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# Very weak carbonaceous asteroid simulants I: mechanical properties and response to hypervelocity impacts

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## Abstract

The two on-going sample return space missions, Hayabusa2 and OSIRIS-REx are going to return to Earth asteroid regolith from the carbonaceous near-Earth asteroids Ryugu and Bennu. The two main processes that lead to regolith production are the micrometeorite bombardment and the thermal cracking. Here we report the production of a weak simulant material, analogue to carbonaceous meteorites with a CM-like composition, following the preliminary compositional results for Bennu and Ryugu. This asteroid simulant has compressive and flexural strength  $1.8 \pm 0.17$  and  $0.7 \pm 0.07$  MPa, respectively. The thermal conductivity (in air) of the simulant at room temperature is between 0.43 and 0.47 W m<sup>-1</sup> K<sup>-1</sup>. In order to distinguish the type of regolith that is produced by each of these processes, we present and discuss the results of the experimental campaign focused on laboratory hypervelocity impacts, using the 2-stage light-gas gun of the University of Kent, that mimic the micrometeorite bombardment. We find that this process produces both monomineralic and multiminerallitic fragments, resulting in a difficulty to distinguish the two processes, at least on these weak materials.

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## 1. Introduction

“Extremely weak, extremely rare on Earth, but probably very common in space”. In the light of the results of the two on-going sample return space missions, JAXA’s Hayabusa2 (Kitazato et al., 2019) and NASA’s OSIRIS-REx (Lauretta et al., 2019), the above statement appears to be an appropriate definition for the type of materials that constitute most of the boulders on the surface of the near-Earth asteroids (162173) Ryugu (Grott et al., 2019; Watanabe et al., 2019; Sugita et al., 2019) and (101955) Bennu (Dellagiustina et al., 2019; Walsh et al., 2019).

10 In the case of Ryugu, information about the weak strength of the boulders was essentially derived from measurements, obtained by the MARA radiometer (Grott et al., 2017) of the thermal conductivity  $\kappa$  of the solid boulder material on which MASCOT, the Hayabusa2 lander (Ho et al., 2017), came to rest. In addition, observations of the entire surface of Ryugu in the thermal infrared 15 indicate that the global value of Ryugu thermal inertia ( $\Gamma$ ), a parameter that measures the resistance of a surface to temperature changes (Delbo et al., 2015, and references therein), lies between 200 and 500 J m<sup>-2</sup> K<sup>-1</sup> s<sup>-1/2</sup> (Sugita et al., 2019) and is thus very similar to the value measured for the boulder observed by MARA ( $\Gamma = 282^{+93}_{-35}$  J m<sup>-2</sup> K<sup>-1</sup> s<sup>-1/2</sup>; Grott et al., 2019). Since the surface 20 is essentially covered by boulders (Sugita et al., 2019), this gives an indication that the typical thermal inertia of Ryugu’s rocks is similar to that measured by Grott et al. (2019).  $\Gamma$  is defined as the square root of the product of thermal conductivity  $\kappa$ , density, and heat capacity of a material, and thus knowledge of the value of  $\Gamma$  allows the estimation of  $\kappa$ . Under reasonable assumptions of the 25 material density and heat capacity, the value of  $\kappa$  for Ryugu’s boulder was found to be lower than expected, namely several factors lower than those measured in the laboratory for meteorites (Opeil et al., 2010, 2012). Specifically,  $\kappa$  was

calculated to be between  $0.06$  and  $0.16 \text{ W m}^{-1} \text{ K}^{-1}$  (at  $230 \text{ K}$ ), whilst meteorites range between  $0.45$ – $5.5 \text{ W m}^{-1} \text{ K}^{-1}$  at  $200 \text{ K}$  (Opeil et al., 2012, and references therein). The lower thermal conductivity of Ryugu’s boulders compared to those of meteorites can be explained as an effect of the higher porosity of Ryugu’s boulder material compared to the meteorites (Grott et al., 2019). Indeed, it should be noted that our sample of asteroid meteoritic material is likely biased, as we could expect the weakest meteoroids to be unable of surviving passage through Earth’s atmosphere.

Initial ground-based spectroscopic observations of Ryugu indicated a CM or CI-type composition (Moskovitz et al., 2013; Perna et al., 2017; Le Corre et al., 2018). Later remote spectral observations of Ryugu from Hayabusa2 confirmed the initial ground-based results (Kitazato et al., 2019), however it was not possible to derive a definite meteoritic analogue. First of all there are no measured meteorite spectra that match perfectly the measured visible and near-infrared spectra of Ryugu in the wavelength range between  $0.45$  and  $3 \mu\text{m}$ . The closest match is produced by thermally-metamorphosed CI chondrites and shocked CM chondrites (Kitazato et al., 2019). Additionally, close inspection of Ryugu’s boulder images revealed mm-size inclusions (Grott et al., 2019). CM and CM2 chondrites, according to our current meteoritic collection have smaller inclusions (e.g. chondrules of  $0.27$ – $0.3 \text{ mm}$ , Burbine, 2016), making a link between Ryugu’s boulder and said meteorites possibly problematic. On the other hand, CI meteorites have less than  $1\%$  calcium aluminium-rich inclusions and chondrules (Burbine, 2016).

Using Hertzian heat conduction theory, MASCOT team (Grott et al., 2019) used the value of  $\kappa$  and the estimated boulder porosity to evaluate the mechanical tensile strength of the material, which was calculated to be between  $0.20$  and  $0.28 \text{ MPa}$ . This is weaker than typical meteoritic material (Ostrowski & Bryson, 2019, and references therein), including the most fragile CM and CI carbonaceous meteorites: for instance, the compressive strength of the CM2 Murchison and Sutter’s Mill meteorites are around  $50 \text{ MPa}$  (Miura et al., 2008) and  $85 \text{ MPa}$  (Jenniskens et al., 2012) respectively. Following common assump-

tion that compressive strength is approximately 10 times the tensile strength,  
60 one can estimate that the tensile strength of the aforementioned meteorites, is  
at best, a few MPa. Observations of the ungrouped C2 Tagish Lake fireball,  
(Brown et al., 2002) estimated that the pre-atmospheric compressive strength  
was about 0.25 MPa, while its material compressive strength is 0.7 MPa, the  
latter being more appropriate for the recovered meteorites. Later laboratory  
65 experiments on samples of Tagish Lake give a tensile strength of  $0.8 \pm 0.3$  MPa  
(Tsuchiyama et al., 2009). The combination of these two independent mea-  
surements, makes Tagish Lake a peculiar case, as its compressive and tensile  
strength have very similar values. Furthermore, the CI Ivuna and Orgueil have  
tensile strengths of  $0.7 \pm 0.2$  and  $2.8 \pm 1.9$  MPa respectively, as measured by  
70 Tsuchiyama et al. (2009), while the CM2 Murray have a value of  $8.8 \pm 4.8$  MPa,  
according to measurements from the same authors.

