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A thermo-elasto-viscoplastic constitutive model for polymers

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Abstract

Tensile tests conducted at different temperatures and strain rates on a low density cross-linked polyethylene (XLPE) have shown that increasing the strain rate raises the yield stress in a similar manner as when the temperature is decreased. The locking stretch also increases as a function of the strain rate, but not to the same extent as by decreasing the temperature. The volumetric straining and self-heating of the specimens were also measured in the experimental campaign: at room temperature the material was close to incompressible, while at the lower temperatures it was found to be moderately compressible. At the lowest strain rate isothermal conditions was observed, while adiabatic heating was seen at the highest strain rate.

In this study, a thermo-elasto-viscoplastic model is developed for XLPE in an attempt to describe the combined effects of temperature and strain rate on the mechanical stress-strain response but also on the thermodynamical response. The proposed model consists of two parts. On one side, Part A models the thermoelastic and thermoviscoplastic response, and incorporates an elastic Hencky spring in series with two Ree-Eyring dashpots. The two Ree-Eyring dashpots represent the effects of the main α relaxation and the secondary β relaxation processes on the plastic flow. Part B, on the other side, consists of an eight chain spring capturing the entropic strain hardening due to alignment of the polymer chains during deformation.

The constitutive model was implemented in a nonlinear finite element (FE) code using a semi-implicit stress update algorithm combined with sub-stepping and a numerical scheme to calculate the consistent tangent operator. After calibration to available experimental data, FE simulations with the constitutive model are shown to successfully describe the stress-strain curves, the volumetric strain, the local strain rate and the self-heating observed in the tensile tests. In addition, the FE simulations adequately predict the global response of the tensile tests, such as the force-displacement curves and the deformed shape of the tensile specimen.

Keywords: Temperature, Constitutive model, Polyethylene, XLPE, Strain rate sensitivity, Self-heating

1. Introduction

The use of polymers in structural applications has increased during the last decades. Some examples are shock absorbers in cars designed for pedestrian protection, thermal insulation of pipelines in the offshore oil industry and electrical insulation of high-voltage cables. The mechanical behaviour of polymers is complex and factors such as strain rate, temperature and stress triaxiality have a great impact on the structural behaviour of polymer components. Thus, it is a challenging task to obtain accurate numerical predictions of the mechanical response of polymeric materials under different loading scenarios. Prototype testing has therefore become a normal way to qualify materials and structural components for given applications in the industry. Qualifying materials in this manner is both costly and time consuming; thus there is a need for sufficiently accurate and easy-to-use material models. By using reliable material models, a limited set of experiments can be conducted for calibration purposes, and subsequently, numerical analyses of the structural component can be used either to optimize geometry or to investigate the effect of using different materials.

There is a number of available material models for polymers. Haward and Thackray [1] were the first to decouple the stress into one part where the elastic response was modelled by Hookean elasticity and a single Eyring dashpot [2] was employed to represent the inelastic flow, and a second part concerning entropic strain hardening using a Langevin spring derived from non-Gaussian chain statistics [3]. This model was extended to a three-dimensional (3D) formulation by Boyce et al. [4], who also incorporated strain softening and pressure sensitivity. Further development of the entropic strain hardening was done by Arruda et al. [5], resulting in the well-known eight chain model used in the current study. Regarding the flow process, Ree and Eyring [6] extended the original model by Eyring [2] to include several relaxation times, which in our work are restricted to two, namely the main α relaxation and the secondary β relaxation [7, 8].

An important aspect regarding the Ree-Eyring flow process is that it does not include strain hardening. A common way of including strain hardening has been to introduce a backstress, see e.g. [1, 4, 9, 10]. A problem that may arise from this approach is that self-heating, due to the viscous flow, can be underestimated. This leads to difficulties when trying to describe thermal softening in polymers at elevated strain rates [11–13]. Another way of including strain hardening was proposed by Hoy and Robbins [14]. Using a multiplicative rate sensitivity formulation where the hardening modulus was scaled by the flow stress, they

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29 obtained good results for the strain rates and temperatures covered in their study. However, investigating
30 different polymers at strain rates yielding isothermal conditions, Govaert et al. [15] showed that the mod-
31 elling approach of Hoy and Robbins [14] did not work in general. Instead they suggested to introduce a
32 backstress in addition to viscous strain hardening, where the viscous strain hardening may either be mod-
33 elled by stress-scaling of the hardening modulus [14], or by introducing a non-constant strain dependent
34 activation volume in the Eyring model as proposed by Wendlandt et al. [16]. The latter approach is thor-
35 oughly evaluated by Senden et al. [17]. Their work shows the problematic behaviour in cyclic loading if
36 the entire strain hardening is incorporated in the strain dependent activation volume (or strain dependent
37 reference strain rate), namely that instead of continuing strain hardening when going from tension to com-
38 pression, the model will predict strain softening since the activation volume will start to decrease when the
39 loading direction is reversed. To avoid this unphysical behaviour, a portion of the strain hardening has to be
40 modelled by an inelastic backstress.

41 The viscous behaviour contributes to self-heating in a material. In the studies performed by Adams
42 and Farris [18] and Boyce et al. [19], it was found that about 50 – 80% of the total mechanical work was
43 converted into heat in glassy polymers. On the other hand, studying high density polyethylene (HDPE),
44 Hillmansen et al. [20, 21] observed that almost the entire mechanical work was converted into heat. A
45 similar observation was also done by Johnsen et al. [11] on a crosslinked low density polyethylene (XLPE).
46 Since heating of the polymer material will introduce thermal softening, it is evident that a correct prediction
47 of heat generation during deformation is crucial in order for the constitutive model to capture the material
48 behaviour over a range of strain rates. Consequently, taking into account thermomechanical coupling is
49 important in this situation, and in particular accounting for heat conduction within the material and heat
50 convection to the surroundings. There are many examples of thermomechanically coupled constitutive
51 models. Arruda et al. [13] and Boyce et al. [19] combined an elastic Hookean response with non-Newtonian
52 viscous flow and kinematic hardening based on the alignment of the polymer chains. Adopting a similar
53 approach, Richeton et al. [22] presented a model able to span the glass transition temperature. More recent
54 developments were made by Garcia-Gonzalez et al. [23] who extended the isothermal model proposed by
55 Polanco-Loria et al. [24] to include thermomechanical coupling. This model combines an elastic Neo-
56 Hookean response with rate-dependent yielding and plastic flow governed by the Raghava yield function
57 [25] and kinematic hardening modelled by an eight chain spring. Another extension of the Polanco-Loria
58 et al. [24] model was done by Ognedal et al. [26], who added isotropic hardening of the Raghava yield

59 surface. Anand et al. [27] and Ames et al. [28] presented a thermomechanically coupled constitutive
60 model describing the large deformation behaviour of amorphous polymers, including loading/unloading
61 and torsion. In another study, Maurel-Pantel et al. [29] proposed a visco-hyperelastic constitutive model to
62 capture large deformations and self-heating in a semi-crystalline polyamide 66. In the study by Srivastava
63 et al. [30], the model presented by Anand et al. [27] was extended to span the glass transition temperature.
64 The material model's ability to span the glass transition temperature is of course desirable, but it inevitably
65 introduces additional parameters and adds complexity to the calibration procedure. Thus, we have chosen
66 to limit our study to temperatures above the glass transition, namely the leathery region [8] between the
67 glass transition and melting temperatures.

68 The thermomechanical behaviour of a cross-linked low density polyethylene (XLPE) material was stud-
69 ied experimentally in Johnsen et al. [11] using the experimental set-up described in Johnsen et al. [31].
70 Similar studies concerned with the effect of low temperatures on the mechanical behaviour have been per-
71 formed, see e.g. Richeton et al. [32], Brown et al. [33], Serban et al. [34] and Bauwens-Crowet [35]. All
72 of these studies revealed the same trends as observed by Johnsen et al. [11], namely that lowering the tem-
73 perature increases the yield stress in a similar manner as an increase in strain rate, indicating that the yield
74 stress may be determined from thermal activation theory [6, 36]. However, in these studies the strains were
75 obtained by mechanical measurement techniques, as opposed to the local measurements made possible by
76 digital image correlation (DIC) in Johnsen et al. [11]. Additionally, self-heating due to elevated strain rates
77 was not reported.

78 In this study, based on the experimental investigation outlined above and described in the next section,
79 we present a thermo-elasto-viscoplastic model to describe the mechanical behaviour of XLPE at different
80 temperatures and strain rates. The proposed model has two parts: Part A consists of an elastic Hencky
81 spring in series with two Ree-Eyring dashpots. The two Ree-Eyring dashpots model the effects of the main
82 α relaxation and the secondary β relaxation processes on the plastic flow. Part B consists of an entropic
83 eight chain spring modelling strain hardening due to alignment of the polymer chains during deformation.
84 The constitutive model is implemented in the commercial finite element (FE) program Abaqus/Standard as
85 a UMAT subroutine. A semi-implicit stress update algorithm is combined with a sub-stepping procedure to
86 ensure convergence. The consistent tangent operator is found by numerical differentiation as proposed by
87 Miehe [37] and Sun et al. [38].

88 This paper is organized as follows: first, we briefly describe the material investigated here followed

89 by a summary of the experimental set-up [31] along with the main experimental results obtained in [11].
 90 Then the constitutive model is presented within a general thermodynamical framework including the heat
 91 equation used to calculate the temperature increase. This is followed by a brief outline of the numerical
 92 integration procedure and the calibration procedure . Finally, the results obtained from simulations are
 93 compared to the experimental findings allowing some concluding remarks to be drawn.

94 2. Material, experimental set-up, methods and experimental results

95 In this study, we consider the material behaviour of a cross-linked low density polyethylene (XLPE)
 96 material. The material is produced by Borealis under the product name Borlink LS4201S [39] and was
 97 received from Nexans Norway as extruded high-voltage cable segments where the copper conductor had
 98 been removed. The dimensions of the cable segments were 128 mm \times 73 mm \times 22.5 mm (length \times
 diameter \times thickness). Material properties of the XLPE material is given in Table 1.

Table 1: Material properties for the XLPE material. All parameters are given for room temperature [11, 31].

Density, ρ (kg/m ³)	Specific heat capacity, C_p (J/(kg·K))	Thermal conductivity, k (W/(m·K))	Heat transfer coefficient to air, h_c (W/(m ² ·K))
922	3546	0.56	21

99
 100 The experimental set-up consisted of a purpose-built transparent polycarbonate temperature chamber,
 101 where a thermocouple temperature sensor mounted close to the test specimen maintained the desired tem-
 102 perature by controlling the flow of liquid nitrogen into the chamber. In contrast to conventional temperature
 103 chambers with non-transparent walls, the polycarbonate chamber made it possible to monitor the test spec-
 104 imen using two digital cameras, enabling the calculation of the strains on two perpendicular surfaces using
 105 digital image correlation (DIC) – a necessity due to the slight transverse anisotropy of the XLPE mate-
 106 rial. It was also feasible to measure self-heating of the specimen with a thermal camera. A sketch of the
 107 experimental set-up is given in Figure 1.

108 Uniaxial tension and compression tests were performed at four temperatures ($T = -30$ °C, $T = -15$
 109 °C, $T = 0$ °C and $T = 25$ °C) and three different cross-head velocities: $v = 0.04$ mm/s, $v = 0.4$ mm/s and
 110 $v = 4.0$ mm/s. Assuming that all deformation happens over the parallel section of the tensile specimen,
 111 these cross-head velocities correspond to initial nominal strain rates $\dot{\epsilon}$ of 0.01 s⁻¹, 0.1 s⁻¹ and 1.0 s⁻¹. All

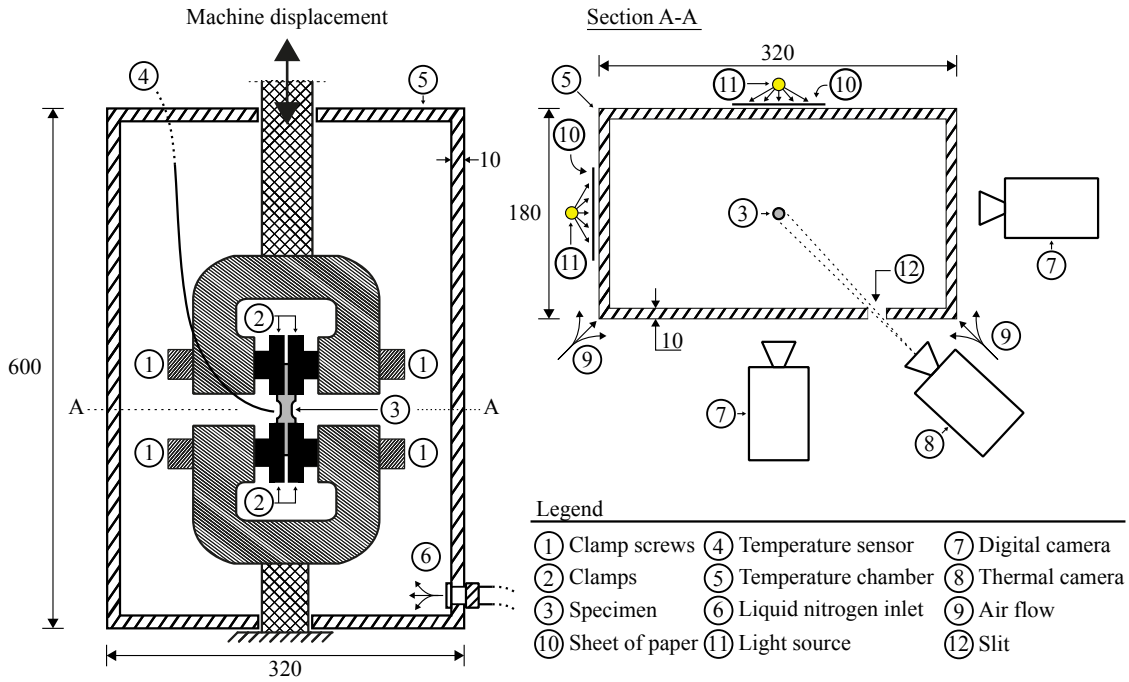


Figure 1: Illustration of the experimental set-up. All measures are in mm. For a detailed description see Johnsen et al. [11, 31].

