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► **To cite this version:**

Antoine Soulier, Caroline Braud, Dimitri Voisin, Danbon Frédéric. High Reynolds investigations on the ability of the full scale e-TellTale sensor to detect flow separation on a wind turbine blade section. 2021. hal-03437712

**HAL Id: hal-03437712**

**<https://hal.science/hal-03437712>**

Preprint submitted on 20 Nov 2021

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# High Reynolds investigations on the ability of the full scale e-TellTale sensor to detect flow separation on a wind turbine blade section

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**Abstract.** The complexity of the flow over a wind turbine blade makes its understanding and monitoring a challenging task, especially on operating wind turbines. The innovative e-Telltale sensor is developed for that purpose : detecting the flow separation on wind turbine blades. In this paper, high Reynolds wind tunnel tests have been performed with different configurations of full scale e-Telltale sensors and wall pressure measurements on a wind turbine blade section. A comparison between the lift curve and the e-Telltale signal was used to evaluate the ability of the sensor to detect flow separation. Results show different interesting properties of the sensor response depending on its size, position along the chord and its fitting process that could be used in real applications.

## 1 Introduction

Increasing the life of wind turbines is one of the major areas of investigation faced by wind farm operators. A cause of premature aging often put forward is the accumulation of loads imposed by the strong shears upstream of the rotor due to a malfunction of the wind turbine or to the atmosphere in which it operates (Rezaeiha et al., 2017). In order to limit the influence of these disturbances on the wind turbine, modern pitch-regulated wind turbines are operational today. Sensors currently used are located on the wind turbine nacelle such as cup anemometers (Smaili and Masson, 2004), on the wind turbine spinner (Pedersen et al., 2015). Upstream flow measurements from a LIDAR mounted nacelle are also under development (Scholbrock et al., 2013). The former sensor will predict flow perturbations (gust, misalignment ...) before a control action of the blade is performed. The control objective is to alleviate turbine blade lift fluctuations and resulting load fluctuations by an adequate rotation of the blade incidence. However, measurements at the wind turbine nacelle do not sufficiently take into account the state of the flow on the aerodynamic surfaces (attached/separated, laminar/transitional or turbulent aerodynamic boundary layer ...) which is critical to decide a control action. Also, pitch control on very large blades is not fast enough to account for the small time scales from small turbulent structures generated in a wind turbine wake for instance (Chamorro et al., 2012), while they have a strong impact on blade loads (Bartholomay et al., 2018). Having local and robust aerodynamic sensors at the blade scale placed at key areas would be an important step for wind turbine monitoring and operation. Furthermore, they could be used together with active devices to further decrease local spatio-temporal loads (Shaqarin et al., 2013; Jaunet and Braud, 2018). Swytink-Binnema and Johnson (2016) have demonstrated the possibility to detect aerodynamic flow separation



25 using distributed tufts over the blade surface and a root-blade embedded camera. A simple and robust alternative of this sensor  
is the use of electronic-Telltale sensors. The system is composed of a silicon strip with a strain gauge at its base. When the  
silicone strip goes away from the surface, the displacement is measured by the strain gauge. The e-Telltale sensor is already  
used on sails of some boats and can be glued on the aerodynamic surface of wind turbine blades for retrofitting purposes.  
First tests of this innovative sensor were conducted at low Reynolds number using a down-scaled device (Soulier et al., 2021).  
30 The strip of the e-Telltale was demonstrated to follow the separation/reattachment dynamics similarly as much more accurate  
detection methods based on Time Resolved PIV measurements. The present paper extends the investigation on the ability of the  
e-Telltale sensor to detect the flow separation over airfoil profiles towards the use of a full scale device and high Reynolds wind  
tunnel tests (the chord based Reynolds number is  $Re_c = 8.85 \cdot 10^5$ ). The same 2D blade section was used here (NACA-65<sub>4</sub>-421)  
and different parameters of the e-Telltale device were investigated, including its position, the strip length and its surface fitting  
35 process. The evaluation has been performed at different angles of attack and through measurements of three lines of chord-wise  
pressure taps acquired synchronously with the strain gauge signal of the e-Telltale sensor.

The paper is divided in three main sections. Section 2 describes the experimental set-up, including the description of the  
wind tunnel facility, the blade manufacturing, the measurements used for the evaluation of e-Telltale sensor and the description  
of the different e-Telltale configurations. Section 3 describes the aerodynamics of the chosen blade profile, focusing on the flow  
40 separation phenomena. The last section present the results including the impact of the strip location, length and fitting process  
on the sensor signal.

## 2 Experimental Setup

### 2.1 The Wind tunnel

The measurements were performed at Nantes (France), in the NSA return wind tunnel of CSTB<sup>1</sup>. The test section is 20m  
45 long with a cross-section of 4 m x 2 m (see figure 1). The turbulence intensity level in this test section is around 1 % and the  
operating speed of the wind tunnel is set to  $U_\infty = 20 \text{ m/s}$ . At this free stream velocity, the Reynolds number of the flow based  
on the chord length ( $c = 0.693\text{m}$ ) is  $8.85 \cdot 10^5$ . The profile was set on a rotating table on its bottom, and guided on the ceiling  
using a bearing. Three pressure lines were used at different span locations, where measurements were performed, to check the  
flow bidimensionality in the area of interest (see section 2.3).

