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# **TOWARDS THE DESIGN OF KRESLING TOWER ORIGAMI AS A COMPLIANT BUILDING BLOCK**

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 *In this paper, the use of the Kresling tower origami as a building block for compliant mechanism design is consid- ered. To help building systems using this origami, models are derived to link the origami pattern geometry to the Kresling tower characteristics. This includes the position of stable configurations, the helical trajectory describing the motion and the orientation of panels during the tower deployment. The provided analytical expressions are helpful to adjust the tower geometry according to desired specification. In addi- tion, an original modification of fold geometry is introduced to modify the tower stiffness. Material removal at specific locations, where maximum fold deformations occur, aims at reducing the actuation force without affecting the kinemat- ics. Experimental evaluation is conducted to assess the rel- evance of the proposed models and evaluate the impact of fold line modification. The proposed simplified models are precise enough for the synthesis. The capacity to strongly reduce the actuation force, thanks to the the fold line modifi-cation, is observed.*

#### <sup>20</sup> 1 Introduction

<sup>21</sup> In the past few years, origami traditional asian art has 22 been increasingly used in several fields of engineering  $\left[\frac{1}{2}\right]$ . <sup>23</sup> Indeed, origami has various interesting mechanical proper- $24$  ties such as deployability  $\boxed{3}$ , advantageous ratio between 25 stiffness and weight  $[4]$ , reconfiguration capabilities  $[5]$ ,  $[6]$ , 26 and for some origami structures, bistability  $[7]$ . This has led <sup>27</sup> to different designs of robotic components and systems based 28 on origami  $\left[\frac{8}{9}\right]$ . Polymer material is then a relevant choice 29 to keep devices lightweight, using for instance POM  $[10]$ , 30 PET  $[8]$  or PP  $[9]$ .

31 Origami structures can be used alone or in combina-32 tion. In [\[11\]](#page-10-10), tubular structures based on origami patterns are assembled to form a metamaterial. Each tube can con- tract along a preferred axis, and has different stiffnesses in other directions. By coupling several tubes in different con- figurations, a compliant structure is obtained that is deploy-able in a given direction and stiff in other directions. In  $\sqrt{2}$ ,

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two origamis with different motions are assembled to gener- 38 ate a perilstatic displacement. Another example is in  $[12]$ ,  $\frac{39}{2}$ where two origamis with different kinematics are combined  $40$ to convert a rotational input motion into a translational output  $41$ motion. For compliant mechanisms, design by association <sup>42</sup> of elementary components is well known with the identifi- <sup>43</sup> cation of so-called building blocks, providing specific kine- <sup>44</sup> matic properties  $[13]$ . This was successfully used in  $[14, 15]$  $[14, 15]$  $[14, 15]$ .  $\overline{\phantom{a}}$ Origamis belong to the general class of compliant mecha- <sup>46</sup> nisms. So, identifying origamis that can be used as build- <sup>47</sup> ing blocks, with tools to help their synthesis, should create <sup>48</sup> opportunities to elaborate more complex architectures to be <sup>49</sup> used in robotics.

The Kresling tower origami  $[16]$  (Fig. [1\)](#page-2-0) is an interesting structure to go in that direction. It is at the same time  $a = 52$ structural element with specific deployment kinematics of its  $53$ panels, and also a mechanical component with the kinemat- <sup>54</sup> ics of an helical joint that can be used as such for mechanism  $55$ design. In  $[17]$ , the origami is used for its deployment capabilities, with the panels used to deploy an antenna. In  $\begin{bmatrix} 8 \\ 1 \end{bmatrix}$  57 the Kresling tower is also used for the motion of its panels,  $\frac{58}{2}$ the inner radius of the tower being used to guide an endo- <sup>59</sup> scope. In  $[12]$ , the tower is used for its kinematics as the 60 helical motion is used to transform a rotational motion into a 61 translation motion to achieve peristaltic locomotion. Moreover, the Kresling tower can possess two stable configura- 63 tions. This bistability is in favor of energy efficiency, as no  $64$ energy is needed to maintain the two stable configurations. It  $\frac{65}{65}$ also brings safety, as in absence of actuation, the mechanism  $66$ remains in a configuration corresponding to one of the two  $67$ stable states.

In this paper, we consider the Kresling tower origami  $\theta$ and bring elements to help the designer building functional  $70$ components from this structure. Two aspects are covered.  $71$ The first one concerns the geometrical synthesis of the  $72$ origami pattern, with the derivation of analytical models describing the deployment of its panels and the overall helical  $\frac{74}{6}$ motion. This description of the structure kinematics has not  $\frac{75}{6}$ been presented in the literature yet. The second one concerns the structural stiffness of the component, and a way to  $77$ modify it through a modification at specific locations, where  $\frac{78}{6}$ 

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<span id="page-2-0"></span>

**FIG. 1**: Prototype of Kresling tower used for assessment

<span id="page-2-2"></span>

**FIG. 2**: Pattern of the Kresling tower. Mountain folds depicted with red lines, valley folds by blue dashed lines. The number of elementary patterns is here chosen as  $n = 8$ 

<sup>79</sup> fold and panels deformations are observed. One new design <sup>80</sup> parameter is introduced accordingly and its impact is anal-81 ysed. By providing a set of relationships to help the design 82 of the Kresling tower and an original crease modification to 83 limit stiffness, this paper will help further use of this origami 84 structure in the design of innovative mechanisms.

 $85$  In section  $\overline{2}$ , the approach we adopt to model the struc-<sup>86</sup> ture and modify the structure stiffness are introduced and jus-87 tified. The design characteristics to derive are also selected. 88 The relationships of interest are then developed in section [3](#page-3-0) <sup>89</sup> as well as the stiffness modification technique we investi-90 gate. Both aspects are assessed experimentally in section  $\overline{4}$ . <sup>91</sup> Finally, discussion on the results and perspectives in terms of 92 origami design are drawn in Section  $\overline{5}$ .

### <span id="page-2-1"></span>93 2 Selection of Methods and Design Characteristics 94 2.1 Desired Characteristics

 The Kresling tower is based on a pattern composed of  $\frac{1}{96}$  triangles (Fig. [2\)](#page-2-2). It was named after B. Kresling, because 97 of her work on buckling effect on paper tubes [\[16\]](#page-11-4), even though similar patterns have been observed earlier in liter-99 ature  $[18, 19]$  $[18, 19]$ . In  $[20]$ , the tower is built to reach the full folded configuration, i.e. with zero height of the tower, as a first stable configuration. A parameter that controls the trian- gles definition is used to adjust the energy needed to switch between two stable configurations. Using the completely folded configuration as one stable state is however restrictive from a design point of view. Interestingly, a parameterization 106 proposed in [\[21\]](#page-11-9) gives the possibility to generate a general- ized Kresling tower with a non-flat stable configuration. In this case, the tower has two stable states, that can be desig-

nated as low and high configurations. The low configuration  $109$ corresponds to the state where the tower is compressed, and 110 the high configuration where the tower is deployed as shown  $111$ on Fig.  $\overline{3}$ . With this work, the height characterizing these 112 states cannot however be tuned. 113

The authors in  $\sqrt{2}$  considered the height of the low stable configuration as an input parameter for the synthesis of 115 generalized Kresling towers. In  $[22]$ , analytical equations,  $116$ based on the geometry of the tower, allow to determine the 117 pattern dimensions according to the desired stable configura-<br>118 tions, characterized by two parameters corresponding to the 119 height of low and high configurations.

