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### <sup>1</sup> Plasma properties conditioned by the magnetic throat location <sup>2</sup> in a Helicon plasma device

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Measurements are conducted in a Helicon plasma device to analyze the spatial distribution of plasma properties as the throat of the magnetic nozzle is axially shifted with respect to the antenna center. Krypton plasma is generated in the sub-kilowatt range and probed using a suite of diagnostics including a rf-compensated Langmuir probe, a planar probe and laser-induced fluorescence. It is found that larger ion currents and increased plasma confinement are achieved when the throat of the magnetic nozzle is located downstream the antenna center, at a distance that equals or exceeds two times the antenna length. The ions, although being accelerated, retain subsonic velocities even beyond the magnetic throat.

### 7 I. INTRODUCTION

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A recent keen interest in electric space propulsion stems 8 <sup>9</sup> from an increasing number of small satellites being deployed <sup>10</sup> for plural applications<sup>1-6</sup>. Simple and compact thrusters can <sup>11</sup> allow for in-space maneuvering of such small spacecrafts<sup>7</sup> <sup>12</sup> which usually do not boast large volume capacity to meet the <sup>13</sup> needs of consolidated technologies, e.g. Gridded Ion thrusters 14 and Hall thrusters. As a result, a portion of research and devel-15 opment efforts are being focused on electrodeless devices and <sup>16</sup> magnetic nozzle (MN) acceleration<sup>8–13</sup>. This kind of technol-<sup>17</sup> ogy can indeed offer a number of advantages compared to the 18 aforementioned state-of-the-art devices. Issues linked to ero-<sup>19</sup> sion of components are expected to be of little consequence 20 due to the absence of direct interaction between plasma dis-<sup>21</sup> charge and electrodes, resulting in a long lifetime. The plasma 22 beam leaving the discharge chamber is quasineutral, thus no 23 neutralizer is required. Furthermore, these plasma sources 24 can more easily operate using condensible propellants with <sup>25</sup> respect to those system that require a dedicated neutralizer, <sup>26</sup> whose lifetime and reliable functioning can be considerably <sup>27</sup> affected<sup>14</sup>.

The Helicon plasma (HP) thruster is one such a device. It mainly comprises three components: *i*) a dielectric tube as plasma discharge region, *ii*) a radiofrequency (rf) antenna for gas ionization and *iii*) an axially directed steady magnetic field for plasma confinement and expansion throughout its diverging region, i.e. the magnetic nozzle. Power radiated by the antenna is deposited in the electron population as thermal energy by means of helicon wave modes<sup>15</sup> and Trivelpiece–Gould wave modes<sup>16</sup>. The energy is used for plasma production and transport phenomena, such as ambipolar electric field, resulting in axial acceleration of the ions in the MN by conversion of the electron thermal energy.

The maturity of this technology is still relatively low<sup>5</sup>, whereby several undesired aspects come along with the condescribed advantages. The most critical one perhaps regards the <sup>43</sup> thrust efficiency. Thus far, direct thrust measurements have <sup>44</sup> revealed efficiencies typically below 10% in many different <sup>45</sup> devices<sup>13,17–24</sup>, with the only recent exception of nearly 20% <sup>46</sup> at several kW of input power<sup>12</sup>. In this scenario, improv-<sup>47</sup> ing the attained performance is required in order to turn HP <sup>48</sup> thrusters into a competitive technology in the near future. The <sup>49</sup> better understanding of the physics governing such rf plasma <sup>50</sup> sources by exploring the different working parameters is cur-<sup>51</sup> rently the preferred approach to provide an exhaustive opti-<sup>52</sup> mization pathway.

In an effort to model HP discharges, previous analy-53 <sup>54</sup> ses have considered different magnetic topologies inside <sup>55</sup> the source tube, e.g. uniform<sup>25,26</sup>, Helmholtz and Maxwell  $_{56}$  configurations<sup>27</sup>. The magnetic field inside the source is then 57 matched with a divergent geometry through a magnetic throat 58 localized in the proximity of the tube exit. This firm condi-<sup>59</sup> tion is not necessarily the most convenient solution, as demon-60 strated in Ref. 28 where a stepped-diameter tube is employed 61 and the magnetic throat is placed downstream the exit cross 62 section of the small main tube. Besides, virtually every nu-63 merical and experimental work studying the influence of mag-64 netic nozzle on the plasma properties, limited the analysis to 65 the exploitation of different magnitudes at the throat. A fruit-<sup>66</sup> ful investigation is reported in Ref. 29, where the location of <sup>67</sup> the magnetic throat is indirectly shifted with respect to the an- $_{68}$  tenna – which is constantly located at the back plate – by <sup>69</sup> changing the discharge tube length. Another relevant analy-<sup>70</sup> sis is reported in Ref. 30, where the authors characterized the 71 plasma and the rf power circuit when the magnetic throat lo-72 calized either within the source or at the exhaust. Bennet et  $_{73}$  al.<sup>31</sup> studied the generation of plasma as a function of the MN 74 throat location. Yet, their setup did not include plasma expan-<sup>75</sup> sion outside the discharge tube and many operating conditions 76 relied on magnetized ions, which is not a typical scenario in 77 propulsion. The shape of the magnetic nozzle and the location 78 of the throat are presumably strongly linked to the amount <sup>79</sup> of the input energy that can be usefully converted into axial <sup>80</sup> thrust. Energetic electrons are mainly produced close to the <sup>81</sup> region occupied by the antenna. Several phenomena occur <sup>82</sup> in the nozzle, such as transport due to pressure gradient and 83 electron cooling via convection and/or conduction, which de-<sup>84</sup> termine the conversion efficiency of inlet thermal energy into

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FIG. 1. Helicon source schematics. The z = 0 position refers to the discharge tube outlet, y = 0 refers to the reactor axis.

85 ion acceleration. Furthermore, as the throat is shifted along <sup>86</sup> the thruster axis with respect to the antenna location, the ex-87 pansion ratio of the plasma beam changes, thus resulting in 88 possibly different thrust levels.

This paper reports on experimental results obtained in a HP 89 <sup>90</sup> source operating with krypton under several configurations of <sup>91</sup> the external magnetic field. Numerous plasma properties are <sub>92</sub> spatially resolved as the throat of the magnetic field is moved 93 with respect to the antenna axial position. A detailed descrip-94 tion of the experimental setup is provided in Section II, in-95 cluding the HP source and the magnetic configurations. In 96 Section III, the diagnostics are extensively described. They in-97 clude: a rf-compensated Langmuir probe to spatially charac-98 terize plasma density, electron temperature, and plasma poten-99 tial; a planar probe to measure the ion current density; laser-<sup>100</sup> induced fluoresce to explore the axial ions and atoms velocity <sup>103</sup> Section V.

