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1 **Assessment of an improved Random Flow Generation method to predict unsteady wind** 2 **pressures on an isolated building using Large-Eddy Simulation**

3
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7 8 **Abstract**

9 A wide variety of improved or modified Random Flow Generation (RFG) methods have
10 recently been proposed for simulating the Atmospheric Boundary Layer (ABL) in a Large-
11 Eddy Simulation (LES) framework. A distinct advantage of RFG methods over precursor-
12 successor methods is the significant savings that can be made in terms of setup and
13 computational costs. A review of the literature indicates that RFG methods are mostly
14 evaluated in terms of velocity statistics and spectra. However, many applications in
15 computational wind engineering ultimately aim to simulate the wind loads acting on a
16 building. It would therefore be useful to evaluate the real capabilities of RFG methods for
17 predicting unsteady wind pressures acting on various shapes of building. In this paper, one of
18 the available RFG methods based on a summation of Fourier harmonic functions is assessed
19 for various shapes of isolated building (a full-scale cube, a wind-tunnel cube, a high-rise
20 building and a low-rise building) in terms of pressure statistics and spectra. Comparisons
21 between measurements and simulations show that the RFG method tested here could be
22 particularly relevant for industrial applications, provided that the inlet turbulence intensity
23 profile is adapted.

24 25 **Keywords**

26 Large-Eddy Simulation, Random Flow Generation, wind pressure, building.

27 28 **1. Introduction**

29 The ability to predict the flow around a building is necessary in a wide variety of applications
30 in wind engineering, including pollutant dispersion in urban areas, wind resistance of
31 buildings and wall cladding, and dynamic containment of airborne pollutants. To assess the
32 effectiveness of dynamic containment when subjected to wind, as encountered in asbestos
33 removal worksites (Papadopoulos et al., 2018) and nuclear plants (Le Roux et al., 2013), it is

34 necessary to know the unsteady differential pressure between the indoor and outdoor
35 environments, together with the local wind characteristics.

36 The first step in identifying and subsequently preventing containment breaches is thus to
37 simulate the airflow around the building in question. Large-Eddy Simulation (LES) is one of
38 the most widely used tools for simulation of the Atmospheric Boundary Layer (ABL), being
39 increasingly adopted due to recent increases in computational power and the growth of
40 unsteady problems investigation.

41 The influence of the majority of simulation parameters used in computational wind
42 engineering is either negligible or can be controlled such as for ground roughness, the
43 subgrid-scale model, the domain size, the near-wall treatment, and sensitivities to mesh and
44 timestep (Blocken, 2015, Franke et al., 2010, Guichard, 2017, Tominaga et al., 2008).

45 However, turbulent inflow conditions are known to have a strong influence on numerical
46 results. Richards and Norris (2015) demonstrated that in the absence of similar turbulence
47 intensity profiles in the simulation and experiment, it was impossible to achieve accurate wind
48 pressures at the building.

49 In order to produce realistic velocity and turbulence inflow fields, various techniques based
50 on either a precursor domain or a synthetic turbulence generator have been proposed and
51 discussed in the literature. Methods based on a precursor domain, with or without recycling,
52 generally involve a higher computational cost due to the use of explicit roughness elements or
53 the iterative adjustment of simulation parameters to obtain the targeted ABL. For these
54 reasons, synthetic turbulence generators that ensure divergence-free inflow fields, like for
55 example the RFG (Kraichnan, 1970, Smirnov et al., 2001), the efficient generation method for
56 street-scale flows (Xie and Castro, 2008), the DSRFG (Huang et al, 2010), the MDSRFG
57 (Castro and Paz, 2013), the improved SEM (Poletto et al., 2013), the CDRFG (Aboshosha et
58 al., 2015) and the NSRFG (Yu et al., 2018) are preferred for practical applications. However,
59 these methods all suffer from a significant decrease in turbulence intensity between the inlet
60 and the building location as the inflow provided is not consistent with the system of equations
61 that is solved in the domain, as explained in Lamberti et al. (2018) and observed by Vasaturo
62 et al. (2018). Although it is then impossible to achieve streamwise homogeneity, Vasaturo et
63 al. (2018) show that the Gaussian velocity spectra produced by the RFG at the inlet rapidly
64 evolve along the domain towards the same shape as those obtained with recycling and Vortex
65 methods, but with a much lower power spectral density. Moreover, previous studies have
66 shown that if the issue of the decrease in turbulence intensity along the computational domain

67 can be overcome, then realistic wind pressure coefficients can be achieved (Yan and Li,
68 2015).

69 The objective of this paper is to evaluate an efficient way for obtaining targeted velocity and
70 turbulence intensity profiles at the building location. Computational and experimental wind
71 pressure coefficients are compared for different shapes of isolated building. The approach was
72 inspired by wind tunnel tests that aim to obtain targeted profiles at the future building location
73 and not necessarily to achieve full streamwise homogeneity, as shown by the partial covering
74 of the wind-tunnel ground by roughness elements. In the framework of practical applications
75 in wind engineering, the RFG method of Smirnov et al. (2001) has been retained here as it
76 does not require any additional developments to the widely used ANSYS Fluent code, and
77 because the code verification has already been done. This is a criterion of choice for engineers
78 as it means that the method can be applied immediately, though it must first be clearly
79 validated for each field of industrial applications. The RFG method, available under the name
80 Spectral Synthesizer, produces a continuous flow field as a superposition of one hundred
81 Fourier harmonic functions. It involves scaling and orthogonal transformation operations to
82 guarantee a flow field that is divergence-free for homogeneous turbulence and nearly
83 divergence-free for inhomogeneous turbulence.

84 The methodology, simulation setup, and reference experiments used to fulfil the objective of
85 this paper are described in the next sections and are followed by a discussion of the results.

