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1 **Modelling building airtightness pressurisation tests with**
2 **periodic wind and sharp-edged openings**

3

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7 **ABSTRACT**

8 This paper presents an approach to model building airtightness pressurisation tests and to
9 quantify uncertainties due to a periodic wind. Our approach assumes the building under test
10 can be represented as a single zone with two sharp-edged openings, one on the windward side
11 and one on the leeward side of the building. We detail the system of equations obtained which
12 deals with stack effect, air compressibility, and inertia of the air subjected to unsteady
13 pressure conditions at the boundaries of the openings. Solving the system of equations gives
14 the dynamic behaviour of the airflow rates and state variables in the enclosure and in the
15 leaks, making it possible to estimate the uncertainty of the air leakage coefficient in one-point
16 pressurisation tests. Our analyses show that: a) the wind fluctuations can yield much larger
17 uncertainties on the air leakage coefficient than the average wind alone; b) the wind frequency
18 influences significantly the uncertainty when it is smaller than the sampling frequency; and c)
19 the contribution of the zero-flow pressure uncertainty to the uncertainty of leakage airflow
20 rate error is of about the same size as that of the flow and pressure uncertainties at the
21 pressure measurement station.

22

23

24 **KEYWORDS**

25 Airtightness; air leakage; building; pressurisation test; infiltration; periodic wind

26

27 **HIGHLIGHTS**

- 28 • We present an approach to model building airtightness pressurisation tests and to
29 quantify uncertainties due to a periodic wind.
- 30 • The approach, based on an idealised building and physical models of the airflow rate
31 through the leaks, gives the airflow rates and state variables as a function of time.
- 32 • Our analyses show the limitations of steady-wind approaches.

33

34 **NOMENCLATURE**

A Area of opening (m²)

Subscripts and superscripts

1,2 Pertaining to opening number 1 or 2

c	Speed of sound ($= \sqrt{\gamma R T_e}$) (m s^{-1})	pd	Pertaining to pressurisation measurement device
c_p	Specific heat capacity at constant pressure ($= 1\,004 \text{ J kg}^{-1} \text{ K}^{-1}$)	$down$	Pertaining to leeward façades
c_v	Specific heat capacity at constant volume ($= 1.4 c_p$) ($\text{J kg}^{-1} \text{ K}^{-1}$)	e	Pertaining to outside the building
C_l	Air leakage coefficient at standard conditions ($\text{m}^3 \text{ s}^{-1} \text{ Pa}^{-n}$)	est	Estimated value
C_p	Wind pressure coefficients defined in Table 2 (-)	f	Pertaining to wind fluctuations
C_{pw}	Wind pressure coefficients (-)		
C_z	Discharge coefficients (-)	i	Pertaining to inside the building
f_w	Wind frequency (s^{-1})	l	Pertaining to leakage path
F	Coefficient defined in Table 2 (-)	l	Pertaining to leakage path
Fr	Froude number (-)		
g	Acceleration of gravity ($= 9.81 \text{ m s}^{-2}$)	$mean$	Mean value
G	Coefficient defined in Table 2 (-)	$nowindstack$	No wind or stack effect
		ref	Referring to reference
L	Length (m)	s	Referring to target pressure station
L_{build}	Reference building length (m)	spl	Referring to values averaged over sampling interval
		up	Pertaining to windward façades
Ld	Leakage distribution ratio (-)	w	Pertaining to wind
l_e	Effective length of opening (m)	zf	Zero-flow pressure measurement
n	Flow exponent (-)	*	Non-dimensional quantity
n_p	Polytropic exponent (-)	0	Referring to standard conditions (293.15 K, 101325 Pa)
p_{abs}	Absolute pressure (Pa)		
p	Pressure relative to external atmospheric pressure (Pa)		
q	Volumetric airflow rate ($\text{m}^3 \text{ s}^{-1}$)		
q_t	Volumetric airflow rate at standard conditions ($\text{m}^3 \text{ s}^{-1}$)		
R	Specific gas constant ($= 287.058 \text{ J kg}^{-1} \text{ K}^{-1}$)		
t	Time (s)		
T	Temperature (K)		
U	Wind speed at the building level (m s^{-1})		
$u(x)$	Uncertainty of a quantity x (units of x)		
V_i	Internal building volume (m^3)		
z	Altitude above ground level (m)		

Greek symbols

$\delta(x)$	Error of x (units of x)
$\Delta(x)$	Difference in the value of x between the outside and the inside of the building (Pa)
Φ	Heat flux (W)
γ	Ratio of specific heats ($\gamma = \frac{c_p}{c_v} = 1.4$)
ρ	Air density (kg m^{-3})
φ	Phase lag (s)

Terms

Error (of a quantity)	Measured quantity value minus a reference quantity value
Uncertainty (of a quantity)	Dispersion of values that could be reasonably attributed to the quantity of interest

Sign convention

The air flow rates are counted positive when air travels from outside to inside

36 **1 INTRODUCTION**

37 Airtightness is an important building characteristic that influences the amount of air passing
38 through building zones and walls. This is why there exists a significant body of literature
39 dealing with building airtightness impacts on energy use [1–6] and indoor air quality [5,7].

40 The general consensus that can be drawn is that, except maybe in mild climates without
41 cooling needs or moisture issues, building airtightness can greatly affect energy performance
42 and indoor air quality. Because the energy use impact is proportionally more significant in
43 low energy buildings which many policies aim to generalise to answer energy conservation
44 and climate change issues, an increasing number of regulations and programmes have
45 recently included requirements on building airtightness [8,9]. These regulations can include
46 requirements for minimum airtightness or default values that strongly penalise poor
47 airtightness; they can also include quality management requirements aiming at securing good
48 airtightness [10].

49 In this context, quantifying airtightness has become a key issue for building professionals
50 because failure to reach minimum targets can have significant legal and financial
51 consequences. Subsequently, there has been over the past 10 years a revived interest among
52 researchers and professionals concerning the reliability of pressurisation tests performed to
53 quantify building airtightness. On the practical side, this has led to revise the international
54 standard to perform building pressurisation tests (ISO 9972) [11] as well as a range of
55 additional details spelled out at national or local levels to specify conditions to perform such a
56 test, including certification programmes for testers [12]. As for the ASTM E779 standard
57 commonly used in North America, it has been revised in 2019 [13]. On the research side,
58 several authors have recently investigated uncertainties in building pressurisation tests [14–
59 18], building on previous work performed in the 1980s and 1990s including evaluations of the
60 impact of climatic conditions, calibration and variation on the value of the flow exponent [19–
61 29]. Note that pressurisation tests are commonly performed at a single or a series of pressure

62 stations. These are also called DC pressurisation tests. However, considerable research and
63 development are ongoing on pulse pressurisation [30], an alternative method consisting in
64 analysing the pressure decay of a pulse released in an enclosure to derive its airtightness
65 characteristics at low pressure.

66

67 More specifically on wind induced uncertainties in DC pressurisation tests , Modera and
68 Wilson (1990) [25] recommend four-surface pressure averaging and time-averaging of
69 pressure and airflow data to reduce scatter in windy conditions. Later, Walker et al. (2013)
70 [14] and Carrié and Leprince (2014; 2016) [15,31] have analysed the impact of wind on
71 uncertainties based on large experimental datasets for the former and an analytical model for
72 the latter. Walker et al. (2013) recommend multipoint testing at low wind speeds ($< 3 \text{ m s}^{-1}$)
73 when the reference pressure is 4 Pa to avoid extrapolation errors due a fixed flow exponent;
74 however, for the same reference pressure (4 Pa), both Walker et al. (2013) and Carrié and
75 Leprince (2016) recommend single point-testing at high wind speeds (above $5\text{-}6 \text{ m s}^{-1}$)
76 because using a default value of the flow exponent is better than trying to estimate it with low
77 pressure points. Prignon *et al.* (2019) [32] have investigated uncertainties in zero-flow
78 pressure measurements and have concluded that, because these could seriously impact
79 pressurisation test results, these should be accounted for in test standard ISO 9972. However,
80 to our knowledge, understanding the physics involved in the influence of wind on building
81 pressurisation tests remains unclear. There exists a threshold for the zero-flow pressure in ISO
82 9972 (5 Pa) beyond which the test is not valid. A wind speed of 6 m s^{-1} is mentioned in ISO
83 9972, but only to state that it is unlikely to have a zero-flow pressure of less than 5 Pa with
84 wind speed beyond that value. We are unaware of a thorough justification either for using the
85 zero-flow pressure criterion, or for linking a wind speed of 6 m s^{-1} to a zero-flow pressure of 5
86 Pa. Although the work of Carrié and Leprince (2014; 2016) gives a simple physical model

87 enabling the analysis of influencing parameters, one important limitation of their work lies in
88 the hypothesis of a steady wind. In the field, it is well known that wind speed fluctuations
89 influence pressure measurements used in the analysis of building pressurisation tests and
90 therefore the resulting building airtightness results.

