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Probabilistic Low Cycle Fatigue criterion for nodular cast-irons

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Abstract

This paper proposes an original method for characterising the Low Cycle Fatigue (LCF) lifetime using probability density functions. The protocol is based on statistics of microstructure heterogeneities taken as damage initiation sites, a qualitative mechanical analysis of the heterogeneities harmfulness and the definition of a micro-crack growth law. The technique is here established and the associated model identified for a nodular cast iron where the graphite nodules are assumed to be the damage initiation zones. The LCF lifetime is characterised from both a large set of experimental test between 300 and 600 °C and damage observations at the micro-scale. Experimental post-mortem observation combined with a simple numerical study first enable to assume the harmfulness of nodules according to their size and their probable role in the damage process A probability density function for the lifetime is then built from the following steps: (i) a quantitative analysis of the material micro-structure, which provides the probability density of nodules occurrence depending of their size (ii) an extreme value analysis using a Gumbel distribution and (iii) a micro-crack growth law associated with LCF conventional terms of energy densities. Its parameters are obtained using an optimisation process applied to laboratory fatigue experiment. The obtained probability function provides a good match for the lifetime and greatly improves results given by conventional criteria. It moreover provides a robust estimate of the lifetime scatter for different types of fatigue tests.

Keywords: High Temperature Low Cycle Fatigue, Thermomechanical fatigue,

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Cast-iron, Criterion, Microstructure, Probability of failure

1 1. Introduction

Cast-iron has been a traditional engineering material in the past centuries. Nev-2 ertheless, its castability, low price and mechanical resistance induces that it is still 3 an actual materials of choice for parts exposed to high temperatures such as auto-4 motive exhaust manifolds or turbochargers. Nodular and compacted cast-irons are 5 indeed still a strong material option despite recent technology innovations, such as 6 cylinder heads with integrated exhaust manifolds [1]. Such structures operate at 7 high temperatures where the material behaviour becomes strongly non-linear and 8 exhibits viscoplastic effects. Moreover, exposed to large thermal gradients as well 9 as severe mechanical boundary conditions, they often experience high tempera-10 ture LCF or even thermo-mechanical fatigue (TMF), that lead to the appearance of 11 cracks during operation [2, 3, 4]. 12

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The state-of-art in design for these structures evolved in the last decades from 14 prototype testing to virtual design performed using mechanical analysis software 15 as proposed in a pioneering paper [5] for an automotive application. In order to 16 keep the computational costs under control, they proposed as a main assumption to 17 decouple material non-linear behaviour and damage provided the constitute model 18 to be precisely identified. The structure lifetime is then associated with the initi-19 ation of a so-called macroscopic crack (of the order of a millimetre). The fatigue 20 criterion then defines a number of cycles which is related to a local dissipated en-21 ergy density as proposed initially by [6, 7]. 22

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The method has since successfully been applied on various structures by sev-24 eral authors, see for example [8, 9, 10, 11, 12, 13]. The fatigue criterion parameters 25 are usually identified from conventional LCF tests [14] at different temperatures. 26 The initial criterion has also been adapted to account for various effects, like mean 27 stress or multi-axial loading [15, 16, 17]. The dissipated energy LCF criterion has 28 thus been found to be rather robust[18]. Moreover, under LCF conditions, several 29 groups [19, 20, 14] demonstrated that the criterion can be physically justified when 30 31 damage and lifetime are mainly controlled mainly the propagation of micro-cracks at the micro-structure scale. 32

33

A major drawback of this method, however, is that it does not allow the estimation of fatigue behaviour randomness. Material fatigue failure has indeed a random aspect that can be observed through the scatter of the lifetimes for a given

load and the occurrence of various crack initiation zones observed during testing. 37 The randomness of material properties, defects or even manufacturing variation 38 cause the uncertainty of damage variable-lifetime relationship and seriously affect 39 the dispersion of fatigue life. The previously discussed criterion has a determin-40 istic formalism that thus prevent a reliable analysis of lifetime scatter, although it 41 is highlighted in various studies especially when LCF is investigated over different 42 temperatures [21, 22, 23, 24]. Moreover, recent probabilistic TMF design proto-43 cols for automotive parts relying on stress-strength interference analysis absolutely 44 need a reliable estimation of this scatter [25]. Randomness is however usually in-45 troduced in the design process by a conservative strategy which induces most often 46 arbitrary safety factors. 47

48

The aim of this paper is to overcome this fundamental drawback by enriching 49 the criterion with a micro-mechanical analysis covering the random character of 50 the damage initiation sites and keeping the computational complexity and costs of 51 the design method at the same level as before. Several attempts have already been 52 proposed in the literature in this direction. For example, Doudard and Calloch [26] 53 included a Weibull distribution for the defects occurrence in order to explain the 54 evolution of self-heating measurements for high-cycle fatigue (HCF). In the LCF 55 domain, Maurel et al. [27] proposed a refined engineering fatigue model based on 56 the dissipated energy and micro-crack growth law based on microstructural char-57 acteristics for stainless steels. A further step was proposed in Tabibian et al.[23] 58 where pores in a lost-foam cast aluminium alloy are considered a initiation site for 59 LCF micro-crack propagation. A statistical analysis of pores sizes is then com-60 bined with a micro-crack basic growth law and a standard LCF criterion in order 61 to successfully predict the experimental scatter. 62

63

These previous attempts are then used as a basis to build a LCF fatigue criterion 64 which explicitly incorporates the statistics of micro-structural defects responsible 65 for the damage initiation. The material under scrutiny is a cast-iron for which a 66 fatigue criterion is sought over a wide temperature range (300 to 600 °C). Such 67 foundry alloys usual incorporate a large range of micro-structural heterogeneities 68 which can evolve into damage initiation sites. In the particular case of cast-irons, 69 carbides and graphite may appear in lamellar, compacted or spheroidal form and 70 coexist with the iron matrix. Casting of complex parts also generate differential 71 shrinkage and conduct to the formation of cavities or pores as documented in 72 73 [28, 29]. The different phases and/or pores are randomly distributed in the material and often exhibit complex mythologies characterised by statistical distributions of 74 size and shape. The study of their role in fatigue damage is commonly highlighted 75 for high cycle fatigue[30] especially for aluminium alloys in [31, 32, 33, 34] and 76

⁷⁷ for cast-iron in [35, 36, 37, 38, 39, 40].

