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Experimental Investigation and 1D Analytical Approach on Convective Heat Transfers in Engine Exhaust-Type Turbulent Pulsating Flows

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Abstract

In many practical engineering situations, such as in the exhaust pipes of Internal Combustion Engines, heat is transferred under conditions of pulsating flow. In these conditions, the heat transfer mechanism is affected by the pulsating flow parameters. The objective of the present work was to experimentally investigate heat transfers for pulsating turbulent pipe flows with large-amplitude oscillations. A specific experimental apparatus able to reproduce a pulsating flow representative of the engine exhaust was designed. A stationary turbulent hot air flow with a Reynolds number of 30000, based on the time average velocity, is excited through a pulsating mechanism and exchanges thermal energy with a water cooled steel pipe. Pulsation frequency ranged from 10 to 95 Hz. An analytical formulation, derived from the 1D energy balance equation for a turbulent pulsating flow, and an ad-hoc experimental procedure evidenced that, in the case of a reversed pulsating flow, a contribution to the energy balance equation through the boundaries, which impacts the total convective heat transfers, may occur. It was also observed that the flow pulsation enhances heat transfers in the entire range of the investigated frequencies. In particular, the results show that when the flow is excited with a frequency equal to a resonance mode of the system, a strong increase in heat transfers occurs. Instantaneous measurements of air velocity and temperature demonstrated that the increase in the energy axial advection due to the oscillating component of the velocity is the major cause of the heat transfer enhancement. The relative Nusselt number was found to be directly linked to the ratio between the velocity oscillation amplitude and the time-average component of the axial bulk velocity, suggesting that this characteristic term is representative of the predominant heat transfer enhancement mechanism.

Keywords: Heat transfer enhancement, Turbulent pulsating flow, 1D analytical analysis, Experimental investigation, Internal combustion engines.

Declarations of interest: none

Highlights

- Convective heat transfers for pulsating turbulent pipe flows are investigated.
- The impact of the pulsation frequency on heat transfers is studied.
- A heat transfer enhancement in the case of a flow in resonant conditions is observed and analyzed.
- Increased Nusselt number is linked to the amplitude of the velocity waveform.

54 In a first study, Dec et al. [3] studied the influence of pulsation frequency, amplitude and mean flow rate on the heat transfers
55 for a pulse combustor tail pipe. Pulsation frequency was varied from 67 to 100 Hz, for a Reynolds number based on the time
56 average velocity varying between 3100 and 4750. Spatially averaged Nusselt numbers were observed to increase linearly with
57 both pulsation frequency and amplitude, while the Nusselt number enhancement decreased with increasing mass flow rate
58 despite accompanying increases in oscillation frequency and amplitude. Cycle-resolved Nusselt number results, discussed in
59 a second study by Dec et al. [4], showed that the Nusselt number was nearly constant throughout the cycle, with the exception
60 of local increases at flow reversal times. Several mechanisms responsible for the heat transfer enhancement in oscillating flows
61 were reviewed and discussed. Among these, acoustic streaming, which corresponds to the appearance of a secondary time-
62 averaged velocity component having the form of large longitudinal recirculation cells, and entrance length effects were
63 considered to have a slight influence on the observed Nusselt number enhancement. Detailed studies of time-resolved velocity
64 and temperature fields [5,6] showed that the heat transfer enhancement was most likely caused by a combination of increased
65 turbulent intensity and strong transverse flows generated during the streamwise velocity reversals.

66 Xu et al. [7] studied the flow properties of a self-excited Helmholtz pulse combustor elbow tailpipe. The results showed that,
67 due to pulsation and flow reversal, Dean Vortex forming, shedding and reforming processes periodically contribute to
68 convective heat transfer enhancement. With the same type of experimental apparatus, Zhai et al. [8] proposed a Nusselt
69 correlation for the pulsating flow, based on the addition of two independent physical properties. Applying the quasi-steady
70 theory and the Vaschy-Buckingham theorem to the convection heat transfer problem, the ratio between the velocity amplitude
71 and the time-averaged velocity and the ratio between the pulsation velocity scale and the time-averaged velocity were found
72 to be the aforementioned independent physical properties. Several studies were also conducted on turbulent and laminar
73 pulsating pipe flows. In the work of Patel et al. [9], the Reynolds number ranged from 7000 to 16500, while the pulsation
74 frequency was varied from 1 Hz to 3.33 Hz. Results showed that the Nusselt number was strongly affected by both pulsation
75 frequency and Reynolds number with a 44.4% maximum enhancement of the heat transfer coefficient at the pulsation
76 frequency of 3.33 Hz. In a study on a pulsating turbulent water stream, Zohir [10] also pointed out that the heat transfer
77 coefficient was strongly affected by pulsation frequency and amplitude and by the Reynolds number. The improvement in heat
78 transfers was attributed to an increased level of turbulence and the introduction of forced convection in the boundary layer.
79 The experimental studies by Said et al. [11] and Nishandar et al. [12] confirmed that the heat transfer coefficient was either
80 increased or decreased as a function of the pulsation frequency in turbulent conditions. Elshafei et al. [13] conducted a
81 numerical study of the heat transfers for a fully developed pulsating turbulent flow over a range of $10^4 \leq Re \leq 4 \cdot 10^4$ and
82 $0 \leq f \leq 70$ Hz and the results were compared with the available experimental data. Results showed a slight reduction in the
83 time-averaged Nusselt number with respect to that of a steady flow. However, in the fully developed established region, the
84 local Nusselt number was either increased or decreased, compared to Nu values for steady flow, depending on the frequency
85 parameter. In recent studies, flow pulsation was also investigated as an active method to enhance heat transfers in industrial

86 applications such as heat exchangers. Kharvani et al. [14] investigated the potential to increase heat transfers by using a rotating
87 ball valve as a pulse generator mounted downstream/upstream of a spiral-coil tube. The Reynolds number ranged from 6220
88 to 16300, while the pulsation frequency was varied from 0 to 20 Hz. It was observed that, although the overall average heat
89 transfer coefficient was strongly affected by the Reynolds number, it was enhanced by up to 26% for a pulsating flow compared
90 to a steady flow. Ghaedamini et al. [15] investigated the potential to increase heat transfers with wavy walled micro-channels.
91 Although such geometries demonstrated their capability of inducing chaotic advection, a substantial pressure drop penalty was
92 observed for highly modulated channels. To overcome this issue, a pulsating flow was introduced, leading to a strong
93 enhancement of heat transfers for channels with slightly modulated walls corresponding to a reasonable pressure drop penalty.
94 This increase was attributed to a chaotic advection phenomenon near a characteristic excitation frequency. In [16], Wantha
95 experimentally studied the enhancement of heat transfer in finned tube heat exchangers by using a pulsating air flow. The
96 influence of velocity amplitude, pulsation frequency, Reynolds number and blockage ratio of heat exchangers was investigated.
97 The results showed that the pulsation frequency and velocity amplitude play major roles in heat transfers. An empirical
98 correlation, based on a Colburn j-factor, was deduced from the experimental results.

99 Several attempts were also made to generalize Nusselt number correlations to pulsating flows in a pipe. Guo et al. [17]
100 numerically tested various forms of the Nusselt number correlation and proposed an improved version describing, on the one
101 hand, a high heat transfer enhancement for a large amplitude pulsation flowrate, and on the other hand both heat transfer
102 enhancement and reduction, depending on the pulsation frequency, for a small amplitude pulsation. As reported by the
103 aforementioned surveys, it has been frequently observed that pulsation frequency may have an impact on the convective heat
104 transfers. However, the variety of the experimental configurations and the variety of the pulsation creation mechanisms have
105 led to some controversies: both enhancement and degradation of convective heat transfers have been observed. Besides, the
106 main physical mechanisms involved have not been fully described.