91

92 In two papers [33,34], Ethridge focusses on fluctuating wind pressures in a naturally
93 ventilated enclosure with two openings. He lays down fundamental equations of four models
94 to address his problem. The flow equations of the most rigorous model—the “Quasi-
95 steady/Temporal inertia model”—are of interest to model other problems involving
96 fluctuating wind pressures, including pressurisation tests with unsteady wind. Note also that
97 in his book ([35], pp. 96-99), Ethridge explains how the quasi-steady assumption and time-
98 averaging can influence the error in the airflow rate. He details the case of an orifice plate
99 meter in a pipe and gives the error in the airflow rate as a function of the Strouhal number in
100 that specific case. However, the natural ventilation and orifice plate meter problems tackled
101 by Ethridge [33,34] are clearly different from that of a pressurisation test involving air forced
102 into or out of an enclosure by a pressurisation device and multiple openings subjected to
103 different pressures because of wind and stack effects.

104

105 In summary, we are unaware of previous work to model pressurisation tests with unsteady
106 winds or to analyse the uncertainty due to wind in such tests. The present paper extends
107 Carrié and Leprince’s model in order to characterise the influence of periodic wind on
108 building one-point pressurisation tests with leaks acting as perfect orifices. We start with
109 physical equations that give the airflow rates in cracks with unsteady pressure boundary
110 conditions in an idealised building and analyse the results of our physical model for several
111 cases.

112

113 **2 QUASI-STEADY TEMPORAL INERTIA MODEL EQUATIONS**

114 **2.1 Idealised building**

115 In accordance with Carrié and Leprince (2016), we assume that the building can be

116 represented by a single zone separated from the outside by two types of walls: walls on the

117 windward side of the building which are subject to the same upwind pressure; and walls on

118 the leeward side which are subject to the same downwind pressure (Figure 1). We further

119 assume that all leaks on the windward (respectively, leeward) side can be represented as a

120 single opening at a given height—e.g., z_l if opening 1 is on the windward side (respectively,

121 leeward) —subjected to the same pressure difference.

122

123 In the field, the roof, the sides (façades n°3 and n°4 in Figure 1) and the leeward façade (n° 2

124 in Figure 1) have negative pressure coefficients on average; the windward façade (n°1) has a

125 positive pressure coefficient on average. In our idealised building, only one average pressure

126 coefficient is considered for all leaks on the windward (resp. leeward) façades. Similarly, only

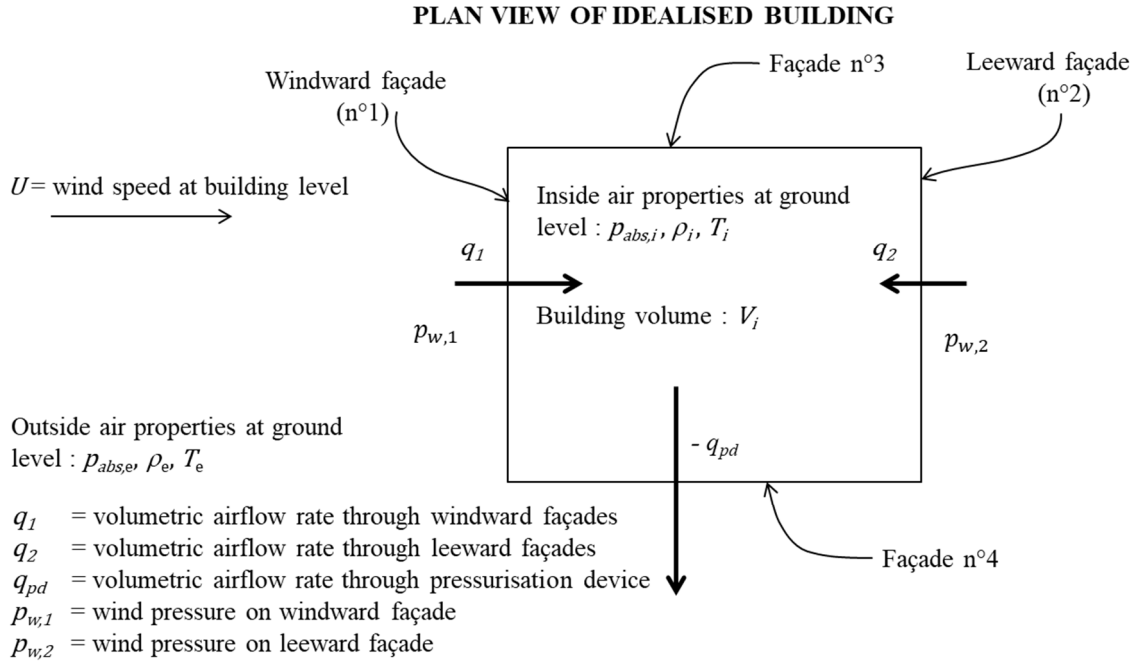
127 one average height is considered for all leaks on the windward (resp. leeward) façades. This is

128 of course a crude representation of the complexity of real airflow paths. Nevertheless, it has

129 the advantage of remaining relatively simple while allowing us to calculate the airflow rates

130 in leaks subjected to different pressures during a pressurisation test, which is the key problem

131 in presence of wind. The system of interest can be represented as shown in Figure 1.



132

133 **Figure 1: Plan view of idealised building with wind facing façade 1**

134 **2.2 Temperature and density of airflows entering and leaving the building**

135 To remain consistent with the complexity level of our idealised building, we assign a density
 136 and temperature to the airflow paths entering or leaving the space as if air was directly
 137 transferred in or out of the enclosure, without interaction with the building fabric or
 138 pressurisation equipment. This leads to the equations detailed in Table 1.

139

Density	Eq. n°	Temperature	Eq. n°
$\rho_{pd} = \begin{cases} \rho_e & \text{if } \text{sign}(q_{pd}) \geq 0 \\ \rho_i & \text{if } \text{sign}(q_{pd}) < 0 \end{cases}$	(1)	$T_{pd} = \begin{cases} T_e & \text{if } \text{sign}(q_{pd}) \geq 0 \\ T_i & \text{if } \text{sign}(q_{pd}) < 0 \end{cases}$	(2)
$\rho_1 = \begin{cases} \rho_e & \text{if } \text{sign}(q_1) \geq 0 \\ \rho_i & \text{if } \text{sign}(q_1) < 0 \end{cases}$	(3)	$T_1 = \begin{cases} T_e & \text{if } \text{sign}(q_1) \geq 0 \\ T_i & \text{if } \text{sign}(q_1) < 0 \end{cases}$	(4)
$\rho_2 = \begin{cases} \rho_e & \text{if } \text{sign}(q_2) \geq 0 \\ \rho_i & \text{if } \text{sign}(q_2) < 0 \end{cases}$	(5)	$T_2 = \begin{cases} T_e & \text{if } \text{sign}(q_2) \geq 0 \\ T_i & \text{if } \text{sign}(q_2) < 0 \end{cases}$	(6)

140 **Table 1: Density and temperature equations at airflow paths**

141

142 **2.3 Pressure difference at leakage sites**

143 The pressure difference at the windward opening is:

$$\Delta p_1(t) = p_{w,1} - (\rho_e - \rho_i)gz_1 - p_i(t) \quad (7)$$

144

145 Similarly, at the leeward opening:

$$\Delta p_2(t) = p_{w,2} - (\rho_e - \rho_i)gz_2 - p_i(t) \quad (8)$$

146 2.4 Flow through leakage sites

147 In his analysis of unsteady flow effects due to fluctuating wind pressures, Ethridge (2000a,
 148 2000b) [33,34] proposes to use a so-called “*Quasi-steady/Temporal inertia model*” (or *QT*
 149 *model*) to describe the flow through the leaks. Applying the same approach in our case leads
 150 to the following flow equations through the leaks:

$$2C_{z,1}A_1l_{e,1}\frac{dq_1(t)}{dt} = -q_1^2(t)\text{sign}(q_1(t)) + 2C_{z,1}^2A_1^2\frac{\Delta p_1(t)}{\rho_1} \quad (9)$$

$$2C_{z,2}A_2l_{e,2}\frac{dq_2(t)}{dt} = -q_2^2(t)\text{sign}(q_2(t)) + 2C_{z,2}^2A_2^2\frac{\Delta p_2(t)}{\rho_2} \quad (10)$$