78

In nodular cast-iron, "Void ratcheting" based on extended plasticity was iden-79 tified as a potential mechanism leading to failure during experimental observations 80 [41] and numerical investigations [42]. In a theoretical problem setting, Mbiakop 81 et al. [43] showed, using finite element computations, that cyclic loading of pores 82 steer the local plastic field and the pores evolution shape indicating the onset of 83 cracks. Moreover, the study highlighted the importance of the local stress triaxial-84 ity and that kinematic hardening decreases the rate of the evolution when compared 85 with isotropic hardening. In the other hand, the appearance of pores at the poles of 86 graphite nodules has been observed in the literature [44] and it seems a reasonable 87 assumption to consider them as potential initiation zones for fatigue micro-cracks. 88 As a consequence, it is therefore relevant to evaluate their role in the case of high 89 temperature Low-Cycle Fatigue. Let us further mention that under these loading 90 regimes, the material exhibits generalised viscoplasticity and that strong differ-91 ences in material properties between the nodules and the matrix naturally induce 92 local strains and stresses concentrations, which may be accelerating factors in the 93 damage process. 94

95

This article starts with a the presentation of the considered nodular cast-iron 96 and the associated LCF tests and post-mortem damage analysis, which allows the 97 hypothesis of considering graphite nodules as initial defects. A quantitative anal-98 ysis of the micro-structure is then performed and the morphology of graphite nod-99 ules is examined by optical microscopy and digital post-processing. The observed 100 data, size and shape characteristics, is then represented using probability distribu-101 tions. Hypotheses are then formulated regarding the evolution of fatigue damage 102 in the material. They are supported by the post-mortem tests analysis and by a 103 qualitative numerical model that allow them to be refined and to finally propose, 104 using the statistical theory of extreme values, the distribution of the largest nod-105 ules inside a representative elementary volume. Under cyclic loading, nodules are 106 indeed considered as the origin of damage that propagates through micro-cracks 107 until failure. A fatigue model is then proposed: it is based on this distribution 108 and includes both micro-initiation and micro-propagation. As such, the path will 109 complete the idea proposed in [5], with the missing micro-structural information 110 and modelling while integrating ideas previously proposed by Seifert et al. [45]. 111 As a consequence it offers both the standard deterministic estimate of lifetime and 112 an estimate of lifetime scatter. The parameters of the model are identified from a 113 series of LCF experiments using a complete optimisation process. The final discus-114 sion exhibits their physical signification and examines the accuracy of the model to 115 describe lifetimes of structures under thermo-mechanical loading by applying the 116

model on an experimental database outside the identification sample.

118 2. Experimental Database

119 2.1. Material and LCF tests

129

SiMo nodular cast-iron is widely used by the automotive industry, especially for exhaust systems as silicon and molybdenum enhance high temperature performances. The specific material used for this study present a ferritic structure up to $850 \,^{\circ}C$. Its chemical composition is given in Table 1.

С	Si	Mo	Mn	S	Р	Ni	Cr	Cu
3.2	3.92	0.58	0.13	0.01	0.01	0.062	0.03	0.016

Table 1: Chemical composition of SiMo cast-iron (% in weight)

A quick analysis of the alloy micro-structure reveals the presence of a ferriticopearlitic matrix but with a ferrite content between 85 to 90% depending on the observed areas. As seen on Figure 1, The micro-structure is also characterised by the presence of graphite nodules with a volume fraction of about 9%.

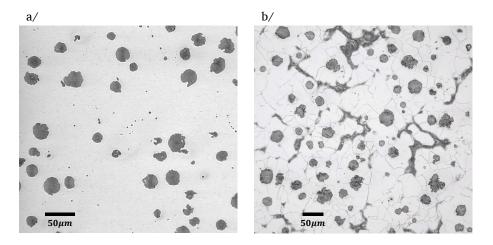


Figure 1: Microstructure of SiMo cast-iron without (a/) and with (b/) chemical etching

The study first focuses here on the Low Cycle Fatigue (LCF) of this cast-iron between 300 and $600^{\circ}C$. These temperatures present indeed a good compromise in terms of material behaviour with an equivalent influence of classical plasticity
(mainly kinematic hardening) and viscosity as shown by Szmytka et al. [46]. With
conventional LCF lifetimes [5], the damage produced by the cyclic viscoplasticity
can then be analysed without risks of high interference with oxidation or effects
induced by the temperature history, typical in the case of TMF [47].

137

A complete experimental database is carried out for this alloy on cylindrical 138 specimens with micro-structures representative of industrial parts (exhaust mani-139 folds and turbochargers). This database is constituted of the tests performed by 140 Constantinescu et al[5] at 350 and $600^{\circ}C$ and completed with new tests at 300 141 and $500^{\circ}C$. For the latter, 120s-long hold times in tension or compression have 142 been added in order to evaluate the influence of the mean stress on the lifetime. 143 The total strain rate was $10^{-3}s^{-1}$ and the proportion between extreme strains was 144 $\varepsilon_{max}/\varepsilon_{min} = -1$. Different strain amplitude $\Delta \varepsilon = \varepsilon_{max} - \varepsilon_{min}$ were tested and the 145 experiments are summarised on Table 2. These tests constitute the identification 146 database for the proposed criterion. 147

148

Temperature (° C)	Number of tests	Dwell phase	$\Delta \varepsilon(\%)$	Lifetime
300	3	none	0.6-0.8	280-1100
300	2	tension	0.6-0.8	1050-1090
300	3	compression	0.6-0.8	260-1650
350	6	none	0.7-2.0	12-610
500	3	none	0.5-0.8	490-2620
500	5	tension	0.5-0.8	490-3700
500	2	compression	0.6-0.8	410-720
600	9	none	0.5-2.0	70-8190

Table 2: Low Cycle Fatigue tests condition summary

Observed lifetimes range between 100 and 8190 cycles and are therefore characteristic of LCF conditions. A single test has a very short lifetime that could be considered as outside the LCF range with 12 cycles. However, it was decided to keep this extreme point in the database, as the loading is still representative of the structures to be designed and provides useful information on the dispersion of the fatigue response.

