



**HAL**  
open science

# Frequency and Amplitude Modulations of a Moving Structure in Unsteady Non-Homogeneous Density Fluid Flow

Tolotra Emerry Rajaomazava, Mustapha Benaouicha, Jacques-André Astolfi, Abdel-Ouahab Boudraa

► **To cite this version:**

Tolotra Emerry Rajaomazava, Mustapha Benaouicha, Jacques-André Astolfi, Abdel-Ouahab Boudraa. Frequency and Amplitude Modulations of a Moving Structure in Unsteady Non-Homogeneous Density Fluid Flow. *Fluids*, 2021, 6 (3), pp.130. 10.3390/fluids6030130 . hal-03795032

**HAL Id: hal-03795032**

**<https://hal.science/hal-03795032>**

Submitted on 3 Oct 2022

**HAL** is a multi-disciplinary open access archive for the deposit and dissemination of scientific research documents, whether they are published or not. The documents may come from teaching and research institutions in France or abroad, or from public or private research centers.

L'archive ouverte pluridisciplinaire **HAL**, est destinée au dépôt et à la diffusion de documents scientifiques de niveau recherche, publiés ou non, émanant des établissements d'enseignement et de recherche français ou étrangers, des laboratoires publics ou privés.

# Frequency and amplitude modulations of a moving structure in unsteady non-homogeneous density fluid flow

T. Emerry RAJAOMAZAVA III <sup>1,\*</sup>, Mustapha BENAOUICHA <sup>2,3</sup>, Jacques-André ASTOLFI <sup>4</sup> and Abdel-Ouahab BOUDRAA <sup>4</sup> 

<sup>1</sup> Computed Wing Sail, 151 Boulevard de l'Hopital, 75013 Paris, France ; emerry.rajaomazava@computedwingsail.com

<sup>2</sup> SEGULA TEHCHNOLOGIES, Research and Innovation Unit in Naval and Energy engineering, 19 Rue d'Arras, 92000 Nanterre, France; mustapha.benaouicha@segula.fr

<sup>3</sup> Normandie University, Unicaen, Cherbourg University Laboratory of Applied Sciences (Lusac)

<sup>4</sup> Ecole Navale/Arts et Metiers Institute of Technology, IRENav, BCRM Brest, CC 600, 29240 BREST Cedex 9, France; Jacques-Andre.astolfi@ecole-navale.fr; abdel.boudraa@ecole-navale.fr

\* Correspondence: emerry.rajaomazava@computedwingsail.com; Tel.: +33-6829-894-85 (E.R.)

**Abstract:** A fluid-structure interactions effects on the dynamics of a hydrofoil immersed in a fluid flow of non-homogeneous density is presented and analyzed. A linearized model is applied to solve the fluid-structure coupled problem. A fluid density variations along the hydrofoil upper surface, based on the sinusoidal cavity oscillations, is used. It is shown that for the steady cavity case, the value of cavity length  $L_p$  does not affect the amplitude of the hydrofoil displacements. However, the natural frequency of the structure increases according to  $L_p$ . In the unsteady cavity case, the variations of the added mass and added damping (induced by the fluid density rate of change) generate frequency and amplitude modulations in the hydrofoil dynamics. In order to analyse this phenomena, the empirical mode decomposition, a well established data-driven method to handle such modulations, is used.

**Keywords:** fluid-structure interaction; added mass; added damping; frequency modulation; amplitude modulation ; non-homogeneous fluid density; cavity oscillations; empirical mode decomposition; intrinsic mode functions

**Citation:** Rajaomazava, T. E.; Benaouicha, M.; Astolfi, J-A.; Boudraa, A.O. Frequency and amplitude modulations of a moving structure in unsteady non-homogeneous density fluid flow. *Journal Not Specified* **2021**, *1*, 0. <https://doi.org/>

Received:

Accepted:

Published:

**Publisher's Note:** MDPI stays neutral with regard to jurisdictional claims in published maps and institutional affiliations.

**Copyright:** © 2021 by the authors. Submitted to *Journal Not Specified* for possible open access publication under the terms and conditions of the Creative Commons Attribution (CC BY) license (<https://creativecommons.org/licenses/by/4.0/>).

## 1. Introduction

Fluid structure interaction (FSI) problems occur when the fluid loading greatly affects the structure's dynamics and the structure displacement locally affects the fluid flow. Initially studied with simplified models, **the simulation of complex coupled problems has developed considerably in recent years**. The state-of-the-art in this field is now **very** mature and several papers with different **fields** of application domains can be found in literature [1–3].

The new challenge of FSI problem analysis consists in taking into account complex phenomena, observed **both in fluid and solid mechanics, especially in the field of fluids where the dynamics are subjected to many physical** quantities such as velocity, pressure, density or temperature. **This work focuses on FSI** effects in a non-homogeneous density flow. **Recent work has pointed out that two-phase flow has an impact on the** fluid structure interaction for various devices, such as propeller blades or hydrofoils [4–9]. However, **very few published works address** the problem of estimating this impact on the structure dynamics. This work is strongly motivated by recent advances in experimental and modeling studies **carried out** by the authors. It is shown that modal response of the structure could be modified in the presence of cavitation [10]. This modification can be attributed to the presence in the flow of a non stationary liquid-vapor mixture with **a strong variation in density** at the fluid structure interface. Previous works proposed the decomposition of the fluid variables into two components: the first component is related

34 to the fluid flow around the non-vibrating structure while the second one describes  
 35 the fluid flow induced by the structure vibrations [11]. This approach can be used to  
 36 compute the added mass and the added damping operators for complex geometries and  
 37 complex fluid flow behaviour. Here, the fluid flow is characterized by oscillating cavity  
 38 on the fluid-structure interface. Unlike to homogeneous fluid case, it is shown that the  
 39 added mass operator is not symmetrical and depends on the flow through fluid density  
 40 variations at the fluid-structure interface. Also, it is evidenced that variation rate of the  
 41 fluid density induces an added damping operator. This suggest to a possible variation  
 42 of the natural frequency of the structure related to the variation of added mass. It is  
 43 reported in [11] that the fluid density variations on the fluid-structure interface have  
 44 an effect on the added mass operator and the variation rate of this density induces an  
 45 added damping operator.

46 The aim of this paper is to study the effect of these variations on the structure  
 47 dynamics. First, the modeling of the structure dynamics is carried out. A rigid section of  
 48 a hydrofoil immersed in a 2D fluid flow and supported by a linear spring, is considered.  
 49 Equations of the hydrofoil motion are thus provided. **Second, a model of the fluid**  
 50 **flow generated by the displacements of the structure is considered to determine the**  
 51 **hydrodynamic loads.** This is given by the solution of a Laplace equation, with the space  
 52 variations of the fluid density taken into account. Applied to the structure dynamics, the  
 53 hydrodynamic loads act as an added mass and an added damping. A simplified model  
 54 of an unsteady cavity, based on a sheet cavitation oscillation, is used to take into account  
 55 the time and space variations of the fluid density on the fluid-structure interface. The  
 56 empirical mode decomposition (EMD) [12] is used to analyze the structure displacement.  
 57 The displacement signal is decomposed by EMD into intrinsic mode functions (IMFs),  
 58 followed by the instantaneous frequencies estimation of these sifted IMFs that evidence  
 59 the frequencies modulations.

## 60 2. Fluid loads acting on the immersed structure

61 A 2D rigid section of hydrofoil type NACA0012 ( $\Omega_S$ ), immersed in a 2D fluid flow  
 62 ( $\Omega_F$ ) and animated by a heave motion, is considered (Figure 1). The fluid domain bound-  
 63 aries are **respectively** the flow inlet  $\Gamma_I$ , the flow outlet  $\Gamma_O$ , the fixed boundary (wall)  
 64  $\Gamma_W$  and the fluid-structure interface (moving boundary)  $\Gamma_{FS} = \Omega_F \cap \Omega_S$ .  $\mathbf{n}$  denotes the  
 65 outward normal unit vector at  $\partial\Omega_F = \Gamma_I \cup \Gamma_O \cup \Gamma_W \cup \Gamma_{FS}$ .  $\vec{U}_\infty$  is the uniform velocity  
 66 field of the fluid upstream of the hydrofoil. Parameter  $\theta$  corresponds to the angle of  
 67 incidence of the hydrofoil.