151 We assume the cracks in the envelope behave like sharp-edged orifices the discharge
 152 coefficient is commonly assumed to be 0.6:

$$C_{z,1} = C_{z,2} = 0.6$$

Regarding the entry length, Ethridge (2000a) used Modera’s experimental data (11)

(Modera, 1989) [36] to show that the opening diameter was a reasonable estimate

of this parameter. Therefore, we write:

$$l_{e,1} = \sqrt{\frac{4A_1}{\pi}}; l_{e,2} = \sqrt{\frac{4A_2}{\pi}} \quad (12)$$

153 **2.5 Equation of state**

154 We assume that the air behaves like a perfect gas, therefore:

$$p_{abs,0} = \rho_0 R T_0 \quad (13)$$

$$p_{abs,e} = \rho_e R T_e \quad (14)$$

$$p_{abs,i} = p_{abs,e} + p_i = \rho_i R T_i \quad (15)$$

155

156 **2.6 Continuity equation**

157 The mass air flow rates through the leakage sites and the pressurisation device are:

$$q_{m,pd} = \rho_{pd} q_{pd} \quad (16)$$

$$q_{m,1} = \rho_1 q_1 \quad (17)$$

$$q_{m,2} = \rho_2 q_2 \quad (18)$$

158

159 Writing the mass balance of the internal volume gives:

160

$$\frac{dm_i(t)}{dt} = V_i \frac{d\rho_i(t)}{dt} = q_{m,pd}(t) + q_{m,1}(t) + q_{m,2}(t) \quad (19)$$

161 **2.7 Energy conservation equation**

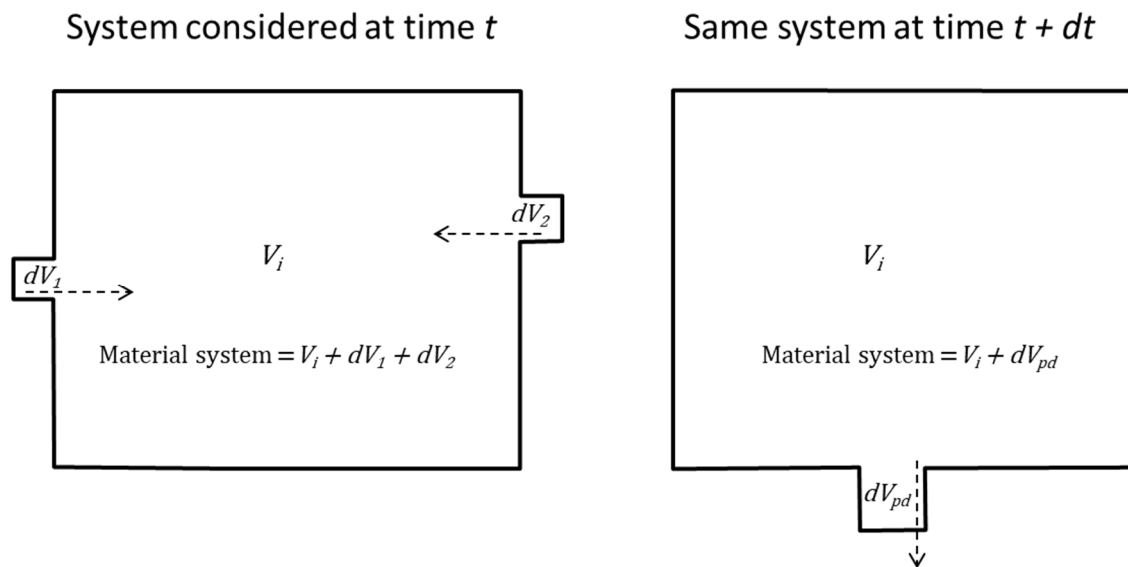
162 Ethridge (2000a; 2000b) assumes that the bulk behaviour of the internal volume follows a

163 polytropic process, yielding:

$$\frac{P_i(t)}{\rho_i(t)^{n_p}} = K \quad (20)$$

164 where K is a constant. For a perfect gas, if the process is adiabatic, $n_p = \gamma$; whereas, if the
 165 process is isothermal, $n_p = 1$. In the adiabatic case, this relationship stems from applying the
 166 state equation and the first principle of thermodynamics to a closed system with homogeneous
 167 state variables. This assumption is convenient because it reduces the energy conservation
 168 equation to a simple mathematical form (equation (20)). Although the air pressure and
 169 temperature inside values differ from outside values, this assumption can be justified with the
 170 Boussinesq approximation—i.e., the density variations are considered only in the buoyancy
 171 terms, in other words, in the terms where gravity appears in equations (1) and (3)—and since
 172 the absolute pressure is nearly identical inside and outside the building (see Figure 2).

173



174 **Figure 2: Closed system considered with the Boussinesq approximation and with $P_i \approx P_e$ (the air is**
 175 **assumed to enter the building through openings 1 and 2 and leave through the pressurisation device).**
 176

177 However, in the more general case, the internal volume is an open system and air at different
 178 temperatures flows in and out of this volume. When we apply the energy conservation
 179 principle to the internal volume, accounting for energy exchanges due to the air entering and
 180 leaving the building, but neglecting other energy transfer through the walls, mechanical stress
 181 or internal sources, we obtain:

$$V_i c_v \frac{d(\rho_i(t) T_i(t))}{dt} = \Phi_{pd}(t) + \Phi_1(t) + \Phi_2(t) \quad (21)$$

182 Where:

183

$$\Phi_{pd} = \rho_{pd} c_p q_{pd} T_{pd} \quad (22)$$

$$\Phi_1 = \rho_1 c_p q_1 T_1 \quad (23)$$

$$\Phi_2 = \rho_2 c_p q_2 T_2 \quad (24)$$

184

185 **2.8 Zero-flow pressure**

186 Depending on wind speed and temperature conditions, the pressure in the building when all
 187 intentional openings are closed is not null. This pressure is called the zero-flow pressure. It is
 188 subtracted from the measured pressures at the pressure stations to derive airtightness
 189 characteristics (see ISO 9972 [11] or ASTM E-779 [13]).

190

191 We obtain the zero-flow pressure with equations (1) through (19) and (21) through (24),
 192 where we set $q_{m,pd}$ to 0.

193 **2.9 Leakage airflow rate error**

194 Assuming a flow exponent of 0.5 (sharp-edged orifices), the estimated airflow rate through
 195 the leaks at the target pressure station is:

$$q_{l,est}(p_s) = C_{l,est} \sqrt{|p_s|} \text{sign}(p_s), \quad (25)$$

196

197 where, accounting for temperature and pressure corrections as described by Carrié (2014)
 198 [37], the estimated leakage coefficient of the building for a one-point measurement test can be
 199 calculated as follows, in pressurisation mode:

$$C_{l,est} = -\sqrt{\frac{\rho_i}{\rho_0}} \frac{\frac{\rho_e}{\rho_i} q_{pd}}{\sqrt{|p_{zf} - p_i|} \text{sign}(p_{zf} - p_i)}, \quad (26)$$

200 and in depressurisation mode:

$$C_{l,est} = -\sqrt{\frac{\rho_e}{\rho_0}} \frac{\frac{\rho_i}{\rho_e} q_{pd}}{\sqrt{|p_{zf} - p_i|} \text{sign}(p_{zf} - p_i)} \quad (27)$$

201 The error in the estimated leakage airflow rate at the pressure station is:

$$\delta_r(q_l) = \frac{\delta q_l}{q_l} = \frac{q_{l,est} - q_{l,nowindstack}}{q_{l,nowindstack}} \quad (28)$$

202 where $q_{l,est}$ and $q_{l,nowindstack}$ are estimated at the same pressure. The leakage airflow rate in
 203 no wind or stack effect condition $q_{l,nowindstack}$ is the leakage airflow rate that would be
 204 observed in standard conditions applying a perfect orifice equation. It is obtained with
 205 equation (25) by substituting $q_{l,est}$ and $C_{l,est}$ with $q_{l,nowindstack}$ and $C_{l,nowindstack}$ where:

$$C_{l,nowindstack} = \sqrt{\frac{2}{\rho_0}} (C_{z,1} A_1 + C_{z,2} A_2) \quad (29)$$

206

207 Note that, in the case of a pressurisation with a unique pressure station, we can infer the air
 208 leakage coefficient error by calculating the error in the leakage airflow rate compared to no
 209 wind or stack effect condition as follows:

$$\frac{\delta C_l}{C_l} = \frac{\delta q_l}{q_l} = \frac{C_{l,est} - C_{l,nowindstack}}{C_{l,nowindstack}} \quad (30)$$

210

211

212 3 QUASI-STEADY COMPRESSIBLE AND ISOTHERMAL MODELS

213 Based on the QT model equations detailed above, we have derived two other models with
 214 additional assumptions. These additional models are simpler. They are meant to evaluate the

215 added-value of the complexity introduced with the inertia term in the flow equations, as well
216 as with variable air density and temperature.