86

87 **2. Methodology**

88 The procedure that was eventually selected to apply the RFG to computational wind
89 engineering cases in this paper is close to that denoted “improved RFG” in Yan and Li (2015).
90 It consists of three steps, as illustrated in Figure 1:

91 **Step 1** – Generation of the targeted inflow (from experimental or theoretical data) using RFG
92 in an empty domain,

93 **Step 2** – Explicit adjustment of inlet boundary conditions using Eq. (1) to achieve targeted
94 profiles at the future building location,

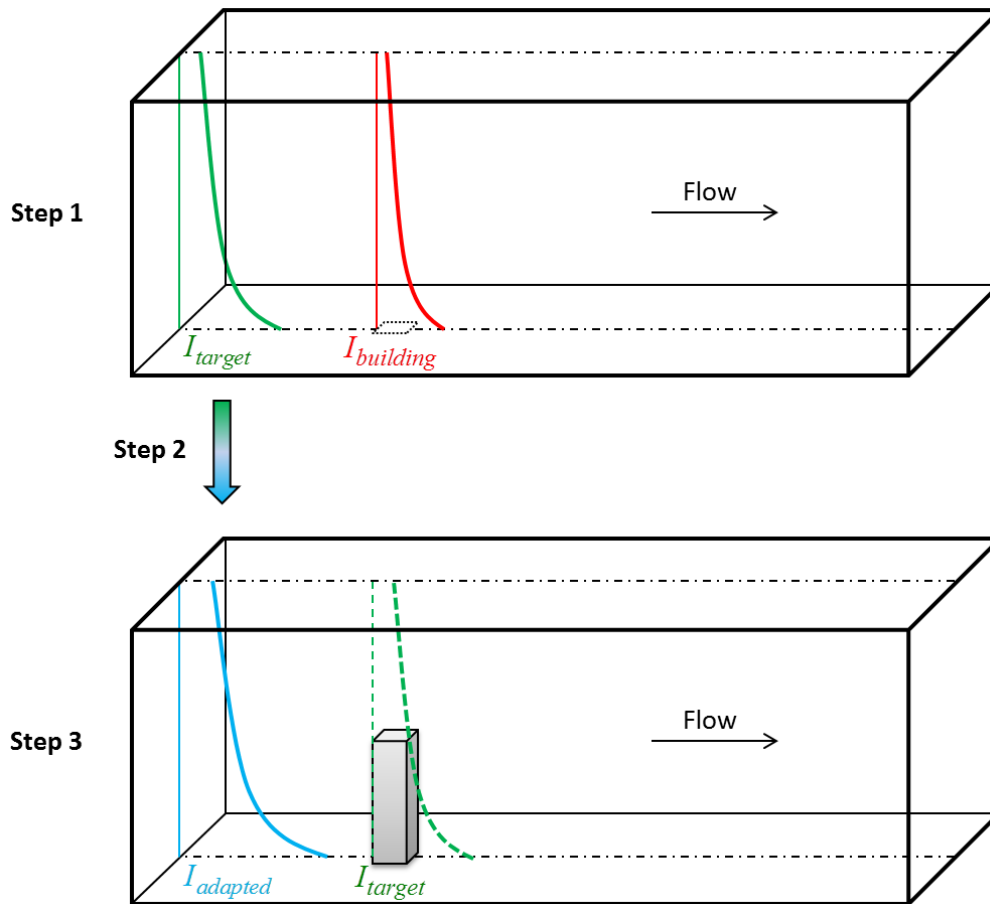
95 **Step 3** – Running the final computation using RFG in the domain with the building.

96 The procedure differs from the “improved RFG” method of Yan and Li (2015) in that only
97 one adjustment is required and the method does not require successive trial-and-error
98 iterations to obtain the targeted profile. It is assumed here that the mean velocity profile is
99 homogeneous, that the length scale remains unchanged and that the turbulence intensity
100 profile is adapted according to:

101
$$I_{adapted}(z) = I_{target}(z) + Dx \left(\frac{I_{target}(z) - I_{building}(z)}{Dx} \right) = 2I_{target}(z) - I_{building}(z), \quad (1)$$

102 where z is the vertical coordinate, Dx is the streamwise distance between the inlet and the
 103 building, $I_{building}$ is the turbulence intensity first obtained at the future building location in
 104 the empty domain, I_{target} is the targeted turbulence intensity and $I_{adapted}$ is the turbulence
 105 intensity to be used in the final computation. When verification is required, it is still possible
 106 to ensure that I_{target} is accurate at the building location in an empty domain. In this paper, the
 107 verification is presented in the Results and Discussion sections.

108



109

110

111 **Figure 1** Strategy applied to obtain the targeted turbulence intensity profile at the building
 112 location using the RFG method

113

114

115

116 **3. Simulation setup**

117 The simulation setup is as far as possible kept unchanged or parameterized. Only the domain
118 size, which is a function of building height, and the inlet profiles are updated for each study
119 case.

120

121 *3.1 Computational domain and grid*

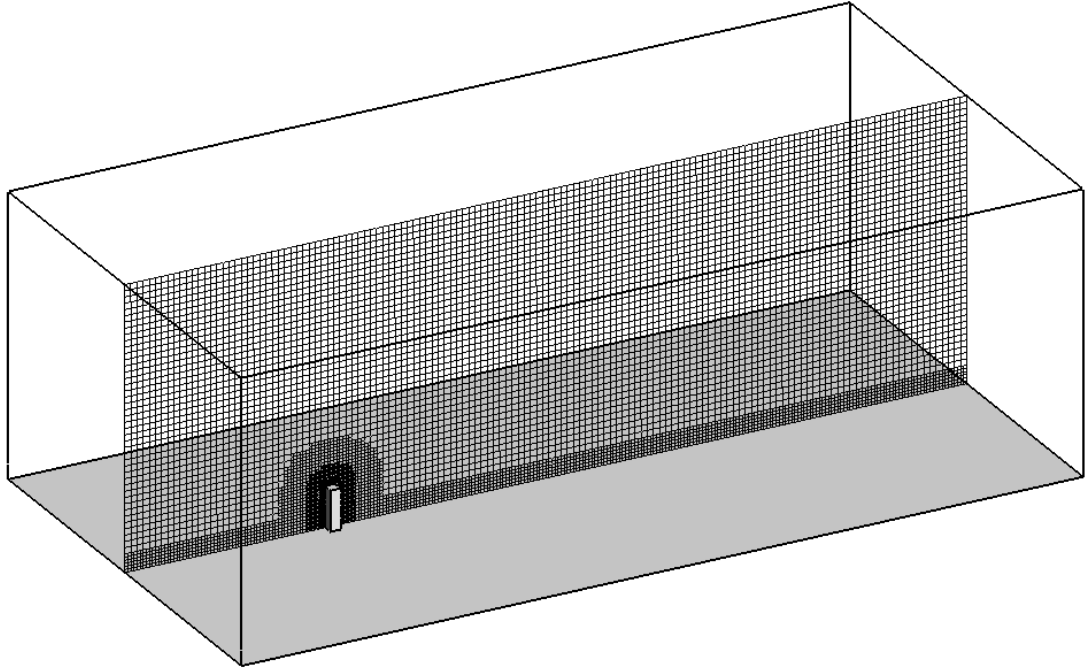
122 The computational domain is built by strictly following the guidelines of Franke et al. (2010),
123 hence the inlet, lateral and upper boundaries are 5H away from the building and the outlet
124 boundary is 15H behind the building (where H is the height of the building).