To use the Kresling tower as a component in a mecha- <sup>121</sup> nism, link between the pattern geometry and other character-<br>122 istics of the tower kinematics is needed. In  $\sqrt{8}$ , it is shown the 123 value of the inner radius of the structure, later named  $r_i$ , with  $124$ the index *i* used to designate this inner dimension (Fig.  $\overline{3}$ ), 125 is to be evaluated. It is then outlined that the evolution of  $_{126}$ the inner radius while the height *h* of the tower (Fig.  $\boxed{3}$ ) is 127 varying can be an issue in practical applications. Therefore, 128 modelling the variation of the inner radius during the motion 129 is of importance. The rotation angle  $\phi$  between the top and 130 bottom surfaces of the tower is also a key information  $\boxed{7}$ . In 131 our opinion, the angle between the panel fold lines and the 132 tower axis later, designated as  $ρ$ , is also to be estimated as it  $133$ helps describing the panel motions with the outer radius of 134 the structure. In addition, what is needed for design is the <sup>135</sup> variation of these characteristics during the evolution of the 136 tower, from a flat configuration up to a fully extended config-<br>137 uration. To our knowledge, there is no existing model giving  $_{138}$ analytical expressions of these properties for configurations 139 outside the stable states. In the literature  $[12, 8, 20]$  $[12, 8, 20]$  $[12, 8, 20]$ , these 140 characteristics are indeed only computed for stable config- <sup>141</sup> urations. Our approach is then to derive such a model to  $_{142}$ access to these characteristics, with explicit relationships be- <sup>143</sup> tween their values and the parameters that define the origami 144 pattern. As a summary, we wish to extract the expressions of  $_{145}$  $(\phi, r_i, \rho)$  for any tower height *h*. In the following, we will use  $_{146}$ the index 1 (resp. 2) for the low (high) stable configurations  $_{147}$  $(Fig. 3)$  $(Fig. 3)$ .

#### 2.2 Modeling Approach 149

Kresling tower origami belongs to the class of non- <sup>150</sup> rigidly foldable origamis  $[23]$ , which means that the tower 151 cannot be folded without panel deformations. Several 152 works have been introduced to tackle this origami model-<br>153 ing. In  $[24]$ , the difficulties associated with the use of finite 154 element methods for origami are outlined. Therefore simpli-<br>155 fied models built by modeling the structure as an assembly 156 of truss are developed. Such a model is of interest as the <sup>157</sup> truss dimensions and arrangement can be linked to the actual 158 origami pattern. The trusses replace the fold lines and their  $_{159}$ axial deformation is taken into account. The authors intro- <sup>160</sup> duce virtual folds between the vertices of panels to model 161 the bending stiffness of 4-sided panels, as encountered in 162 some origami tubular designs. For the Kresling tower, 3-163 sided panels have to be modeled thus the method presented 164

<span id="page-3-1"></span>

**FIG. 3**: Representation of the Kresling tower characteristics of interest for a tower height *h* (left), and for the low (middle) and high (right) stable configurations

165 in  $\sqrt{24}$  is no more applicable. In  $\sqrt{25}$ ,  $\sqrt{26}$ ,  $\sqrt{27}$ , a modification of the previous model makes it possible to study the Kresling tower with similar method. However, there is no introduc- tion of virtual folds so the bending stiffness of panels cannot be modeled, whereas it is at the origin of the bistability ef- fect. In [\[20\]](#page-11-8), a virtual fold line has been added in the Kres- ling tower pattern to allow rigid foldability. Despite this, the mountain folds of the pattern are deleted at the same time. This seems a rather important simplification, as interactions occurs between the panels in this origami of closed shape. Another limit of the previous works is related to the energy oriented approaches that are used to link the origami defor-177 mation to the energy variation during its motion. Energy as- sessment during origami deployment eases the analysis of 179 bistability, as with other bistable mechanism [\[28\]](#page-11-16). However, there is no derived analytical expression that would link the geometry of the tower pattern with the definition of bistable configurations and the aforementioned properties of interest. Our approach is to develop a rigid-bar model based on these previous studies to get such relationships.

#### <sup>185</sup> 2.3 Stiffness Modulation

186 It was shown in  $[26,24,29]$  $[26,24,29]$  $[26,24,29]$  that the origami stiffness can be linked to the stiffness of panels and fold lines. As a conse- quence, the modulation of an origami stiffness is usually ob- tained by changing the thickness or material of the origami 190 sheet  $[30]$ , or by changing the pattern geometry  $[4]$ . Thick- ness variation remains limited, as it impacts the origami kine- matics. Material modification can be restricted by the manu- facturing techniques and by material requirements due to the application context.

<span id="page-3-2"></span>

**FIG. 4**: 3D view of the tower in deployed stable configuration. Top and bottom base polygons are represented in green. Elementary pattern is highlighted in orange and hatched

Another considered fold line modification is a change 195 of geometry all along the fold length. Changing the ratio <sup>196</sup> between strokes and breaks of a dashed line crease pattern is 197 one instance. In  $\boxed{10}$ , this type of approach is studied on a 198 Kresling tower, with a significant reduction of the stiffness 199 of the origami.

In this paper, we propose to investigate a complementary  $201$ approach to have a strong impact on origami stiffness. It con- <sup>202</sup> sists in removing material at a specific location, where maxi-<br>203 mum deformations are encountered along the fold line during 204 the transition between the stable configurations. There are 205 several manufacturing methods to realize a crease, which are 206 dependent on the material or the manufacturing technique of 207 the origami. The double dashed line in  $[10]$  is one of them. 208 The approach presented in this paper can be advantageously 209 used in combination with any fold line manufacturing tech-<br>210 nique. Our objective is in the end to reduce the actuation 211 force required for the tower deployment while respecting the 212 properties adjusted by design of the origami pattern in a first 213 stage. 214

#### <span id="page-3-0"></span>3 Models for Design 215

The Kresling tower has a kinematic behavior which is  $_{216}$ equivalent to an helical joint when considering the relative 217 motion of the top and bottom surfaces. In order to describe 218 such kinematics, we focus first on the derivation of the angle  $\phi$  as a function of the height *h*. This will be designated 220 as the kinematic model of the Kresling tower. The other 221 characteristics will be derived using the model afterwards. 222 Most of the notations are taken from  $[27, 7]$  $[27, 7]$  and some equations presented could be defined for stable configurations <sup>224</sup> in  $[31, 32, 27, 7]$  $[31, 32, 27, 7]$  $[31, 32, 27, 7]$  $[31, 32, 27, 7]$ . The tower kinematics are dependent on  $225$ the geometry of the pattern depicted in Fig.  $2\frac{1}{2}$ . We therefore 226 start by presenting the set of parameters used to define the 227 pattern.