### 104 **II. EXPERIMENTAL SETUP**

A schematics of the HP source employed in this study is 131 105 <sup>106</sup> shown in Figure 1 along with an example of computed mag-<sup>132</sup> tube and the expansion chamber. Their denomination is re-107 vertical position. 108

109 110 <sup>111</sup> open exit arbitrarily identifies the origin of the axial coordi-<sup>137</sup> netic field topologies. For the purpose of this study, five mag-<sup>112</sup> nate z. Krypton gas is introduced off-axis in the tube through <sup>138</sup> netic configurations are examined by powering two electro-<sup>113</sup> its top aperture at a nominal flow rate of  $1 \pm 0.004$  mg s<sup>-1</sup>. <sup>139</sup> magnets at once. In doing so, the location of the magnetic <sup>114</sup> The bottom part is connected to a 30 cm in inner diame-  $_{140}$  throat is intentionally relocated along the z axis. The tested <sup>115</sup> ter and 50 cm in length expansion chamber made in alu-<sup>141</sup> magnetic configurations are progressively labeled as Conf. A <sup>116</sup> minum which is equipped with a turbomolecular/primary <sup>142</sup> to Conf. E. When involved, PB coils are powered with 3 A <sup>117</sup> pumping system. The ultimate base pressure is in the order <sup>143</sup> and GB coils with 6 A of direct current. Figure 2 shows the  $_{118}$  of  $10^{-5}$  mbar, whereas the typical pressure during operation  $_{144}$  computed 2D magnetic field maps, highlighting the stream-<sup>119</sup> reads  $\sim 5 \times 10^{-3}$  mbar.

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FIG. 2. 2D profiles of the externally applied magnetic fields. White line indicates the discharge tube walls, white rectangles represent the magnetic coils, the gray rectangle indicates the antenna region and the black solid line exemplifies the magnetic streamline crossing the tube edge at z = 0.

<sup>101</sup> distribution function. Measurements are presented and dis-<sup>124</sup> that power losses due to intermediate links are prevented. In-102 cussed in Section IV. Eventually, conclusions are drawn in 125 put rf power is generated by a 1 kW-class power supply at 126 13.56 MHz and kept constant to 750 W throughout the re-127 ported measurements. The power carried by higher order har- $_{128}$  monics is certified to be lower than  $-40 \, \text{dBc}$ . Matching of <sup>129</sup> the rf power is attained via a custom  $\pi$ -type network. During  $_{130}$  operation, values of the reflected power are below  $\sim 2$  %.

A set of nine electromagnets surrounds both the discharge netic field lines. The entire experimental setup is oriented in a 133 ported in Figure 1. Each electromagnet is made up of a 2 mm 134 diameter copper wire, constituting 510 loops for PB1-6 and The discharge chamber consists in a borosilicate glass tube 135 430 loops for GB1-3. The presence of multiple electromagwith  $\phi = 9$  cm inner diameter and L = 58 cm length, whose <sup>136</sup> nets enables the investigation of profoundly different mag-145 line that crosses the discharge tube edge at z = 0, thus defin-The rf antenna in use is a double-saddle type with 146 ing the ideal shape of the plasma plume (whether it is assumed  $_{121}$   $d_A = 12$  cm length centered at  $z \approx -11$  cm. It is made of  $_{147}$  fully magnetized). With reference to Figure 2, it is noteworthy <sup>122</sup> pure copper and features two feeding arms which are long <sup>148</sup> to emphasize on the magnetic throat location. It moves from 123 enough to be directly connected to the matching network so 149 upstream the antenna in Conf. A to far downstream the an-



FIG. 3. Externally applied magnetic field on-axis profiles. Shaded area indicates the antenna location. Dashed lines represent simulation profiles, scatter points indicate probe measurements.

<sup>150</sup> tenna in Conf. E. The first axial mode the antenna can generate <sup>151</sup> translates in an axial wavelength as  $\lambda_z = 24$  cm, see Section <sup>152</sup> IV C. Notice that the magnetic throat in Conf. D is approxi-<sup>153</sup> mately one wavelength downstream of the antenna center. The <sup>154</sup> magnitude of the on-axis magnetic field is reported in Figure <sup>155</sup> 3 by comparing computed values and measured values. All <sup>156</sup> magnetic configurations feature a peak value of  $10 \div 13$  mT <sup>157</sup> with a similar qualitative profile but shifted along *z*.

### 158 III. DIAGNOSTICS

### 159 A. RF-compensated Langmuir probe

Plasma density  $n_p$ , electron temperature  $T_e$  and plasma potential  $V_p$  are measured using a rf-compensated Langmuir probe (LP), see Figure 4. A detailed description of the probe has been previously reported in<sup>11</sup>. It is designed in accordance with the results reported in<sup>32–34</sup>. A tungsten wire of 0.38 mm in diameter and 5 mm in length constitutes the probe tip. An additional electrode made of stainless steel contributes for to rf compensation and provides mechanical support to the assembly. The electrode is electrically insulated from the probe tip using a 1 mm in outer diameter alumina tube. A 1 nF axial capacitor is soldered to the compensation electrode and rra connected in parallel with the probe tip and a series of three axial chokes, each of which self-resonates at one of the first three harmonics. The whole assembly is encapsulated inside accommodates the coaxial cable for probe biasing and current



FIG. 4. RF-compensated Langmuir probe.



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FIG. 5. RF-compensated LP frequency spectrum ( $V_{ref} = 10$  V). Vertical dotted lines indicate 13.56 MHz and 27.12 MHz.

176 measuring.

Chokes selection is attained by characterization of their 177 <sup>178</sup> impedance frequency spectrum. A 20 V<sub>pp</sub> sinewave is gener-<sup>179</sup> ated by a function generator and fed to one lead of the testing 180 choke. The other lead is connected to a 46.52 k $\Omega$  resistor and 181 the series circuit closes to the ground. The frequency of the <sup>182</sup> input sinewave is swept while the voltage drop across the re-183 sistor is monitored and used to calculate the current flowing <sup>184</sup> in the circuit. The frequency at which the current is minimum 185 is the choke self-resonating frequency. Functional check of <sup>186</sup> the whole probe assembly is performed by applying a  $20 V_{pp}$ 187 sinewave to the tungsten tip at different frequencies. The 188 probe output is recorded by a digital oscilloscope and shown <sup>189</sup> in Figure 5 as a function of the input frequency. Local minima <sup>190</sup> are evident in correspondence of the harmonic frequencies, as 191 desired.