125 Use of the cut-cell meshing technique, as verified in Iousef et al. (2017) in a close framework,
126 allows a very good mesh quality to be obtained, and it is also easy to manage. This technique
127 also limits the number of parameters needed to describe the mesh if one wants to exactly
128 reproduce the same mesh. A grid sensitivity study has led to application of the following cell
129 sizes:

- 130 • at building faces, the length of the shortest building edge divided by 15,
- 131 • at the ground: twice the cell size set at building faces,
- 132 • elsewhere: twice the cell size set at the ground.

133 In addition, an inflation growth factor of 1.1 is applied, which allows the number of uniform
134 layers before switching from one cell size to another to be controlled. These meshing settings
135 result in a rather coarse mesh of one million cells and an average y^+ of 300. An example of
136 such a mesh for a high-rise building is provided in Figure 2.

137



138

139

140 **Figure 2** Example of a mesh in a center cut plane, generated with the cut-cell technique for a
 141 high-rise building

142

143 *3.2 Boundary conditions*

144 Symmetry conditions are applied to upper and lateral boundaries. This implies zero normal
 145 velocity and zero gradients for all variables. The inlet is a velocity-inlet and the outlet is a
 146 pressure-outlet boundary condition. Both ground and building are smooth walls as various
 147 sensitivity studies have shown that ground roughness has only a weak influence on the
 148 performance on the performance of inflow methods (Köse and Dick, 2010; Vasaturo et al.,
 149 2018). The upstream ground roughness is taken into account through the aerodynamic
 150 roughness length in the inlet profiles.

151 The target profiles are best fits of the available experimental data, based on the wind profile
 152 log law (Oke, 1987):

153
$$U(z) = \frac{u_\tau}{\kappa} \log\left(\frac{z+z_0}{z_0}\right), \quad (2)$$

154 where U is the mean air velocity, $\kappa=0.4$ is the von Kàrmàn constant, z_0 is the aerodynamic
 155 roughness length, and u_τ is the friction velocity defined by

156
$$u_\tau = \kappa \frac{U_H}{\log\left(\frac{H+z_0}{z_0}\right)}, \quad (3)$$

157 where H is the building reference height.

158 Depending on the experiments considered, a better fit can be achieved using the wind profile
159 power law of Counihan (1975):

$$160 \quad U(z) = U_H \left(\frac{z}{H} \right)^\alpha, \quad (4)$$

161 where α is the stability and roughness exponent.

162 As the turbulence intensity rather than the turbulence kinetic energy is often provided in
163 experimental data, the turbulence kinetic energy profile can be computed as:

$$164 \quad k(z) = \frac{3}{2} (U(z)I(z))^2, \quad (5)$$

165 where I is the measured turbulence intensity, which is least-squares fitted using the
166 following expression:

$$167 \quad I(z) = I_u^\infty + (I_u^g - I_u^\infty) \exp(-az), \quad (6)$$

168 where I_u^∞ is the bulk turbulence intensity parameter, I_u^g is the ground turbulence intensity
169 parameter and a is the curvature parameter.

170 Finally, given that the turbulence length scale is not provided in reference experiments, the
171 turbulence dissipation rate is set as:

$$172 \quad \varepsilon(z) = \frac{u_\tau^3}{\kappa(z + z_0)}. \quad (7)$$

173 The quality of the representation of available experimental data by Eq. (2) to (7) will be
174 verified in the Results and Discussion sections below.

175

176 *3.3 Numerical settings*

177 The CFD solver ANSYS Fluent 17.2 was used for all simulations. As the mesh is too coarse
178 to resolve the laminar sub-layer, a wall function must be used. A law-of-the-wall was
179 therefore applied to compute the wall shear stresses with the assumption that the centroid of
180 the wall adjacent cell was in the logarithmic region of the boundary layer. This approach is
181 called Standard Near-Wall Treatment. Because LES is based on the filtering of Navier-Stokes
182 equations, a Sub-Grid Scale model must be used for that part of the turbulence spectrum that
183 is not resolved. The Wall-Adapting Local Eddy-Viscosity (WALE) formulation proposed by
184 Nicoud and Ducros (1999) was used here. In contrast with constant and dynamic
185 Smagorinsky-Lilly models, a spatial operator is added to the eddy viscosity formulation to
186 provide the expected wall asymptotic behavior.

187 Regarding the spatial discretization, a second-order scheme was chosen for pressure and a
188 bounded-central-differencing scheme was chosen for momentum.
189 For the temporal discretization, a Non-Iterative Time Advancement (NITA) was set, along
190 with a fractional-step pressure-velocity coupling. The corresponding algorithms are detailed
191 in the ANSYS Fluent Theory Guide (2016). Fixed timesteps were deduced from RANS k- ϵ
192 Realizable simulations to strictly respect the CFL condition in each cell. The LES
193 initialization run was first conducted over a period of 6 flow-through times, and flow statistics
194 were then computed over 30 flow-through times.

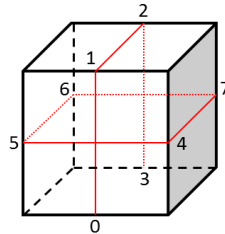
195

196 **4. Reference experiments**

197 *4.1 The Silsoe cube full-scale experiment*

198 The Silsoe cube experiment is one of the most well-documented full-scale investigations of
199 wind velocities and pressures on a building found in the literature (Richards and Hoxey,
200 2012). It is crucial to include full-scale data for validation as reduced-scale models are not
201 always able to reproduce every characteristic of real wind (Richards et al., 2007; Irtaza et al.,
202 2013). Full-scale data therefore represents the experimental data that are the closest to final
203 wind engineering applications. The building considered here is a 6 m cube, instrumented with
204 pressure taps along vertical and horizontal centerlines, as shown in Figure 3.

205



206

207 **Figure 3** Centerlines along which pressure taps are distributed in the Silsoe cube experiment
208 (the front face is windward)

209

210 The flat ground on which the cube is mounted is composed of grass, which leads to an
211 aerodynamic roughness length z_0 of 0.01 m. The mean velocity is $6 \text{ m}\cdot\text{s}^{-1}$ and the turbulence
212 intensity is around 20 % at the height of the building. For this reference experiment, wind
213 mean velocity and turbulence intensity profiles are provided, in addition to the mean, standard
214 deviation, and minimum and maximum pressure coefficients. These parameters allow the

215 quality of both the inflow conditions and the pressure statistics acting on a building to be
 216 evaluated. The pressure coefficients C_p are defined by:

$$217 \quad C_{p,\text{mean}} = \frac{p_{\text{mean}}}{q_{\text{mean}}}, \quad C_{p,\text{min}} = \frac{p_{\text{min}}}{q_{\text{max}}}, \quad C_{p,\text{max}} = \frac{p_{\text{max}}}{q_{\text{max}}}, \quad C_{p,\text{std}} = \frac{p_{\text{std}}}{q_{\text{std}}}, \quad (8)$$

218 where p is the static pressure on the building, q is the reference dynamic pressure and the
 219 subscripts ‘mean’, ‘min’, ‘max’ and ‘std’ are the mean, minimum, maximum and standard
 220 deviation values, respectively.