68

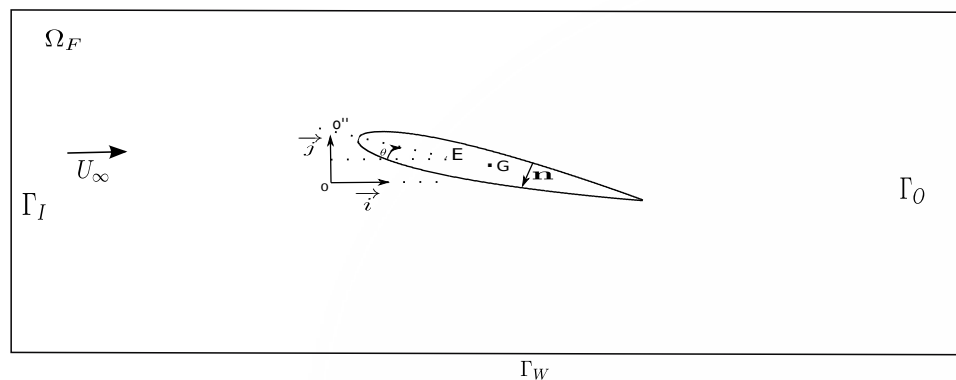


Figure 1. Fluid and Structure domains

In this study, a non-homogeneous inviscid fluid flow is considered. The corresponding conservation equations are given by:

$$\left\{ \begin{array}{ll} \frac{\partial \rho}{\partial t} + \nabla \cdot (\rho u) = 0 & \text{on } \Omega_F \quad (a) \\ \frac{\partial(\rho u)}{\partial t} + \nabla \cdot (\rho u \otimes u) = -\nabla p & \text{on } \Gamma_{FS} \quad (b) \end{array} \right. \quad (1)$$

where  $u$ ,  $p$  and  $\rho$  are respectively the time and space-dependent fluid velocity, fluid pressure and two-phase fluid density. Density  $\rho$  can change from liquid,  $\rho_L$ , to vapor  $\rho_V$  (or vice versa). The boundary conditions are given by:

$$\left\{ \begin{array}{ll} u = u_\infty & \text{on } \Gamma_I \quad (c) \\ -p \mathbf{n} = 0 & \text{on } \Gamma_O \quad (d) \\ u \cdot \mathbf{n} = 0 & \text{on } \Gamma_W \quad (e) \\ u \cdot \mathbf{n} = \dot{\xi}_y n_y & \text{on } \Gamma_{FS} \quad (f) \end{array} \right. \quad (2)$$

69 where  $n_y$  and  $\dot{\xi}_y$  are respectively the  $\mathbf{j}$  components of the normal unit vector  $\mathbf{n}$  and the  
70 velocity  $\dot{\xi}$  of a point  $A(x,y)$  on the interface  $\Gamma_{FS}$  given by Equation (12).

71

Let us assume the following decomposition:

$$u = \bar{u} + u', \text{ and } p = \bar{p} + p' \quad (3)$$

Assuming that  $u'$  is small and uncorrelated to  $\bar{u}$ , the problem described by the system of Equations (1) can be subdivided into two separate problems [11]: the first problem is related to the fluid flow equation around a non-vibrating structure and the second one is about the fluid flow equation induced by the structure vibrations. The later is described by the following system:

$$\left\{ \begin{array}{ll} \Delta p' = 0 & \text{on } \Omega_f \quad (a) \\ \nabla p' \cdot \mathbf{n} = -\rho \ddot{\xi}_y n_y - \frac{\partial \rho}{\partial t} \dot{\xi}_y n_y & \text{on } \Gamma_{FS} \quad (b) \\ p' = 0 & \text{on } \partial\Omega_f \setminus \Gamma_{FS} \quad (c) \end{array} \right. \quad (4)$$

where  $\ddot{\xi}_y$  is the  $\mathbf{j}$  component of the acceleration at the point  $A(x,y)$  on the interface  $\Gamma_{FS}$ . It is given by Equation (12). This formulation is used for cambered hydrofoil and for other geometries [13]. Equations system (4) is coupled to the structure dynamic's equation through the boundary condition (4b), defined on the fluid-structure interface  $\Gamma_{FS}$ . It follows that the structure loading due to the pressure field  $p'$  is given by:

$$F(t) = \int_{\Gamma_{FS}} p' \mathbf{n} ds \quad (5)$$

72 In this paper, we are particularly interested in the effect of  $F(t)$  on the dynamic of the  
73 hydrofoil. Therefore, the main goal is to perform the coupling of the Equation (4) and  
74 the structure dynamics equation (Equation (12)).

### 75 2.1. Added mass and added damping

Due to the linearity of Equation (4), superposition principle holds and the solution can be expressed as  $p' = p_1 + p_2$ , where  $p_1$  and  $p_2$  are respectively the solutions of the following systems:

$$\begin{cases} \Delta p_1 = 0 & \text{on } \Omega_F & (a) \\ \nabla p_1 \cdot \mathbf{n} = -\rho \ddot{\xi}_y n_y & \text{on } \Gamma_{FS} & (b) \\ p_1 = 0 & \text{on } \partial\Omega_F \setminus \Gamma_{FS} & (c) \end{cases} \quad (6)$$

and

$$\begin{cases} \Delta p_2 = 0 & \text{on } \Omega_F & (a) \\ \nabla p_2 \cdot \mathbf{n} = -\frac{\partial \rho}{\partial t} \dot{\xi}_y n_y & \text{on } \Gamma_{FS} & (b) \\ p_2 = 0 & \text{on } \partial\Omega_F \setminus \Gamma_{FS} & (c) \end{cases} \quad (7)$$

76 Solution  $p_1$  of Equation (6) represents the inertial effect of the fluid on the structure as  
 77 it is proportional to the acceleration  $\ddot{\xi}_y$  of the structure. Solution of Equation (7) shows  
 78 that the fluid density rate of change induces a fluid load acting as an added damping  
 79 on the structure, as it is proportional to the velocity  $\dot{\xi}_y$  of the latter (cf. Equation (17)).  
 80 Equation (6-b) and Equation (7-b) show that both solutions  $p_1$  and  $p_2$  depend on space  
 81 and time variations of the fluid density throughout the fluid-structure interface. It is  
 82 easy to see that solution  $p_2$  is zero for the homogeneous case.

In the other hand, it can be shown that the fluid load  $F_a(t) = \begin{pmatrix} F_{a1} \\ F_{a2} \end{pmatrix}$ , defined by the integral

$$F_a(t) = \int_{\Gamma_{FS}} p_1 \mathbf{n} ds, \quad (8)$$

is proportional to the structure acceleration. It can be expressed as

$$\begin{aligned} F_{a1}(t) &= -m_a^{11} \ddot{X}_1 - m_a^{12} \ddot{X}_2 \\ F_{a2}(t) &= -m_a^{21} \ddot{X}_1 - m_a^{22} \ddot{X}_2 \end{aligned} \quad (9)$$

where, for  $A \in \Gamma_{FS}$ ,  $\ddot{X}_1(t) = \ddot{\xi}_x(A, t)$  and  $\ddot{X}_2(t) = \ddot{\xi}_y(A, t)$  are the accelerations according to the 2d-coordinates axis and  $(m_a^{ij})_{i,j=1,2}$  are the added mass coefficients. The matrix  $\mathbf{M}_a$  such that

$$F_a(t) = -\mathbf{M}_a \ddot{\mathbf{X}} = (m_a^{ij} \ddot{X}_j)_{i=1,2} \quad (10)$$

83 is the added mass matrix.

84 By following the same analysis as before, we can define the added damping operator  
 85 (induced by the fluid density rate of change) from Equation (7). The fluid load  $F_d(t) =$   
 86  $\begin{pmatrix} F_{d1} \\ F_{d2} \end{pmatrix}$  is proportional to the velocity of  $\Gamma_{FS}$ .

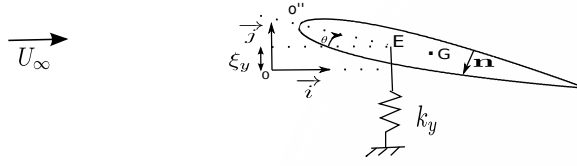
87 The same approach used for Equations ((8) and (9)) leads to the added damping matrix  
 88  $\mathbf{D}_a$ , given by the following relation:

$$F_d(t) = -\mathbf{D}_a \dot{\mathbf{X}} = (d_a^{ij} \dot{X}_j)_{i=1,2} \quad (11)$$

89 where  $\dot{X}_1(t) = \dot{\xi}_x(A, t)$  and  $\dot{X}_2(t) = \dot{\xi}_y(A, t)$  are the velocity and  $(d_a^{ij})_{i,j=1,2}$  are the  
 90 added damping coefficients (induced by the fluid density rate of change).

