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Microstructure and mechanical properties characterization of architected copper aluminum composites manufactured by cold-drawing

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Abstract

The present study investigates the mechanical behavior of severely plastic-deformed Cu-Al composite wires with different diameters and heat-treatments. Each composite holds 61 restacked copper-clad aluminum wires. The bimetal composites were cold-worked up to diameters ranging from 1 mm to 3 mm, without any intermediate heat treatment. All the wires contain 61 hexagonal Cu-Al fibers with a continuous copper network extended from the outer surface into the center of the samples. Tensile tests were then performed on the as-drawn and heat treated wires. The latter were treated at 400°C for 30 min and 6h. The heat treatments firstly induce recrystallization in both constituents, giving rise to a fine-grained microstructure and secondly prompt the formation of several intermetallics. Without heat treatment after processing, the architected composites exhibit a ductility value similar to the conventional copper-clad aluminum wires and larger yield stresses compared to them, regardless of the diameter. The intermetallic compounds, forming as a result of heat treatment, affect the yield stress, ductility and strain hardening mechanisms. Finally, the results are discussed in terms of grain size, texture, intermetallics volume fraction and mechanical coupling between the phases.

Keywords: Metallic composite, wire, mechanical properties, intermetallic compounds, architected microstructure

1. Introduction

The rising price of copper due to a huge demand for this material on the one hand, and its high density (8.96 g.cm⁻³) on the other hand, has convinced manufacturers to replace it, when possible, with aluminum. For instance, Cu-Al bimetallic composites are currently used for such purposes. These composites can be obtained via various techniques including roll-bonding [1, 2, 3], friction stir-welding [4], stir casting [5], electroplating [6], cold-drawing [7], hydrostatic extrusion [8] or spark plasma sintering [9]. Cu-Al composites can reduce weight and costs up to 40% and 30% respectively [10].

In the wire industry, the aforementioned Cu-Al composite, known as Copper-Clad Aluminum Wire, abbreviated as CCAW or CCA, is commonly manufactured by cold-drawing using an outer copper case surrounding an aluminum core which are then simultaneously cold drawn to the desired diameter. In this composite, the copper shell provides

good conductivity whereas the aluminum core serves as a light structural material with lower conductivity than copper. CCAs offer acceptable high conductivity for high frequency currents where the Cu phase acts as the current path because of the skin effect. However, at lower frequencies, the Al core also contributes to the electrical conductance reducing, in turn, the overall conductivity.

In the literature, various architected CCAs have been proposed, creating a continuous copper network (or path) extended across the cross section, intended to increase the conductivity at low frequencies without having to reduce the volume fraction of aluminum. These new composites, manufactured by cold rotary swaging [11, 12] and also restacking cold drawing [13] were obtained using intermediate heat treatments (HT) aiming to recover ductility during the process. However, such heat treatments cause intermetallic compounds growth along the Cu/Al interfaces which could be detrimental to the electrical conductivity and the mechanical behavior.

In a recent work, the authors proposed a new and promising restacking method, without any interoperational annealing, to fabricate an architected Cu-Al composite with a continuous copper network inside the cross section [14]. As dynamic recovery occurs during large plastic deformation, the resistivity of the above-mentioned composite in the as-cold-worked state was found to depend on neither the plastic deformation nor the wire diameter throughout processing. A post-processing heat treatment of these composites increases the resistivity only slightly, due to the formation of intermetallic compounds (IMCs). If the volume fraction of IMCs remains below 12%, the increase in the resistivity stays below 10 % [14]. In service, Cu-Al wires could be subjected to high loads and hereby both good conductivity and proper mechanical properties are required. The large strains undergone by the wires during processing or the intermetallics formation associated with heat treatment can lead to early embrittlement of the wire making it unsuitable for industrial applications. Following the work of Hug and Bellido [15], a compromise between the mechanical properties (ultimate tensile strength and ductility) and the electrical conductivity could be achieved by suitable heat treatments.

The key objective of this work is hence to investigate the mechanical behavior of these new architected copper-aluminum wires under different heat treatment conditions. To this end, samples were cold-drawn to final diameters ranging from 3 mm to 1 mm. They were then mechanically and microstructurally characterized in both as-drawn and heat treated states. Heat treatment conditions of 400°C for 30 minutes and 6 hours were considered in order to understand the role played by plastic deformation and the microstructure in terms of grain size, texture and IMCs volume fraction. The best mechanical properties were observed for the sample heat-treated at 400°C for 30 minutes. The volume fraction of IMCs and the final diameter were also found to influence the mechanical behavior, which is discussed in terms of grain size, texture and composite effect.

2. Elaboration procedure and material/mechanical characterization

The composites were manufactured using 99.97% purity OFHC (Oxygen Free High Conductivity) copper and 99.5% purity aluminum previously annealed during 3h at 500°C for Cu and 300°C for Al. The heat treatment operations were carried out in a secondary vacuum ($P_{O_2} = 10^{-6} \text{ mbar}$) to prevent oxidation and to enable the recrystallization of both materials required to sustain the large strains involved by the process. The first step in the sample manufacture procedure is cold-drawing of a CCA wire up to a diameter of 1 mm, using a copper tube of an outer diameter of 12 mm, with a wall thickness of 2 mm, and a 7.5 mm diameter aluminum rod. The die diameter was gradually decreased throughout the first step to avoid fracture during the process. Each wire took 34 passes for the diameter to change from 12 to 1 mm. 61 1 mm-CCA wires were then restacked into a 12 mm-copper tube with a 1mm-thick wall and were further cold-drawn to obtain the intended architected copper-clad aluminum composite wires (abbreviated as ACCA in the following lines). Three sets of wire samples of final diameters of 1, 1.7 and 3 mm were fabricated. The average plastic strain experienced by the wire during the process ranges between $\eta = 7.37$ and $\eta = 9.57$ (see table 1), estimated from equation 1, in which S_0 and S represent the initial and final cross-sectional areas in the order given. Readers are referred to a previous article [14] for further details about the fabrication process. Figure 1 illustrates the typical cross section of an ACCA wire after the manufacture process observed by optical microscopy.

$$\eta = \ln\left(\frac{S_0}{S}\right) \quad (1)$$

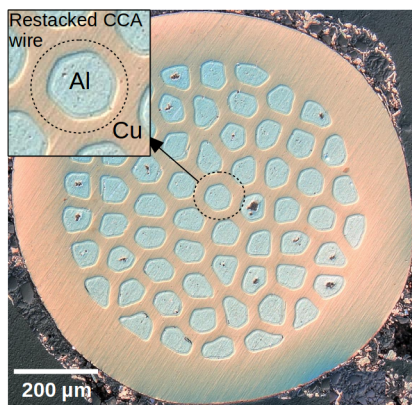


Figure 1: Example of an architected copper-aluminum composite obtained by the restacking method based on cold-drawn.

In order to investigate the role played by the intermetallic compounds and the microstructure on the mechanical behavior, three kinds of wires were investigated: as-drawn (AD), annealed at 400°C during 30 min (HT1) and annealed at 400°C during 6h (HT2). These heat treatments are expected to lead to different volume fractions of IMCs with a low initial dislocation density due to the recrystallization of the heavily strained Al and Cu grains.

These wires were then subjected to tensile tests using specific wire grips in order to reduce stress concentration throughout the test. An electro-mechanical 10 kN-MTS C43 machine was used to perform the crosshead displacement-controlled tests at room temperature. Strain was recorded using a conventional extensometer with a gauge length of 25 mm. During the test, the average strain rate was close to 10^{-4} in order to reduce viscous effects. Three tensile tests were carried out for each set of samples of different diameter and heat-treatment condition to take into account the experimental scattering of the mechanical properties. Nanoindentation tests (MTS NanoXP systems) were also performed on the cross section of the samples to obtain the hardness of each phase, prior to the tensile tests, for the different wire diameters and heat treatment conditions. Berkovich diamond tip (nominal tip radius of curvature of 20 nm) was employed and all hardness measurements were displacement-controlled up to a penetration depth of 200 nm. Spacing between each indent was set to $1 \mu\text{m}$ in order to avoid interaction between plastics zones of each indent but also to capture the hardness of the IMCs. Before analyzing each sample, a preliminary calibration experiment on the same fused quartz specimen was performed to ensure reliable measurements.

