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Experimental assessment of an active (acoustic) liner prototype in an acoustic flow duct facility

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ABSTRACT

In this paper, experimental results of broadband noise reduction in an acoustic flow duct are presented. An active liner composed of an array of electroacoustic absorbers is used. The control law is based on the pressure-based, current driven digital architecture for impedance control with a local control strategy. A wind tunnel test rig named Caïman has been used for the experimental validation. The results confirm the adaptability and the stability of the whole system with the local control strategy. The air flow slightly reduces the efficiency while maintaining the adaptability and the stability.

Keywords: Active liner, Electroacoustic absorber, Impedance control, Broadband noise reduction

1. INTRODUCTION

SALUTE (Smart Acoustic Lining for Ultra High By Pass Ratio (UHBR) Technologies Engines) is a research project funded by the European Union's Horizon 2020 research and innovation programme Clean Sky 2 Joint Undertaking. It aims to develop innovative acoustic treatments for engine nacelles in order to reduce noise pollution of future aircrafts. The main technological objective is to perfect a new generation acoustic liner, based on the Electroacoustic Absorber concept. UHBR Technologies Engines have larger fan and shorter nacelles which present new challenges in terms of noise signature (more low-frequency components and less space for acoustic liners). In order to meet the required noise performance at engine level, the next generation of acoustic treatments should achieve significant noise reduction with limited surface, in order to break the quarter-wavelength rule. It is in this context that this study takes place.

Active Liners can outperform conventional acoustic treatments (Single Degree of Freedom (SDOF) liner for example) and adapt their performances in real-time to varying engine speeds with an external source of energy. The active Noise Control (ANC) strategy proposed here is based on an electro-mechano-acoustical system. The pioneers works on the topic¹ introduced an "electronic sound absorbers" with a loudspeaker and a nearby microphone in a negative feedback loop. A few decades later, an "active equivalent of a quarter wavelength resonator" was achieved through a hybrid passive-active technique.² This strategy was retrieved by³ for liners applications.

Recently, the concept of electroacoustic absorber evolved with shunting techniques,^{4,5} direct feedback,^{6,7} or a sensorless control of the loudspeaker impedance.⁸ This pressure-based current-drive control architecture has

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been applied to design an active lining concept built an arrangement of electroacoustic absorbers. This concept allowed to achieve broadband noise reduction for aircraft engine nacelles applications under grazing flow.⁹

This paper presents a preliminary acoustic characterisation of a 2D active liner based on an array of electroacoustic absorbers. The control law is based on the pressure-based, current driven digital architecture for impedance control designed by Rivet et al.¹⁰ A local control strategy is experimentally validated without and with flow. This paper is organised as follow: section 2 describe the local control strategy, section 3 introduce the test bench facility and the experimental results are finally presented in section 4.

2. LOCAL CONTROL STRATEGY

The active liner is based on an arrays of unit cells. Each unit cell is composed of a transducer (a loudspeaker), 4 microphones and a control card with localized control strategy (Figure 1).

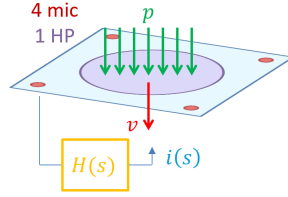


Figure 1. Schema of the cell control.

A short description of the local control is provided here, the pressure-based, current driven digital architecture for impedance control is fully explained in.¹⁰ The equation of the mechanical dynamics of the SDOF piston mode approximation for the loudspeaker is given by:

$$Z_m(s)v(s) = S_d p(s) - Bl i(s), \quad (1)$$

where s is the laplace variable, $v(s)$ is the mechanical velocity of the loudspeaker membrane, $p(s)$ is the average sound pressure on the membrane, S_d is the equivalent piston area, Bl is the force factor of the moving coil, $i(s)$ is the driving current.

The mechanical impedance of the SDOF loudspeaker model in open circuit is expressed as:

$$Z_m(s) = sM + R + \frac{1}{sC}, \quad (2)$$

where M , R and C are respectively the mass, resistance and compliance of the SDOF loudspeaker model.

The transfer function (local control) between the pressure and the current in order to achieve a target impedance Z_{at} on the loudspeaker membrane is defined as:

$$H_{loc}(s) = \frac{i(s)}{p(s)} = \frac{1}{Bl} \left(S_d - \frac{Z_m(s)}{Z_{at}(s)} \right), \quad (3)$$

where Z_{at} is the target impedance:

$$Z_{at}(s) = \frac{p(s)}{v(s)} = \mu_1 \frac{sM}{S_d} + R_{at} + \mu_2 \frac{1}{sCS_d}, \quad (4)$$

with μ_1 and μ_2 are two coefficients used to respectively tune/reduce the mass and the stiffness of the loudspeaker in the control, and R_{at} is a target acoustic resistance to be achieved at the resonance.