217

218 The “Quasi-steady Compressible” (QC) model neglects the inertia term in equations (9) and
219 (10), i.e., the temporal derivative on the left-hand side of the equations. The temporal
220 derivatives of the air density and pressure (equations (19)(21)) remain to be solved. Still, it
221 reduces the dimension of system of ordinary differential equations from 4 in the QT model to
222 2.

223

224 Last, the “Quasi-Steady isothermal » (QS) model neglects the inertia term, the air
225 compressibility, and temperature differences. By construction, this model violates the perfect
226 gas law, therefore the equations listed in § 2.5 do not apply. Its key advantage is that its
227 solution can be derived analytically, with an approach similar to that of Carrié and Leprince
228 (2016).

229

230 Note that the QT and QC models defined in this paper are based on the same assumptions as
231 those derived by Ethridge (2000a), except for the Boussinesq assumption and closed-system
232 assumption implicit in the use of the polytropic law.

233

234 **4 PRESSURISATION TEST CONDITIONS**

235 **4.1 Wind conditions**

236 Users of the present model could use time series of upstream and downstream wind pressures
237 measured simultaneously as model inputs. This could be interesting for comparing measured
238 and model results for real test cases. However, to predict errors in the leakage airflow rates
239 without wind pressure measurements on a specific test case, the users must assume a given

240 form for the wind pressure series. We are not aware of previous work giving simultaneous
241 upstream and downstream wind pressures for given building characteristics, including the
242 time lag, without wind measurements. To overcome this problem, we have assumed in such
243 cases that the wind speed fluctuates around a mean value as follows:

$$U(t) = U_{mean} \left(1 + \frac{U_f}{U_{mean}} \sin(2\pi f_w t) \right), \quad (31)$$

244 and that the wind pressures could be inferred from the wind speed with:

$$p_{w,1} = C_{pw,1} \frac{\rho_e U^2(t)}{2} \quad (32)$$

$$p_{w,2} = C_{pw,2} \frac{\rho_e U^2(t)}{2} \quad (33)$$

245

246 In other words, the wind speed fluctuation is reduced to a single mode of a Fourier transform
247 and the wind pressures are derived from wind pressure coefficients available in the literature.

248 **4.2 Sampling interval**

249 ISO 9972 requires the zero-flow pressure to be measured over 30 seconds with at least 10
250 points but does not detail requirements for sampling intervals at pressure stations. In this
251 paper, we chose to take the same sampling interval for the measurement at zero-flow and for
252 the measurement at the pressure station.

253 **4.3 Air temperature and pressure**

254 The outside air temperature T_e and pressure $p_{abs,e}$ are constant during the pressurisation test.
255 The difference between inside and outside air pressure is controlled by the pressurisation
256 device (see § 4.4), whereas the inside air temperature is an output of the model.

257 **4.4 Pressurisation device pressure control**

258 We assume an ideal pressure control of the pressurisation device, i.e.:

$$\forall t, p_i(t) = p_s \quad (34)$$

259 In other words, at all times, the pressurisation device controller sets the building pressure to
260 the target pressure station by ideally adjusting the pressurisation device airflow rate.

261

262

263 **4.5 Initial conditions**

264 In order to simulate pressurisation tests with the QT or QC models, one must specify initial
265 conditions for the following internal air temperature and pressure, as well as for the leakage
266 flows through the leaks and the pressurisation device.

267

268 Nevertheless, to avoid unrealistic behaviours of the internal air conditions and leakage
269 airflows, we set the initial conditions to the values obtained after a time step short enough to
270 have little variations of the state variables and a steady wind to avoid inertial effects in the
271 leaks.

272 **4.6 Range of input parameters**

273 The sensitivity analysis performed further in this paper is restricted to the following ranges of
274 the input parameters:

- 275 — The mean wind velocity U_{mean} varies between 0 and 10 m s⁻¹;
- 276 — The wind fluctuation U_f is set to 20% of the mean value;
- 277 — The pressure measurement station is set to a target value of 10 or 50 Pa; or to the
278 maximum between those target pressures and a multiple of the zero-flow-pressure.
279 This latter option allows us to simulate the constraint $|p_s| \geq 5 |p_{zf}|$ required in ISO
280 9972 for a test to be valid;
- 281 — The wind pressure coefficients are set to $C_{pw,1} = 0.25$ and $C_{pw,2} = -0.5$ inspired
282 from Liddament (1996) [38] (page 242, data for a low-rise building with a length-to-
283 width ratio of 2:1 and surrounded by obstructions equivalent to half of the building
284 height);
- 285 — The leakage distribution ratio $Ld = \frac{C_{z,2} A_2}{C_{z,1} A_1}$ is set to 4. Because we have one windward
286 façade and four leeward façades, this means that the leakage area is equally distributed
287 on all façades of the building;
- 288 — The initial air conditions are set to standard conditions;
- 289 — The outside air conditions are set either to standard conditions, or in anisothermal test
290 cases, with $T_e = 273.15 K$.

291

292 We do not apply the zero-flow pressure constraint $|p_{zf}| \leq 5 \text{ Pa}$ set in ISO 9972 for the test to
293 be valid. This allows us to analyse the influence of the input parameters on the leakage
294 airflow rate error without discarding results *a priori*.

295

296 **5 NON-DIMENSIONAL EQUATIONS**

297 **5.1 Dimensionless groups and reference values**

298 The equations below are given in non-dimensional form, which reduces the number of inputs
299 and unknowns and helps identify governing dimensionless variables. Let the non-dimensional
300 numbers defined in the appendix. The reference values can be set arbitrarily, but it is
301 convenient to define the reference wind pressure, $p_{w,ref}$, the reference wind speed, U_{ref} , and the
302 reference length, L_{ref} , as follows:

$$p_{w,ref} = p_s \tag{35}$$

$$U_{ref} = \sqrt{\frac{2 |p_{w,ref}|}{\rho_0}} = \frac{q_{pd,ref}}{(C_{z,1} A_1 + C_{z,2} A_2)} \tag{36}$$

$$L_{ref} = \frac{V_i}{(C_{z,1} A_1 + C_{z,2} A_2)} \tag{37}$$

303

304 This choice of reference values implies that:

305 — the relative pressures, including the pressure exerted by the wind on the façades, are
306 compared to the pressure set by the pressurisation device;

307 — the wind velocity is compared to the velocity of the air passing through the leaks;

308 — the time is compared to the characteristic time of air renewal of the enclosure during the
309 pressurisation test, $\frac{V_i}{q_{pd,ref}}$.

310 **5.2 Key QT-model equations**

311 With the non-dimensional groups defined in the appendix and reference values defined in §
312 5.1, the non-dimensional pressure differences and airflow rates at the leakage sites are:

313

$$\Delta p_1^* = C_{p,1} - \frac{2}{Fr_1^2} (\rho_e^* - \rho_i^*) - C_{p,i} \quad (38)$$

$$\Delta p_2^* = C_{p,2} - \frac{2}{Fr_2^2} (\rho_e^* - \rho_i^*) - C_{p,i} \quad (39)$$

$$\frac{dq_1^*}{dt^*} = \frac{F_1}{2} \left(-q_1^{*2} \text{sign}(q_1^*) + \frac{\Delta p_1^*}{\rho_1^*} \right) \quad (40)$$

$$\frac{dq_2^*}{dt^*} = \frac{F_2}{2} \left(-q_2^{*2} \text{sign}(q_2^*) + \frac{\Delta p_2^*}{\rho_2^*} \right) \quad (41)$$

314 The continuity and energy equations become:

$$\frac{d\rho_i^*}{dt^*} = \rho_{pd}^* q_{pd}^* + \frac{1}{1+Ld} \rho_1^* q_1^* + \frac{Ld}{1+Ld} \rho_2^* q_2^* \quad (42)$$

$$\frac{1}{2} \frac{1}{c^{*2}} \frac{d(C_{p,i})}{dt^*} = \rho_{pd}^* T_{pd}^* q_{pd}^* + \frac{1}{1+Ld} \rho_1^* T_1^* q_1^* + \frac{Ld}{1+Ld} \rho_2^* T_2^* q_2^* \quad (43)$$

315 Assuming a closed system with homogenous state variables, which is implicit in the energy
 316 equation (20) used by Ethridge (2000a), the dimensionless air densities and temperatures are
 317 equal to unity. Logically, with this assumption and with only two openings (no pressurisation
 318 device), the energy equation becomes strictly identical to the equation derived by Ethridge
 319 (2000a) (QT model) (referenced (28) in his paper, with $L_{ref} = H$, see appendix).