The microstructure of all samples was analysed by Scanning Electron Microscope (SEM) coupled with Electron Dispersive Spectrum (EDS) and Electron BackScattered Diffraction (EBSD). Due to the small grain size, for the as-drawn samples, Transmission Electron Microscopes (TEM) using different analysis modes, i.e. ASTAR (FEI TECNAI) and STEM (Scanning Transmission Electron Microscopy - JEOL ARM 200F) were also employed for the microstructure analysis. Fracture surfaces of the samples were also analysed by SEM in order to investigate the fracture mechanisms. For SEM and nanoindentation, samples were prepared by conventional SiC polishing followed by a final step of vibratory polishing or electropolishing.

3. Results

3.1. Microstructural characterization

The phase, grain size and texture of all ACCA samples of different diameters and heat treatment conditions (AD, HT1 and HT2) were determined using SEM and TEM (for the sake of conciseness, only a few analysis are reported in the following paragraphs). Figure 2 illustrates the initial microstructure of the as-drawn sample, observed by TEM. A typical Cu/Al interface, at higher magnification, is indicated on figure 2(a) in the bright field STEM mode. Copper is in the top left area of the image whereas aluminum is in the bottom right corner. As shown in this figure of the cross sectional plane, grains are equiaxed in the cross-section, with an average spacing between boundaries of around a few hundred nanometers for each phase. Figure 2(b) shows at higher magnification, in bright field STEM mode, of a typical Cu/Al interface. As reported in a previous article [16], no IMC could be detected along such an interface, except for a small fraction of nanoscaled Al_2O_3 resulting from the cold-drawing process. The grains are mostly

dislocation-free, however, few grains also exhibit dislocation walls/cells (encircled in the figure 2(b)).

Grain orientation maps and pole figures of each phase were acquired, from the cross section, using ASTAR analysis. Cu and Al (001), (011) and (111) pole figures are plotted in figures 2(c) and figure 2(d) for the 3 mm diameter as-drawn sample, respectively. A fiber texture with major $\langle 111 \rangle$ and minor $\langle 001 \rangle$ orientations, was characterized for both phases, in agreement with the results from previous researchs on the influence of cold drawing on the texture of pure Cu [17, 18] and pure Al [19]. Figures 2(e) and 2(f) represent the Cu and Al grain orientation maps, respectively. The post-processing fraction of very low angle grain boundaries (misorientations $\leq 5^\circ$ - VLAGBs) characterized in each phase, is about 20%. This feature is in accordance with the results from the study conducted on wire drawing and accumulative roll bonding of pure copper by Hanazaki et. al. [20]. The fraction of high angle grain boundaries is about 10%.

Figure 3(a), which shows the SEM-EDS map of a heat-treated sample, highlights the copper and aluminum distribution in the cross section of a 3-mm diameter ACCA wire after HT2. InterMetallics Compounds (IMC) are clearly exhibited. For the same sample, figure 3(b) and figure 3(c) illustrate the Cu and Al grain orientation maps, respectively. Preferred grain orientations are clearly visible in both phases. As reported in the as-drawn condition, these preferential orientations lead to a large volume fraction of VLAGBs presented in figure 3(d) for Cu and Al. A large amount of annealing twins were also formed in copper because of its low stacking fault energy.

Figure 4 illustrates the three principal IMCs, Al_4Cu_9 , AlCu and Al_2Cu that have been characterized in two previous papers for the same samples [14, 16], in agreement with literature [21, 22, 23, 24]. For short heat treatment conditions (300 °C during 15 min to enable the TEM observation of the IMC interface), the bright field TEM image of a 1.7 mm diameter ACCA (figure 4(a)) reveals that the AlCu layer is the thinnest one. This trend is not modified for longer heat treatment duration as illustrated in figure 4(b) for a 1.7 mm ACCA wire submitted to HT2. In this case, the thickness of the IMCs layer is about 25 μm . More details about the IMCs characterization can be found in a previous article [16].

(001), (011) and (111) pole figures of the cross section for a 3 mm-diameter ACCA wire, heat-treated under HT2 conditions, are plotted in figure 5(a) and figure 5(b) for copper and aluminum, respectively. The same pole figures for a 1 mm-diameter ACCA wire, after HT2, are plotted in figure 5(c) and figure 5(d) for Cu and Al in the order given. Because fewer grains were analyzed for aluminum, discret spots, rather than a continuous distribution (as seen for Cu), appear in the Al pole figures. Comparing the texture component and their related density for Al and Cu for each heat treatment, figure 5 illustrates that the texture evolution depends on the material.