### 50 2.2 Blade manufacturing

The chosen 2D blade section, NACA 65<sub>4</sub>-421, was manufactured to be installed in the wind tunnel facility of CSTB. It was  
made of a skeleton coated with a supple composite. Details on the blade manufacturing can be found in figure 2. The shape  
of this airfoil profile, already used in previous studies (Sicot et al., 2008; Devinant et al., 2002), is also used in operation on  
stalled regulated wind turbines.

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<sup>1</sup><http://www.cstb.fr/en/>



**Figure 1.** The NSA wind tunnel facility of CSTB with the 2D NACA 654-421 blade section installed

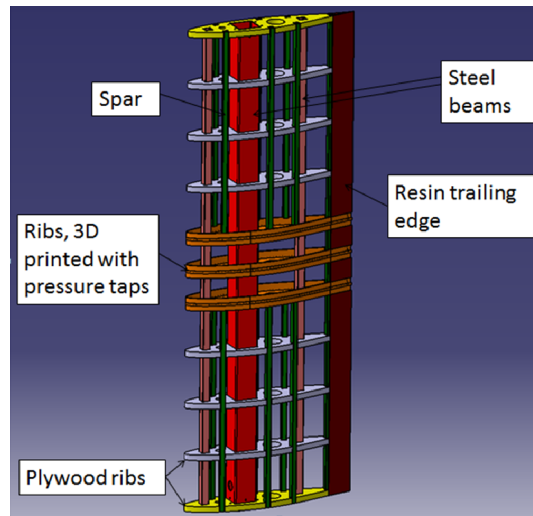
## 55 2.3 Measurements

The framework of the present study is  $x$  in the streamwise direction,  $y$  is the cross direction and  $z$  in the spanwise direction perpendicular to the chord line. The origin is taken at the intersection between leading edge and the chord line, in the middle of the blade span (see figure 3) To get the lift coefficient  $C_L$ , three chord-wise lines of pressure taps were distributed around the profile using three 3D printed ribs equipped with 117 pressure sensors each. They were located in the middle of the profile at  $z = 0$ , and  $z \pm 0.173c$  (see figure 3). Copper tubes of 0.8mm internal diameter were flush mounted using pressure tap holes in the 3D printed ribs. Vinyl tubes were then connected to transport the pressure towards ESP 32HD pressure sensors ranging from 0 to 2.5 kpa with a precision of  $\pm 0.03\%$  of the full scale. The cut-off frequency of the total system (tubes plus sensors) was 256 Hz. The signal was low-pass filtered at 256 Hz and acquired at 512 Hz. For each  $AoA$ , the pressure coefficient,  $C_P$ , was calculated and averaged on the duration on the measurement which was either 1 or 2 minutes, then the lift coefficient  $C_L$  was computed for each line of pressure taps by integration. It has been checked that the statistical convergence of  $C_L$  is reached well before 1 minute. The lift coefficient presented in this study are the results of an average over the three lines of pressure.

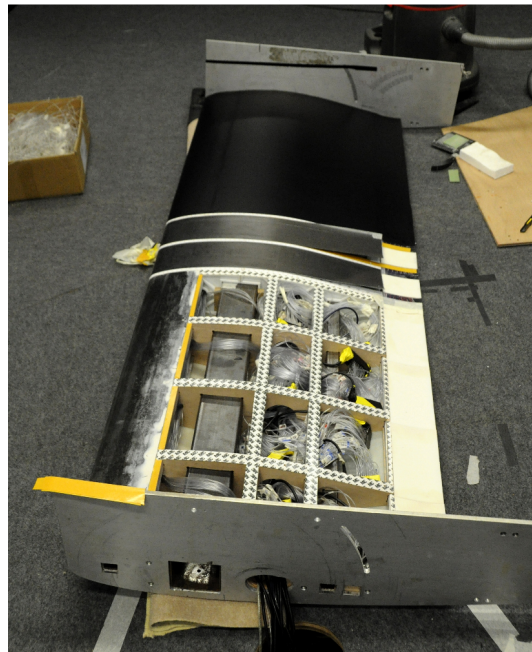
## 2.4 E-Telltale sensors

E-Telltale sensors are composed of a silicon strip with a strain gauge sensor at its base, so that it measures displacements of the strip away from the surface (see figure 4). The signal from the strain gauge sensor was amplified with a specialized amplifier (LTC6915). The signal is then filtered with a low-pass analog filter with a cut-off frequency of 160 Hz and a slope of  $-20 \text{ dB/decade}$ . The filtered signal was then recorded with the same acquisition device than for the pressure measurements.