### 91 3. Structure dynamics modeling

Hydrofoil motion can be defined by its interface displacement  $\zeta = \zeta(t)$  where  $\zeta = \zeta_x \mathbf{i} + \zeta_y \mathbf{j}$ . A linear spring with mass  $m$  and stiffness  $k_y$  is applied in order to model the heave motion  $\zeta_y$  of the hydrofoil in  $\mathbf{j}$  direction (Figure 2). The angle of attack  $\theta$  is assumed to be fixed at 8 degrees. The dynamic of the hydrofoil in heave motion is



**Figure 2.** Modeling of the hydrofoil in heave motion with a spring mass system.

governed by the following equation:

$$m \ddot{\zeta}_y + k_y \zeta_y = F_y \quad (12)$$

92 with the following initial conditions:

$$\begin{cases} \ddot{\zeta}(0) = \ddot{\zeta}_0 & (a) \\ \zeta(0) = \zeta_0 & (b) \end{cases} \quad (13)$$

where  $F_y$  is the **second** component of the force vector  $F$  (Equation (5)), induced by the fluid flow around the vibrating hydrofoil. Moreover, due to linearity, **the** solution of Equations (6) and (7) can be expressed as  $p_1 = -\ddot{\zeta}_y p'_1$  and  $p_2 = -\dot{\zeta}_y p'_2$ , where  $p'_1$  and  $p'_2$  are respectively the solutions of the following systems:

$$\begin{cases} \Delta p'_1 = 0 & \text{on } \Omega_F & (a) \\ \nabla p'_1 \cdot \mathbf{n} = \rho n_y & \text{on } \Gamma_{FS} & (b) \\ p'_1 = 0 & \text{on } \partial\Omega_F \setminus \Gamma_{FS} & (c) \end{cases} \quad (14)$$

and

$$\begin{cases} \Delta p'_2 = 0 & \text{on } \Omega_F & (a) \\ \nabla p'_2 \cdot \mathbf{n} = \frac{\partial \rho}{\partial t} n_y & \text{on } \Gamma_{FS} & (b) \\ p'_2 = 0 & \text{on } \partial\Omega_F \setminus \Gamma_{FS} & (c) \end{cases} \quad (15)$$

It follows that the Lift force  $F_y$  is given by

$$F_y = -\ddot{\zeta}_y \int_{\Gamma_{FS}} p'_1 n_y ds - \dot{\zeta}_y \int_{\Gamma_{FS}} p'_2 n_y ds \quad (16)$$

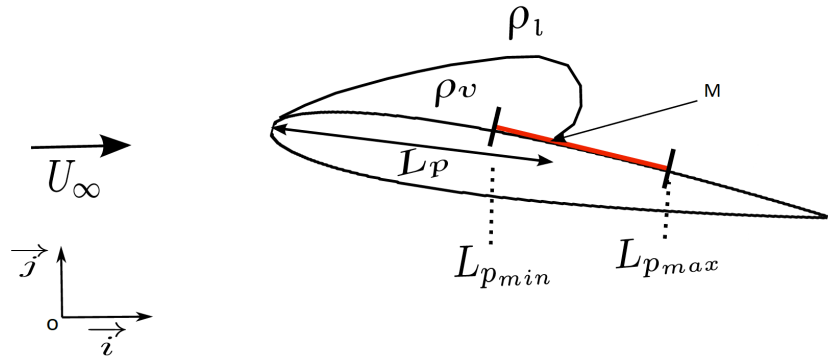
and Equation (12) can be rewritten as:

$$(m + m_a) \ddot{\zeta}_y + d_a \dot{\zeta}_y + k_y \zeta_y = 0 \quad (17)$$

where  $m_a$  and  $d_a$  are respectively the added mass and added damping,

$$m_a = \int_{\Gamma_{FS}} p'_1 n_y ds \quad \text{and} \quad d_a = \int_{\Gamma_{FS}} p'_2 n_y ds$$

93 The resolution of the coupled problem can be summarized by the resolution of Equations  
 94 (14),(15) and (17). On the one hand, Equations (14) and (15) give the added mass and  
 95 added damping (fluid load on the structure). On the other hand, Equation (17) provides  
 96 the structure dynamics (structure displacement, velocity and acceleration). Note that,  
 97 for fixed angle of attack ( $8^\circ$  for our case),  $n_y$  has a fixed value and the variation of the  
 98 solutions  $p'_1$  and  $p'_2$  depend only on the density  $\rho$  and its variation rate ( $\frac{\partial \rho}{\partial t}$ ).



**Figure 3.** Modeling of sheet cavitation.  $M$  oscillation belongs to  $[L_{p_{min}}, L_{p_{max}}]$ .

#### 99 4. Non-homogeneous density model

Modeling of the density variation is made in order to approximate the sheet cavitation behavior on the hydrofoil, as described in the literature [10,14,15] and shown in Figure 17. Sheet cavitation is characterized by unsteady behavior of cavity length  $L_p$  at the hydrofoil upper surface (Figure 3). Attached at the leading edge, the cavity extends on the upper surface and oscillates between the minimum length cavity ( $L_{p_{min}}$ ) to the maximum length ( $L_{p_{max}}$ ). Inside the cavity, the vapor density  $\rho_v$  is equal to  $1\text{kg}/\text{m}^3$ . Outside the cavity, the hydrofoil is surrounded by liquid (water) with the density  $\rho_l$  equal to  $1000\text{Kg}/\text{m}^3$ . At the interface  $\Gamma_{FS}$ , the density  $\rho$  is given by

$$\rho(x, y, t) = \begin{cases} \rho_v = 1 \text{ kg}/\text{m}^3 & \text{if } 0 \leq x < L_p(t) & (a) \\ \rho_l = 1000 \text{ kg}/\text{m}^3 & \text{if } L_p(t) \leq x < L_{p_{max}} & (b) \end{cases} \quad (18)$$

and the variation rate of the density is given by

$$\begin{cases} \frac{\partial \rho}{\partial t} = (\rho_v - \rho_l) \delta(x - L_p(t)) \frac{\partial L_p(t)}{\partial t}, \\ 0 \leq x \leq c, \quad (x, y) \in \Gamma_{FS} \quad \text{and } L_{p_{min}} \leq L_p(t) \leq L_{p_{max}} \end{cases} \quad (19)$$

100 where  $\delta$  is a Dirac function.

101 There are different development phases of sheet cavitation. Firstly, the closing point  $M$   
 102 (Figure 3) has a small variation and the cavity could be considered as a steady. Secondly,  
 103 the cavity length  $L_p$  increases and the closing point oscillates between  $L_{p_{min}}$  and  $L_{p_{max}}$ .  
 104 The cavitation development phases may continue to the destabilization of the cavity,  
 105 followed by a vapor cloud detachment [16,17]. In this paper we focus on the first two  
 106 phases. During the second phase, the cavity length follows a periodic variation [14,18].  
 Let us consider the following simplified model of unsteady cavity

$$L_p(t) = L_{p_{min}} + \frac{L_{p_{max}} - L_{p_{min}}}{2} (1 - \cos(2\pi f_c t)) \quad (20)$$

where  $f_c$  is the oscillation frequency of the closing point  $M$ . The variation rate of cavity length is given by

$$\frac{dL_p}{dt} = (L_{p_{max}} - L_{p_{min}}) \pi f_c \sin(2\pi f_c t) \quad (21)$$

#### 107 5. Numerical resolution

108 A Stainless Hydrofoil (NACA0012) with mass  $m$  equal to  $14.505\text{Kg}\cdot\text{m}^{-1}$  is consid-  
 109 ered. Its natural frequency  $f_N$  in the air is  $58.52\text{Hz}$  and the chord length  $c$  is equal to  
 110  $0.15\text{m}$ . The stiffness is deduced from the previous values.

Newmark scheme [19,20] presented in Equation (22) is used to discretize the structure dynamics Equation (17). The latter is given by :

$$\begin{cases} \zeta_y^n = \zeta_y^{n-1} + \Delta t \dot{\zeta}_y^{n-1} + \frac{\Delta t^2}{4} (\ddot{\zeta}_y^{n-1} + \ddot{\zeta}_y^n) & (a) \\ \dot{\zeta}_y^n = \dot{\zeta}_y^{n-1} + \frac{\Delta t}{2} (\ddot{\zeta}_y^{n-1} + \ddot{\zeta}_y^n) & (b) \end{cases} \quad (22)$$

111 where  $\Delta t$  is the time step and  $\zeta_y^n$  is the value of the displacement  $\zeta_y$  at time  $t_n = n\Delta t$ .  
 112 In this study, the time step is taken equal to  $10^{-3}s$ , which is a good time sampling of  
 113 both the hydrofoil harmonic displacements (with a period of about  $17 \cdot 10^{-3}s$ ) and the  
 114 harmonic variations of the cavity (with a period of about  $44 \cdot 10^{-3}s$ ).

115 The problem (fluid and structure) is solved by using the finite elements code CASTEM  
 116 [21]. Triangular quadratic elements are used. The computational domain is subdivided  
 117 to 34360 elements, which is corresponding to 131720 nodes. As shown in Figure 4, the  
 118 mesh of the subdomain around the hydrofoil is refined in order to improve the accuracy  
 119 of the numerical results.

120

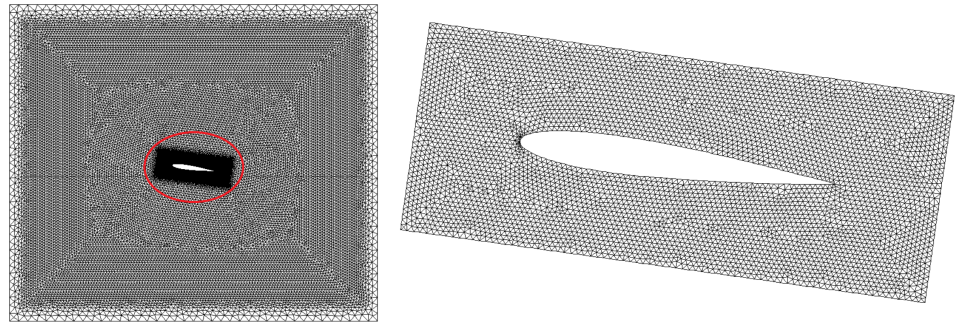


Figure 4. Computational domain and mesh : 131720 elements and 34360 nodes (left). Mesh subdomain around the hydrofoil (right)

121 The same mesh sensitivity study performed in [11] is used here. The Figure 5 show the  
 122 mesh dependence of the numerical added mass value obtained by solving the equation  
 123 14. The chosen mesh corresponds to a relative error of about 1.56%, compared to the  
 124 analytical value of the added mass obtained for a rectangle of the same dimensions as  
 125 the used hydrofoil [11].