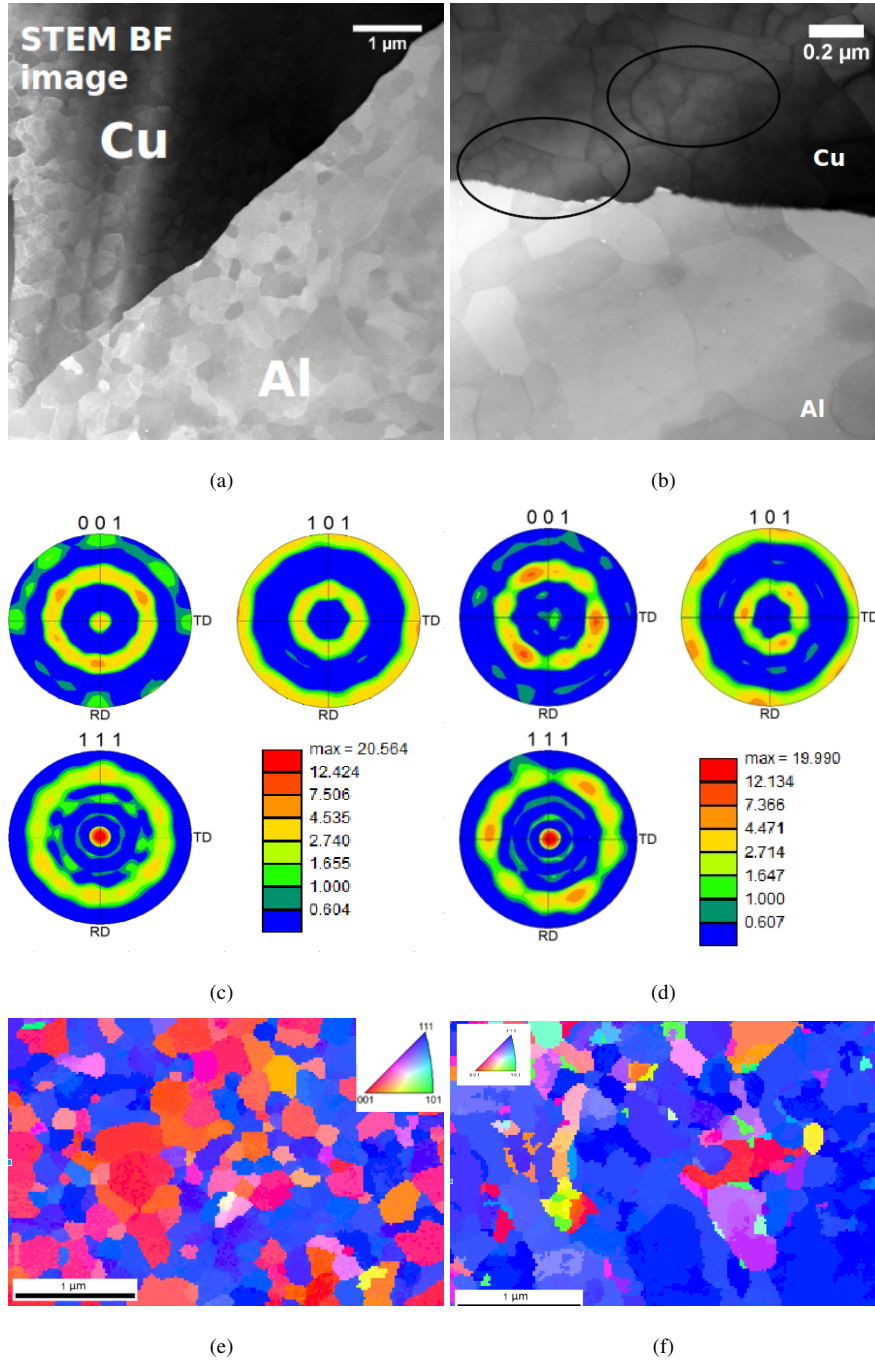


Figure 2: Microstructural analysis of a cross section of 3 mm-diameter architected Cu-Al wire composite in the as-drawn state: (a) and (b) STEM bright field images of Cu and Al, (c) Cu and (d) Al (001), (011) and (111) pole figures, (e) Cu and (f) Al ASTAR grain orientation maps.

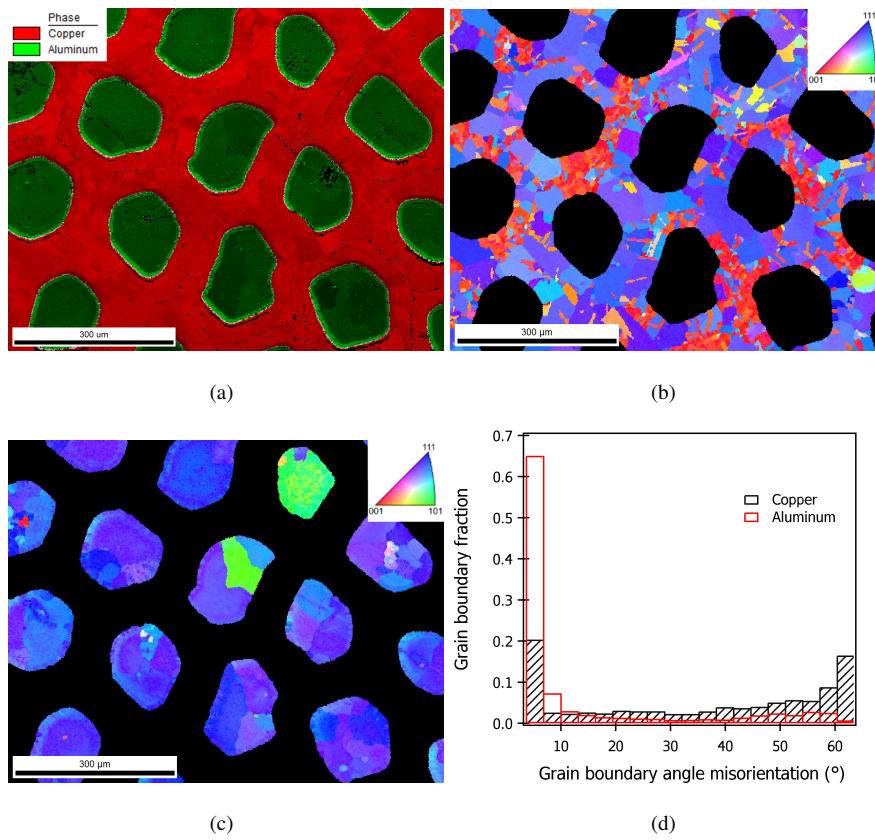


Figure 3: Microstructural analysis of a cross section of an architected composite Cu-Al (3 mm diameter) after heat treatment at 400 °C for 6h: (a) SEM-EDS element map, (b) and (c) Cu and Al EBSD grain orientation maps respectively, (d) Cu and Al grain boundary misorientation diagram.

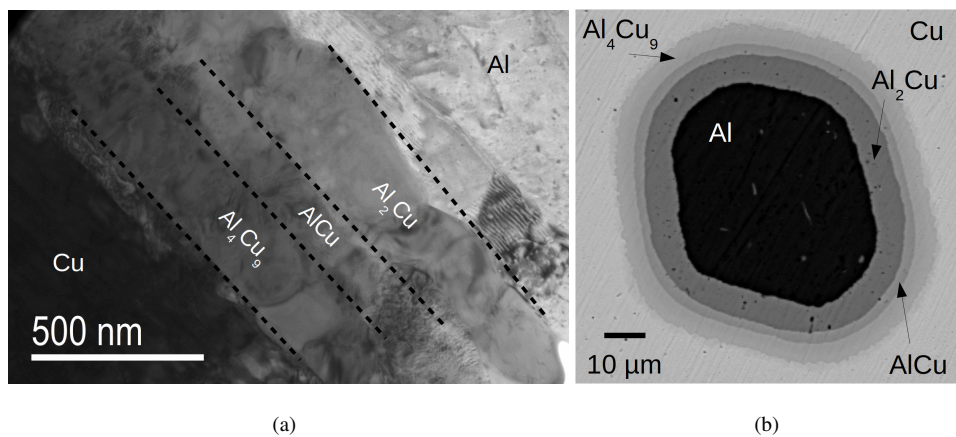


Figure 4: IMC observations for a 1.7 mm diameter ACCA wire : (a) bright field transmission electron microscope image for a short heat treatment at 300 °C during 15 min and (b) scanning electron microscope image, using backscattered electrons, for a heat treatment of 400 °C during 6h (HT2).