OSIRIS-REx provided similar, albeit less constraining, observations com-  
pared to the ones reported by Grott et al. (2019), but still favouring the inter-  
pretation that places Bennu’s materials within the weakest known meteorites. In  
75 particular, thermal infrared data allowed Dellagiustina et al. (2019) to determine  
the thermal inertia of Bennu to be  $350 \pm 20 \text{ J m}^{-2} \text{ s}^{-0.5} \text{ K}^{-1}$ , corresponding to  
thermal conductivities very similar to those derived by Grott et al. (2019) for  
Ryugu. However, these  $\kappa$  values for Bennu, were determined from the thermal  
signal of the entire visible and illuminated surface of Bennu and not just a single  
80 boulder. Nevertheless, boulders constitute a large fraction of Bennu’s surface  
(Dellagiustina et al., 2019; Walsh et al., 2019), implying that they could have  
low  $\kappa$ , large porosity, and therefore low mechanical strength.

For Bennu, initial ground-based spectroscopic observations classified the  
body as a B-type asteroid with a composition more similar to the CM1 me-  
85 teorites (Clark et al., 2011). OSIRIS-REx observations confirmed the blue slope  
that was previously detected and additionally showed an absorption at around  
 $2.74 \mu\text{m}$  (Hamilton et al., 2019), interpreted with the presence of hydrated  
materials. The exact position of this feature gives more information on the  
composition, where in this case is close to CM2.1–CM2.2 meteorites (Hamilton

90 et al., 2019). Unlike Ryugu, Bennu appears strongly hydrated and less heated, consistent with their estimated orbital evolutions (Michel & Delbo, 2010; Delbo & Michel, 2011).

Concerning the abundance of these materials, it has been estimated that Ryugu and Bennu are very likely originated from the re-accumulation of debris  
95 of larger parent asteroids fragmented by impacts with other asteroids in the region of the Main Belt that is closer to the Sun ( $2.1 < a < 2.5$  au) and comprises asteroids with low orbital inclination  $i$ , e.g.  $i < \sim 7^\circ$  (Bottke et al., 2015; de León et al., 2016; Campins et al., 2010; Campins et al., 2013). This area is populated by a large amount of asteroids with spectral and albedo properties similar to  
100 those of Ryugu and Bennu (Delbo et al., 2017; de León et al., 2016; Walsh et al., 2013) implying that their materials could be common in the Main Belt. On the other hand, because of their fragile nature, these materials are extremely rare on Earth as they filtered by the atmosphere: meteoroids carrying fragile materials fragment in the high atmosphere, resulting in the lack of meteorites that could  
105 represent Ryugu's and Bennu's materials.

Even though Ryugu and Bennu have landscapes covered by boulders in the meter-size range, areas with small particles in the size ranges of some cm are reported from the images returned by Hayabusa2 (during the operations approaching the surface) and OSIRIS-REx (Lauretta et al., 2019). Both Hayabusa2 and  
110 OSIRIS-REx will return samples of their regoliths to Earth for detailed laboratory analysis. In particular, the OSIRIS-REx is due to return a minimum of 60 gr of regolith, offering a unique opportunity to analyse a large amount of low-albedo asteroid material (Lauretta et al., 2015). This sample will allow to study the bulk properties of an asteroidal regolith. One of the central aims  
115 of the research related to these missions is to understand how the regolith of these asteroids formed, evolved, and how it is related with the interiors of these asteroids.

As on our Moon, it was thought that the regolith on asteroids originates from the impact of micrometeorites hitting the surface with speeds of several km/s  
120 and breaking up surface rocks into smaller pieces, a process called comminution

(Hoerz et al., 1975; Horz & Cintala, 1997; Basilevsky et al., 2015). However, recent studies have shown from experimental and modelling approaches that regolith could also be produced by another physical processes, the so-called thermal cracking (Delbo et al., 2014; Hazeli et al., 2018; Molaro & Byrne, 2012; 125 Molaro et al., 2017). This is a mechanical stress, which produces damage and eventually causes the failure of the surface rocks, due to the cycles of heating and cooling when the asteroid surface transitions from day to night. These predictions appear validated by images obtained during the preliminary survey of the asteroid Bennu by OSIRIS-REx, which found evidence of boulders broken 130 in place and desegregating boulders in regolith (Lauretta et al., 2019; Walsh et al., 2019), and by other astronomical observations of asteroids (Graves et al., 2019).

Additional laboratory experiments were used to inform thermomechanical and crack propagation models, which indicated that thermal cracks propagate 135 preferentially around different mineral phases (Hazeli et al., 2018), leading to the prediction that the mode of failure of non-homogenous rocks, such as chondrite meteorites would produce monominerallic regolith grains. Indeed monominerallic grains are the major components of the regolith sample returned by the Hayabusa mission from the asteroid (25143) Itokawa (Nakamura et al., 2011).

140 Conversely, the break up of rocks by the much more violent (compared to slow thermal cracking) impacts of micrometeorites should produce debris with different mineralogies in the same grains. This is because the crack propagation velocity is much faster compared to that of thermal cracking. Pioneering experiments by Durda & Flynn (1999) showed mineralogical size segregation 145 from the preferential fracture along mineral grain boundaries in hypervelocity impact experiments into olivine porphyritic basalt. Similar chemical and mineralogical segregation between matrix and chondrule minerals was seen in impact disruption of ordinary chondrite and anhydrous carbonaceous (CV Allende) meteorite specimens (Flynn & Durda, 2004). Flynn et al. (2009) also performed a 150 single hypervelocity impact experiment on hydrous meteorite this time, specifically on Murchison, where they measured the mass distribution of the produced

fragments. Michikami et al. (2007) performed a series of impact experiments where they studied the effect of target porosity and strength on the impact outcome (crater formation and ejecta speeds). However, a comparison between the debris produced by different mechanisms, namely impact experiments and thermal cycling, performed on the very same type of weak natural materials such as those inferred to be present on these low-albedo carbonaceous-like asteroids, have never been performed so far.

Here we perform an initial study of the effects of hypervelocity impacts on these types of materials. In this work we investigate hydrated and mechanically weak asteroid boulder simulants constructed in the laboratory to have composition similar to the CM2 carbonaceous chondrites, but mechanical strength and porosity intermediate between those that have been measured for meteorites and those inferred for low-albedo and spectrally featureless asteroids, including Bennu and Ryugu. In section 2 we describe the mineralogy and preparation of the simulants. In sections 3 and 4 we describe their mechanical and thermal characterisation. In section 5 we explain the set up to carry out the impact experiment, along with their results. In section 6 we interpret our results in comparison to the observations obtained by Hayabusa2 and OSIRIS-REx of Ryugu and Bennu.

## 2. Preparation of simulant materials

Our specimens are based on the UCF/DSI CM Carbonaceous Chondrite Simulant (UCF/DSI-CM-2) that is part of series of asteroid regolith simulants developed in collaboration with the University of Central Florida (UCF) and Deep Space Industries (DSI) and is now produced by UCF's Exolith Lab (Britt et al., 2018; Boivin et al., 2018; Metzger et al., 2019; Britt et al., 2019). In particular, we use the CM2 regolith simulant that is a close mineralogical match to the Murchison CM2 carbonaceous chondrite meteorite and its mineralogy was based on the mineralogical analysis of Murchison by Bland et al. (2004) and Howard et al. (2009).