112 tests were performed in an Instron 5944 testing machine equipped with 2 kN load cell. Figure 2 shows the
 113 cylindrical specimens used in these experiments.

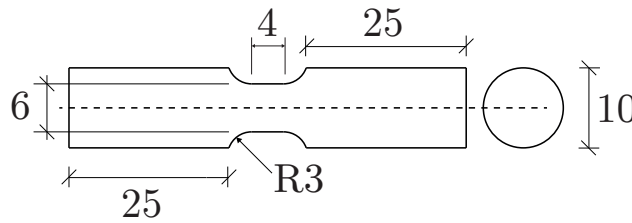


Figure 2: Illustration of the tensile test specimen. All measures are in mm.

114 The transparency of the temperature chamber allowed us to monitor two perpendicular faces of the
 115 specimens during deformation using two digital cameras, an important feature due to the slight transverse
 116 anisotropy of the material [11]. Subsequent digital image correlation (DIC) analyses of the images were
 117 performed to obtain the longitudinal and transverse strains from the section of initial necking on both sur-
 118 faces. Knowing the transverse strains in two perpendicular directions, the current cross-sectional area was
 119 calculated assuming an elliptical cross-section, enabling the calculation of the Cauchy stress as

$$\sigma = \frac{F}{A} = \frac{F}{\pi r_1 r_2} = \frac{F}{\pi r_0^2 \lambda_1 \lambda_2} \quad (1)$$

120 where r_1 and r_2 are the radii recorded by each digital camera, $\lambda_i = r_i/r_0$ (for $i = 1, 2$) are the corresponding
121 transverse stretches with r_0 equal to the initial radius of 3 mm, and F is the global force measured by the
122 testing machine. Furthermore, the volumetric strain ε_V is found by summation of the three principal strain
123 components, i.e.,

$$\varepsilon_V = \varepsilon_L + \varepsilon_1 + \varepsilon_2 \quad (2)$$

124 where ε_L is the longitudinal logarithmic strain and $\varepsilon_i = \ln(\lambda_i)$ are the transverse logarithmic strains.

125 In addition to the two digital cameras used to obtain the strains, an infrared thermal camera was em-
126 ployed to measure the self-heating of the material during the tensile experiments. A slit was added in the
127 front window of the temperature chamber to obtain a free line-of-sight between the camera and the tensile
128 specimen. The thermal camera operated down to a temperature of -20 °C. To ensure that the correct tem-
129 perature was maintained during the experiments, a thermocouple temperature sensor was used to control
130 the flow of liquid nitrogen into the temperature chamber. All specimens were thermally conditioned for
131 a minimum of 30 minutes inside the temperature chamber prior to testing. To avoid icing on the outside
132 of the chamber, and consequently obstruction of the digital camera imaging, fans were used to blow air
133 continuously over the chamber walls.

134 A condensed illustration of the local stress-strain behaviour reported in [11] is given in Figure 3. It
135 appears that temperature-time equivalence applies for the XLPE material, namely that a decrease in temper-
136 ature has a similar impact on Young's modulus and the flow stress as an increase in strain rate. Using two
137 Ree-Eyring [6] dashpots, Johnsen et al. [11] successfully described the flow stress as a function of both tem-
138 perature and strain rate, while they used a phenomenological expression similar to that proposed by Arruda
139 et al. [13] to describe the temperature dependence of Young's modulus. It is also noted from Figure 3 that
140 the locking stretch, defined as the stretch where an abrupt change in strain hardening occurs, increases with
141 increasing strain rate, and decreases slightly with decreasing temperature. This phenomenon is believed
142 to be caused by increased chain mobility due to self-heating at elevated strain rates, and decreased chain
143 mobility at lower temperatures, respectively. The material was also found to be close to incompressible at
144 room temperature, while it is compressible at the three lower temperatures. In terms of self-heating, it was
145 shown in [11] that the lowest strain rate ($\dot{\epsilon} = 0.01$ s $^{-1}$) gave close to isothermal conditions. At the interme-
146 diate strain rate ($\dot{\epsilon} = 0.1$ s $^{-1}$) self-heating was observed, but due to the duration of the test, heat conduction
147 inside the material and heat convection to the surroundings caused the temperature to decrease at the end of
148 the experiment. For the tests performed at the highest strain rate ($\dot{\epsilon} = 1.0$ s $^{-1}$), close to adiabatic conditions

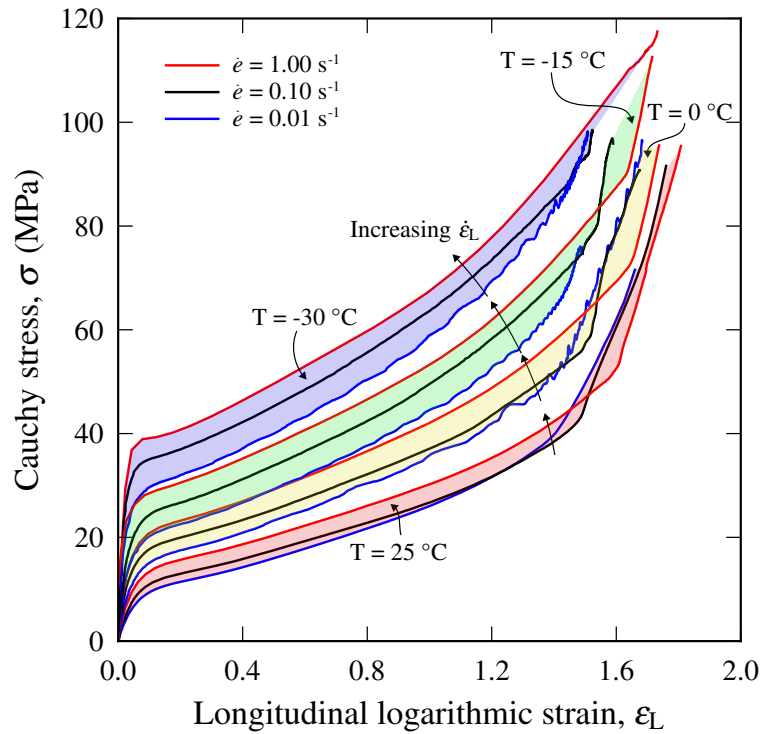


Figure 3: Condensed version of all stress-strain curves from experiments showing how the material behaviour is affected by changing the temperature and the strain rate. Adapted from Johnsen et al. [11].

149 were met, resulting in a temperature increase in the material between 20 °C and 35 °C. Further, uniaxial
 150 compression tests revealed that the yield stress is similar in tension and compression. The test results from
 151 [11] will be shown in full together with predictions from the numerical simulations in Section 6.

152 For a more detailed presentation and discussion of the experimental set-up, the methods used to extract
 153 local stress-strain data and self-heating from experiments, and the experimental results, see Johnsen et al.
 154 [11, 31].

155 3. Constitutive model

156 In this section we present the thermo-elasto-viscoplastic model proposed to describe the thermomechanical
 157 behaviour observed in the experiments on the XLPE material. In addition to the features addressed in
 158 Figure 3, the model also aims at capturing the volumetric response and self-heating. The model has been
 159 implemented in the implicit framework provided by Abaqus/Standard as a user subroutine (UMAT).

160 *3.1. Overview*

161 As seen from the kinematics in Figure 4a, we use a multiplicative split of the deformation gradient
 162 \mathbf{F} to separate between elastic and plastic deformation [40]. Applying the plastic deformation gradient \mathbf{F}^p
 163 maps an undeformed material element from the reference configuration (Ω_0) to the elastically unloaded
 164 intermediate configuration ($\tilde{\Omega}$). Finally, compatibility is obtained by mapping the material element from $\tilde{\Omega}$
 165 to the current configuration (Ω) via the elastic deformation gradient \mathbf{F}^e , viz.

$$\mathbf{F} = \mathbf{F}^e \mathbf{F}^p \quad (3)$$

166 Our material model, see Figure 4b, has two contributions: Part A (intermolecular) describes the hyperelastic
 167 and viscoplastic behaviour, while Part B represents the orientational hardening due to the alignment of the
 168 polymer network. From Figure 4b it follows that the deformation gradient is equal in each part, viz.

$$\mathbf{F} = \mathbf{F}_A = \mathbf{F}_A^e \mathbf{F}_A^p = \mathbf{F}_B \quad (4)$$

where subscripts A and B denote Parts A and B of the rheological model, respectively.

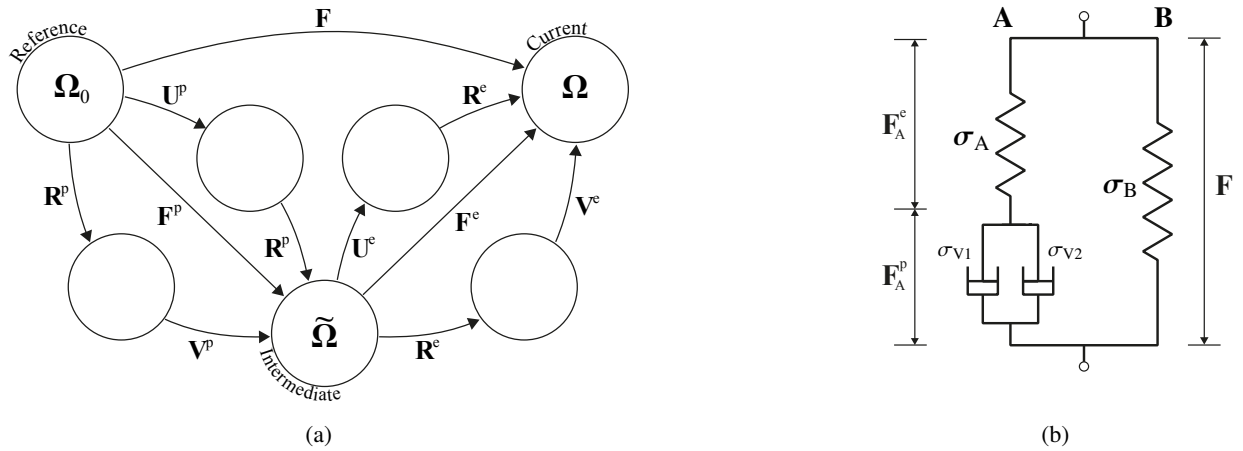


Figure 4: Large deformations kinematics using a multiplicative split of the deformation gradient, \mathbf{F} , is shown in (a), and (b) shows the rheological model.

169

170 Polar decomposition of the elastic and plastic parts of the deformation gradient of Part A yields

$$\mathbf{F}_A^e = \mathbf{V}_A^e \mathbf{R}_A^e = \mathbf{R}_A^e \mathbf{U}_A^e \quad (5)$$

$$\mathbf{F}_A^p = \mathbf{V}_A^p \mathbf{R}_A^p = \mathbf{R}_A^p \mathbf{U}_A^p \quad (6)$$

171 where \mathbf{R} is the rotation tensor, \mathbf{U} and \mathbf{V} are the right and left stretch tensors, respectively, and superscripts
 172 e and p denote the elastic and plastic parts. The isochoric deformation gradient tensor $\bar{\mathbf{F}}$ is defined by

$$\bar{\mathbf{F}} = J^{-1/3} \mathbf{F} \quad (7)$$

173 where $J = \det(\mathbf{F})$ is the Jacobian determinant, thus implying that $\det(\bar{\mathbf{F}}) = 1$. The isochoric left Cauchy-
 174 Green deformation tensor $\bar{\mathbf{B}}$ and the isochoric left stretch tensor $\bar{\mathbf{V}}$ are defined as

$$\bar{\mathbf{B}} = \bar{\mathbf{F}} \bar{\mathbf{F}}^T = J^{-2/3} \mathbf{F} \mathbf{F}^T = J^{-2/3} \mathbf{B} \quad (8)$$

$$\bar{\mathbf{V}} = \sqrt{\bar{\mathbf{B}}} = J^{-1/3} \sqrt{\mathbf{B}} = J^{-1/3} \mathbf{V} \quad (9)$$

175 where $\mathbf{B} = \mathbf{F} \mathbf{F}^T$ is the left Cauchy-Green deformation tensor. Throughout this study the plastic deformation
 176 is assumed to be isochoric, i.e., $J_A^p = 1$ and thus $J_A^e = J$ since the decomposition of the Jacobian determinant
 177 reads $J = \det(\mathbf{F}) = \det(\mathbf{F}_A^e) \det(\mathbf{F}_A^p) = J_A^e J_A^p$. With respect to the elastic and plastic parts of the deformation
 178 gradient tensor, we then obtain the following relations:

$$\bar{\mathbf{F}}_A^e = J^{-1/3} \mathbf{F}_A^e, \quad \bar{\mathbf{B}}_A^e = \bar{\mathbf{F}}_A^e (\bar{\mathbf{F}}_A^e)^T = J^{-2/3} \mathbf{B}_A^e, \quad \bar{\mathbf{V}}_A^e = J^{-1/3} \mathbf{V}_A^e \quad (10)$$

$$\bar{\mathbf{F}}_A^p = \mathbf{F}_A^p, \quad \bar{\mathbf{B}}_A^p = \mathbf{F}_A^p (\mathbf{F}_A^p)^T = \mathbf{B}_A^p, \quad \bar{\mathbf{V}}_A^p = \mathbf{V}_A^p \quad (11)$$

179 According to the rheological model in Figure 4b, the free energy is decomposed as follows

$$\psi = \psi_A + \psi_B \quad (12)$$

180 where ψ_A and ψ_B are the free energies of Parts A and B, respectively. Note that the free energy function is
 181 here defined per unit reference mass. In the same manner, the Cauchy stress tensor is decomposed as

$$\boldsymbol{\sigma} = \boldsymbol{\sigma}_A + \boldsymbol{\sigma}_B \quad (13)$$

182 where $\boldsymbol{\sigma}_A$ and $\boldsymbol{\sigma}_B$ are the Cauchy stress tensors acting in Parts A and B of the rheological model.