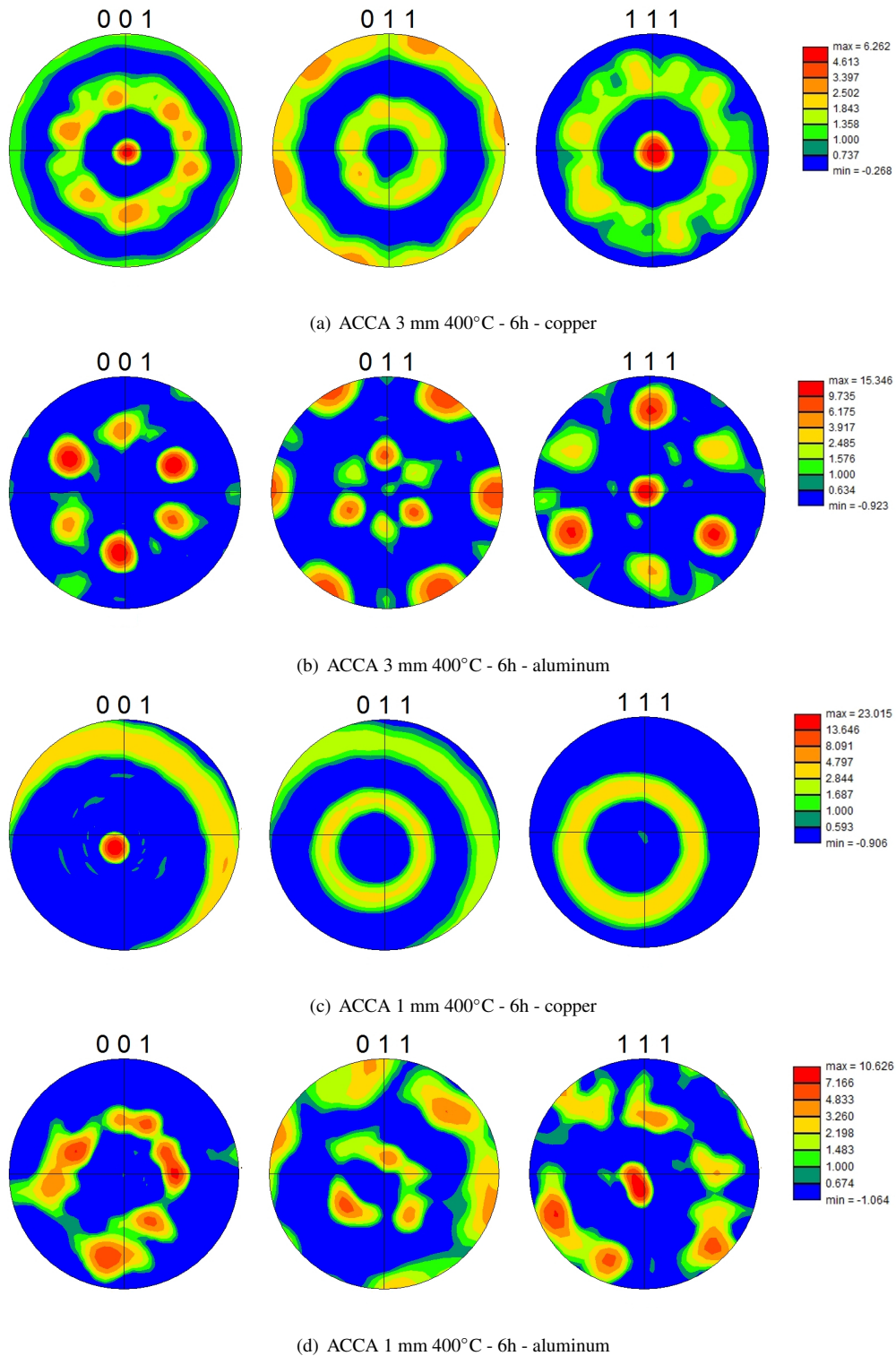


Figure 5: Pole figures associated to the cross section of the architected Cu-Al composite wires, heat treated at 400°C for 6h with the following diameters: (a) 3 mm diameter - Cu, (b) 3 mm diameter - Al, (c) 1 mm diameter - Cu and (d) 1 mm diameter - Al.

Table 1 summarizes the following characteristics of each ACCA sample: final diameter, average plastic strain during the process (estimated from eq.1), Grain Size (GS) of the Cu and Al phases, alongside their corresponding maximal density pole (indicating the texture intensity, I). The volume fractions (f_w) of Al, Cu, Al_4Cu_9 , AlCu and Al_2Cu are also estimated using the average aluminum area and the growth kinetics of the intermetallic compounds reported in previously published papers for the same materials [14, 16]. The same microstructural characteristics for the conventional CCA wires are also included in this table.

Table 1: Microstructural characteristics of all investigated architected and non-architected Cu-Al samples including grain size (GS expressed in μm), maximal density pole (I , expressed in Multiple Random Distribution - M.R.D.), and the volume fraction (f_w) of each phase. AD (As-Drawn), HT1 (400 °C - 30 min) and HT2 (400°C - 6h), represent the heat treatment conditions considered after processing.

Sample type	Plastic strain	State	Phases								
			Cu			Al			Al_4Cu_9	AlCu	Al_2Cu
			GS	I	f_w	GS	I	f_w	f_w	f_w	f_w
ACCA 1 mm	9.5	AD	0.4	23	0.74	0.4	21	0.26		No IMCs	
		HT1	4	18	0.67	6	15	0.23	0.032	0.015	0.053
		HT2	7	23	0.57	10	10	0.16	0.115	0.040	0.115
ACCA 1.7 mm	8.56	AD	0.5	25	0.74	0.4	20	0.26		No IMCs	
		HT1	7	11	0.72	10	14	0.22	0.017	0.01	0.030
		HT2	11	9.5	0.68	60	9	0.17	0.06	0.02	0.07
ACCA 3 mm	7.37	AD	0.4	20	0.74	0.3	18	0.26		No IMCs	
		HT1	5	26	0.73	4.5	5.8	0.24	0.01	0.005	0.017
		HT2	40	6.3	0.69	80	14	0.22	0.03	0.015	0.04
CCA 1 mm	4.97	AD	5	16	0.49	1	16	0.51		No IMCs	
		HT1	7	18	0.485	5	15	0.498	0.005	0.005	0.01
		HT2	10	15	0.47	8	12	0.49	0.017	0.0065	0.02

3.2. Tensile tests

Figures 6(a), 6(b) and 6(c) show the engineering tensile curves for the AD, HT1 and HT2 ACCA wires respectively. Tensile curves of the conventional 1 mm-diameter CCA wires, with the same heat treatment conditions, are also included in the figures for comparison purposes.

For the as-drawn samples (fig. 6(a)), the yield stress for three different diameters is very similar during the tensile test. This result can be attributed to the dynamic recovery occurring in both phases during the process, leading to similar grain size and texture (see table 1). The ductility of both ACCA and CCA wires is generally low (average fracture strain about 0.01). The ACCA wires, however, exhibit larger yield stress compared to CCA ones, probably due to the larger volume fraction of copper in the architected samples and their lower grain size.

For the HT1 conditions (fig. 6(b)), the tensile behavior is strongly dependent on the wire diameter. The yield stress increases from 100 MPa to 180 MPa with a decrease in diameter from 3 to 1 mm. The yield stress of the 1 mm-diameter architected wires is 80 MPa higher than the value for a conventional 1 mm-diameter CCA wires. The recrystallization occurring during the heat treatment clearly enhances ductility with an increase in plastic strain at fracture, for instance, from 0.01 (as-drawn) to 0.2 (HT1) for the 3 mm diameter. Nevertheless, the plastic strain at failure strongly depends on the wire diameter. Similar trends are observed for the HT2 sample (fig. 6(c)). An analogy between the HT1 and HT2 yield stress values reveals that, for a given wire diameter, longer heat treatments lead to higher yield stresses. The smallest-diameter HT2 sample exhibits brittle behavior, probably due to the larger volume fraction of IMCs, as reported in table 1. An unexpected plateau of almost constant stress can be seen beyond the elastic domain on the stress-strain curve of the two sets of heat-treated samples (HT1 and HT2). A number of serrations are also observed along the plateau, as depicted in the figure 6(c).

Figure 7 illustrates the strain hardening rate as a function of strain, for the heat treated samples. It can be observed from this figure that the strain hardening rate evolution (θ) can be divided into four different stages. The first one, characterized by a strong decrease in θ , appears at strains lower than 0.01 and can be ascribed to the gradual generalization of plasticity within the material [25, 26]. This plasticity generalization is expected to occur gradually between the two phases due to their differing Young's modulus and yield stress values (typically around 400 MPa of as-drawn Cu [27, 28] and 200 MPa for as-drawn Al [29]). Moreover, due to the different grain orientations in each phase, a plasticity generalization is also expected from the well-oriented grains (with respect to the tensile axis) to the poor oriented ones. Then, the unexpected plateau of low hardening rate appears in the stage I'. The strain range at this stage depends on the wire diameter, duration of heat treatment and seemingly the IMC volume fraction.