3. TEST BENCH FACILITY

The microphones calibration and the identification of the Thiele-Small parameters of each cell have been done in an acoustic duct. The Caïman wind tunnel is an experimental test bench composed by a straight duct with a square $66 \times 66 \text{ mm}^2$ section. The cutoff frequency (f_c) of the duct is 2600 Hz. The duct termination reproduces an anechoic condition. Liners until 320 mm of length along the axial direction can be tested with a maximum speed flow of Mach 0.4 (137 m/s). Acoustic quantities, such as absorption coefficient (α) and insertion loss (IL) are estimated with a 4 microphones technique following the *B&K* technical review.¹¹ Table 1 summarizes the geometrical parameters. The spacing between pairs of microphones (Δ_{mic}) depends on the frequency range. Experimental tests without flow have been done with a white noise signal between 100 Hz and 6000 Hz sampled at 51200 Hz and an equivalent acoustic level in the duct of 124 dB. The acquisition lasts 30 seconds (30 averages of 1 second). The following list summarizes the experimental equipment (figures 2-3):

- 1 : Computer
- 2 : DAC associated to the electronic cells
- 3 : NI Pxi card (IN/OUT)
- 4 : Audio amplifier
- 5 : Nexus B&K 2692-C (Charge microphones amplifier)
- 6 : Laboratory power supply ($\pm 5V, 2A$ max)
- 7 : Caïman cells and frame
- 8 : 1/4" microphones B&K 4939 and amplifiers B&K 2670
- 9 : Loud speaker
- 10 : Caïman wind tunnel (straight and square section $66 \times 66 \text{ mm}^2$)
- 11 : Anechoic termination
- 12 : Flow until Mach 0.4 (137 m/s)

x_1	x_2	x_3	x_4	d	Δ_{mic}
$x_2 - \Delta_{mic}$	-0.377 m	$x_4 - \Delta_{mic}$	0.846 m	0.266 m	2.2 cm or 10 cm

Table 1. Geometrical parameters.

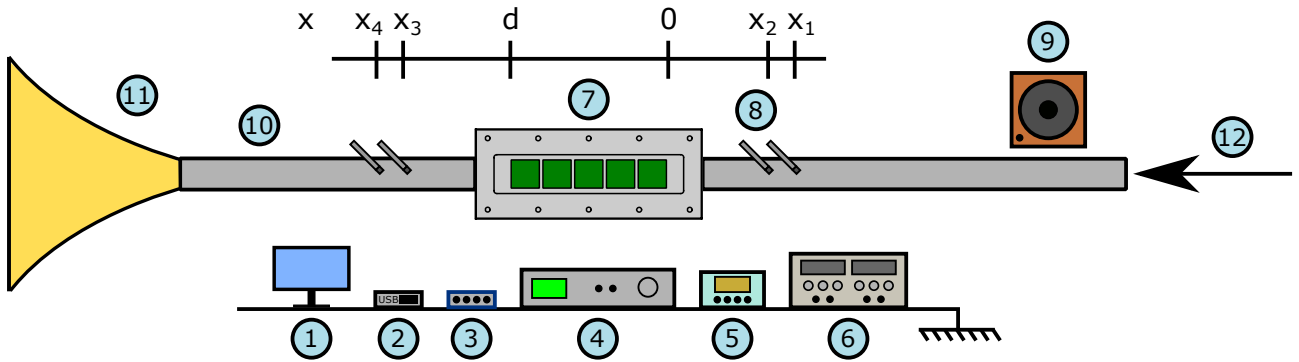


Figure 2. Experimental setup diagram of the Caïman wind tunnel tests.