In the present study, which is a first high Reynolds wind tunnel evaluation of this innovative e-Telltale sensor, we focuses on the ability of the e-Telltale sensors to detect mean load variations highlighted by the lift coefficient versus the angle of incidence and related to the flow separation over the suction side surface of the blade described in section 3. Different parameters of the e-Telltale sensor were investigated and summarized in tabular 1. The first question that arises for such local sensors is where

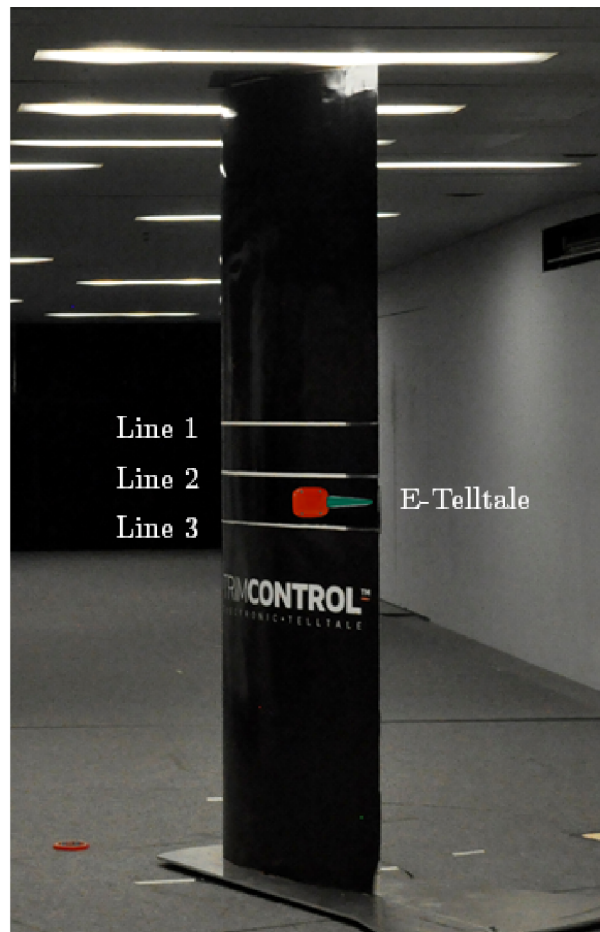


a)

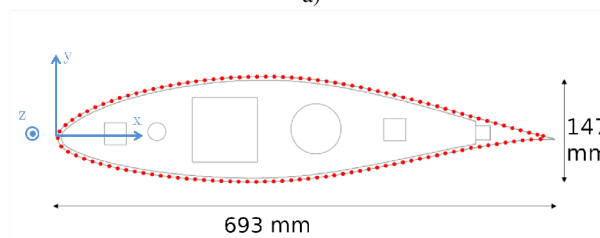


b)

**Figure 2.** The blade manufacturing. a) Elements of the skeleton: 2 steel beams ensure the rigidity, Plywoods ribs are ensuring the blade aerodynamic shape, 3D printed ribs are equipped with pressure taps, The trailing edge is made of resin. b) Picture of the blade with one of the GRP skins removed.

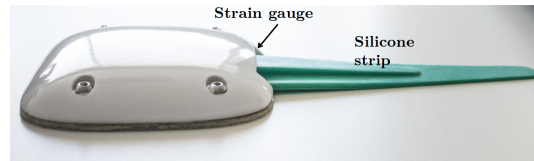


a)



b)

**Figure 3.** a) Positions of the chord-wise pressure lines along the span b) positions of pressure taps around the chord



**Figure 4.** Picture of the innovative E-Telltale sensor (Shell and long strip cases of Table 1).

Name	Short Name	Location	Shape	Size ( $L/c$ )
TE-Shell-Long	TESL	Trailing edge (see figure 5)	with shell	0.19
TE-NoShell-Long	TENSL	Trailing edge (see figure 6)	without shell	0.19
LE-NoShell-Long	LENSL	Leading edge (see figure 6)	without shell	0.19
TE-NoShell-Short	TENSS	Trailing edge (see figure 7)	without shell	0.11

**Table 1.** e-Telltale configurations (with  $L$  the length of the silicon strip and  $c$  the chord of the profile)

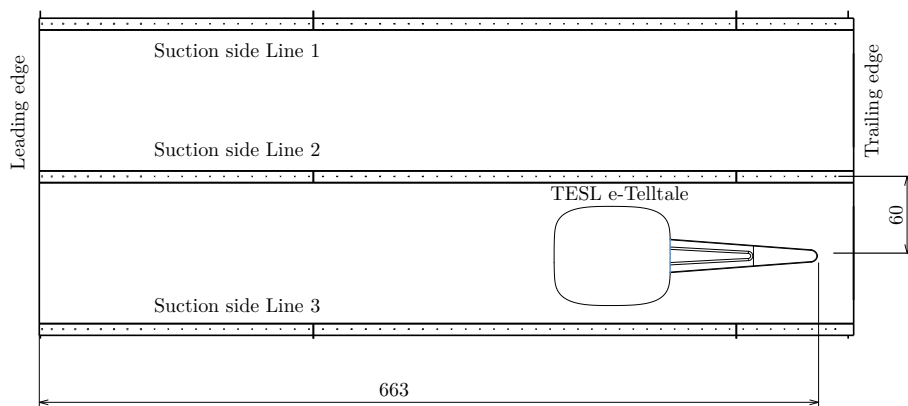
to place them on operating turbines. For this reduced problem (2D blade model), this question can be summarized as follows: which sensor positions are the best to detect the angles at which separation occurs ? Two locations will be targeted as this blade profile shape has two types of flow separation (see section 3 for more details): the flow separation at the trailing edge and the flow separation at the leading edge. Therefore, the end of the strip is first located at 95.6% from the leading edge, the

