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Review on Liquid Piston technology for compressed air energy storage

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Abstract

Compressed air energy storage systems (CAES) have demonstrated the potential for the energy storage of power plants. One of the key factors to improve the efficiency of CAES is the efficient thermal management to achieve near isothermal air compression/expansion processes. This paper presents a review on the Liquid Piston (LP) technology for CAES as a timely documentary on this topic with rapidly growing interests. Various aspects are discussed including the state-of-the-art on LP projects all over the world and the trend of development, the coupled fluid flow and heat transfer during the compression/expansion operations, and different actions proposed and implemented to enhance the heat transfer inside the piston column.

It has been found that LP is a promising concept for isothermal CAES. However, the complex and transient fluid flow and heat transfer behaviors inside the LP are difficult to characterize and master. To enhance the heat transfer and increase the efficiency of the compression/expansion processes many approaches have been tested including liquid spray, wire mesh, porous media, optimal trajectory, hollow spheres and optimal geometry of the piston column. Numerous Nusselt number's empirical correlations have also been proposed to estimate the

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heat transfer in different types of LP, as reviewed and summarized in this paper.

Keywords: Liquid Piston (LP), Compressed Air Energy Storage (CAES),
Compression and expansion, Flow pattern, Heat transfer enhancement,
Thermal management

1 **Nomenclature**

2 **Abbreviations**

3 A-CAES Adiabatic Compressed Air Energy Storage

4 ABS Acrylonitrile Butadiene Styrene

5 CAES Compressed Air Energy Storage

6 CFD Computational Fluid Dynamics

7 CR Compression Ratio

8 D-CAES Diabatic Compressed Air Energy Storage

9 ESS Energy Storage System

10 HDPE High-Density Polyethylene

11 HTE Heat Transfer Enhancement

12 I-CAES Isothermal Compressed Air Energy Storage

13 LP Liquid Piston

14 MDD Mean Droplet Diameter

15 ML Mass Loading

16 Mtoe Million tonnes of oil equivalent

17 PP Polypropylene

18 PPI Pores per inch

- 1 PSH Pumped-Storage Hydropower
2 RTE Round Trip Efficiency
3 SMES Superconducting Magnetic Energy Storage
4 VOF Volumn of fluid

5 **Greek Symbols**

- 6 Δ Difference
7 η Efficiency
8 λ Thermal conductivity
9 μ Dynamic viscosity
10 ν Kinematic viscosity
11 ω_g Swirl velocity
12 ρ Density

13 **Non-dimensional numbers**

- 14 Ca Capillary number
15 Nu Nusselt number
16 Pe Peclet number
17 Pr Prandtl number
18 Ra Rayleigh number
19 Re Reynolds number
20 St Stanton number

21 **Latin symbols**

- 22 A Surface Area

| | | |
|----|-------------------|----------------------------------|
| 1 | c_p | Constant pressure heat capacity |
| 2 | c_v | Constant volume heat capacity |
| 3 | D | Diameter |
| 4 | E | Energy |
| 5 | f_r | Friction factor |
| 6 | g | Gravitational acceleration |
| 7 | H | Global heat transfer coefficient |
| 8 | h | Heat transfer coefficient |
| 9 | L | Length |
| 10 | m | Mass |
| 11 | n | Polytropic index |
| 12 | P | Pressure |
| 13 | Q | Heat flow rate |
| 14 | R | Radius |
| 15 | T | Temperature |
| 16 | t | Time |
| 17 | th | Thickness |
| 18 | U | Velocity |
| 19 | V | Volume |
| 20 | W | Work |
| 21 | X | Position |
| 22 | Subscripts | |

| | | |
|----|---------|-----------------------|
| 1 | 0 | Initial at $t = 0s$ |
| 2 | air | Air |
| 3 | amb | Ambient |
| 4 | ave | Average |
| 5 | c | Compression |
| 6 | e | Expansion |
| 7 | f | Final at $t = t_f$ |
| 8 | fo | Foam |
| 9 | G | Gas |
| 10 | i | Inner |
| 11 | iso | Isothermal |
| 12 | L | Liquid |
| 13 | lam | Laminar |
| 14 | lc | Characteristic Length |
| 15 | o | Outer |
| 16 | ref | Reference |
| 17 | tr | Transition |
| 18 | $turb$ | Turbulent |
| 19 | W | Wall |
| 20 | $water$ | Water |

1. Introduction

The energy consumption worldwide has increased by 21% from year 2009 to 2019 and is expected to grow with more than 50% by 2050 [1]. To meet this demand, the world energy production reached 14 421 Mtoe (million tonnes of oil equivalent) in 2018, with more than 81% driven by fossil fuels (natural gas, coal and oil) [2]. In the meantime, awareness has been raised on the concomitant environmental impacts due to the greenhouse gas emission, evidenced by the signing of the Paris Agreement [3] in 2016 aiming at substantially reducing the global warming and the risks of climate change. In this context, renewable energies, with their increasing contribution year by year to the global energy market, are expected to play a major role to achieve the COP21 objectives for a more sustainable and decarbonized future [3].

Most renewable energy sources (e.g., solar, wind, tidal) show the intermittent character, bringing more difficulties for their exploitation at large scale. In this regard, the Energy Storage Systems (ESSs) have become an essential element for the power generation plants driven by renewable sources so as to augment their reliability and dispatchability [4]. Moreover, the integration of ESSs brings additional economic benefits by storing the excess power produced at times of low demand or low generation cost to be used during the peak periods with a higher price, accelerating thereby the pathway to a flexible, low-cost and electrified future [5, 4]. Different types of ESSs have been developed and utilized in the past decades, classified as follows based on their storage mechanisms.

- Electrical energy storage: electrostatic energy storage including capacitors and super-capacitors, magnetic/current energy storage including Superconducting Magnetic Energy Storage (SMES), etc. [5];
- Thermal energy storage: sensible, latent and thermochemical heat storage [6, 7, 8, 9];
- Mechanical energy storage: kinetic energy storage (flywheels), potential energy storage (Pumped-Storage Hydropower (PSH) and Compressed Air

1 Energy Storage (CAES)), etc. [10];

2 • Chemical energy storage: hydrogen, methane, etc. [5];

3 • Electrochemical energy storage: batteries (lead-acid, lithium-ion, nickel-
4 cadmium, etc.) and fuel cells [11].

5 Among all these energy storage technologies, the PSH and CAES have been
6 proven to be the most adapted one to store electricity in large scale [12]. PSH
7 is a mature technology featured by its large power capacity (100-3000 MW),
8 long storage period (1-24 h) and high Round Trip Efficiency (RTE) (71% to
9 85%) [13]. Therefore, it has dominated the energy storage market accounting
10 for 95% of the global capacity (over 100 GW) [14]. Its main drawback lies in the
11 rareness of available sites for two large reservoirs and one or two dams, a long
12 construction time required (typically about 10 years) and high investment costs
13 (typically hundreds to thousands of million US dollars). Moreover, removing
14 trees and vegetation from the vast land prior to the flooding of reservoirs also
15 appends additional environmental impacts. For these reasons, the planning of
16 new PSH sites has been limited in the past years [13] despite that the overall
17 new power installation is still dominating [15]. In the meantime, the CAES
18 technology has experienced a rapid growth in recent years, demonstrating the
19 potential to become the next major ESS for power plants [12].

20 *1.1. Compressed Air Energy Storage*

21 Storing electrical energy using a compressed air system can be dated back to
22 the early 20th century [13], in a way that it decoupled a conventional gas turbine
23 cycle into separated compression and expansion processes [16]. The energy was
24 stored in the form of elastic potential energy of compressed air and could be
25 restituted by expanding it through turbines to reproduce the electricity. The
26 waste heat of the exhaust gas can be captured through a recuperator before be-
27 ing released to the atmosphere. In modern CAES systems, the compressed air
28 can be stored either in man-made containers at the ground level or underground

1 (salt caverns, hard rock caverns, saline aquifers) [17, 18, 19]. Offshore and un-
2 derwater storage systems have also been tested and are under rapid development
3 [12]. Capital costs for CAES facilities depend on the storage conditions, ranging
4 typically between 400\$ and 800\$ per kW [20]. Above-ground systems usually
5 request higher investments than underground ones [13, 15].

6 Since the gas compression and expansion processes involve a large amount of
7 heat, the efficiency and storage capacity of the CAES systems depend strongly
8 on the heat transfer behaviors. Different variants of CAES exist, distinguishing
9 themselves by the manner of thermal management.

- 10 • Diabatic-CAES (D-CAES): this technology pressurises and heats air by
11 combusting a fuel (usually natural gas) and expands it through turbo tur-
12 bines to generate electricity. It is the basic but the only industrialized
13 version of the CAES at large scale [21, 5, 12]. Nevertheless, gas emis-
14 sions are inevitable due to the fuel combustion required to drive the gas
15 compressor.
- 16 • Adiabatic-CAES (A-CAES): the main improvement for this second gen-
17 eration of CAES is to retain the heat of the compressed air rather than
18 wasting it, to be used then in the expansion process. As a result, it renders
19 higher efficiency than that of D-CAES systems.
- 20 • Isothermal-CAES (I-CAES): this emerging technological variant tries to
21 prevent the temperature increase in the compressors (during charging)
22 and the temperature drop (during discharging) in the expansion devices,
23 permitting isothermal cyclic operations. The main advantage of I-CAES
24 is that fossil-based backup is not needed to start the expansion, implying
25 higher efficiencies and more environmental friendliness [12, 5].

26 1.2. *Liquid Piston technology*

27 It is by far very difficult to achieve isothermal compression and expansion
28 using conventional devices. The heat transfer of the system has to be very effi-
29 ciently mastered to dissipate (almost) all the heat generated in the compression

1 process, and vice versa, to provide necessary heat for the expansion. Therefore,
2 all I-CAES concepts known so far are based on piston machinery capable of
3 performing a comparably slow compression or expansion, leaving enough time
4 for heat exchange to approach a near-isothermal condition.

5 In this context, the interest over Liquid Piston (LP) for CAES usage has
6 been rapidly growing in recent years. In the LP compressor, a column of liquid
7 (usually water) is pumped in to compress a fixed amount of gas (usually air).
8 The main advantages of LPs compared to solid piston machinery is that the gas
9 leakage can be avoided and the dissipations due to friction be largely reduced,
10 leading to a higher efficiency. Many CAES projects with LP concept have been
11 developed or are under construction, as listed in Table 1. For example, the
12 GLIDES project (at ground level) by Oak Ridge National Laboratory using
13 liquid spray injection to enhance heat transfer reported a 66% RTE for their
14 LP prototype [22]. More recently, other modifications to the system have been
15 introduced and several tests have been carried out by replacing the noncon-
16 densable air with a condensable gas (CO_2 , synthetic refrigerants, hydrocarbon
17 refrigerants, etc.) [23]. Near isothermal and near isobaric charging/discharging
18 processes could be achieved.

19 Moreover, the LP concept suits especially the underwater/sub-sea CAES
20 systems in that the heat exchange can be largely enhanced due to the cold water
21 environment with high heat capacity. Park et al. [24] have developed a simplified
22 model (steady-state operation; ideal gas) and a proof-of-concept experiment for
23 a compression process by LP. Their study was later completed by Patil & Ro
24 [25, 26]. It has been estimated that the RTE of their 2 MWh stored energy CAES
25 system could reach 51% with a compressor/expander efficiency of 79%, and up
26 to 73% with a higher compressor/expander efficiency of 95%. More recently, the
27 REMORA project [27, 28, 29] showed that near-isothermal condition could be
28 achieved with a LP-based compression/expansion cycle due to the increased air
29 storage pressure and the reduced heat loss. A 95% compression efficiency could
30 be achieved by the LP, leading to a 70% RTE for the CAES system [27, 29].
31 Table 1 lists different CAES projects, where it could be seen that D-CAES is

¹ the only implemented technology already in service. I-CAES, although still
² underdevelopment, could be promising because it can theoretically offer higher
³ RTE.

| Technology | Project name | Location | Status | RTE | Remarks |
|------------|-----------------------------|------------------------|---------------------------------|--------------|---|
| D-CAES | Huntorf [21, 5] | Huntorf Germany | Commercialized since 1978 | 42% | 1 st commercialized project with 290 MW capacity |
| D-CAES | McIntosh [5] | Alabama, USA | Commercialized since 1991 | 54% | Constructed with a coal plant 110 MW capacity |
| A-CAES | ADELE [30] | Germany | Demonstrator | Up to 70% | Stopped in December 2013 due to uncertain business prospects |
| I-CAES | LightSail Energy [31] | Berkeley, USA | Development stopped in 2017 | 70 % | Water spray injection, LP |
| I-CAES | SustainX [32] | Seabrook, USA | Under development | 70 % | Planned site in Ontario, Canada |
| I-CAES | General Compression [32] | USA | Merged with SustainX in 2015 | – | Underground caverns |
| I-CAES | GLIDES [33] | ORNL, Tennessee USA | Lab-scale prototypes | Up to 80% | Ground-level LP with heat transfer enhancement |
| I-CAES | REMORA [28] | Nantes, France | Lab-scale demonstrator | 70% | LP for compression/expansion |

| Technology | Project name | Location | Status | RTE | Remarks |
|------------|-----------------------------|----------------------------|--|--------------|---|
| I-CAES | FLASC [34] | Malta & Netherlands | Small-scale Prototype (530 MWh) | Up to 75% | LP for compression/expansion Prototype at grand Harbour, Malta |
| I-CAES | Enairys Pow- ertech [35] | Écublens, Switzer- land | Under development; commercialized at small scale | - | LP for compression/expansion |

Table 1: Several CAES projects and plants worldwide (general information and status)

1 *1.3. Objectives of this paper*

2 In the open scientific literature, the LP concept has also received increasing
3 research attention, indicated by the significantly increasing number of yearly
4 documents since 2008 (Fig. 1). A great number of researches have been de-
5 voted to study the heat transfer behaviors through monitoring the evolution
6 of gas temperature [36, 37, 27] and by implementing various techniques to en-
7 hance the heat transfer [26, 33]. Few studies have focused on the flow patterns
8 and their transition, showing that the fluid flow hydrodynamics are rather com-
9 plex and difficult to predict especially when coupled with heat transfer [36, 38].
10 The effects of some design and operational parameters have been studied but
11 their relationships and coherence to the thermofluidic behaviors are still veiled.
12 Moreover, the systematic state-of-art literature review to keep track with all the
relevant studies and the new developments is lacking.