107 In the present study, an experimental investigation was conducted on heat transfer phenomena for a pulsating turbulent pipe
108 flow, in a wide range of variation of the physical parameters. The main purposes were to investigate the impact of the pulsation
109 frequency on heat transfers and to identify the main physical mechanisms involved in the heat transfer modification. The
110 design of an experimental apparatus, able to reproduce a pulsating pipe hot air flow over a range of $10 \leq f \leq 95$ Hz and
111 $10^4 \leq Re \leq 5 \cdot 10^4$, representative of engine exhaust flow operating conditions, is presented. In the test-rig, the pulsating
112 hot air flow exchanges thermal energy with cold water flowing in the opposite direction. The work presented in this paper
113 focuses on the time- and space-averaged characterization of the convective heat transfers. Experimental results are linked to
114 the 1D energy balance equation of the pulsating turbulent pipe flow to identify the characteristic terms representative of the
115 heat transfer modification due to the pulsating conditions. In order to enhance the predictive capability of engine numerical
116 tools, these terms could be included in convective coefficient correlations, to take into account flow velocity fluctuations,
117 instead of using empirically calibrated corrective parameters as in models recently developed [18] to compute the gas

118 temperature drop along the exhaust runner and manifold. The 1D approach is adopted in the present work to be consistent with
 119 engine simulation codes commonly used when accurate modeling of heat transfers in exhaust systems is required [19]. Data
 120 collection and reduction are presented; results in terms of a relative Nusselt number are reported and discussed based on this
 121 assumption.

122 2. ANALYTICAL FORMULATION OF THE PROBLEM

123 The effect of the pulsation on heat transfers can be characterized in terms of the relative Nusselt number Nu_{rel} , defined as the
 124 ratio of the time-averaged Nusselt number for the pulsating flow to the corresponding one for a steady flow with the same
 125 time-averaged Reynolds number. For an air flow into a cylindrical control volume, the time-averaged Nusselt number is
 126 defined as in the following equation:

$$Nu = \frac{h_{air} D}{\lambda} = \frac{Q_{conv} D}{S \Delta T_{lm} \lambda} \quad (1)$$

127 where Q_{conv} is the time-averaged convective heat transfer, D the internal diameter of the pipe, S the exchange surface, ΔT_{lm}
 128 the logarithmic mean temperature difference between the air and the internal wall of the pipe and λ the thermal conductivity
 129 of air. The relative Nusselt number definition has been adopted in several previous studies because of its ability to bring out
 130 the impact of the pulsation frequency on heat transfers and to identify a corrective coefficient for a Nusselt correlation
 131 accounting for pulsating effects. The practical difficulty in such an approach consists in correctly assessing the time-averaged
 132 convective heat transfer Q_{conv} . Since in some applications the measurement of the heat flux exchanged with a solid wall is
 133 difficult to achieve, Q_{conv} is computed starting from the air flow properties by solving the energy balance equation. The time-
 134 averaged convective heat transfer is generally computed from the variation of the time-averaged component of the air enthalpy
 135 through the inlet and outlet sections of the control volume, but it is shown in the following development that this calculation
 136 does not take into account characteristic terms related to the pulsating component of the flow. In the present work, Q_{conv} is
 137 derived from the time-averaged and space-integrated form of the instantaneous energy balance equation for a pulsating
 138 turbulent pipe flow. The terms related to the heat flux propagated from the advection of the oscillating component of the flow
 139 are highlighted.

140 Assuming negligible viscous dissipation, the 2D instantaneous local energy balance equation in cylindrical coordinates, for an
 141 incompressible pipe flow with constant fluid properties, exhibits the following form:

$$\rho c_p \left(\frac{\partial T}{\partial t} + \frac{\partial(uT)}{\partial x} + \frac{\partial(vT)}{\partial r} \right) = \lambda \left(\frac{\partial^2 T}{\partial x^2} + \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial T}{\partial r} \right) \right) \quad (2)$$

142 where u and v represent respectively the axial and the radial instantaneous air velocity components, T the temperature of air,
 143 ρ the fluid density and c_p the specific heat at constant pressure. As proposed by Reynolds et al. [20], in the specific case of a
 144 turbulent pulsating flow, each of the flow properties can be decomposed into three different terms, as expressed in the following
 145 equation:

$$Z(x, r, t) = \bar{Z}(x, r) + \tilde{Z}(x, r, t) + Z'(x, r, t) \quad (3)$$

146 where $\bar{Z}(x, r)$ represents the time-averaged component, $\tilde{Z}(x, r, t)$ is the oscillating term of the coherent cycle-stationary pattern
 147 and $Z'(x, r, t)$ corresponds to the turbulent fluctuations term. Time averaging ($\bar{\quad}$) determines $\bar{Z}(x, r)$ and the phase-
 148 average ($\langle \quad \rangle$), i.e. the average over a large ensemble of points having the same phase with respect to a reference oscillator,
 149 leads to:

$$\langle Z(x, r, t) \rangle = \bar{Z}(x, r) + \tilde{Z}(x, r, t) \quad (4)$$

150 Phase-averaging removes the background turbulence and extracts only the organized motions from the total instantaneous
 151 profile. For the sake of brevity, some useful mathematical properties that follow from the basic definitions of the time and
 152 phase averages are not reported here, they are detailed in [20]. Applying successively the phase averaging and then the time
 153 averaging operators to Eq. 2, in which the triple decomposition is introduced, leads to:

$$\rho c_p \left(\frac{\partial(\bar{u}\bar{T})}{\partial x} + \frac{\partial(\tilde{u}\tilde{T})}{\partial x} + \frac{\partial \langle u'T' \rangle}{\partial x} + \frac{\partial(\bar{v}\bar{T})}{\partial r} + \frac{\partial(\tilde{v}\tilde{T})}{\partial r} + \frac{\partial \langle v'T' \rangle}{\partial r} \right) = \lambda \left(\frac{\partial^2 \bar{T}}{\partial x^2} + \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial \bar{T}}{\partial r} \right) \right) \quad (5)$$

154 The left-hand side of Eq. 5 represents the energy transport by mean flow, oscillating motion and fluctuating motion, through
 155 the control volume, due to axial and radial advection. The right-hand side of the equation describes the heat conduction, through
 156 the control volume, in axial and radial directions. To identify the characteristic terms, in a 1D approach, representative of the
 157 impact of the oscillating motion on heat transfers, Eq. 5 is integrated over the cross section of the pipe Σ . Assuming an
 158 axisymmetric flow, the cross section integration operator is defined as:

$$\{\cdot\}_{\Sigma} = 2\pi \int_0^{R_{in}} (\cdot) r dr \quad (6)$$

159 where R_{in} is the pipe internal radius. The radial advection terms in Eq. 5 are integrated by parts and, by considering a condition
 160 of no-permeability at the wall, it is shown that they are all equal to zero. The radial conduction term is integrated as follows:

$$2\pi \lambda \int_0^{R_{in}} \frac{\partial}{\partial r} \left(r \frac{\partial \bar{T}}{\partial r} \right) dr = 2\pi R_{in} \lambda \left. \frac{\partial \bar{T}}{\partial r} \right|_{wall} \quad (7)$$

161 The heat flux boundary condition corresponds to heat convection at the wall surface:

$$2\pi R_{in} \lambda \left. \frac{\partial \bar{T}}{\partial r} \right|_{wall} = \langle q(x, t) \rangle \quad (8)$$