80 Trailing edge or TE case, and at 31.8% from the leading edge, the Leading-edge or LE case (see figure 5). This last position is slightly modified for shorter strips so that the end of the strip is at 87.6% from the leading edge rather than 95.6% (see figure 5). Another question we target to answer in this study is the ability of this device to detect flow separation for two different surface fitting processes. The first e-Telltale is equipped with an aerodynamic shell that is glued on the surface, the Shell or S case. The second e-Telltale is integrated in the surface of the airfoil so that the surface of the airfoil is less perturbed, the

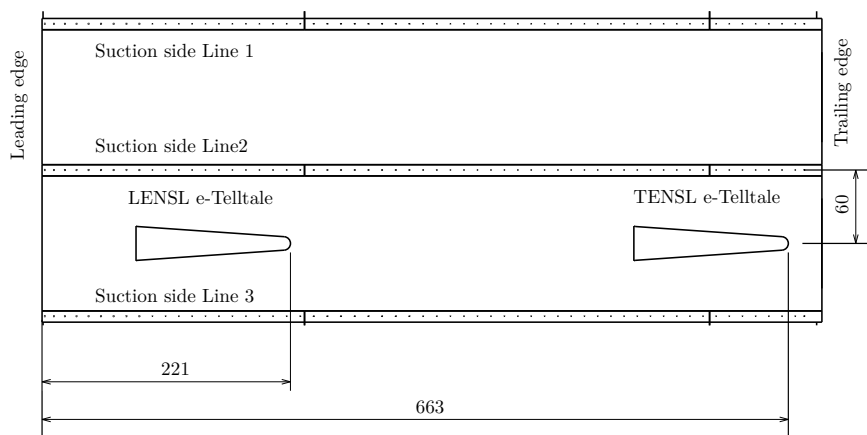
85 NoShell or NS case. The shell case is generally mounted on wind turbines already in operation, while the NoShell case could be part of the blade manufacturing process. Finally, the more rigid the strip is, the less signal is transmitted on the strain gauge. As a first evaluation of this parameter, the length of strip is divided by almost two by keeping only the central part of the strip that is thicker (see figure 4 and 7). Between the two cases, the surface/thickness ratio is modified from 921  $mm$  to 218  $mm$ . The two cases will be referred later as the long (or L) and short (or S) strip.

### 90 3 Blade aerodynamics

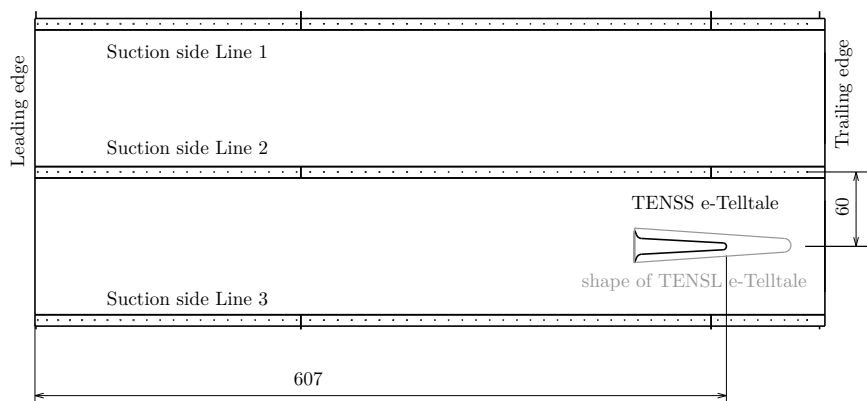
The NACA-65<sub>4</sub>-421 profile has different slope modifications of the static lift coefficient corresponding to different states of flow on the suction side surface of the airfoil. This profile was already studied in the work of (Devinant et al., 2002) and a modified version (rounded trailing edge) of this NACA profile was used in the ANR (French national grant) project SMARTEOLE (Leroy, 2018; Braud and Guilmineau, 2016; Baleriola et al., 2018; Jaunet and Braud, 2018).



