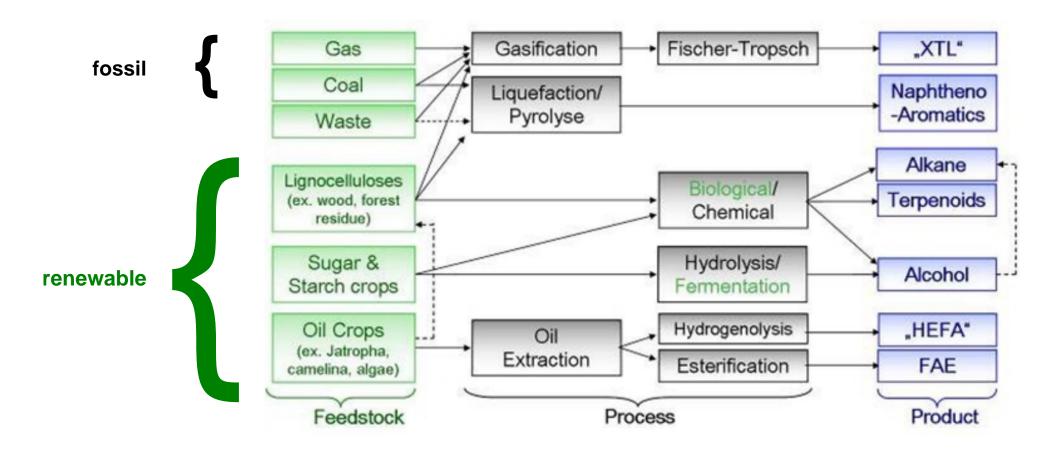
CH_3 + HO_2 => OH + CH_3O.
```



Dagaut & Cathonnet, PECS 32, 48-92, 2006

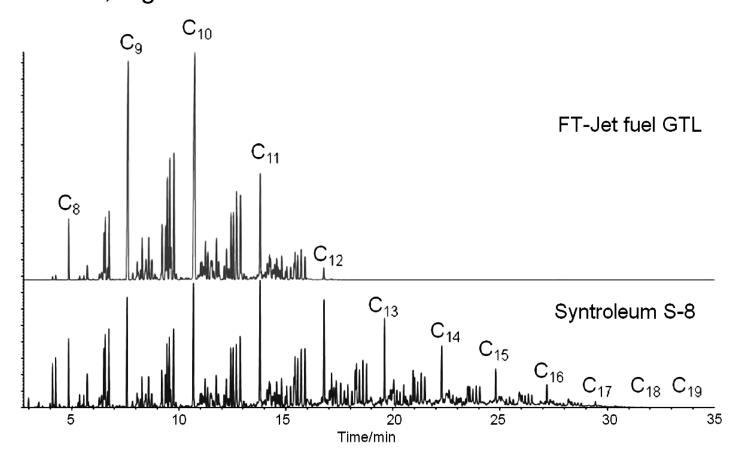
# Synthetic jet fuels

In recent years, research activities on **synthetic and bio-derived jet fuels** have increased significantly in order to reduce dependence of air transportation on oil (petroleum).



\*XTL: Gas/Coal/Waste/Renewable to Liquid

The Fischer-Tropsch (F-T) process allows the production of a kerosene type fuel from synthesis gas also called syngas (CO/H<sub>2</sub>). Frequently, a **synthetic jet fuel** is mainly composed of *n*-alkanes, *iso*-alkanes and *cyclo*-alkanes, but **composition varies** from one source to another, e.g.:



Source: Egolfopoulos et al. (USC)

The very low proportion of aromatic compounds in GtL fuels causes a reduction in emissions of soot and unburned hydrocarbons\*.

The composition of synthetic jet fuel allows also a decrease in emissions of carbon dioxide and soot\*\*.

These fuels are a good alternative to current conventional oil-derived fuels.

<sup>\*</sup> Corporan et al., 2007, *Energy & Fuels* 21, pp. 2615–2626; Kahandalawa et al., 2008, *Energy & Fuels* 22, pp. 3673–3679.

<sup>\*\*</sup> Rye et al., 2010, Energy & Environmental Science 3, pp. 17-27

The kinetics of oxidation of **alternative jet fuels** and **representative surrogates** studied in a JSR under the same conditions (temperature, 550-1150 K; pressure, 10 bar; equivalence ratio, 0.5-2).

To **experimentally represent** the two synthetic fuels we have designed **surrogates** consisting of few representative species among **n-decane**, **iso-octane**, **decalin**, **n-propylcyclohexane**, **and n-propylbenzene**.

The **oxidation of 2 representative mixtures**, **100% GtL** ( $C_{10.45}H_{22.93}$ ; H/C=2.20; M=148.28 g mol<sup>-1</sup>; CN=56\*; density=724 g L<sup>-1</sup>, from Shell), **and 100% CtL** ( $C_{11.06}H_{21.38}$ , H/C=1.934; M=154.12 g mol<sup>-1</sup>; CN=41\*; density=799 g L<sup>-1</sup>, from Sasol) **was performed in a JSR** at 10 atm.

\* ASTM D7668

A **detailed kinetic reaction mechanism** was developed and validated by comparison with the experimental results obtained here and previously\*.

The model was also evaluated under shock tubes conditions by using data from the literature\*\*.

<sup>\*</sup> Mzé Ahmed, A., Dagaut, P., Hadj-Ali, K., Dayma, G., Kick, T., Herbst, J., Kathrotia, T., Braun-Unkhoff, M., Herzler, J., Naumann, C., and Riedel, U., 2012, Energy & Fuels, 26(10), pp. 6070-6079.

Dagaut, P., Karsenty, F., Dayma, G., Diévart, P., Hadj-Ali, K., Mzé-Ahmed, A., Braun-Unkhoff, M., Herzler, J., Kathrotia, T., Kick, T., Naumann, C., Riedel, U., and Thomas, L., 2014, Combustion and Flame, 161(3), pp. 835-847

<sup>\*\*</sup>Wang, H. W., and Oehlschlaeger, M. A., 2012, Fuel, 98(1), pp. 249-258.

The CHEMKIN II computer code was used for the kinetic modeling of the oxidation of the two fuels studied in a jet-stirred reactor.

The chemical kinetic reaction mechanism used contained **2,430 species and 10,962** reversible reactions.

Surrogate model fuels were used for the kinetic modeling

• The synthetic kerosene **GtL** was represented by a mixture of **28.1%w n-decane**, **30% 2-methylheptane**, **33.1% 3-methylheptane**, **and 8.8% decalin**.

This corresponds very well with the GtL composition (GtL%/surrogate% in mass: 28.1/28.1, 63.1/62.8, 8.8/8.8 in mass of n-alkanes, iso-alkanes, and naphthenes, respectively).

• The synthetic kerosene CtL was represented by a mixture of 5.7%w *n*-decane, 11.5% iso-octane, 24.8% 3-methylheptane, 16.1% n-propylcyclohexane, 28.3% decalin, 4% n-propylbenzene, and 9.6% tetralin.

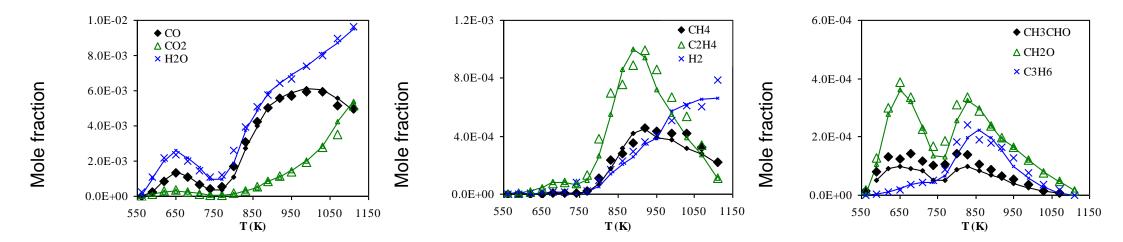
This corresponds very well with the CtL composition (CtL%/surrogate% in mass: 5.7/5.7, 36.3/36.3, 16.1/16.1, 28.3/28.3, 4/4, 9.6/9.6 of n-alkanes, iso-alkanes, mono-naphthenes, di-naphtenes, mono-aromatics, and naphteno-aromatics, respectively).

Sub-models for surrogates components were taken from our previous modeling efforts. *n*-Decane, iso-octane, 2-methylheptane and 3-methylheptane studied previously<sup>#</sup> were used to represent the *n*- and *iso*-paraffins present in the synthetic fuels. Naphthenes were represented by *n*-propylcyclohexane\* and decalin\*\* in the model. Mono-aromatics were represented by n-propylbenzene\*\*\* and tetralin\*\*\*represented naphteno-aromatics.

Experimental data obtained in JSR were compared to simulations in order to validate the chemical kinetic mechanism developed in this work.

- # Sarathy et al., 2011, Combustion and Flame, 158(12), pp. 2338-2357.
  - Karsenty et al., 2012, Energy & Fuels, 26(8), pp. 4680-4689.
  - Mze-Ahmed et al., 2012, Energy & Fuels, 26(7), pp. 4253-4268.
- \* Ristori, A et al., 2001, Combustion Science and Technology, 165(1), pp. 197-228.
- \*\* Dagaut et al., 2013, Proceedings of the Combustion Institute, 34(1), pp. 289-296.
- \*\*\* Dagaut et al., 2002, Fuel, 81(2), pp. 173-184.
- \*\*\*\* Dagaut et al., 2013, Energy & Fuels, 27(3), pp. 1576-1585.

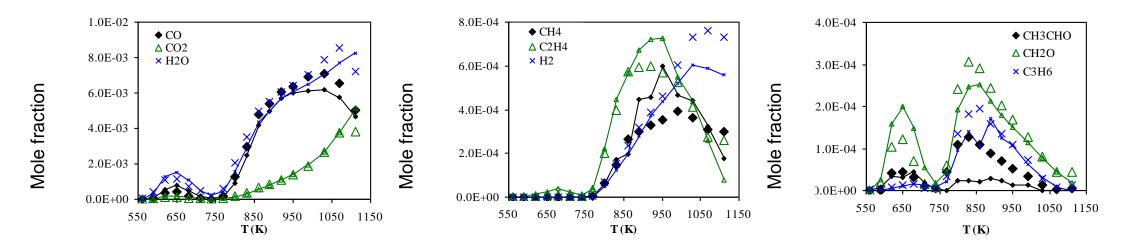
#### **EXPERIMENTAL RESULTS GTL VS. SURROGATE**



Concentrations profiles obtained from the **oxidation of the GTL and the representative mixture** in a JSR at 10 bar,  $\tau$  =0.7 s and  $\varphi$  =1. The initial mole fractions were:  $x_{GTL}$ =0.1%,  $x_{O2}$ =1.6%,  $x_{N2}$ =98.3% mole. The GTL data (large symbols) are compared to those for the surrogate (lines and small symbols, 650 ppm of **n-decane**, 375 ppm of **iso-octane**, and 95 ppm of **decalin**).

Dagaut et al., ICDERS 2015

#### **EXPERIMENTAL RESULTS CTL VS. SURROGATE**

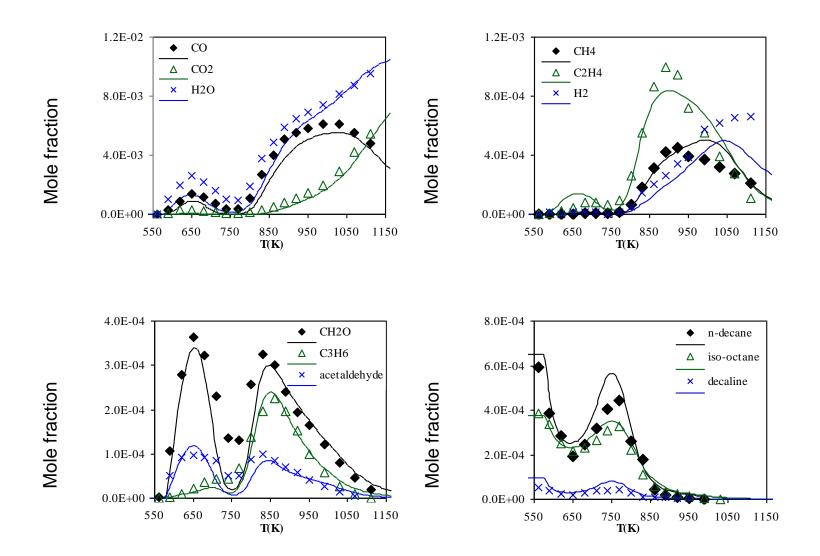


Concentrations profiles obtained from the oxidation of the **CTL** and the representative **mixture** in a JSR at 10 bar,  $\tau$  =0.7 s and  $\varphi$  =1. The initial mole fractions were:  $x_{GTL}$ =0.1%,  $x_{O2}$ =1.5%,  $x_{N2}$ =98.4% mole. The CTL data (large symbols) are compared to those for the surrogate (lines and small symbols, 163 ppm of **n-decane**, 365 ppm of **iso-octane**, 197 ppm of **n-propylcyclohexane**, 317 ppm of **decalin**, and 175 ppm of **n-propylbenzene**).

Dagaut et al., ICDERS 2015

# **EXPERIMENTAL RESULTS SPK VS. SURROGATE** Very similar experimental profiles obtained for the SPKs and their Surrogates Finetic modeling of the oxidation of these surrogates for model validation

## SURROGATE OX'n, EXPERIMENTAL VS. MODELING



Concentrations profiles obtained from the oxidation of a **GTL representative mixture** in a JSR at 10 bar,  $\tau$  =0.7 s and  $\phi$  =1. The data (large symbols) are compared to the **modeling** (lines).

Dagaut et al., ICDERS 2015

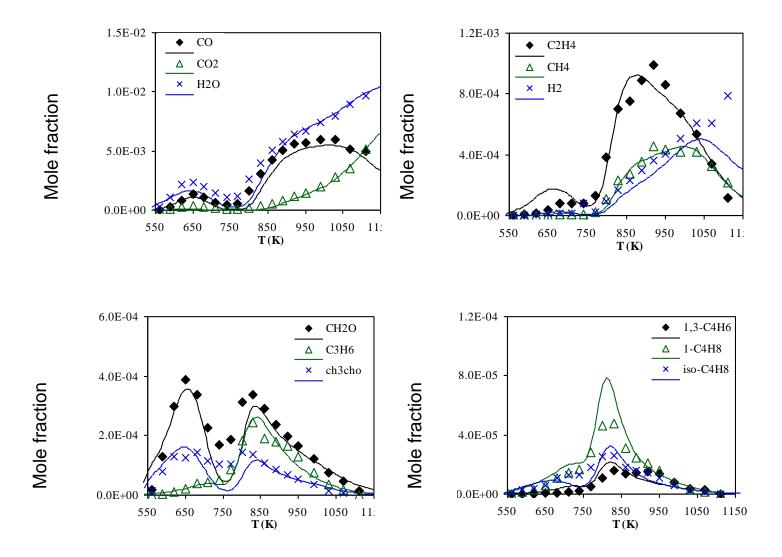
#### **MODELING GTL OX'n**

Composition of final model fuel to simulate the oxidation of the **GtL** fuel ( $C_{10.45}H_{23.06}$ ; H/C=2.20; CN= 57.94; 737.7 g mol<sup>-1</sup>; M=148.46 g mol<sup>-1</sup>) <sup>a</sup>

Component	Initial concentrations (ppm)
<i>n-</i> decane	294
2-methylheptane	390
3-methylheptane	431
decalin	94

<sup>&</sup>lt;sup>a</sup> 1.209×C<sub>8.64</sub>H<sub>18.97</sub> since we used 1209 ppm of model fuel to represent 1000 ppm of GtL

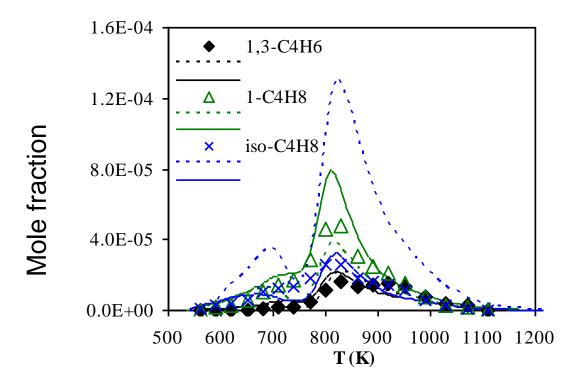
## GTL OX'n, EXPERIMENTAL VS. MODELING



Concentrations profiles obtained from the oxidation of the **GTL fuel** in a JSR at 10 bar,  $\tau$  =0.7 s and  $\phi$  =1. The data (large symbols) are compared to the modeling (lines). Dagaut et al., ICDERS 2015

### GTL OX'n, EXPERIMENTAL VS. MODELING

#### Modeling improvements:



Comparison of computed and experimental concentrations profiles obtained from the oxidation of the **GTL** fuel in a JSR at 10 bar,  $\tau$  =0.7 s and  $\phi$ =1 (experimental data: large symbols; previous model (Dagaut et al., 2015, CNF, 161(3) 835-847): dotted lines; this model: continuous lines).