155 2.2. Fracture surface analysis and damage mechanisms

¹⁵⁶ Some specimens from the experimental basis were analysed to post-mortem

quantify the influence of the micro-structure on the lifetime and on fatigue damage

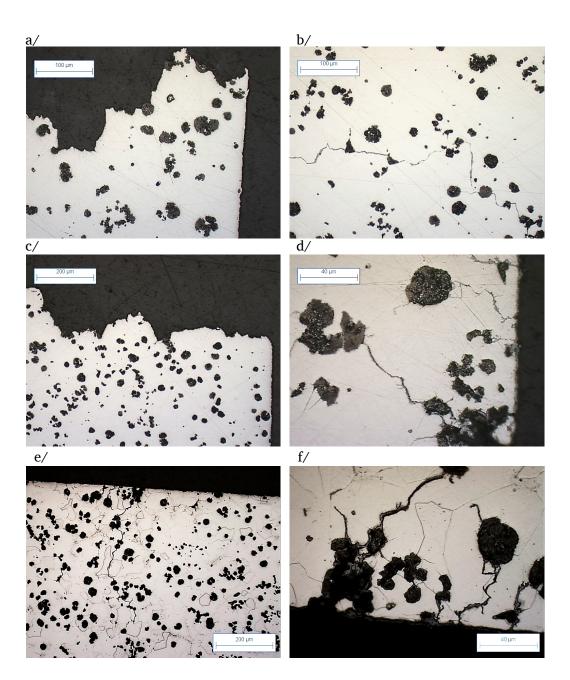


Figure 2: Fractography of LCF tested specimen at $300^{\circ}C$ (a/ and b/) and at $500^{\circ}C$ without (c/ and d/) and with (e/ and f/) chemical etching

mechanisms for this alloy. At $300^{\circ}C$, fatigue propagation area is fairly limited on 158 the specimen, the crack propagation is indeed brittle for 95% of the gauge sec-159 tion. As shown in Figure 2.a and .b, the propagation occurs in the matrix while 160 the crack path seems highly oriented by graphite nodules distribution. At 500 $^{\circ}C$, 161 fatigue propagation area is in the contrary fairly large (about 75% of the gauge sec-162 tion) but the same conclusions could be drawn. Fracture surface are indeed poorly 163 oxidised and many secondary cracks are observed around the main crack, which 164 justifies neglecting oxidation interference with damage mechanisms. 165

166

The observations reveal the important role played by the graphite nodules in 167 the crack micro-propagation inside the ferritic matrix. Moreover, crack paths are 168 enhanced and concentrated around the larger modules, as reported by several anal-169 yses in the literature [48, 41], The application of chemical etching on the observed 170 surfaces unveil a trans-granular crack propagation. Additionally, let us remark that 171 no pores or other casting defects have been observed on the studied surfaces. One 172 can therefore assume that the fatigue crack initiation occurs in the vicinity of the 173 graphite nodules and their size distribution is of prime importance in the micro-174 crack propagation, defining as such the main factors affecting the fatigue damage 175 of this particular cast-iron. 176

177 2.3. Thermal Fatigue test : validation database

Thermal fatigue tests are also performed on a peculiar specimen by Constantinescu et al. [5] to validate their model. The experimental set-up described in their article is designed to induce very severe thermomechanical loads. In this experimental protocol, axial symmetric clamped specimens are heated by the Joule effect. The heating procedure produces a thermal gradient along the specimen axis, which also induces a variation of the mechanical fields in the same direction. This experimental set-up is presented on Figure 3.

185

The set-up, the imposed thermal loads and the adjustable rig stiffness that gen-186 erates thermomechanical strains and stresses in the specimen are detailed in [5]. 187 Eight identical tests were performed on the same nodular cast-iron between 40 and 188 700 °C with a heating rate of 20 °C. s^{-1} . The maximum temperature is obtained 189 in a region of approximately 10mm in the centre of the specimen and with hold-190 ing times between 30 and 900s at maximum temperature. The observed lifetimes 191 are respectively 270, 503, 511, 551, 969, 1194, 1256 and 1388 cycles, underlying 192 an important scatter in the fatigue behaviour. These tests will be our validation 193 sample. 194

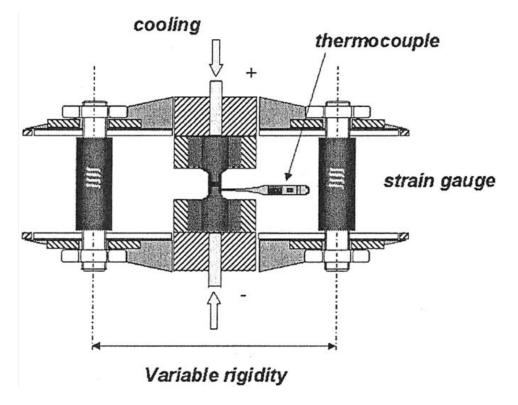


Figure 3: Thermal fatigue set-up from [5]

3. Quantitative analysis of micro-structure

¹⁹⁶ 3.1. Nodule population analysis and distribution modelling

Several samples from both fatigue specimens but also exhaust manifolds or 197 turbochargers were analysed by optical microscopy. The obtained images were 198 processed using Fiji software following a precise procedure for counting the num-199 ber and measuring the size and shape of the graphite nodules, inspired by the work 200 of [23] and [49]. 35 images of identical size were investigated for a studied surface 201 of about $110mm^2$. The studied regions correspond to those which appear as criti-202 cal regarding TMF (examples of exhaust structures and critical zones localisation 203 could be found in [1]). All analyses were performed in two dimensions, for sake 204 of simplicity for industrial parts and as it remains relevant in the global approach 205 of a criterion based on micro-propagation. 206

Nodules Measure	$ ilde{\sigma}_1$	$ ilde{\sigma}_2$	$ ilde{\mu}_1$	$ ilde{\mu}_2$	Mean size - μ m	Mean size - μ m	
					(for population 1)	(for population 2)	
Feret diameter	0.473	0.390	1.559	3.040	4.7	20.9	
\sqrt{area}	0.444	0.402	1.229	2.757	3.4	15.8	

Table 3: Statistical parameters for nodules size.