221

222 *4.2 Tokyo Polytechnic University (TPU) wind-tunnel experiments*

223 An assessment of a proposed approach is not complete if only one study case is considered.
 224 The Tokyo Polytechnic University (TPU) aerodynamic database was therefore exploited in
 225 addition to the Silsoe cube experiment.

226 A particular advantage of the extensive TPU database is that it provides time series of wind
 227 pressure coefficients. Of the various wind-tunnel experiments available in the database, three
 228 cases were retained here:

229 - a reduced-scale (1:400) cube of 0.1 m, to consolidate the full-scale case described in the
 230 previous section with a different approaching flow and a different scale. In this case the flow
 231 exponent was 0.25, the mean velocity was 7 m.s⁻¹ and the turbulence intensity was 23 % at
 232 building height. Tests were run over 32.768 s with a frequency of 1000 Hz.

233 - a reduced-scale (1:400) high-rise building with dimensions of 0.1 m (width) x 0.2 m (length)
 234 x 0.4 m (height), as this type of tall building is often highly exposed to wind. The flow
 235 exponent was 0.25, the mean velocity was 11 m.s⁻¹ and the turbulence intensity was 12 % at
 236 building height. Tests were run over 32.768 s with a frequency of 1000 Hz.

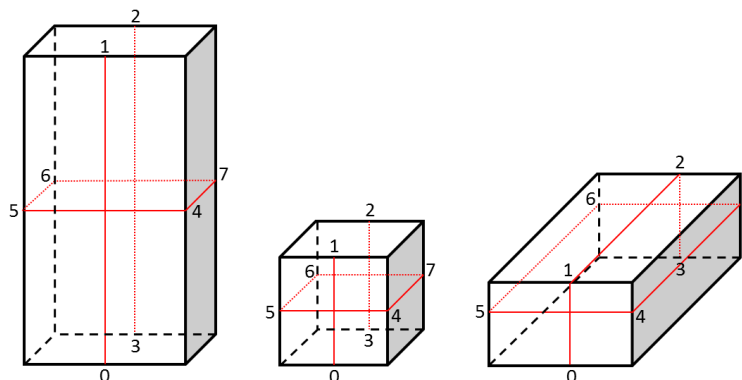
237 - a reduced-scale (1:400) low-rise building with dimensions of 0.16 m (width) x 0.4 m
 238 (length) x 0.08 m (height) in order to evaluate a common shape of industrial building. The
 239 flow exponent was 0.25, the mean velocity was 7 m.s⁻¹ and the turbulence intensity was 24 %
 240 at building height. Tests were run over 18 s with a frequency of 500 Hz.

241 It is important to note that in the TPU database, the pressure coefficients have a different
 242 definition to those of the Silsoe experiment:

$$243 \quad C_{p,\text{mean}} = \frac{p_{\text{mean}}}{q_{\text{mean}}}, \quad C_{p,\text{min}} = \frac{p_{\text{min}}}{q_{\text{mean}}}, \quad C_{p,\text{max}} = \frac{p_{\text{max}}}{q_{\text{mean}}}, \quad C_{p,\text{std}} = \frac{p_{\text{std}}}{q_{\text{mean}}}. \quad (9)$$

244 For the sake of clarity, it is impossible to plot all experimental and numerical pressure
 245 coefficients for each face of each building. The results will therefore be compared in two

246 rings, as illustrated in Figure 4, along which the pressure taps were placed in the wind-tunnel
 247 experiments.



248
 249 **Figure 4** Lines along which pressure coefficients were processed in TPU experiments (front
 250 faces are windward)
 251

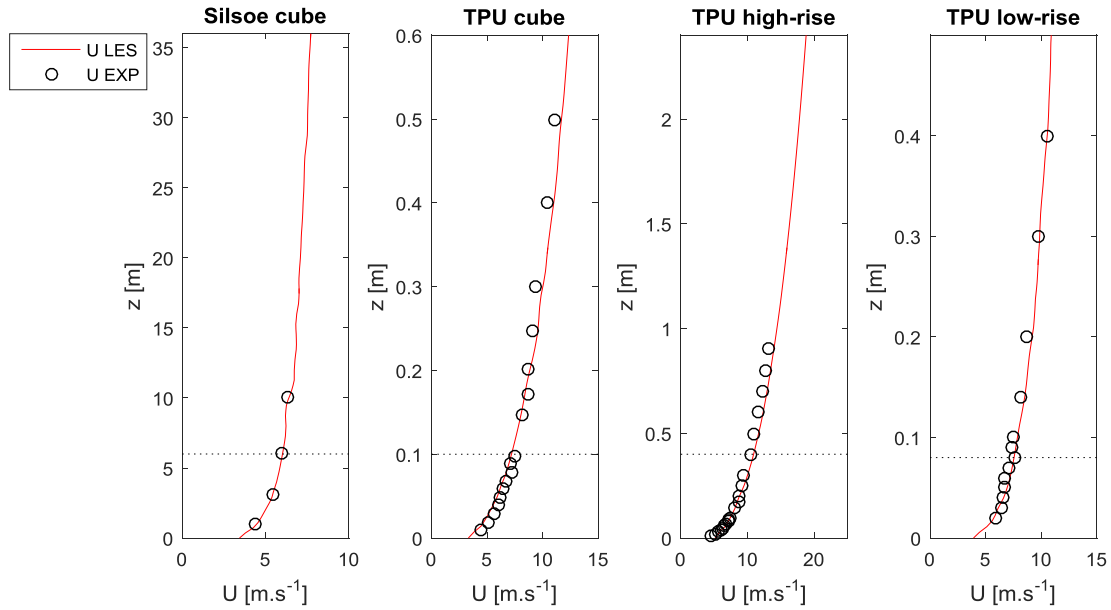
252 5. Results and discussion

253 5.1 Verification of targeted mean velocity and turbulence intensity

254 The first step in the evaluation is to verify that the simulated profiles at the building location
 255 are consistent with the measured profiles when the strategy described in Figure 1 is applied
 256 using RFG as an inlet velocity fluctuations synthesizer. For each case considered, the targeted
 257 mean velocity profiles are reproduced in Figure 5 and the turbulence intensity profiles are
 258 presented in Figure 6. The indices u , v , w denote the streamwise, spanwise and vertical
 259 directions, respectively. Note that only the streamwise component of the turbulence intensity
 260 is provided for the TPU cases.