#### 3.1 Tower Parametrization 229

The Kresling tower is a closed origami of tubular shape 230  $(Fig. \mathbf{A})$ . The elementary pattern, composed of two triangular  $231$ panels, is distributed following an axial repetition around the 232

<span id="page-4-0"></span>

**FIG. 5**: Representation of the elementary pattern on a 3D view of the tower and motion of bar *l* and *b* during deployment.

ass axis  $\Delta$  as shown on Fig.  $\overline{4}$ . The top and bottom contours are base polygons composed of *n* sides, with *n* the number of el-235 ementary patterns (Fig.  $\boxed{2}$ ). As described in  $\boxed{7}$ , we consider that the base polygons are rigid. The vertices  $U, V$  and  $W$  (Fig.  $\overline{5}$ ) that define the triangular panels are constrained to 238 move at the same distance from the axis  $\Delta$ . This is equiva- lent to constrain the vertices to move on the surface of the cylinder of radius  $r$ , which circumscribes the base polygons. Because of that, we can restrict the analysis of the Kresling tower motion to the analysis of a single elementary triangle, using cylindrical coordinates.

244 An elementary pattern is represented in Fig.  $\boxed{2}$ . The <sup>245</sup> parallelogram  $UWVW'$  is composed of two identical trian- $_{246}$  gles *UVW* and *UVW'*. The lengths *a*, *b* and *l*, of respec-<sup>247</sup> tively segments *UW*, *VW* and *UV*, parameterize the ge-<sup>248</sup> ometry of the elementary pattern. The set of parameters  $( a, b, l, n, r)$  then defines the geometry of one Kresling tower. 250 The value of length *a* is linked to *r* and *n* (Fig.  $\overline{6}$ ) with  $a = 2.r.\sin(\pi/n)$ . Thus we define the minimal set of four 252 parameters  $η = (b, l, n, r)$  to describe the tower geometry.

#### <sup>253</sup> 3.2 Derivation of Models

#### <sup>254</sup> 3.2.1 Kinematic Model Determination

 The kinematic model has to relate the angle φ to the tower height *h*. This allows us to describe the evolution of the vertex *V* during the folding, to have then the possibil- ity to reconstruct the panel *UVW* as in Fig. **5** and finally the whole Kresling tower.

 The model derivation is achieved by considering an equivalent rigid-body mechanism. The latter is obtained by replacing the fold lines with bars, connected with spherical joints to the base polygons, also considered as rigid bod- ies. Without loss of generality, the tower we consider is built from the pattern shown in Fig.  $2$ , to get a left-handed helical shape. There are only two configurations where the mecha- nism can be assembled. They correspond to the two configu- rations where the origami tower does not have bending of its <sup>269</sup> panels.

The angle  $\phi$  in Fig. [6](#page-4-1) is defined as the angle between  $\vec{x}$ and the projection of  $\overrightarrow{OV}$ . The bars *l* and *b* have different

<span id="page-4-1"></span>
$$
\frac{1}{\sqrt{\frac{1}{1\sqrt{\frac{1}{\sqrt{\frac{1}{1\sqrt{\frac{1}{1\sqrt{\frac{1}{1\sqrt{\frac{1}{1\sqrt{\frac{1}{1\sqrt{\frac{1}{1\sqrt{\frac{1}{1\sqrt{\frac{1{1\cdot\frac{1}{1\sqrt{1 \cdot \frac{1}{1\sqrt{1 \cdot \frac{1}{\sqrt{\frac{1}{1\sqrt{1 \cdot \frac{1}{1\sqrt{1 \cdot \frac{1}{1\sqrt{1 \cdot \frac{1}{1\sqrt{1 \cdot \frac{1}{1\sqrt{\frac{1}{1\sqrt{1 \cdot \frac{1}{1\sqrt{1 \cdot \frac{1}{1\sqrt{\frac{1}{1\sqrt{\frac{11\cdot\frac{1}{1\cdot \frac{1}{1\cdot \frac
$$

<span id="page-4-3"></span>**FIG. 6**: Projected view of the elementary pattern



**FIG. 7**: Observation of fold line deformations during a tower compression

lengths. They are considered rigid and they rotate respec- 272 tively around *U* and *W* (Fig.  $5$ ). With this model, the two  $275$ bars cannot remain connected in *V* during the origami motion, except for the two stable states. We then define two <sup>275</sup> points  $V_l$  and  $V_b$  that are built from the point V but belonging respectively to the bar *l* and the bar *b*. The points  $V_l$  and 277  $V_b$  are constrained as the other vertices to move at a constant  $278$ distance *r* from the axis  $\Delta$ . This is equivalent in cylindrical 279 coordinates to have a constant radius  $r$ . The Eqn.  $(1)$  links 280 the value of  $\phi$  for the points  $V_l$  and  $V_b$ , respectively  $\phi_l$  and  $\phi_b$  281 to the height *h*: 282

<span id="page-4-2"></span>
$$
\phi_l(h) = 2 \cdot \operatorname{asin}\left(\frac{\sqrt{l^2 - h^2}}{2r}\right) - \frac{\pi}{n}
$$
  

$$
\phi_b(h) = 2 \cdot \operatorname{asin}\left(\frac{\sqrt{b^2 - h^2}}{2r}\right) + \frac{\pi}{n}
$$
 (1)

We have in Eqn. ( $\Box$ ) two expressions of the angle  $\phi$ . 283 Each one corresponds to an hypothesis on the rigidity of <sup>284</sup> the fold line. During the motion between the stable config- <sup>285</sup> urations, there are necessarily deformations in the origami <sup>286</sup> structure. We thus need to make an assumption on the deformation of bars during the deployment, to derive the po-<br>
288 sition of the vertex. In  $[7, 27]$  $[7, 27]$ , the authors faced that same 289 issue and considered the bar  $l$  to be rigid, but without giv ing details. We performed various experiments to observe the behavior of the fold lines, corresponding to the bars *l* and *b* during origami deployment. We could observe, as illus-294 trated in Fig.  $\boxed{7}$  that compression along the fold line leads to panel buckling. On the opposite, tension along a fold leads to panel stretching. The stretching of a panel causes negligi- ble deformations compared to a buckling. Thus, we consider for the model the trajectory of the vertex that causes a fold compression.

<sup>300</sup> With Eqn. [\(1\)](#page-4-2) describing the trajectories of the vertices  $V_i$  illustrated by Fig.  $5$ , we can deduce that the bar *b* is so- $302$  licited in compression in the range of heights  $[h_1h_2]$  when connected to  $V_l$ . Conversely, the bar *l* is solicited in traction when connected to  $V_b$ . As a consequence, the trajectory of  $V_l$  $\frac{1}{205}$  described by  $\phi_l$  is used to model the motion of the tower be-<sup>306</sup> tween the two stable configurations. On the contrary, outside  $307$  the interval  $[h_1h_2]$ , the roles are reversed. The trajectory of  $V_b$  described by  $\phi_b$  is used consequently to model the motion <sup>309</sup> of the tower outside the range defined by the stable configu-<sup>310</sup> rations.