Dagaut et al., ICDERS 2015

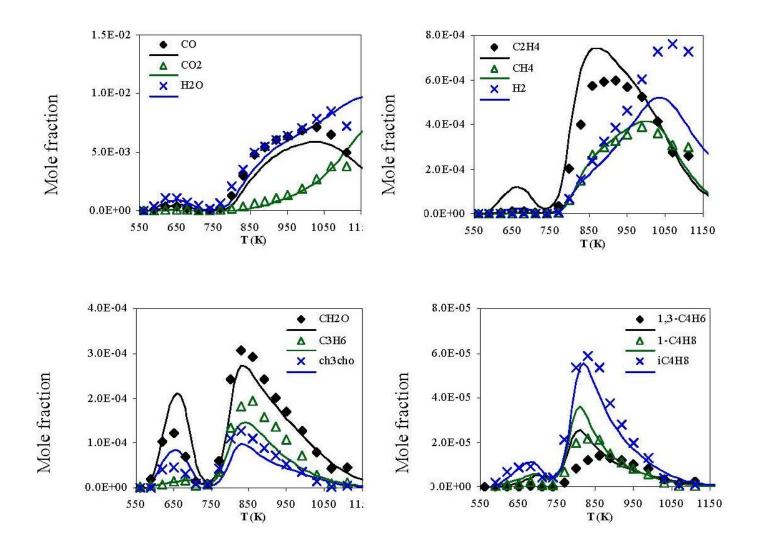
#### **MODELING CTL OX'n**

Composition of final model fuel to simulate the oxidation of the CtL fuel ( $C_{11.06}H_{21.6}$ ; H/C=1.953; CN= 32.7; 815.7 g mol<sup>-1</sup>; M=154.32 g mol<sup>-1</sup>) <sup>b</sup>

Component	Initial concentrations (ppm)
<i>n</i> -decane	62
iso-octane	155
3-methylheptane	335
n-propylcyclohexane	197
decalin	316
n-propylbenzene	52
tetralin	112

<sup>&</sup>lt;sup>b</sup> 1.229×C<sub>9</sub>H<sub>17.4</sub> since we used 1229 ppm of model fuel to represent 1000 ppm of CtL

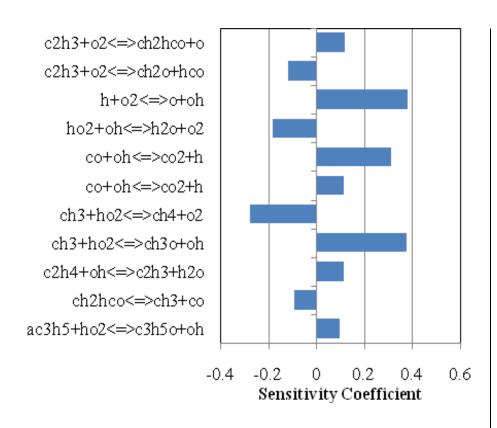
#### RESULTS AND DISCUSSION: CTL OX'n, EXPERIMENTAL VS. MODELING



Concentrations profiles obtained from the oxidation of the CTL fuel in a JSR at 10 bar,  $\tau$  =0.7 s and  $\phi$  =1. The data (large symbols) are compared to the modeling (lines).

Dagaut et al., ICDERS 2015

# Sensitivity analyses and reaction pathways analyses



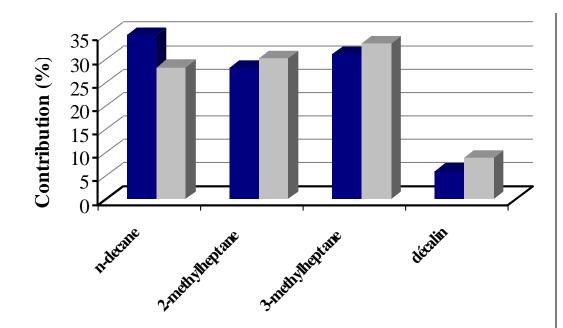
 $c2h3+o2 \le ch2hco+o$ c2h3+o2<=>ch2o+hco  $h + o2 \le > o + oh$  $h+o2(+M) \le ho2(+M)$ ho2+oh<=>h2o+o2  $co+oh \le co2+h$  $co+oh \le co2+h$ ch3+ho2<=>ch4+o2  $ch3+ho2 \le ch3o+oh$ ac3h5+ho2 <=> c3h5o+ohc6h5o<=>c5h5+co c6h5o+h(+M)<=>c6h5oh(+M) 0.2 -0.4-0.20.4Sensitivity Coefficient

Sensitivity analyses for  $CO_2$  at 1030 K during the oxidation of the **GtL** fuel in a JSR ( $\phi$  =1, 10 bar, residence time of 0.7 s

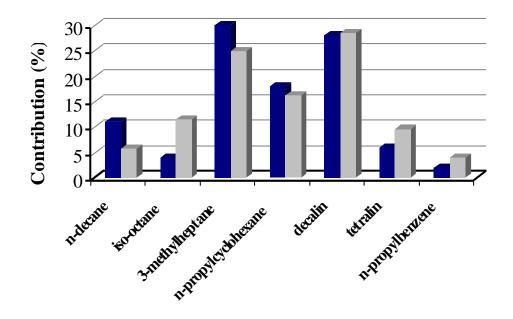
Sensitivity spectrum for  $CO_2$  during the oxidation of the **CtL** fuel in a JSR at  $\phi$ =1 and T=1030 K (P = 10 bar and  $\tau$  = 0.7 s).

These computations show the influence of **OH radicals** during the oxidation of these fuels.

## Sensitivity analyses and reaction pathways analyses



Contribution of the surrogate components to the **formation of OH** (blue) during the oxidation of the **GtL** fuel in a JSR ( $\phi$  =1, 830 K, 10 bar, residence time of 0.7 s). For comparison, the concentrations of the surrogate components are shown in grey.



Contribution of the surrogate components to the **formation of OH** (blue) during the oxidation of the **CtL** fuel in a JSR ( $\phi$  =1, 830 K, 10 bar, residence time of 0.7 s). For comparison, the concentrations of the surrogate components are shown in grey

# **Ignition Delay Times**

Wang and Oehlschlaeger\* measured the ignition delay of a synthetic jet fuel derived from natural gas and provided by Shell ( $C_{10.40}H_{22.88}$ ) in a heated shock tube between 650 and 1290 K at 20 atm and  $\phi$ =1.0 (1.286% fuel, 20.74%  $O_2$ , 77.97%  $N_2$ ). In order to simulate the high temperature regime (T > 1000K), they used the surrogate model developed by Naik et al.\*\* (n-decane: 61%, n-dodecane: 11%, iso-octane: 28% in mole).

Their results showed that the data measured by Wang and Oehlschlaeger are similar to the model predictions at high temperature.

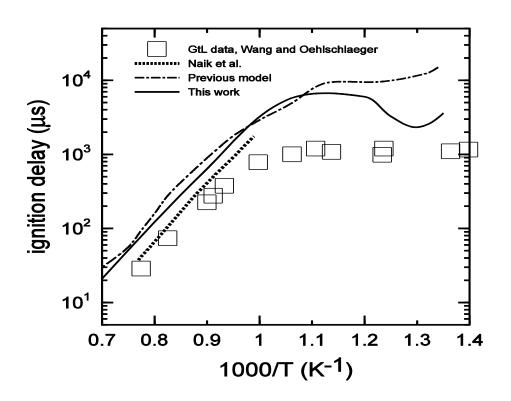
We verified the validity of our model for the ignition in shock tube using the experimental data\* for GtL and the data of Vasu et al.\*\*\* for *n*-dodecane ignition.

\*Wang and Oehlschlaeger, 2012, *Fuel* 98, pp. 249-258

\*\*Naik et al., 2011, Comb. Flame 158, pp. 434-445

\*\*\*Vasu et al., 2009, Proc. Combust. Inst. 32, pp. 173-180

# **Ignition Delay Times in Air**



Comparison between ignition delay times measurements by Wang and Oehlschlaeger (Shell GtL, open symbols) and Vasu et al. (*n*-dodecane, stars), modeling of Naik et al. (dotted line), the present modeling results for GtL (dashed dotted line) and *n*-dodecane predictions (solid line).

• The computed ignition delays > Naik's computations.

Same trends as in the experiments but overestimation of ignition delays (ca. x4 @ 900K).

• The new computed ignition is in better agreement with the data than previously, but the model is too slow.

# Further studies of synthetic jet fuels oxidation

The very low proportion of aromatic compounds in GtL fuels causes a reduction in emissions of soot and unburned hydrocarbons\*. The composition of synthetic jet fuel allows also a decrease in emissions of carbon dioxide and soot\*\*.

These fuels are a good alternative to current conventional oil-derived fuels.

<sup>\*</sup> Corporan et al., 2007, *Energy & Fuels* 21, pp. 2615–2626; Kahandalawa et al., 2008, *Energy & Fuels* 22, pp. 3673–3679.

<sup>\*\*</sup> Rye et al., 2010, Energy & Environmental Science 3, pp. 17-27

#### A GtL, a Naphthenic cut (NC) and a mixture NC/GtL were oxidized in a JSR:

Properties	GtL	NC*	NC/GtL
Formula	C <sub>10.45</sub> H <sub>23.06</sub>	C <sub>12.64</sub> H <sub>23.64</sub>	C <sub>11.54</sub> H <sub>23.35</sub>
M (g mol <sup>-1</sup> )	148.44	175.32	161.83
H/C ratio	2.20	1.87	2.02
DCN <sup>‡</sup>	58.0	39.3	45.8
Density (g I <sup>-1</sup> )	737.7	863.1	800.3

<sup>\*</sup> Naphthenic cut: a representative commercial solvent that fits with typical chemical composition of product coming from coal or biomass liquefaction.

<sup>&</sup>lt;sup>‡</sup> measured by PAC Cetane ID 510, ASTM D7668

A detailed kinetic reaction mechanism was developed and validated by comparison with the experimental results obtained here and previously\*.

The CHEMKIN II computer code was used for the kinetic modeling of the oxidation of the two fuels studied in a jet-stirred reactor.

The chemical kinetic reaction mechanism used contained **2,384 species and 10,368** reversible reactions.

<sup>\*</sup> Mzé Ahmed, A., Dagaut, P., Hadj-Ali, K., Dayma, G., Kick, T., Herbst, J., Kathrotia, T., Braun-Unkhoff, M., Herzler, J., Naumann, C., and Riedel, U., 2012, *Energy & Fuels*, 26(10), pp. 6070-6079.

Dagaut, P., Karsenty, F., Dayma, G., Diévart, P., Hadj-Ali, K., Mzé-Ahmed, A., Braun-Unkhoff, M., Herzler, J., Kathrotia, T., Kick, T., Naumann, C., Riedel, U., and Thomas, L., 2014, *Combustion and Flame*, 161(3), pp. 835-847

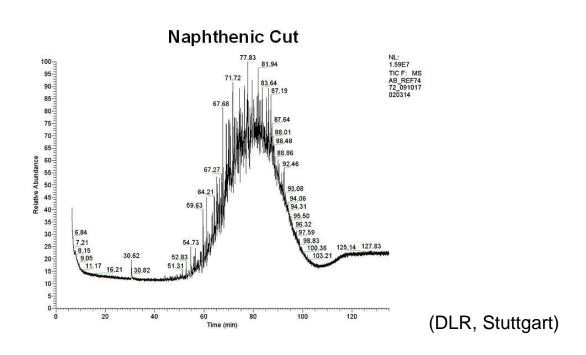
Surrogate model fuels for the kinetic modeling:

The **GtL** was represented by a mixture of **n-decane**, **2-methylheptane**, **3-methylheptane**, **and decahydronaphthalene** (28.1%, 30%, 33.1%, and 8.8% in mass, respectively) which corresponds very well with the GtL mass composition (28.1%, 62.8%, 8.8% of n-alkanes, iso-alkanes, and naphthenes, respectively). The model fuel matches well the GtL cetane number (57.94 vs. 58) and its H/C ratio (2.2 vs. 2.2).

The substitution of the highly branched iso-octane used in a previous model by weakly branched iso-alkanes (2-methylheptane and 3-methylheptane) is beneficial, particularly for better controlling iso-butene production.

#### Surrogate model fuels for the kinetic modeling:

The **naphthenic cut** was represented by a mixture of **decahydronaphthalene**, **tetrahydronaphthalene**, **n-propylcyclohexane**, **2-methylheptane**, **and 3-methylheptane** (27.6%, 23.5%, 10.8%, 12.1%, 25%, and 13% in mass, respectively) which is in line with the naphthenic cut composition (89.9% of paraffins and cycloparaffins and 10.1% of aromatics in mass).



• Composition of the model-fuel to represent the GtL\* fuel in the computations (C<sub>10.45</sub>H<sub>23.06</sub>; H/C=2.20; DCN= 57.94; M=148.46 g mol<sup>-1</sup>) ‡

Component	Initial concentrations (ppm)
<i>n-</i> decane	294
2-methylheptane	390
3-methylheptane	431
decahydronaphthalene	94

<sup>&</sup>lt;sup>‡</sup> 1.209×C<sub>8.64</sub>H<sub>18.97</sub> since we used 1209 ppm of model fuel to represent 1000 ppm of GtL

\*GtL: 28.1% n-alkanes, 62.8% iso-alkanes, 8.8% cyclo-alkanes, and 0.2% aromatics. The composition of the fuels and their molecular weight were determined through gas chromatography (http://www.alfa-bird.eu-vri.eu/)

• Composition of the model-fuel to represent the naphthenic cut\* in the computations (C<sub>12.63</sub>H<sub>23.26</sub>; H/C=1.84; DCN= 39.7; M=174.82 g mol<sup>-1</sup>)<sup>‡</sup>

Component	Initial concentrations (ppm)
decahydronaphthalene	350
tetrahydronaphthalene	312
n-propylcyclohexane	150
2-methylheptane	384
3-methylheptane	200

<sup>&</sup>lt;sup>‡</sup> 1.396× C<sub>9.05</sub>H<sub>16.66</sub> since we used 1396 ppm of model fuel to represent 1000 ppm of NC

\*NC: 4.7% paraffins, 85.2% cyclo-paraffins, 9.6% monoaromatics and 0.5% polyaromatics. The composition of the fuels and their molecular weight were determined through gas chromatography (http://www.alfa-bird.eu-vri.eu/)

• Composition of the model-fuel representing the GtL/naphthenic cut mix (C<sub>11.54</sub>H<sub>23.09</sub>; H/C=2.0; DCN= 48.8; M=161.57 g mol<sup>-1</sup>)‡ in the simulations

Component	Initial concentrations (ppm)
<i>n</i> -decane	147
decahydronaphthalene	222
tetrahydronaphthalene	156
2-methylheptane	387
3-methylheptane	316
n-propylcyclohexane	75

 $<sup>^{\</sup>ddagger}$  1.3024× C<sub>8.86</sub>H<sub>17.73</sub> since we used 1302.4 ppm of model fuel to represent 1000 ppm of GtL/naphthenic cut mixture.