Statistical analysis of the size of graphite nodules was performed from the measurements in term of their Feret diameter and the square root of their surface, denoted next as *a* and \sqrt{area} respectively. Figure 4.a and .b show the histograms of the nodules population distributed to the classes of the two types of measures. 14059 nodules were detected with Feret diameter, *a*, lying between 1.4 and 69 μ m and the square root of their surface \sqrt{area} varying between 0.83 and 53.8 μ m.

For both dimensional measures, *a* and \sqrt{area} , one can recognise two populations of nodules: the first one consists in small nodules covering a majority of the total population while the second one is constituted of medium or large size nodules covering a minority of the total population. Similar observations have also been reported in [49] and accordingly we propose to represent the population by double log-normal distributions expressed as:

$$p_{s} = \frac{1}{s\sqrt{2\pi}\tilde{\sigma}_{1}\tilde{\sigma}_{2}} \left[\tilde{\sigma}_{2} \exp\left(-\frac{1}{2}\left(\frac{\ln(s) - \tilde{\mu}_{1}}{\tilde{\sigma}_{1}}\right)^{2}\right) + \tilde{\sigma}_{1} \exp\left(-\frac{1}{2}\left(\frac{\ln(s) - \tilde{\mu}_{2}}{\tilde{\sigma}_{2}}\right)^{2}\right) \right]$$
(1)

where *s* is the nodule size. $\tilde{\sigma}_{1,2}$ and $\tilde{\mu}_{1,2}$ denote the mean value and the standard deviation of each log-normal distribution. The optimal parameters identified from

the measured data are displayed in Table 3 and the comparison of the measured and
estimated histograms are displayed in Figure 4.c and .d. One can further remark an
excellent match between the measurement and the model.

225

The average sizes of the two populations of nodules are relatively close for both 226 measures as the Feret diameters and the \sqrt{area} with 20.9 and 4.7 μ m and 15.8 and 227 3.4 μ m respectively. There are small differences in the histograms, especially for 228 the population of small nodules which are more numerous and less dispersed for 229 their geometric size. Next, we will only consider the Feret diameter distribution for 230 the construction of the fatigue criterion which seems perfectly suited to describe 231 nodules size even if the square root of the surface \sqrt{area} is a recognised measure 232 of the default size of particles (in terms of results for the identification of fatigue 233 criteria and for all the conclusions of following sections, this choice has anyway 234 very little influence). 235

236 3.2. Shape of nodules

The particular shape of the nodules conducts to stresses and strains concentra-237 tions due to the contrast of mechanical and thermal expansion material parameters 238 between the nodules and the ferritic matrix. Nodules can therefore act as onset of 239 damage zone for the considered material. It is well known from classical elasticity 240 theory that the concentration will be inverse proportional with the curvature of the 241 shape. However it is straightforward that the exact 3D shape of the nodules cannot 242 be recovered without tomography measurements that conducts to explosive costs 243 when applied to large population on industrial parts. Therefore, a single scalar, the 244 circularity c, is further assumed as the appropriate ans statistically representative 245 parameter for representing the shapes of the population of nodules. Circularity c is 246 defined for a particle with an area of A and a perimeter p as : 247

$$c = 4\pi \frac{A}{p^2} \tag{2}$$

It reaches 0 for infinitely elongated polygon and 1 for perfect circles. Figure 248 5.a displays the distribution of nodules according to their Feret diameter and their 249 circularity. A quick analysis shows that the population of smallest nodules presents 250 a circularity essentially close to 1. This is undoubtedly due to the resolution of the 251 images used for the nodules statistical analysis: these very small elements can 252 be assimilated to areas mainly concentrated in squares of 2 pixels on each side. 253 The circularity of the small elements is therefore not necessarily relevant here. 254 However, as will be specified in the following sections, it is essentially the larger 255 nodules that are of interest in the formalism we develop for fatigue. On the other 256 hand, the largest nodules have a lower circularity, between 0.75 and 0.9. Their 257

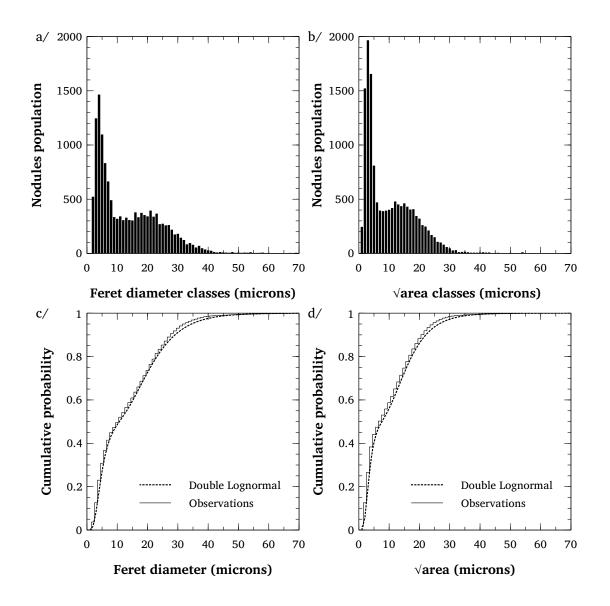


Figure 4: Histogram of nodules populations based on their Feret diameter (a/) and their \sqrt{area} (b/). Nodule size probability estimated by a double Log-normal distribution for Feret diameter (c/) and \sqrt{area} (b/)