320

321 When the pressure inside the building is perfectly controlled, the left hand-side of equation
 322 (43) vanishes, yielding:

$$\rho_{pd}^* T_{pd}^* q_{pd}^* + \frac{1}{1 + Ld} \rho_1^* T_1^* q_1^* + \frac{Ld}{1 + Ld} \rho_2^* T_2^* q_2^* = 0 \quad (44)$$

323

324 Taking the resulting model as a whole, the dimensionless quantities during the measurement

325 (i.e., dimensionless airflow rate, pressure, temperature, and density) are function of 17

326 parameters:

$$\begin{aligned} & q_{pd}^*, T_{pd}^*, \rho_{pd}^*, q_1^*, T_1^*, \rho_1^*, q_2^*, T_2^*, \rho_2^*, p_i^*, T_i^*, \rho_i^* \\ & = f(t^*, U_{mean}^*, U_f^*, f_w^*, C_{pw,1}, C_{pw,2}, Fr_1, Fr_2, Ld, F_1, F_2, p_s^*, c^*, p_{abs,w}^*, p_{abs,e}^*, T_e^*, \rho_e^*) \end{aligned} \quad (45)$$

327 In addition, in equation (45), q_{pd}^* and p_i^* are linked by the pressurisation measurement

328 control. For the zero-flow pressure measurement, the device airflow rate is set to zero and the

329 output quantity of interest is the internal pressure, whereas for the measurement at the

330 pressure station, the internal pressure is controlled depending on the target pressure and the

331 output quantities of interest include in particular the airflow rate through the pressurisation

332 device.

333 **5.3 Uncertainty of zero-flow pressure and leakage airflow rate error**

334 Because in real conditions the zero-flow pressure is estimated with independent

335 measurements before and after the measurements at the pressure station, assuming the wind

336 probability distribution remains the same, the uncertainty of the zero-flow pressure is:

$$u(p_{zf}^*) = \frac{u(p_{zf,spl}^*)}{\sqrt{2}} \quad (46)$$

337 Equation (28) allows us to calculate the leakage airflow rate error for each measurement

338 point, which is identical to the leakage air coefficient error in a one-point measurement

339 pressurisation test. Using dimensionless variables, the relative error in the leakage airflow rate

340 for a given sampling interval and a given measured zero-flow pressure is, in pressurisation:

$$\delta_r(q_l) = -\sqrt{\rho_{i,spl}^*} \frac{\rho_e^*}{\rho_{i,spl}^*} q_{pd,spl}^* \sqrt{\frac{|p_s^*|}{|p_{zf}^* - p_{i,spl}^*|}} \text{sign}(p_{zf}^* - p_{i,spl}^*) - 1 \quad (47)$$

341 and, in depressurisation:

$$\delta_r(q_l) = -\sqrt{\rho_e^*} \frac{\rho_{i,spl}^*}{\rho_e^*} q_{pd,spl}^* \sqrt{\frac{|p_s^*|}{|p_{zf}^* - p_{i,spl}^*|}} \text{sign}(p_{zf}^* - p_{i,spl}^*) - 1 \quad (48)$$

342 where $\rho_{i,spl}^*$, $q_{pd,spl}^*$, and $p_{i,spl}^*$ are values averaged over the sampling interval. Note that
 343 there is an infinite number of possible values of the error $\delta_r(q_l)$ (and of p_{zf}^*) depending on
 344 when the sample starts relative to the wind fluctuations. In our case, the wind gives a leakage
 345 airflow rate error that has both: a random component that depends on when the zero-flow and
 346 pressure-station samples start relative to wind fluctuations; and, a systematic component due
 347 to the effect of wind which can be characterised with equations (47) and (48) for given
 348 sampling intervals. Here, the quantity of interest for our uncertainty analysis is the systematic
 349 component (the systematic error in the leakage airflow rate) and its associated uncertainty.

350

351 To find out a reasonable range of possible values of this leakage airflow rate error—that is, its
 352 uncertainty—with a perfect controller, we set $p_{i,spl}^* = p_s^*$ and we neglect air density variations
 353 which are expected to be small in the vast majority of test conditions. Therefore, in
 354 uncertainty analysis, $\delta_r(q_l)$ may be viewed as a function of $q_{pd,spl}^*$ and p_{zf}^* . Applying
 355 uncertainty analysis principles as described in “GUM - Guide to the expression of uncertainty
 356 in measurement”[39], assuming no correlation between the airflow rate and the zero-flow
 357 pressure, we obtain the uncertainty of $\delta_r(q_l)$ as follows:

358

$$u^2(\delta_r(q_l)) = \left| \frac{p_s^*}{p_{zf}^* - p_s^*} \right| u^2(q_{pd,spl}^*) + \frac{q_{pd}^{*2}}{4} \left| \frac{p_s^*}{(p_{zf}^* - p_s^*)^3} \right| u^2(p_{zf}^*) \quad (49)$$

359 where p_{zf}^* , q_{pd}^* and their associated standard uncertainties represent the expected values of
360 these quantities, i.e., their average value over possible sampling intervals.

361

362 **5.4 Boundary conditions**

363 The dimensionless wind velocity is:

364

$$U^*(t) = U_{mean}^* + U_f^* \sin(2\pi f_w^* t^*) \quad (50)$$

365 f_w^* is a dimensionless frequency also known as Strouhal number.

366 As for the pressure boundary conditions, the definitions of the reference wind speed and

367 pressure yield:

368

$$C_{p,1} = C_{pw,1} \rho_e^* U^{*2} - p_s^*; C_{p,2} = C_{pw,2} \rho_e^* U^{*2} - p_s^* \quad (51)$$

369 The perfect pressurisation device controller gives:

$$p_i^*(t^*) = p_s^* \quad (52)$$

370

371 The boundary conditions for the outside air temperature and pressure, at any time, including

372 $t^*=0$, are:

$$T_e^* = \frac{T_e}{T_0}, \quad (53)$$

$$p_{abs,e}^* = \frac{p_{abs,e}}{p_{abs,0}}, \quad (54)$$

373 whereas the inside air temperature and pressure boundary condition is only set for $t^*=0$:

$$T_i^*(t^* = 0) = \frac{T_i(t = 0)}{T_0} \quad (55)$$

$$p_{abs,i}^*(t^* = 0) = \frac{p_{abs,i}(t = 0)}{p_{abs,0}} \quad (56)$$

374 The model also requires initial conditions for the airflow rates through the pressurisation
375 device and the leaks which are obtained as explained in § 4.5.

376

377 **6 RESULTS ON TEST CASES**

378 **6.1 Code development and checking procedures**

379 We have implemented the model equations detailed above in a numerical code in Python
380 using the scipy package for the numerical integrator (“odeint” functions), with the range of
381 input parameters given in § 4.6. While the kernel of the models implementation uses
382 dimensionless variables, dimensional inputs/outputs can be specified/returned to compare our
383 results with field experience.

384

385 It is good coding practice to cross-check the consistency of results obtained with various
386 methods. In the following sub-sections, we have compared QT-model results obtained with
387 the other models presented in § 3.