311 Because the bar lengths are constant, our modeling has 312 a given domain of validity. The maximum reachable height <sup>313</sup> corresponds to the length of bars *b*. The value of the min-314 imum reachable height depends on the pattern dimensions. <sup>315</sup> If the bar length *l* is smaller than the origami diameter, the 316 height *h<sub>min</sub>* is equal to zero. Else, the minimum reachable height can be derived geometrically with  $h_{min} = \sqrt{l^2 - 4 \cdot r^2}$ . <sup>318</sup> As a consequence, the kinematic model is:

<span id="page-5-3"></span>
$$
\phi(h) = \begin{cases} \phi_b(h) \text{ if } h \in ]h_{min}; h_1[\cup]h_2; b[\\ \phi_l(h) \text{ if } h \in [h_1; h_2], \end{cases}
$$
 (2)

#### 319 3.2.2 Design Relationships

320 With  $\eta = (b, l, n, r)$  the set of geometrical parameters <sup>321</sup> that define the pattern geometry, it is possible to derive the  $322$  heights  $(h_1, h_2)$  of stable configurations. The stable config-323 uration positions are obtained when  $V_l = V_b$ , which can be  $_{324}$  expressed from the Eqn. [\(1\)](#page-4-2):

<span id="page-5-1"></span>
$$
G(h,\eta) = \phi_l(h) - \phi_b(h)
$$
  
=  $2\left(\operatorname{asin}\left(\frac{\sqrt{l^2 - h^2}}{2r}\right) - \operatorname{asin}\left(\frac{\sqrt{b^2 - h^2}}{2r}\right) - \frac{\pi}{n}\right)$  (3)

<sup>325</sup> The designer can then compute the values of heights *h*<sup>1</sup>  $326$  and  $h<sub>2</sub>$  with

<span id="page-5-2"></span>
$$
\begin{cases}\nG(h_1, \eta) = 0 \\
G(h_2, \eta) = 0\n\end{cases}
$$
\n(4)

There is a maximum height  $H = 2 \cdot r \cdot \sqrt{1 - \sin(\frac{\pi}{n})^2}$  of <sup>328</sup> the tower that cannot be exceeded. This height is linked to  $329$  the radius *r* and the number of sides *n*. Thus the heights  $h_1$  and  $h_2$  must be smaller than  $H$ , to find a feasible pattern 331 geometry.

<span id="page-5-0"></span>dz x y z d

**FIG. 8**: Definition of the lead angle γ: global view with the path of a vertex *V* in red (left) and close-up in the tangent plane (right)

The inner radius  $r_i$  is calculated using the projected view  $332$ along the  $\vec{z}$  axis, then calculating the height of the *UOV* 333 isosceles triangle at *O*. In triangle *UOV*, angle in *O* is  $\phi + \frac{\pi}{n}$ *n* <sup>334</sup> with  $\frac{\pi}{n}$  the angle between  $(\overrightarrow{x}, \overrightarrow{OW})$  and  $(\overrightarrow{x}, \overrightarrow{OU})$ . Using 335 trigonometry, one finally obtains:  $336$ 

<span id="page-5-4"></span>
$$
r_i(h) = \sqrt{r^2 - 4 \cdot r^2 \sin\left(\frac{\phi(h) + \frac{\pi}{n}}{2}\right)}
$$
 (5)

The value of the angle  $\rho$  between the panels and the  $337$ plane  $(\vec{x}, \vec{y})$  is equal to: 338

<span id="page-5-5"></span>
$$
\rho(h) = \operatorname{asin}\left(\frac{h}{l(h)\cdot \sin\left(\frac{l(h)^2 + a^2 - b(h)^2}{2 \cdot a \cdot b(h)}\right)}\right) \tag{6}
$$

 $with$  339

<span id="page-5-6"></span>
$$
\begin{cases}\n l(h) = \sqrt{4 \cdot r^2 \cdot \sin\left(\frac{\phi(h) - \frac{\pi}{n}}{2}\right)} \\
 b(h) = \sqrt{4 \cdot r^2 \cdot \sin\left(\frac{\phi(h) + \frac{\pi}{n}}{2}\right)}\n\end{cases} (7)
$$

One possible use of the Kresling tower is to build a com- <sup>340</sup> pliant helical joint. For such type of joint, the designer can <sup>341</sup> be interested by the value of the so-called lead angle, illus-<br><sub>342</sub> trated in Fig.  $\boxed{8}$  and noted γ, which describes the relationship 343 between the translational and rotational degrees of freedom. 344 The angle  $\gamma(h)$  describes the relationship between the variation of the vertex height *V* and its angular position. For <sup>346</sup> a given height *h*, it can be computed (Fig.  $\vert$ **8**) as the ratio  $\vert$ <sub>347</sub> between the infinitesimal variation *dh* and the displacement 348 *r*·*d*φ in the tangential direction,  $\gamma(h) = \tan(dh/r.d\phi)$ , which 349 becomes becomes

<span id="page-6-2"></span>Table 1: List of tower characteristics with required relationships for their derivation

Characteristic	<b>Notation</b>	Eqn.
Tower height	h	111314)
Tower angle	Ø	(12)
Tower inner radius	$r_i$	(5)
Angle between panel and horizontal	ρ	$\left( 617\right)$
Tower helical lead angle evolution	V	

<span id="page-6-3"></span>

<span id="page-6-4"></span>**FIG. 9**: Example of pattern with *FD*=80%. 20% of fold line represented in black are deleted



**FIG. 10**: Prototype of tower 4 (*FD*=80%)

we will assess the impact of this parameter. To do so, the <sup>383</sup> actuation force needed to switch between the two stable con- <sup>384</sup> figurations can be monitored, as it is related to the structural <sup>385</sup> stiffness. For this evaluation, experimental approach is pre- <sup>386</sup> ferred to numerical simulation, as it will allow us to take into 387 account the material behavior, the possible non linearities in- <sup>388</sup> troduced by the specific material removal, the manufacturing 389 variabilities, among others. 390

#### <span id="page-6-0"></span>4 Fabrication And Experimental Assessment 391

### **4.1 Synthesis And Fabrication of Prototypes** 392

The proposed design approach was applied to build pro-<br>393 totypes with different geometries. Here, the goal is to verify <sup>394</sup> the capacity to adjust the position of stable configurations 395 while modifying the behavior of the equivalent helical joint, 396 through modifications of the lead angle value. The design 397 requirements were chosen according to the manufacturing 398 capabilities and the experimental protocol, that are detailed 399  $below.$  400

For the origami structure, PP (polypropylene) sheets of  $401$ 0.5 mm thickness are used. This material has already been 402 proven for robotic applications for example in  $[9]$ . The PP  $_{403}$ sheets are laser-cut with a Speedy 300 machine from Trotec 404

<span id="page-6-1"></span>
$$
\gamma(h) = \begin{cases} \n\text{atan} \left( -\frac{\sqrt{l^2 - h^2} \cdot \sqrt{1 - \frac{l^2 - h^2}{4r^2}}}{h} \right) & \text{if } h \in [h_1; h_2] \\
\text{atan} \left( -\frac{\sqrt{b^2 - h^2} \cdot \sqrt{1 - \frac{b^2 - h^2}{4r^2}}}{h} \right) & \text{otherwise}\n\end{cases} \tag{8}
$$

 The derived relationships are listed in Table  $\boxed{1}$  with their link to the tower characteristics. These relationships allow the designer to compute the tower characteristics for a given pattern, defined by η. For a tower synthesis, the way to use the relationships will obviously depend on the design re- quirements. As there are 4 independent geometrical parame- ters defining the origami pattern, 4 conditions can be fulfilled simultaneously. As an example, we consider the synthesis of a Kresling tower in section 4 starting from given values of  $\{r, n, h_1, h_2\}$ , and solve the corresponding system of equations to obtain the parameter set  $\eta$ .