261 Figures 5 and 6 clearly show that the RFG method can produce velocity and turbulence
 262 intensity profiles that are very close to the targeted experimental data, after adjustment of the
 263 turbulence intensity. When the three components of turbulence intensity are available, the
 264 tested RFG is even able to reproduce the expected relationships between them (Counihan,
 265 1975), namely $I_v = 0.75 I_u$ and $I_w = 0.5 I_u$ from the ground to building height. These simulated
 266 profiles are also much more accurate than those obtained in the wind-tunnel modeling of the
 267 Silsoe cube in Auckland (Richards et al., 2007). Figure 6 also shows that vertical profiles of
 268 turbulence intensity differ significantly from one wind-tunnel test to another, the turbulence
 269 intensity in Figure 6 varying from 25% at the ground to 2% at a height of $6H$ for the high-rise
 270 building, but ranging from 25% to 20% for the low-rise building, being almost flat.

271



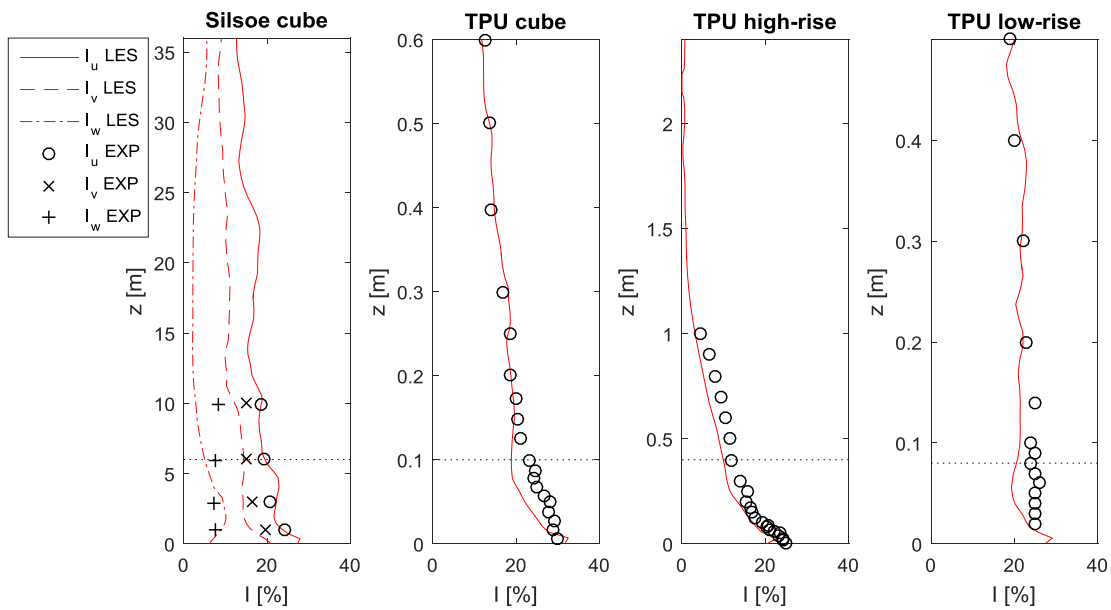
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273

Figure 5 Comparison of simulated and measured vertical profiles of mean velocity at the building location

274

275



276

277

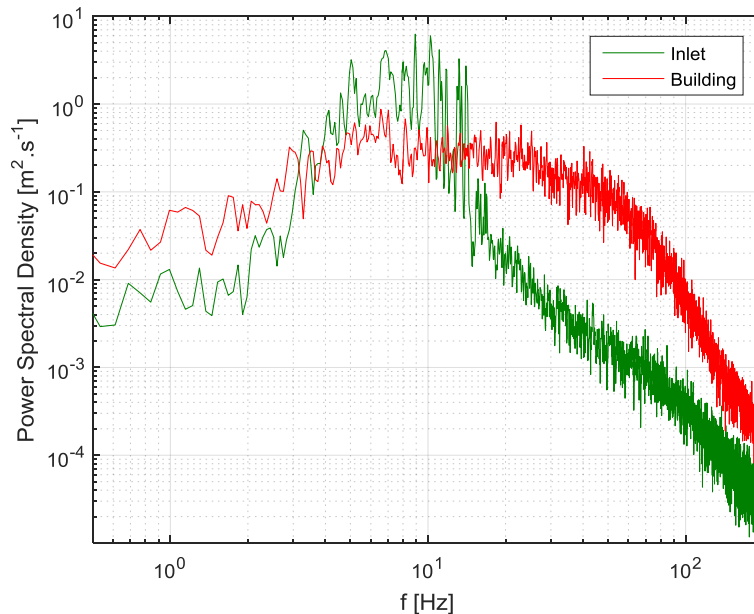
Figure 6 Comparison of simulated and measured vertical profiles of turbulence intensity at the building location

278

279

280 To further evaluate the flow characteristics at the building location, the streamwise velocity
 281 spectrum at building height is plotted in Figure 7, superimposed with the Gaussian spectrum
 282 produced by the RFG method at the inlet. The results highlight that the inlet spectrum evolves
 283 towards a more representative spectrum when approaching the building location, as

284 previously reported by Vasaturo et al. (2018). This means that the flow obtained at the
285 building location might eventually be closer to the experimental flow than expected.



286
287 **Figure 7** Velocity spectra at the building height in the TPU cube case, as simulated at the
288 inlet and building locations

