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▶ To cite this version:

Paolo Guardabasso, Stéphanie Lizy-Destrez. Cislunar CubeSats to Measure Radiation in Support for Human Space Exploration. 12th European Cubesat Symposium, Nov 2021, Paris, France. pp.0. hal-03550624

HAL Id: hal-03550624

https://hal.science/hal-03550624

Submitted on 1 Feb 2022

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Cislunar CubeSats to Measure Radiation in Support for Human Space Exploration

Paolo Guardabasso¹, Stephanie Lizy-Destrez¹

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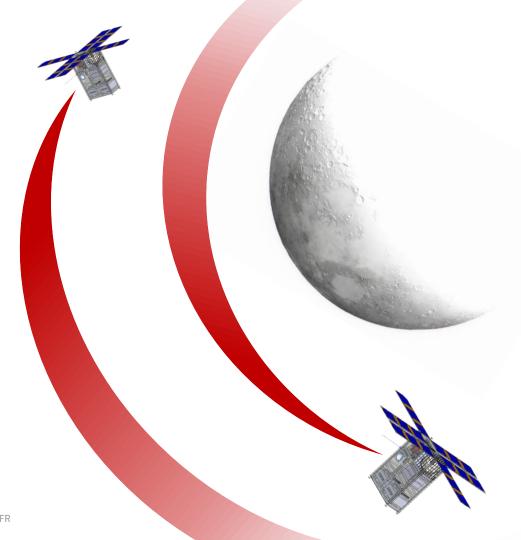
12th European Cubesat Symposium

24-26 November 2021 Ecole polytechnique, Palaiseau, FR



Outline

- Context
- DRACO Mission Analysis
- DRACO System Design
- Conclusion and perspectives



Context

Lunar Exploration

- A new era of exploration begins
- Assembly of a Lunar Gateway in a L2 southern Near Rectilinear Halo Orbit (NRHO).
- Involvement of public and private entities to provide transportation services, logistics and in-situ resources utilisation.



Lunar Gateway Station. Credits: ESA



Blue Moon Lander. Credits: Blue Origin



Orion Crew vehicle. Credits: ESA



Starship Crew and Cargo. Credits: Space X

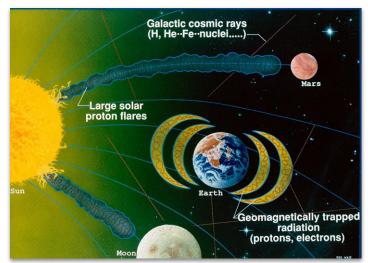
Radiation Environment

Space is filled with radiations, of 2 main types:

- Solar Particle Events (SPEs)
- Galactic Cosmic Rays (GCRs)

Both types have consequences on:

- Electronics
- Materials
- Human body



Recap of space radiation sources.



Supernova explosion (Artist's impression).



Solar flare with Coronal Mass Ejection (Observation).

Beyond Earth

Main Objective

Measurements

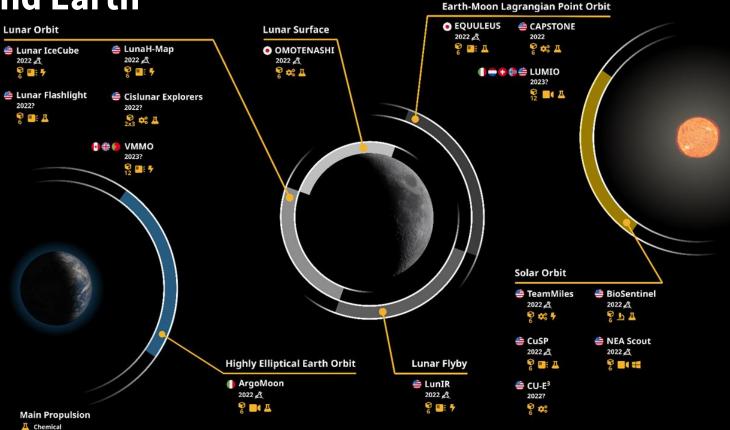
b On-Board Analysis

Tech. Demonstration

F Electric

Solar Sail

[Imagery



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SAC

DRACO Mission Analysis

DRACO = **D**etection of **RA**diation in **C**islunar **O**rbits

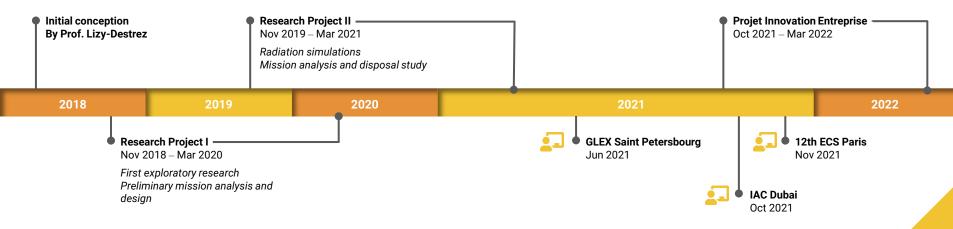
Mission statement

"Characterise the radiation environment and demonstrate the use of CubeSats on Earth-Moon Lagrangian point orbits."

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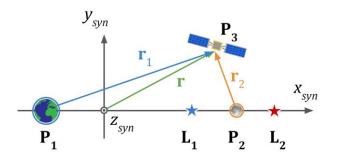
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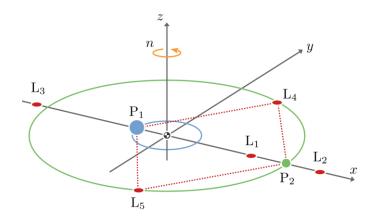


Dynamical Environment

- The Earth and the Moon dominate the "cislunar" space.
- Analysis are run in the Circular Restricted
 Three Body Problem (CR3BP):
- There are 5 libration points around which exist periodic solutions → Orbits!