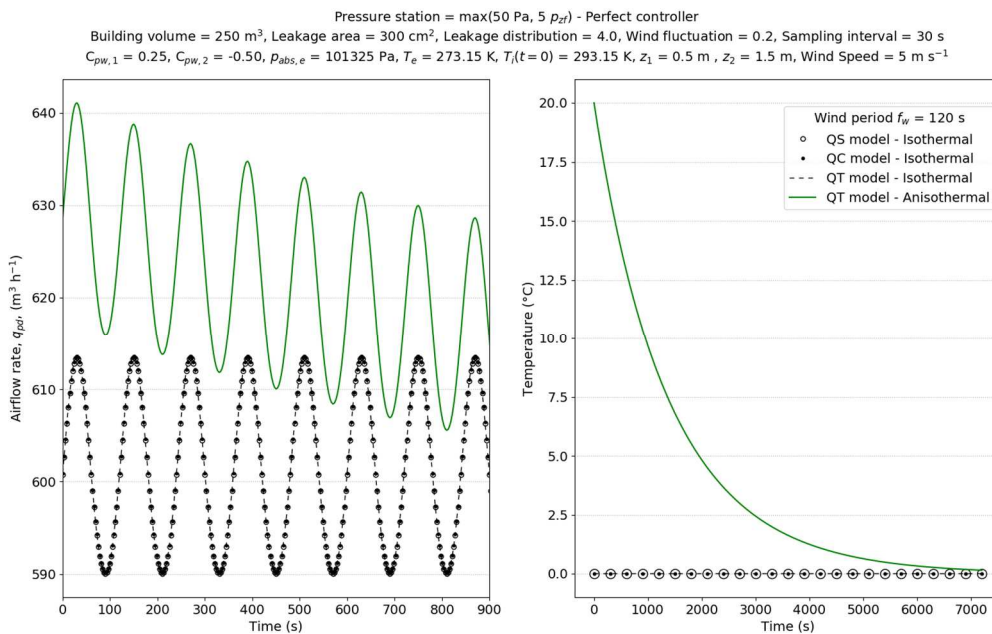
388 **6.2 Anisothermal conditions**

389 In our test cases, the external temperature remains constant over the measurement period.
390 Therefore, as expected, the internal temperature converges towards the external temperature
391 with our QT-model (Figure 3). In the test conditions of Figure 3, the temperature and the
392 airflow rate through the pressurisation device take about two hours to converge. The resulting
393 leakage airflow rate is over estimated by about 5% at the beginning of the simulation

394 compared to the true value, whereas it is underestimated by about 2% at the end of the
 395 simulation due to the wind effect alone (Figure 4). To obtain Figure 4, we had to assume a
 396 given zero-flow pressure. In our case, it was set to the mean zero-flow pressure in the test
 397 conditions of Figure 3 and on same simulation interval (7200 s). Note also, for code checking
 398 purposes, that the QT-, QC-and QS-model results shown in Figure 3 are consistent in
 399 isothermal conditions.

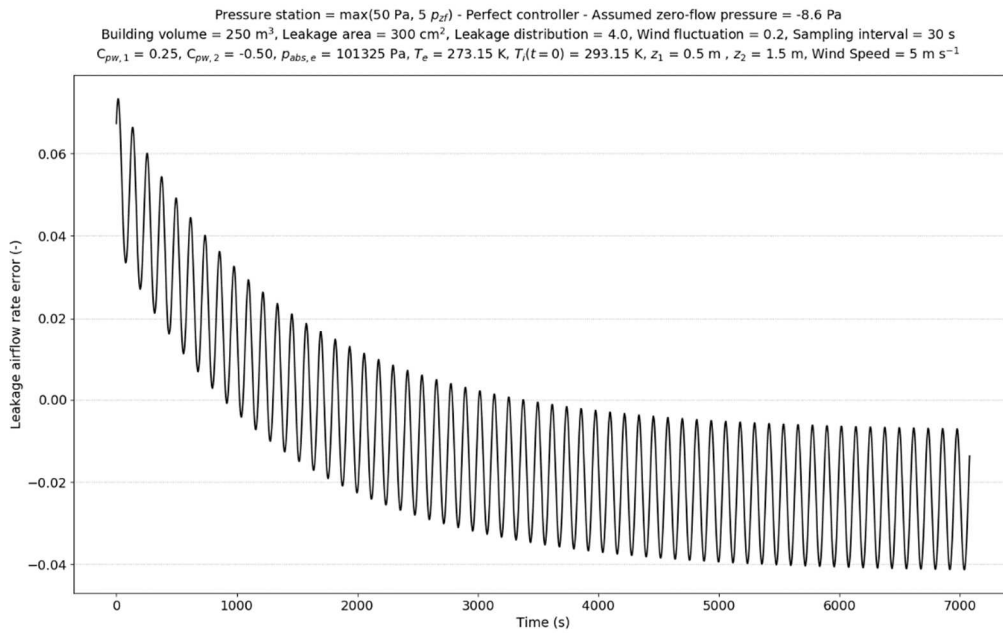
400

401 Anisothermal conditions influence pressurisation test results: depending on when the
 402 technician starts averaging results, the air temperature conditions change and thereby the
 403 measured zero-flow pressures or leakage airflow rates. Therefore, although there is no
 404 intrinsic limitation to our QT model for anisothermal conditions, such conditions considerably
 405 complicate the analysis of the results, including uncertainty estimates. For these reasons, the
 406 analyses of test cases in the following sub-sections focus on isothermal conditions.



407
 408

409 **Figure 3: Leakage airflow rate and temperature time series in isothermal and anisothermal conditions**
 410 ($U_{mean} = 5 \text{ m s}^{-1}$, $f_w = 120 \text{ s}$).



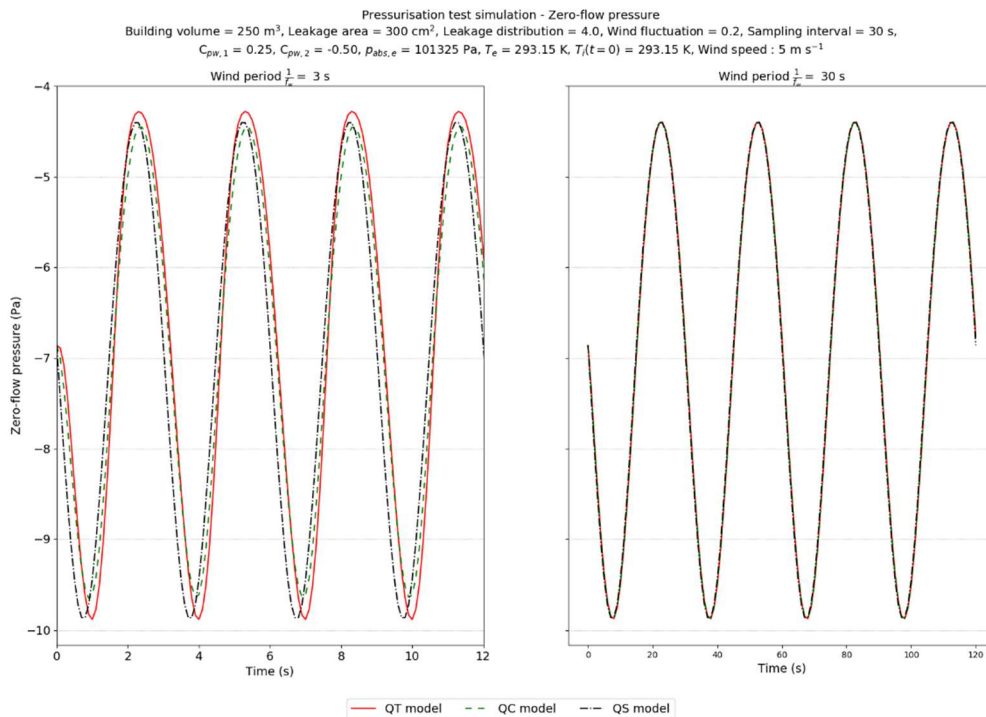
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412
413

Figure 4: Leakage airflow rate error in anisothermal conditions ($U_{mean} = 5$ m s⁻¹, $f_w = 120$ s).

414 **6.3 Inertia in the leaks**

415 In the previous example, the QS, QC and the QT models give nearly identical results in
 416 isothermal conditions. However, for the zero-flow pressure time series, the inertial effects
 417 become visible at high frequencies as shown in Figure 5. We observe the same trend if we
 418 increase the building volume, which we expected from the dimensional analysis since the
 419 time derivatives of equations (40)(41) scale with F_1 and F_2 which increase with increasing
 420 volume. For the pressure station time series, we have to set the frequency or the building
 421 volume to higher values to observe similar trends as with the zero-flow pressure time series.
 422 This result is likely influenced by the pressure control strategy of the pressurisation device,
 423 i.e. the algorithm that controls the pressure during the test. Inertial effects could be more
 424 visible with more realistic control strategies because these yield an unsteady internal pressure.
 425 This would merit further research using realistic wind spectra to find out in which cases
 426 accounting for inertial effects is useful.