183 3.1.1. Part A - Intermolecular

184 Both the elastic and plastic responses of Part A are taken to be isochoric. The elastic response is defined
 185 by the Hencky free energy [41], i.e.,

$$\rho_0 \psi_A = \mu_A(\theta) \text{tr} \left[\left(\ln(\bar{\mathbf{V}}_A^e) \right)^2 \right] \quad (14)$$

186 where ρ_0 is the initial density of the material and θ is the absolute temperature. The shear modulus of the
 187 elastic spring is temperature dependent through the following expression

$$\mu_A(\theta) = \mu_{A,\text{ref}} \exp[-a_A (\theta - \theta_{\text{ref}})] \quad (15)$$

188 where θ_{ref} is a reference temperature, $\mu_{A,\text{ref}}$ is the shear modulus at the reference temperature, and a_A is a
 189 parameter governing the temperature sensitivity.

190 The Kirchhoff stress tensor $\boldsymbol{\tau}_A$ is obtained from the free energy function in Equation (14) as [42]

$$\boldsymbol{\tau}_A = 2\rho_0 \frac{\partial \psi_A}{\partial \mathbf{B}_A^e} \mathbf{B}_A^e \quad (16)$$

191 which after some algebra leads to [41]

$$\boldsymbol{\tau}_A = 2\mu_A(\theta) \ln(\bar{\mathbf{V}}_A^e) \quad (17)$$

192 The Cauchy stress tensor $\boldsymbol{\sigma}_A$ is then given as

$$\boldsymbol{\sigma}_A = \frac{1}{J} \boldsymbol{\tau}_A \quad (18)$$

193 Now we focus on the thermoviscoplastic part of the constitutive model. Since the yield stress in tension and
 194 compression was found to be approximately the same [11], the pressure-insensitive von Mises equivalent
 195 stress is used

$$\sigma_D^{\text{vm}} = \sqrt{\frac{3}{2} \boldsymbol{\sigma}'_D : \boldsymbol{\sigma}'_D} \quad (19)$$

196 where $\boldsymbol{\sigma}'_D = \boldsymbol{\sigma}_D - \frac{1}{3} \text{tr}(\boldsymbol{\sigma}_D) \mathbf{1}$ is the deviatoric part of the driving stress $\boldsymbol{\sigma}_D = \boldsymbol{\sigma}_A$. From the rheological
 197 model (Figure 4b) it is evident that the equivalent driving stress must be balanced by the viscous stress
 198 associated with the Ree-Eyring [6] dashpots. Thus, assuming that the contribution from each dashpot is
 199 additive [7], we obtain

$$\sigma_V = \sigma_{V1} + \sigma_{V2} = \sum_{x=\alpha,\beta} \frac{k_B \theta}{V_x} \text{arsinh} \left(\frac{\dot{p}}{\dot{p}_{0,x}^*} \exp \left[\frac{\Delta H_x}{R\theta} \right] \right) = \sigma_D^{\text{vm}} \quad (20)$$

200 where α and β denote the contributions from the main and secondary relaxation processes, respectively,
 201 k_B is Boltzmann's constant, V_x is the activation volume, \dot{p} is the equivalent plastic strain rate, ΔH_x is the
 202 activation enthalpy, and R is the universal gas constant. Further, $\dot{p}_{0,x}^*$ is the deformation dependent reference
 203 equivalent plastic strain rates given by

$$\dot{p}_{0,x}^* = \dot{p}_{0,x} \exp \left(-\sqrt{\frac{2}{3}} b_x \|\ln(\mathbf{V}^p)\|_2 \right) \quad \text{for } x = \alpha, \beta \quad (21)$$

204 where $\dot{p}_{0,x}$ are the initial values of $\dot{p}_{0,x}^*$, b_α and b_β are the parameters governing the deformation dependence,
 205 and $\|\ln(\mathbf{V}^p)\|_2$ is the Frobenius norm of the Hencky strain tensor.

206 The velocity gradient \mathbf{L}_A and its decompositions are given by

$$\mathbf{L}_A = \dot{\mathbf{F}}_A \mathbf{F}_A^{-1} = [\dot{\mathbf{F}}_A^e \mathbf{F}_A^p + \mathbf{F}_A^e \dot{\mathbf{F}}_A^p] (\mathbf{F}_A^p)^{-1} (\mathbf{F}_A^e)^{-1} \quad (22)$$

$$\mathbf{L}_A = \dot{\mathbf{F}}_A^e (\mathbf{F}_A^e)^{-1} + \mathbf{F}_A^e \dot{\mathbf{F}}_A^p \mathbf{F}_A^{-1} = \mathbf{L}_A^e + \mathbf{L}_A^p \quad (23)$$

$$\mathbf{L}_A = \mathbf{D}_A^e + \mathbf{W}_A^e + \mathbf{D}_A^p + \mathbf{W}_A^p \quad (24)$$

207 where \mathbf{D} and \mathbf{W} are in turn the rate-of-deformation tensor and the spin tensor. Due to isotropy, the plastic
 208 velocity gradient is taken to be irrotational [19, 43], i.e., the plastic spin tensor is equal to zero, $\mathbf{W}_A^p = \mathbf{0}$.

209 The plastic rate-of-deformation tensor is given by the flow rule as

$$\mathbf{D}_A^p = \mathbf{L}_A^p = \lambda \frac{\partial g(\sigma_D)}{\partial \sigma_D} \quad (25)$$

210 where λ is a plastic multiplier and $g(\sigma_D)$ is the plastic potential. Assuming that the plastic flow is isochoric,
 211 the plastic potential is taken as

$$g(\sigma_D) = \sqrt{\frac{3}{2} \sigma_D' : \sigma_D'} = \sigma_D^{\text{vm}} \geq 0 \quad (26)$$

212 where the direction of plastic flow \mathbf{N} is obtained from the gradient of the plastic potential,

$$\mathbf{N} = \frac{\partial g(\sigma_D)}{\partial \sigma_D} = \frac{3}{2} \frac{\sigma_D'}{g(\sigma_D)} \quad (27)$$

213 Equivalence in terms of plastic power yields the relation between the equivalent plastic strain rate, \dot{p} , and
 214 the plastic multiplier, λ , viz.

$$\sigma_D^{\text{vm}} \dot{p} = \sigma_D : \mathbf{D}_A^p \Rightarrow \dot{p} = \lambda \quad (28)$$

215 Combining Equations (23) and (25) and inserting $\lambda = \dot{p}$, we obtain the expression for the evolution of the
 216 plastic deformation gradient, i.e.,

$$\dot{\mathbf{F}}_A^p = \dot{p} (\mathbf{F}_A^e)^{-1} \frac{\partial g(\sigma_D)}{\partial \sigma_D} \mathbf{F}_A \quad (29)$$

217 3.1.2. Part B - Orientational hardening

218 The orientational hardening of the material due to the alignment of the polymer chains is captured by
 219 the eight chain model [5]. Following Miehe [44] we define a modified entropic free energy function, viz.

$$\rho_0 \psi_B = \frac{\kappa(\theta)}{2} (\ln(J))^2 - 3\kappa(\theta)\alpha \ln(J)(\theta - \theta_0) + T(\theta) + \mu_B(\theta) \lambda_{\text{lock}}^2 \left[\left(\frac{\bar{\lambda}_c}{\lambda_{\text{lock}}} \right) \beta + \ln \left(\frac{\beta}{\sinh \beta} \right) \right] \quad (30)$$

220 The shear modulus of Part B is interpreted as a rubbery modulus, i.e.,

$$\mu_B(\theta) = nk_B\theta = \mu_{B,\text{ref}} \frac{\theta}{\theta_{\text{ref}}} \quad (31)$$

221 where n is the chain density, k_B is Boltzmann's constant, and $\mu_{B,\text{ref}}$ is the shear modulus at the reference
 222 temperature. In this study the reference temperature is set equal to 298.15 K, while the initial temperature
 223 is equal to the temperatures at which the experiments were conducted. The temperature dependent bulk
 224 modulus $\kappa(\theta)$ is found by assuming that Poisson's ratio ν is constant, viz.

$$\kappa(\theta) = \frac{2\mu_B(\theta)(1 + \nu)}{3(1 - 2\nu)} \quad (32)$$

225 The linear thermal expansion coefficient α is assumed to be independent of temperature. Further, λ_{lock} is
 226 the locking stretch, $\bar{\lambda}_c = \sqrt{\text{tr}(\bar{\mathbf{B}})}/3$ is an average chain stretch, and

$$\beta = \mathcal{L}^{-1} \left(\frac{\bar{\lambda}_c}{\lambda_{\text{lock}}} \right) \quad (33)$$

227 where \mathcal{L}^{-1} is the inverse Langevin function ($\mathcal{L}(x) = 1/x - \coth x$) approximated by the formula proposed
 228 by Jedynek [45]:

$$\mathcal{L}^{-1}(x) \approx x \frac{3 - 2.6x + 0.7x^2}{(1 - x)(1 + 0.1x)} \quad (34)$$

229 The purely thermal contribution to the free energy, which, assuming that the specific heat capacity, C_p , is
 230 constant, is given as [44, 46]

$$T(\theta) = C_p \left[(\theta - \theta_0) - \theta \ln \left(\frac{\theta}{\theta_0} \right) \right] \quad (35)$$

231 where θ_0 is the initial absolute temperature.

232 The Kirchhoff stress tensor, $\boldsymbol{\tau}_B$, is found after some algebra as [46]

$$\boldsymbol{\tau}_B = 2\rho_0 \frac{\partial \psi_B}{\partial \mathbf{B}} \mathbf{B} = \frac{\mu_B(\theta)\lambda_{\text{lock}}}{3\bar{\lambda}_c} \mathcal{L}^{-1} \left(\frac{\bar{\lambda}_c}{\lambda_{\text{lock}}} \right) \bar{\mathbf{B}}' + \kappa(\theta) \ln(J) \mathbf{1} - 3\kappa(\theta)\alpha(\theta - \theta_0) \mathbf{1} \quad (36)$$

233 with $\bar{\mathbf{B}}' = \bar{\mathbf{B}} - \frac{1}{3} \text{tr}(\bar{\mathbf{B}}) \mathbf{1}$ being the deviatoric part of $\bar{\mathbf{B}}$, and the Cauchy stress reads as

$$\boldsymbol{\sigma}_B = \frac{1}{J} \boldsymbol{\tau}_B \quad (37)$$

234 3.1.3. Self-heating and dissipation

235 The internal energy u , defined per unit reference mass, is given in terms of the free energy ψ and the
 236 entropy $s \equiv -\partial\psi/\partial\theta$ as

$$u = \psi + \theta s \quad (38)$$

237 Local energy balance is expressed as

$$\rho_0 \dot{u} = \boldsymbol{\tau} : \mathbf{D} + r - \text{div}(\mathbf{q}) \quad (39)$$

238 where r is external heat sources and \mathbf{q} is the heat flux. The deformation power per unit reference volume is
239 decomposed according to

$$\boldsymbol{\tau} : \mathbf{D} = \boldsymbol{\tau}_A : (\mathbf{D}_A^e + \mathbf{D}_A^p) + \boldsymbol{\tau}_B : \mathbf{D} = \boldsymbol{\tau}_A : \mathbf{D}_A^e + \boldsymbol{\tau}_D : \mathbf{D}_A^p + \boldsymbol{\tau}_B : \mathbf{D} \quad (40)$$

240 where $\boldsymbol{\tau}_D = J\boldsymbol{\sigma}_D$, and only the deformation power in the two dashpots contributes to the intrinsic dissipation.
241 After some calculations, the rates of change of the free energy and the entropy are obtained as [44]

$$\rho_0 \dot{\psi} = \boldsymbol{\tau}_A : \mathbf{D}_A^e + \boldsymbol{\tau}_B : \mathbf{D} - \rho_0 \dot{\theta} s \quad (41)$$

$$\rho_0 \theta \dot{s} = -\theta \frac{\partial \boldsymbol{\tau}_A}{\partial \theta} : \mathbf{D}_A^e - \theta \frac{\partial \boldsymbol{\tau}_B}{\partial \theta} : \mathbf{D} + \rho_0 C_p \dot{\theta} \quad (42)$$

242 where the specific heat capacity is defined as $C_p = \theta \frac{\partial s}{\partial \theta}$ and

$$\frac{\partial \boldsymbol{\tau}_A}{\partial \theta} = -2a_A \mu_A(\theta) \ln(\bar{\mathbf{V}}_A^c) = -a_A \boldsymbol{\tau}_A \quad (43)$$

$$\theta \frac{\partial \boldsymbol{\tau}_B}{\partial \theta} = \boldsymbol{\tau}_B - 3\kappa(\theta) \alpha \theta \mathbf{1} \quad (44)$$

243 The dissipation inequality may be stated as [42]

$$\mathcal{D} \equiv -\rho_0 (\dot{\psi} + s\dot{\theta}) + \boldsymbol{\tau} : \mathbf{D} - \frac{\mathbf{q}}{\theta} \cdot \frac{\partial \theta}{\partial \mathbf{x}} \geq 0 \quad (45)$$

244 where \mathbf{x} is the position vector in the current configuration. Inserting Equations (40) and (41) yields

$$\mathcal{D} = \boldsymbol{\tau}_D : \mathbf{D}_A^p - \frac{\mathbf{q}}{\theta} \cdot \frac{\partial \theta}{\partial \mathbf{x}} \geq 0 \quad (46)$$

245 The first term represents the intrinsic dissipation and is non-negative by the flow rule. The last term is the
246 dissipation due to heat conduction and is made non-negative by adopting Fourier's law: $\mathbf{q} = -k \frac{\partial \theta}{\partial \mathbf{x}}$, where
247 the conductivity k is positive.