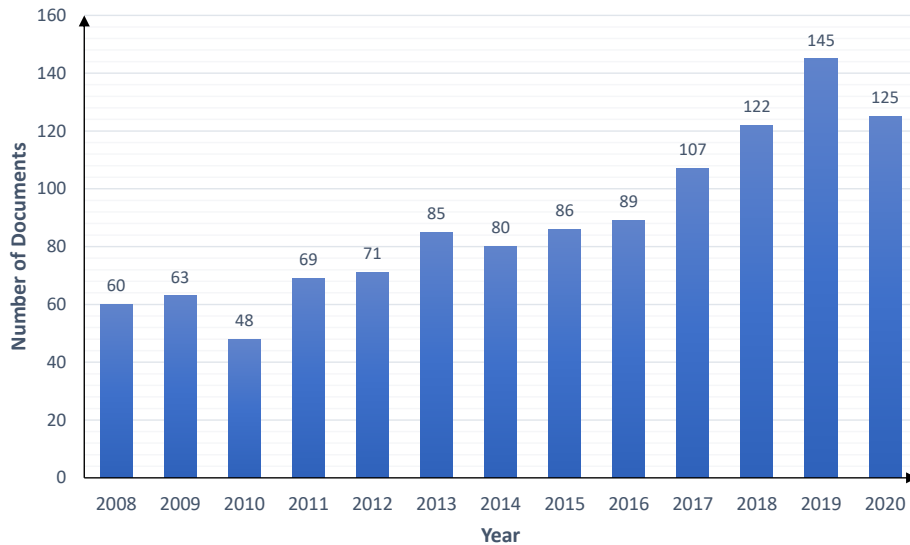


Figure 1: Trend of published documents (journal articles, conference proceedings, book chapters, patents) with "Liquid Piston" in title or abstract for in Dimensions [39] (August 2021)

13

14 This review paper is thereby expected to be the first to gather and survey
15 previous works on LPs for CAES, a research topic with rapid advances in recent
16 years. In particular, a comprehensive and combined knowledge on the thermo-

1 dynamics, the fluid flow and heat transfer mechanisms within the LP is provided.
2 Special focus is given on the thermal management issue, and measures proposed
3 and implemented by different researchers to enhance the heat transfer, aiming
4 at achieving a highly-efficient LP for I-CAES. The contributions of this paper
5 are important because it provides a technology state-of-the-art and a progress
6 overview of the LP technology covering from the fundamental transport phe-
7 nomena to the R&D projects. It may enable interested researchers to timely
8 grasp the latest advancements and existing challenges for the investigation and
9 application of LP technology in future (underwater) CAES systems.

10 The rest of the paper is organized as follows. The operating principles and
11 the thermodynamic background of the LP are described in section 2. The fluid
12 flow patterns and the heat transfer characteristics inside the LP during the
13 compression are presented as well. Various techniques to enhance the heat
14 transfer are summarized and compared in section 3. Finally, conclusions and
15 perspectives are presented in section 4.

16 **2. Liquid piston for energy storage**

17 LP is in fact not a new concept but can be dated back to the Humphrey
18 pumps in 1906 [40], which is a large internal combustion gas-fuelled LP pump
19 used for large-scale water supply projects. Later on, LPs were also used for Stir-
20 ling engines and Stirling pumps. The main applications of these configurations
21 were the usage of a LP as a cooler to serve superconducting computers or as a
22 heat pump [41, 42]. In 2009, Van de Ven & Li first proposed the LP concept
23 for energy storage through an open accumulator [36].

24 *2.1. Operating principle*

25 The LP concept, compared to traditional solid piston, is featured by the
26 use of a compression chamber where a liquid (often water) is used to increase
27 or decrease the pressure of a gas (often air or hydrogen). The gas and liquid
28 phases are naturally separated by their density difference, as long as the piston

1 advance velocity has no particular effects on the gas-liquid interface. But when
2 a portion of gas becomes trained in the liquid due to the high pressure, the
3 decreased bulk modulus of the liquid could cause the cavitation problem in
4 low pressure areas of the hydraulic system. Van de Ven & Li [36] proposed to
5 use a blade diaphragm in the liquid column to separate the LP fluid from the
6 working fluid of the hydraulic system, allowing the pressure transfer across the
7 diaphragm with the minimal loss. Other technological alternatives also exist,
8 i.e., by coupling a LP with a solid piston to handle the trained gas problem [43].

9 The general operation principle of LP, as has already been studied [36] and
10 proved to be feasible for CAES projects [27, 44, 24], is briefly described as
11 follows:

- 12 • During the charging stage, the surplus electricity drives the hydraulic
13 pump to fill the liquid into the closed compression chamber from the bot-
14 tom (Fig. 2a). Since the chamber is hermetically sealed, the trapped gas
15 in the top region undergoes its pressure increase from P_0 up to the setting
16 point P_c until it is discharged to the gas reservoir by opening the outlet
17 valve. Note that the ratio between P_c and P_0 is called the Compression
18 Ratio (CR).
- 19 • During the discharging stage, the compression chamber initially full of
20 liquid is filled with compressed gas released from the gas reservoir. The
21 expansion of the compressed gas in the chamber pushes the liquid to flow
22 back through the turbine to generate electricity (Fig. 2b). Once a certain
23 volume of gas is present in the chamber, the inlet valve is closed.

24 Valves are used accordingly to control and switch the process from compres-
25 sion mode to expansion mode by changing the inlet/outlet domains [27, 36].

26 Besides air or hydrogen, helium was also used as the working gas [45]. Piya
27 et al. [46] made a 2D finite difference modeling to test different combinations
28 of liquid (DTE 25 hydraulic fluid (oil) or water) and gas (helium or air). Their
29 results showed that helium/water combination had the best compression effi-

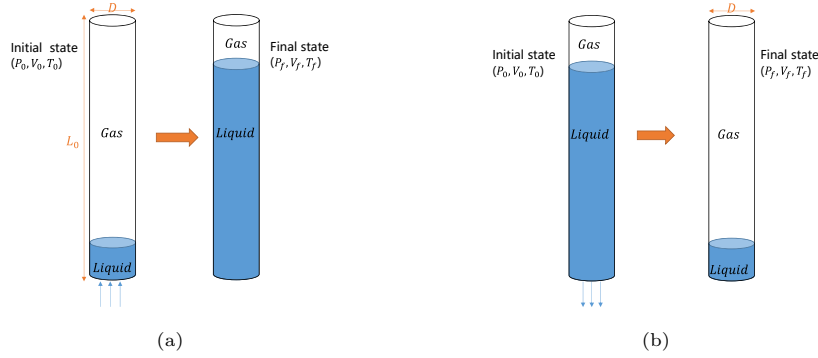


Figure 2: Operation of LP during the compression (a) and the expansion (b) processes

1 efficiency η_c , going up to 61.9% compared to 54.31% for air/water and 23.58% for
 2 air/DTE25 based on the same configuration.

3 2.2. Thermodynamic background

4 Lemofouet & Rufer [47] showed that the usage of LP-based CAES can involve
 5 different cycles, including the isothermal cycle (A-D path as shown in Fig. 3, the
 6 same below), the Joule cycle (A-B-D-E-A: a combination of the adiabatic and
 7 isobaric processes), the Otto cycle (A-B-C-D-E-F-A: a combination of adiabatic
 8 and isochoric processes) and the polytropic cycle (A-B'-C'-D-F' and A-B'-D-
 9 E'-A). The gray area enclosed in each cycle's path represents the energy losses
 10 for that cycle. It can be seen that the isothermal cycle without energy loss is
 11 the optimal one among these cycles. The closer the path to the isothermal one
 12 the less losses the system generates (Fig. 3). In both closed and open systems,
 13 the Joule cycle is more efficient than the Otto cycle for small values of CR
 14 [48, 47, 49]. Dib et al. [48, 49] also studied the thermodynamics of different
 15 compression/expansion cycles with a LP and showed that a Joule cycle is more
 16 adapted for a LP gas compression for energy storage.

17 In an open-accumulator CAES system air at initial state (T_0, P_0, V_0) is com-
 18 pressed by a LP through a $P-V$ trajectory (ζ) to a second state (T_c, P_c, V_c) with
 19 a CR = $\frac{P_c}{P_0}$. Then, the compressed air is ejected into the open accumulator and
 20 cooled isobarically to a smaller volume (V_{iso}), by adding liquid into the bottom

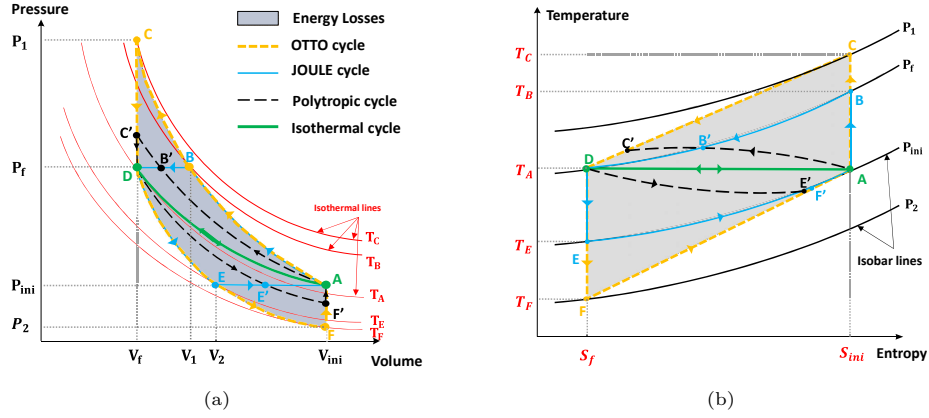


Figure 3: $P - V$ and $T - S$ diagrams for air compression and expansion processes using different cycles (Otto, Joule, Polytropic and Isothermal). Adapted from [47]

1 of the accumulator until the air reaches the final stored state (T_0, P_c, V_{iso}) . For
 2 an internally reversible steady flow device, the specific work could be calculated
 3 by:

$$W = \int_{V_0}^{V_c} P dV \quad (1)$$

5 The total amount of work input to store the air at the final state is then:

$$W_{C_{actual}} = \int_{V_0}^{V_c} \underbrace{(P(t) - P_0) dV}_{\text{Compression flow}} + \underbrace{\left(1 - \frac{1}{CR}\right) (P_c V_c - P_0 V_0)}_{\text{Isobaric cooling}} + \underbrace{(P_c - P_0) V_{iso}}_{\text{Pumping}} \quad (2)$$

7 After the storage stage, the total potential energy E_p that could be extracted
 8 from the stored compressed air is shown as the area shaded by the solid vertical
 9 lines (Fig. 4a).

$$E_p = \underbrace{(P_c - P_0) V_{iso}}_{\text{Air discharge to isobaric tank storage}} + \underbrace{P_0 V_0 \left(\ln(CR) + \frac{1}{CR} - 1 \right)}_{\text{Isothermal expansion}} \quad (3)$$

11
 12 The compression efficiency is then defined as Eq. 4 [50]:

$$\eta_c = \frac{E_p}{W_{C_{actual}}} = \frac{(P_c - P_0) V_{iso} + P_0 V_0 \left(\ln(CR) + \frac{1}{CR} - 1 \right)}{\int_{V_0}^{V_c} (P(t) - P_0) dV + \left(1 - \frac{1}{CR}\right) (P_c V_c - P_0 V_0) + (P_c - P_0) V_{iso}} \quad (4)$$

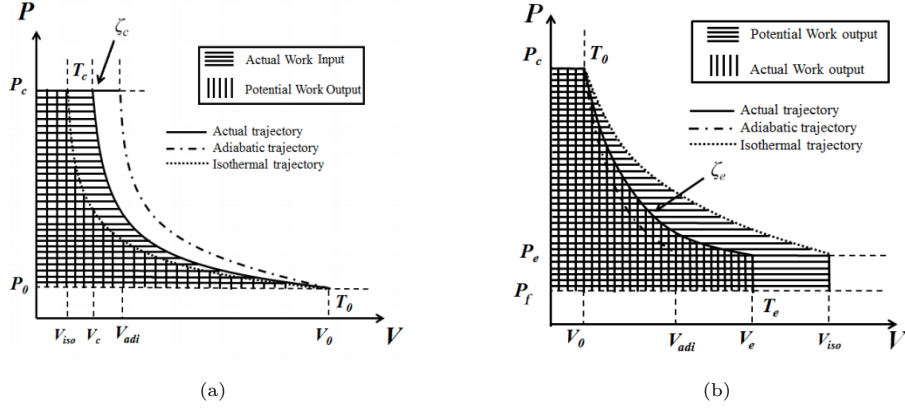


Figure 4: $P - V$ curves showing compression (a) and expansion (b) trajectories for a LP in an open-accumulator system CAES [50]

1 When the actual compression trajectory approaches the isothermal one, the
2 area shaded by vertical lines gets closer in magnitude to that by the horizontal
3 lines, and η_c increases. On the contrary, when the trajectory gets closer to the
4 adiabatic one, the difference between those two areas becomes larger and η_c
5 decreases.