#### **MODELING**

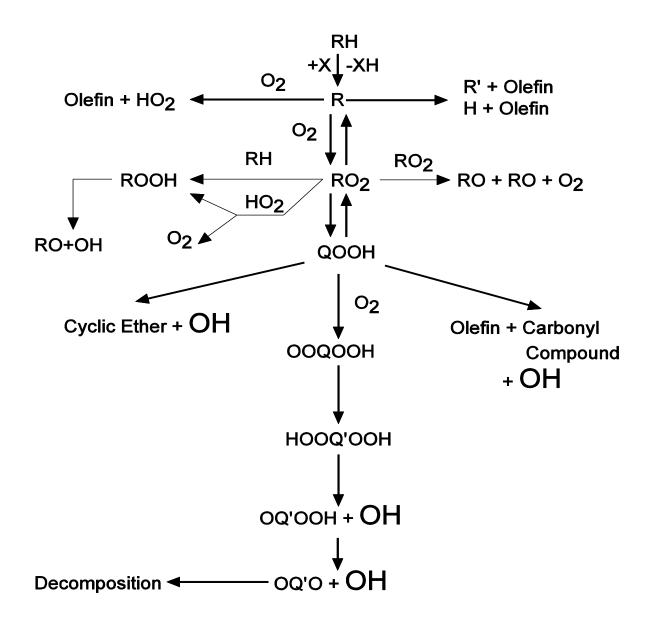
Sub-models for surrogates components were taken from our previous modeling efforts. *n*-Decane, 2-methylheptane and 3-methylheptane studied previously<sup>‡</sup> were used to represent the *n*- and *iso*-paraffins present in the synthetic fuels. Naphthenes were represented by *n*-propylcyclohexane\* and decahydronaphthalene \*\* in the model. Tetrahydronaphthalene \*\*\*represented naphtheno-aromatics.

Experimental data obtained in JSR were compared to simulations in order to validate the chemical kinetic mechanism.

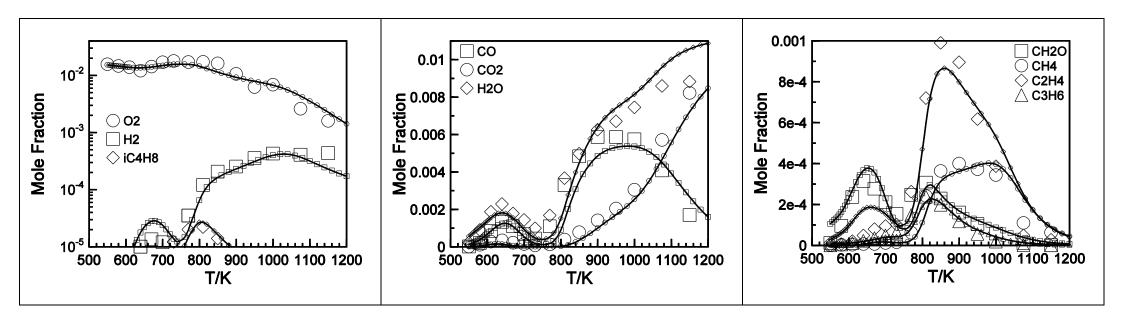
- \$\preceq\$ Sarathy et al., 2011, Combustion and Flame, 158(12), pp. 2338-2357.
   Karsenty et al., 2012, Energy & Fuels, 26(8), pp. 4680-4689.
   Mze-Ahmed et al., 2012, Energy & Fuels, 26(7), pp. 4253-4268.
- \* Ristori, A et al., 2001, Combustion Science and Technology, 165(1), pp. 197-228.
- \*\* Dagaut et al., 2013, *Proceedings of the Combustion Institute*, 34(1), pp. 289-296.
- \*\*\* Dagaut et al., 2013, *Energy & Fuels*, 27(3), pp. 1576-1585.

#### **RESULTS AND DISCUSSION**

The data showed three regimes of oxidation: the cool flame regime (T <  $\sim$ 750 K), the negative temperature coefficient ( $\sim$ 640-750 K) and the high-temperature regime (>750 K).

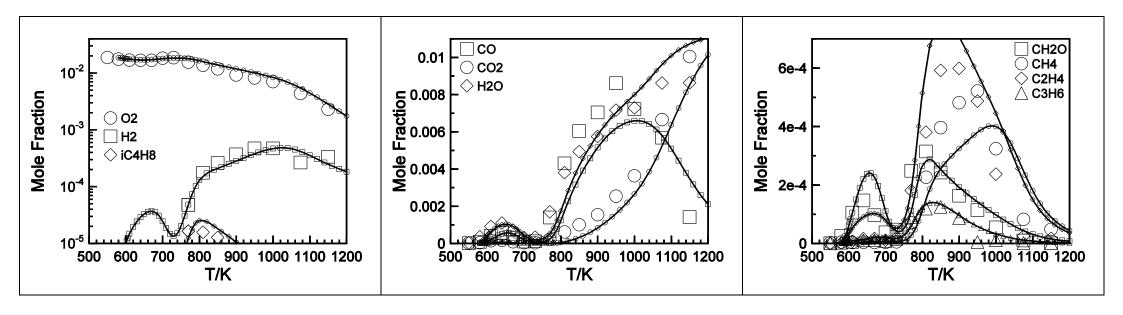


### RESULTS AND DISCUSSION: GTL OX'n, EXPERIMENTAL VS. MODELING



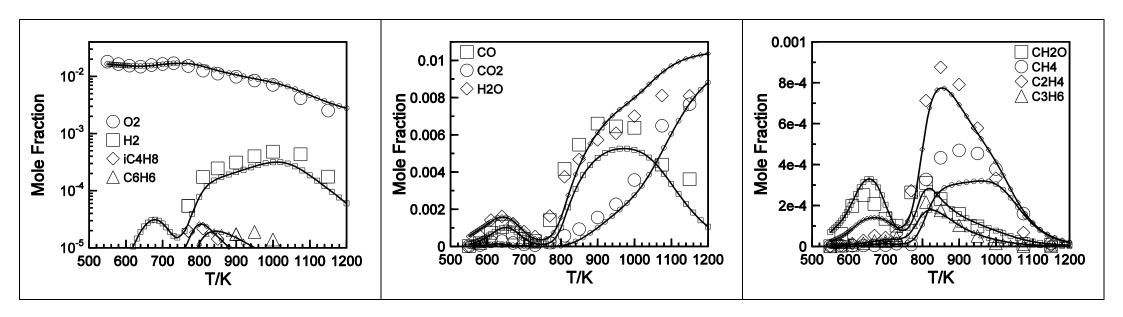
Comparison of experimental and computed concentrations profiles obtained from the oxidation of 1000 ppm of the GtL fuel with 16215 ppm of  $O_2$  in a JSR at 10 bar,  $\tau$  =1 s and  $\phi$ =1 (experimental data: large symbols; computations: lines; dilution by  $N_2$ ).

### RESULTS AND DISCUSSION: NC OX'n, EXPERIMENTAL VS. MODELING



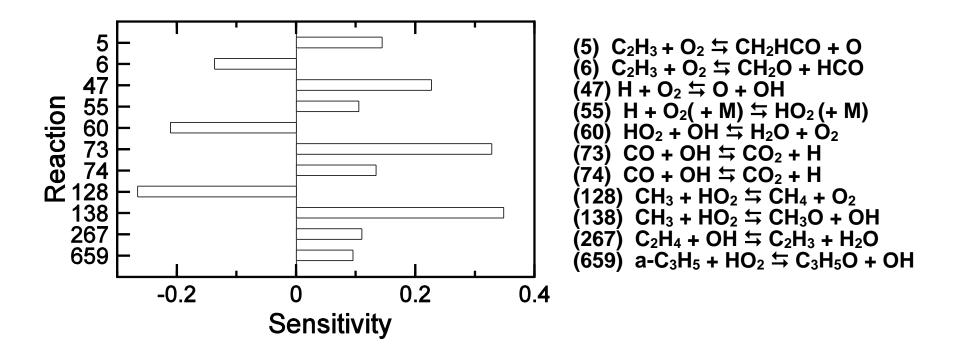
Comparison of computed and experimental concentrations profiles obtained from the oxidation of 1000 ppm of the naphtenic cut with 18570 ppm of  $O_2$  in a JSR at 10 bar,  $\tau$  =1 s and  $\phi$ =1 (experimental data: large symbols; computations: lines; dilution by  $N_2$ ).

### RESULTS AND DISCUSSION: NC/GTL OX'n, EXPERIMENTAL VS. MODELING



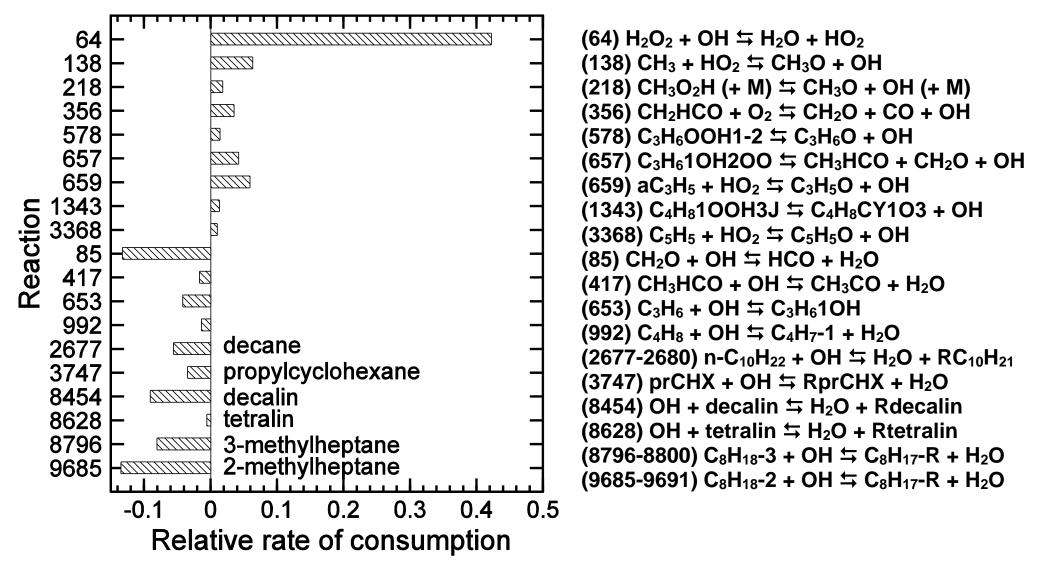
Comparison of computed and experimental concentrations profiles obtained from the oxidation of 1000 ppm of the naphtenic cut/GtL fuel mixture with 17378 ppm of  $O_2$  in a JSR at 10 bar,  $\tau = 1$  s and  $\phi = 1$  (data: large symbols; computations: lines; dilution by  $O_2$ ).

### RESULTS AND DISCUSSION: NC/GTL OX'n, MODELING



Sensitivity analyses for  $CO_2$  at 1040 K during the oxidation of 1000 ppm of the naphthenic cut/GtL fuel mixture in a JSR ( $\phi$  =1, 10 bar, residence time of 1 s).

### RESULTS AND DISCUSSION: NC/GTL OX'n, MODELING



**Consumption/Production** of OH at 790 K during the oxidation of 1000 ppm of the naphtenic cut/GtL fuel mixture in a JSR (φ =1, 10 bar, residence time of 1 s). P. Dagaut, P. Diévart. Proc. Combust. Inst. **36**, 433–440 (2017)

#### **5.4 Biofuels**

# 5.4.1 RME (biodiesel)

Several vegetable oils have also been tested for transport purpose, but their high viscosity, low volatility, and low cetane number (>40) leaded to incomplete combustion. Therefore, the concept of using bio-diesel, consisting of alkyl esters of these vegetable oils obtained by transesterification with an alcohol (mostly methanol, but also ethanol)

# Vegetable oil composition

Vegetable oil	Fatty acid composition (% weight)										
	16:1	18:0	20:0	22:0	24:0	18:1	22:1	18:2	18:3		
Corn	11.67	1.85	0.24	0.00	0.00	25.16	0.00	60.60	0.48		
Cottonseed	28.33	0.89	0.00	0.00	0.00	13.27	0.00	57.51	0.00		
Crambe	20.7	0.70	2.09	0.80	1.12	18.86	58.51	9.00	6.85		
Peanut	11.38	2.39	1.32	2.52	1.23	48.28	0.00	31.95	0.93		
Rapeseed	3.49	0.85	0.00	0.00	0.00	64.4	0.00	22.30	8.23		
Soybean	11.75	3.15	0.00	0.00	0.00	23.26	0.00	55.53	6.31		
Sunflower	6.08	3.26	0.00	0.00	0.00	16.93	0.00	73.73	0.00		
Castor	0.00	0.00	0.00	0.00	0.00	87.0	0.00	0.00	11.2		
Palm	10.2	3.7	0.00	0.00	0.00	22.8	0.00	53.7	8.6		

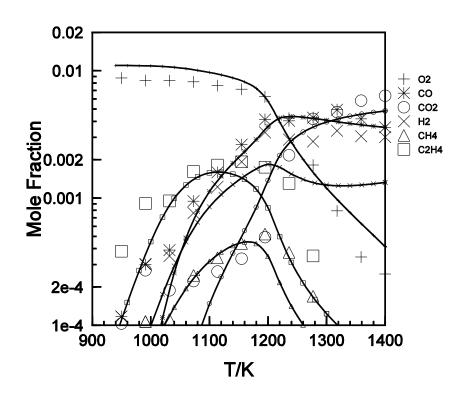
### **Early modeling efforts**

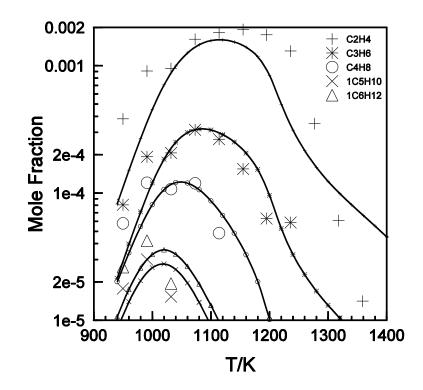
To model combustion of fuels, to predict accurate combustion performance and emission characteristics, a good knowledge of their kinetics of combustion is essential. Since rapeseed is one of the main crop growing Europe, we focus our study on the kinetic of rapeseed oil methyl ester (RME) oxidation.

RME is a complex mixture of  $C_{14}$ ,  $C_{16}$ ,  $C_{18}$ ,  $C_{20}$ , and  $C_{22}$  esters with highly saturated carbon chain. The composition of the fuel was 0.1%  $C_{14}$ , 5.4%  $C_{16}$ , 92.0%  $C_{18}$ , 2.0%  $C_{20}$ , and 0.5%  $C_{22}$ , with mostly one double bond on the acid chain. The equation for the oxidation of RME can be written as follows:

$$C_{17.92}H_{33}O_2 + 25.17 O_2 = 17.92 CO_2 + 16.5 H_2O.$$

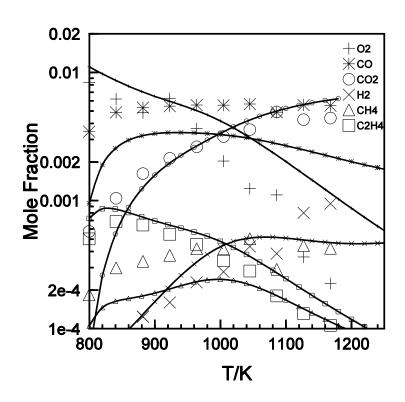
Because of the complexity of this fuel, it is difficult to propose a detailed kinetic scheme for its oxidation, although that could be achieved building on previous kinetics studies involving simpler esters

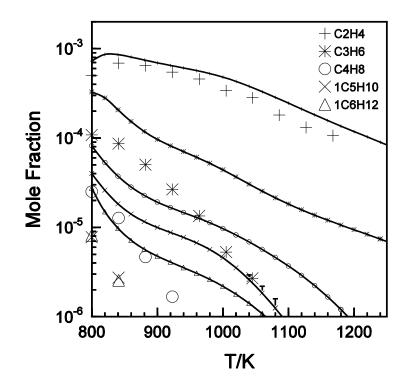




The oxidation of RME in a JSR at 1 atm ( $\phi$  = 1, 0.07 s). The data (large symbols) are compared to the computations (lines, small symbols), n-hexadecane as surrogate model-fuel, initial mole fractions: n-hexadecane, 0.0005625; oxygen, 0.011; nitrogen, 0.9884375).

P. Dagaut et al., Proc. Combust. Inst. 31, 2955–2961 (2007)





The oxidation of RME in a JSR at 10 atm ( $\phi$  = 1, 1 s). The data (large symbols) are compared to the computations (lines, small symbols), n-hexadecane as surrogate model-fuel, initial mole fractions: n-hexadecane, 0.0005625; oxygen, 0.012585; nitrogen, 0.9868525).