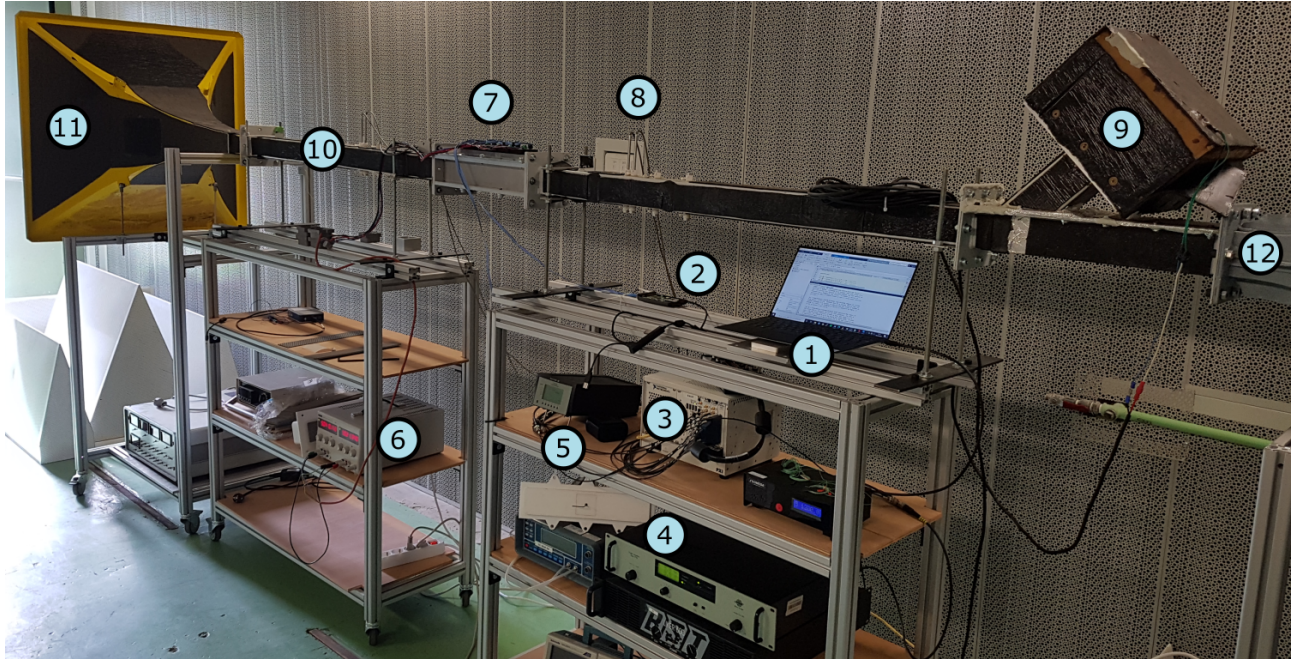


Figure 3. Experimental setup picture of the Caïman wind tunnel tests.

Figure 4a shows the inside view (without wiremesh) of the cells installed in the Caïman wind tunnel. Figure 4b shows the outside view with the electronic cards, the wires for the electrical supply and the wires for the communication between the computer and the cells. A rigid panel is installed on the opposite side of the active liner and a wiremesh is added in front of the cells.

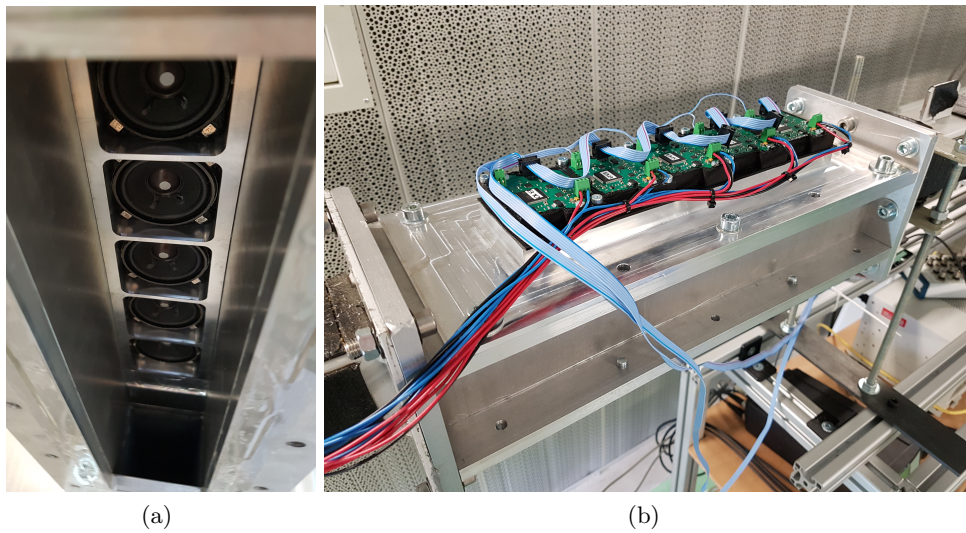


Figure 4. Caïman cells : a) inside view (without wiremesh), b) outside view.