#### <sup>362</sup> 3.2.3 Stiffness Modulation

 To modulate the stiffness of the tower, and hence the energy needed for its actuation, we consider a modification of the fold lines. Our proposition is to perform material re- moval at a specific location along the fold line, to act on the one hand on the stiffness of the fold line, and on the other hand to modify the energy needed for the deformation of the panels adjacent to the fold line. The location of fold mod- ification is identified from experimental observations. The 371 deformations are visible during the folding process such as depicted in Fig. **7.** One can see that the deformation of a fold line is not only the expected bending along the line axis. We also note that the deformations are located around the 375 middle of the crease. Consequently, the fold line material re- moval is being operated at the center of the line as depicted in Fig  $\overline{9}$ . A prototype of the tower cut with this pattern is vis-378 ible in Fig. [10.](#page-6-4) We introduce accordingly a parameter called the "fold density", denoted *FD*, to characterize the modifica- tion. It is defined as the ratio between the preserved sections and total length of the fold line. It can vary between  $100\%$ , when no modification is made, and 0%. In the following,

<span id="page-7-0"></span>Table 2: Designation and description of design conditions for the prototypes

Tower	$h_1$ [mm]	$h_2$ [mm]	$\gamma_{\rm max}$ $^{[o]}$
$\theta$	106.3	166.3	31.5
1	41.5	124.7	54.3
$\mathcal{D}_{\mathcal{L}}$	80.0	130.0	39.4
3	106.3	166.3	35.8

<span id="page-7-1"></span>Table 3: Designation and description of parameter set η for the prototypes



<sup>405</sup> company. The pattern dimensions are in particular compati-<sup>406</sup> ble with the laser cutting process and a manual folding.

 $_{407}$  Tower 0 is considered as a reference tower (Table  $\overline{2}$ ). 408 The value of  $h_2$  is equal to the maximum reachable height  $H$ . The height  $h_1$  is set to 106.3 mm to reach 60 mm of stroke between the two stable configurations. Tower 1 is designed to have the stroke between the two stable configurations cen- tered around *H*/2. Tower 2 is built to reach 50 mm of stroke 413 with an offset from  $H/2$ . For towers 0, 1 and 2, the other two conditions for the design were chosen by a direct selec- tion of parameters *r* and *n*. Finally, tower 3 is built with the 416 same values for  $h_1$  and  $h_2$  as for tower 0, but then adjusting the values of  $r$  and  $n$  to maximize the lead angle along the deployment of the tower. The corresponding η sets for these 419 towers are presented in Table [3.](#page-7-1)

420 A CAD exploded view of prototypes is shown in Fig.  $\boxed{13}$ 421 with designation of the components. The tower is composed <sup>422</sup> of the origami structure which is mounted on rigid ends. The <sup>423</sup> connection is obtained by screwing blocking plates on the <sup>424</sup> periphery of the rigid ends.

 As a first step, fold lines are produced without any spe- cific modification. For their manufacturing, 3 processes have been compared, with the shapes depicted in Fig.  $\boxed{11}$ : a dashed line with cutting of minimal width, an intermittent cutting with an oblong profile, and engraving. The minimal cut width is about 0.3mm given the laser spot characteris- tics. Microscopic observations are used to compare the pro- cesses. The engraving process (C on Fig.  $\boxed{11}$ ) leads to fold anisotropy, as the operation is performed on one side only. One engraving step would be needed on each side of the sheet, to alternate the engraving of the mountain and valley

<span id="page-7-2"></span>

**FIG. 11**: Sketchs of fold design: patterns of minimal thickness intermittent cutting (A), oblong profile cutting (B) and continuous engraving (C)

folds and get isotropic folding. This is time consuming, and 436 positioning errors of the vertices may be introduced. Given 437 the dimensions, the oblong profile (B on Fig.  $\boxed{11}$ ) requires 438 a manual separation of the drops, as they stay fused to the 439 polymer sheet after laser machining. This time-consuming <sup>440</sup> phase is avoided by choosing the intermittent cutting, later 441 named dashed line. In addition, the obtained fold lines have 442 an endurance that is comparable to those with the other cutting profiles.  $444$ 

The dashed line pattern is being used with a 1:1 ratio 445 between strokes and breaks, as shown on the cutting pattern 446 in Fig.  $\boxed{12}$ . The dashed lines are positioned to have 5 mm of 447 material around the vertices, to avoid panel deformation in 448 these regions.  $449$ 

Gluing tabs shown on Fig.  $\boxed{12}$  were integrated to the 450 origami pattern for closing the shape. The tab size was min- <sup>451</sup> imized to limit variation of thickness due to material over- <sup>452</sup> lay. The width of the gluing tabs is fixed to  $10 \text{ mm}$  to have  $453$ sufficient contact surface for gluing. Preliminary tests were 454 set up to define the best bonding process. For initial surface 455 treatments, sanding and cleaning, chemical treatment with 456 PPXP, and plasma treatment by Corona effect were com- <sup>457</sup> pared. Three types of adhesives were considered : neoprene, 458 MS Crystal and cyanoaocrylate. To test the bonding, we used 459 specimens which geometry corresponded to the surface of  $\frac{460}{ }$ the tabs used for the tower. The specimens were tested in ten- <sup>461</sup> sion and shear. Panel gluing should not modify strongly the 462 panel stiffness, as it will affect the symmetry of the structure. 463 Bending stiffness was then also assessed. Finally, the most 464 appropriate process is a combination of surface preparation <sup>465</sup> by sanding, cleaning of the gluing area, plasma treatment, <sup>466</sup> and finally use of a neoprene glue.  $467$ 

The rigid ends are obtained by milling of 10 mm-thick PP. Blocking tabs with rectangular holes on the pattern, are added for positioning the tower. The rectangular hole fits in a milled volume on rigid ends. The blocking plates on Fig.  $\frac{13}{471}$ applies pressure along the tower, to maintain it in position during the folding process. 473

#### 4.2 Validation of Kinematic Behavior 474

Before measurements, the towers are submitted to a sequence of 10 foldings and unfoldings using a tensile testing  $476$ machine, followed by a 30-minute resting period. Loading 477 history is then the same for all prototypes. The number of  $478$ cycles is chosen based on our observations of a stabilization 479 of the tower behavior. Tower assessment is achieved after <sup>480</sup> this phase.  $481$ 

<span id="page-8-1"></span>

**FIG. 12**: Cutting pattern for prototypes

<span id="page-8-0"></span>

**FIG. 13**: CAD exploded view of the prototype of tower

<span id="page-8-2"></span>Table 4: Measurements of tower heights (standard deviations indicated in brackets) and relative error when compared to the tower radius. The index *m* stand for measured to differentiate the theoretical value

Tower	$h_{1m}$ [mm] Error		$h_{2m}$ [mm]	Error
$\theta$	105.5(0.4)	$0.8\%$	$165.8(0.1)$ 0.6%	
1	51.7(5.0)	11.3%	$124.6(0.3)$ 0.1%	
2	82.5(0.4)	$2.7\%$	$131.9(0.2)$ 2.1%	
3	112.7(1.2)	6.4%	165.3(0.1)	$1\%$

 The evaluation was conducted in two steps. First, a di- rect measurement of the position of stable configurations was achieved. The values of tower height are measured after plac- ing the tower in both configurations. The tower is placed on a marble and the height is measured with a gauge at the edge of the polygon. The given value is an average based on four measures, at 90 $^{\circ}$  from each other, to avoid error due to slight non-parallelism between the two polygons of the tower.