427



428
 429 **Figure 5: Zero-flow pressure time series ($U_{mean} = 5$ m s⁻¹, $f_w = 3, 30$ s).**

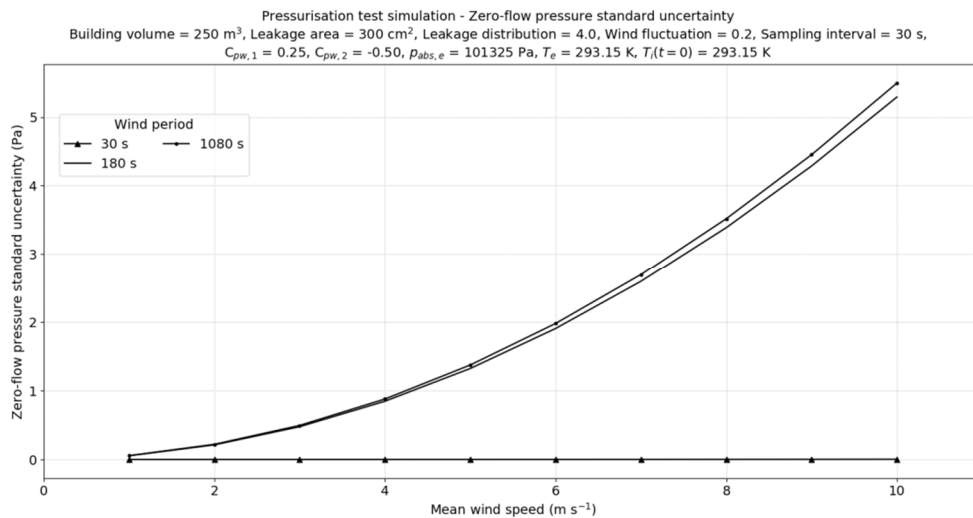
430 **6.4 Zero-flow pressure uncertainty**

431 Figure 6 shows the standard uncertainty of the zero-flow pressures obtained from two
 432 sampling intervals. Logically, the uncertainty increases with decreasing wind frequencies:
 433 indeed, we expected the range to be small for wind periods smaller than the sampling time
 434 intervals because averages are biased when performed on sampling time intervals which are
 435 not multiples of the time period. This observation is consistent with the field results of
 436 Prignon et al. (2019) who concluded that longer sampling periods yielded lower uncertainties.
 437 Prignon's field measurements made under relatively low wind speeds (mean wind speed over
 438 10 minutes lower than 3.8 m s⁻¹) showed estimates on the order of 0.4-1.5 Pa of the zero-flow
 439 pressure standard uncertainty. However, we must exercise caution comparing these field
 440 results with our numerical results because of model limitations (e.g., wind modelling with a
 441 sinusoidal signal versus real wind with many modes and changes in direction) and several
 442 inputs not assessed in the field, including: wind fluctuation amplitude, pressure coefficients,
 443 or leakage distribution. For these reasons, we have not tried to match our test cases with the
 444 inputs available from Prignon's study.

445

446 For our test conditions, our QT model gives zero-flow pressure standard uncertainties in the
447 range of 0.3-2 Pa for wind speeds in the range 3-6 m s⁻¹ and wind periods larger than the
448 sampling time intervals, as shown in Figure 6. This figure also shows that the standard
449 uncertainty can be much greater, for instance, greater than 5 Pa for low-frequency winds (or
450 high wind periods) at 10 m s⁻¹.

451



452

453 **Figure 6: Zero-flow pressure standard uncertainty (averaged over 2 sampling intervals) as a function of**
454 **wind speed for various wind periods.**

455 **6.5 Leakage airflow rate error**

456 Figure 7 shows the leakage airflow rate times series for various wind frequencies as well as
457 the uncertainty estimated with equation (49). The smaller bars represent the uncertainty in the
458 error due to the zero-flow contribution, while the triangles represent that of the leakage
459 airflow rate at the pressure station. In these specific cases, the contributions of the zero-flow
460 pressure measurement (third term of equation (49)) and the leakage airflow measurement
461 (second term of equation (49)) at the pressure station to the combined uncertainty of the
462 leakage airflow rate error are comparable (Figure 7). In Figure 7, as logically as for the zero-
463 flow pressure uncertainty, the uncertainty of the leakage airflow rate error increases with
464 increasing wind periods (or decreasing wind frequencies).

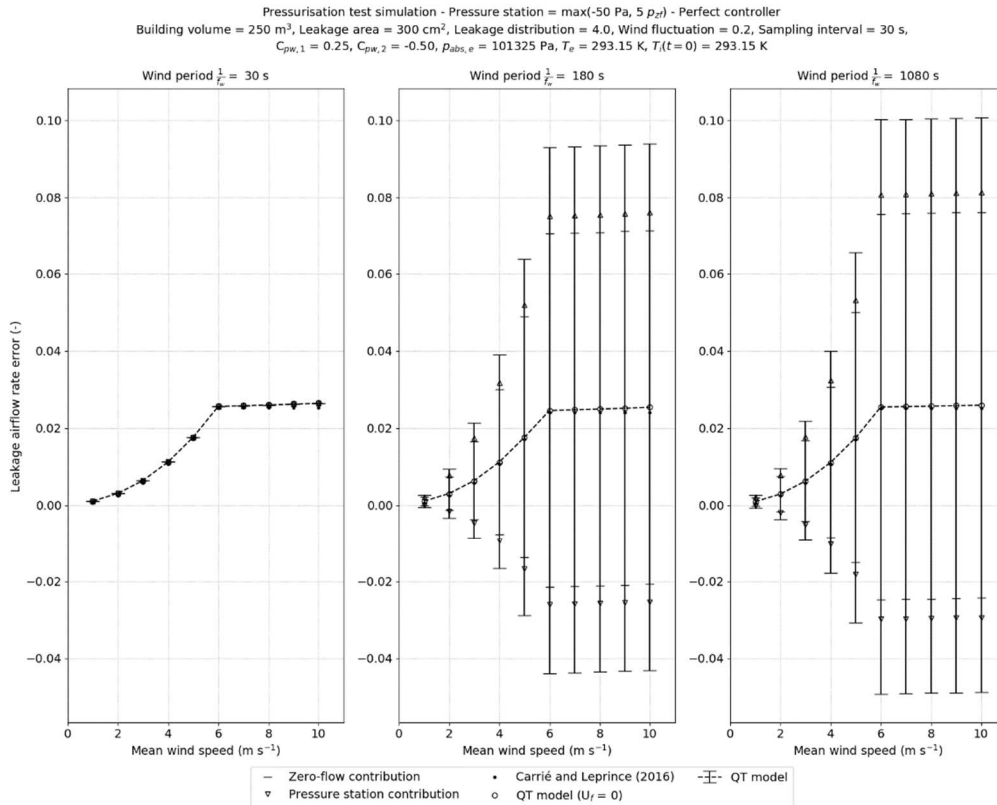
465

466 Note also that, for code checking purposes, our QT-model results with $U_f = 0$ and those

467 obtained with the steady model developed by Carrié and Leprince (2016) are identical.

468

469



470

471

472 **Figure 7: Leakage airflow rate error for various wind speeds and wind periods. The larger bars represent**
473 **the uncertainty (coverage factor = 2) in the error due to wind fluctuations.**

474 7 DISCUSSION

475 Experienced technicians are aware of specific difficulties to obtain repeatable results when

476 performing pressurisation tests in windy conditions, including stable zero-pressure, pressure

477 station and flow readings. This was reported by several authors, sometimes with diverging

478 conclusions regarding the influence of wind. For instance, Persily (1982) [19] reports errors

479 of 15% for an average wind speed of 4.9 m s⁻¹ measured 6 m above the roof of a 2-storey

480 building, whereas Brennan et al. (2013) [40] did not find a clear bias due to wind

481 speeds up to about 10 m s^{-1} . Although surprising, these diverging results could be due to the
482 specificities of the test cases, i.e., diverging parameters that strongly influence the leakage
483 airflow rate error including wind fluctuation amplitude, pressure coefficients, leakage
484 distribution, or leakage area as shown in our analyses.

485

486 Although we observe some deviation between the QT and QC models at high wind
487 frequencies and low building airtightness, the added-value of the QT model is not obvious in
488 our test cases. Note however that these conclusions should not be extended beyond our test
489 cases. In particular, more realistic control strategies yielding short term variations of the
490 leakage airflow rate could show more clearly inertia flow effects in the leaks. This applies to
491 pressurisation at a given pressure station (also called DC pressurisation), but implementing
492 appropriate control strategies would also allow one to simulate AC or Pulse pressurisation
493 [30,41–43] using the key model equations detailed in this paper.