4. EXPERIMENTAL RESULTS

4.1 Tests without flow

This subsection presents the results with a local control law without flow. The efficiency zone (see Table 2) can be tuned by μ_1 and μ_2 and are centred around the frequency:

$$f_{eff} = f_0 \sqrt{\frac{\mu_2}{\mu_1}}, \quad (5)$$

μ_2	0.2	0.4	1	2
f_{eff} [Hz]	475	675	1075	1510

Table 2. Efficiency frequency zone depending on the parameters of the control law ($\mu_1 = 0.4$).

Figures 5 shows the performances of the control when the target impedance takes the following values: $\rho_0 c_0$, $2\rho_0 c_0$ and $0.5\rho_0 c_0$. The μ_1 and μ_2 parameters of the local control law are equal to 0.4. The configuration without control is plotted in black. It can be noticed in Figures 5 that the performance increases around the f_{eff} frequency when the target impedance decreases. The choice of the target impedance is crucial in order to increase the performance compared to the passive solution (without control).

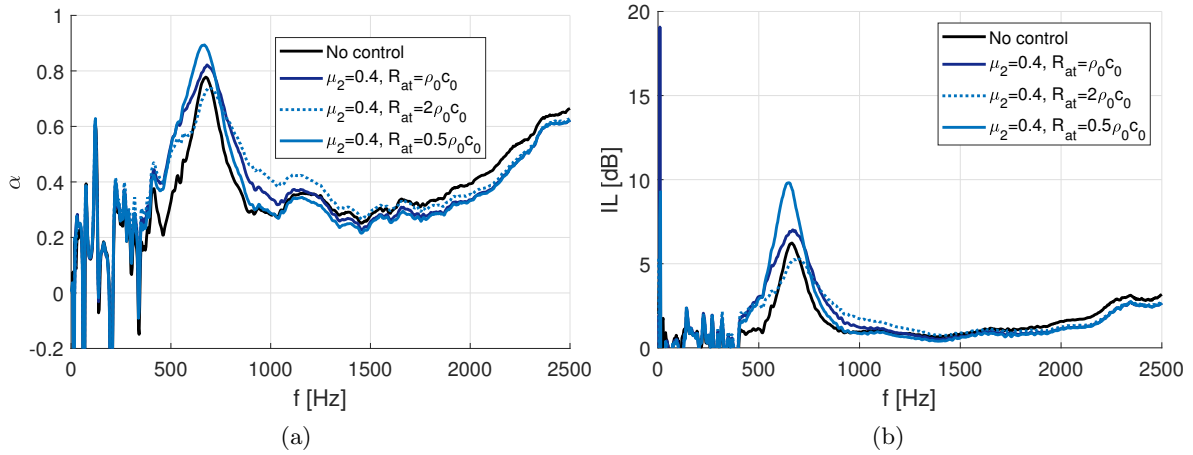


Figure 5. Variation of the target impedance ($R_{at} = [\rho_0 c_0, 2\rho_0 c_0, 0.5\rho_0 c_0]$). Control law parameters ($\mu_1 = \mu_2 = 0.4$): a) Absorption coefficient (α), b) Insertion loss (IL).

In the following the target impedance is now selected as $R_{at} = 0.5\rho_0 c_0$. Figures 6 shows the performances of the control when the parameters of the control law are varied. The μ_1 parameter of the local law remains equal to 0.4 and μ_2 takes the three following values: 0.4, 1, 2. In order to reduce the noise at low frequencies ($\mu_2 = 0.2$), the spacing between pairs of microphones (Δ_{mic}) has been increased from 2.2 cm to 10 cm. Consequently, a zero pressure measurement is present at 1700 Hz corresponding to a half wavelength ($\lambda/2$) equal to 10 cm (the spacing between pairs of microphones). The change in μ_2 value is equivalent to change the efficiency frequency from 500 to 1500 Hz.

The results confirm the adaptability and the stability of the whole system with the local control strategy.