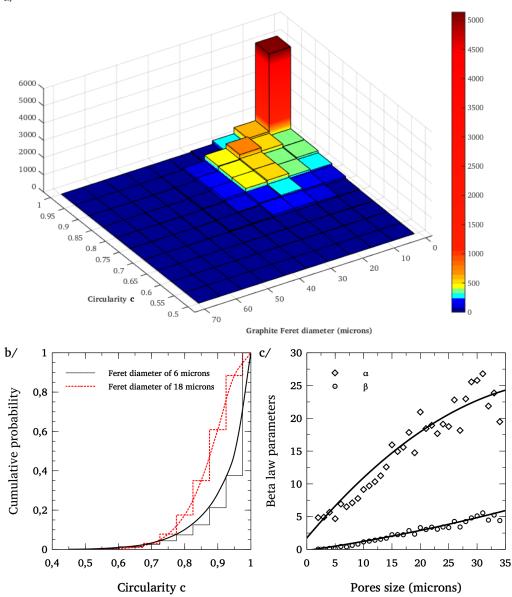


Figure 5: Graphite nodule : observed circularity distribution versus nodule size (a/) Beta distribution for different size (b/) and Beta law parameters (c/)

a/

morphology, visualised here in 2D, presents a segmented border with punctually
zones with an important curvature. This can be explained by their rapid growth
during the alloy solidification and the loss of stability of the material interfaces.
Moreover, the nodule population with a circularity smaller than 0.75 is extremely
reduced. As a consequence, in each class of nodule size, the statistical distribution
of the circularity of the nodules can easily be represented by a beta law, expressed
as :

$$p_c = \frac{\Gamma(\alpha + \beta)}{\Gamma(\alpha)\Gamma(\beta)} c^{\alpha - 1} (1 - c)^{\beta - 1}$$
(3)

with *c*, the circularity. α and β are the parameters of the distribution. The evolution of these two parameters with the nodule size in microns are presented on Figure 5.c. Their evolution can be represented by simple models, as for example the second-order polynomial estimation used here. A good match of the circularity distribution is displayed on Figure 5.b for 2 classes of nodule size.

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With the size and circularity of nodules now fully available and formalised using statistical distributions, the role of nodules in the damage process and the integration of these distributions in a fatigue criterion can now be discussed.

4. From defects to lifetime

276 4.1. Damage initiation and evolution theory

From the engineering point of view, TMF or high-temperature LCF failure of a 277 part, at the macro-scale, is usually related to the apparition of an observable crack, 278 the length of which sufficient to break a material representative volume element 279 (RVE). This failure is the resulting action of the evolution of irreversible processes 280 that occur at the material micro-scale. For nodular cast-irons, macroscopic failure 281 can be linked to the progressive damage of each of its constituent phases as under-282 lined in [28, 52]. From a meso-scale point of view, damage can occur as spheroid 283 rupture, ductile damage in the ferritic matrix or spheroid/matrix debonding. These 284 micro-mechanisms of failure are controlled by the macroscopic load imposed to a 285 RVE [53]. 286

287

From an experimental point of view and in accordance with what was presented in the section detailing the material and the LCF tests, the important role of graphite nodules in the damage cannot indeed be put aside. Several studies in the literature in the context of low cycle fatigue at room temperature have highlighted this role and observed the classic damage mechanisms of cast iron in the

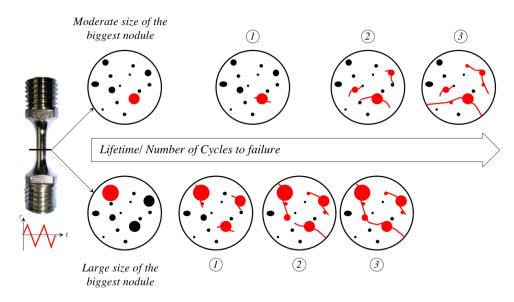
case where plasticity is generalised in the material. Thus, Harada et al.[54] then 293 Komotori et al. [55] have pointed out, on various loading regimes, that cracks oc-294 curring for ferritic cast-irons result from the coalescence of micro-voids originated 295 from debonding of the matrix/graphite interface. Later, Bubenko et al. [56] showed 296 that the low-cycle fatigue cracks always start at the interface between the graphite 297 nodule and surrounding ferrite matrix, while graphite nodules remain generally un-298 broken. Canzar et al.[57] proved that size, shape and distribution of the graphite 290 nodules has no significant influence on cyclic hardening of the material but that 300 they play a major role in the crack initiation and propagation process. They showed 301 that the larger irregularly shaped nodules tend to reduce fatigue strength. All these 302 observations therefore suggest that the damage mechanism corresponds essentially 303 to the location of plastic stresses and strains around the largest nodules which, after 304 decohesion of the matrix/nodule interface, act as pores where micro-cracks initiate. 305 306

From a numerical point of view, nodular cast irons have given rise in the litera-307 ture to numerous studies on plasticity and damage at the microstructural level, and 308 in particular on the behaviour of graphite nodules in a ductile ferritic matrix. Many 309 authors have thus carried out simulations of representative elementary volumes in 310 order to better understand the interactions between the matrix and the nodules and 311 to develop damage models based on the nucleation, growth and coalescence of cav-312 ities in a solid fraction. [58, 59, 60, 61, 62, 63]. In the simulations, considering easy 313 decohesion of graphite nodule and ductile matrix, the graphite nodule is modelled 314 either as a void or as a rigid particle in the cell model [64]. The cyclic elastoplastic 315 computation of a representative element volume with a spherical pore in its cen-316 ter discussed in [67] is thus an interesting starting point. Authors show that the 317 dissipative mechanism are maximal at the interface of the matrix with the pore. 318 Moreover the shape of the zone encompassing the maximal dissipation, which is 319 at the origin of damage are located in most of the cases around the equator of the 320 pore and provide an ideal initiation for the fatigue crack propagation. The fact 321 that the results in [67] have similar shapes for isotropic or kinematic hardening in-322 dicates that the results obtained with the particularly chosen viscoplastic material 323 behaviour can be generalised to other configurations. 324