P. Dagaut et al., Proc. Combust. Inst. 31, 2955–2961 (2007)

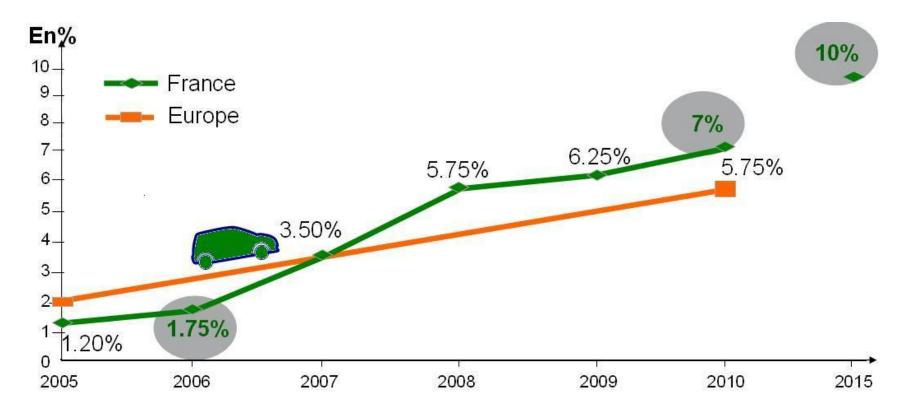
#### 5.4.2 B30

Diesel engines contribute significantly to overall **carbon dioxide emissions** whereas concerns about green-house effect and air pollution favor the investigation of sustainable and environment-friendly Diesel fuels.

Biofuels such as fatty acid methyl esters (FAME) are **mixed** in variable quantities (e.g. B5 contains 5% in volume of FAME and B30 contains 5% in volume of FAME) **with fossil Diesel fuel**.

#### Introduction

**Reduction of engines emissions** in terms of carbon oxides and polyaromatic hydrocarbons (PAH) have been reported, indicating bio-diesel may help preserving our environment.



Reduction of carbon footprint in Europe by increased biodiesel fraction (EU 2010: 5.75% energy HV)

# **Introduction** (cont'd)

The so-called bio-Diesel is a **mixture of FAME** produced from transesterification of triglycerides (oils) with methanol. Current biodiesel fuels are mixtures of ca. C<sub>12</sub>-C<sub>22</sub> highly saturated carbon-chain esters. Their complex composition implies the use of surrogate model-fuels for simulating their combustion kinetics.

Whereas early kinetic studies have demonstrated a strong similitude between the oxidation of rapeseed oil methyl esters (RME) and that of n-hexadecane, long-chain methyl esters exhibiting cool-flames were also proposed as bio-Diesel model fuels.

### **Introduction** (cont'd)

A fossil Diesel fuel consists of an even more complex mixture of thousands of medium-high molecular weight hydrocarbons that participate in thousands of pyrolysis and oxidation reactions. Therefore, surrogates are needed to represent Diesel fuel with a limited number of components.

In Europe, the 'IDEA' surrogate Diesel fuel (70% n-decane + 30% 1-methyl naphthalene) was formulated previously as part of the 'Integrated Development on Engine Action' (IDEA) program.

This fuel mixture matches both the physicochemical properties and combustion behavior of a conventional Diesel fuel. The IDEA fuel has properties similar to those of a conventional Diesel fuel, i.e. a normal density of 0.798 g/L at 20°C, a CN of ca. 53, and a hydrogen-to-carbon ratio of 1.8.

# **Introduction** (cont'd)

The kinetics of oxidation of a **commercial B30 bio-Diesel fuel** and a **B30 surrogate bio-Diesel fuel** were measured and compared.

The experiments were performed in a jet-stirred reactor (JSR), in order to:

- (1) **provide** new information on the kinetic of oxidation of a B30 bio-Diesel fuel over a wide range of conditions,
- (2) **verify** the chemical kinetics of oxidation of a simple B30 surrogate can represent that of a commercial B30 Diesel fuel, and
- (3) **propose and validate** a detailed kinetic reaction mechanism for the oxidation of a B30 bio-Diesel fuel from low to high temperatures.

#### Experimental conditions in the JSR (10300 ppm of C, 560-1030 K, t=0.6 & 1s)

Initial cor	φ	P/ atm				
B30	n-C <sub>10</sub> H <sub>22</sub>	C <sub>11</sub> H <sub>10</sub>	C <sub>9</sub> H <sub>18</sub> O <sub>2</sub>	$O_2$		
600	-	-	-	0.0574	0.25	10
600	-	-	-	0.0287	0.5	10
600	-	-	-	0.0144	1	10
600	-	-	-	0.0096	1.5	10
-	490	210	300	0.0597	0.25	10
-	490	210	300	0.0284	0.5	6, 10
-	490	210	300	0.0142	1	6, 10
_	490	210	300	0.0095	1.5	10

B30 bio-Diesel fuel surrogate: 49% n-decane, 21% 1-methyl naphthalene, and 30% methyl octanoate in mole, i.e. C<sub>10.3</sub>H<sub>18.4</sub>O<sub>2</sub>

Commercial low-S B30 bio-Diesel fuel (CN 54.8, 84.1% C, 12.9% H, and 3% O by wt., d= 845 g/L at 15°C, FAME fraction was rapeseed oil methyl ester): **C**<sub>16.47</sub>**H**<sub>30.83</sub>**O**<sub>0.5</sub>

# Modeling

The computations were performed using the PSR computer code.

The detailed kinetic reaction mechanism is based on previous studies of the oxidation of methyl octanoate, large alkanes, 1-methylnaphtalene and diesel + IDEA surrogate [H.P. Ramirez L, K. Hadj-Ali, P. Diévart, G. Moréac, P. Dagaut, Energy Fuels 24(3) (2010) 1668-1676] where cross-reactions between the main fuel components were considered: metathesis of n-decane with phenyl, benzyl, 1-naphtylmethyl, 1-naphtyl, and indenyl radicals; reactions of decyl radicals with 1-naphtaldehyde; reactions of •C<sub>10</sub>H<sub>21</sub>O<sub>2</sub> with toluene, 1-methylnaphtalene, 1-naphtylmethyl, 1-naphtaldehyde, benzyl, phenyl, and 1-naphtyl.

The proposed kinetic scheme (7748 reversible reactions and 1964 species) represents the 1st attempt to propose a kinetic scheme for the oxidation of Diesel-biodiesel fuel mixtures.

# **Experiment results**

- Oxidation of a B30 bio-Diesel fuel surrogate, 49% n-decane, 21% 1-methyl naphthalene, and 30% methyl octanoate in mole
- Oxidation of a commercial low-sulfur B30 bio-Diesel fuel

They were studied in a jet-stirred reactor over a wide range of conditions:  $\phi$ =0.25-1.5; temperature in the range 560–1030 K, mean residence time constant: 0.6 s at 6atm and to 1 s at 10atm.

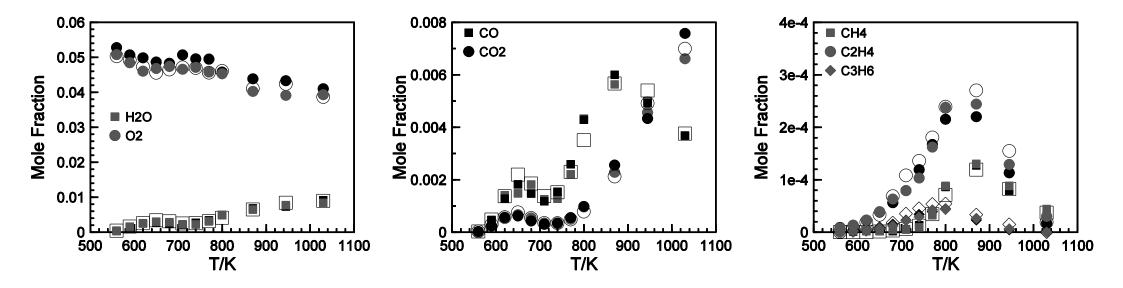
This allowed the observation of the cool-flame oxidation regime, the negative temperature coefficient (NTC) regime, and the high-temperature oxidation regime.

# **Experiment results** (cont'd)

More than 20 species were identified and measured by FTIR, CG-MS/FID/TCD. Experimental concentration profiles were obtained for H<sub>2</sub>, H<sub>2</sub>O, O<sub>2</sub>, CO, CO<sub>2</sub>, CH<sub>2</sub>O, CH<sub>4</sub>, C<sub>2</sub>H<sub>6</sub>, C<sub>2</sub>H<sub>4</sub>, C<sub>2</sub>H<sub>2</sub>, formaldehyde, acetaldehyde, C<sub>3</sub>H<sub>6</sub>, 1-C<sub>4</sub>H<sub>8</sub>, 1,3-C<sub>4</sub>H<sub>6</sub>, 1-C<sub>5</sub>H<sub>10</sub>, 1-C<sub>6</sub>H<sub>12</sub>, 1-C<sub>8</sub>H<sub>16</sub>, n-decane, methyl octanoate, and 1-methylnaphthalene. Other minor species detected at ppm levels were not quantified nor used in the modeling.

The concentration profiles measured from the oxidation of the commercial B30 and the B30 surrogate over the low-, intermediate-, and high-temperature oxidation regimes were compared:

# Experiment results (cont'd)



Experimental concentration profiles from the oxidation of a commercial Diesel fuel (filled symbols), the B30 surrogate (empty symbols) in a JSR at 10 atm,  $\phi$ =0.25, and  $\tau$ = 1s.

H.P. Ramirez L. et al., Proc. Combust. Inst. 33(1), 375–382 (2011)

The concentration profiles obtained for CO, CO<sub>2</sub>, H<sub>2</sub>O, and O<sub>2</sub> during the oxidation of the 2 biofuels are **very similar** over the entire range of experimental conditions; the commercial Diesel fuel used in the B30 mixture reacts **similarly**.

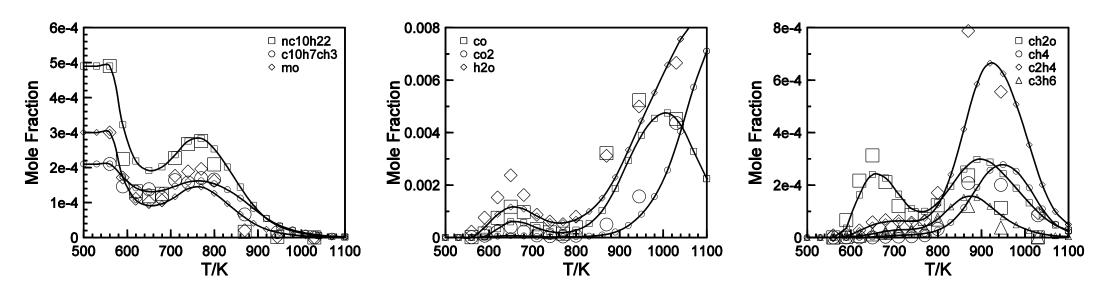
### **Modeling results**

The concentration profiles obtained for the oxidation of the B30 surrogate fuel were compared to the model predictions.

The present model was also successfully tested for the oxidation of pure n-decane, pure methyl octanoate, and pure 1-methylnaphtalene under similar JSR conditions.

Furthermore, the proposed model, not including the methyl octanoate chemistry, was used to simulate the oxidation of commercial and surrogate Diesel fuels [H.P. Ramirez L, K. Hadj-Ali, P. Diévart, G. Moréac, P. Dagaut, Energy Fuels 24(3) (2010) 1668-1676]

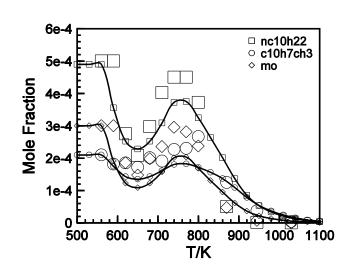
#### Results at 6 atm

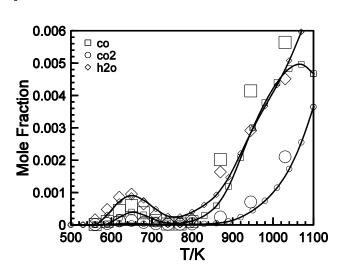


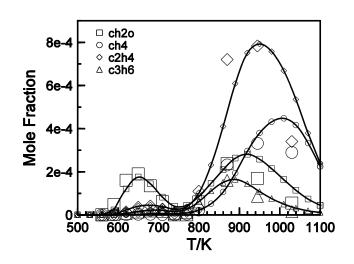
The B30 surrogate Diesel fuel oxidation in a JSR at 6 atm,  $\tau$ = 0.6s, and  $\varphi$ = **0.5**. The experimental data (large symbols) are compared to the computations (lines with small symbols).

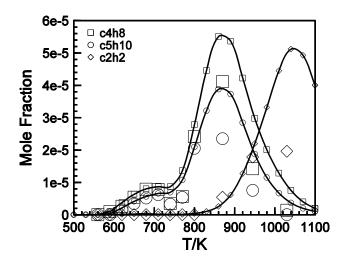
N.B. B30 Surrogate= 490ppm of n-decane + 210ppm of 1-methyl naphthalene + 300ppm of methyl octanoate

#### Results at 6 atm





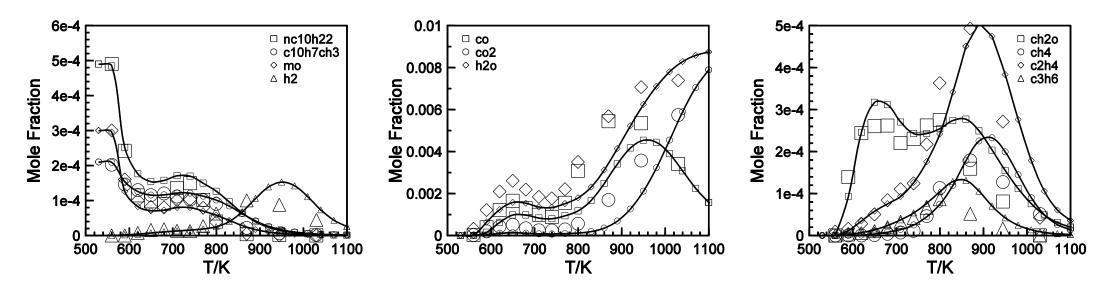




The B30 surrogate Diesel fuel oxidation in a JSR at 6 atm,  $\tau$ = 0.6s, and  $\varphi$ = 1. The experimental data (large symbols) are compared to the computations (lines with small symbols).

N.B. B30 Surrogate= 490ppm of n-decane + 210ppm of 1-methyl naphthalene + 300ppm of methyl octanoate

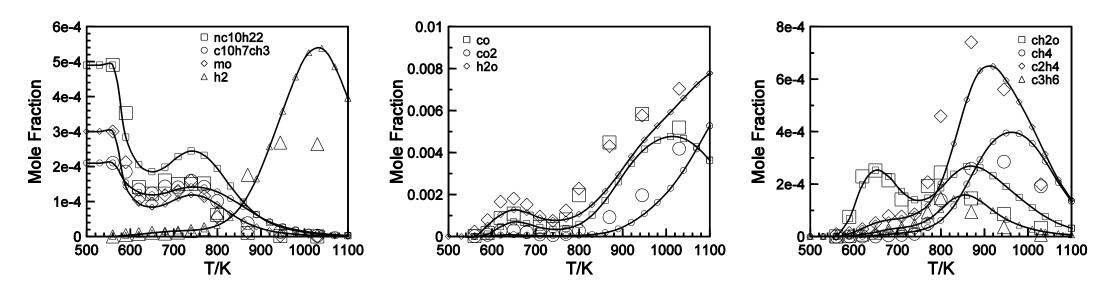
### Results at 10 atm



The B30 surrogate Diesel fuel oxidation in a JSR at 10 atm,  $\tau$ = 1s, and  $\varphi$ = **0.5**. The experimental data (large symbols) are compared to the computations (lines with small symbols).