 The results of the tower height measurements are gath- ered in Table 4. The average difference between the com- puted and measured heights is about 2.9 mm. We note a bet-493 ter accuracy at the positions  $h_2$  than at positions  $h_1$ . Indeed,

the average difference at the position  $h_1$  is about 5.0 mm,  $494$ whereas at  $h_2$ , it is about 0.9 mm. We also note that the  $\frac{495}{495}$ biggest difference is at the position  $h_1$  of tower 1 which has  $496$ the smallest value of  $h_1$  among the 4 prototypes. The mean  $497$ relative error on values of Table  $\frac{4}{1}$  is about 3%. Even though  $498$ refinement of the model might be needed to predict small <sup>499</sup> values of the tower height, the accuracy seems satisfactory, 500 given the simplicity of the model, manufacturing defects and 501 the impacts of PP material non-linearities.

Second, a compression test was performed using a ten- 503 sile testing machine (Zwick/Roell, Z005) as seen on the 504 Fig.  $\boxed{14}$ . The bottom end is linked to the machine with a  $\frac{1}{505}$ cylindrical joint, and the top end is connected to the mov- <sup>506</sup> ing part of the machine, that is linked with prismatic joint  $_{507}$ to the machine frame. Displacement of the top end is ob- <sup>508</sup> tained using the integrated sensor. A camera (Canon EOS 509  $700D$ ) is placed in front of the tower, to measure the rotation  $510$ during the compression test. The camera is used to monitor  $511$ the value provided by protractor, as shown in Fig.  $\boxed{13}$ . The  $_{512}$ measuring tool is integrated to the bottom end of the struc- <sup>513</sup> ture, which rotates during the folding. It is then possible to  $514$ get the time needed to achieve each 1-degree rotation. The 515 translation speed is set by the testing machine, so we are able  $\frac{1}{516}$ to compute the corresponding translation carried out by the 517 tower after an initial registration. Finally, the evolution of  $_{518}$ the tower angle  $\phi$  can be plotted as a function of the relative  $\frac{519}{2}$ displacement between the tower ends.

In order to validate our model, we plot the angle  $\phi$ , according to the height *h* from the Eqn.  $\boxed{2}$  and compare with  $\frac{522}{2}$ the experimental data in black on Fig.  $\overline{15}$ . On this figure,  $\overline{5}$ <sub>23</sub> the yellow curve is the theoretical curve for  $\phi_l(h)$  and the 524 pink one is for  $φ<sub>b</sub>(h)$ .  $φ(h)$  curve is represented in solid line  $525$ from  $\phi_l$ , respectively  $\phi_b$ , according to the definition interval  $\sigma$ <sub>526</sub> presented in Eqn.  $\boxed{2}$ . We note that the distance, between the 527 experimental curve and the theoretical curve of  $\phi(h)$  in solid  $\frac{1}{528}$ line, is always smaller than with the dashed lines. This ob- <sup>529</sup> servation confirms our proposition for the expression of the 530  $\phi$  angle outside the stable configurations.  $531$ 

The mean relative error, computed between the exper-<br>
<sub>532</sub> imental evaluation and the values issued from the proposed 533 models, is in the order of 1.5%, 0.2%, 0.9% and 0.6% respec- <sup>534</sup> tively for towers 0, 1, 2 and 3. These results, below  $2\%$  of  $\sim$  535 error, allow us to validate our model to describe the motion 536 of the tower. <sup>537</sup>

#### **4.3 Validation of Stiffness Modification** 538

For this validation, the same experimental protocol is  $539$ used in terms of initial loading of the prototypes. The <sup>540</sup> compression force is now measured using a force cell <sup>541</sup> (Zwick/Roell Xforce HP 500N), as seen in Fig.  $\boxed{14}$ . The prototypes used to assess the impact of fold density modifica- <sup>543</sup> tion are based on the same geometrical set  $\eta$ :  $b = 174.6$  mm,  $\frac{544}{256}$  $l = 202.2$  mm  $r = 90$  mm and  $n = 8$ . It defines the tower 0, 545 for which  $FD = 100\%$ . Towers 4 and 5 are prototypes produced respectively with  $FD$  set to 80% and 60%.

To assess the impact of the proposed fold modification, <sup>548</sup> tower 6 is produced with an homogeneous modification of  $_{549}$ 

<span id="page-9-0"></span>

**FIG. 14**: Prototype during testing with vision-based measurement of rotation

 the fold line, where the ratio between strokes and breaks dur- ing the laser machining of the fold line has been modified. This tower is characterized by the same amount of material along fold line as the tower 5 with *FD* set at 60 %, but with a homogeneous material distribution along the fold line. This 555 tower is designated by the index  $60 - h$ . Tower 6 is then compared with both tower 0, which also has a homogeneous material distribution along fold line, and tower 5, which has the same amount of material along fold line to highlight the relevance of material removing at a specific location.

 The maximum compression force  $F_{\text{max}}$  is obtained for a compression of about 8 mm as shown in Fig.  $\overline{16}$ . The values for towers 0 and 6 are close (Table  $5$ ). Reducing the effective length of material along fold lines from *FD*=100 % for tower 0 to 60 % for tower 6 causes a reduction of the maximal force by 3.8%. In comparison, the maximal force is lowered by 63 % when using the tower 5, based on the same amount of material along the fold line than the tower 6. The impact of the proposed fold line modification is then significantly higher than the one created by a global modification of fold line pattern.

571 The value of  $F_{\text{max}}$  for tower 4 ( $FD = 80\%$ ) is reduced by  $555\%$  in comparison with the tower 0. That means removing <sup>573</sup> 20 % of the fold line leads to a decrease of the maximal force <sup>574</sup> of more than half. The impact of the *FD* parameter is very <sup>575</sup> significant.