494

495 The models detailed in this paper assume perfect sharp-edged orifices and a rough estimate of
496 the entry length which is a key parameter influencing inertial effects. In addition, the simple
497 representation of parallel orifices with one single orifice questions the validity of Equation
498 (12) to give a proper estimate of the entry length. Because we have a non-linear system of
499 differential equations, it is not possible to extrapolate our results based on two sharp-edged
500 orifices to alternate flow models. Modified flow equations based on power law or a quadratic
501 law could be useful to better represent the flow behaviour in an array of cracks. If work is
502 undertaken in this area, the results of Walker et al. (1998) [29] suggest that the priority should
503 be given to the power law approach.

504

505 One specific added-value of our analysis lies in the quantification of the range of the
506 dispersion of the zero-flow pressure measurements and its contribution to the uncertainty in
507 the estimated leakage airflow rate. Delmotte (2017) [16] as well as Prignon et al. (2019) [32]
508 have recently stressed the importance of the zero-flow pressure in uncertainty analyses. Our
509 results show that indeed, the contributions of the zero-flow pressure measurement and the
510 measurement at the target pressure station have about the same size. Again, these results are
511 limited to the few test cases we have examined; however, they reinforce previous authors'
512 results as well as the need to better account for the zero-flow pressure error. Note that, to our
513 knowledge, present pressurisation test protocols ignore this aspect.

514

515 Without appropriate measurement wind pressure data or models, restricting the analyses to a
516 periodic wind with a simple relationship giving the wind pressures was convenient. Note that,
517 assuming there is no time lag between the upstream and downstream wind signals, our
518 conclusions regarding the sensitivity of the uncertainty to wind period can be extended to
519 each mode of any wind velocity or wind pressures decomposed in Fourier series.
520 Nevertheless, it is unclear how to tackle the time lag issue and how significant its impact
521 could be.

522

523 We have proposed a non-dimensional formulation of the problem. Besides the fact that it
524 reduces the number of parameters used in the kernel of the code and improves its stability, it
525 is useful to understand the cross-influence of parameters, as shown in the analysis of inertial
526 effects. This approach would also be useful for statistical analyses to identify key influencing
527 parameters or to determine protocols that would minimise uncertainty, possibly using local
528 climatic data. Finally, the dimensional analysis proves to be useful for our current work to
529 design a laboratory experiment to investigate the validity of our model.

530

531 **8 CONCLUSION**

532 Our model allows one to simulate the dynamic behaviour of the airflow rates and state
533 variables in an enclosure with sharp-edged openings subjected to a pressurisation test with
534 periodic wind. This makes it possible to estimate the uncertainty on the air leakage
535 coefficient. The analyses are restricted to one-point pressurisation tests at a given pressure
536 station (DC pressurisation) in this paper but they could be extended to multi-point DC tests or
537 alternative test methods such as AC or Pulse pressurisation. This would be useful because
538 these other methods may give smaller uncertainties, in particular at low wind speed and low
539 reference pressure. The fundamental equations would remain the same but some new analyses
540 would be required to derive the airtightness characteristics from the airflow rate, pressure and
541 temperature time series. Extending our model to other alternate flow models to better describe
542 the flow behaviour in leaks (e.g., quadratic or power law) would require additional research
543 efforts on the flow equations.

544

545 Our analyses show that a steady approach to uncertainties in building pressurisation tests is
546 quite limited because the wind fluctuations can yield much larger uncertainties than the
547 average wind alone. In addition, we have shown the significant impact of the wind frequency
548 on the results and have confirmed that ignoring the zero-flow pressure uncertainty is
549 inappropriate because of its significant contribution to the uncertainty of the leakage airflow
550 rate error. Although our model has clear limitations, we expect these key conclusions
551 generally hold in field pressurisation tests.

552

553 Overall, our approach could be used to develop pressurisation test protocols to minimise
554 uncertainties, in parallel to investigating several aspects highlighted in this paper, including

555 the sensitivity of the results to the leakage flow model or the lack of adequate experimental
 556 wind and pressurisation test data.

557

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 560 of this publication lies with the authors.

561

562 10 APPENDIX

563 10.1 Dimensionless groups

564 Let the non-dimensional numbers defined in Table 2.

565

$$\begin{aligned}
 t^* &= t \frac{U_{ref}}{L_{ref}}; f_w^* = \frac{L_{ref}}{U_{ref}} f_w \\
 p_{abs,e}^* &= \frac{p_{abs,e}}{p_{abs,0}}; p_{abs,i}^* = \frac{p_{abs,i}}{p_{abs,0}}; p_{abs,w}^* = \frac{\frac{\rho_0 U_{ref}^2}{2}}{p_{abs,0}} \\
 \rho_e^* &= \frac{\rho_e}{\rho_0}; \rho_i^* = \frac{\rho_i}{\rho_0}; \rho_{pd}^* = \frac{\rho_{pd}}{\rho_0}; \rho_1^* = \frac{\rho_1}{\rho_0}; \rho_2^* = \frac{\rho_2}{\rho_0}; \\
 T_e^* &= \frac{T_e}{T_0}; T_i^* = \frac{T_i}{T_0}; T_{pd}^* = \frac{T_{pd}}{T_0}; T_1^* = \frac{T_1}{T_0}; T_2^* = \frac{T_2}{T_0}; \\
 Fr_1 &= \frac{U_{ref}}{\sqrt{g z_1}}; Fr_2 = \frac{U_{ref}}{\sqrt{g z_2}} \\
 Ld &= \frac{C_{z,2} A_2}{C_{z,1} A_1}; F_1 = \frac{L_{ref}}{l_{e,1}}; F_2 = \frac{L_{ref}}{l_{e,2}} \\
 G_{pd} &= \frac{L_{ref} (C_{z,1} A_1 + C_{z,2} A_2)}{V_i}; G_1 = \frac{L_{ref} C_{z,1} A_1}{V_i} \\
 q_{pd}^* &= \frac{q_{pd}}{(C_{z,1} A_1 + C_{z,2} A_2) U_{ref}}; q_1^* = \frac{q_1}{C_{z,1} A_1 U_{ref}}; q_2^* = \frac{q_2}{C_{z,2} A_2 U_{ref}} \\
 q_{m,pd}^* &= \rho_{pd}^* q_{pd}^*; q_{m,1}^* = \rho_1^* q_1^*; q_{m,2}^* = \rho_2^* q_2^*; \\
 \Phi_{pd}^* &= q_{m,pd}^* T_{pd}^*; \Phi_1^* = q_{m,1}^* T_1^*; \Phi_2^* = q_{m,2}^* T_2^* \\
 U^* &= \frac{U}{U_{ref}}; c^* = \frac{c}{U_{ref}}
 \end{aligned}$$

$$p_i^* = \frac{p_i}{\frac{\rho_0 U_{ref}^2}{2}}; p_s^* = \frac{p_s}{\frac{\rho_0 U_{ref}^2}{2}}; p_{w,ref}^* = \frac{p_{w,ref}}{\frac{\rho_0 U_{ref}^2}{2}}$$

$$C_{p,1} = \frac{C_{pw,1} \frac{\rho_e U^2}{2} - p_{w,ref}}{\frac{\rho_0 U_{ref}^2}{2}}; C_{p,2} = \frac{C_{pw,2} \frac{\rho_e U^2}{2} - p_{w,ref}}{\frac{\rho_0 U_{ref}^2}{2}}; C_{p,i} = \frac{p_i - p_{w,ref}}{\frac{\rho_0 U_{ref}^2}{2}}$$

$$\Delta p_1^* = \frac{\Delta p_1}{\frac{\rho_0 U_{ref}^2}{2}}; \Delta p_2^* = \frac{\Delta p_2}{\frac{\rho_0 U_{ref}^2}{2}}$$

566 **Table 2: Non-dimensional numbers**

567 **10.2 Non-dimensional continuity and energy equations**

568 The continuity and energy equations become:

$$\frac{d\rho_i^*}{dt^*} = G_{pd} \rho_{pd}^* q_{pd}^* + G_1 \rho_1^* q_1^* + G_1 Ld \rho_2^* q_2^* \quad (57)$$

$$\frac{1}{2} \frac{1}{c^{*2}} \frac{d(C_{p,i})}{dt^*} = G_{pd} \rho_{pd}^* T_{pd}^* q_{pd}^* + G_1 \rho_1^* T_1^* q_1^* + G_1 Ld \rho_2^* T_2^* q_2^* \quad (58)$$

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