The probe is manually displaced within the plasma using 192 <sup>193</sup> a single-axis translation stage with a resolution of 10 µm and <sup>194</sup> a stroke of 150 mm. Probe alignment is ensured by the use 195 of a cross-line laser pendulum. I-V curves are recorded us-<sup>196</sup> ing a ALP control unit by Impedans Ltd. Thereafter, data is <sup>197</sup> post-processed relying on OML theory<sup>34,35</sup>. Assuming local <sup>198</sup> quasineutrality, plasma density  $n_p$  is inferred from the linear 199 fit of the ion current squared versus the probe potential. The 200 ion current fit is then subtracted from the total current to ob-201 tain a better estimation of the electron current. Assuming the <sup>202</sup> electrons are described by a Maxwell-Boltzmann distribution <sup>203</sup> function, the logarithmic profile of the electron current is lin-204 early fitted, enabling the estimation of the electron tempera-<sup>205</sup> ture  $T_e$ . The plasma potential  $V_p$  is computed as the probe bias 206 voltage where the first derivative of the I-V curve features its 207 peak value.

<sup>208</sup> Uncertainty associated with the probe measurements is es-<sup>209</sup> timated through a statistical approach. The detailed procedure <sup>210</sup> has been previously reported in<sup>11</sup>. The outline of the analysis <sup>211</sup> shows that all plasma parameters are measured with a statisti-<sup>212</sup> cal deviation as high as  $\sim 2 \%$ .

### 213 B. Planar probe with guard ring

The ion current density  $j_i$  is measured using a planar probe with a guard ring (PPGR). It mainly comprises a disk-shaped

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(a) 1.0-0- $\lambda_D$ 0.5 $s_f$ Length [mm] -0 $s_{hi}$ 0.0 0.3(d) 0.20.10.00.0 2.50.0 2.50.0 2.5 $x \, [\mathrm{cm}]$ 

FIG. 6. Ion current profiles at (x, y = 0; z = 15 cm). Subplots denomination is linked to magnetic configurations Conf.s A-E.

<sup>216</sup> collector and an outer ring, both made in stainless steel. The <sup>217</sup> role of the guard ring is to concentrate sheath edge effects far <sup>218</sup> from the collector, thus ensuring that the ion collection area <sup>219</sup> exactly corresponds to the collector geometrical area<sup>36</sup>. For <sup>220</sup> correct functioning of the probe, it is required that i) the width of the guard ring is much larger than the local plasma sheath 221 <sup>222</sup> thickness and *ii*) the gap between the two electrodes is smaller 223 than the local sheath thickness to introduce negligible poten-224 tial irregularities. The collector is 5.6 mm in diameter and <sup>225</sup> 1 mm in thickness, whereas the ring width is 1 mm. A gap <sup>226</sup> of 100 µm is maintained between the two electrodes to ensure 227 electrical insulation. Each electrode features a separated elec-228 trical connection for polarization and current measurement. Voltage is applied to the collector by way of a Keithley 2410 229 230 SourceMeter, which is also used to measure the flowing cur- 266 231 rent. The guard ring is biased using a TTIEX752M power 267 measured by means of laser-induced fluorescence (LIF) spec-232 supply.

Functional check of the probe is carried out by recording 233 <sup>234</sup> the I-V curves at (x, y = 0; z = 15 cm) for the five magnetic 235 configurations denominated Conf.s A-E. The curves obtained <sup>236</sup> with a biased and floating guard ring are compared in Figure Saturation of the ion current is achieved for Conf.s C-D-E 237 6. at voltages lower than about -40 V, while no complete satu-238 239 ration is recorded for Conf.s A and B due to the fact that the 240 guard ring width is not compatible with the local values of <sup>241</sup> the sheath thickness. This can be confirmed by computing the <sup>242</sup> local values of the characteristic lengths over the radial coor-<sup>243</sup> dinate using the values of  $n_p$  and  $T_e$  later discussed in Section <sup>244</sup> IV. Floating and high voltage sheath thickness,  $s_f$  and  $s_{hv}$  re- $_{245}$  spectively, are computed in accordance with  $^{37}$ , using -75 V <sup>246</sup> for Conf.s A-B and -50 V for Conf.s C-D-E. Outcomes are <sup>247</sup> plotted in Figure 7. It results evident that values of  $s_{hv}$  for Conf.s C-D-E are much smaller than the guard ring width, 248 whereas those for Conf.s A and B approach its size. It shall 249 be noted that the attainment of a perfectly flat profile of the 250 ion saturation current would require a guard ring width  $5 \div 10$ 251 <sup>252</sup> times larger than the actual one. Thus, the overall probe size would probably lead to serious plasma perturbations. Further-253 <sup>254</sup> more, the scope of the presented measurements is not to, e.g., 255 focus on indirect propulsive performance estimation but rather

FIG. 7. Debye length, floating and high-voltage sheath thickness at z = 15 cm. Subplots denomination is linked to magnetic configurations Conf.s A-E.

<sup>256</sup> to perform a more qualitative-like comparison of the five mag-257 netic configurations.

Given the high level of reproducibility recorded during pre-258 <sup>259</sup> liminary measurements, the uncertainty in  $j_i$  is mainly given <sup>260</sup> by the uncertainty in the measured current and the probe col-<sup>261</sup> lecting area. This consideration leads to a value of < 8 %.

The planar probe is manually actuated from outside the 262 263 chamber using the single-axis translation stage described in 264 Section III A.

### С. Laser-induced fluorescence setup 265

Ions and atoms velocity distribution functions (VDFs) are <sup>268</sup> troscopy. The technique and the optical bench in use have 269 been extensively described in previous works<sup>38,39</sup>. Excita-270 tion of metastable levels is achieved via an amplified tunable  $_{271}$  single-mode laser diode in the 810 - 840 nm spectral range. 272 The primary laser beam is split in multiple beams with the  $_{273}$  purpose of *i*) instantaneous monitoring of the laser mode and 274 detection of any mode hop by means of a Fabry-Pérot interfer-275 ometer, *ii*) laser power check using a photodiode and *iii*) accu-276 rate wavelength measurement by way of a calibrated waveme- $_{277}$  ter whose absolute uncertainty is  $\sim 60 \,\mathrm{m\,s^{-1}}$  (80 MHz). Mod-278 ulation of the laser beam is achieved using a mechanical chop- $_{279}$  per at  $\sim 1.9\,kHz.\,$  The laser is coupled to a 50  $\mu m$  optical 280 fiber which allows transporting the laser beam from the optical 281 bench to the Helicon source setup. Laser is injected inside the <sup>282</sup> vacuum chamber through a borosilicate glass window located

TABLE I. Kr I and Kr II probed optical transitions.

		$\lambda_{exc}$ [nm]	$\lambda_{fluo}$ [nm]
	Transition	(vacuum)	(air)
Kr I	$5s^2[3/2]_1^\circ \to 5p^2[3/2]_2$	819.23095	760.15457
Kr II	$5p^2D^{\circ}_{5/2} \to 5s^2P_{3/2}$	820.4959	461.91658

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FIG. 8. Photographs of the plasma plume in (a) Conf. A, (b) Conf. B, (c) Conf. C, (d) Conf. D and (e) Conf. E. Windows are centered at z = 12 and z = 24 cm.