However, there are the following differences compared to the meteorite: the organic component of volatile rich carbonaceous chondrites contains polycyclic aromatic hydrocarbons (PAHs), some of which are known carcinogens and mutagens. Our research suggests that rough chemical fidelity can be maintained by substituting much safer sub-bituminous coal as our organic analogue. A second safety consideration is the choice of serpentine group materials. While most serpentine polymorphs are quite safe, we avoid fibrous chrysotile because of its asbestos content. In the same vein, pyrite was substituted for troilite because of troilite’s tendency toward combustion and/or explosion in small particle sizes (Yang et al., 2011).

The simulant was produced by first sourcing the individual mineral components, which are then comminuted down to approximately 70  $\mu\text{m}$  in size. This size can be larger than the original CM meteorites with grain sizes ranging from sub-micron to micron (Britt et al., 2019), and could potentially affect the thermal conductivity of the material. The constituent minerals are then mixed together with water and sodium metasilicate, which acts as a binder. Approximately one ml of deionized water was added for every 4 grams of minerals. The resulting mixture is then cured at a high temperature ( $\sim 80^\circ\text{C}$ ) to dry and remove the water, then mechanically ground in a rock crusher to achieve the desired particle size distribution. The initial grain density of the simulant is about 2,750  $\text{kg m}^{-3}$ , which is in the range of the measured grain densities of the CMs, 2,570–2,870  $\text{kg m}^{-3}$  (Britt & Consolmagno, 2003). Next, we created regolith with simulated chondrite inclusions by adding glass spherules 600–800  $\mu\text{m}$  in size and 2,500  $\text{kg m}^{-3}$  in density so that the resulting mixture contained 15% chondrules-analogues and 85% minerals by volume, resulting to a final grain density of  $\rho_{\text{grain}}=2,710 \text{ kg m}^{-3}$ . Mineral constituents, glass spherules, water, and sodium metasilicate were combined, as described above, and then packed into a rectangular cast with approximate dimensions of  $95 \times 95 \times 44 \text{ mm}$  in order to simulate the blocks present of the surface of Bennu and Ryugu. The simulant was cast in and then cured at a high-temperature to remove moisture. Using a laboratory balance with precision  $\pm 0.1 \text{ gr}$ , we measured the mass of

Table 1: Mineralogy and bulk chemistry of CM Carbonaceous Chondrite Simulant (UCF/DSI-CM-2).

Mineral	Wt. %	Oxide	Wt. %
Mg-serpentine	72.5	SiO <sub>2</sub>	32.5
Magnetite	10.4	TiO <sub>2</sub>	0.3
Olivine	7.8	Al <sub>2</sub> O <sub>3</sub>	3.1
Sub-bituminous coal	3.6	Cr <sub>2</sub> O <sub>3</sub>	0.2
Pyrite	2.6	FeO	20.2
Pyroxene	2.1	MgO	32.1
Siderite	1.0	CaO	3.1
		Na <sub>2</sub> O	6.2
		K <sub>2</sub> O	0.2
		P <sub>2</sub> O <sub>5</sub>	0.4
		SO <sub>3</sub>	1.5
Total	100.0	Total	100.0

the target samples before the impacts, while their accurate dimensions (x,y,z) were measured using callipers. The ratio between mass and volume resulted in a density of the targets of  $\rho_{\text{bulk}}=1,980 \text{ kg m}^{-3}$ , very similar to the average of  
215 CMs ( $2,200 \text{ kg m}^{-3}$  from Britt et al., 2002). The porosity,  $\phi = 1 - \frac{\rho_{\text{bulk}}}{\rho_{\text{grain}}}$ , of the simulant blocks is calculated to be 26%.

We initially performed Thermogravimetry/Evolved Gas Analysis (TG/EGA) in order to measure the mass loss by volatile sublimation and its composition as a function of the temperature of the specimens. We found that total volatile  
220 release of 11.2% for the CM simulant, with the majority of the volatile content being the H<sub>2</sub>O from the hydrated minerals. The mineralogy and bulk chemistry of our CM2 asteroid regolith simulant is reported in Tab. 1. The specimens were then shipped from Florida to France for further analysis.

### 3. Mechanical properties of simulant materials

225 Next, we performed experiments to measure the mechanical properties of the  
simulants. Simple mechanical compression tests were carried out to determine:  
(1) the simple compression elastic modulus,  $E_c$ , and (2) the maximum compressive  
stress,  $\sigma_c$  of our asteroid simulants. In addition, three-points bending tests  
were also carried out to determine (3) the flexural strength,  $\sigma_f$ , which can be  
230 related to the tensile strength (but see caveats below). In total, 12 flexural tests  
and 6 compression tests were performed (see Fig. 1).

For the aforementioned mechanical tests, the samples were prepared by cutting  
the simulants into parallelepipeds of roughly  $20 \times 20 \times 50$  mm for the  
compression tests and  $10 \times 25 \times 75$  mm for bending tests. Parallelepipeds of  
235  $10 \times 20 \times 40$  mm were also cut out of the main samples for additional bending  
tests. After the cutting, the samples were oven-dried at 378 K for at least 24  
hours in dry air in order to remove the humidity from the Earth's atmosphere  
that have been absorbed by the simulants. We note that this heating process is  
too weak to drive off water present within the hydrated minerals as it has been  
240 shown that heating significantly above 373 K is necessary to alter the 2.7  $\mu\text{m}$  hy-  
dration spectral band of CM2 Murchison meteorite (Hiroi et al., 1996). However  
this preparation improved the repeatability of the mechanical tests by limiting  
the influence of ambient humidity.

For brittle materials (such as the tested simulants), and for rocks in general,  
245 the compressive strength is significantly larger than tensile strength. Therefore,  
in a bending test, failure is triggered at the location of maximum tensile stress,  
namely at the vertical of the indenter on the opposite (lower) surface. At this  
location the stress is composed of a vertical component and a shear component.  
However, the shear stress is significantly lower than the tensile stress; this can  
250 be demonstrated analytically (Timoshenko & Goodier, 1982). For all the sam-  
ple dimensions used in this study, we neglected the shear stress as it affects  
the failure stress only by a few percents in relative value, that is lower than  
experimental discrepancy between repeated measurements (see Tab. 2). Once

we have neglected the shear stress, the point where the crack appears is in pure  
255 tension and therefore the flexural stress can be assimilated to the tensile stress.  
However, in a general way, the presence of defects in the material tends to cause  
the failure strength to be lower in a tensile test than in a bending test. The  
reason for this lies in the fact that in the bending test, only a small portion  
of the sample is loaded at the maximum tensile stress, whereas in the tensile  
260 test the stress is approximately homogeneous, so that the whole volume of the  
sample is submitted to the same stress. A larger volume implies more defects  
susceptible of triggering failure, thus reducing the tensile strength compared to  
flexural strength. As a conclusion, the flexural strength can be considered as an  
upper bound for the tensile strength.

265 On the other hand, even for the samples with a higher aspect ratio, the  
flexural modulus from the flexural tests was significantly different from that  
obtained from the simple compression ones, indicating that it is a combination  
of the compressive and flexural moduli. Therefore, no conclusion could be drawn  
concerning the tensile elastic modulus.