325

Considering all these observation, a high-temperature LCF damage initiation and propagation process can be proposed. The mechanical behaviour of spheroidal graphite cast irons is indeed clearly influenced by their generally heterogeneous microstructure (composed at first order of a ductile ferritic matrix and fragile graphite spheroids). These latter naturally induce stress concentrations and quickly lead to debonding and pore generation. It then seems realistic to consider that they will be the sites of early LCF or TMF damage. The stress concentration is a function

of the size and the shape of the considered nodules ; the most significant damage 333 shall thus appear first on the largest and least spherical particles. In the rest of this 334 article, we will decide not to take into account either the influence of neighbouring 335 nodules on the mechanical fields around a considered nodule, or the existence of 336 possible aggregates of particles that could be the consequence of the solidification 337 process and generate an increase in local mechanical fields. These two elements 338 can legitimately play a role in the damage mechanisms and may be investigated 339 later. 340



1. Micro-crack initiation 2. Micro-crack propagation 3. Coalescence and failure

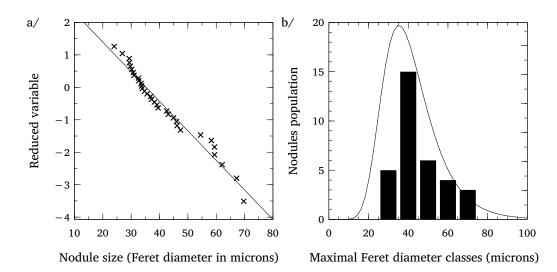
Figure 6: Simple hypothesis of damage evolution as a function of the maximum nodule size within a volume element loaded by an identical load

Based on the preceding results, we can make the simplifying assumption that 341 within a representative elementary volume containing a large family of graphite 342 nodules in cast-iron, the damage will then localise around the largest graphite nod-343 ule with the lowest circularity. The localised dissipation mechanisms will initiate 344 a micro-crack which will propagate from this nodule as illustrated on Figure 6. 345 The crack initiation will relax the local load and with increasing number of cycles 346 further micro-cracks will appear on smaller nodules. With further cycles micro-347 cracks will start to propagate until coalescence inducing the complete failure of the 348

volume element under consideration. The failure of the volume element represent
the initiation of a macroscopic crack and the onset of the structural failure.

351

We shall further infer that propagation is mainly driven by the material behaviour of the ferritic matrix. Comparable phenomena have been observed on aluminium alloys under similar loading conditions [66]. One can conclude that the lifetime of a representative volume element, i.e. number of cycle to the initiation of a macroscopic crack is dependent on its the largest graphite nodule and the properties of the matrix.



358 4.2. Extreme value statistics

Figure 7: a/ Gumbel probability plot ; b/ Gumbel identified distribution for given classes of maximal nodule size for each sample

Once the damage mechanism are assumed, a discussion has to be conducted on the type of distribution to be used to represent graphite nodules in a fatigue context. Tabibian et al.[23] introduce a simple distribution of pore size into a micro-crack propagation law for an aluminium alloy. This choice is simple and can be improved as it is strongly related to the size of the zone used to identify the distribution without taking into account of distribution tail estimation..

365

For high-cycle fatigue, the statistical theory of extreme values is very often used to represent the distribution of the largest particles or defects observed on

a plane section of fixed area [30]. In our case, considering the total distribu-368 tion of nodules would be possible but certainly too simplistic. Indeed, the over-369 representation of small nodules would undoubtedly favour the calculation of very 370 long lifetime, which is far to be conservative. Here, the statistical measurements of 371 nodules size and circularity have first been carried out on surfaces with a constant 372 area chosen to be representative both of critical areas observed on the parts to be 373 designed and to specimens failure area (size equivalent of circle with 1mm radius). 374 A sample of observations of maximum sizes of nodules for each investigated op-375 tical microscopy picture can then be obtained and serve as a basis for an extreme 376 values distribution. As explained by Beretta et al.[65], extreme value theory allows 377 us to know the asymptotic behaviour of the maximum taken by the values of identi-378 cally distributed and independent random variables. This law includes parameters 379 that can be estimated on the basis of the extreme values taken in blocks of fixed 380 size for the available data, which is the case here as detailed in section 3.1. Here, 381 the extreme value distribution arises as the possible limit distributions for the max-382 imum value of nodule size considering a given area. For the 35 pictures analysed, 383 we obtained the following sample 384

Picture	Max. size (µm)	Picture	Max. size (µm)	Picture	Max. size (µm)
1	37.2	13	44.9	25	54.5
2	30.2	14	47.4	26	39.5
3	33.9	15	38.2	27	46.0
4	42.5	16	43.1	28	30.8
5	58.3	17	35.6	29	59.4
6	69.7	18	33.7	30	67.2
7	69.8	19	39.1	31	29.7
8	69.5	20	29.3	32	59.4
9	46.2	21	29.4	33	62.1
10	32.6	22	34.7	34	37.6
11	32.7	23	26.9	35	31.2
12	31	24	24.0		

Table 4: Maximal nodule size value per investigated image.

The Gumbel plot assess the fit of data sample to the Gumbel distribution. It implies first that the maximum values to be arranged in increasing order. It induces a so-called reduced variable which is equal to $\ln(\ln(i/n + 1))$ with *n* the sample size and *i*, the *i*th smallest maximum value. This variable is plotted versus the corresponding value of nodule maximal size. A straight line indicates a good fit to the distribution which is here the case as seen on Figure 7.a. A Gumbel distribution is then identified for the sample and plotted on Figure 7.b. The probability that the biggest nodules on a critical zone has a size equal to *a* is expressed as

$$G(a) = \exp\left(-\exp\left(-\frac{a-\lambda}{\delta}\right)\right) \tag{4}$$

with λ and δ , two parameters respectively identified to 35.28 and 10.97. This 393 distribution will then be taken as the probability of finding maximum nodules ei-394 ther in the fatigue specimens or in the part critical zone[65]. Here, circularity is 395 not directly taken into account. It simply allows us to consider the largest nodules 396 as the most critical. A finer integration by considering, for example, a combination 397 of the largest nodules with the lowest circularity could be envisaged but it would 398 require a heavy additional work of fine analysis of the role of the nodules morphol-399 ogy in the fatigue process. 400