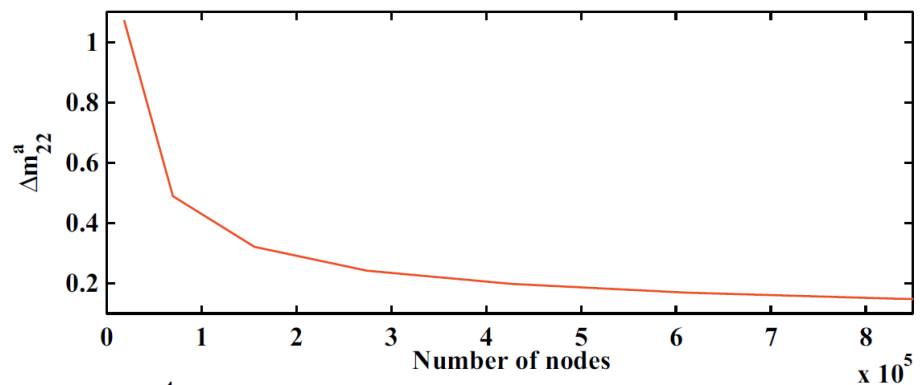


Figure 5. Mesh sensitivity [11]

### 126 5.1. Steady cavity length

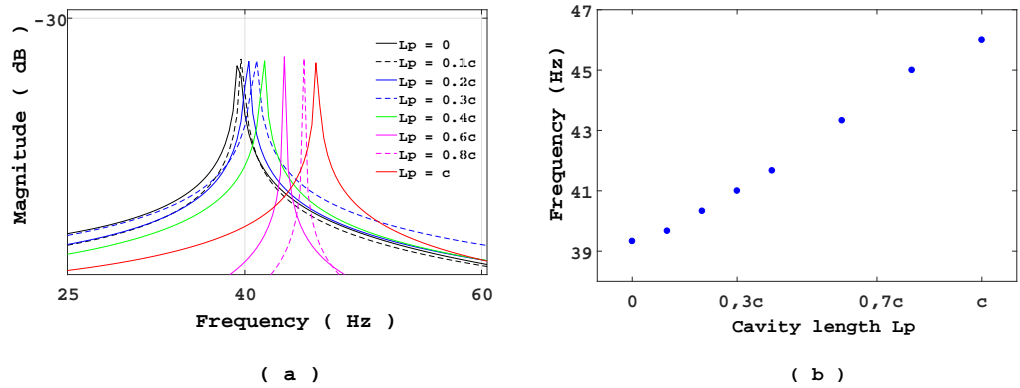
127 Steady cavity length is firstly studied in order to understand the effect of cavity  
 128 length  $L_p$  on the structure dynamics. In this case, the cavity length  $L_p$  is considered as  
 129 constant. Hence, the fluid density is only space dependent and its variation rate is zero.



130 Hence,  $\frac{\partial \rho}{\partial t}$  and  $p'_2$  are equal to zero. It follows that the added mass  $m_a$  is constant and the  
 131 added damping  $d_a$  is zero.

132

133 The simulation is performed for one value of steady  $L_p$  equal to  $0.4c$ . Only Equations (   
 134 14) and (17) are solved for the coupled problem. It follows that the induced movement  
 135  $\zeta_y$  of the hydrofoil is periodic. Thus, it can be defined by the induced frequency  $f_I$  and  
 136 the corresponding amplitude. The induced frequency  $f_I$  of the structure oscillations into  
 the fluid flow can be obtained by Fast Fourier Transform (FFT).



137 **Figure 6.** (a) Spectrum of hydrofoil movement  $\zeta_y$  for different values of  $L_p$ . (b) Frequency of  $\zeta_y$   
 versus  $L_p$

137 The same study was reproduced for different values of  $L_p$  ( $L_p = 0$  to  $L_p = c$ ). The  
 induced frequencies versus cavity length are presented in Figure 6. It can be shown that  
 the value of  $L_p$  does not affect the amplitude of the hydrofoil displacement (Figure 6-a).  
 However, the frequency increases according to the cavity length (Figure 6-b). This is  
 expected because the surface covered by vapor expands as  $L_p$  increases. Furthermore,  
 the added damping  $d_a$  is zero because of the steady cavity length. So the induced  
 frequency  $f_I$  can be deduced from Equation (17) as following :

$$f_I = \frac{1}{2\pi} \sqrt{\frac{k_y}{m + m_a}} \quad (23)$$

The frequency  $f_I$  can be approximated by using the formula (24) given in [22]

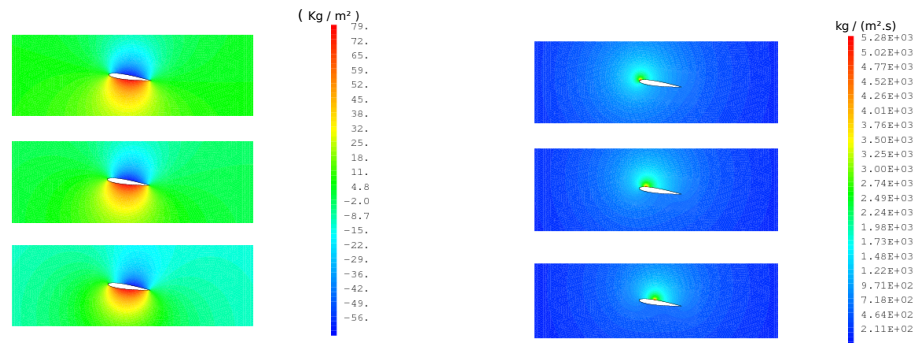
$$\frac{f_I}{f_N} = \frac{1}{\left(1 + \frac{m_a}{m}\right)^{\frac{1}{2}}} \quad (24)$$

### 138 5.2. Unsteady cavity length

139 The simulation of the coupled problem is now performed with unsteady cav-  
 140 ity length. The same values of mass, natural frequency ( $f_N$ ) and chord length used  
 141 previously are applied. Equation (20) is applied for cavity length oscillation; where  
 142  $L_{p_{max}} = 0.4c$ ,  $L_{p_{min}} = 0$  and  $f_c = 22.5\text{Hz}$ . The value of  $f_c$  is chosen to be close to the  
 143 experimental observation [10].

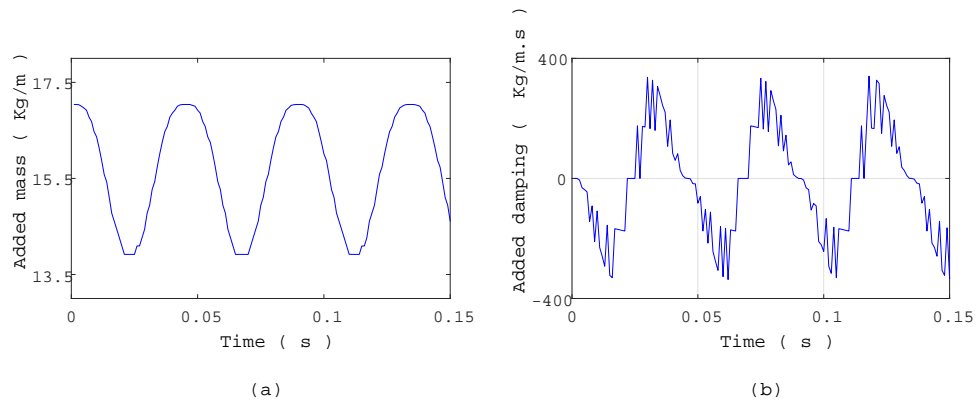
144

145 Solutions  $p'_1$  of Equation (14) in a fluid domain at three different moments, corresponding  
 146 respectively to  $L_p \approx L_{p_{min}}$ ,  $L_p \approx \frac{L_{p_{max}}}{2}$  and  $L_p \approx L_{p_{max}}$ , are shown in figure 7 (left). It is  
 147 easy to see that the values of  $p'_1$  at the upper surface are smaller than those of the lower  
 148 surface. Indeed,  $p'_1$  is proportional to the fluid density and the hydrofoil is surrounded  
 149 by the vapor at the upper surface.



**Figure 7.** Solutions  $p'_1$  (left) and  $p'_2$  (right) for three times corresponding respectively to  $L_p \approx L_{p_{min}}$ ,  $L_p \approx \frac{L_{p_{max}}}{2}$  and  $L_p \approx L_{p_{max}}$ .