248 The heat equation is obtained by combining Equations (38) to (44), and the result comes out as

$$\rho_0 C_p \dot{\theta} = \boldsymbol{\tau}_D : \mathbf{D}_A^p + \boldsymbol{\tau}_B : \mathbf{D} - \theta \left[a_A \boldsymbol{\tau}_A : \mathbf{D}_A^e + 3\kappa(\theta) \alpha \text{tr}(\mathbf{D}) \right] + r - \text{div}(\mathbf{q}) \quad (47)$$

249 By solving for the temperature rate, the heat equation is used to calculate the self-heating of the material at
250 elevated strain rates.

251 3.2. Numerical integration

252 The governing equations of Part A of the constitutive model are compiled in Box 1.

Box 1: Governing equations of Part A.

$\sigma_A = \frac{2}{J} \mu_A(\theta) \ln(\bar{\mathbf{V}}_A^e)$	elastic response
$\sigma_D = \sigma_A$	driving stress
$g(\sigma_D) = \sqrt{\frac{3}{2} \sigma'_D : \sigma'_D} = \sigma_D^{\text{vm}} \geq 0$	plastic potential
$\mathbf{D}_A^p = \dot{p} \frac{3\sigma'_D}{2\sigma_D^{\text{vm}}} = \mathbf{F}_A^e \dot{\mathbf{F}}_A^p \mathbf{F}_A^{-1}$	plastic rate-of-deformation
$\sigma_V = \sum_{x=\alpha,\beta} \frac{k_B \theta}{V_x} \operatorname{arsinh} \left(\frac{\dot{p}}{\dot{p}_{0,x}^*} \exp \left[\frac{\Delta H_x}{R\theta} \right] \right)$	viscous stress

253 A semi-implicit stress-update algorithm is used to integrate these equations in time, which implies that
 254 the direction of plastic flow \mathbf{N} and the absolute temperature θ lag one time step behind. Using the relation
 255 for the plastic rate-of-deformation tensor in Box 1, the inverse plastic deformation gradient is estimated by
 256 the relation

$$\left(\mathbf{F}_{A,n+1}^{p,i} \right)^{-1} = \left(\mathbf{1} - \Delta p_{n+1}^i \mathbf{F}_{n+1}^{-1} \mathbf{N}_n \mathbf{F}_{n+1} \right) \left(\mathbf{F}_{A,n}^p \right)^{-1} \quad (48)$$

257 where i denotes the current iteration in time step $n + 1$, $\Delta p_{n+1}^i = \dot{p}_{n+1}^i \Delta t_{n+1}$ is the equivalent plastic strain
 258 increment, and \mathbf{N}_n is the direction of plastic flow calculated from the previous time step, i.e.,

$$\mathbf{N}_n = \frac{3}{2} \frac{\sigma'_{D,n}}{\sigma_{D,n}^{\text{vm}}} \quad (49)$$

259 The elastic deformation gradient is then calculated as

$$\mathbf{F}_{A,n+1}^{e,i} = \mathbf{F}_{n+1} \left(\mathbf{F}_{A,n+1}^{p,i} \right)^{-1} \quad (50)$$

260 which gives us the driving stress, $\sigma_{D,n+1}^i$ and the von Mises equivalent stress $\sigma_{D,n+1}^{\text{vm},i}$, see Box 1. The
 261 constitutive relations for the two dashpots give a residual function in the form

$$f(\dot{p}_{n+1}^i) = f_{n+1}^i = \sigma_{D,n+1}^{\text{vm},i} - \sigma_{V,n+1}^i = 0 \quad (51)$$

262 where the viscous stress $\sigma_{V,n+1}^i$ is defined in Box 1. Using the secant method, an updated value of the
 263 equivalent plastic strain rate is obtained by

$$\dot{p}_{n+1}^{i+1} = \dot{p}_{n+1}^i - f_{n+1}^i \frac{\dot{p}_{n+1}^i - \dot{p}_{n+1}^{i-1}}{f_{n+1}^i - f_{n+1}^{i-1}} \quad (52)$$

264 The iteration procedure continues until a convergence criterion is fulfilled. Note that the iterative scheme
 265 is not self-started. In iteration $i = 1$ of the first increment the equivalent plastic strain rates \dot{p}_1^0 and \dot{p}_1^1 have
 266 to be estimated, while in the remaining increments \dot{p}_n^1 is set equal to the converged value from the previous
 267 increment \dot{p}_n and \dot{p}_n^0 is kept constant and equal to \dot{p}_1^0 .

268 Concerning Part B of the rheological model, the stress tensor $\boldsymbol{\sigma}_{B,n+1}$ is given explicitly by the deforma-
 269 tion gradient \mathbf{F}_{n+1} and the temperature from the previous timestep θ_n , i.e.,

$$\boldsymbol{\sigma}_{B,n+1} = \frac{\mu_B(\theta_n)\lambda_{\text{lock}}}{3\bar{\lambda}_{c,n+1}} \mathcal{L}^{-1}\left(\frac{\bar{\lambda}_{c,n+1}}{\lambda_{\text{lock}}}\right) \bar{\mathbf{B}}'_{n+1} + \kappa(\theta_n) \ln(J_{n+1})\mathbf{1} - 3\kappa(\theta_n)\alpha(\theta_n - \theta_0)\mathbf{1} \quad (53)$$

270 Following the work of Miehe [37] and Sun et al. [38], the consistent tangent operator, \mathbf{C}^t , is found
 271 by numerical differentiation. The deformation gradient is perturbed in such a way that only one of the six
 272 unique components of the rate-of-deformation tensor is changed at the time, i.e.,

$$\Delta\mathbf{F}_{\pm}^{(kl)} = \pm \frac{\epsilon}{2} [(\mathbf{e}_k \otimes \mathbf{e}_l)\mathbf{F} + (\mathbf{e}_l \otimes \mathbf{e}_k)\mathbf{F}] \quad (54)$$

273 where ϵ is the perturbation coefficient set equal to 10^{-8} and \mathbf{e}_k for $k = 1, 2, 3$ are the Cartesian base vectors.

274 The perturbed deformation gradient, $\hat{\mathbf{F}}^{(kl)}$, is then obtained as

$$\hat{\mathbf{F}}_{\pm}^{(kl)} = \mathbf{F} + \Delta\mathbf{F}_{\pm}^{(kl)} \quad (55)$$

275 For each of the twelve deformation gradients thus obtained, the Cauchy stress tensor $\boldsymbol{\sigma}(\hat{\mathbf{F}}^{(kl)})$ is calculated.

276 Using a central difference scheme, the consistent tangent operator \mathbf{C}^t is estimated as

$$\mathbf{C}_{ij(kl)}^t = \frac{\sigma_{ij}(\hat{\mathbf{F}}_+^{(kl)}) - \sigma_{ij}(\hat{\mathbf{F}}_-^{(kl)})}{2\epsilon} \quad (56)$$

277 In Voigt notation this means that for each plus-minus perturbation of the deformation gradient, we obtain
 278 column (kl) in the 6×6 tangent operator $[\mathbf{C}^t]$ with row indices $ij = 11, 22, 33, 12, 13, 23$.

279 To ensure convergence, sub-stepping is used to limit the strain increment during the time step. The
 280 number of sub-steps, N , is controlled by the criterion

$$N = \max \left\{ \text{nint} \left[\frac{\Delta\epsilon_{\text{eq}}}{\epsilon_{\text{cr}}} + 0.5 \right], 1 \right\} \quad (57)$$

281 where nint is the *nearest integer* function, $\Delta\epsilon_{\text{eq}} = \sqrt{\frac{2}{3}\Delta\boldsymbol{\epsilon}' : \Delta\boldsymbol{\epsilon}'}$ is the equivalent logarithmic strain incre-
 282 ment, $\Delta\boldsymbol{\epsilon}' = \Delta\boldsymbol{\epsilon} - \frac{1}{3} \text{tr}(\Delta\boldsymbol{\epsilon})\mathbf{1}$ is the deviatoric logarithmic strain tensor obtained by integrating the rate-of-
 283 deformation tensor \mathbf{D} over the time increment [47]

$$\Delta\boldsymbol{\epsilon} = \int_{t_n}^{t_{n+1}} \mathbf{D} dt \quad (58)$$

284 where n and $n + 1$ denote the previous and current time step, respectively. Furthermore, ε_{cr} is a critical value
 285 set equal to strain-to-yield. If $N > 1$, new deformation gradients are calculated from the velocity gradient
 286 at the beginning of the time step, i.e.,

$$\mathbf{L}_n = \frac{\mathbf{F}_{n+1} - \mathbf{F}_n}{\Delta t_{n+1}} (\mathbf{F}_n)^{-1} \quad (59)$$

287 For sub-step number q , the deformation gradient, \mathbf{F}_q is then calculated as

$$\mathbf{F}_q = \left(\mathbf{1} + \frac{q\Delta t_{n+1}}{N} \mathbf{L}_n \right) \mathbf{F}_n \quad \text{for } q \in [1, N] \quad (60)$$

288 4. Material model calibration

289 Direct calibration from the experimental data was performed to obtain initial values of the parameters
 290 in the constitutive model. These initial values were then used in an optimization procedure. A brief review
 291 of the direct calibration procedure is given in the following.

292 4.1. Shear modulus

293 Young's modulus E was found by linear regression of the Cauchy stress vs. longitudinal logarithmic
 294 strain curve for strain magnitudes of $\varepsilon_L \in [0, 0.025]$. The shear modulus μ could then be computed from
 295 the relation

$$\mu = \frac{E}{2(1 + \nu)} \quad (61)$$

296 where $\nu = \left| \frac{d\varepsilon_T}{d\varepsilon_L} \right| \approx 0.49$ is Poisson's ratio found by numerical differentiation of the transverse strain (ε_T) vs.
 297 longitudinal strain (ε_L) curves given in Johnsen et al. [11].

298 As shown in Figure 5, a clear strain rate and temperature dependence of the shear modulus was ob-
 299 served. This strain rate dependence of the shear modulus has been neglected in Equation (15). The material
 300 parameters in Equation (15) were found to be equal to $\mu_{A,ref} = 46$ MPa and $a_A = 0.03$ K⁻¹ from a least
 301 squares fit to the experimentally obtained shear moduli, see Figure 5.

302 4.2. Flow stress

303 The coefficients in the Ree-Eyring flow model [6] were identified from the stress-strain curves by using
 304 the flow stress, $\sigma_{0.15}$, at a fixed strain magnitude of $\varepsilon_L = 0.15$ for all investigated temperatures and strain
 305 rates. The least squares fit of Equation (20) to the experimental data is shown in Figure 6 along with the
 306 obtained parameters in Table 2.

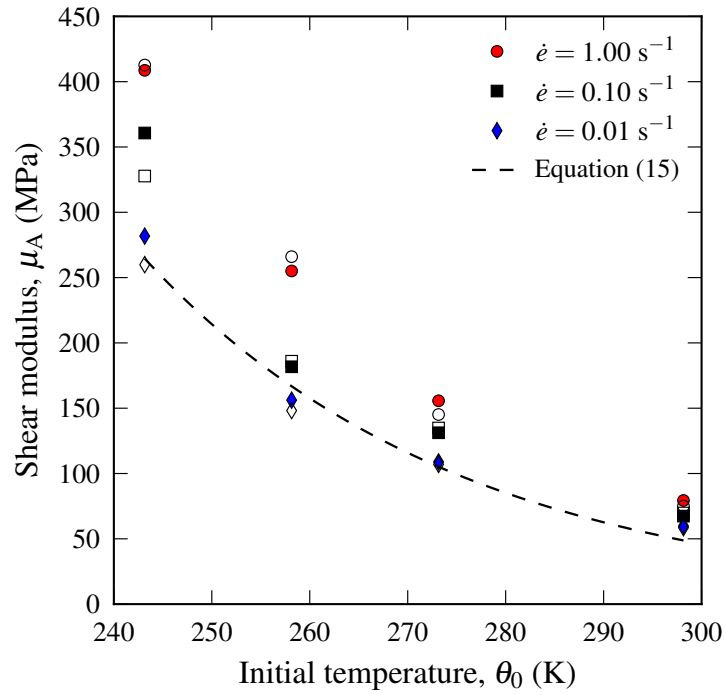


Figure 5: Temperature and strain rate dependence of the shear modulus of the material. Data adapted from [11].