162 where $q(x, t)$ is the local specific convective heat transfer per unit length ∂x . Considering that axial diffusive transport is
 163 negligible compared to the advective transport of energy, the 1D local energy balance equation becomes:

$$\rho c_p \left(\frac{\partial \{\bar{u}\bar{T}\}_{\Sigma}}{\partial x} + \frac{\partial \{\tilde{u}\tilde{T}\}_{\Sigma}}{\partial x} + \frac{\partial \langle (u'T') \rangle_{\Sigma}}{\partial x} \right) = \langle q \rangle \quad (9)$$

164 By integrating Eq. 9 over the total length L of the pipe (corresponding to the control volume, between the inlet and outlet
 165 measuring sections, in the following), the energy conservation equation is written as:

$$\rho c_p \left(\underbrace{\{\bar{u}\bar{T}\}_{\Sigma} \Big|_0^L}_A + \underbrace{\{\bar{u}\bar{T}'\}_{\Sigma} \Big|_0^L}_B + \underbrace{\{<u'T'>\}_{\Sigma} \Big|_0^L}_C \right) = Q_{conv} \quad (10)$$

166 The total time-averaged convective heat transfer Q_{conv} is equal to the sum of three different terms: the term A which physically
 167 represents the energy variation of the time-mean component of the flow across the pipe, the term B which represents the energy
 168 variation due to the oscillating component of the flow and the term C corresponding to the advective transport of energy by
 169 fluctuating motion due to turbulence fluctuations. The statement of Eq. 10 clarifies that, whenever a direct measurement of
 170 Q_{conv} is not available, instantaneous measurements of the air velocity and temperature are required to correctly compute all
 171 the terms in the left-hand side of Eq. 10. In this study, a water flow was used to cool the external surface of the pipe. In addition
 172 to keeping the wall temperature cold, constant and quite homogeneous in the pipe section where heat transfers are
 173 characterized, this experimental setup allows Q_{conv} to be evaluated directly from the water temperature measurements. In this
 174 manner, it is possible to avoid performing instantaneous measurements of the flow properties at the inlet and the outlet of the
 175 test-section.

176 3. EXPERIMENTAL SETUP AND PROCEDURES

177 3.1. Experimental Setup

178 A schematic diagram of the pulsating flow facility is depicted in Fig. 1. It comprises three main parts: the first one produces a
 179 hot stationary air flow, the second one transforms the stationary flow to a pulsating flow, and in the last part, in which the flow
 180 develops, heat transfers are estimated. In the first part of the test rig, the dry compressed air mass flow rate is measured and
 181 regulated by a Brooks SLA5853S {1}, a thermal effect mass flow meter with a maximum flow rate of 2500 NI/min and with
 182 a calibration uncertainty of 0.73% of full scale. Then, air is heated by three Sylvania inline air heaters {3} with a total electric
 183 power of 12 kW, ensuring a maximum air temperature of 400°C at the maximum mass flow rate. Hot flow is finally stored in
 184 a 30-litre steel tank {4} designed to withstand a maximum air pressure of 10 bar and to dampen the flow pulsation coming
 185 back from the pulse generator. Once the hot air flow has been generated it is forced to pass through the pulsating mechanism
 186 {5-7}: a mono-cylinder head was chosen to produce a pulsating flow with a maximum frequency of 95 Hz, equipped with a
 187 classical pushrod valve train entrained by an electric engine with a power of 3 kW and a nominal velocity of 3000 rpm. In
 188 detail, hot air flows from the bottom of the cylinder head and only one of the two intake valves is alternately closed and opened
 189 to create the pulsating flow. The intake valve was chosen because of its higher diameter. Air leakages in the cylinder head
 190 were experimentally estimated to be below 0.5 kg/h, leading to an error on the mass flow rate measurement of <1%. In order
 191 to determine the camshaft position and rotation velocity, an encoder with a resolution of 0.1° is placed on the camshaft. Once
 192 the pulsating flow is generated, it is forced to pass through a steel pipe {8-10} in which it develops and exchanges thermal
 193 energy. Finally, at the end of the pipe, a 6-litre tank {11} is placed to muffle pressure pulsations. The entire flow facility is
 194 then linked to the exhaust line of the laboratory. As shown in Fig. 1, the steel pipe, with an internal diameter of 57.76 mm, is

195 composed of three different sections: the first one, called *developing section* {8}, has a length of 2.65 meters and a pipe
 196 length/internal diameter ratio of 48.4. It was designed to be long enough to completely develop the velocity flow field in the
 197 case of a steady turbulent flow. The pipe is thermally insulated to avoid high heat energy losses. The test section {9} (detailed
 198 in Fig. 2), with a length of 1 meter, is placed at the outlet of the developing section and is designed to be able to characterize
 199 heat transfers.

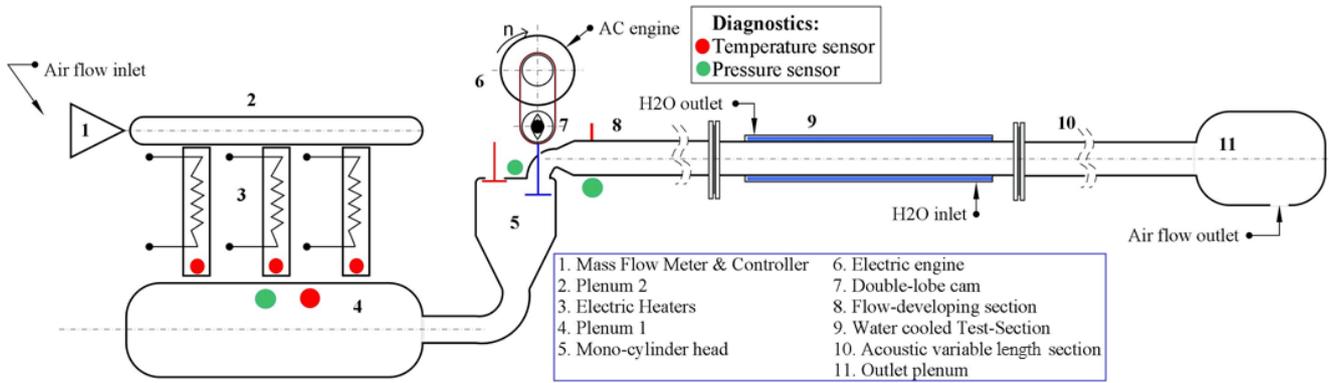


Figure 1: Scheme of the experimental apparatus

200 The test section consists of a double-wall pipe, in which the pulsating air and cold water flow in opposite directions,
 201 respectively in the internal and external parts of the pipe. Water cooling has the advantage of making it possible to manage the
 202 wall temperature and of having a homogeneous temperature field in the internal wall of the pipe. Inlet and outlet water
 203 temperatures are measured by two 0.5 mm sheathed K-type thermocouples for each section in order to evaluate the total time-
 204 averaged convective heat transfer Q_{conv} (see Eq. 10). The two thermocouples by section enable to control the homogeneity of
 205 the temperature field and allow to be confident in the temperature measurement at each section. A Kistler Type 2621F
 206 conditioning unit, with a maximum cooling power up to 1500 Watt, is used to provide a maximum water flow rate of 6.1
 207 L/min. As shown in Fig. 2, at a distance from the beginning of the test-section of 10 times the internal pipe diameter, several
 208 sensors are placed to measure the air velocity, temperature and pressure. In particular, in order to calculate the air bulk
 209 temperature, which corresponds to the integral of temperature on the cross section, in the first measuring section (A – A' in
 210 Fig. 2) four sheathed K-type thermocouples with a 0.5 mm diameter are placed at different distances from the wall, respectively
 211 1, 0.5, 0.125 and 0.0625 times the pipe radius. Furthermore, a Kulite pressure transducer is placed to measure the instantaneous
 212 static pressure of air. The same measuring configuration was used for the outlet section of the test-section. For the remaining
 213 sections, except the section B – B', only one 0.5 mm sheathed K-type thermocouple is placed at the centerline of the pipe.
 214 The thermocouples protrusion lengths have been chosen to compute accurately the bulk air temperature. The choice of the
 215 number and position of the thermocouples was based on the minimum number required to describe a presumed temperature
 216 radial profile. As, for a pulsating flow, a theoretical temperature profile is unknown, the temperature radial profile for a
 217 stationary turbulent pipe flow, developed by Johnk et al. [21] and described by Eq. 11, has been adopted:

$$\frac{(T_w - T)}{(T_w - T_{centerline})} = (1 + s) \left(\frac{y}{R_{in}} \right)^{1/m} - s \left(\frac{y}{R_{in}} \right)^{(1+s)/sm} \quad (11)$$