6 For the expansion stage, the stored compressed air (T_0, P_c, V_0) is firstly
7 ejected from the open-accumulator isobarically and then expanded in the LP at
8 (T_e, P_e, V_e) . The process also follows a trajectory between the isothermal one
9 and the adiabatic one (Fig. 4b). Similarly, the expansion efficiency η_e can be
10 calculated in Eq. 5 [50]:

$$11 \quad \eta_e = \frac{W_{e_{actual}}}{E_p} = \frac{\int_{V_0}^{V_e} (P(t) - P_f) dV + (P_c - P_f)V_0}{P_c V_0 (\ln(\text{CR})) - (P_f(V_{iso} - V_0)) + (P_c - P_f)V_0} \quad (5)$$

12 It can be seen from Eq. (4) that the compression efficiency η_c depends on the
13 compression time as well as the CR. It also depends on the modeled cycle. Since
14 many cycles have been studied, different formulas to calculate the efficiencies
15 exist. Patil & Ro [51] defined compression/expansion formulas (Eqs. 6 and 7)

1 only related to CR and n the polytropic index :

$$2 \quad \eta_c = \frac{\ln(CR) + \frac{1}{CR} - 1}{\frac{CR^{\frac{n-1}{n}} - 1}{n-1} + CR^{-\frac{1}{n}} - 1 + (CR - 1) \left(CR^{-\frac{1}{n}} - \frac{1}{CR} \right)} \quad (6)$$

$$3 \quad \eta_e = \frac{1 - \frac{\left(\frac{1}{CR}\right)^{\frac{n-1}{n}}}{n-1} - \left(\frac{1}{CR}\right)^{\frac{n-1}{n}} + \frac{1}{CR}}{\ln(CR) + \frac{1}{CR} - 1} \quad (7)$$

4 Mutlu & Kiliç [52] have numerically examined the effect of piston speed, the
5 CR and the cylinder geometry on η_c and required power of a LP. It was found
6 that the η_c decreased from 87.93% to 68.89% when CR increased from 5 to 80.
7 The piston velocity had only a small impact (0.5%) on η_c once a certain velocity
8 ($U_{pist} = 0.05 \text{ m/s}$) had been reached. Decelerating the piston velocity at higher
9 pressures did not affect the total input work nor the efficiency of the process.
10 Their study recommended to use long cylinders with small diameters to achieve
11 a higher η_c , e.g., up to 91.42% with the proper length/diameter ratio L/D
12 [52]. For low pressure compression, an important flow rate (velocity) is usually
13 imposed to compensate the power level and the system performance. Both the
14 convective heat transfer coefficient and the frictional pressure drop will increase
15 with the increasing piston advance velocity, leading to higher energy density at
16 the cost of higher hydraulic loss. The same thing applies to the L/D ratio and
17 a trade-off needs thereby to be made.

18 2.3. Flow patterns during air compression

19 The fluid flow regime inside a LP column during dynamic operation is rather
20 complex, involving unsteady compressible multiphase flow (liquid/gas) with con-
21 jugated heat transfer. The Reynolds number Re is usually used to characterize
22 the flow pattern within a cylindrical column (uniform cross-sectional area):

$$23 \quad Re_L = \frac{\rho_L U D}{\mu_L} = \frac{U D}{\nu_L} \quad (8)$$

24 Or :

$$25 \quad Re_G = \frac{\rho_G U D}{\mu_G} = \frac{U D}{\nu_G} \quad (9)$$

1 Where ρ is the density of the fluid, U is the advance velocity (of the interface)
 2 in the piston column, D is the (hydraulic) diameter of the column, μ and ν the
 3 dynamic and kinematic viscosity, respectively.

4 However, one should distinguish the Re for the liquid phase (Re_L) and that
 5 for the gas phase (Re_G). Since a constant liquid volume flow rate is often
 6 assumed by the liquid pump, the liquid velocity as well as the Re_L are considered
 7 thereby as constant during the compression process. On the contrary, the gas
 8 flow is much more complex than a simple pipe flow because the gas does not
 9 have the same velocity as water all over the piston column, as reported by Mutlu
 10 & Kiliç [52] and Schober et al. [45]. It has been observed by Neu et al. [53, 27]
 11 that during compression many re-circulation zones could appear next to the
 12 walls (upper, bottom and lateral) of the piston column. The numerical results
 13 of Gouda et al. [54] also shown that the maximum local velocity of air could
 14 reach 8 times the velocity of water. The gas phase actually undergoes regime
 15 transition from laminar to turbulent along with the advancement of the piston
 16 (air and water are the working fluids), as illustrated in Fig. 5.

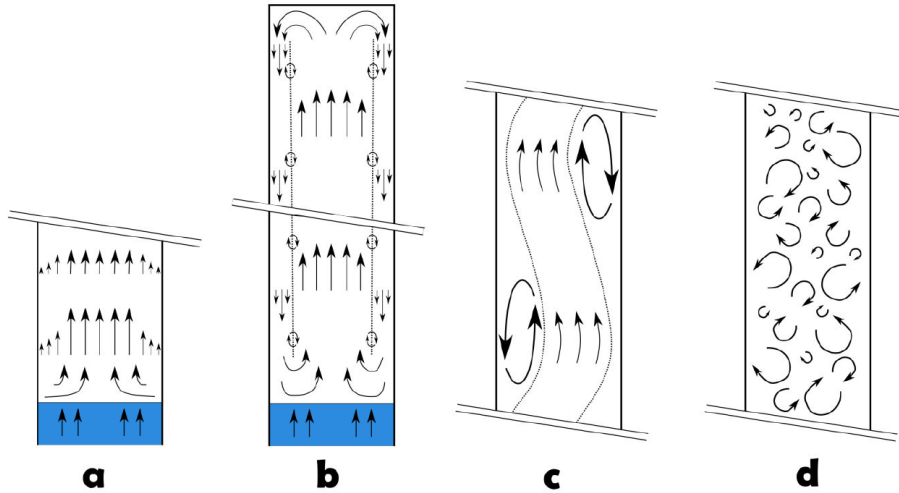


Figure 5: Evolution of air flow pattern along with the advancement of the water during the compression operation [53]

17 At first, an ascending central air flow becomes established during the short

1 start-up period of compression (Fig. 5 (a)). This central rising flow having a
 2 velocity magnitude greater than the water velocity will then disturb the station-
 3 ary air at the upper part of the column, initiating the air movement. In a little
 4 while, the central rising air flow occupies the entire compression chamber while
 5 the falling flow occurs close to the walls of the column (Fig. 5 (b)). Recirculation
 6 zones are thereby present at both ends of the compression chamber, i.e. at the
 7 column head where the central rising air is deviated to the outer periphery and
 8 at the air-water interface where the descending air rejoins toward the center of
 9 the column. At the boundary between the ascending and descending air flows,
 10 small instabilities would cause small amplitude of movements between the two
 11 velocity regions. When the central air velocity is significantly higher than the
 12 water velocity, the Rayleigh-Taylor instabilities [55] become stronger and will
 13 trigger swirling flows and vortices (Fig. 5 (c)). This transition regime is rather
 14 short and in a brutal way, the global structure of the air flow disintegrates into
 15 a multitude of small vortices as shown in Fig. 5 (d), indicating the turbulent
 16 feature of the flow. Such transition of flow regime has also been confirmed by
 17 Gouda et al. [54] through CFD (Computational Fluid Dynamics) simulation of
 18 the compression process with the VOF (Volume-Of-Fluid) model.

19 Many other formulations of Re number have therefore been proposed to
 20 better characterize the gas flow regimes and their transition. For example,
 21 Adair et al. [56] and Liu & Zhou [57] used the transient equivalent diameter
 22 and the swirl velocity (ω_g) instead of the averaged piston speed:

$$23 \quad Re = \frac{\rho(D/2)^2\omega_g}{\mu} \quad (10)$$

24 Van de Ven & Li [36] used the fully-developed pipe flow model to calculate the
 25 Re (Eq. 11):

$$26 \quad Re = \frac{U_{ave}D}{\nu} \quad (11)$$

27 where U_{ave} is the average gas velocity, estimated as half of the piston (liquid)
 28 velocity and D is the characteristic length (diameter of the piston).

29 Neu et al. [27, 38] observed that the transition from laminar to turbulent
 30 flow seems to happen at different stroke times depending on the operational

1 conditions (D , L , U_{pist} , etc.). Their experimental measurements of air temper-
 2 ature in the compression column also showed a switch from a slow evolution
 3 to a fast oscillation during the compression. An empirical correlation has been
 4 proposed to predict the position of the piston front at the moment of regime
 5 transition (Eq. 12):

$$6 \quad L_{tr}^* = \left(-0.0344L_0 + 109U_{pist}D^2 + \frac{0.0227}{D} \right) \left(\frac{P_0}{P_{ref}} \right)^{-0.645*\sqrt{U_{pist}}} \quad (12)$$

7 Where L_0 is the initial piston length, L_{tr}^* is the non-dimensional piston
 8 position (water/air interface) when the transition occurs, U_{pist} is the piston
 9 inlet velocity, P_{ref} is the reference pressure, generally equal to the atmospheric
 10 pressure and P_0 the initial pressure. This correlation has been experimentally
 11 verified within the range of $P_0 = 101325 \text{ Pa}$, $2 \text{ m} \leq L_0 \leq 6 \text{ m}$, $0.08 \text{ m/s} \leq$
 12 $U_{pist} \leq 1.25 \text{ m/s}$ and $3 \times 10^{-2} \text{ m} \leq D \leq 10^{-1} \text{ m}$.

13 *2.4. Heat transfer analysis*

14 The heat transfer characteristics inside a LP column is also complex, due
 15 to various heat transfer modes involved. The increasing gas (air) temperature
 16 during the compression will cause the heat transfer between the gas and liquid
 17 which are in direct contact. Moreover, the convection heat transfer, which is
 18 the most dominant, occurs between the piston wall and the gas or/and liquid.
 19 Depending on the advancement of the gas-liquid interface, the flow regimes, the
 20 thermal resistance of wall and the temperature of the external environment, the
 21 liquid may absorb heat from the wall, or vice-versa, release the heat towards
 22 the wall.

23 Nikanjam & Greif [58] measured the wall temperature variation of a LP to
 24 estimate the heat flux. The liquid chosen was oil, and two gases (argon and air)
 25 were compressed with a thin film gauge placed at the end wall of the compression
 26 chamber. Only conduction heat transfer was supposed to happen at the wall.
 27 Kermani & Rokni [59] developed a detailed heat transfer model considering both
 28 the heat transfer through the gas-liquid interface and through the wall. Some

- 1 critical parameters have been identified to maximize the amount of heat that can
 2 be extracted from the compressed gas for a hydrogen compression technology.