N.B. B30 Surrogate= 490ppm of n-decane + 210ppm of 1-methyl naphthalene + 300ppm of methyl octanoate

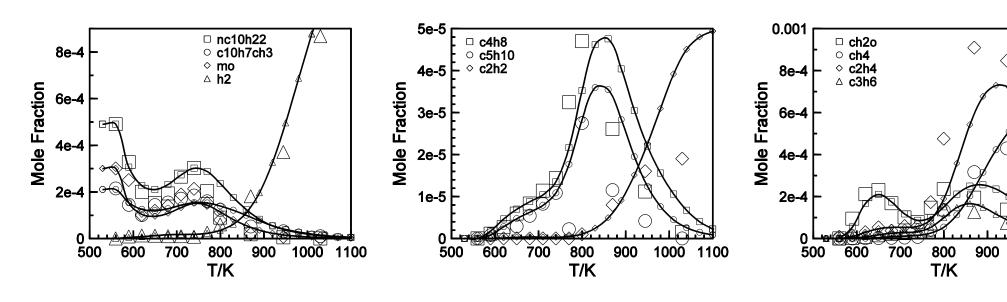
### Results at 10 atm



The B30 surrogate Diesel fuel oxidation in a JSR at 10 atm,  $\tau$ = 1s, and  $\varphi$ = 1. The experimental data (large symbols) are compared to the computations (lines with small symbols).

N.B. B30 Surrogate= 490ppm of n-decane + 210ppm of 1-methyl naphthalene + 300ppm of methyl octanoate

#### Results at 10 atm

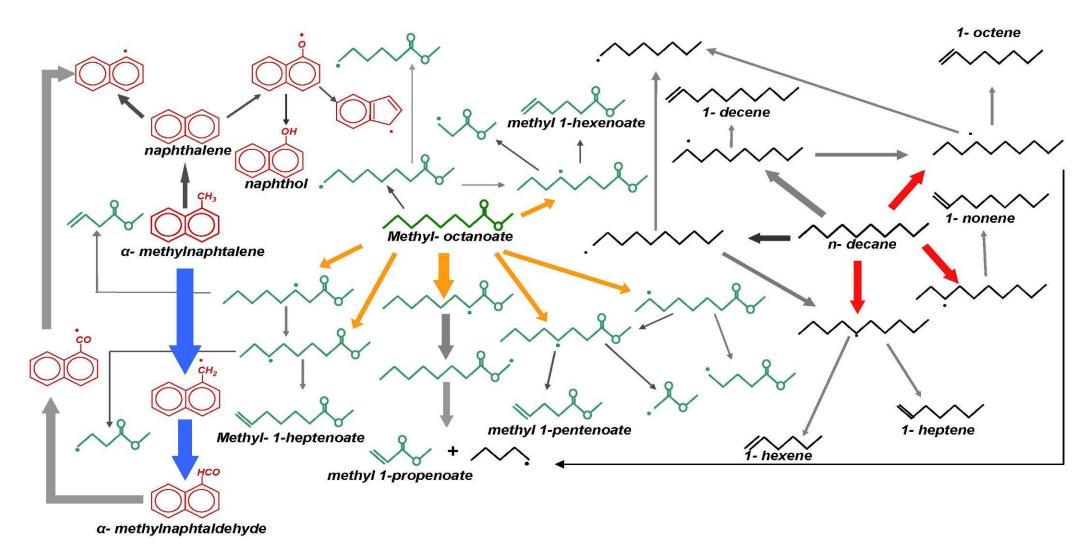


The B30 surrogate Diesel fuel oxidation in a JSR at 10 atm,  $\tau$ = 1s, and  $\varphi$ = **1.5**. The experimental data (large symbols) are compared to the computations (lines with small symbols).

N.B. B30 Surrogate= 490ppm of n-decane + 210ppm of 1-methyl naphthalene + 300ppm of methyl octanoate

H.P. Ramirez L. et al., Proc. Combust. Inst. 33(1), 375–382 (2011)

1000



Rxn pathways analysis for B30 surrogate oxidation (10 atm,  $\phi$ =1, 1030K)

According to the present computations, at 620 K and in fuel-lean conditions ( $\phi$ =0.5 at 10 atm), OH radicals are mostly responsible for the oxidation of n-decane (ca. 95%), methyl octanoate (ca. 89%), and 1-methylnaphtalene (ca. 80%) via

$$n-C_{10}H_{22} + OH = C_{10}H_{21} + H_2O$$

$$C_9H_{18}O_2 + OH = C_9H_{17}O_2 + H_2O$$

$$C_{10}H_7CH_3 + OH = C_{10}H_7CH_2 + H_2O$$

Under these conditions, their formation mainly occurs via the decomposition of alkylhydroperoxy (O<sub>2</sub>QOOH and OQ'OOH) deriving from the oxidation of n-decane and methyloctanoate.

Above ca. 750 K, the transition to the high-temperature oxidation regime occurs. The fuel is rapidly consumed through metathesis reactions with OH and larger amounts of products are formed.

The model predicts the experimentally observed overall reactivity of the fuel and products' formation, although it tends to underestimate the overall rate of oxidation above ca. 800 K. This behavior results from the too strong inhibiting effect of 1-methylnaphtalene on n-decane and methyl octanoate oxidation.

We did not attempt to improve the present simulations by modifying the kinetic parameters used in previous modeling efforts in order to keep this model valid for representing the neat oxidation of the surrogate fuel components, i.e. n-decane, 1-methylnaphtalene, and methyl octanoate.

At 1040 K, OH radicals are still mostly responsible for the oxidation of n-decane (ca. 80%), methyl octanoate (ca. 80%), and 1-methylnaphtalene (ca. 89%) via the same reactions.

The reactions of n-decane with O (ca. 10%) and H (ca. 5%) also contribute to its consumption.

Also, methyl octanoate reacts with H (ca. 8%). Similarly, H-atoms also consume 1-methylnaphtalene (ca. 8%).

Under these conditions, the production of ethylene mainly occurs via  $\beta$ -scissions of alkyl radicals (1-butyl and 1-propyl 30%) whereas the oxidation of ethyl radicals by  $O_2$  also contributes to ethylene formation (20%).

# **Modeling results (cont'd)** Local, first-order sensitivity analyses

Sensitivity analyses showed that at 620K, besides the C<sub>0</sub>-C<sub>2</sub> reactions, the overall reactivity is positively sensitive to the rates of oxidation of n-decane by OH, and the peroxidation of methyl octanoate radicals, i.e. to reactions

$$CH_3(CH_2)_5CH(\bullet)C(=O)OCH_3 + O_2 <=> CH_3(CH_2)_5CH(OO_\bullet)C(=O)OCH_3$$

 $CH_3CH_2CH_2CH_2CH_2CH(OO\bullet)C(=O)OCH_3 <=> CH_3CH_2CH_2CH(\bullet)CH_2CH(OOH)C(=O)OCH_3$ 

As expected, at 1040K, the system is mostly sensitive to the kinetics of the C<sub>0</sub>-C<sub>1</sub> sub-scheme, i.e.

$$H+O_2 <=> OH+O$$

$$HO_2 + OH <=> H_2O + O_2$$

$$CO+OH <=> CO_2+H$$

#### 5.4.3-Pentanol

Because they are renewable, biofuels are attracting great interest as transportation fuels. They can be locally produced, may be less polluting, sometimes more biodegradable, and could reduce net greenhouse gas emissions [1].

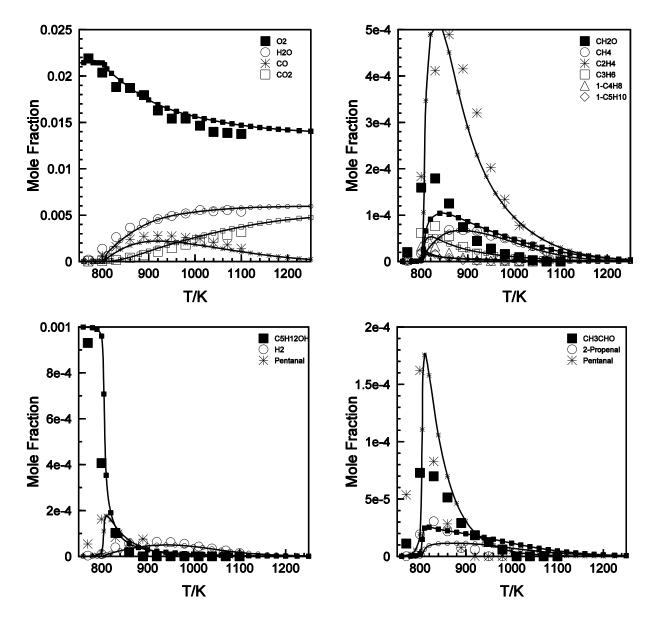
Ethanol accounts for over 90% of all biofuels' production worldwide [2]. However, mixing stability issues may appear with simple alcohols whereas larger alcohols would mix better with petrol-derived fuels thanks to their longer alkyl carbon chain.

Since 1-butanol was announced to be sold soon as a gasoline blending constituent [3], Dagaut and Togbé studied the oxidation of butanol-gasoline surrogate mixtures (85-15 vol%) in a JSR at 10 atm and a kinetic reaction mechanism was derived for modeling the oxidation of butanol-gasoline surrogate mixtures [4].

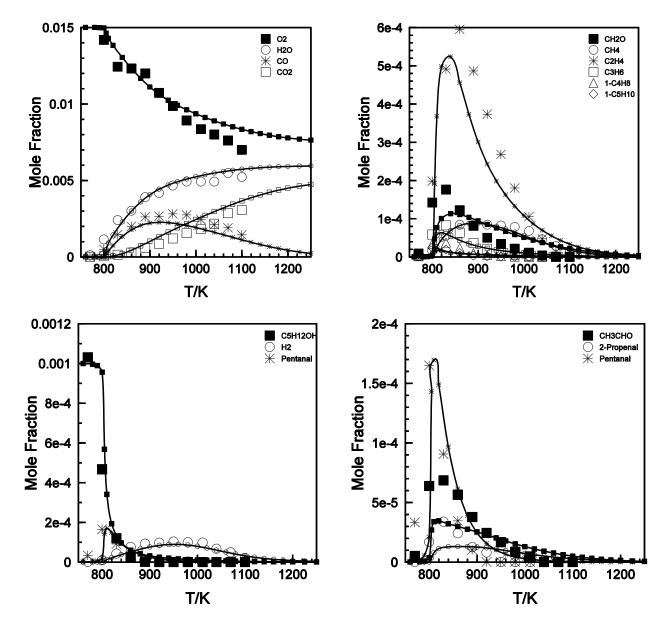
- 1. A. Demirbas, Prog. Energy Combust. Sci. 33 (1) (2007) 1-18.
- 2. IEA World Energy Outlook (2006), ISBN 92-64-10989-7, 500p.
- 3. Dupont Corp. (2006) available at http://www2.dupont.com/Biofuels/en\_US/facts/BiobutanolFactsheet.html
- 4. P. Dagaut and C. Togbé, Energy and Fuels 22 (2008) 3499–3505.

1-Pentanol is among the longer carbon-chain alcohols that could be blended with conventional fuels. However, so far, it received little attention since only engine experiments were reported in the literature [5,6] whereas bio-pentanol could be produced [7,8].

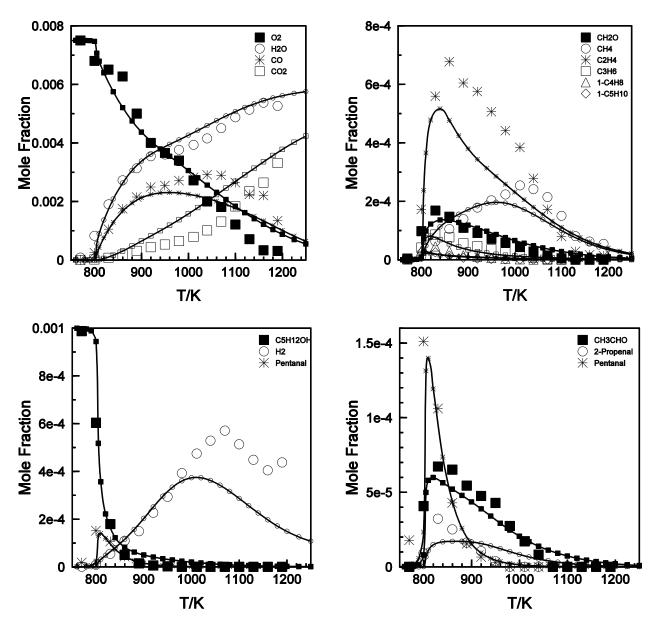
- 5. M. Gautam, D.W. Martin, Proc Instn Mech Engrs Part A 214 (2000) 165-182.
- 6. M. Gautam, D.W. Martin, D. Carder, Proc Instn Mech Engrs Part A 214 (2000) 497-511.
- 7. A. F. Cann, J.C. Liao, Appl. Microbiol. Biotechnol. 85 (2010) 893-899.
- 8. K. Zhang, M.R. Sawaya, D.S. Eisenberg, J.C. Liao, Proc. Natl. Acad. Sci. USA 105 (2008) 20653-20658.



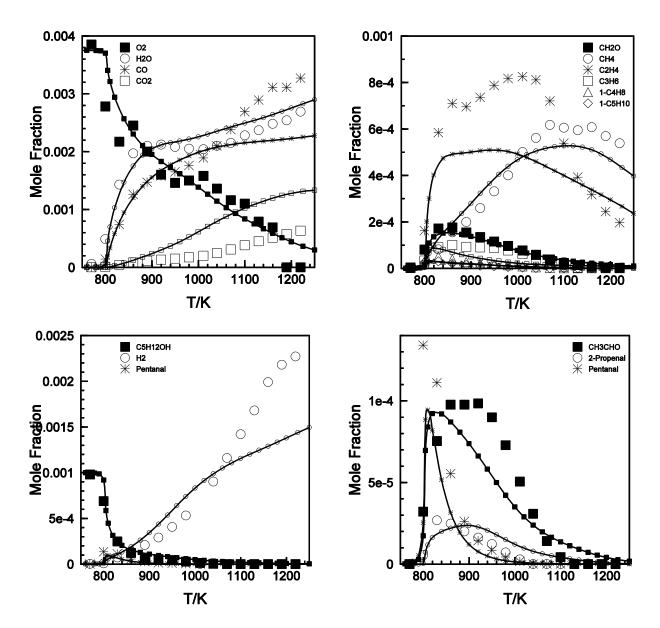
Experimental (large symbols) and computed (lines and small symbols) concentration profiles obtained from the oxidation of 1-pentanol in a JSR at  $\phi$  = 0.35, P = 10 atm,  $\tau$  = 0.7s. **C. Togbé et al.**, Proc. Combust. Inst. **33**(1), 367–374 (2011)



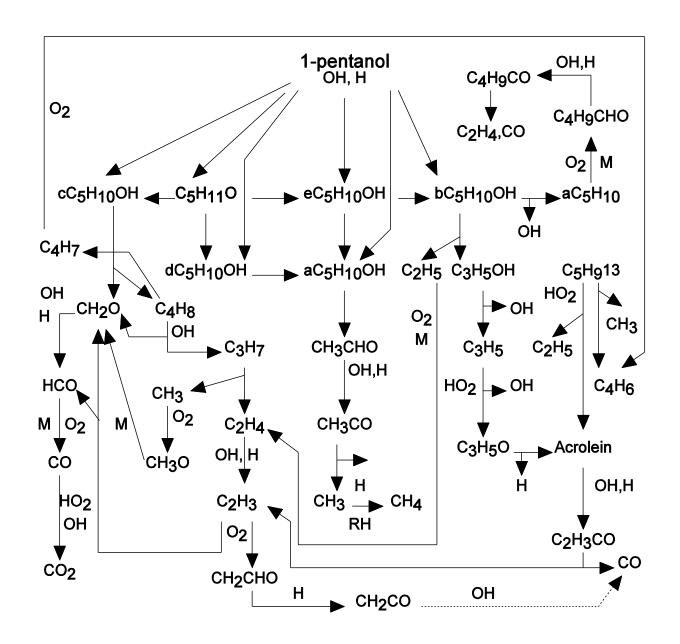
Experimental (large symbols) and computed (lines and small symbols) concentration profiles obtained from the oxidation of 1-pentanol in a JSR at  $\phi$  = 0.5, P = 10 atm,  $\tau$  = 0.7s. **C. Togbé et al.**, Proc. Combust. Inst. **33**(1), 367–374 (2011)



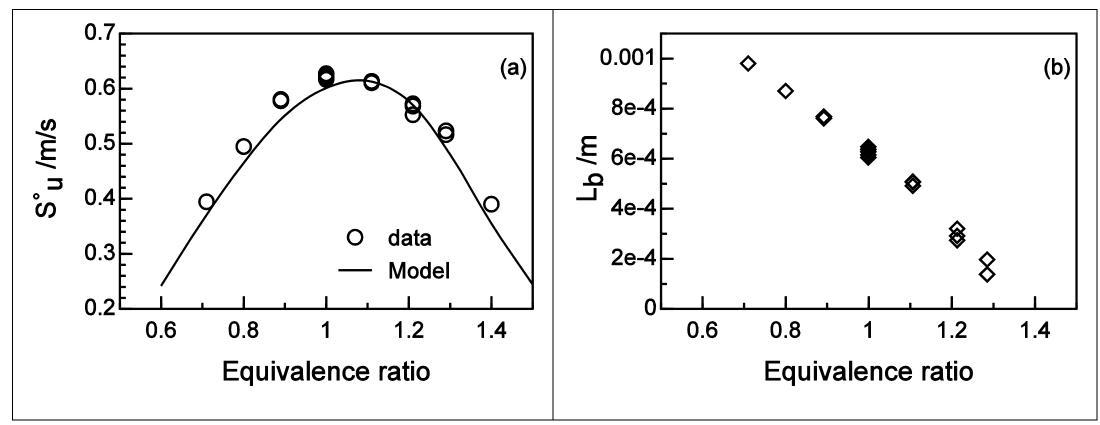
Experimental (large symbols) and computed (lines and small symbols) concentration profiles obtained from the oxidation of 1-pentanol in a JSR at  $\phi$  = 1, P = 10 atm,  $\tau$  = 0.7s. **C. Togbé et al.**, Proc. Combust. Inst. **33**(1), 367–374 (2011)



Experimental (large symbols) and computed (lines and small symbols) concentration profiles obtained from the oxidation of 1-pentanol in a JSR at  $\phi$  = 2, P = 10 atm,  $\tau$  = 0.7s. **C. Togbé et al.**, Proc. Combust. Inst. **33**(1), 367–374 (2011)



Reaction paths from the kinetic modeling of 1-pentanol oxidation in a JSR at 10 atm. **C. Togbé et al.,** Proc. Combust. Inst. **33**(1), 367–374 (2011)



Laminar burning velocities of 1-pentanol/air mixtures at T=423 K and 1 atm (a) and burnt gases Markstein lengths (b).