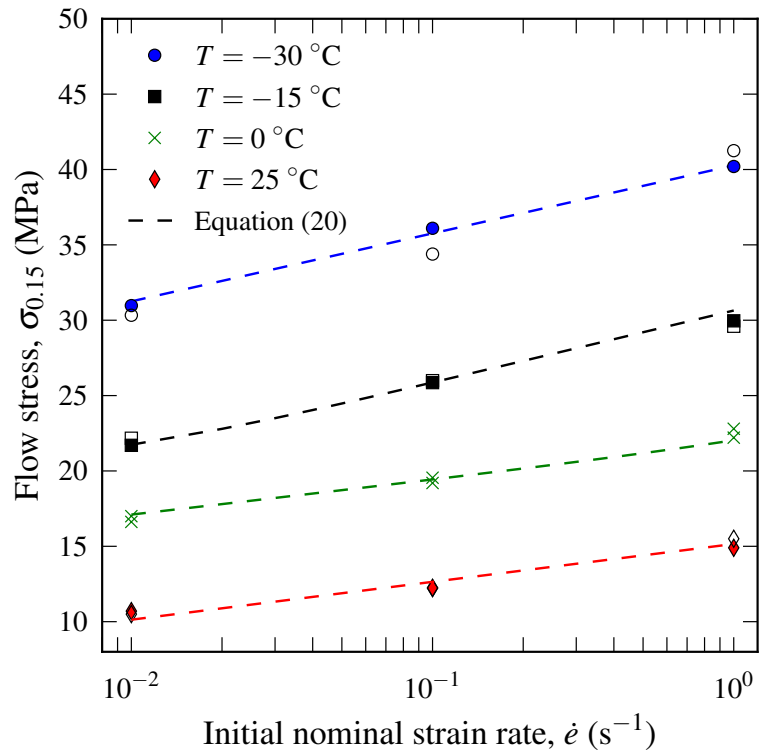


Figure 6: Temperature and strain rate dependence on the flow stress of the material. Data taken from [11].

Table 2: Initial material parameters (before optimization) in the Ree-Eyring model, Equation (20).

k_B (J/K)	R (J/(mol·K))	V_α (nm ³)	$\dot{\rho}_{0,\alpha}$ (s ⁻¹)	ΔH_α (kJ/mol)	V_β (nm ³)	$\dot{\rho}_{0,\beta}$ (s ⁻¹)	ΔH_β (kJ/mol)
$1.38 \cdot 10^{-23}$	8.314	3.45	$1.38 \cdot 10^{28}$	188.6	3.10	$5.79 \cdot 10^{39}$	204.3

307 4.3. Strain hardening

308 There are two contributions to strain hardening in the model: (1) orientational hardening σ_B in Part B
 309 capturing the effect of polymer chain alignment, and (2) isotropic hardening from the deformation dependent
 310 reference strain rates in the viscous dashpots in Part A.

311 The orientational hardening is modelled by the eight chain spring [5]. Simply put, the spring accounts
 312 for how the polymer chains align due to stretching and give rise to the abrupt change in strain hardening
 313 when approaching the locking stretch. To estimate the value of the reference shear modulus $\mu_{B,\text{ref}}$ and
 314 the locking stretch, λ_{lock} , a simple one-dimensional (1D) model was used. First we calculate the axial
 315 component of the stress from Equation (37) as

$$\sigma = \frac{\mu_B(\theta)\lambda_{\text{lock}}}{3J\bar{\lambda}_c} \mathcal{L}^{-1}\left(\frac{\bar{\lambda}_c}{\lambda_{\text{lock}}}\right) (\bar{\lambda}^2 - \bar{\lambda}_c^2) \quad (62)$$

316 where $J = \lambda^{1-2\nu}$ and $\bar{\lambda}_c = \sqrt{\frac{1}{3}\left(\bar{\lambda}^2 + \frac{2}{\bar{\lambda}^{2\nu}}\right)}$. Using a Poisson's ratio ν equal to 0.49 and comparing the onset
 317 of strain hardening from Equation (62) with that from the experimental stress-strain curve at the reference
 318 temperature $\theta_{\text{ref}} = 298.15$ K, we find the values $\mu_{B,\text{ref}} = 2.0$ MPa and $\lambda_{\text{lock}} \approx 5.2$.

319 Next, the deformation dependent reference strain rates are found by fitting the expression for the viscous
 320 stress, σ_V in Equation (19), to the flow stress minus the stress contribution from Part B at different levels of
 321 deformation while keeping all parameters except the reference strain rate constant. From Figure 7 it is read-
 322 ily seen that there is a decrease in the reference strain rates as the deformation is increased. Equation (21) is
 323 proposed to describe the deformation dependence of the reference strain rates $\dot{\rho}_{0,\alpha}^*$ and $\dot{\rho}_{0,\beta}^*$. A least squares
 324 fit of Equation (21) to the data in Figure 7 yielded the initial values: $b_\alpha = 7.2$ and $b_\beta = 12.0$.

325 4.4. Material parameters

326 The material parameters obtained in the previous sections were used as initial values in a numerical op-
 327 timization procedure where simulations were run and the parameters varied manually to fit the experimental

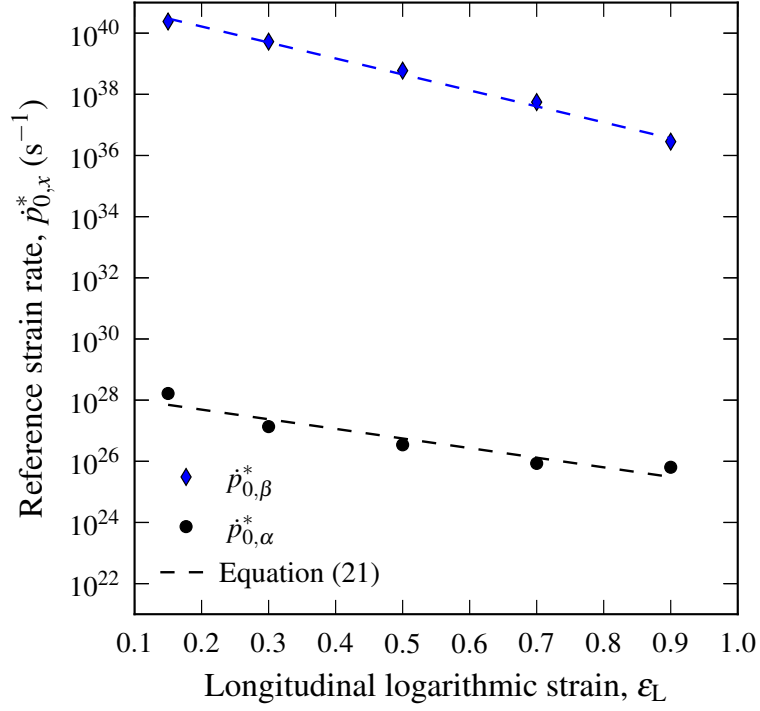


Figure 7: Reference strain rates, $\dot{\rho}_{0,x}^*$, as a function of longitudinal logarithmic strain.

328 data. An alternative would have been to use an optimization software. The material parameters used in the
 329 subsequent numerical simulations are presented in Table 3.

Table 3: Optimized parameters in constitutive model.

Part A	$\mu_{A,ref}$ (MPa)	a_A (K^{-1})	θ_{ref} (K)	ΔH_α (kJ/mol)	V_α (nm^3)	$\dot{\rho}_{0,\alpha}$ (s^{-1})	b_α (-)	ΔH_β (kJ/mol)	V_β (nm^3)	$\dot{\rho}_{0,\beta}$ (s^{-1})	b_β (-)
	46	0.028	298.15	179.5	4.72	$2.36 \cdot 10^{25}$	3.0	196.1	3.19	$6.13 \cdot 10^{36}$	10.0
Part B	$\mu_{B,ref}$ (MPa)	λ_{lock} (-)	ν (-)								
	2.0	5.2	0.49								

330 5. Finite element model

331 All simulations were run in the commercial finite element program Abaqus/Standard, with the constitu-
 332 tive model implemented through a UMAT subroutine. Due to the symmetry of the tensile specimen and to

333 save computational time, axisymmetric boundary conditions were employed in addition to one symmetry
 334 plane, as indicated in Figure 8. Consequently, the transverse deformation anisotropy observed in the exper-
 335 imental tests is not included. Four-node axisymmetric elements with reduced integration and one thermal
 336 degree of freedom (CAX4RT) were used in all simulations with an element size of approximately 0.1 mm
 $\times 0.05 \text{ mm}$ in the parallel part. Only a 1 mm portion of the grips was included in the model to further

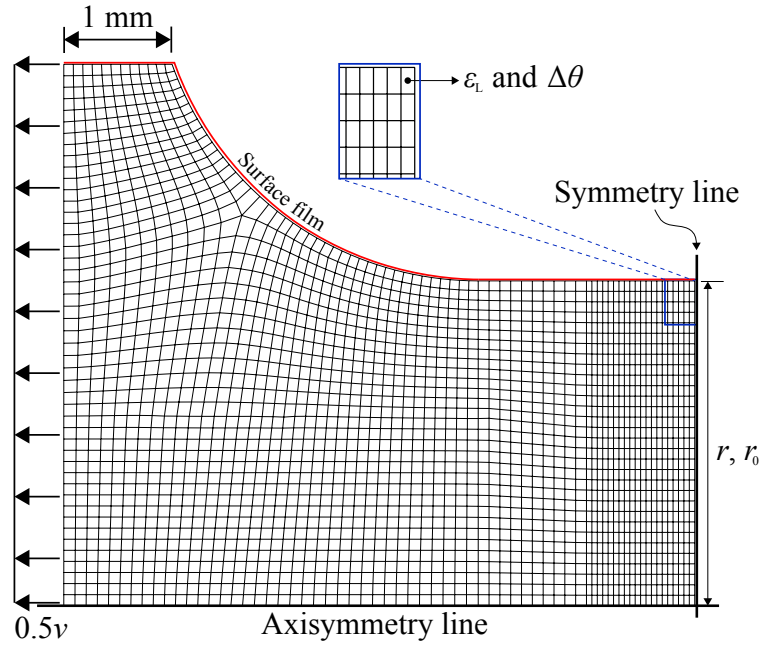


Figure 8: Axisymmetric finite element model with mesh and boundary conditions.

337
 338 reduce the computational time. The cross-head velocity, v , of the testing machine was applied as a velocity
 339 boundary condition at the positions indicated in Figure 8. Self-heating, $\Delta\theta$, and longitudinal strain, ε_L , were
 340 extracted from the indicated element in Figure 8, while the transverse strains were calculated as an average
 341 over the cross section at the symmetry line, i.e., $\varepsilon_1 = \varepsilon_2 = \ln(r/r_0)$, where r and r_0 are the current and initial
 342 radius of the parallel section, respectively. The Cauchy stress was then found using Equation (1), where
 343 $\lambda_1 = \lambda_2 = \exp(\varepsilon_1)$ is used to calculate the current area A and the force F is extracted from the boundary
 344 conditions on the symmetry line.

345 In addition to the mechanical boundary conditions, a surface film was applied on the free surface of the
 346 tensile specimen, see the area highlighted with red in Figure 8. The surface film was used to simulate heat
 347 convection to air. Heat conduction within the material itself and heat convection to the surroundings were
 348 handled by the thermal solver in Abaqus. The values of the heat convection to air parameter, h_c , the thermal

349 conductivity, k , and the specific heat capacity, C_p , are given in Table 1. Lastly, the entire axisymmetric
350 model was given an initial temperature equal to the surrounding temperature using the predefined field
351 feature in Abaqus/Standard.

352 **6. Results and discussion**

353 A comparison of the numerical results and the experimental results obtained by Johnsen et al. [11] are
354 presented in the following. All numerical and experimental values were obtained from uniaxial tension
355 tests. Note that the results from the repeat tests presented in [11] are omitted, thus only the representative
356 experimental results are included in this study.

357 *6.1. Stress-strain curves*

358 Figure 9 presents the axial component of the Cauchy stress tensor as a function of the longitudinal
359 logarithmic strain from both simulations and experiments. Twelve configurations of temperature and strain
360 rate were investigated in total: four temperatures T of 25 °C, 0 °C, -15 °C and -30 °C and for each
361 temperature three nominal strain rates $\dot{\epsilon}$ of 0.01 s⁻¹, 0.1 s⁻¹ and 1.0 s⁻¹.

362 As shown in Figure 9, the overall behaviour of the material is well described by the constitutive model,
363 although the strain rate effect on Young's modulus (Figure 5) is not captured since viscoelasticity is not
364 incorporated. It appears from Figure 9 that the yield stress is accurately represented for all test configura-
365 tions by the incorporated Ree-Eyring [6] flow theory. Furthermore, we see that the strain hardening is well
366 described up to the onset of network hardening for all configurations except at room temperature. At room
367 temperature the onset of network hardening occurs too early in the simulations. However, as seen from
368 Figure 9, the onset of network hardening is continuously shifted to higher strain levels as the temperature
369 is decreased. This is caused by the constant locking stretch in combination with the reduced shear modulus
370 (Equation (31)) for decreasing temperatures in Part B of the constitutive model.