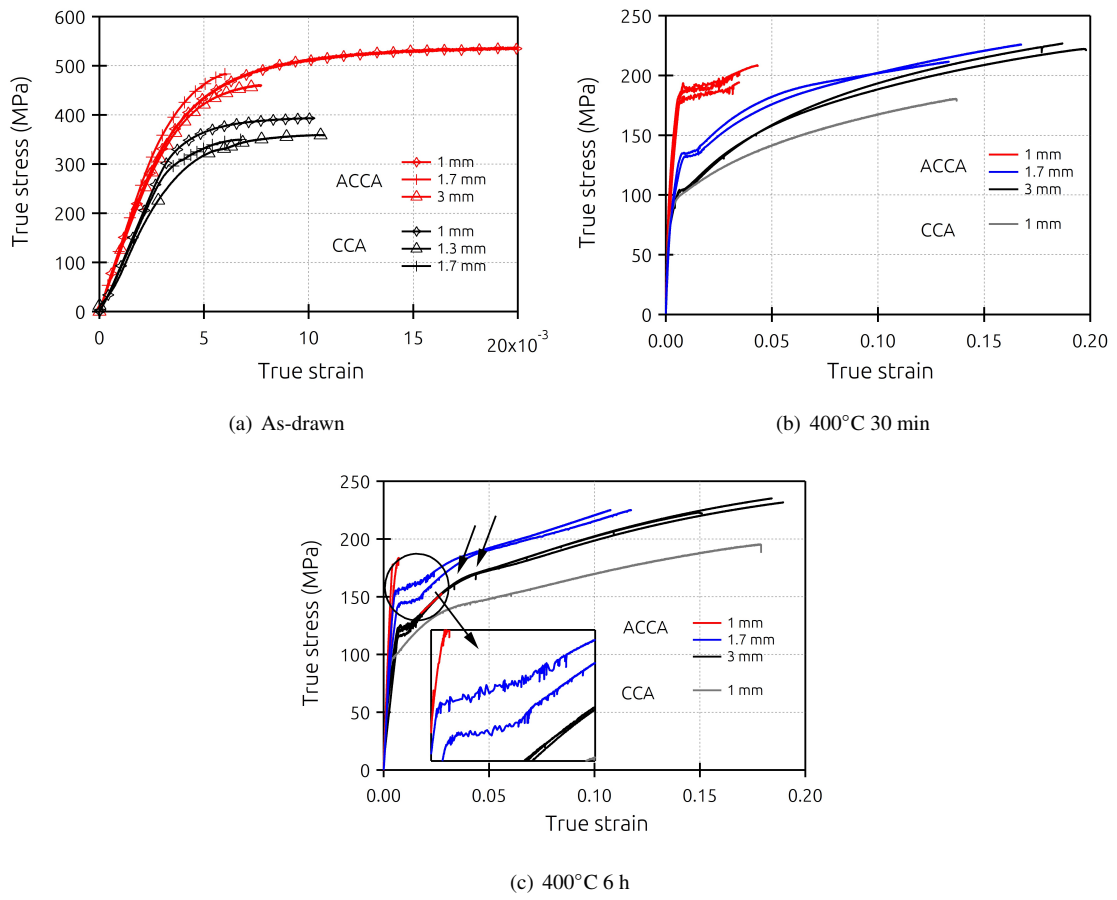


Figure 6: Engineering tensile curves of the architected copper-aluminum composites: (a) as-drawn (AD), (b) heat-treated at 400 °C for 30 min (HT1) and (c) heat-treated at 400 °C for 6 h (HT2).

The stage II begins as the plateau fades away. Here, the curve exhibits θ values around 2000 MPa, the same as the strain hardening rates reported for large-grain pure copper [30] and aluminum [31]. Following a short transition, the stage III, with an almost constant θ value, extends up to fracture. Furthermore, this stage begins at lower strain values for HT2 compared to HT1. All the above-mentioned stages, including the transition (labelled trans.) between stage II and stage III, are designated in figure 7(b) for the HT2 3 mm-diameter wire.

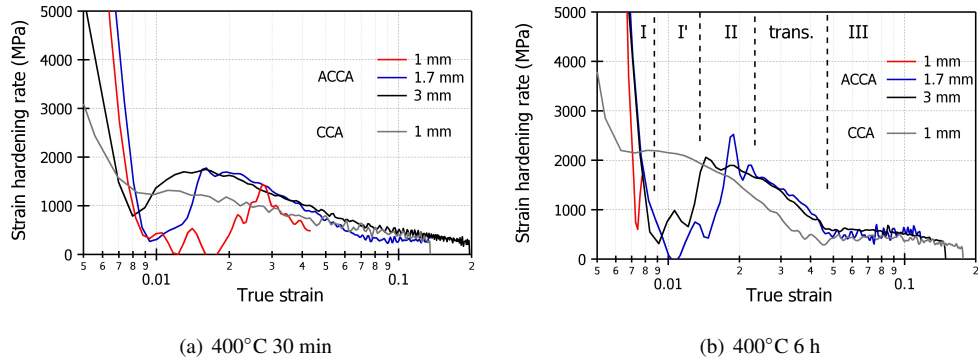


Figure 7: Evolution of the strain hardening rate versus strain for the following copper-aluminum composites: heat treated at 400°C (a) for 30 min and (b) for 6h.

3.3. Nanohardness

Nanoindentation tests were carried out in order to examine the effect of the sample characteristics (architecture, wire diameter and heat treatment) on the mechanical behavior of each phase. To this end, the cross-sections of 1 mm and 3 mm-diameter samples of CCA and ACCA in the as-drawn and HT2 states were analyzed. Modulus and hardness profiles across the Cu/Al and Cu/IMC/Al interfaces were acquired for all samples. Figure 8 represents the aforesaid type of profile for the HT2 3-mm diameter ACCA sample. Two intermetallic compounds named IMC1 and IMC2 with modulus values of about 90 and 130 GPa respectively, can be observed in this figure in agreement with the previous works on CCA [15]. It can be concluded from the literature [32, 33], that IMC1 and IMC2 are related to Al_2Cu and $AlCu$ in the order given. The hardness is almost the same for both IMCs with values around 9 GPa, being also consistent with the literature [21].

Using the above-mentioned profiles, average values of hardness and modulus for Cu and Al are calculated for each sample and a summary of the results is presented in table 2. The experimental values and their corresponding experimental uncertainty values are reported with a confidence level of 95%.

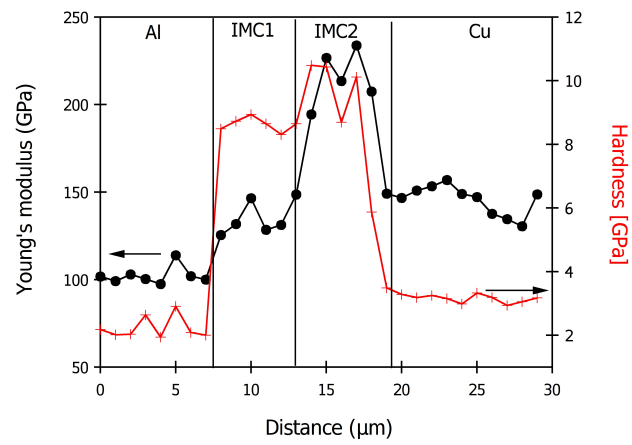


Figure 8: Modulus and hardness profile for a 3 mm diameter architected copper-aluminum composite obtained by cold-drawing via restacking method after the 400 °C 6h heat treatment.