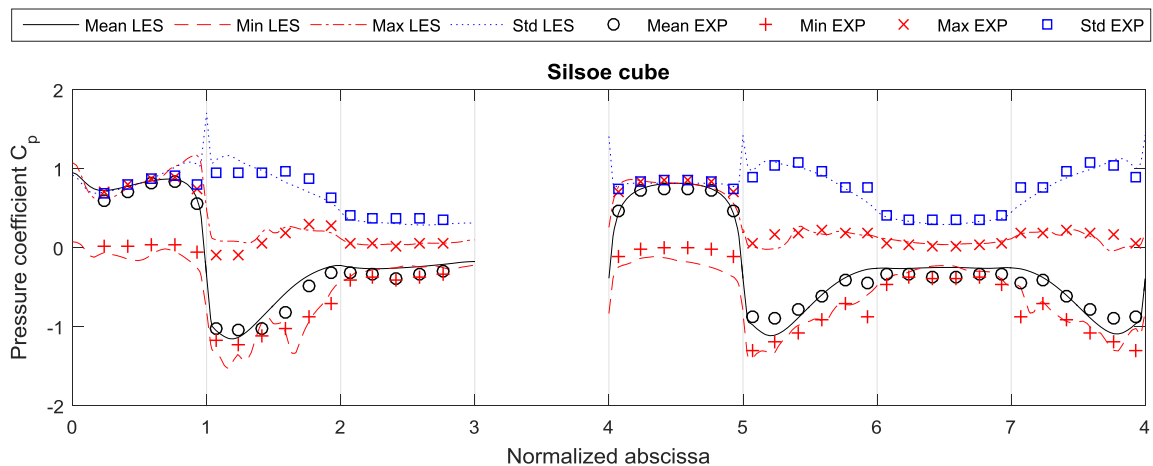
289
290 *5.2 Pressure coefficients*

291 The mean, minimum, maximum and standard deviation of pressure coefficients are presented
292 in Figure 8. Results obtained by Large-Eddy Simulation are plotted using lines and are
293 denoted by LES in the legend. Experimental data are plotted with symbols and denoted by
294 EXP. The normalized abscissa corresponds to the numbers depicted in Figures 3 and 4.
295 Overall, the computed pressure coefficients are in good agreement with experimental data for
296 the buildings considered. The major deviation is observed for peak values, which are known
297 to be more difficult to accurately predict and will be discussed below. For the Silsoe cube case
298 however, the simulations provide a better prediction of standard deviation and peak pressure
299 coefficients than others in the literature (Richards and Norris, 2015). This is mainly due to the
300 fact that the flow generated at the building location in the present paper is much closer to the
301 full-scale experiment than in previous studies. One of the implications of this for current
302 industrial applications is that the best results can be obtained if the focus is more on flow
303 characteristics at the building location than on the streamwise flow homogeneity, though such
304 a choice of emphasis would not be relevant in the framework of fundamental research.
305 Another observation is that the discrepancy between measured and computed pressure
306 coefficients is particularly high at specific locations in the TPU wind-tunnel cases. At the

307 same time, the symmetry of pressure coefficients at building sides (abscissas from 5 to 6 and
 308 from 7 to 4) is not systematic. One of the measurement points for the high-rise building is
 309 even out of range, the reason for which will be examined in more detail in the next section.
 310 This suggests that the uncertainty associated with the peak values for the wind-tunnel TPU
 311 measurements might be very high. Unfortunately, the experimental uncertainties are not
 312 provided in the TPU database and cannot be estimated without extra information. It is
 313 important to note that such behavior is not observed in the Silsoe full scale experiment, for
 314 which the expected symmetry is visible in all measurements.
 315 Very good agreement between simulated and measured pressure coefficients, for both mean
 316 and standard deviation, were obtained in a recent study for a low-rise building with eaves
 317 (Ricci et al., 2017). However, the method employed involved explicit roughness elements and
 318 implementation of the MDSRFG synthesizer, which, while very relevant for benchmarking,
 319 would not be appropriate for industrial applications due to the long setup and computational
 320 times, as discussed below.

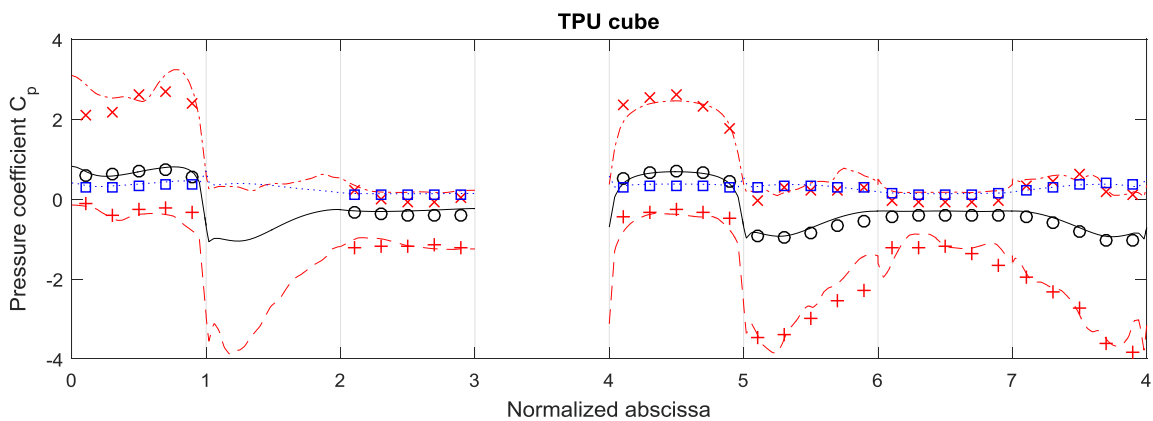
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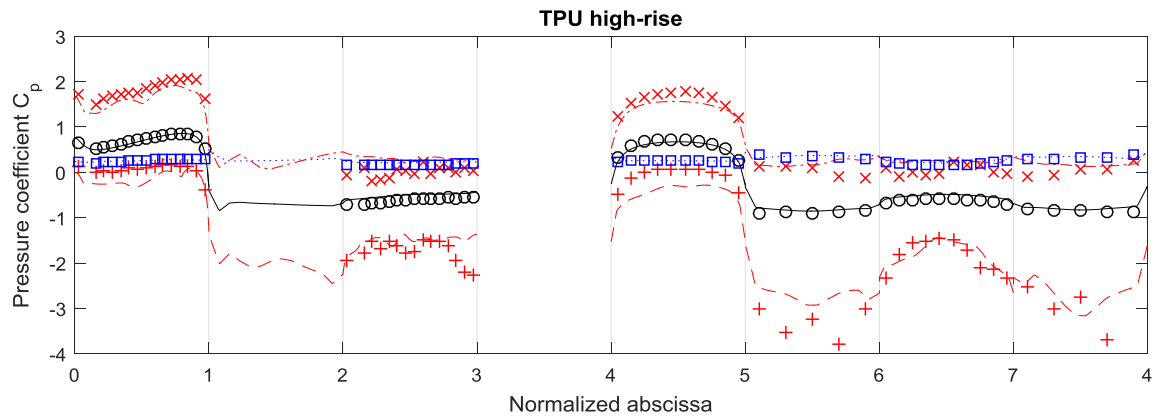
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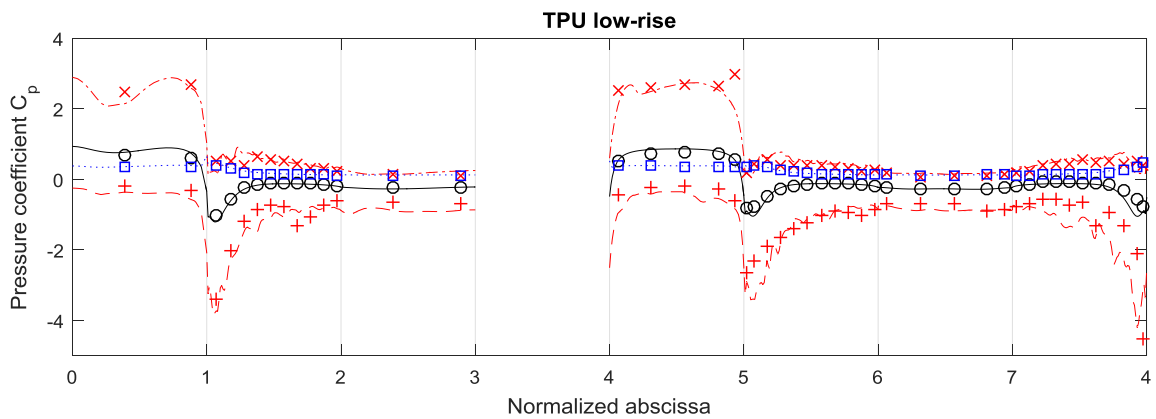
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328