**Figure 5.** Position of the TESL e-Telltale



**Figure 6.** Positions of LENS and TENS e-Telltales



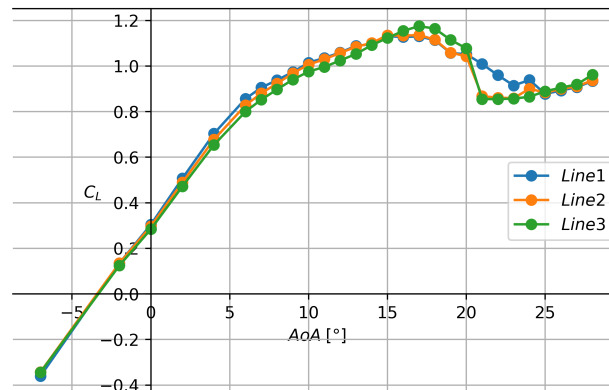
**Figure 7.** Position of TENS e-Telltale





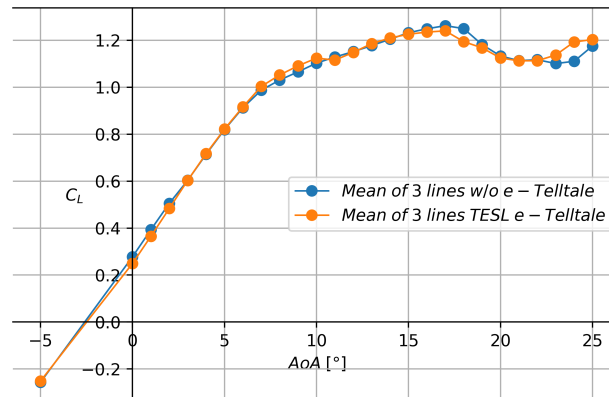
95 Measurements of the present study were performed with a turbulence intensity of 1%, which may be at the origin of some  
discrepancies in the exact values of the separations angles. However, similarly to previous studies, this profile shape presents  
different states of the flow depending on the angles of incidence that have an impact on the shape of the lift curve and on the  
chord-wise pressure distribution. These flow states correspond to a progressive displacement of the flow separation from the  
trailing edge to the leading edge, until the stall occurs. For the present study we can identify the following angles of attack (see  
100 figure 8):

- Until  $AoA \simeq 6^\circ$ , the lift rises linearly with the angle of attack, the flow is attached to the surface of the profile.
- Between  $AoA \simeq 6^\circ$  to  $8^\circ$ , the flow is transitioning from the separated state to the attached state, as can be seen on  
the trailing edge suction side of the airfoil from the intermittent apparition of a plateau on the pressure distribution  
(corresponding to a zero gradient pressure).
- 105 – From  $AoA \simeq 8^\circ$  to  $18^\circ$ , the flow separation can now be observed in average and move progressively towards the leading  
edge. This corresponds to a linear evolution of the lift with the angle of attack, with however a smaller slope than the  
previous flow state.
- From  $18^\circ$  to  $AoA \simeq 20^\circ$ , the evolution of the separation point towards the leading edge is faster and not always linear,  
the flow is transitioning towards stall.
- 110 – Over  $AoA \simeq 20^\circ$  the separation point has reached the leading edge, the flow is stalled and the wall pressure is constant on  
the whole suction side. The flow behaves like an asymmetric bluff body with shear layers on each side of the profile (from  
the leading edge and the trailing edge), a recirculating area in the close wake, and a wake behavior further downstream.



**Figure 8.** Lift coefficient from the 3 lines of pressure taps

Figure 8 demonstrates that the flow remains 2D between  $z = 0.173c$  (line 1) and  $z = -0.173c$  (line 3) until  $AoA \simeq 20^\circ$ , where the stall occurs and thus the flow becomes 3D. Even for these large angles of attack, lines 2 and 3 present similar



**Figure 9.** Lift coefficient without e-Telltale and with th TESL e-Telltale

115 pressure distributions. In the following, the strain gauge signal of the e-Telltale will be compared to the average of these lift  
coefficient lines for different values of the angle of attack. It is thus possible to evaluate the ability of the e-Telltale sensor to  
detect the trailing edge separation angle as well as the stall separation angle. Figure 9 presents the lift coefficient measured  
without any e-Telltale and with an TESL e-Telltale. There is not any significant difference seen between these two coefficients,  
showing that the e-Telltale have not any influence on the time average lift coefficient.

## 120 4 Results

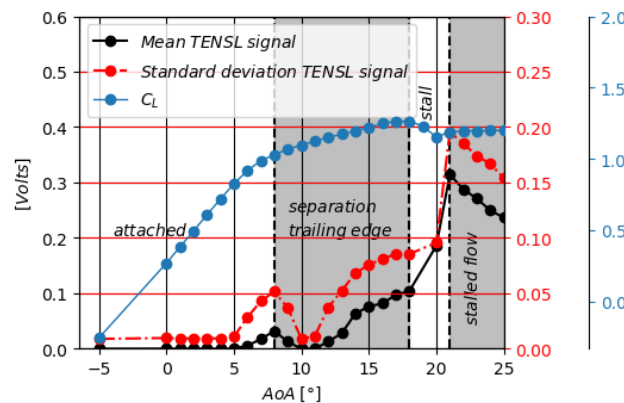
During these experiments the attention was drawn to evaluate the ability of the e-Telltale to detect the slopes changes on  $C_L$   
corresponding to the flow separation at the trailing edge ( $AoA \simeq 8^\circ$ ) and the leading edge flow separation, corresponding to  
the stall angle ( $AoA \simeq 21^\circ$ ) as explain in section 3. Each modification of the lift slope corresponds to an evolution of the flow  
separation state, which ideally induces a different movement of e-Telltale strip and thus a difference in the strain gauge signal  
125 of the e-Telltale device. In this section, the time average strain gauge signal and the standard deviation are computed for each  
angle of incidence and presented together with the lift curves.