<sup>283</sup> at the bottom of the reactor. Note that  $\mathbf{k} \cdot \mathbf{v} < 0$  here, where  $\mathbf{k}$  <sup>323</sup> IV. EXPERIMENTAL RESULTS AND DISCUSSION  $_{284}$  is the laser wavenumber and **v** the particle velocity. The laser 285 beam is collimated using a combination of a 150 mm and a 286 1000 mm focal length plano-convex lenses. Uncertainty on <sup>287</sup> the alignment with respect to the HP source axis is estimated  $_{288}$  to be  $\lesssim 1.6$  deg. In order to ensure a linear regime of the 289 studied optical transitions, the laser power density delivered <sup>290</sup> to the plasma is kept below  $1.2 \,\mathrm{W}\,\mathrm{mm}^{-2}$ , a threshold value <sup>291</sup> that has been determined during preliminary measurements. <sup>292</sup> A detection branch is placed perpendicularly to the excitation 293 beam featuring a pair of plano-convex lenses (250 mm and <sup>294</sup> 100 mm focal lengths) focusing on a 200 µm core diameter <sup>295</sup> optical fiber. Both the excitation and detection branches are <sup>296</sup> mounted on a dedicated computer-controlled high precision <sup>297</sup> linear translation stage to enable spatially resolved measure-<sup>298</sup> ments along the y axis. The detection optical train can be <sup>299</sup> additionally displaced along the *z* axis within a few centime-<sup>300</sup> ters using a manual stage. The 200 µm optical fiber transports 301 the fluorescence light to a monochromator which isolates the 302 desired line from the whole spectrum. Afterwards, a photo-303 multiplier tube is used to convert the light signal into volt-<sup>304</sup> age, which is fed into a lock-in amplifier operating at the laser <sup>305</sup> modulation frequency to distinguish the signal from the nat-306 ural plasma emission. The VDFs are inferred by scanning <sub>307</sub> over the laser wavelength  $\lambda$  and computing the particles axial velocity  $v_z$  as  $v_z = c(\lambda - \lambda_0)/\lambda_0$ , where  $\lambda_0$  is the unshifted 308  $_{309}$  transition wavelength and c is the speed of light. The mea-<sup>310</sup> sured LIF profiles result from the Doppler broaden line. The <sup>311</sup> presence of Kr isotopes leads to broadening due to isotopic <sup>312</sup> shift and hyperfine structure<sup>38</sup>. Moreover, due to the pres-313 ence of a steady magnetic field, the Zeeman effect splits the 347 <sup>317</sup> rescence line profile images the local VDF.

318 322 literature<sup>38–42</sup>.

### 324 A. Visual inspection

The very first piece of information regarding the way the 325 326 plasma responds to the applied magnetic field topology is pro-327 vided by visual inspection. The photographs of the plasma <sup>328</sup> plume reported in Figure 8 are taken using a common DSLR <sub>329</sub> camera framing the region 8 < z < 28 cm and set at f/8 and  $_{330}$  1/40 s exposure time. It results evident how significant differ-<sup>331</sup> ences characterize the extracted plasma plume when employ-<sup>332</sup> ing the different magnetic topologies. A faint plume is visible <sup>333</sup> when the reactor operates in Conf. A and Conf. B. Instead, a <sup>334</sup> well defined plasma beam is noticed when Conf.s C, D and E <sup>335</sup> are chosen. In the latter cases, the plasma radius visibly delin-<sup>336</sup> eates the local magnetic streamlines, cf. Figure 2 and Figure 8. 337 It is interesting to notice that the local magnitude of the ax-<sup>338</sup> ial component  $B_z$  is nearly equal for Conf. C and Conf. E at  $_{339}$  z = 12 cm (see Figure2), i.e. at the center of the top window <sup>340</sup> in Figure8. Yet, the two plasma beams show a very different <sup>341</sup> structure. As a matter of fact, local values of the field radial <sub>342</sub> component  $B_r$  and gradient  $\partial B_z/\partial z$  significantly differ. This <sup>343</sup> finding agrees with previous numerical models<sup>26,43</sup> and intrinsically points out that both  $|\mathbf{B}|$  and  $\nabla \mathbf{B}$  have an impact on the 345 plasma behavior.

### 346 B. LP measurements

Plasma properties are characterized using the diagnos- $_{314}$  spectral lines into sub-levels<sup>38</sup>. Nevertheless, since the spec-  $_{348}$  tics described in Section III. The region delimited by  $_{315}$  trum is dominated by the <sup>84</sup>Kr isotope and the magnetic field  $_{349}$  0 < y < 4 cm and -4 < z < 21 cm is probed by means of the 316 is relatively weak, it is herein assumed that the observed fluo- 350 rf-compensated LP. I-V characteristics are recorded with a 351 spatial resolution of 5 mm. Data is post-processed as dis-The probed optical transitions are reported in Table I. The 352 cussed in Section III A and eventually interpolated to enhance 319 zero-velocity wavelength for Kr I is estimated from prelimi- 353 visualization. Figures 9, 10 and 11 illustrate normalized 2D <sub>320</sub> nary measurements in the HP source shining the laser along  $_{354}$  maps of  $n_p$ ,  $T_e$  and  $V_p$ , respectively, for the five magnetic <sub>321</sub> the x axis. The other listed wavelengths are given in the  $_{355}$  topologies. Normalization values are:  $6 \times 10^{17} \,\mathrm{m}^{-3}$  as re-<sub>356</sub> gards to  $n_p$ , 6 eV for  $T_e$  and 25 V for  $V_p$ . The solid black lines

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FIG. 9. Normalized plasma density for the five magnetic configurations. Normalization factor is  $6 \times 10^{17} \text{ m}^{-3}$ . Solid black lines represent magnetic streamlines.



FIG. 10. Normalized electron temperature for the five magnetic configurations. Normalization factor is 6 eV. Solid black lines represent magnetic streamlines. 381



FIG. 11. Normalized plasma potential for the five magnetic configurations. Normalization factor is 25 V. Solid black lines represent magnetic streamlines.