218 where T_w is the wall temperature, $T_{centerline}$ is the temperature at the pipe centerline, y is the distance from the pipe centerline,
 219 $s = 0.02$ for all Reynolds numbers, m is function of the Reynolds number and is equal to 5.7 for a Reynolds number of
 220 30000. From Eq. 8, it can be shown that, for the pipe radial interval ranging from $y/R_{in} = 0.1$ to the pipe centerline, the
 221 polynomial which approximates the temperature profile with the higher coefficient of determination is a 3rd order polynomial.
 222 Consequently, four temperature measurements were chosen since they correspond to the minimum number of values required
 223 to determine a 3rd order polynomial.

224 In the section B – B', in addition to a Kulite pressure transducer, a Constant Temperature Anemometer and two unsheathed
 225 micro-thermocouples are placed to measure the instantaneous radial profiles of air velocity and temperature. In particular, a
 226 Dantec 55P71 double wire probe (*HWA* in Fig. 2) was used to measure the air velocity magnitude and direction. The phase-
 227 averaged wire voltage was corrected to account for air temperature variations, and the flow reversal was detected thanks to the
 228 spatial orientation of the wire probe. When placed parallel to the flow direction, this probe is widely used to measure the spatial
 229 gradient of the flow velocity in wind tunnel application. In this study, it has been placed perpendicular to the flow in order to
 230 make possible the detection of the velocity direction by comparing the voltage signals of the two parallel wires, outdistanced
 231 of 0.4 mm. Because of the distance between the wires, flow reversal event, corresponding to the minimum of the wire voltage
 232 signal, shows a phase delay when the wire signals are compared. The wire that firstly senses the flow reversal has a phase of
 233 the minimum of the signal advanced in comparison to the other wire. Once the flow reversal is detected, the velocity is
 234 computed only from the first wire in the flow direction. Furthermore, a higher standard deviation of the phase-averaged velocity
 235 of one wire in comparison to the other one, resulting from the perturbations created by the first wire on the second one, was
 236 used as a second method to validate the detection of the flow direction.

237 The energy balance equation applied to the hot junction of a thermocouple describes the temperature difference between the
 238 gas and the hot junction with a thermocouple temperature delay due to the finite mass of the hot junction and due to the
 239 convective heat transfer between the fluid and the thermocouple. Assuming negligible thermal conduction and radiation in this
 240 equation, the hot junction temperature can be modelled as a first-order system, where the time constant represents the time the
 241 sensor requires to reach the gas temperature. For rapid temperature measurements, the time constant has to be compared to the
 242 dynamic of the flow properties variation in order to know if a compensation of the time delay has to be applied to the
 243 thermocouple signal to compute the real fluid temperature. A Kalman Filter method [22] was applied to the signal of the two
 244 micro-thermocouples (*T1-kf*, *T2-kf* in Fig. 2) to calculate in-situ the time constant of the sensor in order to correct the raw
 245 thermocouple measurements and to estimate the actual air temperature. The method was experimentally validated using a
 246 reference temperature signal measured with a cold wire with a diameter of 1 μm which has a frequency bandwidth up to 1
 247 kHz. The employed micro-thermocouples had a diameter respectively of 25 μm and 50 μm . An ad-hoc experimental
 248 characterization of the thermocouple had shown that for an unsheathed micro-thermocouple with a diameter of 25 μm the

249 sensor time-constant ranges from 25 msec to 10 msec for a flow velocity respectively of 2 m/s and 60 m/s, which is in
 250 agreement with the results of Tagawa et al. [23]. It was also observed that the time-constant evolution in function of the air
 251 velocity follows a logarithmic trend. An uncertainty less than 3% was found. The *HWA* and the micro-thermocouples are
 252 connected to the pipe with a concentric screw system with a thread pitch of 1 mm.
 253 As shown in Fig. 1, downstream the test-section, a pipe with a variable length was used. The total length of the pipe can vary
 254 from 3.61 meters to 5.91 meters. A NI-9035 cRIO and the LabView software were used to control all the devices in the
 255 experimental apparatus and to acquire data.

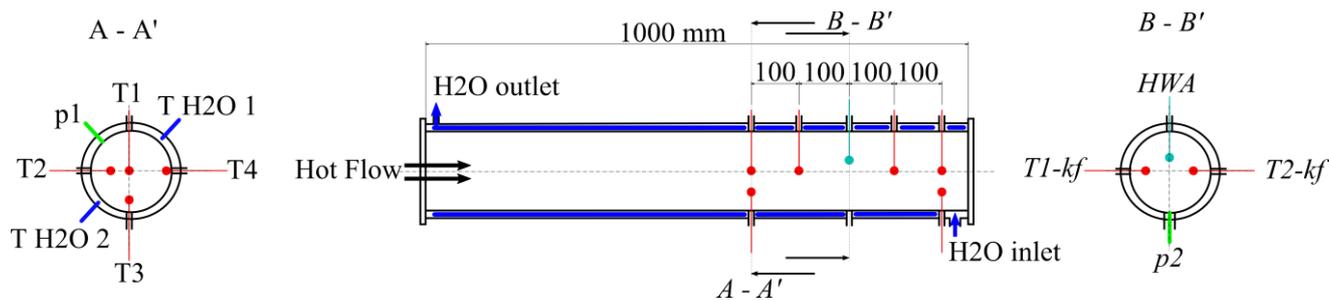


Figure 2: Test-section

3.2. Procedure

256
 257 The study of the impact of the flow pulsation on heat transfer phenomena was achieved by exciting a steady hot air flow with
 258 a pulsation frequency ranging from 0 to 95 Hz. The minimum attainable frequency is 10 Hz, for lower frequencies the electric
 259 motor is unable to perform constant speed revolution. A time-averaged mass flow rate of 110 kg/h is forced to flow through
 260 the mono-cylinder head. The centerline air temperature at the inlet of the test-section is maintained constant at 150°C for all
 261 experiments by regulating the electric power of the air heaters with a PID controller. These flow conditions correspond to a
 262 time-averaged Reynolds number of 30000, corresponding to turbulent flow conditions. The cooling water temperature at the
 263 inlet of the test-section is kept constant at 17.4°C for each experiment, with a maximum test-by-test variation of around 0.2°C.
 264 Accounting for the limitations of the hydraulic circuit used to cool the test-section and given the water mass flow rate, 17.4°C
 265 was the lowest achievable and repeatable temperature in order to maximize heat transfers from the hot air to the cold water to
 266 reach large temperature variations. The instantaneous mass flow rate profile imposed on the flow, in this manner, is dependent
 267 on the flow pulsation frequency and the pipe acoustic responses. This means that while the time-averaged component of the
 268 mass flow rate is kept constant for all the flow pulsations, the oscillating component is not constant. In order to extract coherent
 269 phenomena responsible for the convective heat transfer mechanisms, three different pipe lengths were investigated (3.69, 4.69
 270 and 5.91 meters). In this manner both the flow pulsation frequency and the acoustic resonance modes of the pipe were varied,
 271 making it possible to study the influence of one characteristic on the other. To characterize the acoustic resonance modes of
 272 the test-rig pipe for all the different lengths, a further experiment was conducted: after thermal stabilization of the experimental
 273 apparatus, by generating a steady hot air flow with an air temperature of 150°C at the inlet of the test section, the system was

274 subjected to a pressure impulse and then allowed to resonate. Instantaneous pressures were measured at four different axial
275 positions along the pipe and analyzed to compute the resonance mode frequencies of the system.