270 In both bending and simple compression tests, the displacement  $u$  of the  
press arm and the force  $F$  applied were recorded together. The samples were  
loaded at  $1 \text{ mm min}^{-1}$ . The compression stress was defined as  $F/S$  ( $S$  being the  
sample's cross-sectional area), the bending stress as  $F\ell/4wt^2$ , where  $\ell$ ,  $w$  and  $t$   
are the support span, width and thickness of the sample. The compression strain  
275 was defined as  $u/L_0$ , where  $L_0$  is the initial sample length, and the bending  
strain as  $6tu/\ell^2$ .

In Fig. 2 we present the average stress-strain curves for flexural and com-  
pression tests. The maximum of the curves give the compressive  $\sigma_c$  and flexural  
 $\sigma_f$  strengths of the simulant. As described before, we approximate the tensile  
280 strength of the material to be equal to the flexural strength for this particular  
experiments. For the compressive stress,  $E_c$  is the slope of the linear part of  
the stress-strain curve. Samples were cut in two perpendicular directions to as-  
sess the anisotropy of the simulant material. However, no significant differences  
were observed in the resulting measurements of the values for  $\sigma_f$ ,  $\sigma_c$ , and  $E_c$ .

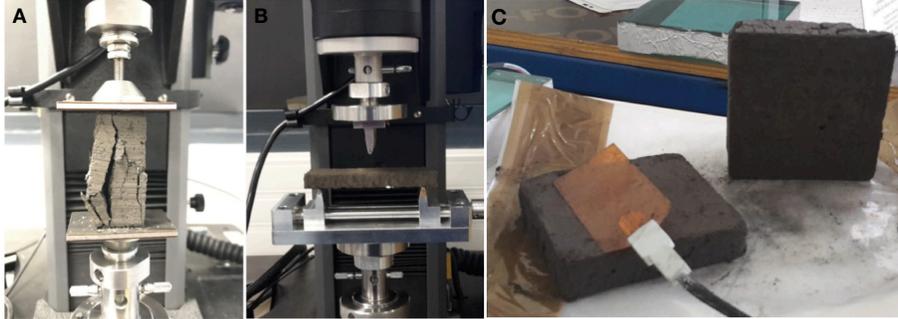


Figure 1: Experimental devices for simple compression (A), 3-point bending test (B) and measurements of the thermal conductivity (C) .

Table 2: Mechanical properties of CM Carbonaceous Chondrite Simulant (UCF/DSI-CM-2).

	$E_c$ (MPa)	$\sigma_c$ (MPa)	$\sigma_f$ (MPa)
mean value	151.67	1.82	0.72
90% conf. interval	17.48	0.17	0.07

285 Therefore, no further mention of anisotropy will be made, and we processed all the results of the mechanical tests assuming the samples to be equivalent with respect to anisotropy of the mechanical properties.

Due to the weak and brittle nature of the simulant material, it was difficult to obtain perfectly parallel surfaces for the samples. Broken corners, holes,  
 290 bad surface roughness and poor parallelism between the sample's faces affect the repeatability of the test. This inaccuracy, to which should be added the inhomogeneities of the material itself, induced an uncertainty in the measurement. Therefore, it was assumed that the measured quantities  $\sigma_f$ ,  $\sigma_c$  and  $E_c$  followed a normal distribution and accordingly, a 90% confidence interval was  
 295 calculated. The results of the mechanical tests are summarised in Tab. 2. The smaller samples ( $10 \times 20 \times 40$  mm), were used for the bending tests in order to study if the results are significantly affected by aspect ratio of the samples. We find that results are consistent regardless the dimensions of the samples.

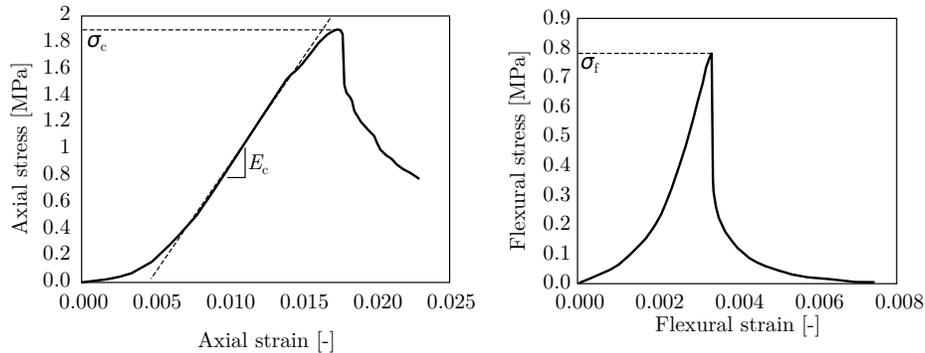


Figure 2: Typical stress-strain curves with indication of compression strength  $\sigma_c$  and elastic modulus  $E_c$  (left) and the flexural strength  $\sigma_f$  (right).

#### 4. Thermal properties of simulant materials

300 We also measured the thermal conductivity of our asteroid simulants in the laboratory. We adopted the hot wire method, using a probe developed by the *Centre Scientifique et Technique du Batiment* CSTB (Scientific and Technical Center for Building), in Grenoble, that has a typical accuracy of 5%. To do this, one  $95 \times 95 \times 44$  mm sample was cut in two pieces of about  $95 \times 95 \times 20$  mm, 305 to place the probe between them, as shown in Fig. 1C. The measurement was repeated three times and attention was paid to limit the maximum temperature transmitted to the sample to be below 310 K. The three measurements were done at ambient temperature (the average temperature was 295 K) and atmospheric pressure in air. The measurement was quite repeatable, showing a value of 310 thermal conductivity being between  $0.43$  and  $0.47 \text{ W m}^{-1} \text{ K}^{-1}$ .

#### 5. Impact experiments: ejecta, crater sizes and morphology

In order to study the response of our asteroid simulant material to collisions with particles at typical impact speeds occurring in the asteroid Main Belt, we performed a series of laboratory hypervelocity impact experiments. We used 315 the facilities of the Impact Lab of the University of Kent in Canterbury (UK). The main instrument used here is a 2-stage light-gas gun (LGG), which can

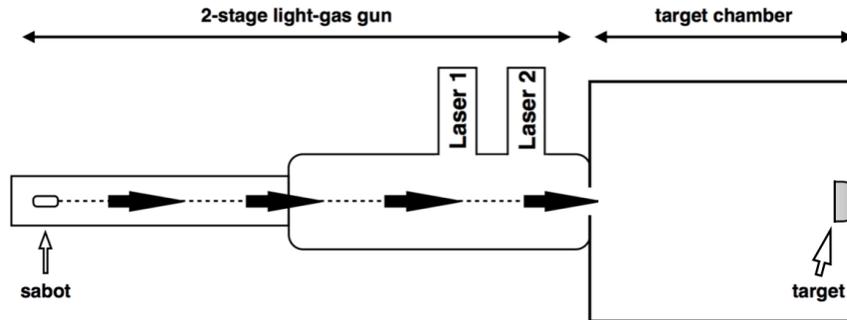


Figure 3: Schematic of the 2-stage light-gas gun of the University of Kent.

achieve speeds up to  $7.5 \text{ km s}^{-1}$ , while the typical impact speed in the Main Belt is around  $5 \text{ km s}^{-1}$  (Bottke et al., 1994). In our experiments we used spherical projectiles made of stainless steel, with density of  $7,870 \text{ kg m}^{-3}$ . It should be noted that this value is much higher than the typical densities of the micrometeoroids that impact asteroid boulder in the Main Belt and near-Earth space, which are of cometary and asteroidal origin (Nesvorný et al., 2010, and references therein). However, by choosing a high density projectile we wanted to ensure the production of a substantial amount of ejecta, even at the lower speeds.