<sup>357</sup> included in the figures represent the magnetic streamlines de-<sup>358</sup> limited by the discharge tube edge at z = 0. Furthermore, on-<sup>359</sup> axis profiles of  $n_p$ ,  $T_e$  and  $V_p$  are shown in Figure 12. These <sup>360</sup> experimental measurements show that:

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- 1. when the HP source operates in Conf. A, the extracted plasma plume is characterized by a low monotonically decreasing profile of  $n_p$  ranging between  $\sim 1 \times 10^{17}$  and  $\sim 9 \times 10^{15}$  m<sup>-3</sup> along the reactor axis. The profile of  $T_e$  features a monotonic behavior as well, with a peak value of  $\sim 3$  eV. In a similar fashion,  $V_p$  drops from  $\sim 24$  V to  $\sim 15$  V. All plasma properties peak onaxis at z < 0. These quantitative observations agree with the faint light emitted in the nozzle region, as stated beforehand. Visual check of the plasma also reveals that light is instead concentrated within the discharge tube. It suggests that the radial and back boundaries represent a major loss term of plasma<sup>26,44</sup>.
- 2. when the HP source operates in Conf. B, analogous arguments can be put forward. The profile of  $n_p$  likewise peaks on-axis at z < 0 and decays monotonically downstream. Altogether, slightly larger values of  $n_p$  are measured with respect to Conf. A. In the bulk of the plasma,  $T_e$  remains in the order of  $\sim 3 \text{ eV}$  with lower values downstream. However, a population of relatively more energetic electrons ( $T_e \sim 3.5 \text{ eV}$ ) appears off-axis, in correspondence of the magnetic nozzle edge, visible at z < 0 in Figure 10(b). Although qualitatively very

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similar, the distribution of  $V_p$  for Conf. B presents lower values with respect to Conf. A of a few volts.

3. when the HP source operates in Conf. C, results similar to those previously reported in<sup>11</sup> using the same HP source are obtained. As a matter of fact, in both the experiments the magnetic throat is located at z = 0. In contrast to what was discussed in points (i) and (ii), all plasma properties peak off-axis. Data acquisition in the region (0 < r < 2, z < 0) was not possible due to discharge instability possibly induced by the probe presence. The largest value of  $n_p$  is recorded at z > 0and it reads  $\sim 3 \times 10^{17}$  m<sup>-3</sup>, whereas  $T_e$  reaches a peak of 4 eV at z < 0. The distribution of  $V_p$  is relatively isotropic, ranging in the  $8 \pm 0.5$  V interval. The reason why  $n_p$  and  $T_e$  present separated regions of the respective maximum values has been explained in<sup>11,45</sup> in terms of pressure balance. When considering the z component of the electron momentum conservation equation in the limit of negligible contribution due to momentum transfer collisions and due to the Lorenz force, the potential gradient and the pressure gradient balance each other. In the region where  $B_r/B_z \ll 1$ , i.e. where the local magnetic field divergence angle is small, the reduced cross field electron mobility yields a relatively large axial conductivity, resulting in a nearly null potential gradient. Hence, for equilibrium reasons, the pressure gradient is required to be small, which results in a larger  $n_p$  where  $T_e$  is lower and vice versa. Similar results are numerically found and discussed in<sup>46</sup>.

4. when the HP source operates in Conf. D, light emission upstream the antenna is almost absent, hinting that plasma losses at the back and lateral walls are mitigated with respect to the previous cases. With reference to Figure 9(d), it is evident that  $n_p$  follows the axial gradient of  $B_z$ . The maximum value of  $\sim 5 \times 10^{17}$  m<sup>-3</sup> is therefore located in correspondence of the magnetic throat. In Figure 10(d), it is relevant to notice the distinct transport pattern of a relatively more energetic electron population along the magnetic field lines at the edge of the nozzle, which exhibits  $T_e \approx 3.5 \div 4.2$  eV. A relatively small  $V_p$  is measured throughout the probed region, with a peak value in the order of 10 V in the area of maximum  $n_p$ .

5. when the HP source operates in Conf. E, similar observations to those stated in point (4) can be raised. Profiles of  $n_p$  and  $T_e$  have a strong dependence on the spatial distribution of  $B_r$  and  $B_z$ . The highest plasma density value among all the studied magnetic configurations is herein recorded to be  $\sim 5.5 \times 10^{17}$  m<sup>-3</sup> nearby the magnetic throat. As highlighted in point (4), also the magnetic throat. As highlighted in point (4), also the magnetic dalong the external surface of the magnetic uncertain nozzle, cf. Figure 10(e). At  $z \cong 10$  cm, all the electrons with  $T_e \gtrsim 3$  eV are concentrated in a radius of about 445 3 cm, as visually confirmed by Figure 8(e).



FIG. 12. On-axis profiles of (top) plasma density, (middle) electron temperature and (bottom) plasma potential. Data is extracted from Figures 9-11 and down-sampled to enhance visualization.



FIG. 13. Ion density profile in the proximity of the back-plate for the five magnetic configurations.

<sup>439</sup> Eventually, it is worth observing that the different config-<sup>440</sup> urations of *B* possibly influence the power deposition map. <sup>441</sup> This, in turn, very likely relates to the different spatial distri-<sup>442</sup> butions of  $n_p$  and  $T_e$  as discussed in points (1)-(5).

transported along the external surface of the magnetic transported to the measurements performed using the rfaddition to the measurements performed using the rfaddi

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<sup>448</sup> No significant distortions due to rf oscillations are expected in <sup>449</sup> this part of the I-V curve<sup>33</sup>. For obtaining this particular set <sup>450</sup> of data, the reactor injection plate has been modified in order <sup>451</sup> to accommodate the probe and its related wiring. The pro-<sup>452</sup> pellant is still injected off-axis but it goes through a plenum <sup>453</sup> first and eventually reaches the discharge tube via a multi-<sup>454</sup> perforated surface, resulting in a axial gas flow. An uncom-<sup>455</sup> pensated probe is chosen due to limitations on the available <sup>456</sup> room. The measured profiles of  $n_p$  are shown in Figure 13 <sup>457</sup> for Conf. A to E. Since the magnetic throat in Conf. A is lo-<sup>458</sup> cated upstream the antenna, the ions are mostly transported <sup>459</sup> towards the back plate by following the convergent-divergent <sup>460</sup> shape of the magnetic field. As the magnetic throat moves far-<sup>461</sup> ther downstream from the antenna center, lower densities are <sup>462</sup> measured in the proximity of the injection plate, as expected.