329

330 **Figure 8** Simulation (Lines, LES) and experimental (Symbols, EXP) pressure coefficients for
 331 various shapes of isolated building

332

333 *5.3 Pressure time series and spectra*

334 In order to identify why the statistics of simulated pressure coefficients differ from
 335 experimental data, pressure time series were plotted for two particular locations (Figure 9).

336 These time series are only available for the TPU cases.

337 The first location (point A) is on the high-rise building, where a vortex is produced from
 338 interactions between the airflow and the obstacle. This leads to the highest fluctuations in
 339 wind pressure, as seen in Figure 8 near abscissas 5 and 4. This is also the location where
 340 discrepancies between simulation and experiment were greatest.

341 An equivalent location was also retained on the cube (point B) for comparison, as a good
 342 agreement between simulation and experiment was obtained here, despite remaining the
 343 location where pressure fluctuations were highest.

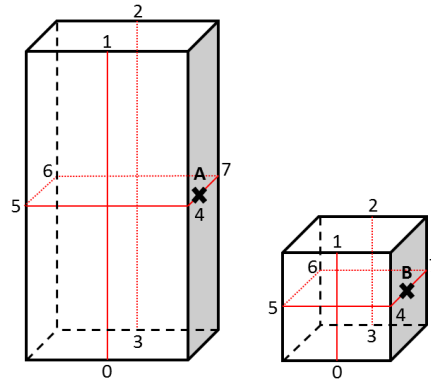
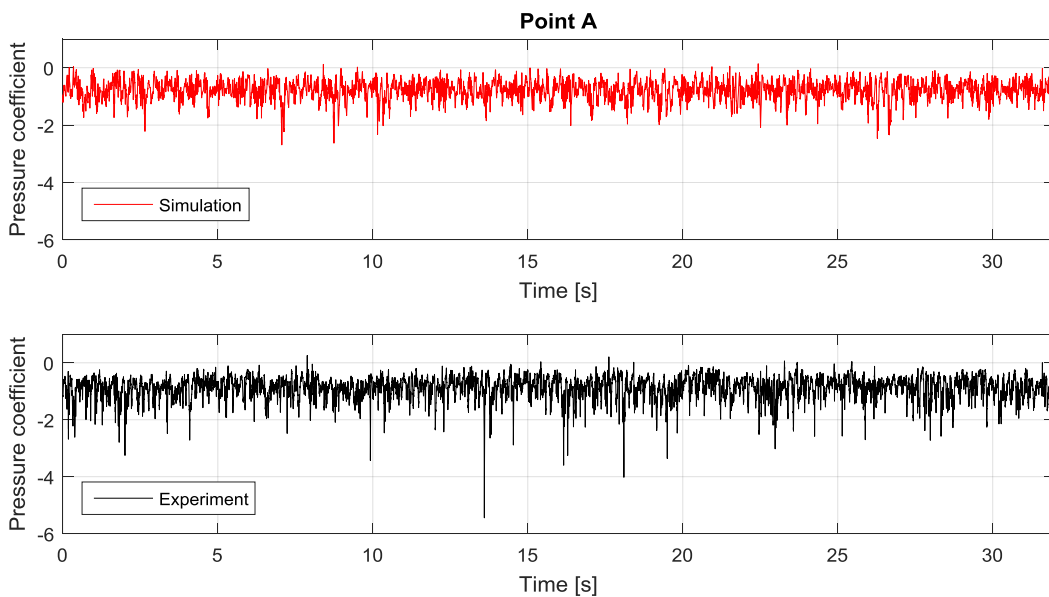
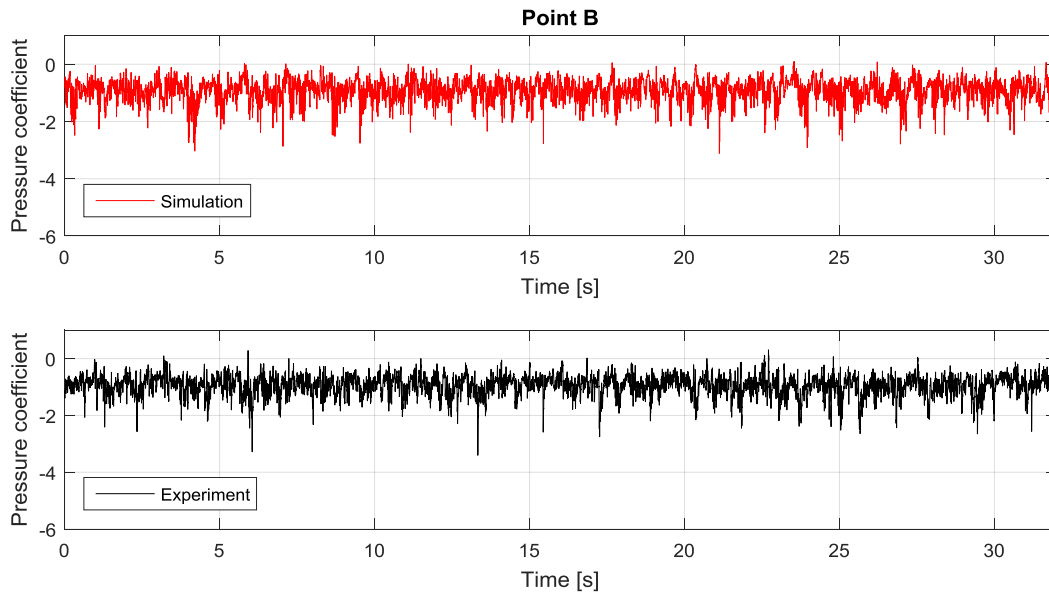


Figure 9 Locations of points A and B used for monitoring pressure coefficients

In Figure 10, time series of pressure coefficients from simulations and experiments are plotted for points A and B over a 32.768 s period. It is clear that the mean, maximum and standard deviation of the pressure are well predicted by the simulation. However, the minimum values differ due to a single negative peak in the experiment at point A. This anomalous peak is likely an experimental artifact since it occurs only once over the period of measurements. Thus, if another run had been carried out, different results might have been obtained. This would support our earlier suggestion that high uncertainties might be associated with the TPU wind-tunnel measurement peak values. Such behavior is not observed at point B, where peak values are reached at least twice over the measurement period, as they are in simulations, making these results more reliable.