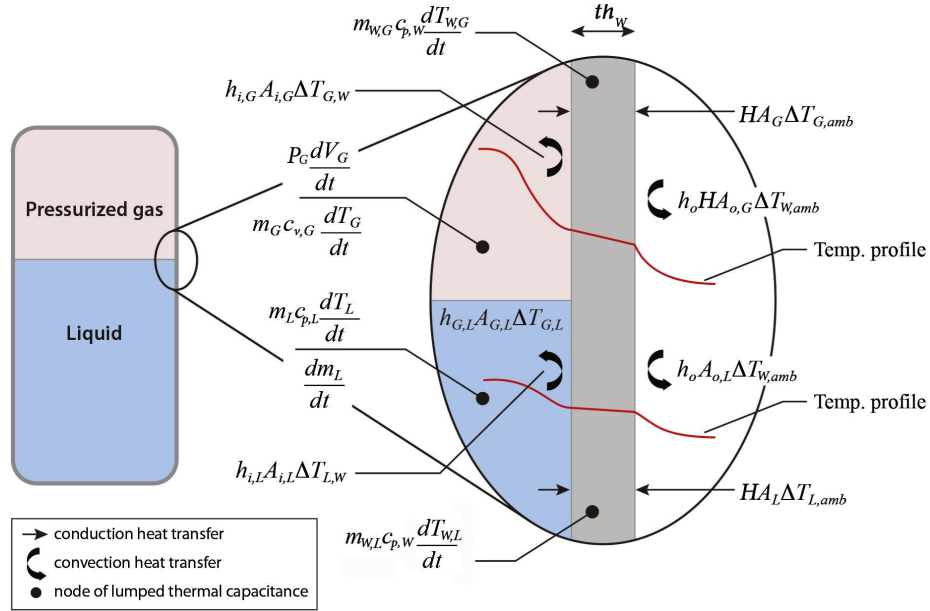


Figure 6: Heat transfer modeling for LP compression/expansion, including the convection between the gas and liquid through the interface, the convection at the inner and outer walls of the piston column and the conduction through the wall. Adapted from [22]

3 A detailed modeling of heat transfer inside the LP has been performed by
 4 Odukomaiya et al. [22], as shown in Fig. 6. It has been supposed that the upper
 5 wall temperature in contact with gas has a uniform temperature ($T_{W,G}$) and the
 6 one in contact with liquid ($T_{W,L}$) is also uniform. These temperatures evolve in
 7 time as the piston front moves upward (compression) or downward (expansion).
 8 The internal energy variation of the gas or liquid phase can be calculated by
 9 Eq. 13 or Eq. 14, respectively. Note that Redlich-Kwong equation of state was
 10 used to model the gas inside the column.

$$\underbrace{m_G c_{v,G} \frac{dT_G}{dt}}_{\text{change of the gas energy}} = \underbrace{-h_{G,L} A_{G,L} (T_G - T_L)}_{\text{heat transfer with the liquid}} - \underbrace{H A_G (T_G - T_{amb})}_{\text{heat transfer with ambient through the wall}} - \underbrace{P_G \frac{dV_G}{dt}}_{\text{boundary work}} \quad (13)$$

$$\underbrace{m_L c_{p,L} \frac{dT_L}{dt}}_{\text{change of the liquid energy}} = \underbrace{h_{G,L} A_{G,L} (T_G - T_L)}_{\text{heat transfer with the gas}} - \underbrace{H A_L (T_L - T_{amb})}_{\text{heat transfer with ambient through the wall}} + \underbrace{\dot{m} c_{p,L} (T_{amb} - T_L)}_{\text{mass flow}} \quad (14)$$

Where m is the mass, c_v and c_p are the specific heat capacity at constant volume and constant pressure, respectively, h is the heat transfer coefficient, A is heat transfer surface area, H is the global heat transfer coefficient and $\dot{m} = \frac{dm}{dt}$ is the mass flow rate. The subscripts G , L and amb represent the gas phase, the liquid phase and the ambient, respectively.

The variation of wall temperature in contact with the gas phase or liquid phase can be calculated as:

$$\underbrace{m_{W,G} c_{p,W} \frac{dT_{W,G}}{dt}}_{\text{change of wall energy}} = \underbrace{h_{i,G} A_{i,G} (T_G - T_{W,G})}_{\text{heat transfer between gas and inner wall}} - \underbrace{h_o A_{o,G} (T_{W,G} - T_{amb})}_{\text{heat transfer between outer wall and ambient}} \quad (15)$$

$$\underbrace{m_{W,L} c_{p,W} \frac{dT_{W,L}}{dt}}_{\text{change of wall energy}} = \underbrace{h_{i,L} A_{i,L} (T_L - T_{W,L})}_{\text{heat transfer between liquid and inner wall}} - \underbrace{h_o A_{o,L} (T_{W,L} - T_{amb})}_{\text{heat transfer between outer wall and ambient}} \quad (16)$$

Where the subscripts i , o and W represent the inner surface, the outer surface and the wall, respectively. The global heat transfer capacity $H A_G$ or $H A_L$ is calculated based on the sum of effective thermal resistances according to the convection on the inner and outer wall surfaces and the conduction through the wall.

$$H A_G = \frac{1}{\left(\frac{1}{h_{i,G} A_{i,G}}\right) + \left(\frac{t_{W,G}}{\lambda_{W,G} A_{ave,G}}\right) + \left(\frac{1}{h_o A_{o,G}}\right)} \quad (17)$$

$$H A_L = \frac{1}{\left(\frac{1}{h_{i,L} A_{i,L}}\right) + \left(\frac{t_{W,L}}{\lambda_{W,L} A_{ave,L}}\right) + \left(\frac{1}{h_o A_{o,L}}\right)} \quad (18)$$

1 where th_W is the thickness of the wall and the subscribe *ave* stands for average.
2 Kermani & Rokni [59] developed a LP model based on the first law of ther-
3 modynamics to predict the evolution of temperature and pressure. The amount
4 of heat transferred from the gas (hydrogen) towards the wall was estimated ac-
5 cording to different heat transfer correlations. The results showed small changes
6 in hydrogen temperature, between 0.2% and 0.4%, compared to the adiabatic
7 case. Patil et al. [37] experimentally tested a LP compressor with different
8 chamber materials and various stroke times of compression. It was observed
9 that the heat transfer rate increased with the increasing piston velocity but at
10 the expense of additional compression work and a higher gas temperature. The
11 convective heat transfer coefficient was observed to be high at the initial phase
12 of the compression but then decreased to a stable value. In fact, the interior
13 convective heat transfer coefficient $h_{i,G}$ for the gas phase is a key parameter
14 that should be carefully estimated in the modeling of heat transfer behaviors of
15 LP. For engineering applications, it is usually a convenient way to present and
16 compare the results in terms of the dimensionless Nusselt number Nu . Consid-
17 ering that $h_{i,G}$ varies along the piston (tube), it is usually the average value of
18 $h_{i,G}$ used to calculate the Nu :

$$19 \quad Nu_D = \frac{1}{L_0} \int_0^{L_0} \frac{D}{\lambda} h_{i,G}(L) dL = \frac{h_{i,G} D}{\lambda} \quad (19)$$

20 Where λ is the thermal conductivity of gas. Many correlations have been pro-
21 posed to estimate the heat transfer performance of the working gas, mainly for
22 a reciprocating piston application or for a compression fluid in tubes. In these
23 correlations the Nu number is usually written as a function of the Reynolds
24 number Re and the Prandtl number Pr , in a generic form shown in Eq. 20:

$$25 \quad Nu = A Re^a Pr^b \left(\frac{\mu}{\mu_0} \right)^c \quad (20)$$

26 Where A, a, b and c are constants fitting different working conditions. μ and
27 μ_0 are the fluid viscosity at fluid's average temperature and at the heat transfer
28 boundary surface temperature, respectively.

1 Some widely used empirical $Nu - Re$ correlations are listed in Table 3.
2 Among them, Dittus & Boelter [60] correlation was proposed for fully developed
3 turbulent flow in circular tubes, with smooth surfaces and moderate tempera-
4 ture difference between the wall and fluid mean temperature. Sieder & Tate
5 [61] extended the Dittus & Boelter correlation by taking the wall effect into
6 account for the condition of large temperature difference between the wall and
7 the fluid. Hamilton [62] proposed a new correlation more adapted to the com-
8 pressor chamber, especially for $L/D \leq 60$. Likewise, Annand [63], Adair et al.
9 [56], Liu & Zhou [57] and Hsieh & Wu [64] proposed other correlations for the
10 reciprocating compressors based on their proper experimental data.

11 More recently, Van de Ven & Li [36] numerically compared the evolution of
12 Re , Nu and gas temperature in a single stage LP and in a reciprocating solid
13 piston. Their results showed that the LP could have significantly lower Nu but
14 much higher $h_{i,G}$ than that of the solid piston, mainly due to a smaller diameter
15 of the compression chamber. Moreover, the LP can reach a near-isothermal
16 compression process which can hardly be achieved by the solid piston. Two
17 $Nu-Re$ correlations were proposed, one for laminar flow regime (Eq. 21) and
18 the other for turbulent flow regime (Eq. 22) [65].

$$19 \qquad \qquad \qquad Nu = 0.664Re^{1/2}Pr^{1/3} \qquad \qquad \qquad (21)$$

$$20 \qquad \qquad \qquad Nu = 0.023Re^{0.8}Pr^{0.3} \qquad \qquad \qquad (22)$$

22 Neu et al. [27] experimentally studied a LP prototype for CAES. A set of
23 configurations were tested with the column diameter D varying from 30 *mm*
24 to 100 *mm* and the piston length L_0 from 2 *m* to 6 *m*. Based on the data
25 of 73 compression tests, $Nu-Re$ correlation has been proposed for the laminar
26 flow regime (Eq. 23) and for the turbulent flow regime (Eq. 24), respectively.
27 Good agreement could be achieved between the correlation prediction and the
28 experimental data [27].

$$Nu_{lam} = 6.67 \left(Re_D Pr \frac{D}{L} \right)^{0.36} \quad (23)$$

$$Nu_{turb} = 6.17 \left(Re_D Pr \frac{D}{L} \right)^{0.48} \quad (24)$$

1 The transition from Nu_{lam} to Nu_{turb} could be determined by using the
 2 non-dimensional formula shown in the previous section (Eq. 12).

3 3. Heat transfer enhancement technologies for LP

4 The description in the above sections clearly indicates that how to handle
 5 the heat transfer to approach an isothermal gas compression and expansion
 6 is actually a key issue to increase the LPs efficiency and the RTE of CAESs.
 7 Therefore, different techniques for Heat Transfer Enhancement (HTE) have been
 8 proposed and implemented within the LPs. This section introduces several
 9 representative types of HTE with special focus on the concept, the efficiency
 10 increase and the related $Nu-Re$ correlation.

11 3.1. Spray injection

12 Spray injection concept can be utilized to increase the heat transfer during
 13 the compression/expansion process where a high pressure spray of water droplets
 14 is injected into the top of LP column. The water droplets having a high heat
 15 capacity, scattered in the air, can absorb heat from the air during compression
 16 or inject heat into air during expansion, as explained in Fig. 7. The mean
 17 droplet diameter (MDD), the mass loading (ML defined as the ratio of water
 18 spray mass injected into a cycle and the mass of dry air drawn into the piston
 19 chamber) and the injected spray temperature are key operational parameters
 20 that determines the efficiency of this thermal management concept.

21 Qin & Loth [43] studied the efficiency of LP with droplet spray through a
 22 1-D simulation validated with experimental data. Two ways of spray injection
 23 were considered: premixed injection upstream of the intake valve and direct
 24 injection within the compression chamber. For the direct injection, the nozzles

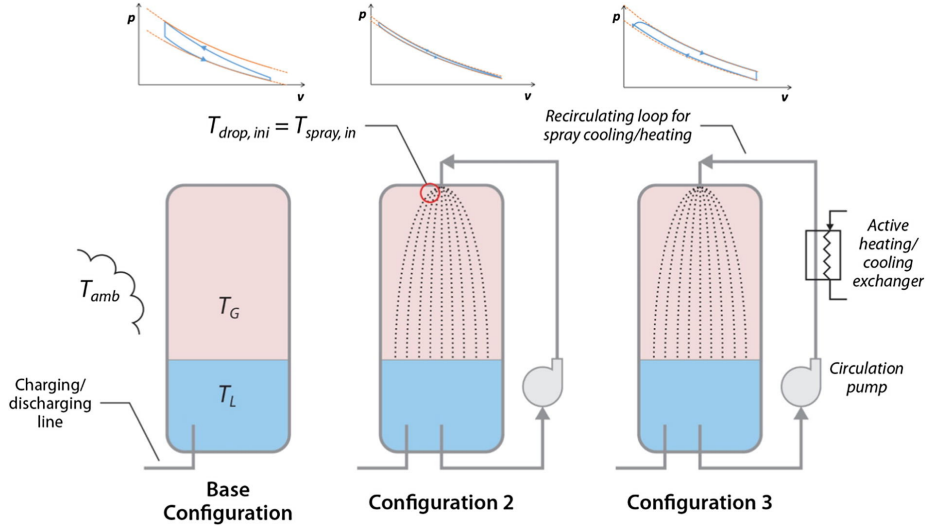


Figure 7: Investigated alternative liquid spray configurations of GLIDES system [22]

1 were installed inside the piston to inject and uniformly distribute the liquid
 2 droplets. For the premixed injection, the nozzle was placed upstream the intake
 3 valve. Various conditions were tested including a CR ranging from 6 to 12, a
 4 MDD of 10, 20, 50 and 100 μm and a ML of 0.5, 1, 2 and 5. Their simula-
 5 tion results showed that the direct injection concept of spray cooling (CR=10;
 6 MDD=20 μm ; ML=1) could reach a η_c higher than 95% compared to about
 7 71% for adiabatic compression. When ML increased to 5, the η_c could be fur-
 8 ther augmented up to 98%, as shown in Fig. 8. A $Nu-Re$ correlation (Eq.
 9 25) based on Ranz-Marshall form [66] was used to estimate the heat transfer
 10 characteristics of LP with liquid spray.

$$11 \quad Nu = 2 + 0.6Re^{1/2}Pr^{1/3} \quad (25)$$

12 With a similar configuration, Zhang et al. [67] only obtained a 10% gas
 13 temperature abatement compared to the adiabatic case, mainly due to a small
 14 $L/D \sim 1$. Guanwei et al. [68] has built an experimental setup to test the spray
 15 cooling technique for LP. Experiments have been done in the summer and in
 16 the winter with different ambient temperatures to compare the effects of the
 17 nozzle diameter (0.1 – 1 mm) and spray droplet diameters (10 – 100 μm) on the

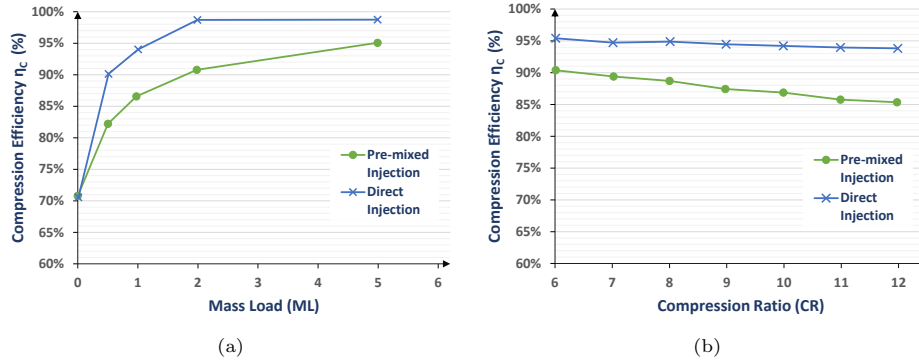


Figure 8: Efficiency increase of Liquid Piston by the liquid spray injection. (a) Compression Ratio (CR)=10 and variable Mass Loading (ML); (b) variable CR, Mean Droplet Diameter (MDD)=20 μm). Adapted from [43]

1 heat transfer rate and thus on the η_c . The best η_c (92,8%) has been achieved
 2 with CR=2 and $D = 0.3 mm$ nozzle diameter. Further increasing the nozzle
 3 diameter required a higher work to generate the spray due to higher mass of
 4 water injected in the piston.