150 Values of  $p'_2$  at the same three moments, corresponding to the three values of  $L_p$ , are  
 151 shown in Figure 7 (right). High values match with the closure points  $M$  where the  
 152 density changes. Indeed,  $p'_2$  is proportional to the variations rate of the density as  
 153 shown in Equations (15) and (19), and that formulation includes Dirac function. So,  
 the variations of  $L_p$  can be observed within the solution  $p'_2$ . It follows that the added

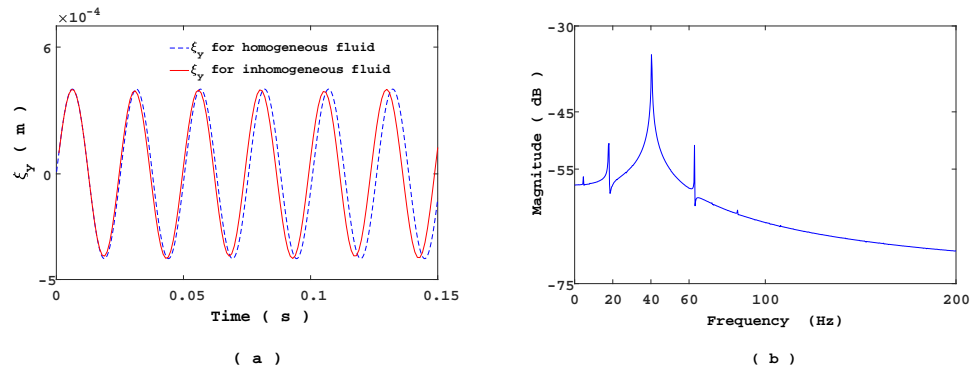


**Figure 8.** (a) Added mass variation versus time. (b) Added damping variation versus time

154 mass  $m_a$  is time dependent. Its variation are shown in Figure 8-a. It oscillates between  
 155 13.92 Kg.m<sup>-1</sup> and 17.04 Kg.m<sup>-1</sup>. These values correspond respectively to values of  $L_{p_{max}}$   
 156 and  $L_{p_{min}}$ . The maximum value corresponds to that obtained in the homogeneous fluid  
 157 case. Hence, it is assumed that the added mass variations is periodic and has the same  
 158 frequency as the cavity length variation. It can be conclude that a frequency modulation  
 159 of the structure is expected in this case.

160 The added damping variation is shown in Figure 8-b. It is periodic with frequency equal  
 161 to  $f_c$  and it can take negative values. It may cause structure instabilities or amplitude  
 162 modulation.  
 163

164  
 165 Hydrofoil motions in both homogeneous and non-homogeneous cases are shown in  
 166 Figure 9-a. The dynamics of the structure are modified by the cavity length oscillation  
 167 and the phase shift between the two motions increases over time. The spectrum analysis  
 168 obtained by FFT shows one fundamental frequency centered between two harmonics  
 169 (Figure 9-b). These harmonics specifically characterize an amplitude modulation. The  
 170 study is reproduced for different values of maximum cavity length ( $L_{p_{max}}$ ). It is shown



**Figure 9.** (a) Comparison of hydrofoil displacements in homogeneous and non-homogeneous cases. (b) Frequency of hydrofoil displacements in non-homogeneous case.

171 that the frequency spectrum are still composed by the fundamental frequency and the  
 172 two harmonics (Table 1). It is noted that the fundamental frequency increases with  
 173 the cavity length. Variation of the added damping from positive to negative sign and  
 174 vice versa is observed. This can induce an amplitude modulation of the hydrofoil  
 175 displacements.

**Table 1.** Frequencies spectrum of the hydrofoil motion for different  $L_{p_{max}}$

$L_{p_{max}}$	1 <sup>st</sup> Harmonic (Hz)	Fundamental (Hz)	2 <sup>nd</sup> Harmonic (Hz)
0.2c	17.54	39.47	61.4
0.6c	19.37	41.16	62.95
0.8c	19.22	41.26	63.61
c	17.47	43.67	69.87

### 176 5.3. Frequency analysis

177 In the previous section, the natural frequency  $f_N$  in air, the cavity length  $L_{p_{max}}$  and  
 178 the cavity length frequency  $f_c$  were fixed to be close to the experimental observation.  
 179 However, in this case the effects of variations in added mass and added damping on the  
 180 hydrofoil dynamics are difficult to highlight. Indeed, for one period of the hydrofoil  
 181 oscillation ( $\xi_y$ ), the cavity length changes from 0 to  $\approx L_{p_{max}}/2.6$  and at the same time  
 182 added mass and added damping vary with the same frequency as the cavity.

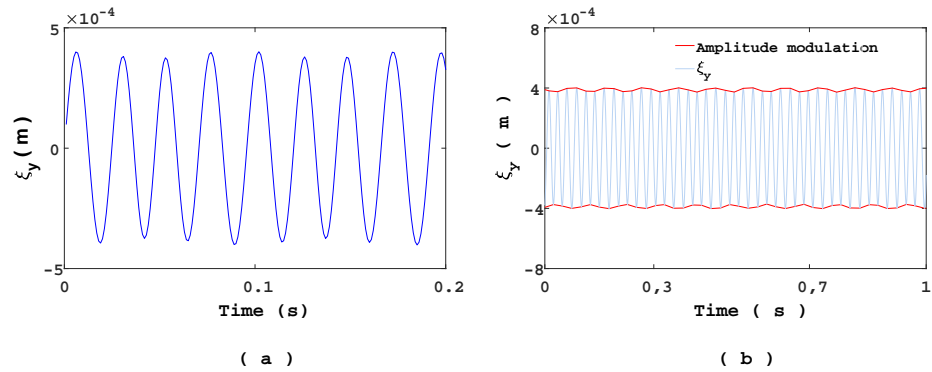
183 So, in order to highlight these effects, a smaller cavity length frequency  $f_c = 11$  Hz and  
 184 a larger cavity length  $L_{p_{max}} = 0.8c$  are used. It consists of increasing the gap between  
 185  $f_N$  and  $f_c$ . The values of  $c$ ,  $m$ ,  $f_N$  and  $L_{p_{min}}$  are the same as in previous section. The  
 186 solution of the coupled problem (Equations (14), (15) and (17)) is shown in Figure 10-a.  
 187 An extended analysis performed over a long period of time is reported in Figure 10-b.  
 188 Upper and lower envelopes of the signal  $\xi_y$  is represented in black curve. This represents  
 189 the amplitude modulation of the structure dynamics.

190  
 191 In order to highlight the expected frequency modulation of the structure dynamics,  
 192 a spectrogram analysis of the signal  $\xi_y$  is performed. However, an accuracy on the  
 193 frequency induces automatically a less clearly observable frequency time variation, as  
 194 shown by Figure 11. This corresponds to the best spectrogram obtained, according to  
 195 the characteristics presented in Table 2. Indeed, a frequency range is observed and it  
 196 oscillates with a frequency close to the cavity length variations one. Thus, the classical  
 197 frequency analysis methods can not take into account the frequency modulation phe-  
 198 nomena. Hence, application of EMD method followed by the Hilbert spectral analysis  
 199

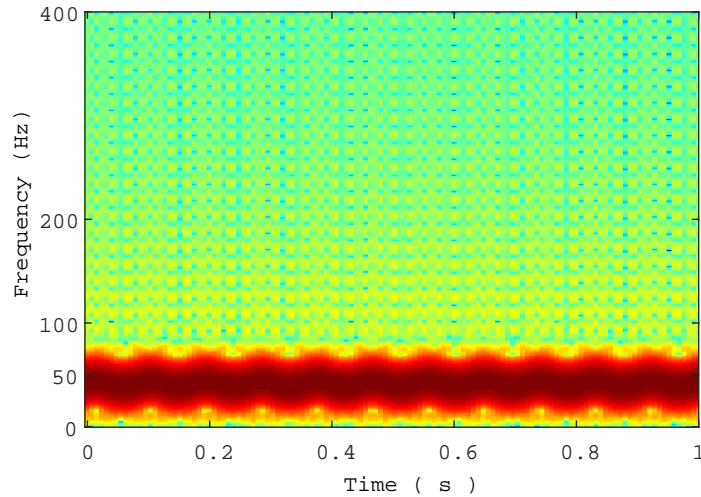
200 are used for the estimation of the instantaneous frequency (IF).  
 201

**Table 2.** Spectrogram parameters.

Block length	Frequency discretization	Time lapse between blocks	Sampling frequency (Hz)
64	1024	8	1000



**Figure 10.** (a) Heave displacement  $\xi_y$  of the hydrofoil,  $t \in [0, 0.2]$ . (b) Heave displacement  $\xi_y$  of the hydrofoil over a long period time,  $t \in [0, 1]$ .



**Figure 11.** Spectrogram of  $\xi_y(t)$  in non-homogeneous case.

### 202 5.3.1. Empirical Mode Decomposition

203 EMD, introduced by Huang et al., is an adaptive and data-driven decomposition well  
 204 suited to decompose non-stationary signals derived or not from linear systems [12].  
 205 More precisely, no *a priori* basis functions are required for the decomposition. The  
 206 algorithm decomposes the multi-component signal into a linear combination of set of  
 207 reduced number of additive oscillatory components termed as IMFs (Intrinsic Mode

208 Functions). Each extracted IMF, a mono-component signal, must satisfy the following  
209 conditions:

- 210 (i) The number of local extrema and the number of zero-crossings must either equal  
211 or differ at most by one.  
212 (ii) The local trend value (mean) of the envelope defined by local maxima and the  
213 envelope defined by the local minima is zero

This requirement ensures that the IMFs have no positive local minima and no negative maxima [12]. Furthermore, these conditions allow us to obtain physically meaningful IF estimates from the extracted IMFs. The core of the EMD is called the sifting process and the resulting adaptive expansion can be seen as a type of wavelet decomposition, whose sub-bands are built up as needed to separate the different components of the signal. To be successfully decomposed into IMFs, a signal  $s(t)$  must have at least two extrema: one minimum and one maximum [12],[23]. At the end of the sifting, the signal  $s(t)$  can be expanded as the sum of mode time series  $\text{IMF}_i(t)$  and a residual  $r_K(t)$ :

$$s(t) = \sum_{i=1}^K \text{IMF}_i(t) + r_K(t) \quad (25)$$

214 where  $K$  is the number of modes determined automatically. Based on a dyadic filter bank  
215 conjecture of the EMD algorithm, the number of sifted modes  $K$  is usually limited to  
216  $K \leq \log_2(L)$ , where  $L$  is the number of samples of the signal  $s(t)$  [24]. The signal  $r_K(t)$ ,  
217 called residual, is a monotonic function that represents the trend within  $s(t)$ .