### 463 C. Plasma density profile model

Using a simplified description of the plasma, the measured axial profiles of  $n_p$  can be compared with theoretical profiles retrieved from the helicon wave dispersion relation, which is thereafter derived accounting for the non-uniformity of the external magnetic field along *z*. This approach implicitly assumes that the plasma production entirely relies on the wavemode power coupling. Although the helicon regime is not proven here by direct measurement of the excited wave, previus experiments performed on the same device operated with similar ranges of magnetic field strength and input power have that the electric  $\tilde{\mathbf{E}}$  and magnetic  $\tilde{\mathbf{B}}$  components of the exited wave are described<sup>47,48</sup> as

$$\tilde{\mathbf{E}}, \tilde{\mathbf{B}} \sim \exp[i(m\theta + k_z z - \omega t)],$$

<sup>477</sup> where *m* is the azimuthal mode number,  $\theta$  is the azimuthal <sup>478</sup> angle and  $k_z$  is the axial wavenumber, Maxwell equations as-<sup>479</sup> sume the form

$$\nabla \cdot \mathbf{\tilde{B}} = 0$$
,

(2a) 507

(2b) 509

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 $(2c)^{511}$ 

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(3) 514

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$$\nabla \times \mathbf{\tilde{B}} = \mu_{0} \mathbf{j}.$$

 $\nabla \times \tilde{\mathbf{E}} = i\omega \tilde{\mathbf{B}}$ 

<sup>482</sup> The electron momentum equation is written as

$$-en_{p}\tilde{\mathbf{E}}=i\frac{m_{e}}{e}\mathbf{j}_{e}(\boldsymbol{\omega}+i\boldsymbol{v})-\mathbf{j}_{e}\times\mathbf{B},$$

<sup>483</sup> where  $\mathbf{j}_{e}$  is the electron current density, **B** is the external mag-<sup>516</sup> <sup>517</sup> <sup>484</sup> netic field and  $\nu$  is a collision rate accounting for all dissi-<sup>517</sup> <sup>485</sup> pation phenomena. Additional comments and derivation of <sup>518</sup> <sup>519</sup> <sup>519</sup> <sup>519</sup> <sup>519</sup> <sup>519</sup> <sup>519</sup> <sup>520</sup> <sup>520</sup> <sup>521</sup> <sup>521</sup> <sup>521</sup> <sup>521</sup> <sup>522</sup> <sup>522</sup>

TABLE II. Helicon wave parameters at z = 10 cm.

	Conf. A	Conf. B	Conf. C	Conf. D	Conf. E
<i>r<sub>p</sub></i> [cm]	7.3	8.1	5.1	4.0	3.5
$k_{z}  [\mathrm{cm}^{-1}]$	0.26	0.26	0.26	0.26	0.26
$\lambda_z$ [cm]	24	24	24	24	24
$\kappa_1  [\rm cm^{-1}]$	0.46	0.41	0.68	0.88	1.03
$\kappa_2  [\mathrm{cm}^{-1}]$	0.14	0.17	0.07	-0.01	-0.08

<sup>490</sup> field along *z* is accounted for. Combining Equation 2a-(2c) <sup>491</sup> and Equation 3, one obtains

$$\left(\frac{\boldsymbol{\omega}+i\boldsymbol{v}}{\boldsymbol{\omega}_{c_e}}\right)\nabla\times(\nabla\times\tilde{\mathbf{B}}) + k_z\nabla\times\tilde{\mathbf{B}} + \frac{\mu_0\boldsymbol{\omega}en_p}{B}\tilde{\mathbf{B}} = \mathbf{0}, \qquad (4)$$

<sup>492</sup> where the explicit dependence on *z* has been omitted,  $\omega_{c_e}$  is <sup>493</sup> the electron cyclotron frequency and the rest is conventional. <sup>494</sup> Detailed derivation of Equation 4 is likewise included in<sup>47</sup>. <sup>495</sup> Eventually, in the limit of  $m_e \rightarrow 0$ , the local dispersion relation <sup>496</sup> is retrieved in the same shape of the well known 0D dispersion <sup>497</sup> relation, i.e.

$$k(z)k_z = \frac{\mu_0 \omega e n_p(z)}{B(z)},\tag{5}$$

where k(z) indicates the total wavenumber. The wave bound-499 ary conditions are fixed by<sup>48</sup>

$$mk(z)\mathbf{J}_{m}[k_{r}(z)r_{p}(z)] + k_{z}\mathbf{J}'_{m}[k_{r}(z)r_{p}(z)] = 0, \qquad (6)$$

<sup>500</sup> where  $J_m$  and  $J'_m$  are the Bessel function of the first kind (1) <sup>501</sup> of order *m* and its derivative, respectively,  $k_r(z)$  is the lo-<sup>502</sup> cal radial wavenumber and  $r_p(z)$  is the local plasma rathal <sup>503</sup> dius. The value of  $k_z$  is fixed by the antenna length, as- <sup>504</sup> i.e.  $k_z = \pi/d_A$ ,  $3\pi/d_A$ ,  $5\pi/d_A$ , etc. Pretending that only the <sup>505</sup> first order axial mode is coupled with the antenna, the compu-<sup>506</sup> tation algorithm includes:

- the axial wavenumber is estimated as  $k_z = \pi/d_A$ , therefore the axial wavelenth is  $\lambda_z = 24$  cm;
- r<sub>p</sub>(z) is set equal to the radius of the magnetic streamline crossing the tube edge at z = 0, see Figure 2;
  - considering the geometry of the antenna and the direction of the external magnetic field, it is imposed that m = +1;
- two cases are distinguished:
  - 1.  $k_r(z)$  is computed numerically from Equation 6 and it is named as  $\kappa_1(z)$  for the sake of clarity in the following;
  - 2.  $k_r(z)$  is evaluated as  $\kappa_1(z) \sin[\alpha(z)]$  referred to as  $\kappa_2(z)$  hereafter. Here  $\alpha(z)$  is the local magnetic field divergence angle. The purpose of this heuristic strategy is to provide the model with the information that the applied magnetic field is not

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FIG. 14. Plasma density at r = 0 computed using the helicon waves dispersion relation in the  $k_r(z) = \kappa_1(z)$  and  $k_r(z) = \kappa_2(z)$  cases compared to measurement data.

purely axial. In doing so, it is postulated that the helicon wave resonance cone reduces to a line at the magnetic throat, i.e. where  $\alpha = 0$ , thus k equals  $k_z$ .

• the profile  $n_p(z)$  is consequently computed from Equation 5 for the aforementioned two cases.