Table 2: Modulus and hardness measurements of copper and aluminum phases in 1 mm and 3 mm-diameter samples of CCA and ACCA in as-drawn (AD) and HT2 (400 °C, 6h) states.

sample	plastic strain	state	Cu outer			Cu inner			Al	
			modulus (GPa)	Hardness (GPa)	modulus (GPa)	modulus (GPa)	Hardness (GPa)	modulus (GPa)	Hardness (GPa)	
CCA 3 mm	2.77	AD	144 ± 4	2.01 ± 0.07	-	-	82 ± 1	0.86 ± 0.04		
		HT2	120 ± 5	1.9 ± 0.1	-	-	75 ± 4	0.8 ± 0.05		
CCA 1 mm	4.97	AD	149 ± 2	2.21 ± 0.05	-	-	84 ± 1.4	1.07 ± 0.04		
		HT2	126 ± 10	1.8 ± 0.2	-	-	78 ± 10	0.9 ± 0.03		
ACCA 3 mm	7.37	AD	140 ± 4	2.28 ± 0.07	144 ± 3	2.53 ± 0.07	86 ± 1	1.23 ± 0.04		
		HT2	102 ± 3	2.39 ± 0.14	106 ± 3	2.26 ± 0.15	70 ± 2	1.10 ± 0.17		
ACCA 1 mm	9.57	AD	135 ± 4	2.29 ± 0.10	130 ± 5	2.55 ± 0.09	85 ± 3	1.24 ± 0.02		
		HT2	94 ± 3	2.18 ± 0.10	95 ± 3	2.30 ± 0.14	70 ± 1	1.40 ± 0.10		

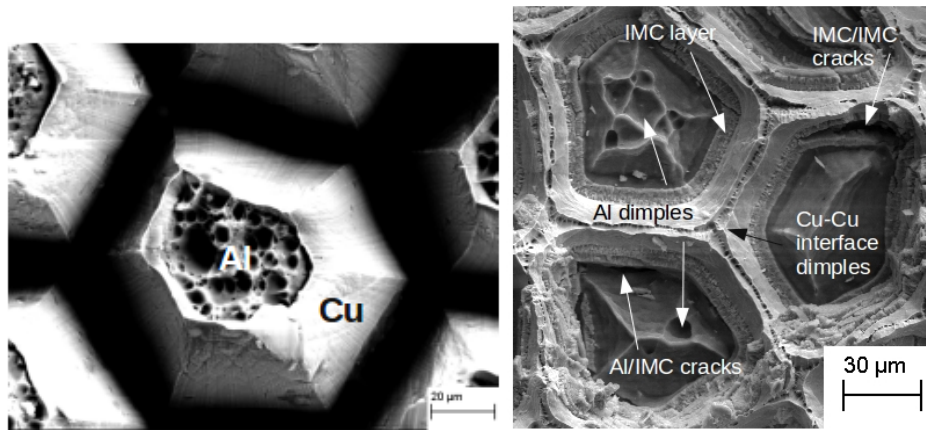
Taking into account the dispersion of the experimental modulus values in the as-drawn state, the sample characteristics show almost no impact on this parameter revealing no phase transformation or damage as a result of the process. Regarding the ACCA wires, the copper hardness is lower at the outer surface than in the inner part. This feature is probably due to the larger plastic strain sustained by the restacked CCA wires throughout the process. A comparison between the hardness values of Al and Cu in the architected and non-architected samples shows that the ACCA wires exhibit higher hardness values that could be attributed to the higher plastic deformation undergone during wire manufacturing.

In the HT2 heat-treated conditions, no statistically significant differences in hardness and module appear between the 1 and 3 mm diameter samples due to the sample recrystallization. A comparison between the aluminum and copper hardness values in the as-drawn and heat-treated states shows insignificant variations. However, for the two investigated sample diameters, the heat treatment leads to a decrease in modulus for Al and Cu phases, probably due to the texture evolution during the HT. This decrease is larger for copper than for aluminum. Regardless of the samples heat treatment condition and diameter, it is worth noticing that the average ratio of Al hardness to Cu for the ACCA samples is about 0.5 indicating the occurrence of a similar hardening mechanism in each phase.

3.4. Fracture surfaces

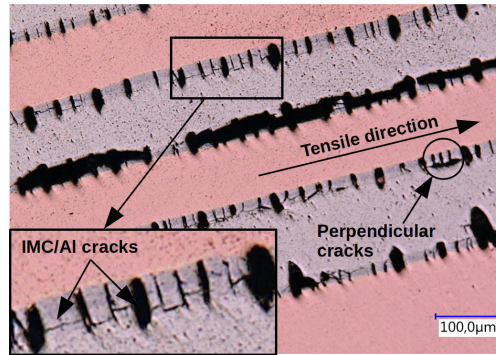
Figures 9(a) and 9(b) illustrate the fracture surfaces of two 1 mm-diameter ACCA specimens, the former in the as-drawn and the latter in HT1 state (the results are similar for the other diameters and the second heat treatment condition). In the as-drawn state, a transgranular ductile fracture of Al and Cu can be observed in figure 9(a). Contrary to Al, where large dimples can be seen, no dimples are observed in the copper phase. Moreover, this figure reveals that fracture at the Cu-Cu interface between the re-stacked wires occurs, most probably due to a weak mechanical bonding during the process [3]. Post-heat treatment Al fracture surface (figure 9(b)) shows no modification of the fracture mechanism. This figure is, furthermore, indicative of better bonding built at the Cu-Cu interface of the restacked wires. Few dimples appear across the interface, presumably because of the superficial contaminants resting across the copper wires during the process as reported in [16]. The IMC layer cracks perpendicularly to the tensile direction (perpendicular to the image plane) as expected and indicated in figure 9(b). This figure also shows several IMC/IMC and Al/IMC cracks, developed locally.

Those IMC and IMC/metal crack formations are confirmed by the longitudinal section observation, below the fracture surface, illustrated in figure 9(c). At fracture, all IMCs layers are regularly broken into parts of about 30 μm . IMC/Al cracks are frequently observed contrarily to IMC/Cu which are not observed revealing the higher strength of the Cu-IMC bonding compared to the IMC-Al one.



(a) as-drawn - 1 mm diameter

(b) 400°C 30 min - 1 mm diameter



(c) 400°C 6 h - 3 mm diameter

Figure 9: Microstructure observation at fracture for architected Cu-Al composites: (a) fracture surface for a 1 mm-diameter wire in as-drawn conditions, (b) fracture surface for a 1 mm-diameter wire heat treated at 400°C for 30 min and (c) longitudinal section for a 3 mm-diameter wire heat treated at 400°C for 6h.

4. Discussion

The results presented above show the influence of architecture and heat treatments on the strain and fracture mechanisms of the architected Cu-Al composites. These results are discussed in terms of the architecture, intermetallic compounds volume fraction and grain size in the following paragraphs.

4.1. Effect of the manufacture process on the microstructure

Taking into account the microstructural analysis reported in section 3.1, the grain size and texture of the as-drawn ACCA samples do not seem to be dependent on the wire diameter. Submicrometer grains are observed in Cu and Al for all as-drawn ACCA samples with the crystallographic directions of $\langle 111 \rangle$ and $\langle 001 \rangle$ as major and minor components of the texture, respectively. These results suggest the occurrence of dynamic recovery throughout the process, leading to an almost steady microstructural state. For the CCA specimens, the grain size is larger than ACCA

ones due to the lower plastic strain associated with the process. Additionally, the texture of the Al and Cu phases of the CCA and ACCA wires are similar to those of the drawn pure FCC metals, suggesting that the strain distribution pattern brought about by the process (which influence the texture formation), does not vary in the cold drawn Cu-Al composites.

The volume fraction of the IMCs in the heat-treated samples depends on the final diameter of the wire and the heat treatment conditions, varying between 0.03 (3 mm diameter - 400 °C 30 min) and 0.26 (1 mm diameter - 400 °C 6h). Al_2Cu has the largest volume fraction among the IMCs in all the samples. However, predominance of Al_2Cu is in contradiction with the observations reported by Kim and coworker [23] for Cu/Al/Cu trilayers obtained by Roll-Bonding (RB) which exhibit a larger volume fraction of Al_4Cu_9 .