### 4.1 The reference case (TENSL)

In order to compare different parameters of the e-Telltale (as presented in the tabular 1) the TENSL configuration (near the  
trailing edge, no shell, long strip) was selected as the reference case. The figure 10 shows the evolution of the mean and the  
standard deviation of the signal of the e-Telltale corresponding to the TENSL case as function of  $AoA$ . The lift curve is plotted  
130 together to allow us to locate the state of the flow with the angle of incidence which are recalled in the figure. From  $-5^\circ$  to  $5^\circ$ ,  
the linear evolution of the lift indicates an attached flow state. For this  $AoA$  range, both the mean and the standard deviation of  
the signal of the e-Telltale are near 0. From  $5^\circ$  to  $8^\circ$ , corresponding to the flow separation apparition at the trailing edge, both



the mean and the standard deviation rise up linearly with the  $AoA$ . Then from  $8^\circ$  to  $10^\circ$  they decrease linearly to values near  
 135 0. Interestingly, the e-Telltale has a particular sensitivity to the transition flow state from the fully attached flow to the trailing  
 edge flow separation. The advantage of this observation is to make this innovative sensor appropriated to predict the trailing  
 edge flow separation, at least for this type of profile. At  $AoA > 10^\circ$ , both the mean and the standard deviation of the e-Telltale  
 signal increases again until  $AoA = 18^\circ$ . Contrary to the  $AoA$  range corresponding to the apparition of the trailing edge flow  
 separation, between  $AoA = 5^\circ$  to  $AoA = 8^\circ$ , the evolution is not linear. A progressive saturation of the signal appears slowly.  
 140 This may be explained by the displacement of the separation point and its associated shear layer, that are moving further away  
 from the sensor with the increase of the  $AoA$ . Between  $AoA = 18^\circ$  and  $AoA = 20^\circ$  the e-Telltale signal is marked by a sudden  
 rise in two steps. A first moderate step between  $18^\circ$  and  $20^\circ$  and a sudden rise between  $20^\circ$  to  $21^\circ$ . This can be explained by  
 the strip that is flipping towards the leading edge due to the strength of the reverse flow after  $20^\circ$ , in good agreement with low  
 Reynolds number experiments performed by Soulier et al (Soulier et al., 2021). After the stall angle,  $AoA = 21^\circ$ , the mean  
 145 signal reaches its maximum value while the standard deviation has doubled. This indicates that the strip is not only flipping in  
 the reverse flow in average, but it is also fluctuating in the flow with larger oscillation than when the stall has occurred, again  
 in good agreement with low Reynolds observations (Soulier et al., 2021). The significant rise which occurs at the stall angle  
 allows a clear detection of the stall angle with the e-Telltale sensor, even though not ahead its apparition. After this  $AoA$  of  $21^\circ$ ,  
 the stalled flow or the bluff-body flow is settled, both the mean and the standard deviation of the e-Telltale signal are linearly  
 150 decreasing, probably due to the deportation of the shear layer further away from the location of the e-Telltale.



**Figure 10.** Mean and Standard deviation of the e-Telltale TENSLE signal and lift  $C_L$  as function of  $AoA$

#### 4.2 Influence of the strip stiffness or length (TENSS case)

Compared to the previous case (section 4.1), the TENSS case presents a shorter strip length, reduction of 58% ( $L_{TENSLE}/c = 0.19$  and  $L_{TENSS}/c = 0.11$ ), and a more rigid strip measured through the surface-thickness ratio reduction of 77% (from



155  $A_{TENS L}/T_{TENS L} = 921 \text{ mm}$  to  $A_{TENS S}/T_{TENS S} = 218 \text{ mm}$ ), which parameter is inversely proportional to the stiffness of the strip. The short strip is therefore much more stiff than the long one.