### 218 5.3.2. Hilbert spectral analysis

219 With the extracted modes  $\text{IMF}_i(t)$ , Hilbert spectral analysis can be applied to each  
220 mode in order to estimate the associated IF  $f_i(t)$ . To compute the IF, the analytic signal  
221 (also called Gabor's complex signal) associated to a real signal  $\text{IMF}_i(t)$  is calculated, as  
222 follows

$$\begin{aligned} z_i(t) &= \text{IMF}_i(t) + j\mathcal{H}[\text{IMF}_i(t)] \\ &= a_i(t)e^{j\phi_i(t)} \end{aligned} \quad (26)$$

where  $a_i(t)$  and  $\phi_i(t)$  are the instantaneous amplitude and phase of  $\text{IMF}_i(t)$ .  $\mathcal{H}[\text{IMF}_i(t)]$  is the Hilbert transform of  $\text{IMF}_i(t)$  and it is given by

$$\mathcal{H}[\text{IMF}_i(t)] = \frac{1}{\pi} \text{PV} \int_{-\infty}^{+\infty} \frac{\text{IMF}_i(\tau)}{t - \tau} d\tau \quad (27)$$

223 where PV is the Cauchy principal value of the integral. Finally, the IF  $f_i(t)$  of  $\text{IMF}_i(t)$  is  
224 calculated as follows [25]:

$$\begin{aligned} f_i(t) &= \frac{1}{2\pi} \frac{d\phi_i(t)}{dt} \\ \phi_i(t) &= \tan^{-1} \left( \frac{\mathcal{H}[\text{IMF}_i(t)]}{\text{IMF}_i(t)} \right) \end{aligned} \quad (28)$$

### 225 5.3.3. IMFs and IFs of the signal $\zeta_y$

EMD is applied to the signal given by the hydrofoil displacement  $\zeta_y$  and ten IMFs are extracted (Figures 12 and 13). Following EMD definition,

$$\zeta_y(t) = \sum_{i=1}^{10} \text{IMF}_i(t) + r_{10}(t) \quad (29)$$

226 Here  $K$  is set to 10. In our case, two classes of IMFs can be defined: the high frequency  
227 class composed by the three first modes and the low frequency composed by the re-

228 maining modes. Note that the first mode,  $IMF_1(t)$ , corresponds to the highest frequency  
 229 component of the signal. In our case, it has the highest amplitude for the high frequency  
 230 class. Zoom of the signal is shown in Figure 14-a. Overall, the hydrofoil movement is  
 231 mainly composed by  $IMF_1(t)$  and the remaining low frequency mode.

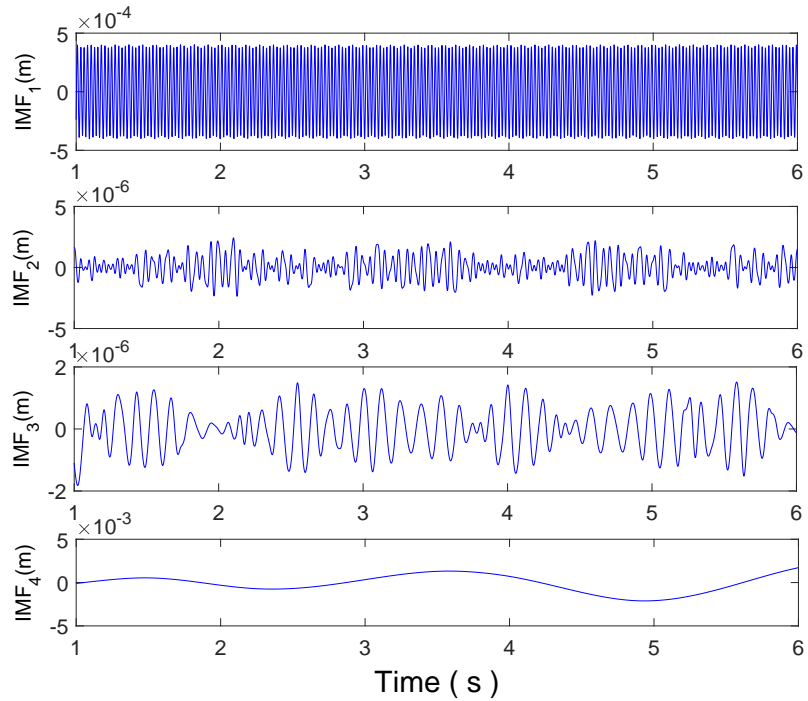


Figure 12. High frequency class:  $IMF_1$ ,  $IMF_2$ ,  $IMF_3$  and low frequency class  $IMF_4$  extracted from  $\xi_y$ .

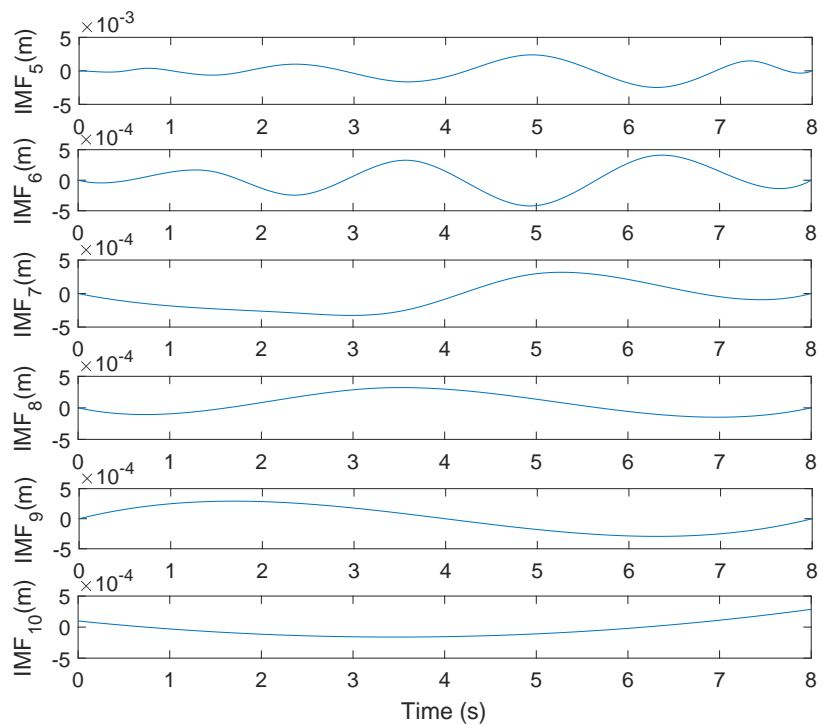


Figure 13. Low frequency class:  $IMF_5$  to  $IMF_{10}$  extracted from  $\xi_y(t)$ .

232 Hilbert spectral analysis of the modes has been performed. The IF  $f_1(t)$  of the first mode  
 233 ( $IMF_1(t)$ ), is shown in Figure 14-b. The frequency modulation is explicitly shown. These  
 234 oscillations are attributed to the variation of the cavity length. Both variations ( $f_1(t)$  and  
 235  $L_p$ ) have the same frequency. The component  $f_1(t)$  oscillates from 39.38 Hz to 44.98 Hz  
 236 except at the beginning of the simulation.

237

238 The IFs of the modes 2 and 3 are shown in Figure 15. They show many peaks (or spikes)  
 239 which are similar to Dirac functions. If the peaks are omitted, complex oscillation of the  
 240 IFs  $f_2(t)$  and  $f_3(t)$  are observed. For the low frequency class, the average of IF variations  
 241 is in the order of  $10^{-2}$  Hz (Figure 16). It can be concluded that the frequency modulations  
 242 of the signal  $\zeta_y$  come from the first three IMFs.

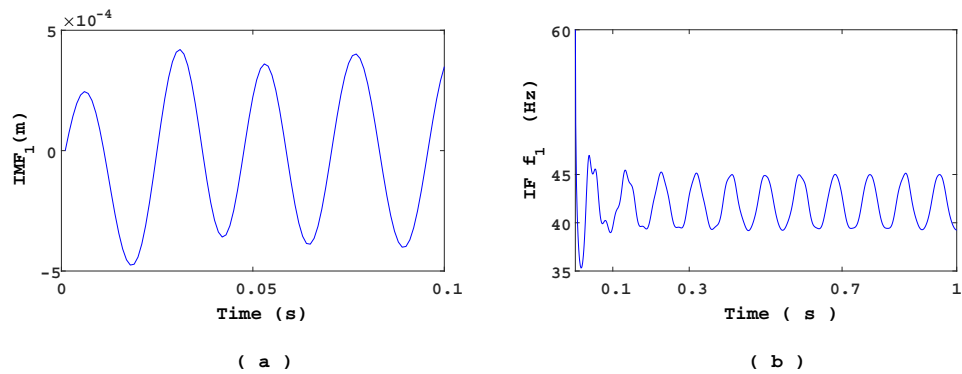


Figure 14. (a)  $IMF_1(t)$  mode. (b) IF  $f_1(t)$ .