5 Odukumaiya et al. [22] numerically and experimentally studied the effect of
 6 liquid spray for enhancing the heat transfer. Three configurations were tested
 7 as shown in Fig. 7: without liquid spray (base configuration), liquid spray with
 8 or without heat regeneration. The tests conducted under the base configuration
 9 showed a temperature rise up to 40 K during compression and a fall of 15
 10 K during expansion. The introduction of spray droplets could attenuate the
 11 temperature variation, to 13 K (for configuration 2) and 3 K (for configuration
 12 3), respectively. Nevertheless, this active heating/cooling for regulating the
 13 temperature of spray via an external heat exchanger needs extra energy input.
 14 Note that the same correlation of Eq. 25 has been used to model the heat
 15 transfer.

16 The impacts of other parameters, i.e., the compression time, the injection
 17 pressure and the spray angle on the heat transfer and the efficiency have also
 18 been studied by Patil et al. [69, 70]. Their results showed the positive effect of
 19 injection pressure when it increased from 10 psi (69 kPa) to 30 psi (483 kPa),

1 but the benefit became marginal beyond 30 psi. The nozzle angle at 60° was
2 shown to be the most efficient configuration whereas the compression time had
3 no significant impact.

4 The spray injection concept has proven its efficiency in both multi-stages
5 and one stage compressions especially for obtaining a trade-off between power
6 density and efficiency. For one stage compression, the tested CR is relatively low
7 (maximum 10), which are not adapted for an industrial application. Another
8 issue is the pumping power for the spray injection at high pressures and high
9 flow rates [22]. The trade-off between all these factors should be determined
10 carefully.

11 *3.2. Porous media*

12 Another largely used measure to enhance the heat transfer is the porous me-
13 dia inserts. The used porous media could have several forms (e.g., structured
14 interrupted-plates or unstructured foams) and with the use of different mate-
15 rials and densities (Fig. 9). They can cover the whole compressed/expanded
16 gas domain or just a part of it. It has been reported that an insert close to
17 the piston top is the most efficient [71]. On one hand, the porous inserts serve
18 as a good media for heat transfer between the gas and liquid. On the other
19 hand, they can reduce the time needed to complete the compression/expansion
20 process, increasing thereby the power density for a given thermodynamic effi-
21 ciency, or augmenting the thermodynamic efficiency at a fixed power density.
22 Note that the power density ($\frac{E}{tA}$) is defined as the energy stored or recovered
23 during a process (E), divided by the total time it took to complete the process
24 (t) and the surface area of the piston column (A). It has been reported that
25 in compression, the porous inserts could increase the power density by 39-fold
26 at 95% efficiency, or augment the efficiency by 18% at $10^5 W/m^3$ power density
27 [72]. In expansion, the power density could be increased threefold at 89% effi-
28 ciency or the efficiency be increased by 7% at $1.5 \times 10^5 W/m^3$ power density [73].

29

30 Zhang et al. [74] compared several heat transfer correlations to estimate

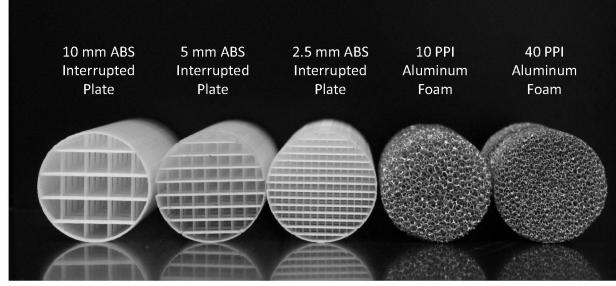


Figure 9: Five types of porous inserts used in compression/expansion experiments [50], including 3 Acrylonitrile Butadiene Styrene (ABS) interrupted-plates with 2.5 mm, 5 mm and 10 mm spacing between plates and 2 aluminium foams with 10 and 40 pores per inch (PPI)

1 the heat transfer in their studied geometry with porous media inserts. These
 2 correlations were proposed by many researchers, including Wakao & Kaguei
 3 [75], Kuwahara et al. [76], Zukauskas [77], Nakayma et al. [78], Kamiuto &
 4 Yee [79] and Fu et al. [80] (cf. details in Table 3). Among them, the Kamiuto
 5 & Yee correlation [79] fitted the best their experimental results with 10 PPI
 6 (Pores Per Inch) insert, and was then chosen for a further numerical study [81].
 7 A new correlation has been proposed based on the obtained simulation results
 8 (Eq. 26):

$$9 \quad Nu = 9.700 + 0.0879 Re_{Dh}^{0.792} Pr^{1/3} \quad (26)$$

10 Later on, an improved correlation has also been proposed (Eq. 27) [73]:

$$11 \quad Nu = 8.456 + 0.325 Re_{Dh}^{0.625} Pr^{1/3} \quad (27)$$

12 Where Re_{Dh} is the Reynolds number based on the hydraulic diameter D_h of
 13 the piston column.

14 Besides, the Stanton number (St) and a dimensionless heat flow rate term
 15 (Q^*) were used to characterize the effect of porous inserts.

$$16 \quad St = \frac{h_{i,G}}{\rho c_p \frac{U}{2}} \quad (28)$$

17 A linear relation St versus Q^{*-1} has been found [73], written as:

$$18 \quad -St = a(Q^{*-1}) + b \quad (29)$$

1 Where a , b are coefficients depending on the compression velocity.
 2
 3 Yan et al. [50] conducted an experimental investigation to measure the
 4 effects of porous media inserts on the heat transfer, the power density and the
 5 compression/expansion efficiencies. Compression experiments were run using
 6 5 different inserts and a baseline (Fig. 9). Power densities ranged from $3.8 \times$
 7 $10^3 W/m^3$ to $1.82 \times 10^5 W/m^3$ while similar constant flow profiles were imposed.
 8 In compression, porous media could increase the power density by a factor of
 9 39 at a constant efficiency, or increase the efficiency by as much as 18% to
 10 reach a maximum of 95% (Fig. 10). For the expansion, the results showed an
 11 increase in power density by a factor of 3 for a fixed efficiency, and an increased
 12 process efficiency by as much as 7% up to 90%. Fig. 10 (b) shows the expansion
 13 efficiency vs. the normalized power with or without porous inserts, indicating
 14 that an efficiency improvement of 5% could be achieved at a certain mean power
 15 density.

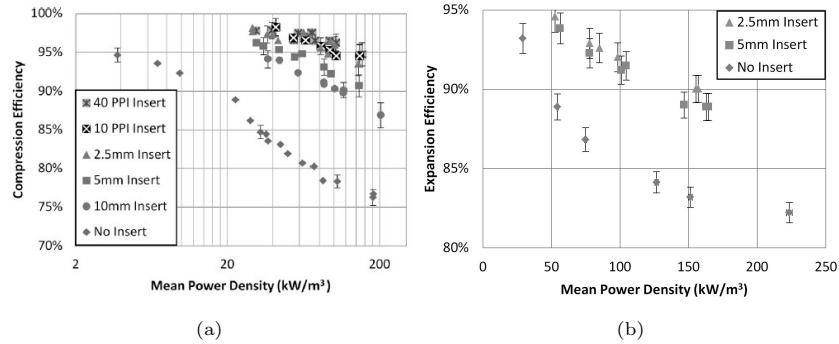


Figure 10: Comparison of compression/expansion efficiencies with or without the porous media inserts [50]. (a). Compression efficiency vs. power density (CR=10); (b). Expansion efficiency vs. power density (expansion ratio=6)

16 More recently, a LP structure with a two layer porous medium was proposed
 17 by Ren et al. [82], allowing enhanced air cooling so as to approach the isother-
 18 mal compression. It was reported that this configuration could also reduce the
 19 pressure and the compression work needed.

1 A thermodynamic and heat transfer analysis was conducted by Khaljani
2 et al. [83] to evaluate the use of porous media as an effective for thermal
3 management. The porous medium has been modeled as metal parallel plates
4 inserted in the piston column. The results showed that the inclusion of five and
5 nine plates could reduce the final temperature of the air by 15 K and 20 K,
6 respectively, compared to the base case without plate inserts. The porous insert
7 also had a significant influence on the wall temperature and on the flow pattern
8 evolution. With 9 plates insert, the η_c could be increased by 2%, going from
9 86.91% for the base case to 88.91%. Nevertheless, the energy storage capacity
10 would decrease, since a part of the piston volume was occupied by the porous
11 media, leaving smaller space for the gas phase.

12 In brief, porous media insert could be considered as the most widely used
13 HTE technology for LP, adapted to different CR for both the compression and
14 the expansion. It can offer high power densities while maintaining the efficiency
15 and can be easily combined with the other HTE concepts as will be introduced
16 in following sub-sections. Further researches could be focused on finding the
17 optimal geometries of the porous media by using shape or topology optimization
18 methodologies [84].

19 *3.3. Optimal trajectories*

20 Another proposed HTE concept consists in optimizing the manner of air
21 compression from the view point of finite-time thermodynamics by choosing the
22 best trajectory (ζ) (Fig. 4), with the purpose of significantly improve the energy
23 storage/generation capacity. Moreover, it tries to find the trade-off between the
24 HTE and the pressure drop increase by imposing a pressure-linked flow rate, i.e.,
25 a high flow rate at the beginning of the compression/expansion but gradually
26 decreasing with time.

27 Saadat et al. [85] proposed a general numerical approach to determine the
28 optimal compression profile by using more general heat transfer models. The
29 viscous friction and system constraints were also considered in the optimization
30 process. A fully developed laminar pipe flow model was used to estimate the

1 heat transfer coefficient inside cylinders with high $L/D \gg 1$ (Eq. 30):

$$2 \quad h = 3.66 \frac{\lambda}{D} \quad (30)$$

3 An average Nusselt number Nu_{ave} was also proposed based on $Re_{air,ave}$ for
 4 cylinders with small $L/D \sim 1$ where the flow is supposed to still be laminar
 5 (Eq. 31):

$$6 \quad Nu_{ave} = 0.664 Re_{air,ave}^{1/2} Pr^{1/3} \quad (31)$$

$$7 \quad Re_{air,ave} = \frac{\rho_{air} D U_{air,ave}(t)}{2\mu_{air}} = \frac{2m}{\pi D \mu_{air}} \frac{U_{air,ave}(t)}{L_0 - L(t)} \quad (32)$$

9 Where $U_{air,ave}(t)$ is the air average velocity and $L(t)$ is the liquid/air interface
 10 position. Based on the Pareto optimization procedure, their results [85] showed
 11 that the optimal compression profile increased the storage power up to 10% for
 12 a η_c of 90% compared to the A-PI-A (Adiabatic-PseudoIsothermal-Adiabatic
 13 process) and more than 40% when compared to a conventional compression
 14 profile such as sinusoidal or constant speed compression trajectories (Fig. 11).
 15 Several other trajectories were tested and compared. Note that the study was
 16 limited to a relatively small CR=10, the effect of optimal trajectory could be
 17 expected to be more significant at higher CR and with more complex geometries.

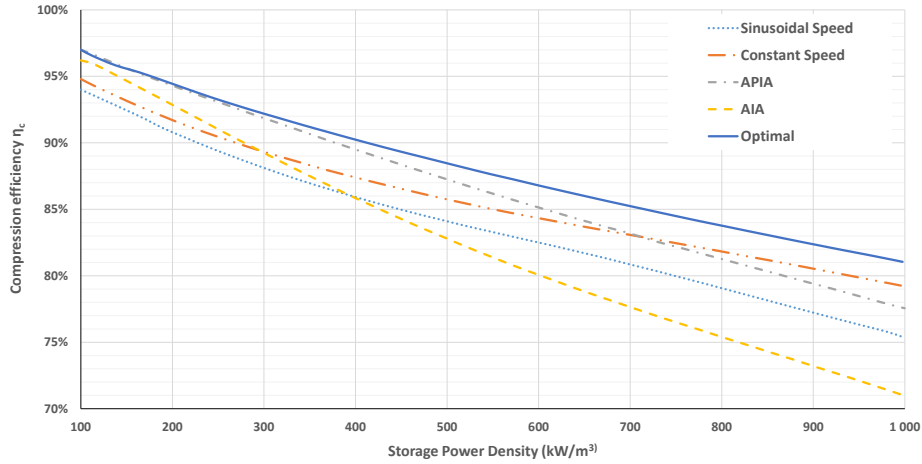


Figure 11: Comparison on the efficiency as a function of storage power density for different compression trajectories (CR=10). Adapted from [85]

1 3.4. Honeycomb geometry

2 The honeycomb concept consist in substituting the large piston column by
 3 many long and thin tubes, the liquid flowing through these tubes to compress
 4 air. The thin tubes could have a circular geometry (Fig. 12a), or as hexagonal
 5 cells stacked in layers (Fig. 12b). The main advantages of this concept are the
 6 easy implementation of the inserts which gives a higher power density and an
 7 increased L/D in each cell which enhances the heat transfer.