576 Direct measurements of the height for the stable config-577 urations were also performed. All the towers from Table [5](#page-10-13) use the same set of parameters  $\eta$ , thus the theoretical val- ues of  $h_1$  and  $h_2$  are identical and equal to 106.3 mm and 166.3 mm. The average position difference with theoretical values for towers 0 and 6 is about 0.5 mm, and about 3.6 mm for towers 4 and 5. For these towers, the mains differences are related to the values of  $h_1$ , which describes the configu-ration where the tower is compressed at most. The impact

<span id="page-9-1"></span>

**FIG. 15**: Tower angle φ as a function of the tower height *h*. In pink, the theoretical curve for  $\phi_h$ , in yellow, the theoretical curve for  $\phi_l$  and in black, the measured curve. The solid line compose the curve for theoretical  $\phi(h)$  according to the definition interval defined in Eqn. [\(2\)](#page-5-3)

of fold line modification can be seen as acceptable, as it re- <sup>585</sup> mains in the order of 5  $\%$ , even though that will depend on  $\frac{586}{2}$ exact application requirements. This means the synthesis of 587 the tower could be achieved in two phases. First the pattern 588 geometry can be defined to achieve the two stable configura-<br><sub>589</sub> tions. Then the stiffness level can be modified using the fold 590 modification approach. 591

<span id="page-10-12"></span>

**FIG. 16**: Force-displacement curves obtained from compres-sion tests of towers described in Table [5](#page-10-13)

<span id="page-10-13"></span>Table 5: Measured values of  $h_1$ ,  $h_2$  and maximum force for prototypes with modified fold lines

Tower	$FD [\%]$		$F_{\text{max}}$ [N] $h_{1m}$ [mm]	$h_{2m}$ [mm]
$^{(1)}$	100	54.8	105.5	165.8
4	80	24.2	114.8	167.1
5	60	14.9	110.3	167.3
6	60-h	52.7	105.6	166.4

#### <span id="page-10-11"></span><sup>592</sup> 5 Conclusion

 In this paper, we provided tools to the designer to build compliant components based on the Kresling tower for robotic applications. The origami provides specific motion of its panels, with an overall behavior that can be applied as a bistable helical joint. The tower is then of great interest if associated to a building block design approach. Relation- ships have been established between the origami pattern and the characteristics of interest which were defined as the posi- tion of bistable configurations, the inner radius of the struc- ture, and the orientation of the panels. The experimental re- sults on origami, achieved using materials and processes for suitable with robotic applications, have confirmed the valid- ity of the proposed model.Additionally, an original modifi- cation of fold lines at specific locations was introduced to modulate the stiffness of the Kresling tower. Experimental observations confirmed the provided models can be used to select an origami pattern, according to design input param- eters. The impact of our specific fold line modification was investigated experimentally as well. Results are confirming the significant impact on stiffness, while the error on the po- sition of stable configurations is not significantly increased by fold modification. This means it seems possible to cor-rect the origami stiffness independently from its kinematics.

 Two approaches can be envisioned to be capable of tun-<sup>617</sup> ing the tower stiffness. The first one would be purely ex- perimental, using a trial and error process. The other one would be based on a model, that can be established using fi- nite element simulation and identification. Model building and comparison of the two approaches constitute the main perspectives of the work.

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#### <span id="page-10-0"></span> $References$  629

- [1] Greenberg, H., Gong, M., Magleby, S., and Howell, L., 630 2011. "Identifying links between origami and compli- 631 ant mechanisms". *Mechanical Sciences,* 2(2), pp. 217– <sup>632</sup>  $225.$  633
- <span id="page-10-1"></span>[2] Liu, T., Wang, Y., and Lee, K., 2017. "Three- <sup>634</sup> dimensional printable origami twisted tower: Design, 635 fabrication, and robot embodiment". IEEE Robotics 636 *and Automation Letters,* 3(1), pp. 116–123. 637
- <span id="page-10-2"></span>[3] Zirbel, S. A., Lang, R. J., Thomson, M. W., Sigel, 638 D. A., Walkemeyer, P. E., Trease, B. P., Magleby, S. P., 639 and Howell, L. L., 2013. "Accommodating thickness in 640 origami-based deployable arrays". *Journal of Mechan-* <sup>641</sup> *ical Design,* **135**(11). <sup>642</sup>
- <span id="page-10-3"></span>[4] Onal, C. D., Tolley, M. T., Wood, R. J., and Rus, D., 643 2014. "Origami-inspired printed robots". *IEEE/ASME* <sup>644</sup> *transactions on mechatronics*, **20**(5), pp. 2214–2221. 645
- <span id="page-10-4"></span>[5] Belke, C. H., 2020. From modular origami robots to 646 polygon-based modular systems: a new paradigm in re- <sup>647</sup> configurable robotics. Tech. rep., EPFL. 648
- <span id="page-10-5"></span>[6] Berre, J., Geiskopf, F., Rubbert, L., and Renaud, 649 P.,  $2021$ . "Origami-inspired design of a deployable  $650$ wheel". In New Advances in Mechanisms, Mechani- 651 cal Transmissions and Robotics, E.-C. Lovasz, I. Ma- <sup>652</sup> niu, I. Doroftei, M. Ivanescu, and C.-M. Gruescu, eds., 653 Springer International Publishing, pp. 114–126. 654
- <span id="page-10-6"></span>[7] Bhovad, P., and Li, S., 2018. "Using Multi-Stable 655 Origami Mechanism for Peristaltic Gait Generation: <sup>656</sup> A Case Study". In International Design Engineering 657 Technical Conferences and Computers and Information 658 in Engineering Conference, Vol. 5B: 42nd Mechanisms 659 and Robotics Conference. V05BT07A061. 660
- <span id="page-10-7"></span>[8] Sargent, B., Butler, J., Seymour, K., Bailey, D., Jensen, 661 B., Magleby, S., and Howell, L., 2020. "An origami- 662 based medical support system to mitigate flexible shaft  $663$ buckling". *Journal of Mechanisms and Robotics,* 12(4), <sup>664</sup>  $p. 041005.$
- <span id="page-10-8"></span>[9] Wu, S., Ze, Q., Dai, J., Udipi, N., Paulino, G. H., and 666 Zhao, R., 2021. "Stretchable origami robotic arm with  $667$ omnidirectional bending and twisting". *Proceedings of* <sup>668</sup> *the National Academy of Sciences,* **118**(36). 669
- <span id="page-10-9"></span>[10] Hwang, H.-Y., 2021. "Effects of perforated crease line  $670$ design on mechanical behaviors of origami structures". 671 *International Journal of Solids and Structures,* 230- <sup>672</sup> **231**, p. 111158.  $673$
- <span id="page-10-10"></span>[11] Filipov, E. T., Tachi, T., and Paulino, G. H., 2015. 674 "Origami tubes assembled into stiff, yet reconfigurable  $675$ structures and metamaterials". Proceedings of the 676 *National Academy of Sciences,* 112(40), pp. 12321– <sup>677</sup> 12326. 678