371 *6.2. Volume change*

372 The volumetric strain from the simulations was calculated using the longitudinal strain from the indi-
373 cated element in Figure 8 and the average transverse strain over the cross section, viz.

$$\varepsilon_V = \varepsilon_L + 2\varepsilon_1 = \varepsilon_L + 2 \ln\left(\frac{r}{r_0}\right) \quad (63)$$

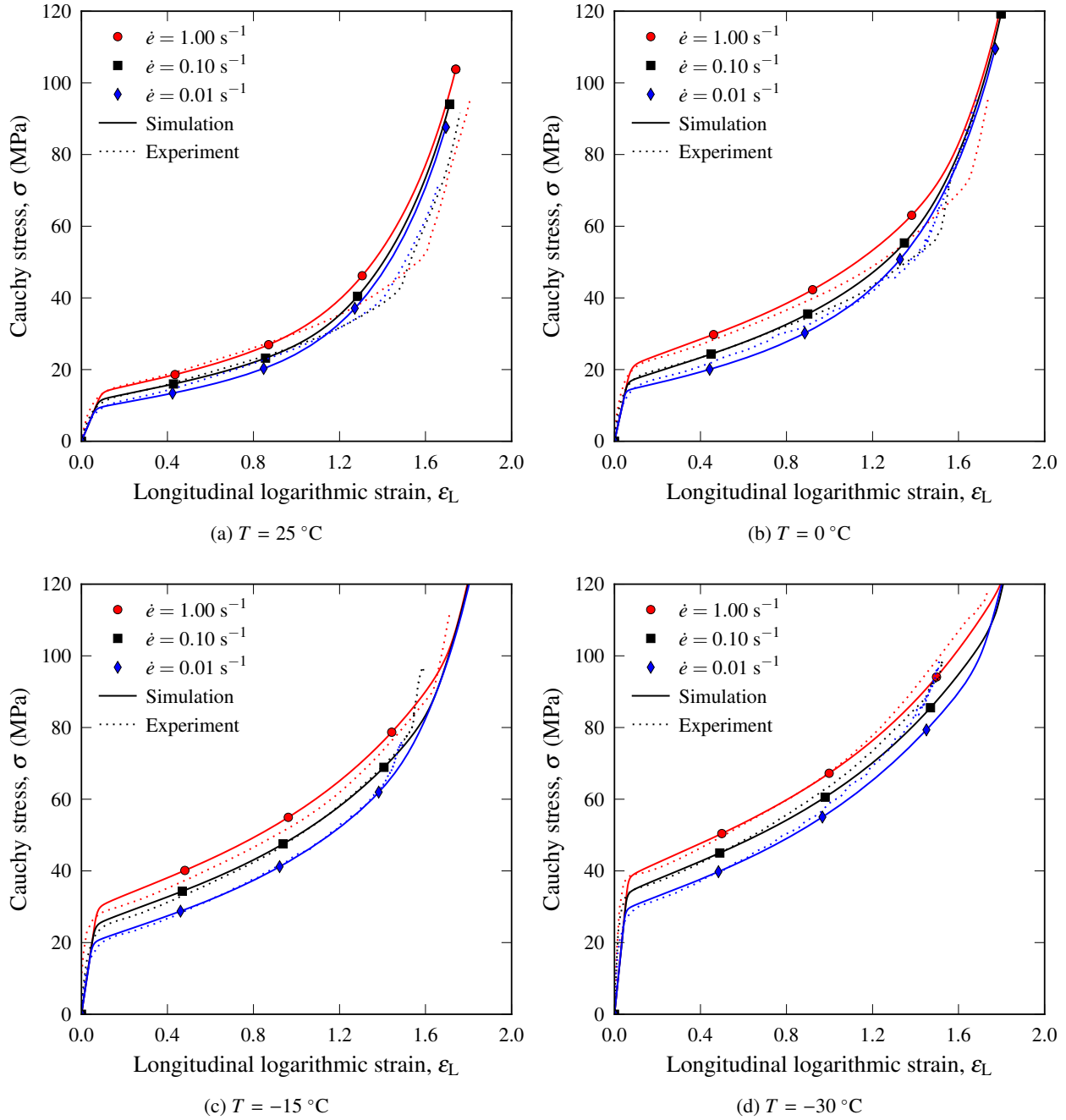


Figure 9: Cauchy stress vs. longitudinal logarithmic strain from uniaxial tension tests and numerical simulations at three different nominal strain rates, $\dot{\epsilon} = 0.01\text{ s}^{-1}$, $\dot{\epsilon} = 0.1\text{ s}^{-1}$, and $\dot{\epsilon} = 1.0\text{ s}^{-1}$, and at four different temperatures, (a) $T = 25^\circ\text{C}$, (b) $T = 0^\circ\text{C}$, (c) $T = -15^\circ\text{C}$ and (d) $T = -30^\circ\text{C}$.

374 Figure 10 compares the volumetric strain from simulations and experiments for all test configurations.

375 Qualitative agreement between numerical predictions and experimental results is achieved at all investigated

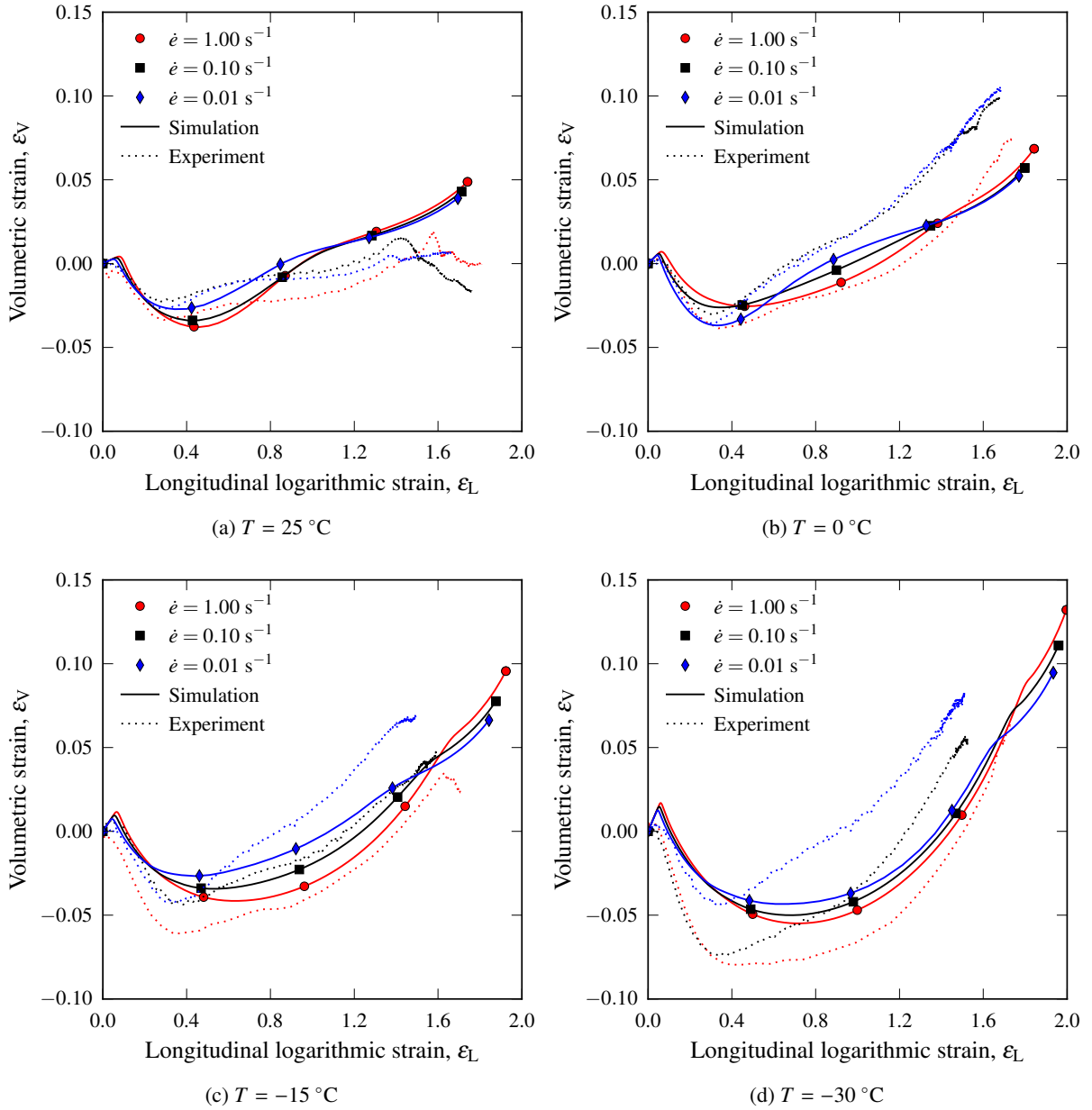


Figure 10: Volumetric strain vs. longitudinal logarithmic strain from uniaxial tension tests and numerical simulations at three different nominal strain rates, $\dot{\epsilon} = 0.01\text{ s}^{-1}$, $\dot{\epsilon} = 0.1\text{ s}^{-1}$, and $\dot{\epsilon} = 1.0\text{ s}^{-1}$, and at four different temperatures, (a) $T = 25^\circ\text{C}$, (b) $T = 0^\circ\text{C}$, (c) $T = -15^\circ\text{C}$ and (d) $T = -30^\circ\text{C}$.

376 temperatures. Due to the assumption of a constant Poisson's ratio, ν , and the entropic nature of the bulk
 377 modulus in Part B, the model also captures the observed transition from an approximately incompressible
 378 material behaviour at room temperature to a more compressible behaviour at the lower temperatures.

379 In agreement with what is observed in experiments [11], reducing the initial temperature results in
380 more negative volumetric strain at moderate deformations in the numerical simulations. This is due to
381 the formation of a more prominent neck, causing the strain field to become more heterogeneous. The
382 heterogeneity of the strain field causes our method of calculating the volumetric strain, i.e., using the average
383 longitudinal and transverse strain over the cross-section, to be less representative of the actual state inside the
384 material, leading to the fictitious negative evolution of the volumetric strain in the beginning. A method to
385 avoid this problem is to try estimating the heterogeneity of the strain field in the experiments, as proposed by
386 Andersen [48] and used by Johnsen et al. [31]. However, since the volumetric strain presented in Figure 10
387 is calculated in a similar manner in experiments and simulations, this method was not further explored in
388 this study.

389 6.3. Self-heating

390 The temperature increment due to self-heating in the material is given as a function of longitudinal
391 logarithmic strain in Figure 11. Good qualitative agreement is found between simulations and experiments.
392 In the uniaxial tension tests at the lowest strain rate, close to isothermal conditions are predicted. At the
393 intermediate strain rate the predicted temperature increment from simulations is in good agreement with
394 experimental observations. However, at the highest strain rate, the model does not generate enough heat.
395 This is due to the interplay between the elastic and plastic components of Part A, see Figure 4b. Since the
396 elastic stiffness in Part A is reduced for increasing temperature the consequence is a negative contribution
397 to heat generation, which has to be compensated by the plastic dissipation in the viscous dashpots and the
398 entropic spring in Part B. Furthermore, as the initial temperature decreases, the elastic stiffness increases,
399 thus reducing elastic deformation and in effect the elastic rate-of-deformation. This is the reason why the
400 constitutive model predicts a higher temperature increase as the initial temperature is lowered.

401 Another possible explanation for the observed discrepancies could be inaccuracies in the measured
402 heat on the surface of the specimen during testing, along with uncertainties in the experimentally obtained
403 thermal constants. The laser flash method [49] was used to obtain the specific heat capacity and the thermal
404 conductivity. Due to limitations in the testing apparatus, it was not possible to measure the parameters at
405 low temperatures. Consequently, the specific heat and thermal conductivity were estimated at three elevated
406 temperatures of 25 °C, 35 °C and 50 °C. The thermal conductivity ($k = 0.56 \text{ W}/(\text{m}\cdot\text{K})$) was more or less
407 constant over the investigated temperatures with a standard deviation of $0.048 \text{ W}/(\text{m}\cdot\text{K})$, while the specific
408 heat varied almost linearly with temperature, see Johnsen et al. [31]. However, the values obtained at room