The configuration of the shotgun cartridges used in the first stage, and gas pressure in the second stage of the LGG, were calibrated to produce controlled impact speeds varying between  $0.8$  and  $5 \text{ km s}^{-1}$ . The actual impact speed for each shot is measured by timing the passage of the projectile through two laser curtains placed down-range in the gun and is indicated in Tab. 3. For all the tested impact speeds we used projectiles with diameter  $1 \text{ mm}$  with mass of  $0.0041 \text{ gr}$ . The shot with impact velocity of  $4.7 \text{ km s}^{-1}$  was repeated tree times using stainless steel projectiles with diameters of  $1$ ,  $1.5$  and  $2 \text{ mm}$ , the latter two having masses  $0.0139 \text{ gr}$  and  $0.0330 \text{ gr}$ , respectively. This procedure was carried out in order to compare impacts at similar speeds, but with different kinetic energies. The uncertainty of the measurements of projectile diameters

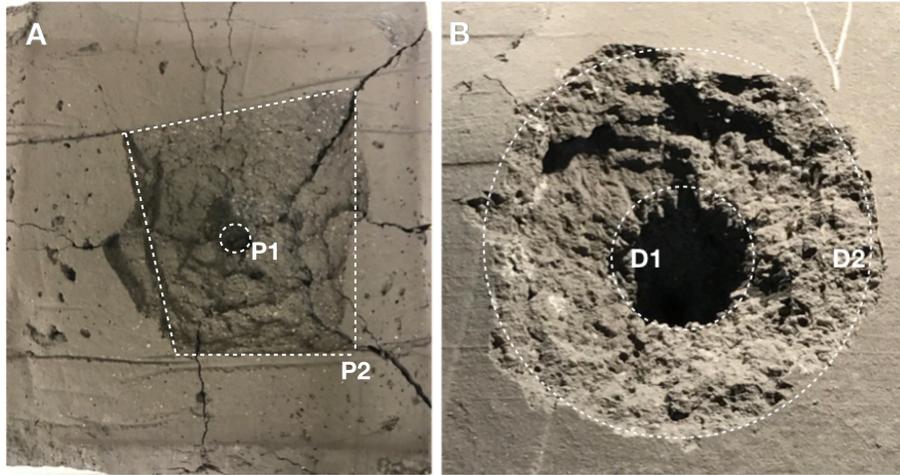


Figure 4: The produced craters of two experiments. (A) Crater of the shot G181109#3 with impact speed  $4.71 \text{ km s}^{-1}$  and 2 mm size projectile. The area between P1 and P2 was used to estimate the roughness of the cratered area; (B) Crater of the shot G181109#2 with impact speed  $4.81 \text{ km s}^{-1}$  and 1 mm size projectile. Both images show the central crater (inner white dashed circle, in B panel with diameter D1), and the outer spallation area (zone zone in between D1 and D2) with incipient spallation. Diameters, D1 and D2, and depths are in Tab.3.

is about  $10 \text{ }\mu\text{m}$ .

Targets comprised of the blocks of CM simulants prepared as described in section 2 with approximate dimensions  $95 \times 95 \times 44 \text{ mm}$  thickness. The target  
 340 samples were placed in a holder which was mounted on the door of the target chamber and were impacted horizontally (Fig. 3). The impact chamber was evacuated prior to each shot, the air pressure inside the target chamber was measured and is reported in Tab. 3. At the bottom of the target chamber were placed several layers of aluminium foil that enabled the collection of the ejected  
 345 fragments after each shot for later inspection.

With these experiments we primarily aimed to examine the crater sizes, shapes and morphology, along with the type of the produced ejecta (e.g. monominerallic, multiminerallic). After each shot the targets were removed with great care from the chamber door in order to preserve the shape of the produced

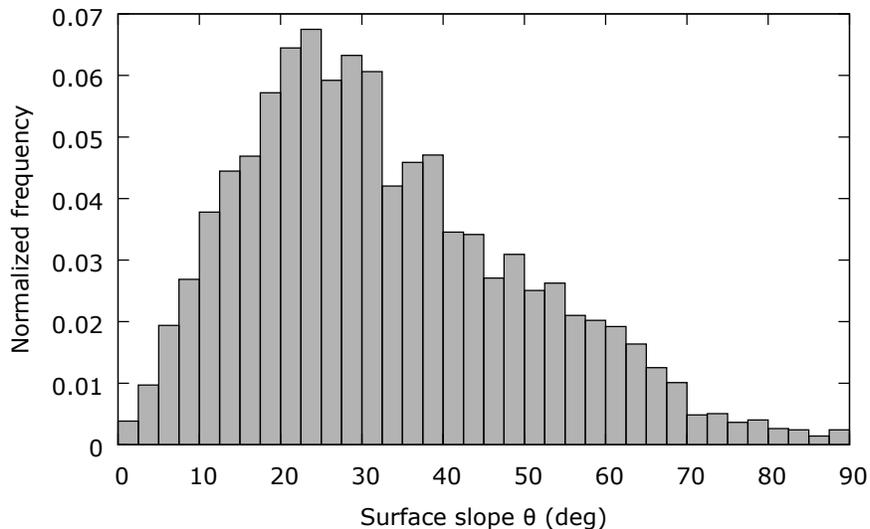


Figure 5: The distribution of the surface slopes within the damaged area of the surface of the simulant after the shot G181109#3 (see Fig. 4A) as measured from the 3D scan of the simulant.

350 crater. For each crater it was measured the total depth  $d$ , the crater diameter  $D1$  and the larger diameter encompassing the area of the spall  $D2$ , which are presented in Tab. 3. The spallation appears when there is tensile failure as the compressive stress is reflected at the surface and becomes tensile stress. In addition, we present the  $d/D1$  and  $d/D2$  ratios of each crater. In the case  
 355 of three impacts where the impact speed is constant but the impact energy is different the  $d/D1$  and  $d/D2$  ratios appears also constant. In the case of the shot G181109#3 the surface of the target was not smooth but had large cracks that were formed during the production phase. The resulted crater was not circular, but had an irregular shape following those cracks (see Fig. 4A).