Numerical values of the relevant parameters are reported 529 <sup>530</sup> in Table II for all magnetic configurations. In Figure 14, the 531 obtained theoretical curves are compared with measurements  $_{532}$  of  $n_p$  extracted from Figure 9 at x, y = 0 for all the tested  $_{553}$  A further insight into the properties of the plasma when 533 magnetic configurations. In general, it is found that the case 554 operating with the different magnetic topologies is ren $k_{r}(z) = \kappa_{1}(z)$  provides an overestimation of the plasma den- 555 dered by the employment of the PPGR described in Secsity. When dealing with Conf.s D and E, the  $k_r(z) = \kappa_2(z)$  case site tion III. The probe is displaced radially along the x axis sign is in very good agreement with the data in correspondence strain z = 5 cm and z = 15 cm in order to analyze the diver-537 of the magnetic throat, that is where the strong assumption 558 gence of the extracted plasma plume. The measurements of  $_{538}$  **B** =  $B(z)\hat{z}$  is more representative. Ultimately,  $n_p$  measured in  $_{559}$   $j_i$  are shown in Figure 15(a-b). Table III gathers the val- $_{539}$  Conf. C is in relatively good agreement with the latter case.  $_{560}$  ues of the divergence parameter  $\delta$  computed as the ratio s40 It is noted that even the  $k_r(z) = \kappa_2(z)$  case overestimates the s61  $j_i(x = -11.5 \text{ cm})/j_i(x = 0 \text{ cm})$ . Results confirm that a low-<sub>541</sub> measured values of  $n_p$  for Conf.s A and B, although match- <sub>562</sub> current high-divergence ion beam is extracted from the HP 542 ing the qualitative profile. This suggests that the helicon wave 563 source when operating in Conf. A and B. Differently, the ion 543 is evanescent in this region of the plasma since the measured 564 beams related to Conf.s C, D and E feature a similar low- $_{544}$   $n_p$  is lower than the one required by the dispersion relation.  $_{565}$  divergence profile of  $j_i$  in the proximity of the tube exit, see 545 In conclusion, these profiles of  $n_p$  computed from the disper- 566 Figure 15(a). Yet, the divergence angle of Conf. C rapidly 546 sion relation represent a first effective estimation of the actual 567 grows as the plasma expands downstream, cf. Figure 15(b). 547 spatial evolution of plasma density. Therefore, these quanti- 568 The highest value of  $j_i$  is recorded in Conf.E at z = 15 cm, tative information can be of some convenience for guessing  $_{569}$  in agreement with the largest  $n_p$  probed in the same configu-<sup>549</sup> the propagation region of helicon waves and for preliminary <sup>570</sup> ration. Larger values of  $j_i$  relate well with higher propellant <sup>550</sup> modeling of the plasma dynamics and/or design of the axial <sup>571</sup> utilization efficiency, which is a desirable working condition <sup>551</sup> magnetic field profile.



FIG. 15. Ion current density at (a) z = 5 and (b) z = 15 cm for the five magnetic configurations.

TABLE III. Plume divergence parameter  $\delta$  from data in Figure 15.

	Conf. A	Conf. B	Conf. C	Conf. D	Conf. E
$\delta$ at $z = 5$ cm	0.218	0.359	0.041	0.014	0.021
$\delta$ at $z = 15$ cm	0.523	0.567	0.224	0.028	0.035

### 552 D. PPGR measurements

572 for space propulsion applications.

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TABLE IV. Order of magnitude of relevant velocity quantities.

	$\mathscr{O}(\mathbf{v})$
Ion acoustic velocity	$v_B = \sqrt{kT_e/m_i} \approx 2000 \text{ m s}^{-1}$
Thermal velocity	$v_{th} = \sqrt{8kT_i/\pi m_i} \approx 275 \text{ m s}^{-1}$

### 573 **E.** LIF measurements

574 Using LIF spectroscopy, the velocity distribution functions 575 of krypton atoms Kr I and ions Kr II are inferred. Infor-576 mation about the velocity of the ions is of particular inter-577 est for driving conclusions about propulsive performance. On 578 the other hand, knowledge of the atoms velocity is useful for 579 modeling of the plasma flow. Table IV exemplifies the or-580 der of magnitude of the relevant velocity terms in the HP <sup>581</sup> source operating with Kr. Radial profiles of Kr I VDF are see evaluated at z = 12 cm and shown in Figure 16 for all mag-<sup>583</sup> netic configurations. The spatial resolution is 10 mm. In this 584 case, the reported data did not require any post-processing as <sup>585</sup> a adequate signal-to-noise ratio was found. These measure-<sup>586</sup> ments show that the most probable velocity of Kr I is nearly <sup>587</sup> unchanged over the different magnetic topologies, reading  $_{588} \sim 250 \div 300 \text{ m s}^{-1}$ , therefore in the order of the thermal ve-<sup>589</sup> locity assuming  $T_i \approx 300$  K, cf. Table IV. No dependence of <sup>590</sup> the velocity on the radial location y is recorded. Dissimilar <sup>591</sup> values of velocity spread are found between the magnetic con-<sup>592</sup> figurations, with a broader VDF for Conf. C and Conf. D.

Axial VDFs of Kr II are probed in the same radial region 593 with a spatial resolution of 5 mm at three distinct axial posi-594 <sup>595</sup> tions, i.e. z = 10.5 cm, z = 12 cm and z = 25 cm. At some <sup>596</sup> y locations, the LIF signal was not distinguishable from the <sup>597</sup> background noise due to small local values of  $n_p$ , therefore <sup>598</sup> mapping of the VDFs was not possible. For the very same <sup>599</sup> reason, Conf.s A and B did not allow obtaining an appropri-600 ate signal-to-noise ratio in any of the probed locations. Raw VDFs are post-processed using a least squares Gaussian fit. 601 <sup>602</sup> The resulting profiles when the HP source operates in Conf.s 603 C, D and E are reported in Figure 17, 18 and 19, respectively. <sup>604</sup> In addition, the most probable velocity is extracted and plot <sup>605</sup> in Figure 20 thus to facilitate comparison among the various <sup>606</sup> magnetic configurations. It is noticed that the profiles related 607 to Conf.s C and D feature a nearly identical trend, whereas the <sup>608</sup> ion velocity in Conf. E generally retains lower readings. This  $_{609}$  is in agreement with the measured values of  $V_p$  which remain 610 below those concerning Conf. C and D, cf. Figure 11. In the <sub>611</sub> further upstream regions, namely z = 10.5 cm and z = 12 cm, velocities smaller than  $\sim 250 \text{ m s}^{-1}$  are measured, therefore in 612 <sub>613</sub> the order of the thermal velocity assuming  $T_i \approx 300$  K. Con- $_{614}$  versely, at z = 25 cm, all the velocities increase by a factor of  $_{622}$  size of the vacuum chamber. Similar small values of ion ve-616



FIG. 16. Kr I velocity distribution function for the five magnetic configurations measured along the y axis at z = 12 cm.



FIG. 17. Kr II velocity distribution function for Conf. C measured along the y axis at z = 10.5, 12 and 25 cm.

 $_{615}$  2 to 3, reaching values as high as ~ 500 m s<sup>-1</sup>. These small  $_{623}$  locity are reported in Ref. 49 using LIF on a large scale HP velocities somewhat concur with the fact that the magnetic  $_{624}$  source. In all of the magnetic configurations, the drop of  $V_p$  $_{617}$  field is weakly divergent in the probed region for Conf.s C and  $_{625}$  is fairly small (in the order of 1 V) within z < 21 cm, cf. Fig- $_{618}$  D, while z = 25 cm represents the throat location for Conf. E.  $_{626}$  ure 11. Considering Conf. E for instance, since at z < 25 cm 619 A weakly divergent magnetic field does not allow for a signifi- 627 the magnetic field is convergent, it is expected that the favored 620 cant momentum gain imparted by the Lorentz force. Measure- 628 inward transport of electrons do not allow for a significant po-621 ments farther downstream were prevented due to the limited 629 tential drop in this region. In the purely electrostatic case,

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FIG. 18. Kr II velocity distribution function for Conf. D measured along the y axis at z = 10.5, 12 and 25 cm.