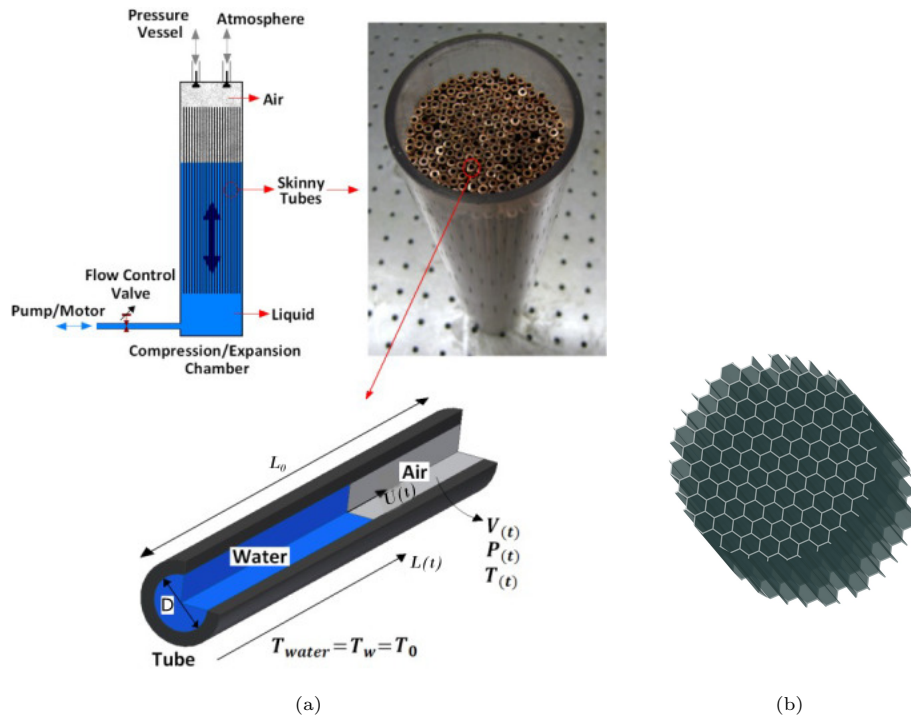


Figure 12: Honeycomb insert configuration for HTE in LP. (a). thin tubes [86]; (b). honeycomb matrix [72]

8 Zhang et al. [86, 72] numerically studied a LP compressor inserted with
 9 honeycomb geometry by CFD approach. Each tube channel had a L/D ratio of
 10 100. The liquid (moving boundary assumed as water) had a constant velocity
 11 and the walls had a constant temperature. In their study a transient Re_{Dh}
 12 number ranging from 130 to 1300 has been calculated based on the diameter (of

1 the largest cylinder), the mean flow velocity, and a transient density. Moreover,
 2 a $Nu-Re$ correlation was proposed for predicting the heat transfer based on the
 3 local Darcian velocity Re_{l_c} , where l_c is the characteristic length [87]:

$$4 \quad Nu = 2f_r \left(\frac{l_c}{L_d} Re_{l_c} Pr \right)^{1/2} \quad (33)$$

5 Where f_r is the friction factor defined as Eq. 34 :

$$6 \quad f_r = \frac{0.564}{\left[1 + (0.664Pr^{1/6})^{9/2} \right]^{2/9}} \quad (34)$$

7 Zhang et al. [72] has compared two concepts, a honeycomb insert and a
 8 porous media interrupted-plate heat exchanger, on the efficiency improvement
 9 of LP. The impacts of different parameters (material, length scale of the plate
 10 elements, specific surface area and porosity) on the temperature evolution were
 11 studied. A honeycomb insert with higher power density showed better air tem-
 12 perature abatement, and thus better efficiency. Nevertheless, the difference of
 13 η_c among all tested configurations was below 4% (between 60% and 64%).

14 In summary, the honeycomb concept with mini-tubes insert can take the
 15 maximum benefit from slow compression/expansion using a LP. They are easy
 16 to implement, but more investigations are still required to further investigate
 17 the effect of the materials and the CR on the heat transfer. Studies on the
 18 expansion will also help the application of this HTE concept.

19 3.5. Hollow spheres

20 The hollow sphere concept consists in adding a layer of floating spheres at the
 21 gas-liquid interface, which absorb the heat from the gas and transfer it to water.
 22 These hollow spheres move along with the interface during the compression or
 23 expansion. In this way, the overall heat transfer rate would be increased without
 24 adding much the compression work.

25 A numerical and experimental study has been conducted by Ramakrishnan
 26 et al. [88] on the effects of inserting hollow spheres (Fig. 13). An analytical

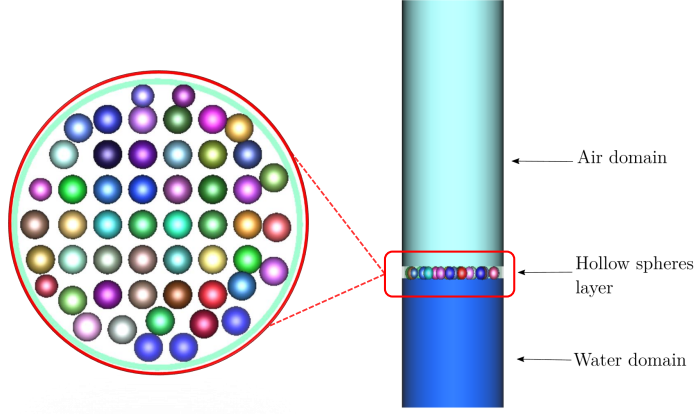


Figure 13: Hollow spheres concept for LP. Full LP domain and Hollow spheres layer at the water/air interface

1 model has been developed for the compression, by assuming the temperature-
 2 dependent physical properties of air. The viscosity has been calculated by using
 3 the Sutherland formula:

$$4 \quad \mu(T) = \mu_0 \left(\frac{T}{T_0} \right)^{1.5} \left(\frac{T_0 + 110.4}{T + 110.4} \right) \quad (35)$$

5 where μ_0 is the viscosity at $T_0 = 273 \text{ K}$. An average Nu correlation for free
 6 convection was chosen from the literature [88]:

$$7 \quad \overline{Nu}_D = 2 + \frac{0.589 Ra_D^{0.25}}{\left[1 + \left(\frac{0.469}{Pr} \right)^{9/16} \right]^{4/9}} \quad (36)$$

$$8 \quad Ra = \frac{g \rho^2 c}{T \lambda \mu} (T - T_W) D^3 \quad (37)$$

9 where Ra is the Rayleigh number, D the diameter of the spheres, and c the
 10 specific heat of air.

$$11 \quad c(T) = 1002.5 + 275 \times 10^{-5} (T - 200)^2 \quad (38)$$

12 The thermal conductivity of air λ is calculated by:

$$13 \quad \lambda(T) = 0.02624 \left(\frac{T}{300} \right)^{0.8646} \quad (39)$$

1 Many influencing parameters were tested including the spheres diameter, the
 2 stroke time, the CR and the sphere's material (for the experimental part). The
 3 results showed clearly the effect of the material of the spheres on the temperature
 4 abatement compared to the base case. HDPE (High-Density Polyethylene) and
 5 SiC (Silicon Carbide) showed better performance than PP (Polypropylene) (Fig.
 6 14). Moreover, the smaller spheres diameter ($D=6.5$ mm) showed better η_c
 7 than the larger one ($D=9$ mm) because of the higher total heat transfer surface
 8 area. A maximum temperature abatement of 32 K could be achieved by a fast
 compression with hollow spheres insert.

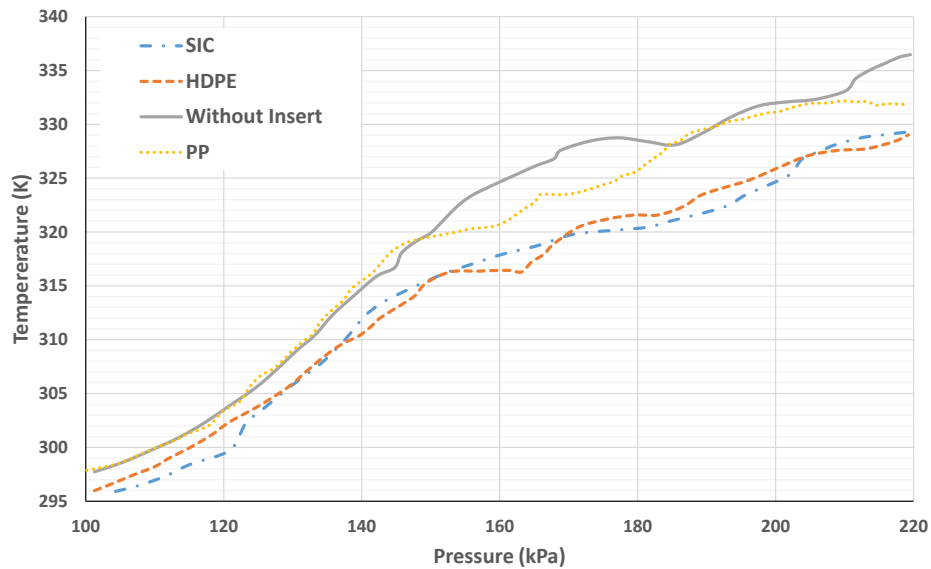


Figure 14: Experimental results on the effect of hollow spheres with different materials and sizes for CR=2. Adapted from [88]

9
 10 Although the study was limited to a small CR and a limited range of spheres'
 11 diameter/material, this concept has showed its effectiveness and easy imple-
 12 mentation. Yet, more investigations are still needed to further test this concept
 13 under high pressures, high CR and high flow rates conditions.

1 *3.6. Aqueous foam*

2 This concept consists in generating foam (with water additives) from the
 3 bottom of the piston which rises to the liquid/gas interface and increases the
 4 heat transfer surface between the liquid column and the compressed gas. This
 5 novel technique has been proposed and patented by SustainX [89] and studied
 6 experimentally by Patil & Ro [90] for temperature abatement in a LP com-
 7 pressor. Compared to the droplet spray, the mixture of air and water as a
 8 homogeneous foam had several advantages, including the higher contacting sur-
 9 face area which can be maintained during the entire compression or expansion
 10 stroke. With the use of aqueous foam, it has been observed that the rise of air
 11 temperature could be reduced especially in the end of the compression process
 12 with higher temperature gradients, i.e., about 7 to 20 K at a CR=2.5. This
 13 led to a η_c increase of 4-8% to 86% of isothermal η_c . Moreover, the use
 14 of several foam generators (up to 4) could further increase the η_c up to 90 %.
 15 The operating principle and the influence of the aqueous foam on the gas flow
 16 dynamics are illustrated in Fig. 15.

17 Patil & Ro [90] used Skelland correlation (Eq. 40) [91] and Attia correlation
 18 (Eq. 41) [92] to model the heat transfer with aqueous foam in the LP. Never-
 19 theless, none of them could give appropriate results due to the complexity of
 20 the physical phenomenon and the geometry.

$$21 \quad Nu_x = 1.41 \left(\frac{3k+1}{4k}\right)^{1/3} \left(\frac{D_h Re Pr}{x}\right)^{1/3} \quad \text{and} \quad (40)$$

$$Nu_\infty = \frac{8(5k+1)(3k+1)}{31k^2+12k+1}$$

22 where Nu_x and Nu_∞ are the Nusselt number in the entry region and in ther-
 23 mally fully developed region, respectively. k is the flow behavior index and x is
 24 the length from the entrance.

$$25 \quad Nu_{fo,x} = a C(\chi)^{-\frac{1}{6}} (Ca_{fo})^{\frac{1}{8}} \left(\frac{3k+1}{4k}\right)^{1/3} \left(\frac{D_h Re_{fo} Pr_{fo}}{x}\right)^{1/3} \quad (41)$$

26 where a is a constant to be determined from experimental data, $C(\chi)$ is an
 27 empirical function dependent on surfactant mass fraction ξ in weight percentage
 28 and Ca_{fo} is the capillary number. The subscript fo denotes the foam.