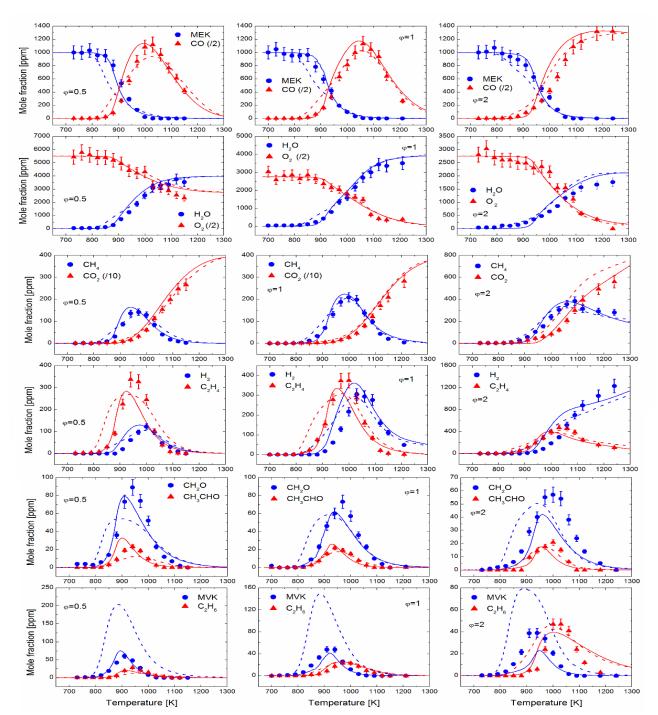
C. Togbé et al., Proc. Combust. Inst. 33(1), 367-374 (2011)

### **5.4.4 2-Butanone**

Methyl Ethyl Ketone (MEK) is a four carbon linear ketone that can be produced through either chemical and biological conversion of furfural [1] or oxidation of 2-butanol. Besides its potential application as a fuel substitute [2], MEK is also used as solvent in the paint and adhesive industry. With these considerations, and since MEK is the smallest ketone exhibiting secondary C-H bonds, this fuel is a molecule of choice to investigate the specificities of keto groups oxidation.

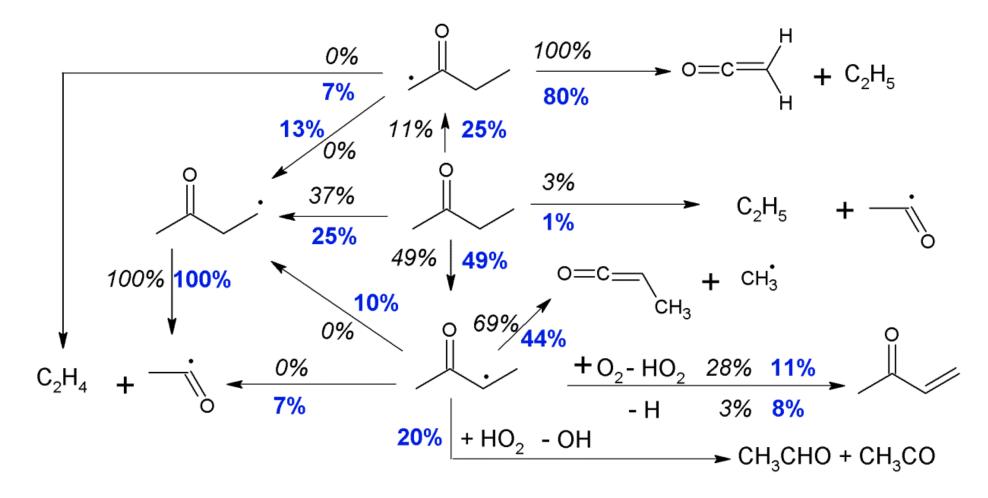
[1] E.R. Sacia, M. Balakrishnan, M.H. Deaner, K.A. Goulas, F.D. Toste, A.T. Bell, ChemSusChem, 8 (10)(2015) 1726-1736.

[2] F. Hoppe, U. Burke, M. Thewes, A. Heufer, F. Kremer, S. Pischinger, Fuel, 167 (2016) 106-117.



Comparison between experimental (symbols) and computed (Solid line: this work, dashed line: Serinyel et al. [10]) concentration profiles for the oxidation of MEK at 10 atm at different equivalence ratios.

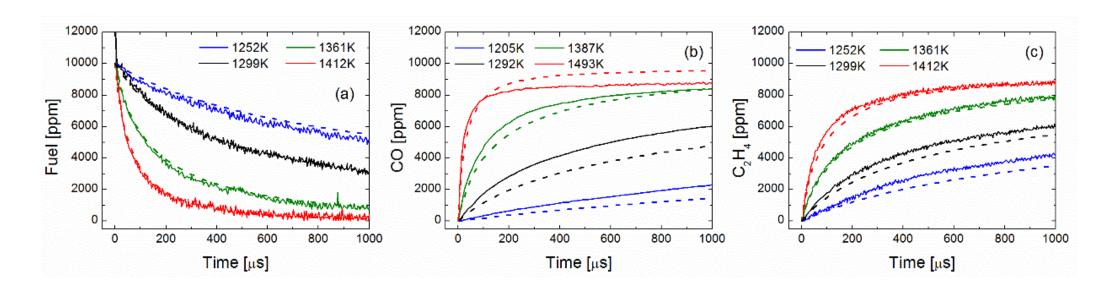
## Rate of production analysis



Rate of production analyses of MEK oxidation at 950K,  $\phi$ =1 and 10 atm. Blue values: This work, black italic values: Serinyel et al. .

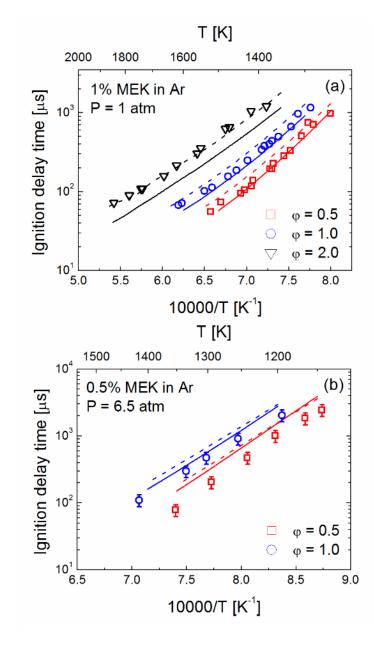
S. Thion, Proc. Combust. Inst. 36, 459-467 (2017)

# Pyrolysis and high temperature oxidation

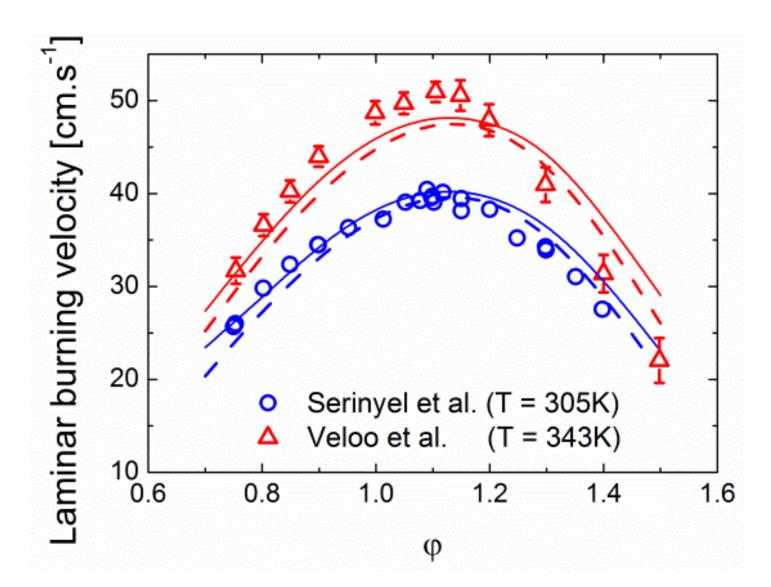


Species concentration profiles for the pyrolysis of 1% MEK in argon at an average pressure of 1.5 atm (solid line: experiments by Lam et al., dashed line: this work.)

S. Thion, Proc. Combust. Inst. 36, 459-467 (2017)



Ignition delay times of MEK/O<sub>2</sub>/Ar mixtures. Solid line: this work, dashed line: Serinyel et al. S. Thion, Proc. Combust. Inst. **36**, 459–467 (2017)



Laminar burning velocities of MEK in air. Dashed line: Serinyel et al., solid line: this work. S. Thion, Proc. Combust. Inst. **36**, 459–467 (2017)

#### **5.4.5 ML and DEE**

Among proposed chemical platforms, levulinic acid is one of the most interesting

Alkyl levulinates produced from levulinic acid esterification contain keto and ester functional groups. The synthesis of these compounds starts with hemicellulose and cellulose hydrolysis to xylose and glucose, respectively. They can be converted to furfural and 5-hydroxymethylfurfural which in turn can be converted to levulinic acid.

Methyl levulinate (DCN ≈7.8) is considered here.

• Another interesting biofuels, produced via dehydration of bio-ethanol, is **diethyl ether** suitable for C.I. engines (CN >125).

#### **MODELING**

The CHEMKIN II computer code was used for the kinetic modeling of the oxidation of the two fuels studied in a jet-stirred reactor.

The chemical kinetic reaction mechanism for ML oxidation contained **704 species involved** in **3870 reversible reactions**; that for DEE oxidation contained **471 species involved in 2861 reversible reactions**\*.

Core mechanism: C<sub>0</sub>-C<sub>3</sub> oxidation mechanism extended to model the oxidation of other oxygenates [a]

[a] S. Thion et al., Combust. Flame 185 (2017) 4-15; A.M. Zaras et al., Energy & Fuels 31 (6) (2017) 6194-6205.

<sup>\*</sup> sub-mec included in DBE oxidation mechanism.

We previously reported computed rate constants for H-abstractions by OH, H and CH<sub>3</sub> on ML [a].

H-abstraction reactions by other radicals were not found to be sensitive, and simple analogies were applied with no specific corrections for k(T).

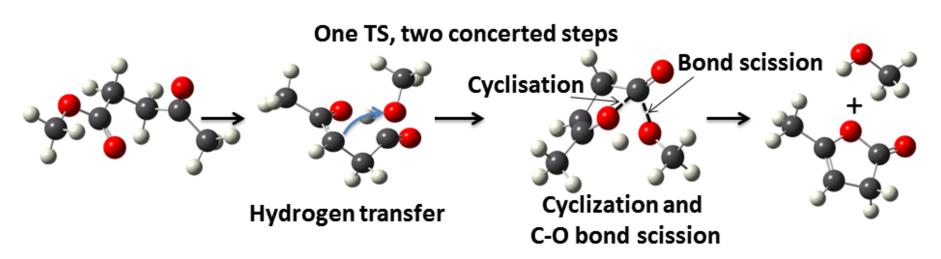
[a] S. Thion, A.M. Zaras, M. Szori, P. Dagaut, Phys. Chem. Chem. Phys. 17 (36) (2015) 23384-23391

Additional theoretical calculations using the same computational strategy were performed in order to elucidate the decomposition pathways of ML and to obtain missing thermochemical properties.

These calculations were carried out using the Gaussian09 code [a] at the G3//MP2/aug-cc-pVDZ and G3B3 levels of theory.

[a] M.J. Frisch et al., Gaussian 09, Revision D.01; Wallingford CT, 2009

The presence of oxygenated groups, and in particular of the ester group, favors molecular reactions. Ethyl (and larger) esters can easily decompose by H–transfer to produce an acid and an olefin. This type of reaction cannot take place here because a carbon chain is needed on the alcohol side, while methyl levulinate has only one carbon. Therefore, other possible pathways for the molecular reaction decomposition of methyl levulinate were explored by theoretical chemistry methods and a reaction similar to that of esters has been identified. It involves a complex TS:



Structure of the transition state during the molecular reaction yielding methanol and 5-methyl-2(3H)-furanone from ML.

The high-pressure limit rate constant was computed at the G3B3 and G3//MP2/aug-cc-pVDZ levels of theory by following the strategy described in our previous work. We assumed hindered rotors cancel out, as in the work of Al Abbad et al.[a].

This molecular reaction is much slower than that observed in the case of esters: Its rate constant is 100 times lower at 1500 K and almost 200 times at 1000 K.

However, its low activation barrier allows it to play an important role.

[a] M. Al Abbad, B.R. Giri, M. Szori, A. Farooq, Proc. Combust. Inst. 36 (1) (2017) 187-193.

A second reaction has also been identified. It involves another interaction between the two oxygenated groups in ML. It consists of a H–transfer from the C–"5" to the oxygen atom in C=O of the ester group. This transfer is accompanied by cyclization between the oxygen atom of the ketone group and C–"1" and the formation of a C=C double bond to give 2-methylene-5-methoxy-5-hydroxy-tetrahydrofuran ("oxyTHF"):



Formation of 2-methylene-5-methoxy-5-hydroxy-tetrahydrofuran.

The rate constant for this reaction was calculated with G3B3 and G3//MP2/aug-cc-pVDZ levels of theory.