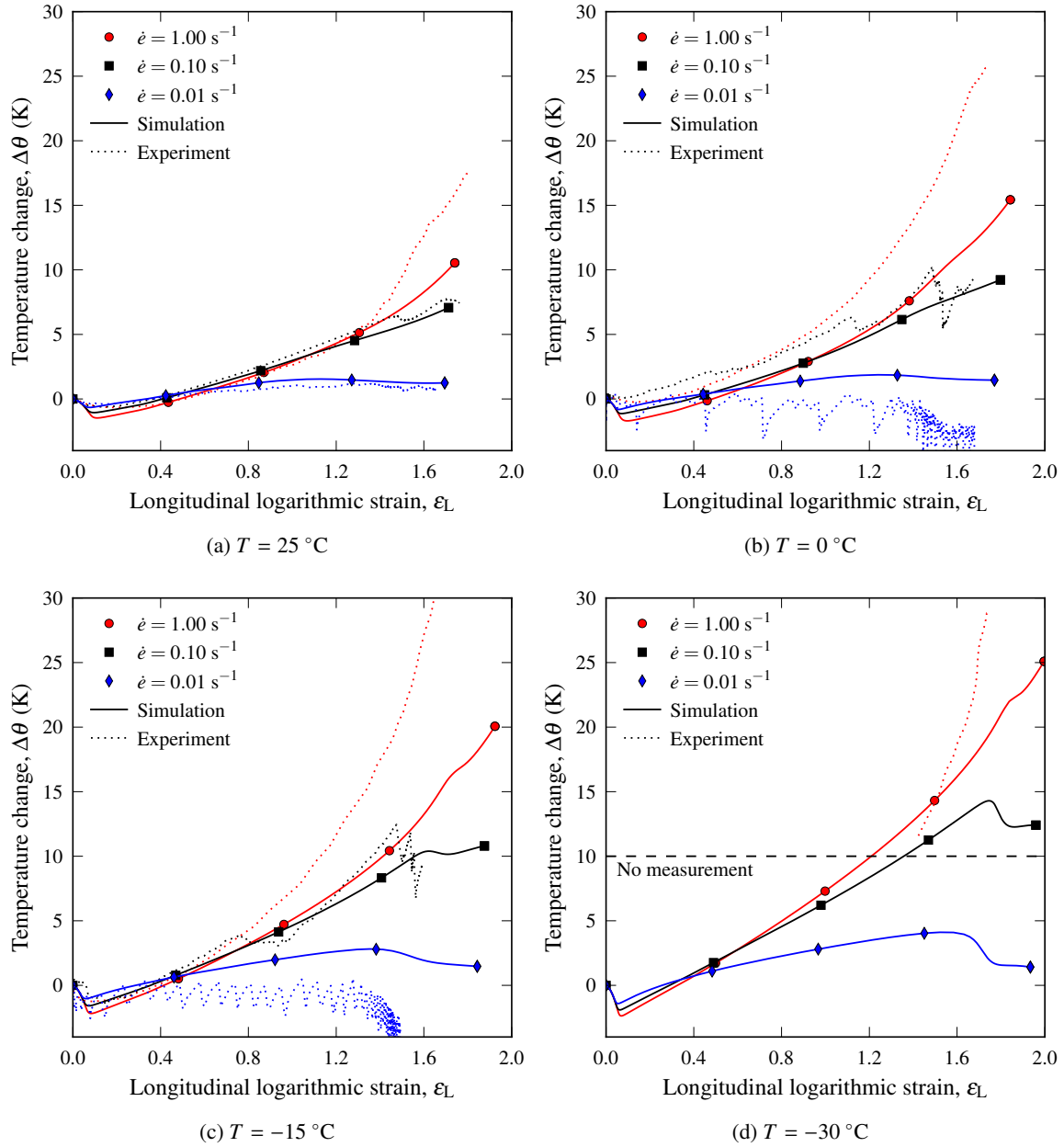


Figure 11: Temperature change vs. longitudinal logarithmic strain from uniaxial tension tests and numerical simulations at three different nominal strain rates, $\dot{\epsilon} = 0.01 \text{ s}^{-1}$, $\dot{\epsilon} = 0.1 \text{ s}^{-1}$, and $\dot{\epsilon} = 1.0 \text{ s}^{-1}$, and at four different temperatures, (a) $T = 25 \text{ }^\circ\text{C}$, (b) $T = 0 \text{ }^\circ\text{C}$, (c) $T = -15 \text{ }^\circ\text{C}$ and (d) $T = -30 \text{ }^\circ\text{C}$.

409 temperature were used for both the specific heat capacity and the thermal conductivity in the simulations.

410 Note that the thermal camera used in the experiments by Johnsen et al. [11] was limited to temperatures
 411 above $-20 \text{ }^\circ\text{C}$, as indicated by the dashed line in Figure 11d. It should also be mentioned that the jagged

412 shape of the temperature increment vs. longitudinal strain curves at temperatures below 25 °C is caused by
413 the influx of liquid nitrogen during the tension test.

414 6.4. Force-displacement curves

415 As a further validation incorporating the response of the entire tension test sample, force vs. displace-
416 ment curves are shown in Figure 12. The evolution of the force up to the peak value is well captured,
417 along with the subsequent force drop. In the simulations of the room temperature experiments, the force
418 levels are in general overestimated. This is attributed to a too high value of the shear modulus in Part B, in
419 combination with a too low value of the locking stretch, thus overestimating the strain hardening. For the
420 temperature 0 °C, good agreement is found between simulation and experiment for the two lowest strain
421 rates. At the highest strain rate there is not enough reduction in force after the peak force is reached, which
422 for this configuration is caused by the deformation dependent reference strain rates. For the two lowest
423 temperatures, a combination of the aforementioned effects is observed. At -30 °C the force reduction is
424 overestimated due to the reduced shear modulus in Part B ($\mu_B \approx \mu_{A,ref} \cdot \frac{243.15}{298.15} = 0.81\mu_{A,ref}$), while at -15 °C
425 the model underestimates the force reduction because of the isotropic hardening of the viscous dashpots.

426 6.5. Strain rate

427 As shown in Figure 13, there is an overall good agreement between the strain rate from simulations,
428 extracted from the indicated element in Figure 8, and the strain rate from experiments. At room temperature
429 the strain rate in the simulations decreases too rapidly. This is due to strain hardening from Part B of the
430 model, which decreases the strain rate by propagating the neck too early. As seen from Figure 13 this
431 effect is continuously decreased as the initial temperature is reduced, which is caused by the reduced shear
432 modulus in Part B. The reduced shear modulus delays the onset of network hardening, which again allows
433 for a more prominent neck to form causing, or maintaining, the strain rate for a longer period before the
434 neck starts to propagate and the strain rate goes down. Furthermore, when the neck is fully propagated, the
435 strain rate stops decreasing and a sudden increase in strain rate is observed in all experiments and in the
436 simulations at the two lowest temperatures. This is caused by the re-straining of the specimen which occurs
437 when the neck is fully propagated to the shoulders.

438 6.6. Strain-displacement curves

439 A comparison of the local strain in the most deformed section of the specimen vs. the global displace-
440 ment curves from simulations and experiments is given in Figure 14. The displacement in the finite element

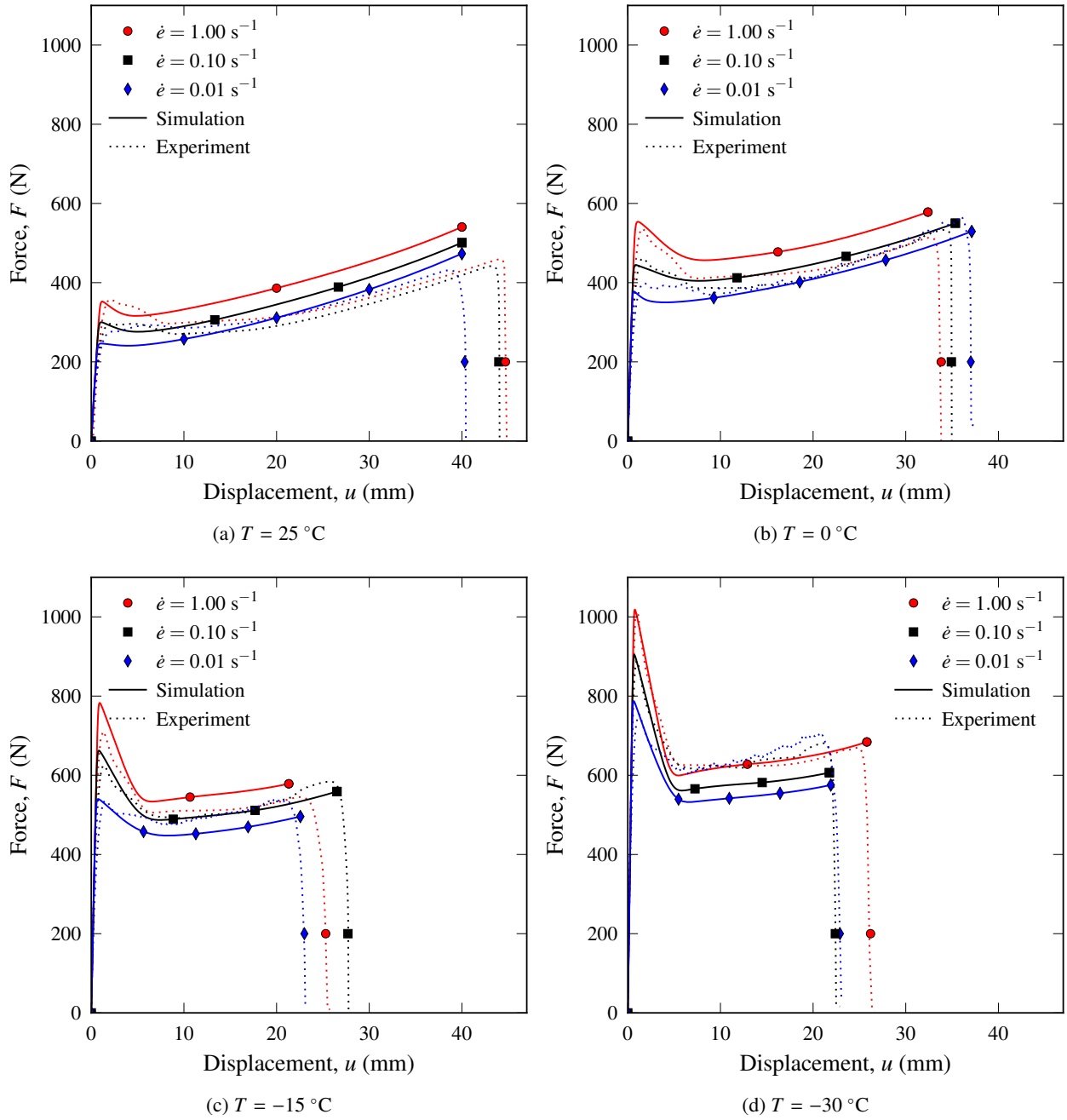
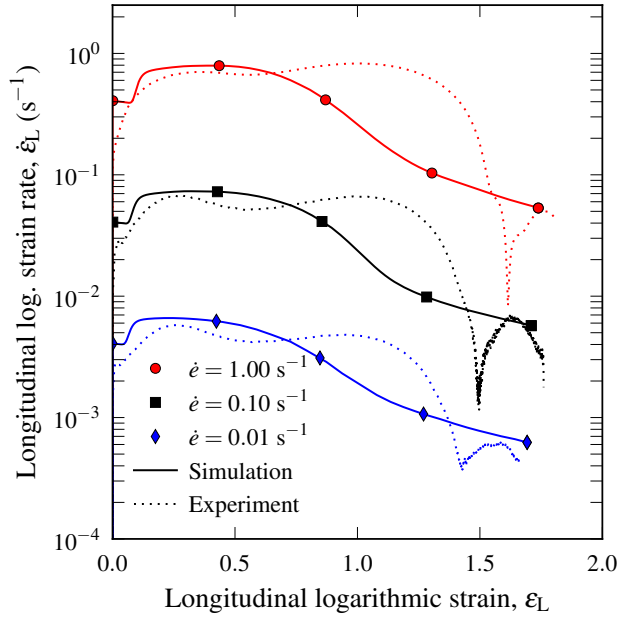


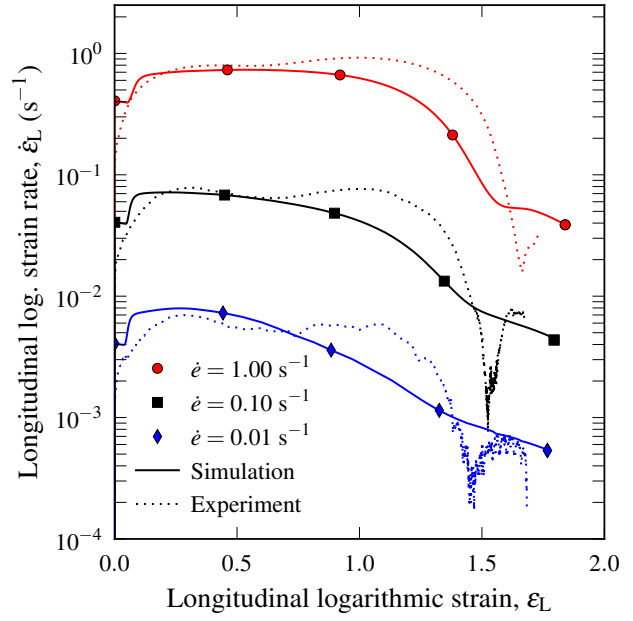
Figure 12: Force vs. displacement curves from uniaxial tension tests and numerical simulations at three different nominal strain rates, $\dot{\epsilon} = 0.01 \text{ s}^{-1}$, $\dot{\epsilon} = 0.1 \text{ s}^{-1}$, and $\dot{\epsilon} = 1.0 \text{ s}^{-1}$, and at four different temperatures, (a) $T = 25 \text{ }^\circ\text{C}$, (b) $T = 0 \text{ }^\circ\text{C}$, (c) $T = -15 \text{ }^\circ\text{C}$ and (d) $T = -30 \text{ }^\circ\text{C}$.

441 model was extracted at the nodes where the velocity boundary condition was applied, see Figure 8.