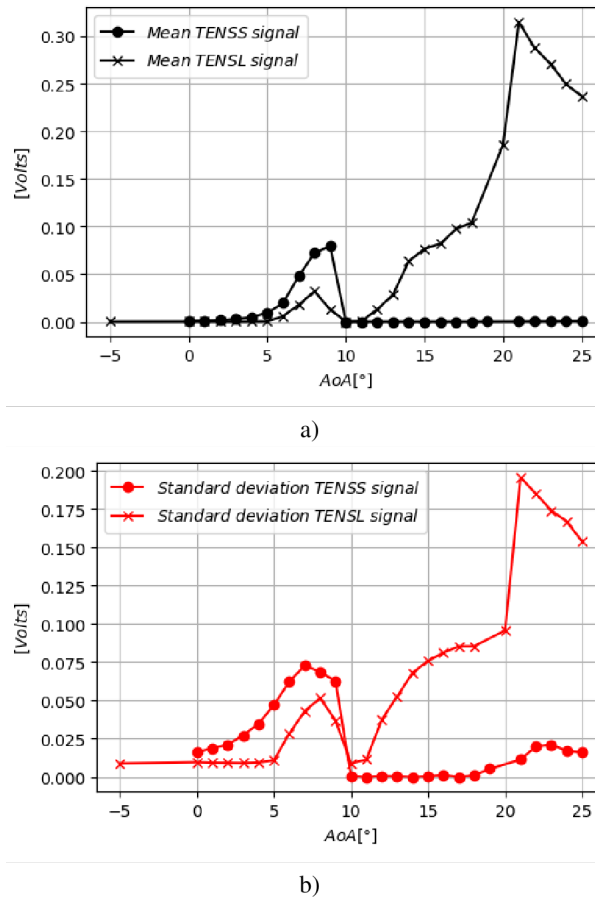
The e-Telltale is located at the same position of the reference TENS L case (the root of the strip at around 77% of the chord). For this test, the same device is used but the strip has been cut, ensuring that the strain gauge and the position of the strip are identical to the reference case. The figures 11a and 11b show respectively the mean and standard deviation of the TENS S case signal versus the angle of attack, plotted together with the reference case. It clearly shows that there is almost no information  
160 about the stall with the shorter strip case (i.e. for  $AoA > 19^\circ$ ), only a slight increase of the standard deviation value with a peak around  $21^\circ$  is observed (see figure 11b). On the contrary, for angles of attack corresponding to the apparition of the flow separation at the trailing edge, from  $AoA = 5^\circ$  to  $AoA = 8^\circ$ , the mean signal is 1.5 higher than the reference case (longer strip case). Also, the mean and standard deviation of the signal suddenly drops at  $AoA = 9^\circ$ . Having a higher sensitivity of the strain gauge signal for the shorter strip case which is stiffer, highlights that the length of the strip is the most relevant parameter  
165 for the detection of trailing edge separation phenomena of this profile. For the stall phenomena it seems that the reversed flow close to the wall is not the relevant phenomena at the origin of the displacement of the strip, but rather the separated shear layer and its distance to the wall. However, further spatio-temporal explorations are needed to investigate this point. For further understanding of full scale blade aerodynamics, the high sensitivity of the short sensor case to the apparition of the trailing edge separation and not the stall, makes it really interesting to discriminate them.

#### 170 4.3 Influence of the position of the e-Telltale (LENS L)

The influence of the position of the e-Telltale is discussed in this section. The e-Telltale located near the trailing edge in the area of high favorable pressure gradient, the LENS L case, is compared to the reference case, at the trailing edge (see figure 6). When located at the leading edge, the e-Telltale is clearly unable to detect the angle of incidence corresponding to the trailing edge flow separation,  $AoA = 8^\circ$ . On the contrary, both the mean and the standard deviation of the signal starts to rise just after  
175 the second linear part of the lift,  $AoA = 18^\circ$ , with a stronger slope than when the e-Telltale is located at the trailing edge. It leads to a higher level of the signal, more than doubled for the mean signal, and around 50% higher for the standard deviation. Also, the peak of the signal occurs at higher angles of incidence than when the sensor is located at the trailing edge. All these observations are in good agreement with the fact that the stall phenomena is originated at the leading edge, with a separated shear layer that remains, for the sensor located near the leading edge, close to the wall and thus close to the sensor, even at high  
180 angle of incidence. Due to its strong signal value, this configuration is well suited to detect the stall phenomena compared to other e-Telltale cases.

#### 4.4 Influence of the shell (TES L case)

The figure 13a and 13b show respectively the mean and standard deviation of the TES L case versus the angle of incidence, compared with the reference case. The magnitude of both, the mean and standard deviation is larger considering both the trailing  
185 edge separation and the leading edge flow separation angles. However, both phenomena are detected at higher  $AoA$ ,  $AoA = 9^\circ$  and  $AoA = 21^\circ$  respectively, when compared with the sensor integrated in the aerodynamic surface. Further investigations are

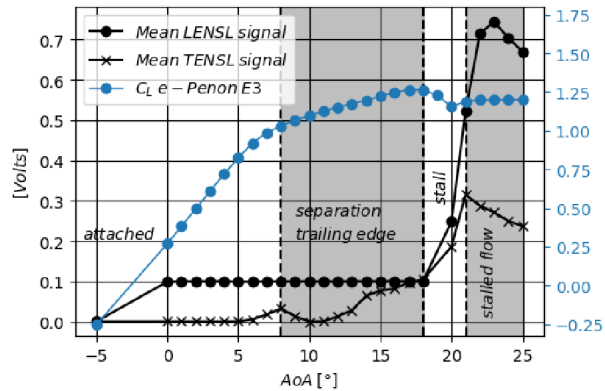


**Figure 11.** Comparison of the mean a) and the standard deviation b) of the signal of the TENSLS (reference case) and TENSSE e-Telltales as function of  $AoA$