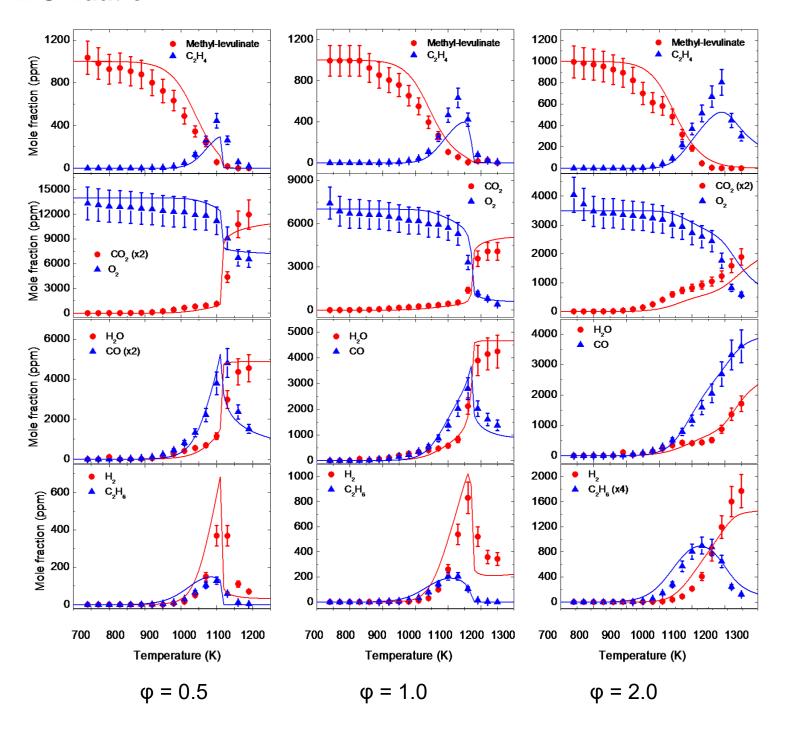
The α-angelica lactone is likely to undergo a molecular decomposition reaction similar to that of cyclopentanone yielding methyl vinyl ketone and CO:

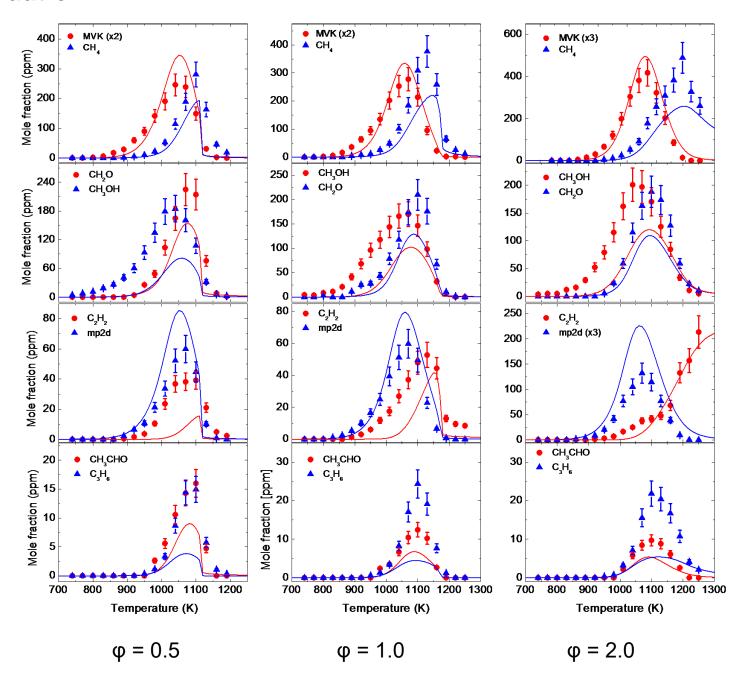
Molecular decomposition of  $\alpha$ -angelica lactone.

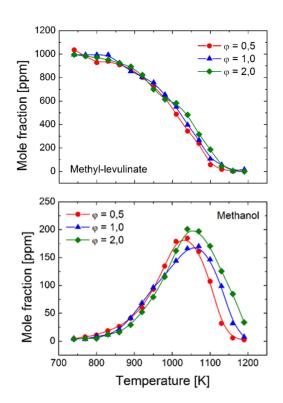
The rate constant for this reaction was calculated using the G3B3 method and the transition state theory.

12 intermediate stable species were identified and quantified in addition to the reactants (O<sub>2</sub>, ML) and the final products (H<sub>2</sub>O, CO<sub>2</sub>).

No reactivity below 750K



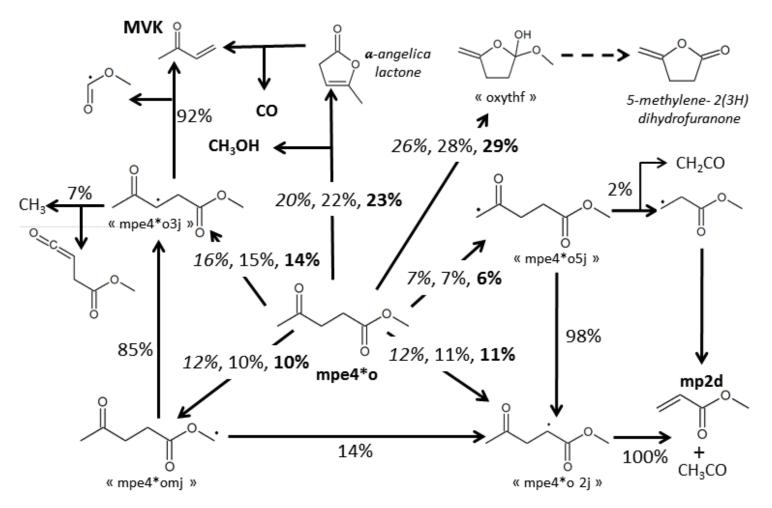




Consumption of ML and production of methanol during the oxidation of ML in a JSR.

Methanol production starts at the same temperature as fuel consumption (around 850 K) and in the same proportions for the 3 equivalence ratios.

Differences are observed ~1000 K, when the consumption of methanol > formation. These experimental observations indicate that a large fraction of the fuel is consumed by molecular reactions yielding methanol.

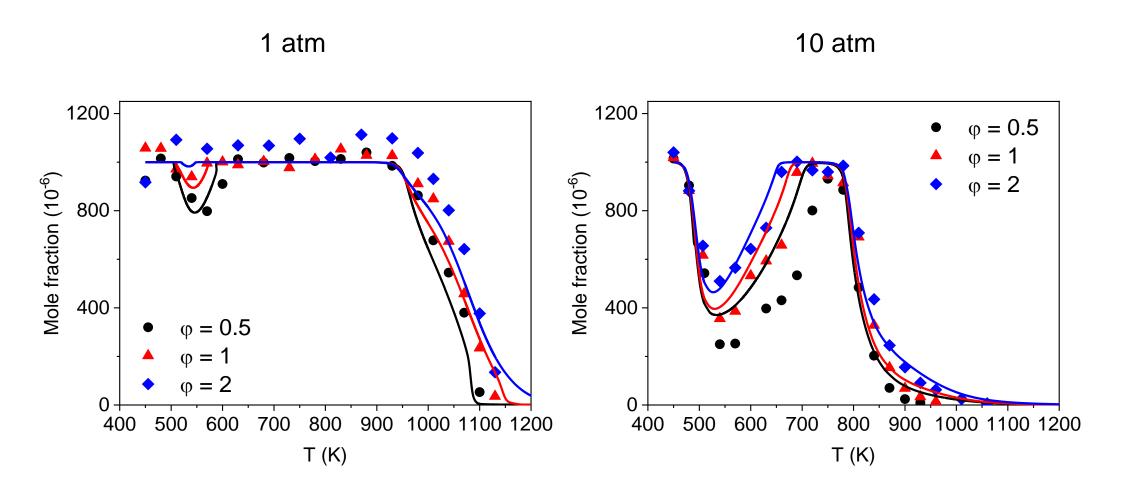


Normalized rates of reaction analysis at  $\varphi$ = 1, 1 atm and 1000 K corresponding to ~ 50% of fuel consumption. Values are also given at  $\varphi$  = 0.5 (italics) and 2 (bold) for the primary reactions. Bold species are measured and italic species are detected in trace amounts.

Proc. Combust. Inst. **37** (1), 381–388 (2019)

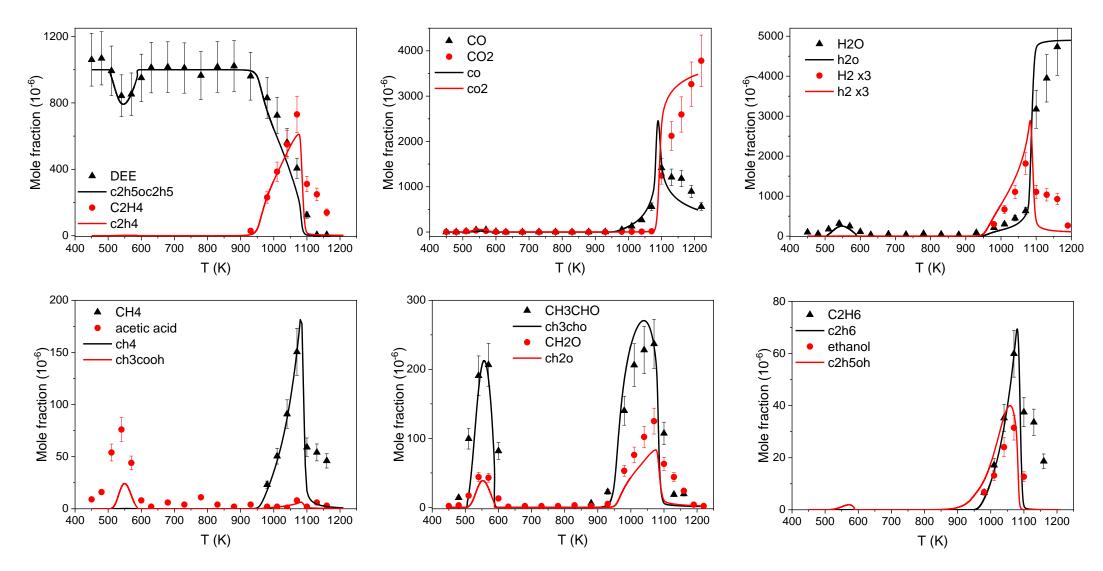
#### **MODELING: DEE Oxidation**

- Beta-scission reactions of fuel radicals and QOOH radicals are adopted from the CBS-QB3 calculations of Sakai et al. [a], and from our previous calculations on DBE [b].
- Other reactions related to low-temperature chemistry are taken analogous to our previous DBE study [b].
- Unimolecular decomposition reactions of DEE were taken from the study of Yasunaga et al. [c].
- Thermochemistry of the fuel, fuel radical as well as all related low-temperature species were taken from the theoretical study of Sakai et al. [a], and for other species these were calculated using using the group additivity method of Benson [d].
- [a] Y. Sakai et al. Proceedings of the Combustion Institute 36 (2017) 195–202.
- [b] S. Thion et al. Combustion and Flame 185 (2017) 4-15.
- [c] K. Yasunaga et al. Journal of Physical Chemistry A 114 (2010) 9098-9109.
- [d] S.W. Benson, Thermochemical Kinetics, Wiley, New York, 1976.



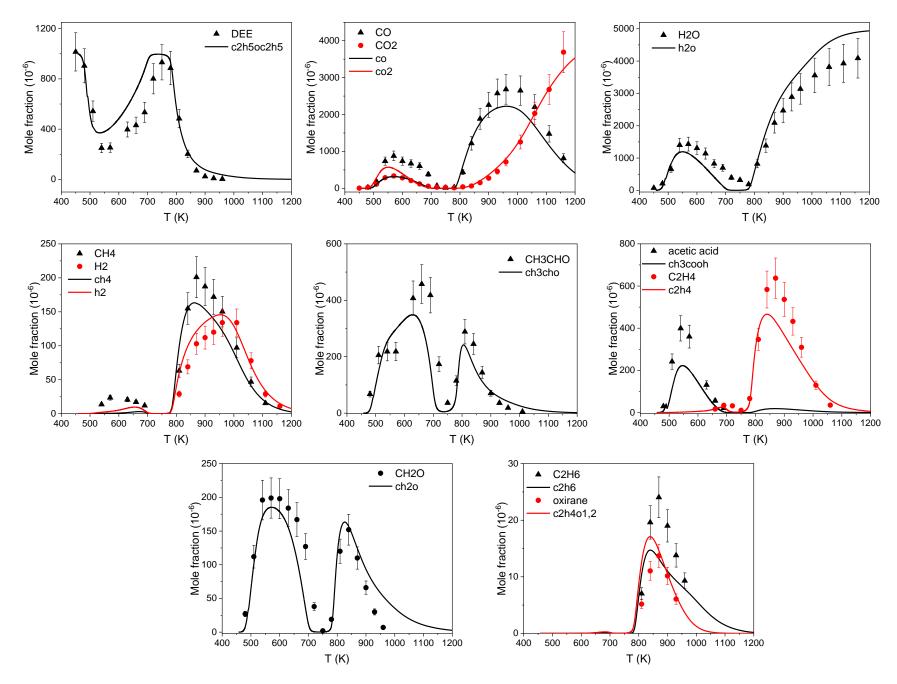
DEE mole fraction evolution as a function of temperature; lines represent simulations.

Z. Serinyel et al., Combust. Flame 193, 453–462 (2018)



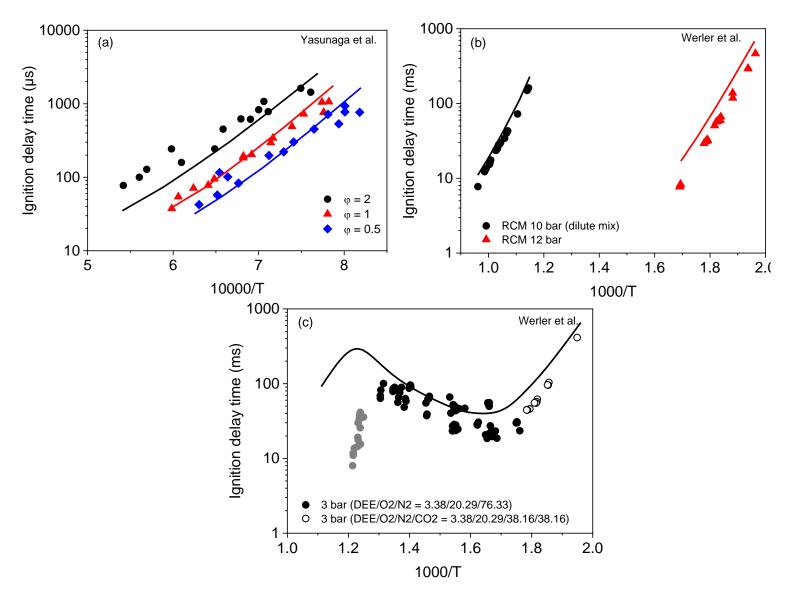
Mole fraction for the  $\varphi$  = 0.5 experiment at 1 atm, initial mole fraction of DEE: 1000 ppm, t = 0.07s.

Z. Serinyel et al., Combust. Flame 193, 453-462 (2018)



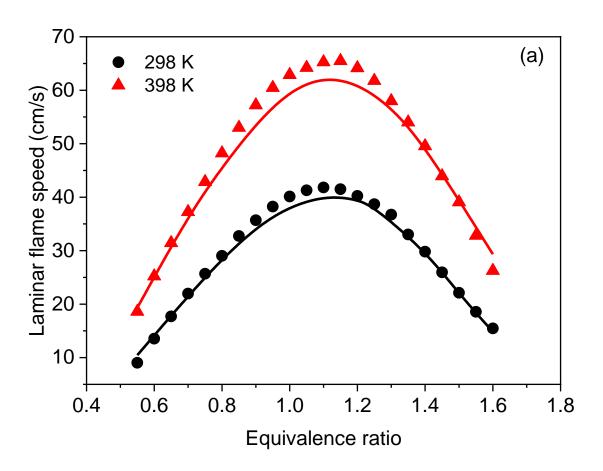
DEE oxidation  $\varphi = 0.5$  experiment at 10 atm, t = 0.7s.

Reaction pathways at (a) 510 K and (b) 690 K (10 atm,  $\varphi$  = 0.5)



Shock tube and RCM ignition (a) 1% DEE in Ar, p = 1 atm; (b) 0.698% DEE in Ar,  $\varphi = 1$ , p = 10-12 bar; (c) DEE in air,  $\varphi = 1$ , RCM by Werler et al. 2015

Z. Serinyel et al., Combust. Flame 193, 453-462 (2018)

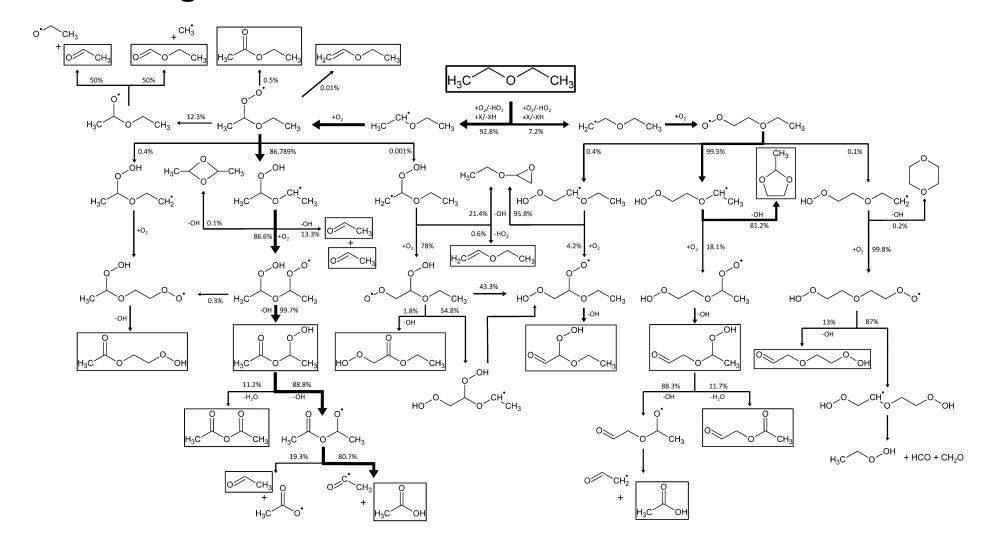


Laminar flame speed of DEE/air mixtures [a] as a function of  $\varphi$  at 1 atm,  $T_u$  = 298 and 398 K.