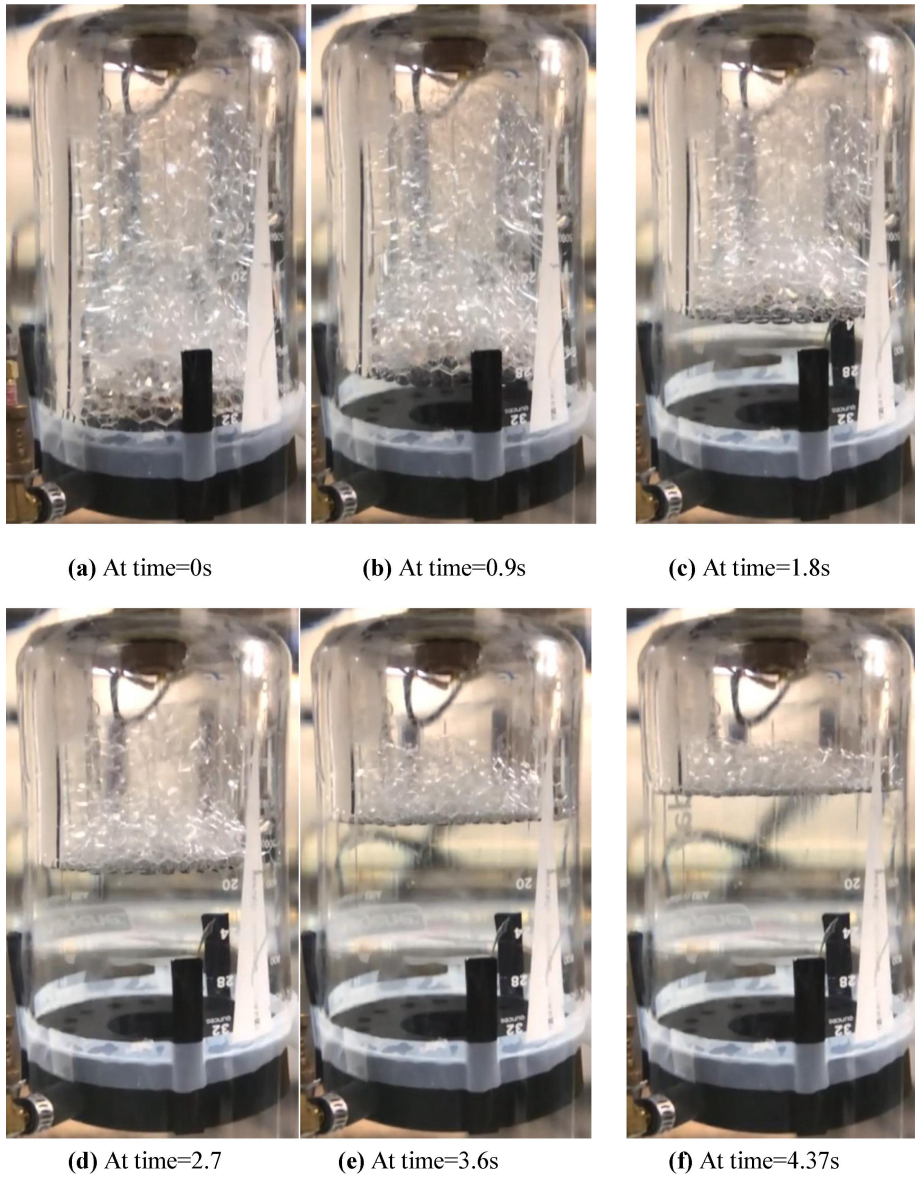


Figure 15: Aqueous foam bubble dynamics during the compression process [90]

- 1 The presence of foams in the gas domain inside the LP shows relatively
- 2 high heat transfer rate with a low mass flow rate. Therefore, the ratio of heat
- 3 transfer to coolant density is higher for foams. However, the accumulation of
- 4 residual foams after a few cycles could change the heat transfer characteristics

1 and flow dynamics inside the LP and could lead to the corrosion in some parts
2 of the system. Effects of cyclic operation and variability of foams geometries
3 have to be further studied so as to obtain more experimental feedback. Better
4 scalability of the foam-based HTE could be ensured by further studies focusing
5 on the use of microfoams, controlled macrofoams generation and under higher
6 CR operations.

7 *3.7. Wire mesh*

8 Metal wire meshes can also be installed in the piston column in contact
9 with both the gas and the liquid to enhance the heat transfer (Fig. 16a). For
10 cylindrical column, an Archimedean spiral form could be used, facilitating the
11 heat transfer in both the axial and radial directions (Fig. 16b). This concept
12 could in fact be considered as a special case of porous media inserts.

13 Patil et al. [69] compared experimentally the effect of wire mesh on the
14 pressure, the temperature and the efficiency of a LP compressor. Two types
15 of wire mesh were tested, one in copper and another in aluminium (Fig.16)
16 (b). Their results showed a significant temperature abatement (about 25 K)
17 by adding the wire mesh. Meshes with larger wire size (0.7112 mm or 3.5
18 wire/cm) were more efficient than that with smaller wire size (0.4572 mm or
19 6 wire/cm). The compression speed also had an important influence on the
20 final gas temperature. At medium velocity condition (4 s compression time),
21 about 28 K temperature abatement could be achieved (Fig. 17). In general,
22 the isothermal η_c could be increased from 82-84% to 88-92% by using the wire
23 mesh [69].

24 Only two materials (copper and aluminium) have been tested while more sys-
25 tematic investigations on many factors (e.g., wire diameters, mesh sizes, forms of
26 the spiral and mesh position) are still needed, so as to achieve an optimal mesh
27 design and to obtain the maximum efficiency. Also, the utility of metal wire
28 meshes for high CR should be studied to ensure that the obtained efficiencies
29 will also be maintained at high power levels.

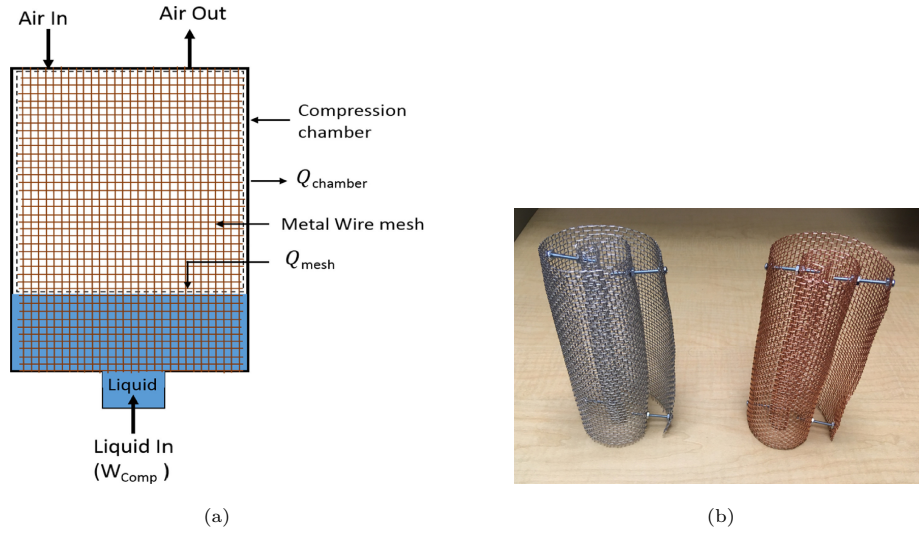


Figure 16: Wire mesh insert for LP [69]. (a). schematic model; (b) photo of aluminum and copper wire mesh in Archimedean spiral form

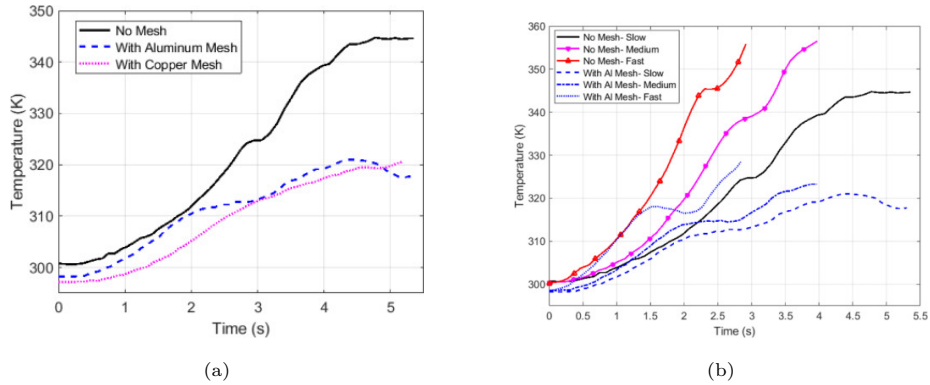


Figure 17: Wire mesh insert on the temperature evolution of LP [69]. (a) Effect of mesh materials; (b) Effect of compression speed

1 3.8. Optimal geometry of the piston column

2 Another method to augment the compression/expansion efficiency is to op-
 3 timize the shape of the piston column instead of the conventional cylinder one
 4 with uniform cross-section area, shown as an example in Fig. 18. Zhang et al.
 5 [93] made a CFD-based design analysis for a LP compression chamber by vary-
 6 ing five geometrical parameters, including the inlet radius R_i , the maximum

1 radius R_m , the top cap radius R_t , the location of the maximum radius X_m with
 2 respect to the inlet and the column length L (Fig. 18). 16 designs were proposed
 3 and studied. Their numerical results showed that a longer column and a steeper
 4 change in cross-sectional radius would agitate the gas flow, enhancing thereby
 5 the heat transfer (Fig. 19). The optimal shape of the column as shown in Fig.
 6 19b (b) had the following parameters: $P_1 = \frac{L}{R_m} = 8$, $\frac{R_m}{R_t} = 6$, $P_3 = \frac{X_m}{L} = 0.5$,
 7 $P_4 = \frac{R_m}{R_t} = 4$, improving the η_c from 62% to 69%.

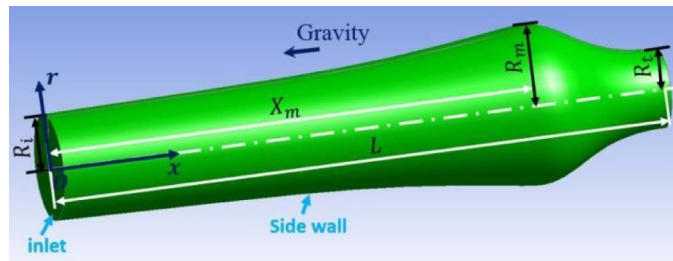
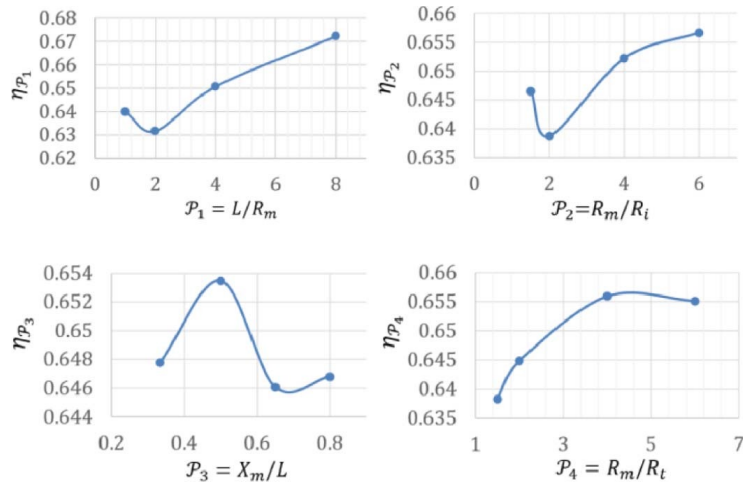


Figure 18: Schematic of LP chamber and shape parameters [93]

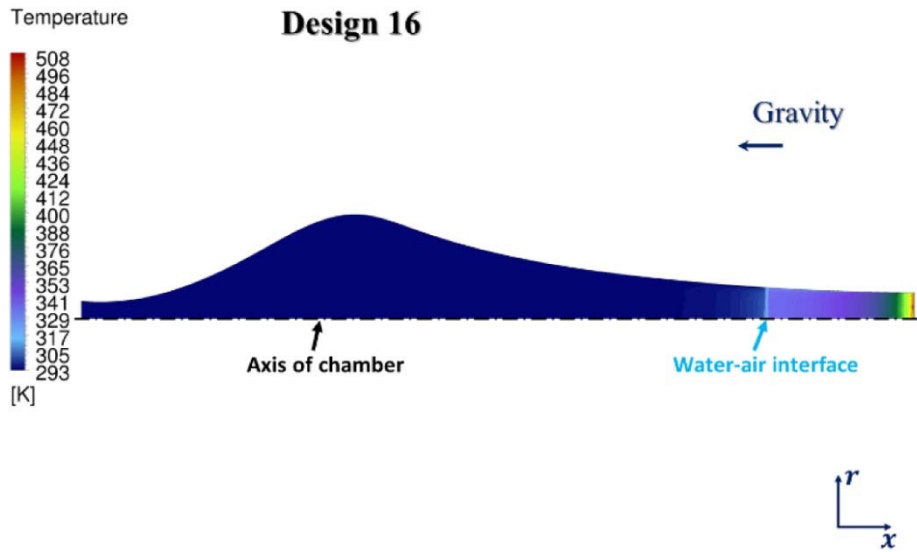
8 Generally speaking, parametric study or topology optimization (geometry,
 9 piston velocity, materials, CR) are the best way to better understand and opti-
 10 mize the processes to help achieve the highest efficiencies. While it is costly to
 11 do such studies experimentally, numerical modeling based on CFD simulations
 12 could be more adapted and play an important role in the future. Once the op-
 13 timal geometry of the column is determined, another problem is the fabrication
 14 difficulty and cost. Therefore, trade-off needs to be achieved for this concept
 15 for large-scale industrial application considering the computational cost, the
 16 efficiency improvement, the production cost, the material resistance and the
 17 installation in a real site.