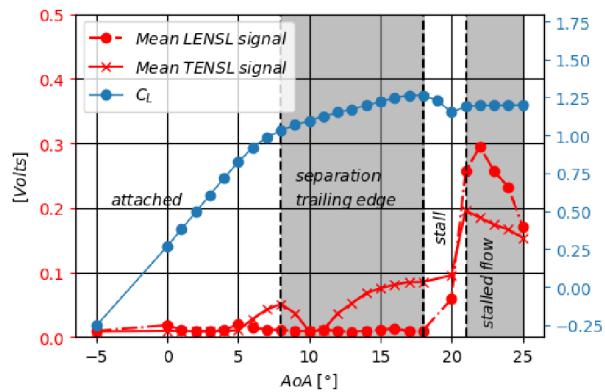
needed to understand that point which might be originated in the interaction of the shell aerodynamics with the separated shear layer.

## 5 Conclusion

190 In order to evaluate the ability of an innovative aerodynamic sensor, the e-Telltale, to detect the flow separation on wind turbine blades, high Reynolds wind tunnel tests ( $Re_c = 8.85 \cdot 10^5$ ) were performed on a 2D blade section using a full scale e-Telltale sensor. This innovative sensor is made of a strip with a strain gauge at its base, detecting displacement of the strip away from the aerodynamic surface. These tests highlight the impact of different parameters on the signal of the e-Telltale, and also the possible different use of the sensor in agreement with these findings. First, the e-Telltale with a long  
 195 strip located at the trailing edge is able detect both the trailing flow separation angle and the stall angle. This e-TellTale



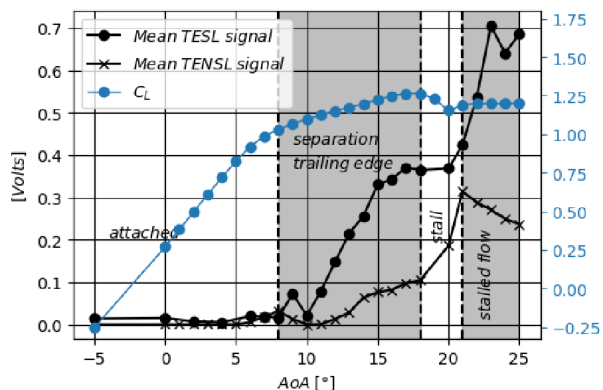
a)



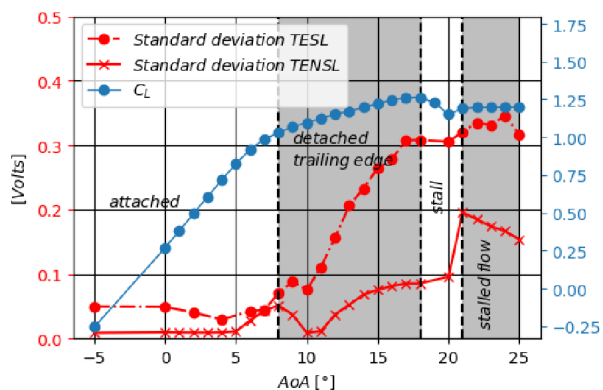
b)

**Figure 12.** Comparison of the mean a) and the standard deviation b) of the signal of the LENS and TENS (reference case) e-Telltales as function of  $AoA$

detects only the stall angle when located at the leading edge, with however a higher amplitude in the mean and standard deviation of the sensor's signal. For a sensor located at the trailing edge, the strip can be shortened to only detect the trailing edge separation angle with a higher sensitivity, this configuration being however ineffective to detect the stall angle. The limit size of the strip have not been identified yet but a relative length of  $L_{TENS}/c = 0.11$  (associated with a thickness ratio of  $A_{TENS}/T_{TENS} = 218 \text{ mm}$ ) is short enough for that purpose while a length of  $L_{TENS}/c = 0.19$  (associated with a thickness ratio of  $A_{TENS}/T_{TENS} = 218 \text{ mm}$ ) is too long. When the e-Telltale is not integrated to the aerodynamic surface, which may concerns all the wind turbines already in operation, the same conclusions can be drawn: this sensor is able to detect both the trailing edge flow separation angle and the stall angle.



a)



b)

**Figure 13.** Comparison of the mean a) and the standard deviation b) of the signal of the TESL and TENSLE e-Telltales as function of  $AoA$

*Author contributions.* This work was performed during the PhD of AS under the supervision of CB, DV and FD. CB provided her scientific expertise in turbulent shear flows and wind turbine blade aerodynamics, DV shared his expertise on the innovative e-TellTale sensor, FD supervised the wind tunnel experiments at CSTB.

*Competing interests.* No conflict of interest.

*Acknowledgements.* Authors would like to thank the technical staff at CSTB who carried experiments in the NSA wind tunnel.



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