- <span id="page-11-0"></span><sup>679</sup> [12] Angatkina, O., Chien, B., Pagano, A., Yan, T., Alleyne,
- <sup>680</sup> A., Tawfick, S., and Wissa, A., 2017. "A metameric <sup>681</sup> crawling robot enabled by origami and smart materi-
- <sup>682</sup> als". In Smart Materials, Adaptive Structures and Intel-<sup>683</sup> ligent Systems, Vol. 58257, American Society of Me-
- <span id="page-11-1"></span><sup>684</sup> chanical Engineers, p. V001T06A008.
- <sup>685</sup> [13] Moon, Y.-M., Trease, B. P., and Kota, S., 2002. "De-<sup>686</sup> sign of large-displacement compliant joints". *Journal* <sup>687</sup> *of Mechanical Design,* 36533, pp. 65–76.
- <span id="page-11-2"></span><sup>688</sup> [14] Rubbert, L., Bitterli, R., Ferrier, N., Fifanski, S., Vardi, <sup>689</sup> I., and Henein, S., 2016. "Isotropic springs based on <sup>690</sup> parallel flexure stages". *Precision Engineering,* 43,
- <span id="page-11-3"></span> $_{691}$  pp. 132–145. <sup>692</sup> [15] Merriam, E. G., 2016. *Stiffness Reduction Strategies*
- <sup>693</sup> *for Additively Manufactured Compliant Mechanisms*. <sup>694</sup> Brigham Young University.
- <span id="page-11-4"></span><sup>695</sup> [16] Kresling, B., 2008. "Natural twist buckling in <sup>696</sup> shells: from the hawkmoth's bellows to the deployable <sup>697</sup> Kresling-pattern and cylindrical Miura-ori". In Pro-<sup>698</sup> ceedings of the 6th International Conference on Com-<sup>699</sup> putation of Shell and Spatial Structures, IASS-IACM <sup>700</sup> 2008, pp. 1–4.
- <span id="page-11-5"></span><sup>701</sup> [17] Jape, S., Garza, M., Ruff, J., Espinal, F., Sessions, D., <sup>702</sup> Huff, G., Lagoudas, D. C., Hernandez, E. A. P., and <sup>703</sup> Hartl, D. J., 2020. "Self-foldable origami reflector an-<sup>704</sup> tenna enabled by shape memory polymer actuation". <sup>705</sup> *Smart Materials and Structures,* 29(11), p. 115011.
- <span id="page-11-6"></span><sup>706</sup> [18] Nojima, T., 2002. "Modelling of folding patterns in flat <sup>707</sup> membranes and cylinders by origami". *JSME Interna-*<sup>708</sup> *tional Journal Series C Mechanical Systems, Machine* <sup>709</sup> *Elements and Manufacturing,* 45(1), pp. 364–370.
- <span id="page-11-7"></span><sup>710</sup> [19] Guest, S. D., and Pellegrino, S., 1994. "The Folding <sup>711</sup> of Triangulated Cylinders, Part I: Geometric Consid-<sup>712</sup> erations". *Journal of Applied Mechanics,* 61(4), 12, <sup>713</sup> pp. 773–777.
- <span id="page-11-8"></span><sup>714</sup> [20] Pagano, A., Yan, T., Chien, B., Wissa, A., and Taw-<sup>715</sup> fick, S., 2017. "A crawling robot driven by multi-<sup>716</sup> stable origami". *Smart Materials and Structures,* 26(9), <sup>717</sup> p. 094007.
- <span id="page-11-9"></span><sup>718</sup> [21] Zhang, Q., Cai, J., Li, M., and Feng, J., 2018. "Bistable <sup>719</sup> behaviour of a deployable cylinder with kresling pat-<sup>720</sup> tern". In Proceedings of the 7th International Meet-<sup>721</sup> ing on Origami in Science, Mathematics and Education <sup>722</sup> (7OSME), Oxford, UK, pp. 4–7.
- <span id="page-11-10"></span><sup>723</sup> [22] Lang, R. J., 2017. *Twists, tilings, and tessellations:* <sup>724</sup> *mathematical methods for geometric origami*. CRC <sup>725</sup> Press.
- <span id="page-11-11"></span><sup>726</sup> [23] Tachi, T., 2010. "Geometric considerations for the <sup>727</sup> design of rigid origami structures". In Proceedings <sup>728</sup> of the International Association for Shell and Spatial <sup>729</sup> Structures (IASS) Symposium, Vol. 12, Elsevier Ltd, <sup>730</sup> pp. 458–460.
- <span id="page-11-12"></span><sup>731</sup> [24] Filipov, E., Liu, K., Tachi, T., Schenk, M., and Paulino, <sup>732</sup> G. H., 2017. "Bar and hinge models for scalable anal-<sup>733</sup> ysis of origami". *International Journal of Solids and* <sup>734</sup> *Structures,* 124, pp. 26–45.
- <span id="page-11-13"></span><sup>735</sup> [25] Zhai, Z., Wang, Y., and Jiang, H., 2018. "Origami-<sup>736</sup> inspired, on-demand deployable and collapsible me-

chanical metamaterials with tunable stiffness". *Pro-* <sup>737</sup> *ceedings of the National Academy of Sciences,* 115(9), <sup>738</sup> pp.  $2032-2037$ .

- <span id="page-11-14"></span>[26] Liu, K., and Paulino, G., 2017. "Nonlinear me- <sup>740</sup> chanics of non-rigid origami: an efficient computa- <sup>741</sup> tional approach". *Proceedings of the Royal Society* <sup>742</sup> *A: Mathematical, Physical and Engineering Sciences,* <sup>743</sup>  $473(2206)$ , p. 20170348.
- <span id="page-11-15"></span>[27] Pagano, A., Leung, B., Chien, B., Yan, T., Wissa, <sup>745</sup> A., and Tawfick, S., 2016. "Multi-Stable Origami <sup>746</sup> Structure for Crawling Locomotion". In Smart Ma- <sup>747</sup> terials, Adaptive Structures and Intelligent Systems, <sup>748</sup> Vol. 2: Modeling, Simulation and Control; Bio- <sup>749</sup> Inspired Smart Materials and Systems; Energy Harvest- <sup>750</sup> ing, p. V002T06A005. 751
- <span id="page-11-16"></span>[28] Qiu, J., Lang, J. H., and Slocum, A. H., 2004. "A 752 curved-beam bistable mechanism". *Journal of micro-* <sup>753</sup> *electromechanical systems,* **13**(2), pp. 137–146.  $\frac{754}{6}$
- <span id="page-11-17"></span>[29] Nayakanti, N., 2016. "Flexigami: folded polygonal 755 unit cells for deployable metamaterials and mecha- <sup>756</sup> nisms". PhD thesis, Massachusetts Institute of Technology. The contract of the co
- <span id="page-11-18"></span>[30] Yellowhorse, A., and Howell, L. L., 2018. "Three 759 approaches for managing stiffness in origami-inspired 760 mechanisms". In International Design Engineering 761 Technical Conferences and Computers and Information 762 in Engineering Conference, Vol. 51814, American So- <sup>763</sup> ciety of Mechanical Engineers, p. V05BT07A056. 764
- <span id="page-11-19"></span>[31] Jianguo, C., Xiaowei, D., Ya, Z., Jian, F., and Yong- <sup>765</sup> ming, T., 2015. "Bistable Behavior of the Cylindrical 766 Origami Structure With Kresling Pattern". *Journal of* <sup>767</sup> *Mechanical Design,* **137**(6), 06. 061406.
- <span id="page-11-20"></span>[32] Butler, J., Morgan, J., Pehrson, N., Tolman, K., Bate- 769 man, T., Magleby, S. P., and Howell, L. L., 2016. 770 "Highly compressible origami bellows for harsh en- <sup>771</sup> vironments". In Proceedings of the ASME 2016 In- <sup>772</sup> ternational Design Engineering Technical Conferences 773 and Computers and Information in Engineering Con- <sup>774</sup> ference, Vol. 5B: 40th Mechanisms and Robotics Con- <sup>775</sup> ference. V05BT07A001. 776