Z. Serinyel et al., Combust. Flame 193, 453-462 (2018)

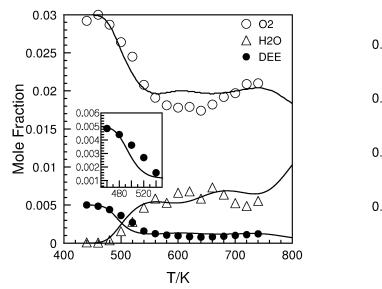
[a] F. Gillespie et al. Energy 43 (2012) 140-145.

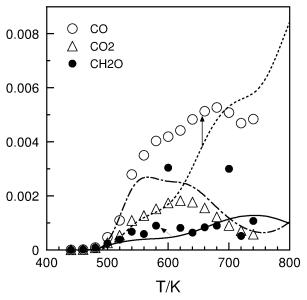
## Further investigations of DEE cool-flame

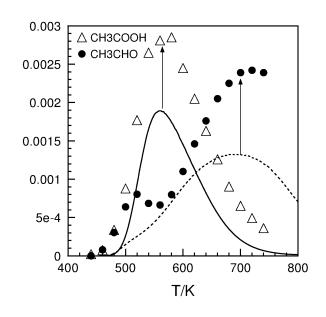


Reaction paths for the oxidation of DEE at 530 K using the kinetic mechanism of Tran et al. [Proc. Combust. Inst. 37 (1) (2019) 511-519]. Species in boxes were detected in the present study. Thick arrows indicate major reaction routes.

## Further investigations of DEE cool-flame

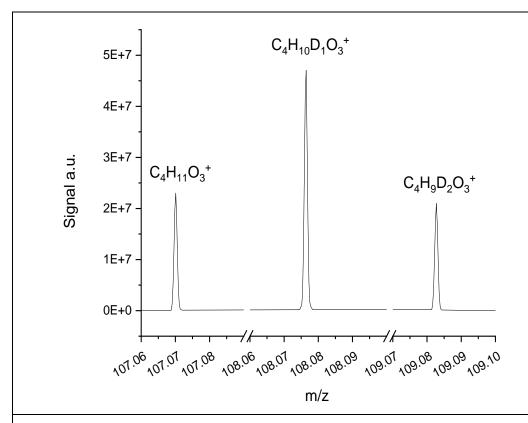


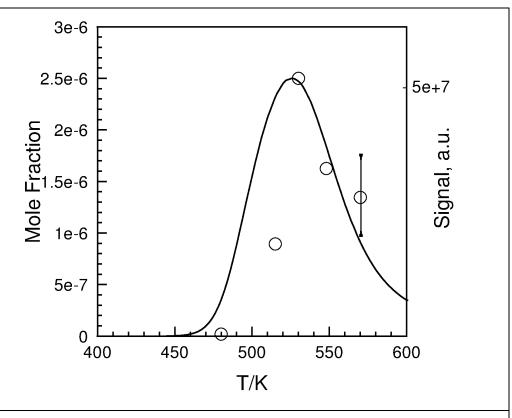




Oxidation of 5000 ppm of diethyl ether in a JSR at 10 bar. Experimental results (symbols) and computations (lines) are presented. Combust. Flame 228, 340-350 (2021) https://doi.org/10.1016/j.combustflame.2021.02.007

# Further investigations of DEE cool-flame: ROOH and diols



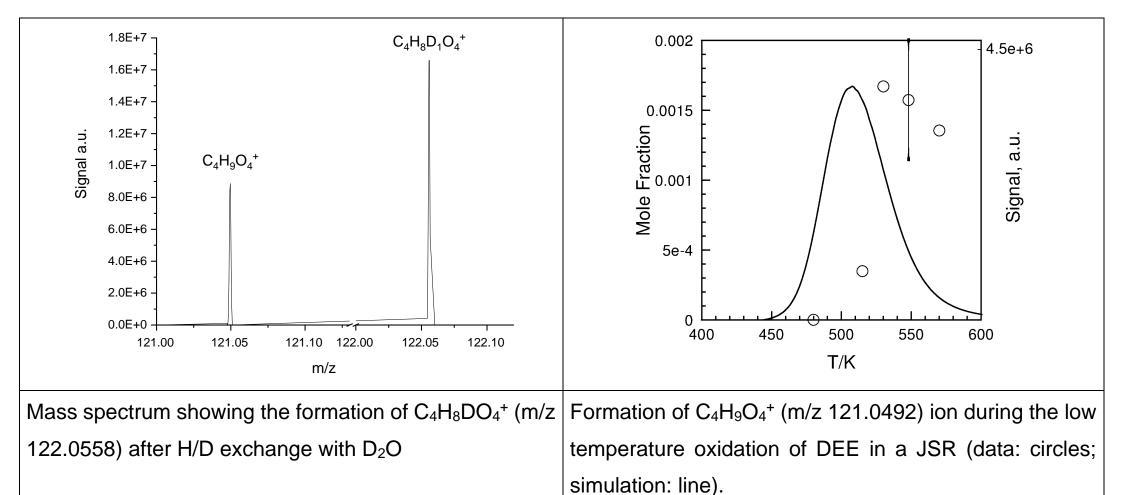


Mass spectrum showing the formation of  $C_4H_{10}DO_3^+$  (m/z 108.0764) and  $C_4H_9D_2O_3^+$  (m/z 109.0827) after H/D diolshydroperoxides exchange with  $D_2O$ . Analyses were performed in FIA/APCI (+). No signal for  $C_4H_{10}DO_3^+$  and  $C_4H_9D_2O_3^+$  could be observed before reaction with  $D_2O$ 

Formation of  $C_4H_{10}O_3$  during the low temperature oxidation of DEE in a JSR (data: circles, simulation: line). The integrated signal was obtained in FIA/APCI positive mode of  $C_4H_{11}O_3^+$ (m/z 107.0699) ion

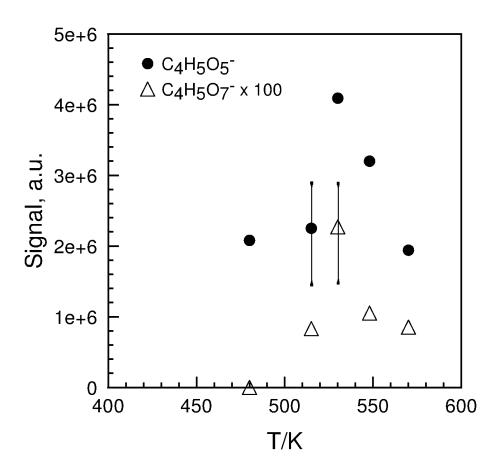
Combust. Flame **228**, 340-350 (2021) <a href="https://doi.org/10.1016/j.combustflame.2021.02.007">https://doi.org/10.1016/j.combustflame.2021.02.007</a>

# Further investigations of DEE cool-flame: KHPs



Combust. Flame 228, 340-350 (2021) https://doi.org/10.1016/j.combustflame.2021.02.007

# Further investigations of DEE cool-flame: HOMs



Formation of highly oxygenated molecules during the low temperature oxidation of DEE in a JSR: ( $\bullet$ ) di-keto-hydroperoxides  $C_4H_6O_5$  corresponding to  $C_4H_5O_5^-$  ion (m/z 133.0142). ( $\Delta$ ) di-keto-dihydroperoxides  $C_4H_6O_7$  corresponding to  $C_4H_5O_7^-$  ion (m/z 165.0041).

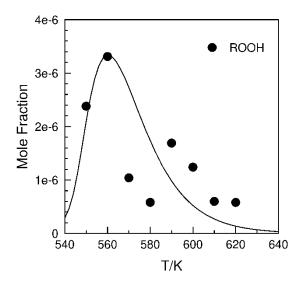
Combust. Flame 228, 340-350 (2021) https://doi.org/10.1016/j.combustflame.2021.02.007

## **THF cool-flame: KHPs formation**

Initial radicals formed by H-atom abstraction on THF	
α- position	β-position
HC. O	0
	сн-
ROOHs structure	
НО	0
	HO-0
KHPs structure	
НО	но
ο α, β	ο΄ β, α
0 0	0
ο HO΄ α, β'	ο΄΄ ο΄ β, β'
0 O O O O O O O O O O O O O O O O O O O	OOOO
α, α'	β, α'

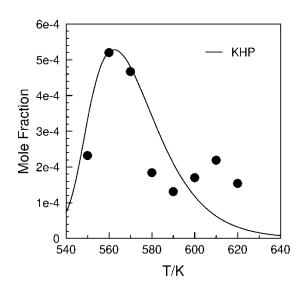
Energy & Fuels **35**(9) 7242–7252 (2021) https://doi.org/10.1021/acs.energyfuels.0c03291

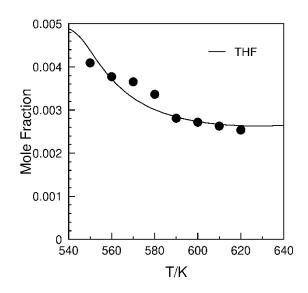
### **THF cool-flame: ROOH formation**



Formation of C<sub>4</sub>H<sub>8</sub>O<sub>3</sub> in a JSR where 5000 ppm of fuel are oxidized. Analyses were performed in FIA and APCI (+) mode. The data (symbols) represent the signal recorded at m/z 105.0545, scaled to the maximum computed mole fraction (line, Fenard et al.). Energy & Fuels 35(9) 7242–7252 (2021) https://doi.org/10.1021/acs.energyfuels.0c03291

### THF cool-flame: KHPs formation



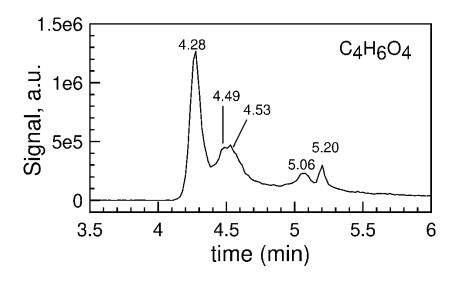


Formation of C<sub>4</sub>H<sub>6</sub>O<sub>4</sub> in a JSR where 5000 ppm of fuel are oxidized. Analyses were performed in FIA and APCI (+) mode. The data (symbols) represent the signal recorded at m/z 119.0338 (C<sub>4</sub>H<sub>7</sub>O<sub>4</sub><sup>+</sup>), scaled to the KHPs maximum computed mole fraction

Consumption of THF under the same conditions based on m/z 73.0647 ( $C_4H_9O^+$ ). The data (symbols) are compared to simulations (lines).

Energy & Fuels **35**(9) 7242–7252 (2021) <a href="https://doi.org/10.1021/acs.energyfuels.0c03291">https://doi.org/10.1021/acs.energyfuels.0c03291</a>

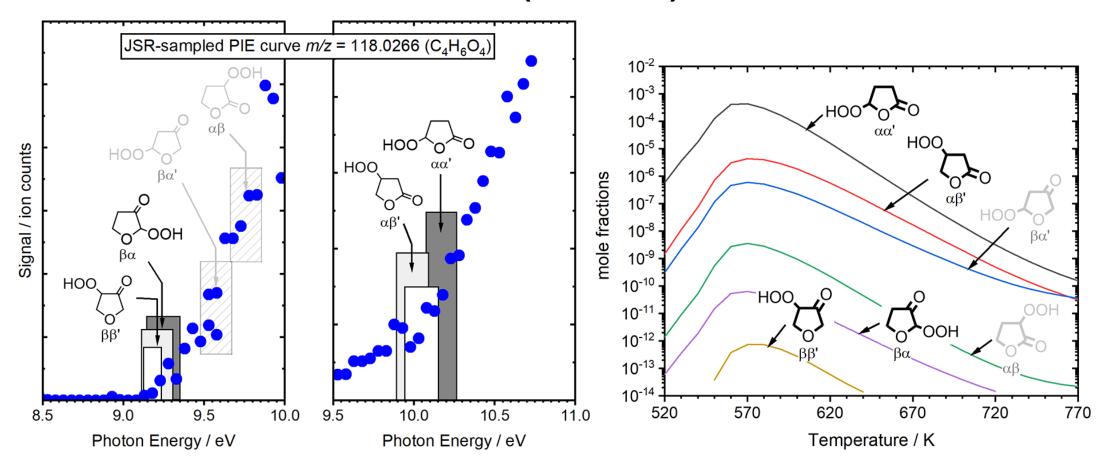
### THF cool-flame: KHPs formation



Chromatographic separation on a Hypercarb PGC column (100% ACN, 100 µL/min, 40°C) of KHPs isomers (C<sub>4</sub>H<sub>6</sub>O<sub>4</sub>) obtained by THF oxidation at 590 K. The APCI + mode was used.

Energy & Fuels **35**(9) 7242–7252 (2021) <a href="https://doi.org/10.1021/acs.energyfuels.0c03291">https://doi.org/10.1021/acs.energyfuels.0c03291</a>

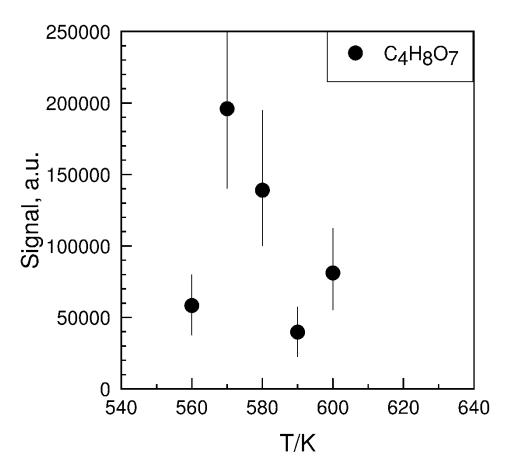
# THF cool-flame: KHPs formation (ALS data)



**Left**: Experimentally observed photoionization efficiency curve of m/z = 118.0266 (C4H6O4) (symbols) after molecular-beam sampling of intermediates of THF oxidation in a jet-stirred reactor from 8.5 to 10.0 eV (left panel) and 9.5 to 11.0 eV (right panel). The experimentally observed ionization thresholds are indicated with white boxes and are compared with theoretically predicted ionization energies (marked in gray) of the six conceivable keto-hydroperoxide isomers. **Right**: Kinetic modeling using the model of Fenard et al.

From Hansen et al., DOI: 10.1021/acs.jpca.9b07017

### **THF cool-flame: HOMs formation**



Formation of  $C_4H_8O_7$  in a JSR where 5000 ppm of fuel are oxidized. Analyses were performed in FIA and APCI (-) mode. The data (symbols) represent the signal recorded at m/z 167.0191 ( $C_4H_7O_7$ -).

Energy & Fuels 35(9) 7242-7252 (2021) https://doi.org/10.1021/acs.energyfuels.0c03291

## **SUMMARY**

Gasoline

Diesel

Jet fuel

Biofuels: biodiesel, ketones, alcools, ethers, levulinates. Emphasis on low-T products

## Literature

- I. Glassman et al., Combustion, Elsevier, ISBN: 9780124079137
- C.K. Law, Combustion physics, Cambridge press, <a href="https://doi.org/10.1017/CBO9780511754517">https://doi.org/10.1017/CBO9780511754517</a>
- W.C. Gardiner Jr, Combustion Chemistry, Springer, doi 10.1007/978-1-4684-0186-8
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- R.G. Gilbert, Theory of Unimolecular and Recombination Reactions, John Wiley and Sons Ltd, ISBN10 0632027495
- M.J. Pilling, Low-temperature Combustion and Autoignition, Elsevier, ISBN 0444543791
- M.J. Pilling, I.W.M. Smith, Modern Gas Kinetics: Theory, Experiment and Application, Blackwell Science Ltd, ISNB 0632016159