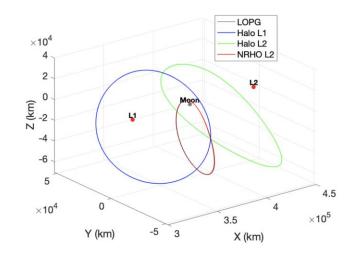
The Earth-Moon system in the rotating "synodic" frame



A rotating frame, used to study the Restricted Three Body Problem. The 5 Lagrange points are highlighted.

DRACO mission Conops

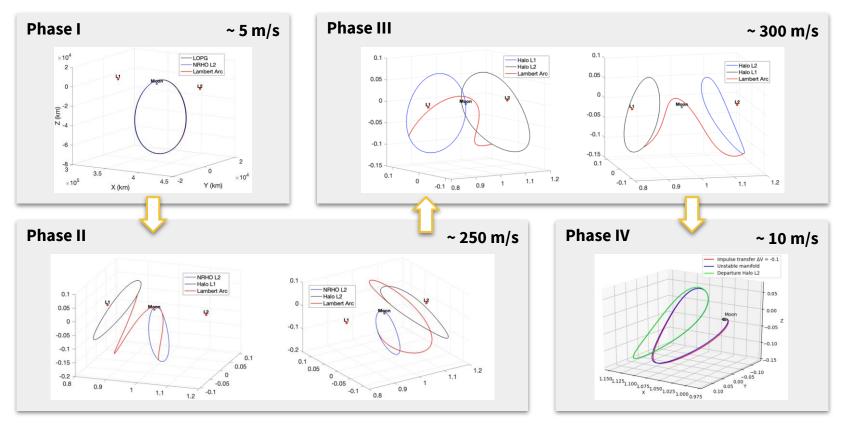
- Phase I. Release from Gateway of DRACO-1 and DRACO-2 to NRHO-L₂ and calibration.
- Phase II. Transfer DRACO-1 and DRACO-2 to science Halo orbits around L₁ and L₂.
- Phase III. Exchange science Halo orbits.
- Phase IV. Disposal of both CubeSats with impact on the lunar surface.



The orbits involved in the DRACO preliminary study [1]

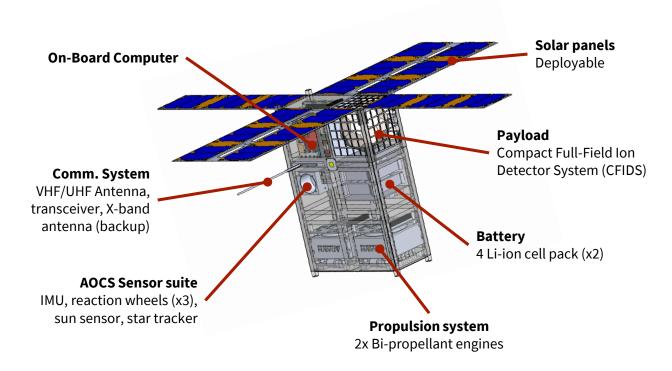
[1] M. Burnichon, E. V. Llamazares, Á. C. Serrano, P. Guardabasso, and S. Lizy-Destrez, "Mission Analysis to Measure Radiation Dose About the Earth-Moon Lagrangian Points" presented at the Global Space Exploration Conference (GLEX), Saint Petersburg, Russia, 2021.

Trajectory Analysis



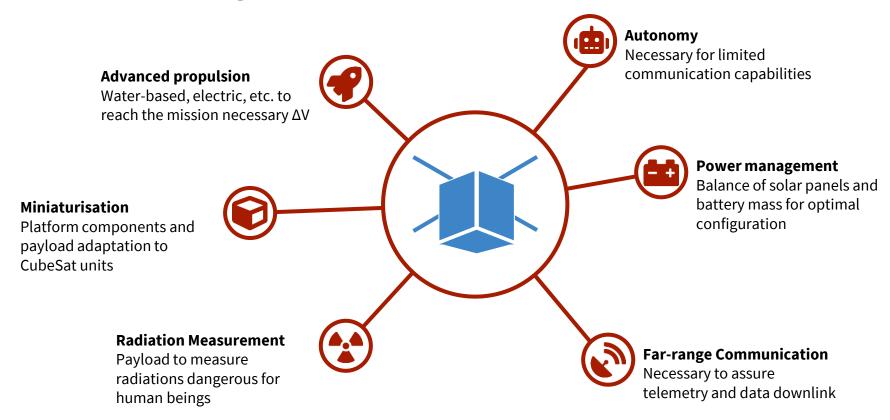
DRACO System Design

Preliminary System Design





Key technologies



Conclusion and future work

Conclusions...



Science and Payload

Review of available models

Preliminary dose simulations



Mission Analysis

Analysis of science orbits Computation of transfer ΔV



System design

Components technology scouting
Preliminary mass and power budget

Conclusions...and future work



Science and Payload

Review of available models
Preliminary dose simulations



Detailed **review** of existing missions and sensors

Experiment **requirements** preparation



Mission Analysis

Analysis of science orbits Computation of transfer ΔV



Transfer and **stationkeeping** redesign and optimisation

High-fidelity trajectory refinement



System design

Components technology scouting
Preliminary mass and power budget



Evaluate other **CubeSat configuarations** (3U)

Prepare **parametric model** for generalised **system budgets**

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Thank you for your attention!
Any questions?



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