360 Next we estimated the roughness of the impacted surface. For this measurement we used the target with the largest formed crater (with a diameter of  $\sim 5$  cm) from the shot G181109#3 (Fig. 4A). The cratered simulant block was scanned using a commercial 3D software (Qlone) and the object exported into a *wavefront* (*.obj*) file format representing a mesh of 344,312 triangular facets

365 (with a side-length of 0.3 mm). A region of interest with 32,911 facets corre-  
 sponding to the surface of the simulant damaged by the impact and without  
 the central hole (region between polygons P1 and P2 of Fig. 4A) was selected  
 to perform the roughness analysis. The mean plane of the surface was calcu-  
 lated along with the angle between the normal of this plane and the  $i$ -th facet  
 370 that we call  $\theta_i$ . Following Hapke (1984), we characterise the roughness of the  
 cratered region of the simulant by its mean surface slope  $\bar{\theta}$ , which is defined as:  
 $\tan(\bar{\theta}) = (2/\pi) \int_0^{\pi/2} \tan(\theta) a(\theta) d\theta$  where  $a(\theta)$  is the distribution of the surface  
 slope normalised such that  $\int_0^{\pi/2} a(\theta) d\theta = 1$  (as displayed in Fig. 5). We find  
 that  $\bar{\theta} = 32^\circ$  for the surface of our sample damage by the impact. In section  
 375 6, we compare this value with the Ryugu boulder’s surface from Grott et al.  
 (2019) data.

For four out of six impact experiments ejecta was recovered. Ejecta could  
 not be collected for the low speed shots S181106#1 and G181105#1 as the  
 craters were very small and the excavated crater mass was comminuted to a  
 380 fine powder, which was impossible to collect using the setup. For the rest of the  
 shots, G181105#2, G181109#1, G181109#3, G181109#2, that were performed  
 at higher and narrow speed range between 4.07 and 4.81 km s<sup>-1</sup>, we see similar  
 characteristics of the ejecta. First of all ejecta has a size range from centimetres  
 to sub-microns. Furthermore, as can be seen from the different panels of Fig. 6,  
 385 the ejecta consist of both multiminerallitic and monomineralic fragments, where  
 the glass inclusions and the pyrite dominate the monomineralic population of  
 sub-mm-sized fragments.

At any impact speed, the largest fragments, such as the one at the centre  
 of Fig. 4A, which measures 8 mm along its longest axis, are multiminerallitic:  
 390 the glass spherules and the pyrite are still embedded in the CM-like matrix.  
 The multiminerallitic nature of impact fragments was expected as described in  
 the introduction of this work. Additionally, we see many small fragments which  
 consist primarily of a glass spherule or a pyrite attached to a smaller piece of  
 the CM matrix. On the other hand, several smaller fragments ( $\sim 500 \mu\text{m}$  in  
 395 size) are clearly monomineralic (see morphological units  $\alpha$ ,  $\beta$  in the Fig. 6).

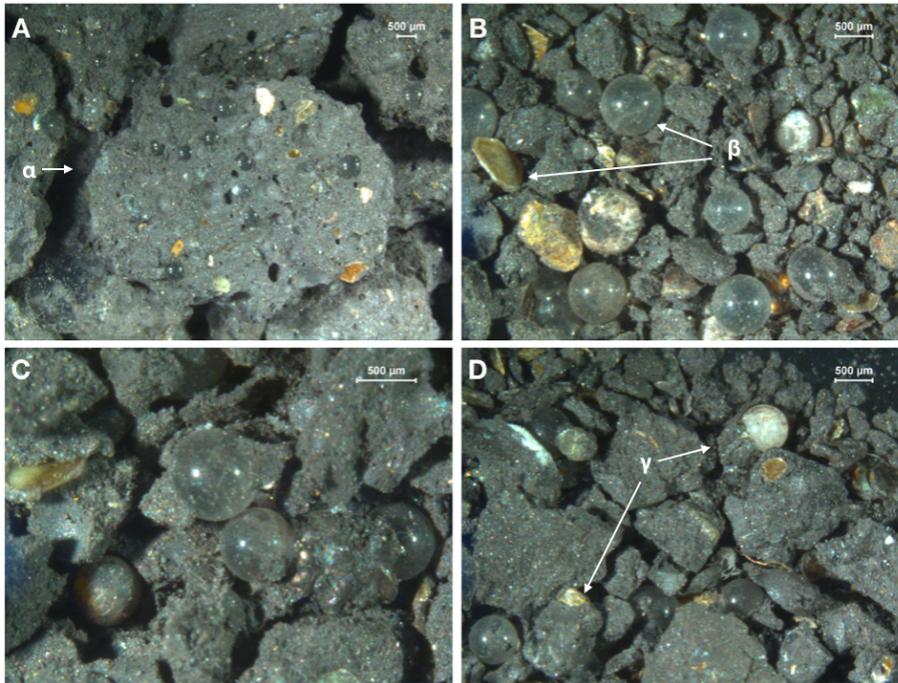


Figure 6: Parts of collected ejecta produced at four impact experiments. In each case we found multiminerale large fragments ( $\alpha$ ), monominerale fragments ( $\beta$ ) and individual inclusions that were attached to a small amount of CM matrix.

In all the ejecta material collected we discovered severely shocked individual glass spherules (Fig. 6). These spherules have white colour indicating that are totally shattered and very possibly originate from the centre (or close to the centre) of the impact (see an example in Fig. 7).

## 400 6. Discussion

The first action we did was to measure the mechanical properties of the simulant material and compare them to the properties of some known weak carbonaceous meteorites. In particular, the compressive strength of the simulant is found to be  $1.82 \pm 0.17$  MPa, at least an order of magnitude smaller than  
 405 the one measured for the CM2 meteorites Murchison (Miura et al., 2008) and Sutter's Mill (Jenniskens et al., 2012). The value of the compressive strength

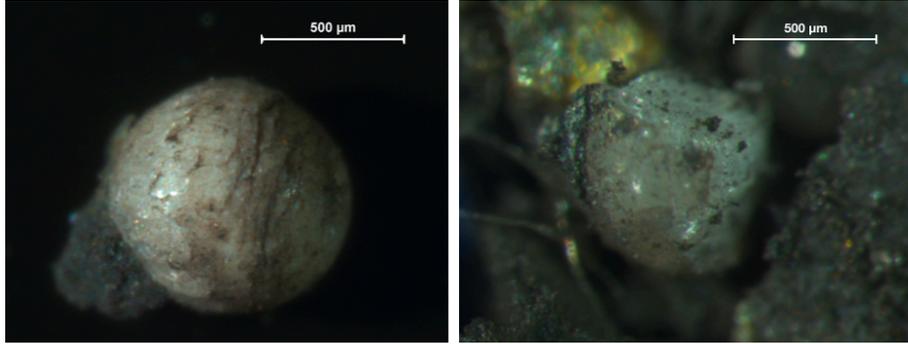


Figure 7: Examples of a shocked, shattered and broken spheres that were used as inclusions in the CM matrix.

Table 3: Summary of the impact experiments where are given: the speed of each impact ( $V_{imp}$ ), the air pressure inside the impact chamber ( $P$ ), the diameter of the projectile ( $D_{proj}$ ), the impact energy ( $E_{imp}$ ), the depth ( $d$ ) and diameters of pit and spall ( $D1$  and  $D2$ ) of each crater. In addition, are given the depth-to-diameter ratio for both cases ( $d/D1$  and  $d/D2$ ) and the spall-to-pit ratio ( $D2/D1$ ).