276 **4. RESULTS AND DISCUSSION**

277 **4.1. Acoustic characterization of the pipe**

278 When the experimental apparatus is excited with a rapid mass injection, a pressure wave travelling the pipe is produced. By
279 measuring the instantaneous air static pressure inside the pipe, it is possible to compute the acoustic resonance modes of the
280 system. The rapid mass injection was created by compressing the air inside the cylinder head and by opening the cylinder head
281 valve only one time during one pulsation cycle. The acoustic resonance frequencies were determined by computing the power
282 spectral density (PSD) of the pressure signal, which describes the distribution of power into frequency components. These
283 resonance frequencies correspond to the local maxima of the PSD. In this study, a second order low-pass filter with a cut-off
284 frequency of 1 kHz was applied to the sensor signals before calculating the PSD. Sensor signals were acquired at a frequency
285 of 20 kHz, high enough to respect the Nyquist-Shannon sampling theorem. Because the behavior of the pressure response to a
286 mass injection is qualitatively the same for each pipe length, in the following figures only the pressure signal for one pipe
287 length is shown. Fig. 3 shows the instantaneous pressure, measured at the center of the test-section, and Fig. 4 shows the PSD
288 of the signal obtained for a specific pipe length. The other pressures, measured at different pipe axial positions or for the other
289 pipe lengths, show the same PSD results and are not reported here. Local maxima of the PSD are identified by the red points
290 on Fig. 4. Table 1 shows the first six calculated resonance frequencies for the three tested pipe lengths. They result from the
291 same resonance phenomenon occurring in the pressure field when the pulsation source reaches a pulsation frequency equal to
292 one of the acoustic characteristic frequency of the system. Results show that the first resonance mode of the system is quite
293 different from the resonance frequency calculated for a pipe closed at one end and open to the surrounding air at the other end,
294 which is equal to $f_r = nc/4L$, where n is a natural integer number equal to 1 for the first resonance mode, c is the speed of
295 sound and L is the pipe length. This means that the volume of the outlet plenum is not large enough to be representative of an
296 open exhaust at pressure and temperature ambient conditions. This implies that a detailed numerical modeling of the
297 experimental apparatus should take into account this result through the description of the boundary conditions.

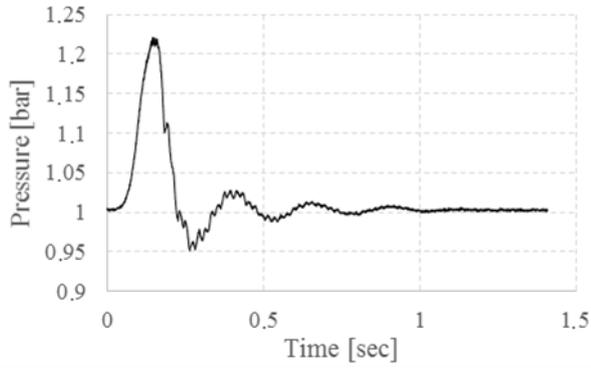


Figure 3: Instantaneous pressure, $L = 5.91m$

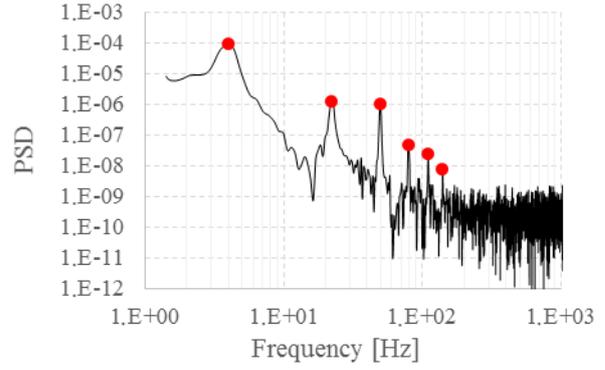


Figure 4: Power Spectral Density, $L = 5.91m$

| Resonance Mode | 1 | 2 | 3 | 4 | 5 | 6 |
|--------------------------------------|-----|------|------|-------|-------|-------|
| Resonance Frequency [Hz] $L = 3.69m$ | 6.4 | 36.1 | 79.6 | 126.3 | 176.2 | 239.8 |
| Resonance Frequency [Hz] $L = 4.69m$ | 4.7 | 26.2 | 63.1 | 101.3 | 138.9 | 175.8 |
| Resonance Frequency [Hz] $L = 5.91m$ | 3.9 | 22.2 | 49.5 | 80.5 | 110 | 139.8 |

Table 1: Resonance frequencies [Hz] of the test-bench for different pipe lengths

4.2. Validation of measurements procedure: steady state operation and boundary conditions

Given the difficulty of measuring the three terms in the left-hand side of Eq. 10 experimentally, the total time-averaged convective heat transfer Q_{conv} was evaluated from the water temperature measurements. In practice, a constant water volumetric flow rate of 4.74 L/min is forced to pass through an annular section, with an internal diameter of 67.3 mm and a thickness of 7 mm, in order to cool the pipe wall. In these operating conditions, the Reynolds number of the water flow is 788, leading to a laminar velocity profile under steady state conditions. Consequently, from the energy balance equation for the water, in which the convective heat transfer with the exterior ambient air was estimated to be less than 10 W and was therefore neglected, Q_{conv} can be solved as the time-averaged enthalpy difference of the water between the inlet and the outlet of the test-section. The computed Q_{conv} and the term A of Eq. 10 are plotted in Fig. 5 and Fig. 6 as a function of the pulsation frequency. It should be noted that pressure pulsation amplitude varies with the frequency (which is illustrated in Figs. 8 and 9 for two different pulsation frequencies). Uncertainties are depicted by error bars. The total uncertainty on convective heat transfers evaluation was determined by applying the error propagation methodology on the energy balance equation for the water flow: $\Delta Q_{conv}/Q_{conv} = \Delta \dot{m}_{H_2O}/\dot{m}_{H_2O} + \Delta(\Delta T_{H_2O})/\Delta T_{H_2O}$ where \dot{m}_{H_2O} is the water mass flow rate and ΔT_{H_2O} is the temperature difference between the inlet and outlet water temperatures. The relative uncertainty for Q_{conv} due to the mass flow rate measurement was 3% for all the pulsation frequencies. To evaluate the relative uncertainty due to the measurement of the water temperature difference, the experimental standard deviation of this temperature difference has been determined based on a set of ten tests for each frequency. From the instantaneous temperature measurements corresponding to an acquisition period of 2 minutes (the data rate was fixed at 10 Hz), the instantaneous inlet and outlet water temperatures were computed by averaging the two thermocouples measurement for each section. The instantaneous inlet/outlet temperature difference was then computed. From this temporal profile, the time-averaged value and the standard deviation were calculated for each test.

319 Finally, the mean temperature difference was obtained by averaging the ten time-averaged temperature difference and the
 320 standard deviation was taken as the largest standard deviation obtained for the set of ten tests.