402 4.3. A probabilistic fatigue criterion

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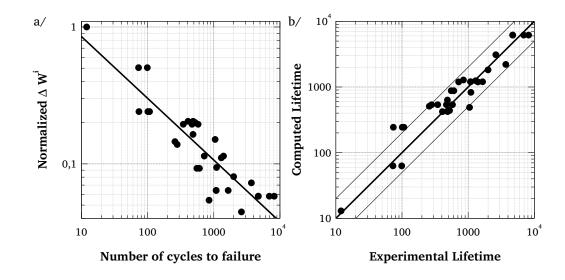


Figure 8: Deterministic fatigue criterion

The relation between viscoplastic dissipated energy over the stabilised cycle and the number of cycles to failure has proven to be a robust fatigue criterion starting with the first results by Skelton, the development including structural computational proposed in [5] and the evolution including mean stress in [16, 17]. If the viscoplastic dissipated energy over the stabilised hysteresis loop is denoted W^p and the number of cycles to failure, N_f , the criterion takes the following form:

$$W^{p}N_{f}^{b} = \int_{cycle} \underline{\underline{\sigma}} : \underline{\dot{\underline{s}}} dt N_{f}^{b} = c$$
⁽⁵⁾

where b and c denote two material parameters identified using a database of 409 isothermal LCF tests. The direct application of this fatigue criterion, proposed 410 by [5] over the present cast-iron LCF database appears to be fairly robust as it 411 revealed by the results in Figure 8. However, the scatter of the data characterised 412 by the coefficient of determination R^2 equal to 0.80, stay important and cannot be 413 precisely determined. As a partial conclusion, one can state that the energy density 414 remains a reliable marker for fatigue damage, but can neither accurately estimate 415 the variability of the fatigue response nor explain the role of the microstructure. 416 417

As an outcome, we propose to combine the description of the microstructure variability proposed Charkaluk et al.[23] which will provide the mathematical form of the criterion with the micro-crack propagation law advanced in Maurel et al.[27] for ferritic stainless steels. As matter a fact, the failure probability p_f of an elementary volume element can be described using the probability of finding a graphite nodule of maximal size a_0 within this RVE according to the formula:

$$p_f = \int_{a_0}^{a_f} \frac{da}{\left(\frac{aW^p}{\gamma_p}\right)^{m_p} + \left(\frac{aW^e}{\gamma_e}\right)^{m_e}} \tag{6}$$

$$W^{e} = \frac{1}{3} \int_{cycle} tr\left(\underline{\underline{\sigma}}\right) tr\left(\underline{d\underline{\varepsilon}^{e}}\right)$$
(7)

with $\underline{\varepsilon}^{e}$, the elastic part of the strain tensor and $\gamma_{p}, \gamma_{e}, m_{p}, m_{e}$ are four parameters 424 to be identified. a_0 is here a random variable representing the size of the largest 425 nodule and taken as a initial defect size. a_f is the final length of micro-crack, prop-426 agated from the nodule. It is associated with a critical or unacceptable crack for the 427 part or specimen under scrutiny. The size of a macroscopic crack on a laboratory 428 fatigue test specimen that induces final failure is in mm size, while on a structure 429 final failures are induced by cm or dm size crack. As already discussed in [23], it 430 seems reasonable to choose a critical size $a_f = 2$ mm. 431

432

The final form of the criterion is however relatively simple. This is a direct consequence of a series of underlying hypothesis as well as the absence of possible interaction of the different micro-cracks, i.e. the one starting at large nodules with the ones starting at smaller ones. In order to be conservative, a shorter value of *1mm* is therefore used.

438 5. Results and discussion

439 5.1. Criterion identification

Let us first remark that the number of cycles to failure N_f are considered next 440 as a random variable. Its probability distribution is directly linked with the previ-441 ously identified Gumbel distribution for maximal nodule size. Starting from this 442 observation, one can identify the criterion fatigue parameters from the LCF exper-443 imental lifetime results, by matching the probability distribution for the number of 444 cycles to failure and failure probability expressed in equation 6. The probability 445 for the number of cycles to failure is directly dependant of the loading condition 446 on a part or more easily of the load applied on a specimen. These conditions are 447 represented by the couple (W^p, W^e) that can easily be computed from the moni-448 tored strains and stresses evolution. Nf is here considered to follow a log-normal 449 distribution for the sake of simplicity: 450

$$p_{f,(W^{p},W^{e})}(N) = \frac{1}{2\pi N \sigma_{(W^{p},W^{e})}} \exp\left(\frac{(\ln(N) - \mu_{(W^{p},W^{e})})^{2}}{2\sigma_{(W^{p},W^{e})}}\right)$$
(8)

The distribution parameters $\sigma_{(W^p W^e)}$ and $\mu_{(W^p W^e)}$ can be identified from the 451 nodule size Gumbel distribution. A number of cycles to failure can indeed be com-452 puted for each class of nodule maximal size obtained from the Gumbel distribution 453 (an example of classes is represented in Figure 7). For each class, the correspond-454 ing maximal value is introduced as a_0 in equation 6, introducing chosen values for 455 γ_p, γ_e, m_p and m_e . This step leads, for a given loading condition (W^p, W^e) to a com-456 plete histogram for the number of cycles to failure and the log-normal parameters 457 are then easily identified with the computation of the mean value and the standard 458 deviation of the identified population. 459

460

It is then easy to compare the mode of this distribution (depending on the 4 parameters γ_p , γ_e , m_p and m_e of equation 6) with the actual result of the LCF test for which (W^p , W^e) were induced by the loading conditions on the specimen. A leastsquare cost function is then calculated and its minimisation allows us to obtain the optimal values for γ_p , γ_e , m_p and m_e . The results of this simple least-square optimisation are presented in Table 5.