18 3.9. Summary of HTE concepts for LP

19 Patil et al. [94] have compared several HTE concepts used in LP for CAES
 20 application. The comparison was done on the RTE value through a typical
 21 configuration of a global CAES system, composed of 4 LPs having a $L/D = 10$
 22 and a $CR = 6$. The total stored energy was 2 MWh [44]. By improving nothing



(a)



(b)

Figure 19: CFD-based shape optimization of the LP column [93]. (a). Effect of single shape parameter on the efficiency improvement; (b). Optimal geometry of the LP column and the temperature field at the end of compression

- 1 more than the compression/expansion efficiency through thermal management,
- 2 the RTE of the CAES system could rise from 45% to up to 62% (Fig. 20).
- 3 The RTE improvement is different by using different HTE concepts, i.e., 5%

1 by the optimal trajectory concept, 7% by hollow spheres insert, and more than
 2 17% by spray cooling or by porous media inserts. Further improvement of
 3 LP efficiency could be expected by combining two or several HTE methods,
 4 stepping towards near-isothermal compression/expansion. For example, Ahn et
 5 al. [95] combined wire mesh insert with the droplet spray but the effect of each
 6 HTE measure could not be identified clearly. Further efforts should thereby
 7 be devoted to the development and implementation of combined HTE methods.
 8 Table 2 recapitulates the studies on the HTE technologies in LP with their main
 9 results, conditions, advantages and shortcomings.

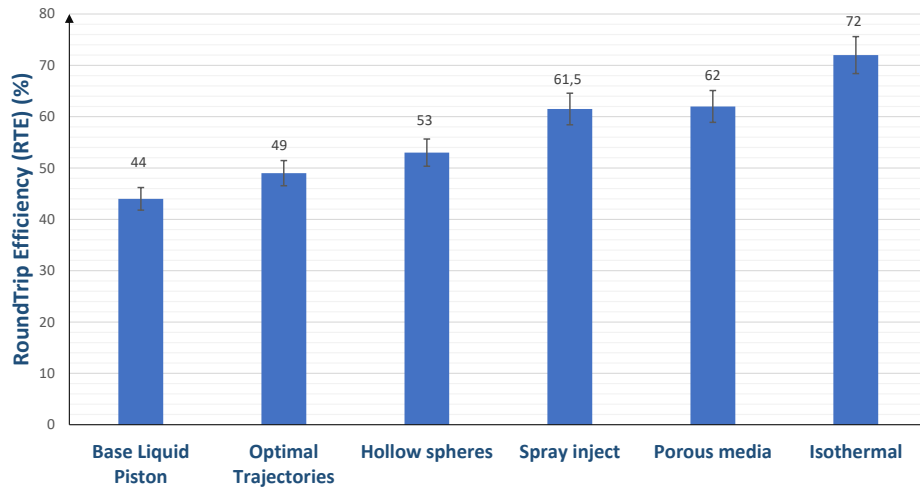


Figure 20: RTE efficiency improvement of a LP-based CAES system by implementing various HTE concepts (uncertainty bars represent 95% confidence interval), adapted from [94]

| HTE method | Spray subject | Optimal trajectories | Porous media insert | Honeycomb geometry | Hollow spheres | Wire mesh | Aqueous foam | Optimal column geometry |
|----------------------|--|--|---|---|---|---|--|--|
| References | [22], [96, 493], [98], [97] | [98, 99] | [100, 71], [73, 101], [81], [102] | [86] | [88] | [69] | [90] | [93], [98] |
| CR | 2-10 350 (Three stages) | 10 | 30 | 10 | 2-5 | 2-3 | 2-3 | 25 |
| L/D | 1 [96] | 10-20 | 6 | 100 | - | 2 | 2 | 1-8 |
| Compression time (s) | 0.3-5 | 0.3 | 2-428 | 3 | 5-15 | 5s | 4.5 | - |
| Maximal efficiency | 89% to 95% | 91.5 | 95 | 64% | - | 88-90% | 91.5% | 69 |
| Results | $\eta_c = 89\%$ for CR=350 [96] Up to 7% η_c increase to 95% for CR=33 | 10% η_c increase. 40% power density increase | 18% η_c increase. 20% η_c increase at high power densities | Up to 70K temperature abatement (for compression) | Up to 30K temperature abatement | Up to 6-8% η_c increase | Up to 20K temperature abatement (for compression). Up to 8% η_c increase | 6% η_c increase. L/D is a key parameter towards an optimal geometry |
| Advantages | Effective for HTE easy coupling with other methods | Easy coupling with other methods | Effective for HTE proved for high CR easy to implement | Effective for HTE easy to implement | Easy to implement | Easy to implement | Well compatible with LP | Take the advantages of numerical modeling and optimization algorithms easy coupling with other methods |
| Shortcomings | External power needed higher pressure drop | Advanced modeling needed to identify critical parameters | Maintenance and cleaning due to moisture accumulation | Only low CR tested | Only low CR tested, more experimental feedback needed | Only low CR tested, more experimental feedback needed | Only low CR tested, accumulation of residual foams, corrosion due to additives | lacking experimental results, fabrication cost and difficulty |

Table 2: Summary table of the HTE technologies from the literature with main results and conditions

1 Nusselt number Nu has been shown to be the most used non-dimensional
2 number for estimating the heat transfer inside the LP during compression/expansion
3 processes. Different correlations have been proposed and used based on their
4 own applicable conditions. The generic form used to write Nu as a function of
5 Re and Pr numbers is defined in Eq. 20 and summarized in Table 3 together
6 with other proposed correlations.

| Reference | Year | A | a | b | c | Remarks |
|-----------------------|------|--------|---|-------|------|--|
| Dittus & Boelter [60] | 1930 | 0.023 | 0.8 | 0.3 | 0.0 | Fully developed turbulent flow in circular tubes; valid only for $L/D > 60$ |
| Sieder & Tate [61] | 1936 | 0.023 | 0.8 | 0.3 | 0.14 | Fully developed flow $0.7 < Pr < 160$; $Re > 10000$; $L/D > 60$ |
| Anand [63] | 1963 | 0.70 | 0.7 | 0.7 | 0.0 | Internal combustion engines |
| Skelland [91] | 1967 | | $Nu_x = 1.41 \left(\frac{3k+1}{4k} \right)^{1/3} \left(\frac{D_h Re Pr}{x} \right)^{1/3}$ | | | Can be applied to foams in a pipe [90] |
| Adair et al. [56] | 1972 | 0.053 | 0.8 | 0.6 | 0.0 | Reciprocating compressors |
| Hamilton [62] | 1974 | 0.0245 | 0.8 | 0.6 | 0.0 | Inner duct turbulent flow |
| Wakao & Kagueli [103] | 1983 | | $1.748 + 1.19 Re_D^{0.6} Pr^{1/3}$ | | | Porous media; Re based on mean pore diameter |
| Liu & Zhou [57] | 1984 | 0.75 | 0.8 | 0.6 | 0.0 | Reciprocating compressors |
| Zukauskas [77] | 1987 | | $9.48 + 0.29 Re_D^{0.6} Pr^{1/3}$ | | | Porous media Re based on mean pore diameter |
| Hsieh et al. [64] | 1996 | 0.16 | 1.1 | 0.0 | 0.1 | Reciprocating compressors |
| Cengel [65] | 1998 | 0.664 | 0.5 | 1/3 | | Laminar flow in a pipe [36] |
| Cengel [65] | 1998 | 0.023 | 0.8 | 0.3 | | Turbulent flow in a pipe [36] |
| Kamimoto and Yee [79] | 2005 | 0.124 | 0.791 | 0.791 | | Open-cellular porous materials based on equivalent strut diameter |

| | | | | | | |
|--------------------------|------|---|-----|------|---|---|
| Lu et al. [104] | 2006 | 0.76 | 0.4 | 0.37 | 0 | Porous media $1 \leq Re_D \leq 40$; $D = (1 - e^{-\frac{1-\epsilon}{0.04}D_f})$ where D_f is the filament diameter |
| Lu et al. [104] | 2006 | 0.76 | 0.4 | 0.37 | 0 | Porous media $1 \leq Re_D \leq 40$; $D = (1 - e^{-\frac{1-\epsilon}{0.04}D_f})$ where D_f is the filament diameter |
| Lu et al. [104] | 2006 | 0.52 | 0.5 | 0.37 | 0 | $40 \leq Re_D \leq 10^3$; Porous media |
| Lu et al. [104] | 2006 | 0.25 | 0.4 | 0.37 | 0 | $10^3 \leq Re_D \leq 2 \times 10^5$; Porous media |
| Zhang et al. [81] | 2013 | $Nu = 9.700 + 0.0879Re^{0.792}Pr^{1/3}$ | | | | Porous media; $L/D > 1$; Interrupted-plate |
| Zhang et al. [73] | 2014 | $8.456 + 0.325Re^{0.625}Pr^{1/3}$ | | | | Porous media; $L/D > 1$; obtained numerically |
| Zhang et al. [72] | 2014 | $Nu = 2f_r \left(\frac{l_c}{L_d} Re_{lc} Pr \right)^{1/2}$ | | | | Honeycomb geometry |
| Ramakrishnan et al. [88] | 2016 | $2 + \frac{0.589RaD^{0.25}}{1 + \left(\frac{0.469}{Pr} \right)^{9/16}} \frac{1}{4/5}$ | | | | Used for hollow spheres [88] |
| Neu[27] | 2020 | $6.67 \left(Re_D Pr \frac{D}{L} \right)^{0.36}$ | | | | Laminar regime; $60 < L/D < 200$ |
| Neu [27] | 2020 | $6.17 \left(Re_D Pr \frac{D}{L} \right)^{0.48}$ | | | | Turbulent regime; $60 < L/D < 200$ |
| Patil & Ro[26] | 2020 | $Nu_{fo,x} = a C (X)^{-\frac{1}{6}} (Ca_{fo})^{\frac{1}{8}} \left(\frac{3k+1}{4k} \right)^{1/3} \left(\frac{D_h Re_f Pr_{fo}}{x} \right)^{1/3}$ | | | | Aqueous foam insert |

Table 3: Summary of some empirical heat transfer correlations used for solid pistons and liquid pistons

1 4. Conclusion and perspectives

2 This review paper is dedicated to gather and survey the advances of the LP
3 technology for CAES. The basic principles, the thermodynamic background,
4 the flow and heat transfer characteristics related to LP are presented in de-
5 tail. Especially the technological state-of-the-art on the HTE concepts imple-
6 mented in LP has been surveyed, aiming at approaching the isothermal com-
7 pression/expansion. The main findings may be summarized as follows.

- 8 • Compared to solid piston, LP can allow high L/D geometry and a flexible
9 compression/expansion speed. Moreover, it is a promising way to achieve
10 a near-isothermal process, which will have a bright foreground for I-CAES
11 application.
- 12 • The fluid flow and heat transfer inside the LP during the compression
13 or expansion operation are rather complex and difficult to characterize,
14 especially for the transition and turbulent regimes.
- 15 • Many $Nu-Re$ correlations were employed to predict the heat transfer of the
16 compression/expansion processes inside the LP, as summarized in Table
17 3. Almost all of them are based on the average value of Nu number or
18 convective heat transfer coefficient.
- 19 • Different concepts were proposed to enhance the heat transfer in the LP,
20 including liquid spray, wire mesh, porous media, optimal trajectory, hol-
21 low spheres and optimal geometry of the piston column. By using these
22 HTE techniques the compression/expansion efficiency of the LP could be
23 significantly improved up to 10%, leading to the RTE increase of the CAES
24 system. Each HTE concept has their own advantages and short comings
25 while the combination of two or several concepts seems more promising.
26 By implementing the proper HTE technique, the LP in a CAES system
27 can achieve a 95% compression/expansion efficiency.

1 Some scientific and technological barriers remain to be overcome for the
2 widespread industrial application of LP, which are also the key issues and chal-
3 lenges of the current research and development:

- 4 • Fine models that describe the evolution of the flow & heat transfer inside
5 a LP during compression/expansion processes are still lacking. Therefore,
6 systematic experimental and numerical studies are especially needed on
7 the local coupled fluid flow and heat transfer behaviors, for better charac-
8 terization, understanding and mastering the transient gas-liquid two phase
9 flow transport phenomena.
- 10 • Critical geometric and operational parameters (flow rate, CR, compres-
11 sion time, geometry, energy density, etc.) and the suitable combinations
12 of them remain to be identified. This requires the development and appli-
13 cation of advanced modeling and optimizing algorithms and experimental
14 verification.
- 15 • Most of the LP studies are still on the laboratory scale with small LP
16 dimensions and small CR values. The field testing at pilot or industrial
17 scale with realistic CR are still rare but fully necessary for the possible
18 application of LP in CAES systems.
- 19 • Further investigations on the new or combined HTE concepts, their testing
20 and validation at pilot-scale prototypes are still required.
- 21 • Finally, studies are still required on full cyclic compression/expansion cy-
22 cles considering all the auxiliary components (pumps, valves, etc.). De-
23 tailed thermodynamic and techno-economic analyses can also help de-
24 termine the optimal operation conditions and adapted control strategies
25 especially at pilot or industrial level.

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