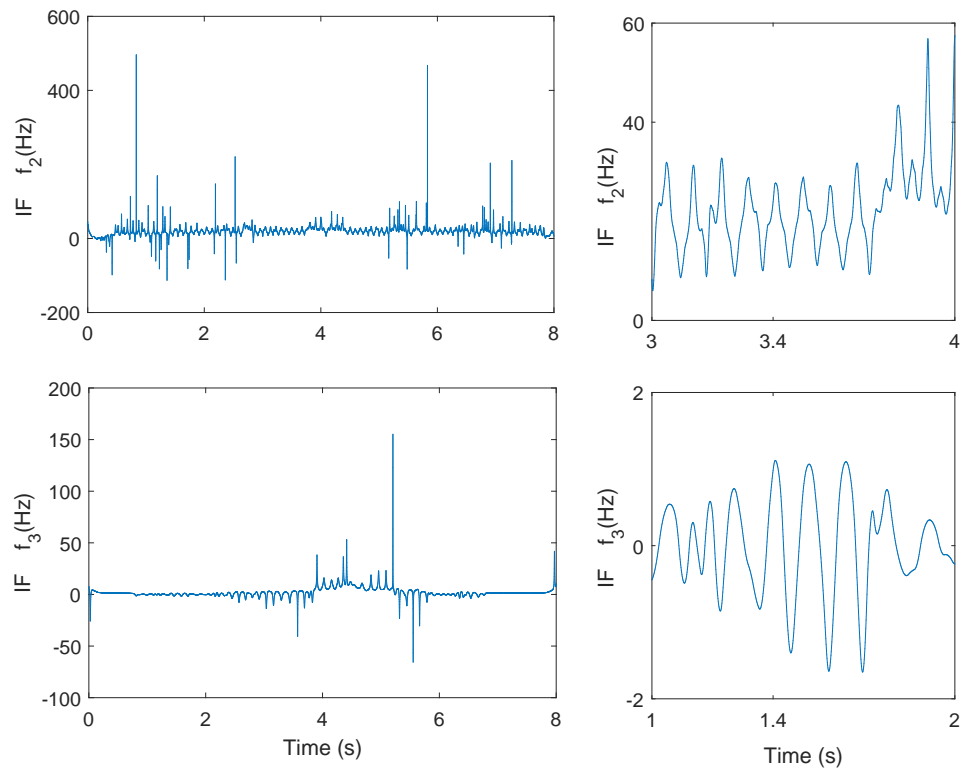
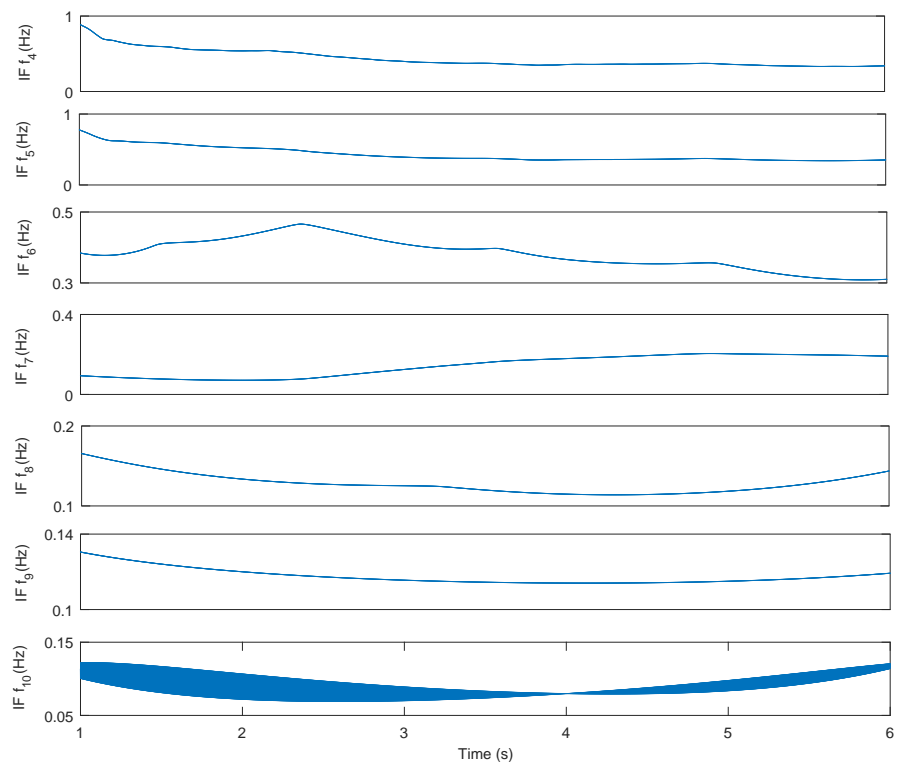


Figure 15. IFs  $f_2(t)$  and  $f_3(t)$  (left), zoom of IFs  $f_2(t)$  and  $f_3(t)$  (right).

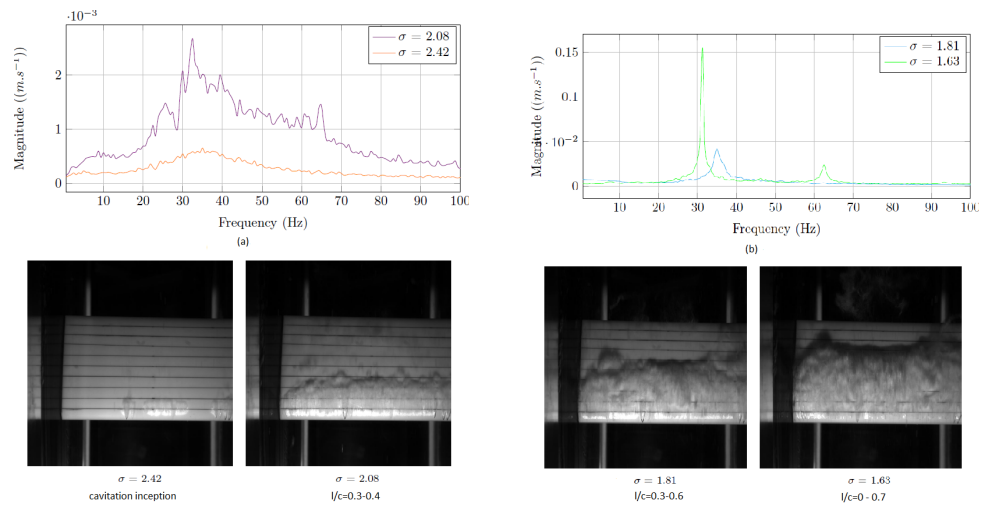


**Figure 16.** IFs from  $f_4(t)$  to  $f_{10}(t)$ .

243 Some experimental evidence can be found in [15] from experiments conducted  
 244 on cavitation induced vibration, performed on a hydrofoil in a hydrodynamic tunnel.  
 245 Typical vibration spectra and the corresponding cavity snapshots on the suction side  
 246 are shown on Figure 17 for various cavity lengths, according to the cavitation number  
 247  $\sigma$  obtained on a hydrofoil. The smaller  $\sigma$ , the larger the maximum cavity length.  $\sigma$   
 248 is defined as  $(P_o - P_v) / (0.5\rho U_\infty^2)$ , where  $P_o$  is the pressure in the test section and  $P_v$  is the  
 249 vapor pressure [4,10,15,26].

250 On Figure 17.a,  $\sigma = 2.42$  corresponds to cavitation inception with small spots of  
 251 vapor attached to the leading edge (bottom of the picture). The corresponding vibration  
 252 spectrum exhibits a rather large peak corresponding to the structural bending mode. For  
 253  $\sigma = 2.08$ , a sheet cavitation was attached at the leading edge and oscillated periodically  
 254 between about 30% and 40% of the chord length. This leads to an increase of the vibration  
 255 level over several peaks ranging from about 25Hz slightly below the bending structural  
 256 mode frequency at 32Hz up to the cavity frequency close to 65Hz. That is the sign of a  
 257 complex response including frequency modulation probably. As the cavitation number  
 258 decreases again (Figure 17.b,  $\sigma = 1.81$ ), the maximum cavity length increases up to 60%  
 259 of the chord length and oscillates at about 35Hz close to the structural frequency. By  
 260 decreasing again the cavitation number ( $\sigma = 1.63$ ), the cavity frequency and the bending  
 261 frequency merge inducing a strong coupling resulting in a very high level resonant peak  
 262 of vibration at the bending/cavity frequency and harmonics.





**Figure 17.** Vibration spectra in cavitating flow and corresponding cavity snapshots on a hydrofoil for various cavitation number ( $\theta = 8$ ,  $U_\infty = 6$  m/s) [15].