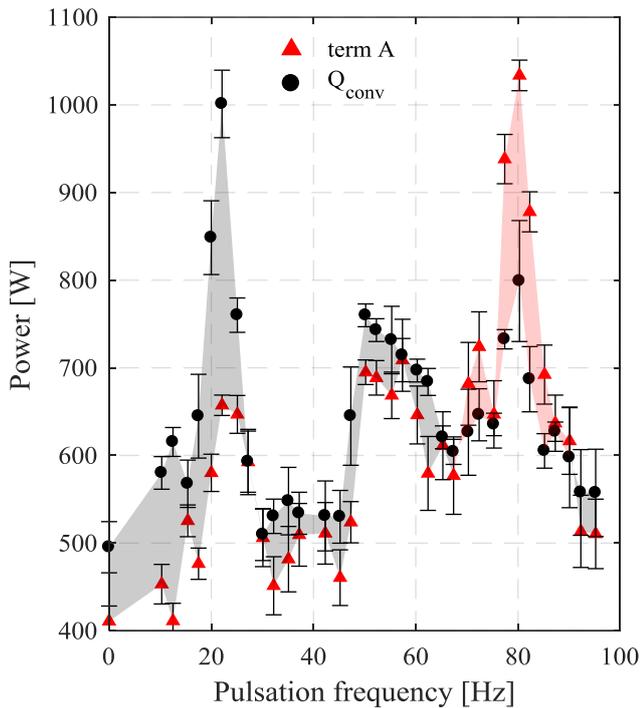


Figure 5: Energy balance equation terms, w/o dry ice, $L = 5.91m$

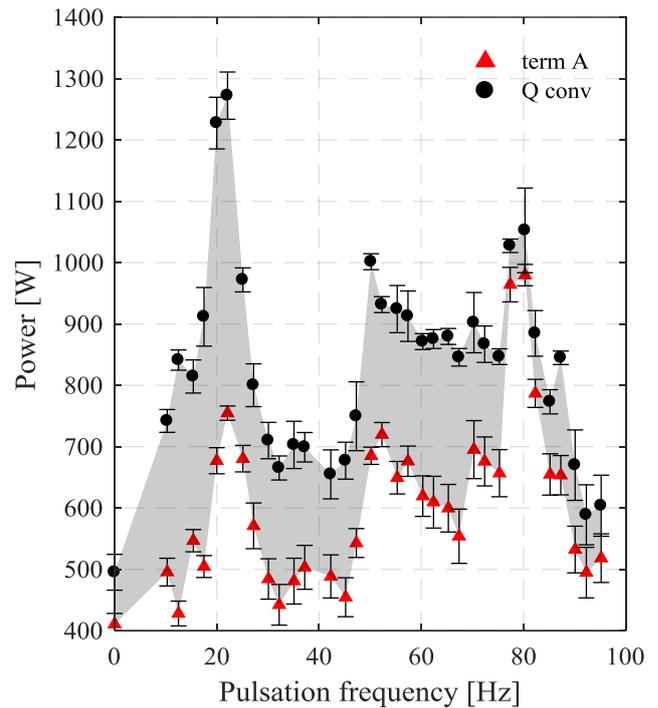


Figure 6: Energy balance equation terms, w/ dry ice, $L = 5.91m$

321 The air steady turbulent state (for the particular case with a time-averaged Reynolds number of 30000) is taken as the reference
 322 case. In such particular conditions, the term B in Eq. 10 is null, making it possible to compare Q_{conv} computed from
 323 experiments to the theoretical predicted one. The Nusselt number was evaluated using the Colburn and the Dittus-Boelter
 324 correlations and applying the Bhatti and Shah corrective coefficient [24] to take entry effects into account. For a Re of 30000,
 325 the space-averaged Nusselt number was found to be equal to 89.4 with the Colburn correlation and 92 with the Dittus-Boelter
 326 correlation. The total convective heat transfer linked to the Colburn correlation is 482 W, which corresponds to a difference
 327 of 2.7% with the experimental value. With the Nusselt number calculated from the Dittus-Boelter correlation the agreement is
 328 even better, with a difference of less than 1%. Consequently, the small differences between the theoretical predictions and the
 329 experiment validate the evaluation of the total convective heat transfer based on water temperature measurements. In order to
 330 validate the evaluation of the term A based on the experimental measurements, a 2D axisymmetric numerical model was
 331 developed using the commercial CFD code Fluent. The computational domain corresponds to the total length of the pipe and
 332 simulations were carried out by using a constant mass flow rate as inlet condition, by specifying an operating temperature at
 333 the inlet of the *developing section* and by using a constant pressure at the outlet section. Boundary conditions were defined as
 334 the measured wall temperatures along the pipe. An incompressible turbulent model, using the Reynolds stress model of
 335 turbulence to account for the effects of streamline curvature, swirl, rotation, and rapid changes in strain rate, was solved.
 336 Simulation result for Q_{conv} is 472 W, i.e. a difference of around 4% with the experimental result, which verifies the

337 computation accuracy. Comparison between experimental and simulated results for the term A shows a difference of less than
338 3%. These agreements validate the determination of the terms in the energy balance equation (Eq. 10) in steady state operation
339 using the experimental setup. Comparison between Q_{conv} and term A of Eq. 10, in the case of the steady turbulent flow, shows
340 that the calculation of the variation of the time-averaged enthalpy of air is not sufficient to correctly compute the total time-
341 averaged convective heat transfer: not taking into account the turbulent term may lead to an underestimation of the total time-
342 averaged convective heat transfer of around 15%.

343 It is observed from Fig. 5 that the difference between Q_{conv} and term A shows a significant variation as a function of the
344 pulsation frequency: the values of Q_{conv} can be higher (grey zone) than, equal to or lower (red zone) than the term A . Assuming
345 that the turbulent term in Eq. 10 cannot be negative, which physically would represent a creation of turbulent heat flux inside
346 the test-section, it can be concluded that a negative value of the term $\left\{ \overline{uT} \right\}_{\Sigma} \Big|_0^L$ may occur. It is of interest to note that a negative
347 value of term B (for instance for a pulsation frequency of 80 Hz) describes an energy contribution to the control volume,
348 through the outlet section, due to the pulsating component of the flow. This mechanism is experimentally made possible by
349 the combination of a warming of the air, once it has left the test-section, followed by a flow reversal. Downstream the test-
350 section, the steel pipe was not water cooled and the experimental measurements showed that, after the test-section outlet, the
351 air near the wall had a lower temperature than the external wall of the pipe. Moreover, the calculation of the particle
352 displacement from the instantaneous air velocity profile showed that, when a flow reversal occurs, the air passes through the
353 boundaries of the test-section several times. To experimentally demonstrate the importance of this mechanism, the steel pipe
354 downstream the test-section was cooled by using dry ice in contact with the external wall of the pipe. Dry ice is in solid state
355 at a temperature of -78.5°C and, once subjected to the ambient conditions, it sublimates, absorbing energy from the ambient
356 to ensure the phase change. In this manner, it was possible to impose an external wall temperature of around -40°C for all the
357 experiments. In these conditions, the results for Q_{conv} and term A are shown in Fig. 6. For each of the pulsation frequencies
358 investigated, it can be observed, on the one hand, that Q_{conv} is always higher than the variation of the time-averaged air
359 enthalpy and, on the other hand, that Q_{conv} exchanged during the use of dry ice is also always higher than that without dry ice
360 (see Fig. 5). As a consequence, these results show that, in the present experimental configuration, an energy contribution
361 coming from the ambient downstream the test-section can occur in the case of reversing pulsating flows.