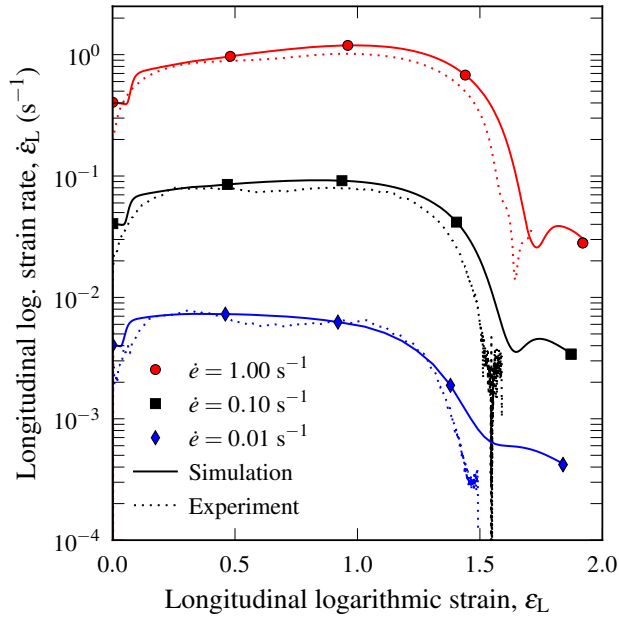
442 Due to the constant locking stretch, the longitudinal strain saturates at approximately the correct level



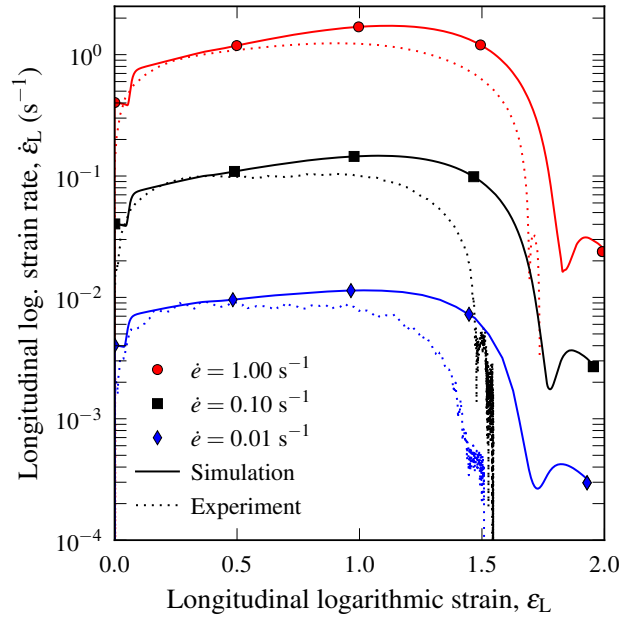
(a) $T = 25\text{ }^{\circ}\text{C}$



(b) $T = 0\text{ }^{\circ}\text{C}$



(c) $T = -15\text{ }^{\circ}\text{C}$



(d) $T = -30\text{ }^{\circ}\text{C}$

Figure 13: Longitudinal logarithmic strain rate vs. longitudinal logarithmic strain from uniaxial tension tests and numerical simulations at three different nominal strain rates, $\dot{\epsilon} = 0.01\text{ s}^{-1}$, $\dot{\epsilon} = 0.1\text{ s}^{-1}$, and $\dot{\epsilon} = 1.0\text{ s}^{-1}$, and at four different temperatures, (a) $T = 25\text{ }^{\circ}\text{C}$, (b) $T = 0\text{ }^{\circ}\text{C}$, (c) $T = -15\text{ }^{\circ}\text{C}$ and (d) $T = -30\text{ }^{\circ}\text{C}$.

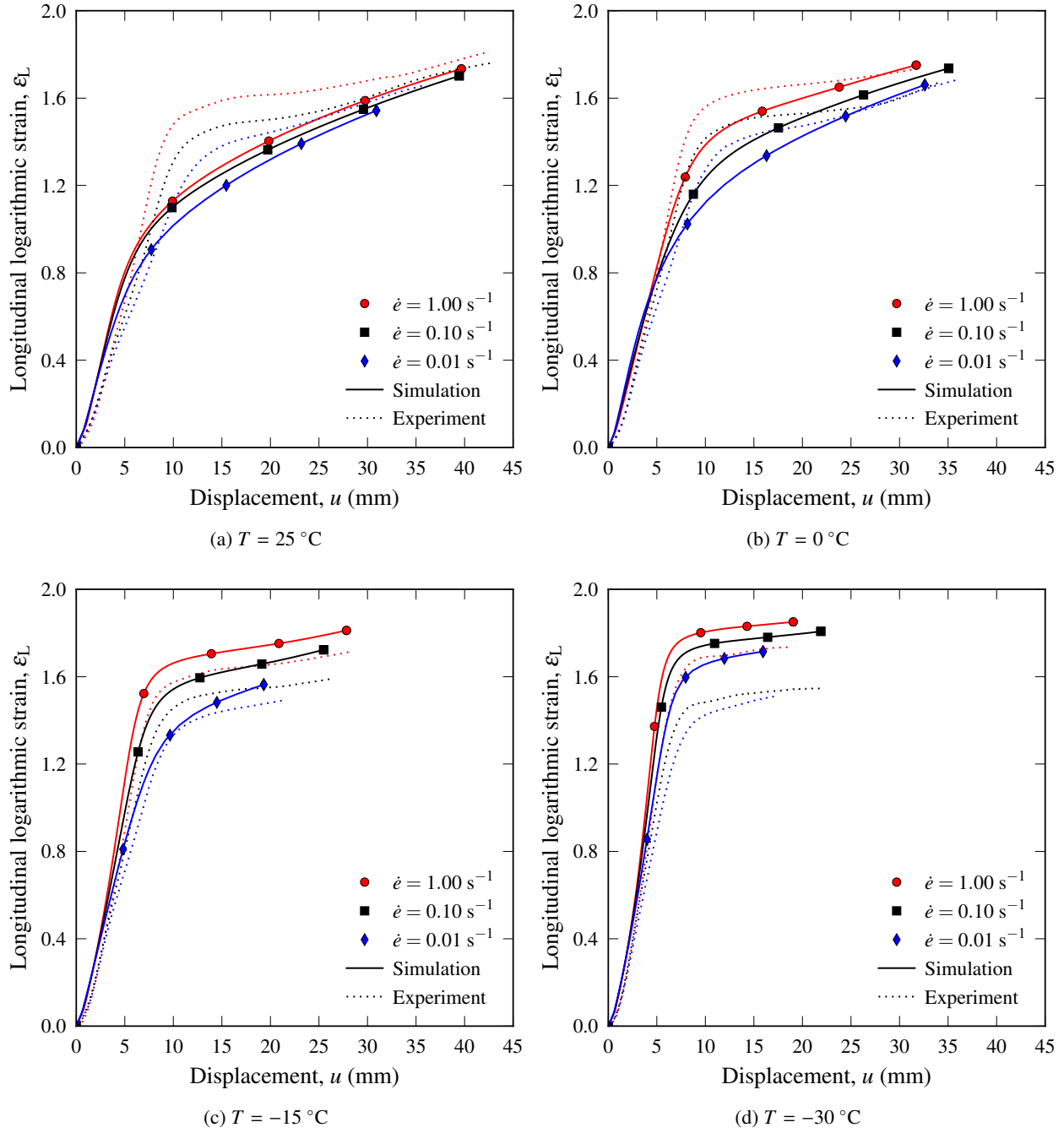


Figure 14: Local longitudinal logarithmic strain vs. global displacement from uniaxial tension tests and numerical simulations at three different nominal strain rates, $\dot{\epsilon} = 0.01 \text{ s}^{-1}$, $\dot{\epsilon} = 0.1 \text{ s}^{-1}$, and $\dot{\epsilon} = 1.0 \text{ s}^{-1}$, and at four different temperatures, (a) $T = 25 \text{ }^\circ\text{C}$, (b) $T = 0 \text{ }^\circ\text{C}$, (c) $T = -15 \text{ }^\circ\text{C}$ and (d) $T = -30 \text{ }^\circ\text{C}$.

443 for all simulations. However, as has been the case for previous simulation results, the change in the shear
444 modulus in Part B of the model is clearly evident. At room temperature, the strain saturates more gradually,
445 as seen in Figure 14a. As the temperature is decreased, the shear modulus in Part B is continuously reduced
446 leading to a rather accurate prediction of the longitudinal strain as a function of global displacement at
447 a temperature of $-15\text{ }^{\circ}\text{C}$ (Figure 14c). At a temperature of $-30\text{ }^{\circ}\text{C}$ (Figure 14d), the shear modulus has
448 been reduced too much, causing the longitudinal strain to saturate at a level which is too high. However, it
449 should be noted that the global displacement measured in the experiments is not directly comparable to the
450 displacement in the simulations. The reason for this is twofold: (1) the specimen was clamped in the testing
451 machine which could have caused some slippage between the clamping rig and the tensile specimen, and (2)
452 the machine stiffness could have affected the displacement recorded by the testing machine. Nevertheless,
453 Figure 14 demonstrates the constitutive model's capability of capturing both the local and global material
454 behaviour of the tensile specimen.

455 6.7. Comparison of deformed shape

456 Figure 15 shows a comparison between the deformed shape of the specimen from experiments and
457 simulations at room temperature and a strain rate of $\dot{\epsilon} = 1.0\text{ s}^{-1}$. The deformed shape of the finite ele-
458 ment model is outlined in red on the images from the experiments. As evident from Figure 15, there are
459 some discrepancies between simulation and experiment. At a relatively small displacement of $u = 3\text{ mm}$
460 (Figure 15a) the agreement between simulation and experiment is excellent. However, at a displacement
461 of 8 mm , the simulation deviates from experiment. The specimen has not contracted enough due to the
462 network hardening from Part B which limits the neck formation and accelerates neck propagation. All these
463 observations can be explained from Figure 14a where we see that at $u = 3\text{ mm}$ there is excellent agreement
464 between simulation and experiment. After $u \approx 6\text{ mm}$ the simulation starts to deviate from the experiment
465 due to the network hardening in Part B limiting the longitudinal strain, and a displacement of approximately
466 35 mm has to be reached before the longitudinal strain from simulation and experiment agrees again.

467 7. Concluding remarks

468 We have presented a thermo-elasto-viscoplastic constitutive model describing the thermomechanical
469 behaviour of a cross-linked low density polyethylene (XLPE) at different temperatures and strain rates.
470 The constitutive model consists of two parts: Part A represents thermoelasticity and thermoviscoplasticity,

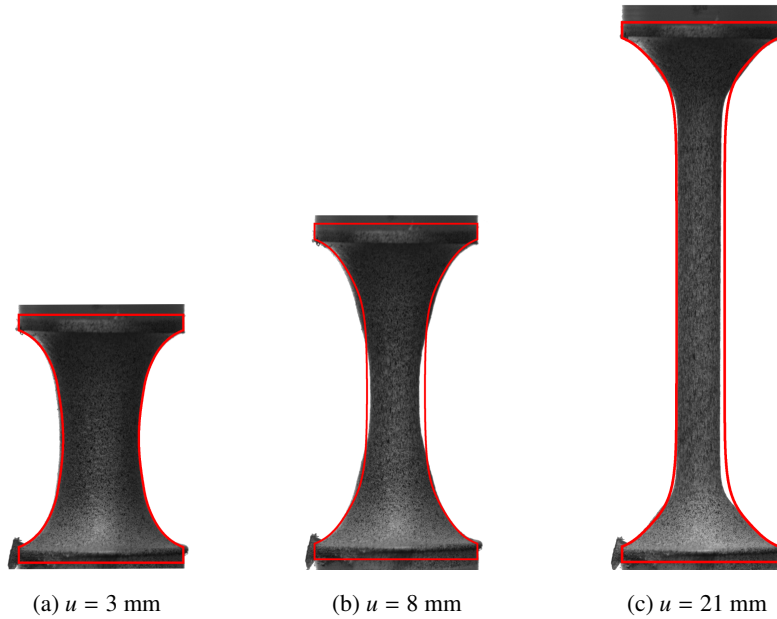


Figure 15: A comparison of the deformed shape of a specimen tested at $T = 25\text{ }^{\circ}\text{C}$ and $\dot{\epsilon} = 1.0\text{ s}^{-1}$ from finite element analysis and experiment at three magnitudes of displacement: (a) 3 mm, (b) 8 mm and (c) 21 mm. The deformed shape from the finite element analysis is outlined in red on the images from the experiment.

471 whereas Part B represents entropic strain hardening due to alignment of the polymer chains during defor-
 472 mation. Assuming that the contributions from the main α and the secondary β relaxation processes are
 473 additive, Ree-Eyring dashpots were successfully used to describe yielding as a function of temperature and
 474 strain rate. In addition, the yield stress of the material was modelled as pressure insensitive, and the plastic
 475 flow was taken to be isochoric. There were two contributions to strain hardening in the model: (1) kinematic
 476 hardening from the eight chain spring in Part B, and (2) isotropic hardening introduced by the deformation
 477 dependent reference strain rates in the viscous dashpots. A phenomenological expression was proposed to
 478 describe the increase in Young's modulus as the material was cooled down. The constitutive model was im-
 479 plemented in a nonlinear finite element (FE) code using a semi-implicit stress update algorithm combined
 480 with sub-stepping and a numerical scheme to calculate the consistent tangent operator.

481 The constitutive model was calibrated from the stress-strain curves obtained in uniaxial tension tests
 482 performed at four different temperatures and three nominal strain rates, as reported in [11]. Considering
 483 the stress-strain curves, good agreement between simulations and experiments was achieved, as evident by
 484 Figure 9. For the temperature increase, qualitative agreement was obtained between numerical predictions
 485 and experimental values. The predictions by the FE model in terms of volumetric strain, force vs. global

486 displacement, local strain vs. local strain rate, global displacement vs. strain and the deformed shape of the
487 tensile specimen were in good overall agreement with the experimental counterparts, and these results serve
488 as validation in the sense that the material model, which is calibrated from local stress-strain data, is able to
489 predict the global response adequately.

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