The effect of processing on the heat treated samples, in terms of grain size and texture, depends on the material. As expected, an increment in the grain size of copper, for a given diameter following longer heat-treatments, is observed. The grain size is also affected by the wire diameter for a given heat treatment since recrystallization is dependent on the accumulated strain energy during the process. Figure 5(a) and figure 5(c) show that texture still exists in the copper phase of the heat treated samples after recrystallization. Moreover, the texture components and intensities depend on the heat treatment duration and wire diameter. Table 3 summarizes, for Cu and Al, the grain volume fraction with $\langle 111 \rangle$, $\langle 001 \rangle$ and $\langle 110 \rangle$ crystal orientation along the tensile axis, for the different samples characterized in this study.

For the shorter heat treatments (HT1), the $\langle 001 \rangle$ fiber component prevails over the $\langle 111 \rangle$, independently on the wire diameter. Park and Lee [18] suggested that the aforementioned increase in the volume fraction of the $\langle 001 \rangle$ fiber component indicates that only primary recrystallization, controlled by the stored strain energy, takes place in copper. In this case, the maximal density pole associated with the texture is still high.

For longer-duration heat treatments, the ACCA samples with 1.7 mm and 3 mm diameter exhibit a slightly higher volume fraction of $\langle 111 \rangle$ oriented grains compared to the $\langle 001 \rangle$. The occurrence of primary and secondary recrystallization (controlled by grain growth) in these two samples leads to a reduced texture intensity [18]. As observed in figure 3(b), during this secondary recrystallization, grain growth for copper is more pronounced in $\langle 111 \rangle$ oriented grains compared to the $\langle 001 \rangle$, in agreement with previous work published on Ag wires [34]. As a result the $\langle 111 \rangle$ component volume fraction increases according to table 3. For the 1 mm diameter sample in the HT2 conditions, no further texture evolution compared to shorter heat treatment is characterized, revealing that secondary recrystallization did not occur for this sample due to the high plastic strain involved during the process.

The post-heat-treatment aluminum grain orientations are similar to those already reported for the cold-drawn wires

[35, 36]. Both primary and secondary recrystallization take place within shorter heat-treatment periods in this material due to the lower homologous temperature of aluminum compared to copper. Subsequently, figure 3(c) shows the majority of grains orientated towards $\langle 111 \rangle$ along the wire axis. Park and Lee [18] concluded that the volume fraction of $\langle 111 \rangle$ increases by extending the heat treatment duration, prompting a rise in the maximal density pole value, as already reported for aluminum alloys processed by multi directional forging [37].

Table 3: Volume fraction of the $\langle 111 \rangle$, $\langle 001 \rangle$ and $\langle 110 \rangle$ texture components for the different samples characterized in this study. These values are computed using a 10° misorientation with respect to the crystal directions.

Sample type	Plastic strain	State	Texture component volume fraction					
			$\langle 111 \rangle$	Cu	Al	$\langle 001 \rangle$	$\langle 110 \rangle$	$\langle 110 \rangle$
ACCA 1 mm	9.5	AD	0.63	0.12	0.07	0.65	0.1	0.05
		HT1	0.3	0.5	0.05	0.4	0.3	0.05
		HT2	0.04	0.85	0.02	0.4	0.03	0.2
ACCA 1.7 mm	8.56	AD	0.70	0.09	0.05	0.68	0.12	0.03
		HT1	0.08	0.70	0.03	0.55	0.18	0.05
		HT2	0.45	0.30	0.05	0.6	0.03	0.14
ACCA 3 mm	7.39	AD	0.78	0.04	0.08	0.76	0.14	0.09
		HT1	0.03	0.9	0.04	0.4	0.04	0.122
		HT2	0.35	0.19	0.16	0.69	0	0.12
CCA 1 mm	4.97	AD	0.47	0.27	0.07	0.65	0.2	0.01
		HT1	0.19	0.69	0.04	0.5	0.4	0.03
		HT2	0.78	0.08	0.02	0.7	0.1	0.02

4.2. Effect of the architecture on the mechanical behavior of the as-drawn samples

Figure 6(a) demonstrates a difference between the yield stress values of the conventional and architected Cu-Al composites in the as-drawn state. Because of the significant dynamic recovery that takes place in both Cu and Al during processing, the wire diameter seems to have little impact on the yield stress of either architected or non-architected samples. These results are consistent with those reported for pure copper samples manufactured by wire drawing and accumulative roll bonding, for which the flow stress depends on the process plastic strain only trivially if this parameter is larger than 2 [20]. Assuming that a conventional rule of mixture can be employed for the CCA and ACCA samples for which there is no size effect involved (large characteristic dimensions for the copper and aluminum path), the average experimental yield stress for CCA and ACCA can be compared, as a first approximation, to the predictions of the Voigt (upper bound - linear relationship) and Reuss (lower bound - inverse relationship) boundaries of the rule of mixture. To this end, the mechanical behavior of copper and aluminum in tension were separately investigated with the same test configuration for the composite wires. The pure Cu and Al samples (same material as for the composite wires) were fabricated under the same conditions as the CCA and ACCA specimens to have the different degrees of plastic strain undergone by CCA and ACCA wires into consideration.

In order to achieve this aim, 1 mm-copper-copper wires were first obtained from cold drawing of a copper tube and a copper rod of the same dimensions of the Cu and Al counterparts used for manufacturing the CCA sample. For this pure 1 mm diameter Cu wire similar to CCA, the yield stress is about 380 MPa. 61 of these all-copper wires were then restacked into a copper tube and further cold-drawn to a diameter of 1 mm to ensure reaching the same strain level as the 1 mm-ACCA sample (i.e $\eta = 9.5$). The yield stress of this full-copper ACCA counterpart is about 410 MPa in agreement with the 380 MPa saturation stress reported in previous works for strains up to 10 [27, 28]. An aluminum rod and an aluminum tube of the same dimensions of those employed to fabricate the ACCA samples were first cold-drawn to 1 mm ($\eta = 2.5$). The yield stress of the full-aluminum CCA counterpart is about 197 MPa. Further drawing of these wires, when restacked into an annealed aluminum tube, was unsuccessful due to the large hardness contrast between them. Assuming the ratio of the yield stress of aluminum to that of copper to be similar for CCA and ACCA (as observed for the nanohardness values - see table 2), the yield stress of full pure aluminum-ACCA counterpart is expected to be about 210 MPa.

Using these values of yield stress for Cu and Al, and considering a pure plastic perfect behavior (associated with the dynamic recovery during the process), the upper and lower boundaries for the theoretical rule of mixtures are plotted for the yield stress in figure 10, as a function of the aluminum volume fraction for CCA and ACCA wires. The average experimental values for the CCA (51% of Al) and ACCA (26% of Al) are also indicated for comparison purposes.

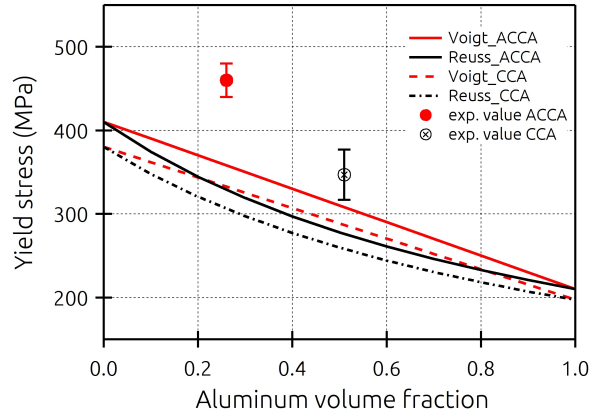


Figure 10: Comparison between the Voigt and Reuss prediction for the yield stress of Cu-Al composites.