Shot ID	$V_{imp}$ ( $\text{km s}^{-1}$ )	$P$ (mbar)	$D_{proj}$ (mm)	$E_{imp}$ (kJ)	$d$ (mm)	$D1$ (mm)	$D2$ (mm)	$d/D1$	$d/D2$	$D2/D1$
S181106#1	0.85	0.76	1.0	1.48	7.5	1.97	4.20	3.80	1.78	2.13
G181105#1	2.13	0.90	1.0	9.25	9.0	2.95	12.05	3.05	0.74	4.08
G181105#2	4.07	0.70	1.0	33.79	4.0	6.78	23.25	0.59	0.17	3.43
G181109#1	4.70	0.49	1.5	152.13	12.0	9.24	27.20	1.29	0.44	2.94
G181109#3	4.71	0.25	2.0	362.16	17.0	14.00	51.62	1.21	0.33	3.7
G181109#2	4.81	0.55	1.0	47.00	5.5	5.92	18.88	0.93	0.29	3.2

of our simulant is closer to the one of the C2 ungrouped Tagish Lake meteorite that was found to be around 0.7 MPa. The flexural strength of the simulant was measured to be  $0.72 \pm 0.07$  MPa: this value, according to our approach  
410 presented in section 3, represents an upper limit for the tensile strength of the simulant and thus probably even lower than the tensile strength of the CI meteorite Ivuna ( $0.7 \pm 0.2$  MPa) and C2 Tagish Lake ( $0.8 \pm 0.3$  MPa) as given by Tsuchiyama et al. (2009). However, the inferred tensile strength of our simulant is at least one magnitude lower the one derived for CM2 meteorite  
415 Murchison and Sutter’s Mill, and Murray ( $8.8 \pm 4.8$  MPa; Tsuchiyama et al., 2009). Here we have to mention that our experiments were conducted in air, but this does not affect the measurements since the compressibility of the air is below that of the solid materials.

The measured thermal conductivity  $\kappa$  of the CM carbonaceous chondrite  
420 Cold Bokkeveld at the temperature of 200 K is  $0.50 \text{ W m}^{-1} \text{ K}^{-1}$  (Opeil et al., 2010) and is temperature dependent ( $\kappa$ -values were measured with the meteorite samples inside a cryostatic chamber in the temperature range between 300 K and 5 K and at pressure  $< 1.33 \times 10^{-4}$  Pa). On the other hand, the  $\kappa$ -value of our simulant was measured to be between  $0.43$  and  $0.47 \text{ W m}^{-1} \text{ K}^{-1}$   
425 at room temperature (295 K) in air. However, because thermal conductivity is temperature dependent, we need to correct the measured  $\kappa$ -values for Cold Bokkeveld and our simulant to the same temperature at which Ryugu’s boulder thermal conductivity was derived. The boulder of Ryugu was calculated to have a  $\kappa=0.06\text{--}0.16 \text{ W m}^{-1} \text{ K}^{-1}$  at 230 K (Grott et al., 2019). Using the temperature  
430 dependence curve of the Cold Bokkeveld CM meteorite, we make the correction on the basis of a linear fit to that data of Fig. 3 of Opeil et al. (2010) for which temperature is greater or equal to 150 K and we get  $\kappa=0.53 \text{ W m}^{-1} \text{ K}^{-1}$  at a temperature of 230 K for the CM Cold Bokkeveld. Applying the same temperature dependence of the Cold Bokkeveld to our thermal conductivity measurement after having subtracted the air thermal conductivity ( $0.03 \text{ W m}^{-1} \text{ K}^{-1}$ ),  
435 we result in  $\kappa=0.36 \text{ W m}^{-1} \text{ K}^{-1}$  at 230 K. This shows that the  $\kappa$  of our simulant is lower than the one measured for Cold Bokkeveld. The bulk porosity

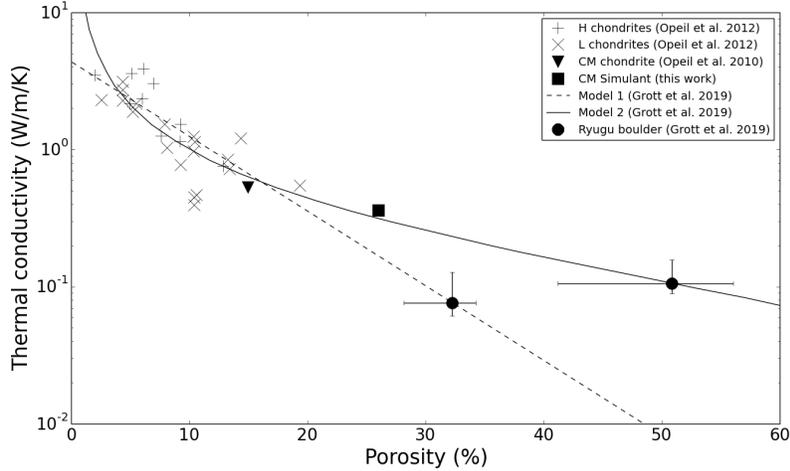


Figure 8: Thermal conductivity of H, L and CM meteorites, in comparison with the measured thermal conductivity of the surface boulders of Ryugu and with our asteroid surface simulant. Adapted from Grott et al. (2019).

$\phi$  of our simulant is higher than those of meteorites, placing it in between the value of Cold Bokkeveld and the one estimated for the boulder of Ryugu. The  
 440 thermal conductivity vs. porosity data of our simulant, the Cold Bokkeveld meteorite and Ryugu’s boulder plot along the prediction of the Model 2 ( $\kappa = 0.11 \times (1-\phi)/(\phi)$ ) of Grott et al. (2019) and are presented in Fig. 8.

We therefore summarise that the mineralogy, water content and amount of inclusions of our simulant is close to CM meteorites, while the thermal conduc-  
 445 tivity and mechanical properties are in between those of CM meteorites and the boulder of Ryugu. In the light of the new space mission data, it appears that we miss asteroid material from our meteoritic collection, perhaps not in terms of composition, but very likely in terms of thermo-mechanical properties. This is very reasonable as Earth’s atmosphere filters out the weakest fragments.  
 450 Therefore, the novelty behind this work is not to precisely reproduce the existing sample of weak carbonaceous meteorites but to mimic more accurately the surfaces of carbonaceous asteroids.

The morphology of the craters created by hypervelocity impact of mm-sized

projectiles on our simulants is typical of those of brittle materials (Dufresne  
455 et al., 2013). All craters show a central pit with diameter  $D1$  and a spallation  
area around. There was a high density difference between our simulant material  
(1,980 Kg m<sup>3</sup>) and the steel projectiles (7,870 Kg m<sup>-3</sup>), which resulted into  
large spalling areas and also large depth-to-diameter ratio. Our impact craters  
simulate the result from micrometeorite bombardment and the spall-to-pit di-  
460 ameter ratio ( $D2/D1$ ) ranges between 2 and 4. The latter is slightly larger than  
the one for lunar microcraters that have spall-to-pit ratio around 3 and more  
precisely falling in the range 1.5–3, as it is reported by (Nakamura, 2017). The  
depth of our craters is large (see Tab. 3) and this can be explained by the large  
ratio of projectile-to-target density. In addition, our depth-to-diameter ratio is  
465 also large, ranging between 0.2 and 0.7.