359
 360 **Figure 10** Measured and calculated time series of pressure coefficients at two selected
 361 locations, A and B

362
 363 As experimental data and simulation results have different sampling frequencies, a
 364 quantitative comparison can be made by superimposing pressure power spectra, as shown in
 365 Figure 11 for points A and B from 0.1 to 500 Hz. This comparison reinforces the observation
 366 that the magnitude of pressure is well represented by Large-Eddy Simulation at lower
 367 frequencies. However, the comparison also highlights abnormal representations of some high
 368 frequencies in the wind-tunnel test at point A, which occur at 228 Hz and 456 Hz, with lower
 369 peaks at 114 Hz and 342 Hz. As these frequencies are exact multiples of 114 Hz, they appear
 370 to be related to a measurement issue rather than being a real characteristic of the wind
 371 turbulence. Once again, no singular frequency is observed in Figure 11 for the experimental
 372 pressure power spectrum at point B.

373 Finally, the particular locations where the deviations between experiment and simulation were
 374 highest appear to correspond to those locations where the experimental data raised questions
 375 about the consistency of minimum values.

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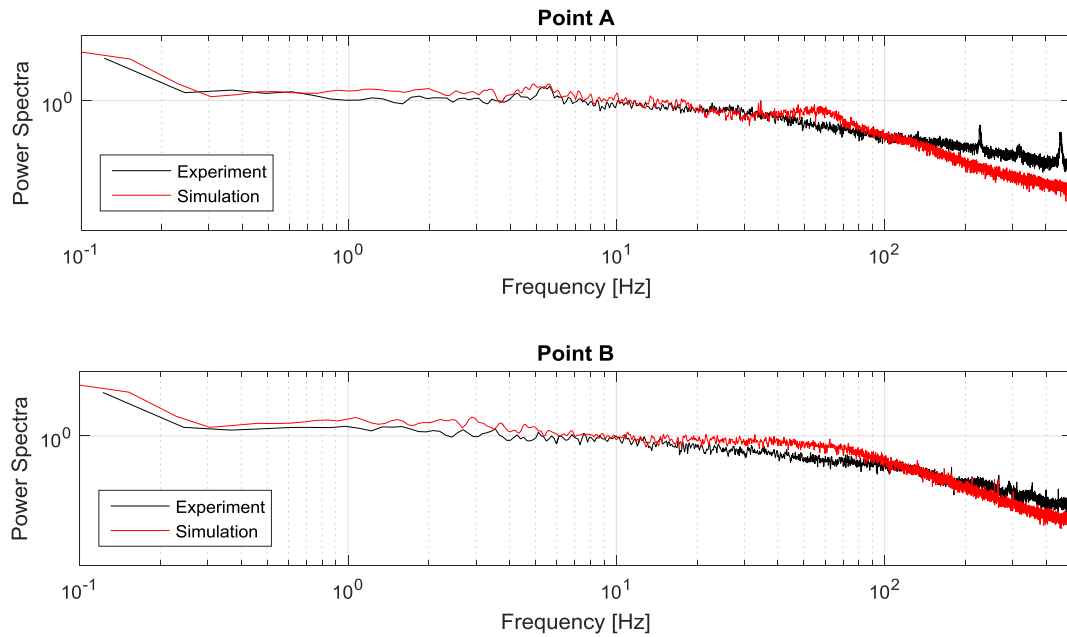


Figure 11 Measured and computed power spectra of pressure coefficients at two selected locations A and B

5.4 Computational cost

The performance of the proposed approach is now assessed for various cases in terms of accuracy, by comparison with reference full scale and wind-tunnel experiments. A major advantage of the method proposed here is claimed to be the significant reduction in both setup and computational times compared to existing approaches.

The setup time is minimized by the use of the automatic cut-cell meshing method (about one million cells) and the Spectral Synthesizer RFG method, a method that has already been implemented and verified in commercial codes.

The computational wall-clock time of one simulation was consistently under 10 hours, covering both initialization and statistical runs, which corresponds to about 1200 CPU hours on the INRS cluster using 128 CPUs (8 nodes, 2 eight-cores Intel Xeon@1.7 GHz and 64 GB RAM per node). This represents 4 times fewer CPU hours than the Silsoe case of Richards and Norris (2015), 20 times fewer CPU hours than the low-rise building case of Ricci et al. (2017), and 4 times fewer CPU hours than the high-rise building case of Yu et al. (2018), without taking into account the fact that the frequencies of the cores used at INRS are always lower than cores used in other studies.

401 **6. Conclusions**

402 Most recent LES studies aimed at predicting unsteady wind pressures on a building have
403 sought to obtain streamwise flow homogeneity, which is particularly relevant in terms of
404 fundamental research. However, it is often difficult to directly apply these recent findings to
405 wind engineering problems because of the high computational costs involved and the time-
406 consuming implementation requirements. The strategy proposed here consists of putting aside
407 the quest for homogeneity and focusing instead on finding a more efficient way to attain
408 targeted flow conditions at the building location.

409 The method proposed is based on the use of an existing and standard RFG method to produce
410 a divergence-free velocity field at the domain inlet. The inlet turbulence intensity profile must
411 be adapted once using an explicit equation in order to obtain the desired local wind
412 characteristics. The approach was inspired by wind-tunnel experiments in which the flow is
413 first verified without the building and the building is then added to run the final tests.

414 The main findings of our assessment in the context of wind engineering and industrial
415 applications are as follows:

- 416 1. Even though the RFG method produces a Gaussian spectrum at the inlet, it naturally
417 evolves along the domain towards a more realistic spectrum.
- 418 2. When the characteristics of the simulated wind are representative of the real flow at the
419 building location, it is possible to obtain an accurate prediction of unsteady wind pressures
420 acting on an isolated building.
- 421 3. For wind engineering applications, it seems more cost-effective to aim at achieving realistic
422 wind characteristics at the building location rather than streamwise homogeneity throughout
423 the whole simulation domain.
- 424 4. Compared to existing methods (recycling, precursor-successor, iterative methods and other
425 synthetic turbulence generators), the proposed approach involves a lower computational cost
426 and needs no additional developments under commercial codes, making it possible for
427 engineers to apply it immediately.

428 Next steps will consist of extending the evaluation of the approach to more complex shapes of
429 buildings and to non-isolated buildings, and then using the results to design industrial
430 ventilation systems that are subjected to wind.

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