FIG. 19. Kr II velocity distribution function for Conf. E measured along the y axis at z = 10.5, 12 and 25 cm.

 $_{631}$  crease of  $\sim 1.5 \text{ km s}^{-1}$ . Hence, the acceleration of the ions  $_{662}$  the mechanisms occurring within the nozzle. The exploited <sup>632</sup> throughout the MN does not fully exploit the available en- <sup>663</sup> parameters are: input power (750 W and 900 W); mass flow  $_{633}$  ergy. This phenomenon is related to some dissipation phe- $_{664}$  rate (0.5 mg s<sup>-1</sup>, 1 mg s<sup>-1</sup> and 2 mg s<sup>-1</sup>); current in the elec-634 nomenon, e.g. collisions, as later discussed. Moreover, al- 665 tromagnets (6 A and 8 A). The magnetic configuration is fixed 635 though no sonic condition is achieved at the magnetic throat 666 to Conf. D for this experiment because of its divergent shape  $_{636}$  in Conf.s C and D, ions experience a further acceleration in  $_{667}$  at z = 25 cm and larger signal-to-noise ratio with respect to 637 the divergent part of the nozzle, see Figure 20. It suggests that 668 Conf. C. The resulting VDFs are shown in Figure 21, which 638 the sonic point, if ever reached, is shifted downstream, in con- 669 includes the raw data, the Gaussian fittings and the unfold 639 trast with the ideal postulation of gas-dynamics-like plasma 670 most probable velocities obtained at each working condition. 640 expansion<sup>50</sup>. In this experiment, the ions might reach the 671 The overall tendency results clear. Indeed, higher ion veloci-<sub>641</sub> sonic speed only at the sheath edge that forms downstream <sub>672</sub> ties are attained as the mass flow rate decreases and the mag-<sup>642</sup> between the chamber wall and the plasma plume. Numer- <sup>673</sup> netic field is increased, whereas slower ions are recorded with <sup>643</sup> ous works focusing on modeling of magnetic nozzle dynam- <sup>674</sup> larger mass flow rates. These outcomes suggest that collisions <sup>644</sup> ics have assumed that the ion Mach number equals unity at <sup>675</sup> resulting from a larger pressure do play a role in the acceler-<sup>645</sup> the throat<sup>51–53</sup>. Albeit this condition conveniently allows sep- <sup>676</sup> ation process. At lower pressures, a larger mean free path is  $_{646}$  arating the processes of plasma generation and plasma accel- $_{677}$  expected, therefore allowing for greater values of  $T_e$ .



FIG. 20. Kr II most probable axial velocity extracted from Figure 17-19. Error bar is  $60 \text{ m s}^{-1}$  (typical instrument absolute error).

647 eration, it is not representative in this case as well as in some other plasma beams<sup>24,54</sup>. Collard *et al*<sup>24</sup> have examined a list <sup>649</sup> of phenomena which are potentially responsible for separation 650 of the magnetic throat and the sonic point, namely electron-<sup>651</sup> neutral collisions, charge-exchange collisions and ionization 652 within the plume. In the device herein analyzed, a combina-<sup>653</sup> tion of several dissipative phenomena is likely to contribute in 654 introducing a drag force term that prevents the ions to reach 655 the sonic speed. In Conf. C, for instance, a significant level 656 of ionization is localized in the near-field plume downstream 657 the magnetic throat, as previously noted when commenting on <sup>658</sup> Figure 9. This suggests that the energy stored in the electrons 659 is spent for ions generation rather than acceleration.

A parametric evaluation of the Kr II axial VDF is addition-660  $_{630}$  a potential drop of  $\sim 1$  V translates into an ion velocity in-  $_{661}$  ally performed at z = 25 cm for the purpose of better grasping

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FIG. 21. Kr II axial VDF for Conf. D at z = 25 cm as function of  $P_{IN}$ ,  $\dot{m}$  and electromagnets current. Scatter points indicate raw data, solid lines exemplify least squares Gaussian fittings.

### 678 V. CONCLUSION

In summary, several plasma properties are inferred via di-679 680 rect measurement to establish empirical approaches to thruster 725 681 design and optimization. The magnetic throat is displaced 726  $_{\rm 682}$  along the reactor axis and the resultant plasma is studied in  $^{727}$ 728 683 the near-field plume using electrostatic diagnostics and LIF 729 <sub>684</sub> spectroscopy. It is found that a low-current high-divergence <sub>730</sub> <sup>685</sup> plume is extracted from the plasma source when the magnetic <sup>731</sup> 686 throat is located upstream or in correspondence of the antenna 732 687 center point. Plasma density measurements at the back plate 734  $_{588}$  emphasize that, in those cases, the generated plasma mainly  $_{735}$ <sup>689</sup> remains within the discharge tube and is lost at the radial and <sup>736</sup> <sup>690</sup> back boundaries. Larger ion currents and higher confinement <sup>737</sup> <sup>691</sup> levels are attained when the magnetic throat is located down-<sup>692</sup> stream the antenna, namely at a distance at least equal to the 740 <sup>693</sup> wavelength corresponding to the first order axial mode of the 741 694 helicon wave. Hence, optimization of the plasma generation 742 <sup>695</sup> process in a Helicon thruster would include the separation of <sup>743</sup> 744 <sup>696</sup> the magnetic throat from the antenna location in accordance 745 <sup>697</sup> with the excited helicon wavelength. 746

Theoretical axial profiles for the plasma density are directly derived from the dispersion relation under the hypothesis of non-uniform magnetic field. Comparison with probe meausurements show relatively good agreement therefore hinting the applicability of this theoretical description for guessing the applicability of this theoretical description for guessing

the helicon wave propagation region. Furthermore, preliminary modeling and design of Helicon thrusters can perhaps
make use of this approach to shape the axial magnetic field
profile.

The measured ion axial velocities exhibit small values, beros ing below the sonic condition at the magnetic throat. By mearos suring the most probable ion velocity obtained under differron ent working parameters (input power, magnetic strength and run mass flow rate), it is concluded that ionization and collisions roccurring in the near-field plume contribute to limiting the ion run acceleration process.

### 714 ACKNOWLEDGMENTS

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### 719 CONFLICT OF INTEREST

The authors have no conflicts to disclose.

### 721 DATA AVAILABILITY

The data that support the findings of this study are available from the corresponding author upon reasonable request.

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