362 **4.3. Time-averaged heat transfer calculation and analysis**

363 The time-averaged Nusselt number $Nu = h_{air}D/\lambda$ was evaluated by modelling the heat transfer between the hot air and the
364 water with three thermal resistances placed in series to describe the internal forced convection of the air, the radial conduction
365 through the wall and the forced convection of the water. The air convective heat transfer coefficient h_{air} assumes the following
366 form:

$$h_{air} = \left[2\pi R_{in} L \left(\frac{\Delta T_{lm}}{Q_{conv}} - \frac{\ln(R_{out}/R_{in})}{2\pi L \lambda_{wall}} - \frac{1}{h_{water} 2\pi R_{out} L} \right) \right]^{-1} \quad (12)$$

367 where R_{in} and R_{out} are the inner and outer radii of the annular section of the pipe, L is the length of the test-section, λ_{wall} is
 368 the thermal conductivity of the wall, ΔT_{lm} is the logarithmic temperature difference usually used in the case of heat exchangers
 369 and h_{water} is the convective heat transfer coefficient of the water. The latter was calculated according to Rohsenow et al. [25]
 370 with a Nusselt number of 4.4, and ΔT_{lm} was calculated according to the following equation [26]:

$$\Delta T_{lm} = \frac{(T_{b_{in}} - T_{H2O-in}) - (T_{b_{out}} - T_{H2O-out})}{\ln(T_{b_{in}} - T_{H2O-in}) - \ln(T_{b_{out}} - T_{H2O-out})} \quad (13)$$

371 where $T_{b_{in}}$ and $T_{b_{out}}$ are respectively the input and output bulk temperatures of the air, calculated as the surface integral of the
 372 four measured temperatures. $T_{H2O-out}$ and T_{H2O-in} are the outlet and inlet water temperatures. The relative time-averaged
 373 Nusselt number (defined as the ratio of the time-averaged Nusselt number for the pulsating flow to the corresponding one for
 374 a steady flow with the same time-averaged Reynolds number), Nu_{rel} , is reported on Fig. 7. Results show that, for the entire
 375 frequency range, the relative time-averaged Nusselt number is always greater than 1: flow pulsation has a positive effect on
 376 heat transfers so that an enhancement of the internal forced convection, in comparison to the steady flow, is observed. Cooling
 377 the pipe downstream the test section only adds a positive offset to the relative time-averaged Nusselt number on the entire
 378 frequency range, without modifying the finding of the heat transfer enhancement. This offset is due to the increase in Q_{conv}
 379 because of the increase in the heat energy advection by the oscillating component of the flow. It can also be observed that, at
 380 the frequencies of 22.5, 50 and 80 Hz, a local maximum of the relative time-averaged Nusselt number appears. Thanks to the
 381 previous acoustic characterization of the system, it is possible to identify these frequencies as the 2nd, 3rd and 4th resonance
 382 modes of the system (see Table 1). Similar results were found by varying the pipe length for this time-averaged Reynolds
 383 number: for each of the pipe lengths tested when the flow is excited, with a pulsation frequency equal to one of the resonance
 384 modes, a heat transfer enhancement occurs. However, the magnitude of the heat transfer enhancement differs for each
 385 resonance mode and pipe length. This kind of result suggests that a coherent heat transfer enhancement mechanism exists when
 386 the flow is in resonant state.

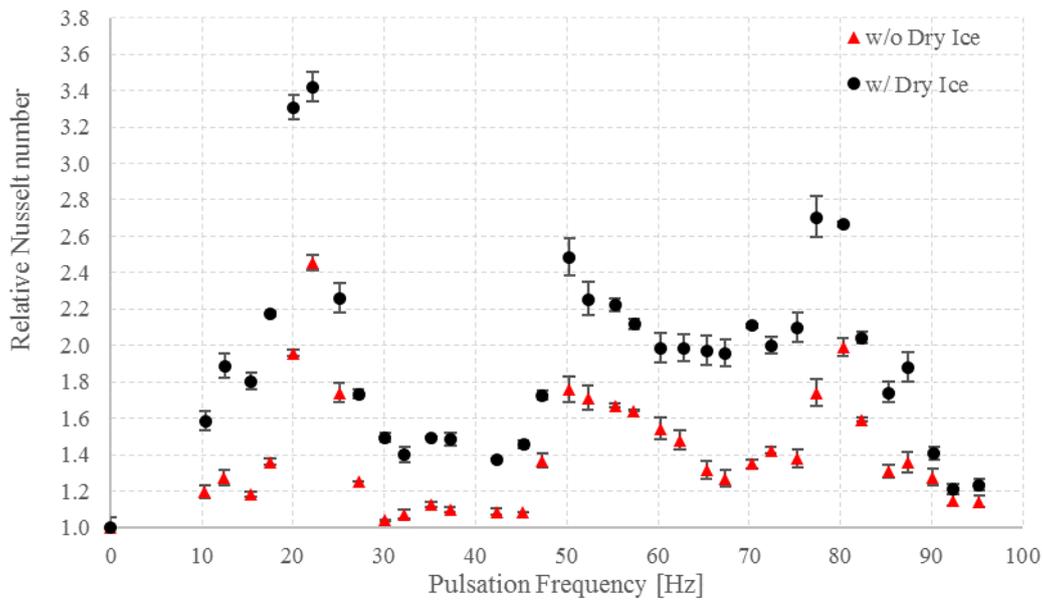


Figure 7: Relative time-averaged Nusselt number, $L = 5.91m$

387 In order to describe the mechanism of resonance and to investigate the phenomenon of heat transfer enhancement, the
388 instantaneous measurements of the radial profile of the air velocity and of the static pressure at the middle of the test-section
389 (section B – B' in Fig. 2) were analyzed. The phase-averaged axial component of the air velocity and the phase-averaged air
390 static pressure are shown in Figs. 8 and 9. Two particular frequencies of 10 and 22.5 Hz have been chosen. It can be observed
391 that at 22.5 Hz, which corresponds to the 2nd resonance mode, the amplitudes of velocity and pressure pulsations increase in
392 comparison with the 10 Hz case: this is the consequence of the system acoustic resonance. The pulsation source excites the
393 system with a mass injection that has a frequency equal to the system resonance mode (characterized and shown in the previous
394 section). At this frequency, the response of the system to sound waves becomes very large. Such acoustic resonance,
395 characteristic of the pressure field in the pipe, is transferred to the velocity field, as the phase-averaged velocity shows. Results
396 also show that flow reversals appear for both frequencies and that the pulsation amplitude of the velocity variation is
397 proportional to the pressure pulsation amplitude, in accordance with the acoustic linear theory for incompressible flows. It can
398 also be noted that the instantaneous profiles do not have a sinusoidal waveform, because of the system cannot be assumed to
399 behave as an open-end pipe due to the presence of the outlet plenum that modifies the reflection of the acoustic waves once
400 they reach the end of the straight pipe. Although the pressure and velocity profiles are not in agreement with the quarter-wave-
401 tube resonance, a phase delay (different from 45° for the case of a theoretical open-end pipe) between the maximum points of
402 the air velocity and of the pressure exists.