Parameter	$\gamma_p(mJ.mm^{-2})$	$\gamma_e(mJ.mm^{-2})$	m_p	m _e
Optimized value	4.29	5.51	2.57	2.02

Table 5: Optimized criterion parameters

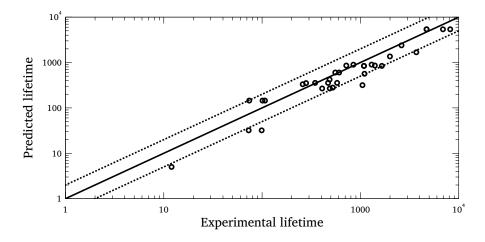


Figure 9: Experimental versus predicted lifetime with the proposed criterion

The comparison of these results with those obtained by Maurel et al.[27] on 467 stainless steel only at 300°C is interesting. The LCF tests are indeed carried out 468 under identical conditions and Maurel had chosen the grain size as the initial crack 469 size for deterministic results. The parameter m_e has been set at 2 in the case of 470 stainless steel, consistent with traditional databases, which is very close to the value 471 we obtained through optimisation. This result, for two ferritic matrices, underlines 472 the robustness of our identification process. Taking into account Maurel's formula-473 tion which introduces a correction related to grain size in mathematical formalism, 474 we find very similar values also for γ_e . The contribution of the "elastic" part for 475 two ferritic matrices therefore seems to be similar here. However, the viscoplastic 476 part is much more pronounced in the context of the considered cast-iron: indeed, 477 m_p is here higher (2.57 against 1.62), thus accentuating the viscoplastic contribu-478 tion to propagation, which also sees its attenuating factor γ_p decrease by a factor of 479 10 (once again by taking into consideration the grain size correction from Maurel 480 and his co-authors). These differences can be partly explained by the high vis-481 coplasticity of the considered cast-iron for high temperatures, much higher than 482 that of stainless steel considered by Maurel and his co-authors at 300°C as pre-483 viously shown in a comparative study[47]. This high viscoplastic activity greatly 484 reduces the observed stress levels and ultimately the W^p ones. To obtain a suffi-485 cient and significant driving force for the damage, the adjustment of the parameters 486 of the propagation law appears logical. For a given pair (W^p, W^e) , the distribution 487 mode is furthermore given by: 488

$$N_{computed} = \exp\left(\mu_{(W^i, W^e)} - \sigma_{(W^i, W^e)}^2\right)$$
(9)

This value is compared to the experimental one on Figure 9 which underlines a 489 very good matching between computed and experimental lifetime. The correlation 490 coefficient R^2 is here equal to 0.92, reaching a higher value than the deterministic 491 criterion detailed previously. The proposed method supplies therefore a lifetime 492 estimation at least as effective as the deterministic criterion. However, additional 493 information in the probability density functions of failure are an important im-494 provement as it is possible to now estimate the scattering of the fatigue strength for 495 the material. 496

497 5.2. Criterion validation

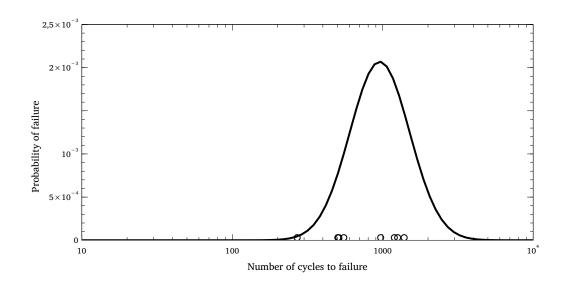


Figure 10: Probability of failure distribution for thermal-mechanical structural tests[5]. Circles represent experimental lifetimes.

The thermal fatigue tests presented in section 2.3 is here used as a validation sample for the proposed criterion. They are simulated in Abaqus using an elastoviscoplastic behaviour detailed in [68]. The clamp value due to the variable stiffness (proportional to the flexural moment of the beams) is here fixed at 183 000 N/mm and the dwell time at 700 °C at 60s. The thermal load of this experiment was numerically estimated by an electric and a thermal FEM computation on Abaqus.

The results of the thermal-electrical computation is then introduced in a mechanical 504 computation. The FE model is here constituted of 43934 nodes and 28805 C3D10 505 10-node quadratic tetrahedral elements. The numerical protocol that enables to de-506 tect the most critical finite element for thermomechanical fatigue are explained in 507 [5]. For these tests, the obtained values of W^p and W^e for the most critical element 508 are respectively 1.44 and $0.44 m J.mm^{-3}$ and allow the identification of a number of 509 cycles to failure distribution using the parameter set identified on the LCF database 510 as shown on Figure 10 where the experimental lifetimes are figured by circles. The 511 Figure represents the probability density for the number of cycles to failure and 512 the considered load. An fair good representation of the variability of the fatigue 513 response is highlighted with most of the tests lifetimes in the high probability zone 514 and a very relevant estimate of the average lifetime (about 1150 cycles versus 850 515 experimentally). Thus, despite the limited temperature range for model identifica-516 tion, satisfactory initial results in thermomechanical fatigue can be obtained, which 517 constitutes a first validation of the criterion. 518

519 6. Conclusions

This paper proposes a method for estimating the probability density distribu-520 tions of lifetime in high temperatures LCF and by extension in TMF for nodu-521 lar cast irons. It is based on the estimation of the probability density of nodule 522 sizes using optical microscope image analysis and identifying a double log-normal 523 distribution. After analysis of the fatigue test results supplemented by literature 524 analyses, nodules are identified as the preferred initiation zone. A probability den-525 sity function over the lifetime is then proposed based on an extreme value Gumbel 526 distribution for nodule maximal size and a micro-crack propagation law integrat-527 ing energy densities identified on the loading cycle. The four parameters of the 528 proposed criteria are optimised to have the most probable estimated lifetimes that 529 correspond to the experimental observations. The proposed criterion provides pre-530 dictive results for the studied cast iron, while the lifetime probability is explic-531 itly calculated and successfully estimates the results obtained with thermal fatigue 532 tests. The parameters identified for the model are also consistent with the literature 533 and physics of damage for such cast-iron. 534

535

The discussed model is based on a series of material assumptions, i.e. the type and shape of the nodule, which must be adapted each time the method is extended to a class of materials. The ability of such a model to correctly describe low cycle fatigue behaviour at higher temperatures (presence of oxidation, decarburisation) is not guaranteed and it goes without saying that it should be modified to be effective. The ability to predict lifetime in TMF should also be explored in more detail, 542 particularly for wider range of thermal load.

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