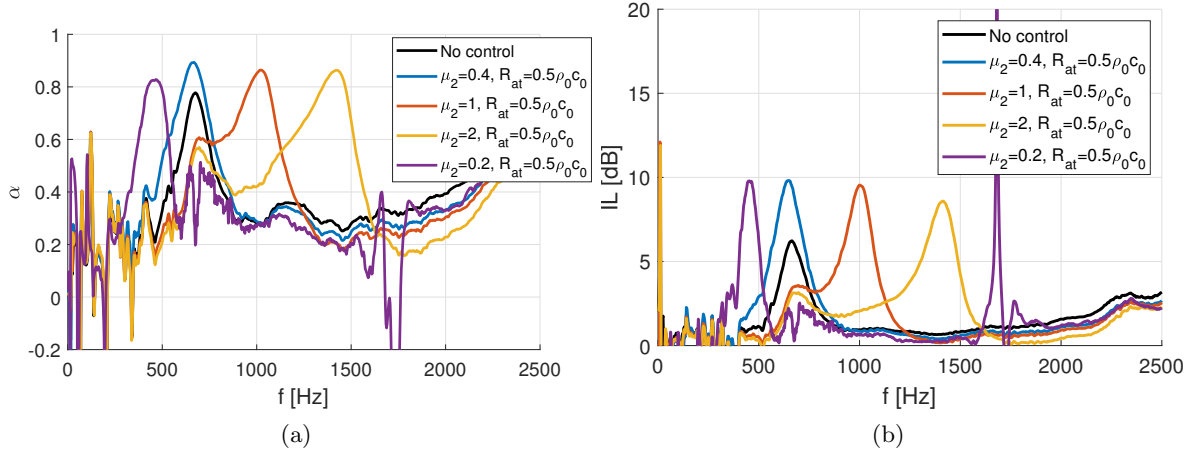


Figure 6. Variation of the parameters of the control law ($\mu_1 = 0.4$, $\mu_2 = [0.2, 0.4, 1, 2]$). Target impedance ($R_{at} = 0.5\rho_0c_0$): a) Absorption coefficient (α), b) Insertion loss (IL).

4.2 Tests at different flow speeds

This subsection presents the results with a local control law with flow. The target impedance has been selected as $R_{at} = 0.5\rho_0c_0$.

With flow, the ratio between the signal and the noise was too low with a white noise or a sweep sinus. Therefore, a pure sine has been used to increase the signal over noise ratio and decorrelate the turbulent boundary layer from acoustics. In order to reduce the noise at low frequencies, the spacing between pairs of microphones (Δ_{mic}) has been set at 10 cm.

The flow speed is equal to 20, 40, 60, 80, 100 and 120 m/s (*Mach* 0.06, 0.12, 0.18, 0.24, 0.29 and 0.35). Figures 7 present the results when the flow speed increases. The μ_1 and μ_2 parameters of the local control law are equal to 0.4. The reference test without flow is plotted with a deep blue line.

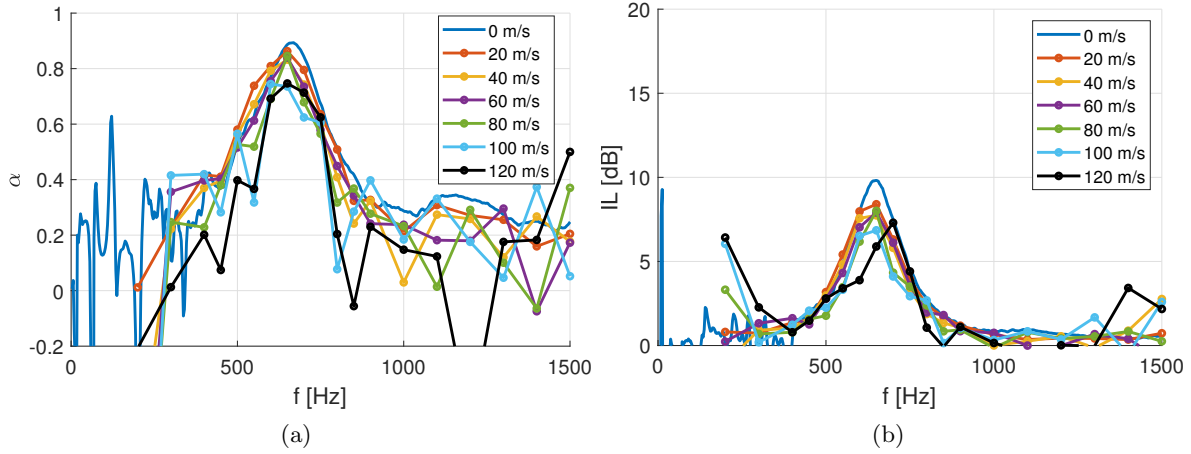


Figure 7. Variation of flow velocity [0, 20, 40, 60, 80, 100, 120] m/s, Control law parameters ($\mu_1 = \mu_2 = 0.4$): a) Absorption coefficient (α), b) Insertion loss (IL).

The air flow slightly reduces the efficiency while maintaining the adaptability and the stability.

5. CONCLUSION

In this article, experimental results in acoustic flow duct are presented. The active liner is based on an array of electroacoustic absorbers. The control law is based on the pressure-based, current driven digital architecture

for impedance control with a local control strategy. The adaptability and the stability of the system between 500 and 1500 Hz with the tested parameters of the local control law have been validated. The air flow slightly reduces the performance.

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