On the other hand, the depth-to-diameter ratio derived from the large craters  
(with diameters between  $\sim 10$  and 290 m) found on asteroid Ryugu was measured  
to be 0.14–0.2 (Sugita et al., 2019), and is smaller than the ones we derive in this  
work. This difference in crater depths between our experiments and Ryugu’s  
470 craters can be explained by a series of observations: Sugita et al. (2019) explain  
the shallow craters on Ryugu as due to the presence of loosely unconsolidated  
material that could have moved, producing landslides, and is also responsible for  
the observed crater floor morphologies. Moreover, another reason to reduce the  
crater sizes is the presence of m-sized boulders, comparable to the impactors’  
475 sizes, which absorb a large portion of impact energy. As a conclusion, our  
experiments are more comparable with the meteoroid impacts on the surface  
boulders of Ryugu or Bennu which form mini-craters (also known as cavities;  
R. Ballouz, private communication). These mini-craters have been detected in  
high spatial resolution images from the detailed survey phase of the OSIRIS-  
480 REx mission on boulders of Bennu. The cavities have diameters ranging from  
about 3 to 30 cm (Ballouz et al., 2019). A quantitative comparison between  
the properties of our experimental craters and the mini-craters will be possible  
when spacecraft data will be publicly available. Moreover, our experimental  
craters resemble, at least morphologically, the surface at similar cm-scales that

485 MASCOT revealed on Ryugu boulders (Grott et al., 2019; Jaumann et al.,  
2019). The roughness at a scale smaller than the MARA footprint – about 5 cm  
in diameter – of the Ryugu boulder observed by MASCOT can be estimated  
from the best fit parameters of the thermal model used by Grott et al. (2019) to  
interpret MARA radiometric measurements. In particular, Grott et al. (2019)  
490 find that a thermal model with 0.34 of the surface area observed by MARA  
covered by hemispherical craters produce the best fit to the data. From the  
crater opening angle ( $\gamma_C$ ) and the crater areal density ( $\rho_C$ ) it is possible to  
calculate the mean slope of surface  $\bar{\theta}$  as described by Delbo (2004) and shown  
by Hanuš et al. (2018). We find that  $\gamma_C=90^\circ$  and  $\rho_C=0.34$  yield  $\bar{\theta}=28.7^\circ$ . This  
495 value is very similar to the mean surface slope of the region of our simulant  
damaged by the hyper-velocity impact that we performed in the laboratory  
( $\bar{\theta}=32^\circ$ ). We thus conclude that the degree of roughness of the surface created  
by the impact on our simulant and that observed by MASCOT/MARA at very  
similar spatial scale are very similar. It is thus possible that MASCOT observed  
500 a surface resulting from the bombardment of micrometeorites over the age of  
Ryugu.

The large multiminerale fragments, like the one in Fig. 6A, most probably  
originate from the spalling area, where the shock was already disseminated. On  
the other hand the shattered and broken glass inclusions should come from the  
505 central crater. This idea could be supported by recent hypervelocity impact  
experiments conducted on L ordinary chondrite meteorites (Michikami et al.,  
2019) where the impact induced cracks appear to grow through the chondrules  
that are located near the impact point, and around the chondrules that are  
located away from the impact point. Although there are indications for a differ-  
510 ence in the crack propagation, the authors did not measure a strong correlation  
between the type of crack propagation in chondrules and the distance of those  
chondrules from the impact point concluding that for the near-Earth asteroid  
Itokawa impacts is the main mechanism of regolith production with preference  
to multiminerale fragments. On the other hand, early impact experiments on  
515 basalt with olivine inclusions showed a preference in the generation of olivine

dominated fragments, indicating a crack propagation around the density boundaries between olivine and basalt matrix (Durda & Flynn, 1999). From our impact experiments, where both monomineralic and multiminerallitic fragments are present, does not lead to an obvious separation of the production mechanism and thermal cracking is not the only process that produces monomineralic fragments. By broadly combining these three different impact experiments we could argue that there should be a difference on how strongly are bound the inclusions to the matrix of different materials, natural and simulant.

The key characteristic for a clear answer, when sample will return from Ryugu and Bennu, could be the presence of shocked fragments where here are represented by the shattered glass inclusions.

## 7. Conclusions

We created in the laboratory a mineralogically analogue material to the CM carbonaceous meteorites but with mechanical and thermal properties intermediate between those of the CM meteorites, that we receive on Earth, and those measured by space missions for the boulders on the C-complex near-Earth asteroids Ryugu and Bennu. We added to the matrix of a CM-like composition spherical glass beads in order to simulate the multiminerallitic nature of known CMs due to the presence of chondrules and pyrite, the latter being occasionally present in CM meteorites, such as in the case of the Murchison meteorite (Fuchs et al., 1973).

We measured the compressive and tensile (flexural) strength of the simulants, that we found to be around 1.8 and 0.7 MPa, respectively (with a 0.17 and 0.07 MPa 90% confidence interval). We also measured the thermal conductivity (in air) of the simulants at room temperature (295 K), which resulted in a value between 0.43 and 0.47 W m<sup>-1</sup> K<sup>-1</sup>. The values of the mechanical and thermal properties are intermediate between the typical ones of carbonaceous chondrites (Brown et al., 2002; Miura et al., 2008; Opeil et al., 2010; Jenniskens et al., 2012) and those measured and/or inferred for the boulders of the asteroid

545 Ryugu (Grott et al., 2019; Sugita et al., 2019). Preliminary observations of  
the OSIRIS-REx mission indicate that also the boulders covering the surface  
of Bennu have similarly low thermal conductivities and likely low mechanical  
strengths (Dellagiustina et al., 2019).

We performed a series of hypervelocity ( $0.8\text{--}5\text{ km s}^{-1}$ ) impact experiments  
550 in order to study how these materials fragment, simulating their response to  
collisions of micro-meteorites in the asteroid Main Belt. We determined that  
craters resulting from these impacts have typical morphologies, with large depth-  
to-diameter ratios that is characteristic of impacts on porous media. Morpho-  
logically, the surface resulting from the impact on the simulants present a degree  
555 of roughness that qualitatively and quantitatively resembles those observed by  
MASCOT on the boulders of Ryugu (Grott et al., 2019).

We find that the material fragmented and ejected by an impact at a velocity  
of  $4\text{--}5\text{ km s}^{-1}$  contains both multi- and monomineralic fragments, including  
single glass spherules that were easily separated from the matrix by the impact  
560 shock (we had originally embedded these glass spherules in the matrix of our  
simulants to mimic chondrules).

The results from this study indicate that it could be more difficult than pre-  
dicted (see introduction) to distinguish grains produced from thermal fragmen-  
tation from those created by the impact of micro-meteorites in the regoliths of  
565 asteroids that have CM-like composition with mechanically weak surface boul-  
ders. This is because it was originally predicted that thermal fragmentation  
should produce monomineralic fragments Hazeli et al. (2018) than impacts  
would do. A further work will investigate the impact effects on CI-like materials  
and also the effect of thermal cracking on both CM- and CI-like simulants with  
570 weak mechanical strengths similar tho those inferred for boulders on Ryugu.

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