## 263 6. Conclusions

264 The effect of the fluid density variations, at the fluid-structure interface, on the  
 265 structure dynamics is studied and analysed. A decomposition method is used to linearize  
 266 the fluid-structure coupled problem, which is separated into two components. The first  
 267 one describes the fluid flow around the fixed hydrofoil while the second one is related  
 268 to the flow induced by the structure vibrations. A model of the fluid density variation  
 269 along the upper interface of the hydrofoil, based on the sheet cavitation behaviour, is  
 270 used. The governing equations are solved numerically using Finite Element Method. In  
 271 this study, the hydrofoil is considered to be animated by a free heave motion. For steady  
 272 cavity length, the added mass remains constant and the added damping (induced by  
 273 the fluid density rate of change) is zero. The study was reproduced for different values  
 274 of cavity length. It was highlighted that the frequency increases according to the cavity  
 275 length. However, the amplitude of the displacement is kept at the same value.

276 For unsteady cavity length, its oscillations along the fluid-structure interface in-  
 277 duces variations in the added mass values. In addition, the fluid density rate of change  
 278 generates a fluid load acting as an added damping on the structure dynamics, which  
 279 can be negative and thus at the origin of instabilities of the structure. Although clas-  
 280 sical methods, such as spectral analysis, make it possible to highlight both amplitude  
 281 modulation (AM) and frequency modulation (FM) phenomena, in structural dynamics  
 282 requires the use of suitable tools to handle such AM-FM signals. Thus, empirical mode  
 283 decomposition (EMD) method, well suited to analyse AM-FM components, was applied  
 284 to the signal obtained from the hydrofoil displacement. Such a decomposition makes it  
 285 possible to obtain the instantaneous frequencies (IFs) of the signal from the extracted  
 286 Intrinsic Mode Functions (IMFs). Therefore, FM is explicitly given through the time  
 287 variations of the frequency, obtained from EMD method. It is shown that the IF derived  
 288 from the first IMF, sifted by EMD decomposition of the hydrofoil displacement signal  $\zeta_y$ ,  
 289 corresponds to the cavity frequency.

290 This signal processing method allows us to highlight the FM phenomenon which  
 291 occurs in the dynamics of a structure immersed in a fluid flow with unsteady non-  
 292 homogeneous density. In this study, only the effects of the added mass and added  
 293 damping (induced by the fluid density rate of change) on the structure dynamics are  
 294 analysed. As future work, we plan to extend this study in order to investigate the  
 295 potential of the EMD method in this case, by analysing the information and the related  
 296 physics, which could be extracted from all the sifted IMFs and the associated IFs.

## 297 7. Author Contributions

298 Conceptualization, T. Emerry RAJAOMAZAVA III and Mustapha BENAOUICHA;  
 299 Formal analysis, T. Emerry RAJAOMAZAVA III and Abdel-Ouahab BOUDRAA; Investi-  
 300 gation, T. Emerry RAJAOMAZAVA III; Methodology, Mustapha BENAOUICHA and  
 301 Abdel-Ouahab BOUDRAA; Project administration, Jacques-André ASTOLFI; Software,  
 302 T. Emerry RAJAOMAZAVA III; Supervision, Mustapha BENAOUICHA and Jacques-  
 303 André ASTOLFI; Validation, Mustapha BENAOUICHA, Jacques-André ASTOLFI and  
 304 Abdel-Ouahab BOUDRAA; Visualization, T. Emerry RAJAOMAZAVA III; Writing –  
 305 review editing, Mustapha BENAOUICHA, Jacques-André ASTOLFI and Abdel-Ouahab  
 306 BOUDRAA.

307

- 308 1. Morand, H.J.P.; Ohayon, R. Fluid structure interaction; John Wiley, 1995.
- 309 2. Axisa, F. Modélisation des systèmes mécaniques: Interactions fluide-structure; Hermès  
310 Science publ., 2001.
- 311 3. Sigrist, J.F. Fluid-structure interaction: an introduction to finite element coupling; John  
312 Wiley & Sons, 2015.
- 313 4. Coutier-Delgosha, O.; Stutz, B.; Vabre, A.; Legoupil, S.; others. Analysis of cavitating flow  
314 structure by experimental and numerical investigations. Journal of Fluid Mechanics **2007**,  
315 578, 171–222.
- 316 5. Ross, M.R.; Felippa, C.A.; Park, K.; Sprague, M.A. Treatment of acoustic fluid–structure  
317 interaction by localized Lagrange multipliers: Formulation. Computer methods in applied  
318 mechanics and engineering **2008**, 197, 3057–3079.
- 319 6. Ross, M.R.; Sprague, M.A.; Felippa, C.A.; Park, K. Treatment of acoustic fluid–structure in-  
320 teraction by localized Lagrange multipliers and comparison to alternative interface-coupling  
321 methods. Computer Methods in Applied Mechanics and Engineering **2009**, 198, 986–1005.
- 322 7. Young, Y. Time-dependent hydroelastic analysis of cavitating propulsors. Journal of Fluids  
323 and Structures **2007**, 23, 269–295.
- 324 8. Young, Y. Fluid–structure interaction analysis of flexible composite marine propellers.  
325 Journal of Fluids and Structures **2008**, 24, 799–818.
- 326 9. Amromin, E.; Kovinskaya, S. Vibration of cavitating elastic wing in a periodically perturbed  
327 flow: excitation of subharmonics. Journal of fluids and structures **2000**, 14, 735–751.
- 328 10. Benaouicha, M.; Astolfi, J.; Ducoin, A.; Frikha, S.; Coutier-Delgosha, O. A numerical study of  
329 cavitation induced vibration. ASME, 2010.
- 330 11. Benaouicha, M.; Astolfi, J.A. Analysis of added mass in cavitating flow. Journal of Fluids  
331 and Structures **2012**, 31, 30–48.
- 332 12. Huang, E.N.; Zheng, S.; Steven, R.; Manli, C.; Hsing, H.; Quanan, Z.; Nai-Chyuan, Y.; Chi, C.;  
333 Henry, H. The empirical mode decomposition and the hilbert spectrum for non-linear and  
334 non-stationary times series analysis. Proceedings of the Royal society **1998**, 454, 903–995.
- 335 13. Benaouicha, M.; Astolfi, J. On Some Aspects of Fluid-Structure Interaction in Two-Phase  
336 Flow. ASME 2013 Pressure Vessels and Piping Conference. July 14-18, 2013, Paris, France,  
337 2013, p. 9.
- 338 14. Leroux, J.B.; Coutier-Delgosha, O.; Astolfi, J.A. A joint experimental and numerical study  
339 of mechanisms associated to instability of partial cavitation on two-dimensional hydrofoil.  
340 Physics of fluids **2005**, 17, 052–101.
- 341 15. Astolfi, J. A. Some Aspects of Experimental Investigations of Fluid Induced Vibration in a Hy-  
342 drodynamic Tunnel for Naval Applications. In Flinovia-Flow Induced Noise and Vibration  
343 Issues and Aspects-III; Ciappi, Ed.; Springer, Feb. 2021. ISBN 978-3-030-64806-0.
- 344 16. Stutz, B.; Reboud, J. Experiments on unsteady cavitation. Experiments in Fluids **1997**,  
345 22, 191–198.
- 346 17. Jousselein, F.; Delannoy, Y.; Sauvage-Boutar, E.; Goirand, B. Experimental investigations on  
347 unsteady attached cavities. Cavitation, ASME-FED-116 **1991**, 91, 61–66.
- 348 18. Frikha, S. Étude numérique et expérimentale des écoulements cavitants sur corps portants.  
349 PhD thesis, Arts et Métiers ParisTech, 2010.
- 350 19. Kane, C.; Marsden, J.; Ortiz, M.; West, M. Variational integrators and the Newmark algorithm  
351 for conservative and dissipative mechanical systems. International Journal for Numerical  
352 Methods in Engineering **2000**, 49, 1295–1325.

- 
- 353 20. Kane, C. Variational integrators and the Newmark algorithm for conservative and dissipative  
354 mechanical systems. PhD thesis, Citeseer, 1999.
- 355 21. Combescure, A.; Hoffmann, A.; Pasquet, P. The CASTEM finite element system. In Finite  
356 Element Systems; Springer, 1982; pp. 115–125.
- 357 22. Blevins, R. Formulas for natural frequency and mode shape; Van Nostrand Reinhold  
358 NewYork, 1979.
- 359 23. Boudraa, A.; Cexus, J. EMD-based signal filtering. IEEE Trans. Instrum. Meas. **2007**,  
360 56, 1597–1611.
- 361 24. Wu, Z.; Huang, N. A Study of the characteristics of white noise using the empirical mode  
362 decomposition method. Proc. R. Soc. Lond. A **2004**, 460, 1597–1611.
- 363 25. Boashash, B.P. Estimating and interpreting the instantaneous frequency of a signal. Part I:  
364 Fundamentals . IEEE **1992**.
- 365 26. Balyts' Kyi, OI and Chmiel, J and Krause, P and Niekrasz, J and Maciag, M. Role of hydrogen  
366 in the cavitation fracture of 45 steel in lubricating media. Materials Science **2009**, 45, 651.