The prediction of the rule of mixture is 50 MPa lower than the experimental yield stress value for the CCA sample, demonstrating some potentially weak extra hardening because of the two phases involved and their corresponding interfaces. The yield strength of the ACCA sample, however, shows a positive 100 MPa deviation from the prediction of the rule of mixtures and is additionally higher than the yield stress of the full-copper ACCA counterpart. For Cu/Al/Cu trilayers obtained by RB, an opposite trend was reported for the ultimate tensile strength which is slightly lower than the value predicted by the linear rule of mixtures [23]. A clear hardening effect of the composite architecture is then characterized.

In the as-drawn state, the dominant $\langle 111 \rangle$ crystal direction alongside the wire axis exhibits the highest Young's modulus difference between Cu and Al. Using the values reported in the literature [38], obtained from the compliance matrix for the $\langle 111 \rangle$ orientation, the elastic modulus of copper and aluminum are 191 and 76 GPa, in that order. The coupled effect of this difference in elastic properties and yield strength results in residual stresses induced during the process [39, 40, 41]. Following the analytical computations by Bullough and Hartley [41] of residual stress in a 50%-Al CCA wire strained up to 4 (equivalent to a 1 mm-diameter CCA wire in this study), axial compressive stresses of about 80 MPa are expected in aluminum, whereas there would be around 70 MPa of tensile stress in copper. Tensile residual stresses in copper, however, decrease for lower volume fractions of aluminum and increases at higher plastic strains. Low Al volume fraction and large processing strains lead to a higher level of compressive stresses in the aluminum component of the ACCA samples compared to the CCA samples. The stress level leading to plastic flow in the aluminum phase is then higher in the ACCA sample compared to the CCA sample, which in part explains the wider gap between the experimental yield stress and the Voigt prediction.

Furthermore, at the onset of plasticity in copper (which deforms plastically first), strong plastic strain incompatibilities will appear between this phase and aluminum (which still behaves elastically). This strain incompatibilities may enhance the formation of strong backstresses and geometrically necessary dislocations in copper phase near the interfaces between the two phases. As the interface density by volume unit is larger for the ACCA compared to CCA, this phenomenon may also explain the extra-hardening of the architected composite characterized in this study. An extensive investigation of the role played by the architecture on the process induced residual stresses and the influence of the interface must be performed to better understand the origin of this such elevated yield stresses.

4.3. Influence of the heat-treatments on the mechanical properties of the Cu-Al composites

Figures 6(b) and figure 6(c) clearly show the effect of the volume fraction of the intermetallic compounds on the tensile curves. To make comparisons between the IMCs effect on the mechanical behavior of the samples with different heat treatment conditions, the yield stress and Ultimate Tensile Strength (UTS) as a function of IMC volume fraction in HT1 and HT2 conditions are plotted in figure 11. For the same volume fraction of IMCs, the yield stress of the sample heat treated at 400°C for 30 minutes, is significantly larger than the one heat treated at 400 °C for six hours. As summarized in table 1, the grain size of the HT2 sample is larger than the HT1 one. In agreement with those results, the grain size strengthening is then lower for HT2 than for HT1. Moreover, the difference in Schmid factor between each texture component (0.401 for a theoretical $\langle 001 \rangle$ versus 0.27 for $\langle 111 \rangle$ one) can impact the yield stress, depending on the type of the recrystallization (primary or secondary) occurring during the heat-treatment. The yield stress of the ACCA wires is then a function of the heat treatment-dependent characteristics such as the volume fraction of IMCs, grain size and texture.

Regardless of the grain size, larger volume fractions of IMCs decrease the UTS as a result of early fracture. Such early fracture, happening in the samples with larger volume fractions of IMCs, may result from the large tensile stresses undergone by the brittle compounds but also from the weak metal/IMC interface in the samples with large IMC thicknesses, according to the literature [2, 3].

The volume fraction of IMCs also affects the strain hardening stages of the architected Cu-Al composites. Firstly, the occurrence of the stage I' with no or weak hardening is clearly related to the IMCs as no static strain ageing is expected for this material. This stage is connected with the serrated flow, observed on the stress-strain curve, similar to those observed during dynamic strain aging (see figure 6(c) inset). In some cases, these serrated parts are observed for larger plastic strains (see black arrows in figure 6(c)) and have been also reported by Kocich and coworkers for Cu/Al/Cu trilayers [12]. These authors incriminated the propagation of shear bands within the metallic composites. However, since the oscillations primarily appear at the onset of plasticity in the samples with different volume fractions of IMCs, the gradual fracture of the intermetallic compounds which cannot bear rather high tensile

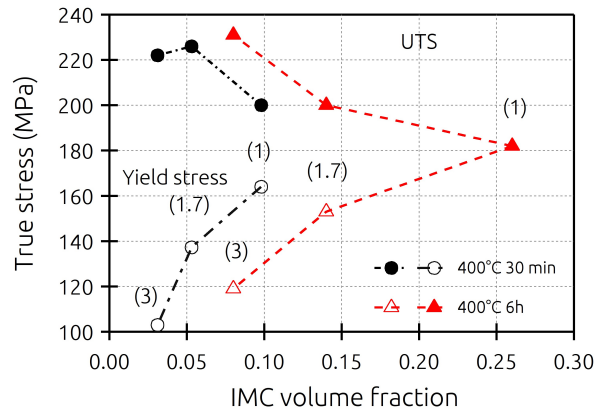


Figure 11: Evolution of the yield stress and Ultimate Tensile Strength (UTS) as a function of the volume fraction of the intermetallic compounds. Wire diameter is indicated between brackets for yield stress.

strains (see figure 9(c)), is believed to cause the stage I' to appear.

Second of all, the stage III, with a constant strain hardening rate of around 800 MPa, reveals that dislocation annihilation processes (which generally act in pure copper and aluminum) may be counterbalanced by a dislocation generation mechanism. The dimples observed on the fracture surface of the copper network in the HT1 sample are typical of its larger stress triaxiality compared to as-drawn samples [42]. This may result from a change in the stress state within copper and aluminum phases because of the radial deformation restriction imposed by the IMCs (only few interfacial cracks observed in figure 9). Moreover, according to the literature [43], stress concentration in Al and Cu phases can occur at large plastic strains, as a result of fracture in IMCs. The stress concentration points suggests that dislocation generation probably counterbalances the dislocation annihilation mechanism by cross-slip.

Conclusions

In this study, architected Cu-Al composite wires were manufactured via cold drawing applying a restacking method. The mechanical properties of the wires, including hardness and tensile behavior, were examined, under the following heat treatment conditions: as-drawn, 400 °C - 30 min and 400 °C - 6h. The conclusions made are listed below:

- The microstructure, in terms of grain size, texture, IMC type and its volume fraction strongly depends on the manufacturing conditions;
- in the as-drawn state, the improvement of the mechanical properties for ACCA, partly coming from the larger volume fraction of copper, highlights the important role played by the architecture, the difference between the elastic modulus of Cu and Al and the density of the Cu/Al interface.
- a comparison between the as-drawn and heat-treated samples shows that the two heat treatments lead to a reduction in the yield stress and an increase in the ductility, if the IMC volume fraction remains low, because of the recrystallisation of both Al and Cu phases. The recrystallization is also responsible for heat-treatment dependence of texture components;
- the volume fraction of the intermetallics significantly influences the yield stress, fracture strain and strain-hardening mechanisms, by inducing new stages on the strain-hardening curves, compared to those of the conventional pure copper and aluminum.

The architected composite wires, with low intermetallic volume fractions, exhibit slightly improved mechanical properties, compared to those of conventional copper-clad aluminum wires. Considering the good electrical conductivity of those architected wires, they could be considered as innovative alternatives to the conventional copper-clad aluminum wires, for carrying electric currents over an even wider range of frequencies. An ongoing study is already carried out to better understand how both the architecture and intermetallics influence the mechanical properties of Cu-Al composite wires.

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Data availability

The raw/processed data required to reproduce these findings cannot be shared at this time as the data also forms part of an ongoing study

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