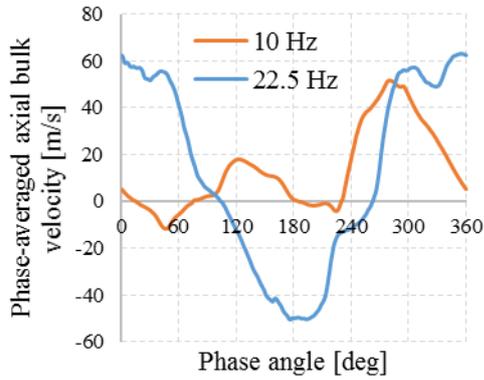


Figure 8: Axial component of the phase-averaged bulk air velocity, axial position B-B', $L = 5.91\text{m}$, $Re = 30000$

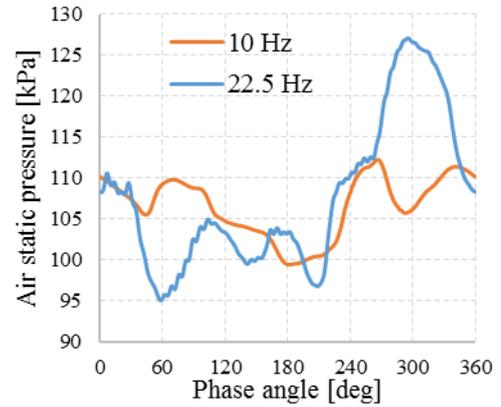


Figure 9: Phase-averaged air static pressure, axial position B-B', $L = 5.91\text{m}$, $Re = 30000$

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Using the radial instantaneous profiles of the axial velocity and of the temperature, it is possible to calculate the local values of the terms which spatial derivatives appear in Eq. 9. In order to be in agreement with the 1D approach, the physical quantities correspond to the integral of the measured profiles on the cross section. As previously reported, the wall cooling by dry ice does not change the main finding on the heat transfer enhancement. For this reason, in the remaining part of the paper, only the results for the experimental conditions corresponding to the case without the pipe cooling by dry ice have been reported. Instantaneous measurements were conducted only for pulsation frequencies below 30 Hz since the size of the thermocouple implies that measurements for higher frequencies cannot be exploited due to the time constant of the thermocouple that cannot be compensated using a Kalman filter method. The measurement procedure of the instantaneous profiles of air axial velocity, using hot-wire anemometry, and air temperature, with micro unsheathed thermocouples, requires a correction of the hot-wire signals to account for the temperature variation of air as well as a thermocouple signal compensation for the sensor time delay. For frequencies higher than 30 Hz, the implicit filtering of the real temperature variations, due to the thermal inertia of the sensors, makes compensation of the thermocouple signal unfeasible, nor does it allow temperature compensation of the hot-wire signal. The pulsation frequencies below 30 Hz were chosen as a function of particular heat transfer conditions: 10 Hz and 12.5 Hz were selected for the light impact on heat transfers, 20 Hz and 22.5 Hz because of their vicinity to the second resonance mode of the pipe, and 30 Hz because it has the smallest relative Nusselt number. Results for the terms $\{\overline{\tilde{u}\tilde{T}}\}_\Sigma$ and $\Delta\langle u \rangle/\bar{u}$ are reported in Fig. 10 and Fig. 11.

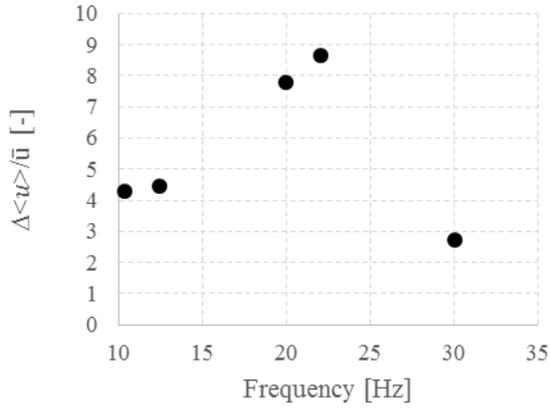


Figure 10: $\Delta\langle u \rangle / \bar{u}$ as a function of the pulsation frequency

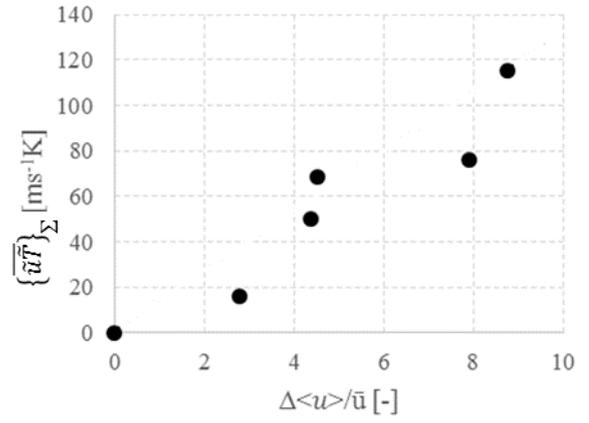


Figure 11: $\{\tilde{u}\tilde{T}\}_\Sigma$ as a function of $\Delta\langle u \rangle / \bar{u}$

420 $\Delta\langle u \rangle / \bar{u}$ represents the weight of the phase-averaged velocity oscillation amplitude on the time-averaged velocity
 421 component and characterizes the increase in the oscillating component of the flow due to pulsation. Results in Fig. 10 show a
 422 maximum of the term $\Delta\langle u \rangle / \bar{u}$ corresponding to the resonance frequency of the system, indicating that a flow resonance
 423 implies a wide oscillating component of the velocity. Furthermore since $\Delta\langle u \rangle / \bar{u}$ values are all higher than 1, it can be
 424 concluded that flow reversal occurs for all the frequencies analyzed in Fig. 10, as confirmed by the instantaneous measurements
 425 of the air velocity with the double wire probe. Results in Fig. 11 show the term $\{\tilde{u}\tilde{T}\}_\Sigma$ as a function of $\Delta\langle u \rangle / \bar{u}$. By linking
 426 Figs. 10 and 11 it can be observed that large velocity oscillations, which correspond to high $\Delta\langle u \rangle / \bar{u}$ values, are favored
 427 for pulsation frequencies close to the resonance frequency (the highest values of $\Delta\langle u \rangle / \bar{u}$ correspond to the frequencies
 428 around 22.5 Hz, i.e. the second resonance mode of the system) and lead to an increase of $\{\tilde{u}\tilde{T}\}_\Sigma$ as a function of $\Delta\langle u \rangle / \bar{u}$.
 429 Trend analysis observed on Fig. 11 can be performed, under the assumption that the axial velocity is constant inside a given
 430 cross section, by considering only the first harmonics of the oscillating components of the velocity and temperature profiles,
 431 which represent the biggest structures in the velocity and temperature fields containing the major part of the kinetic and thermal
 432 energies:

$$\tilde{u} = \Delta\langle u \rangle \sin(\omega t), \quad \tilde{T} = \Delta\langle T \rangle \sin(\omega t + \varphi) \quad (14)$$

433 From the definition of $\{\tilde{u}\tilde{T}\}_\Sigma$, it can be written:

$$\{\tilde{u}\tilde{T}\}_\Sigma = \Delta\langle u \rangle \Delta\langle T \rangle \int_{\text{period}} \sin(\omega t) \sin(\omega t + \varphi) dt \quad (15)$$

434 In the right-hand side of Eq. 15, the integral over a cycle period assumes a definite constant value, which leads to a
 435 proportionality of $\{\tilde{u}\tilde{T}\}_\Sigma$ with the product $\Delta\langle u \rangle \Delta\langle T \rangle$. Due to the complex turbulent pulsating problem, it is not possible
 436 to propose an exact analytical solution for $\Delta\langle u \rangle$ and $\Delta\langle T \rangle$. Nevertheless, according to the analytical development
 437 proposed for a pulsating laminar pipe flow by Fagri et al. [27], it can be assumed, using a first-order approximation in ω , that

438 $\Delta \langle T \rangle$ is proportional to the amplitude of the velocity oscillations. According to Eq. 14, $\{\overline{\tilde{u}T'}\}_\Sigma$ is then also found proportional
 439 to the amplitude of the velocity oscillations $\Delta \langle u \rangle$ in a quadratic form when only the biggest structures of the flow are
 440 considered. As the time-averaged velocity component \bar{u} is constant for all experiments, the experimental results reported on
 441 Fig. 11 seem to be consistent with this monotonic trend. As a conclusion of this analysis, the heat transfer enhancement for
 442 these frequencies, observed on Fig. 7, is therefore mainly attributed to a large oscillating component of the fluid velocity which
 443 increases the oscillating heat advection.

444 In order to characterize the impact of the pulsation frequency on the turbulent energy advection, the term C in Eq. 10 should
 445 be evaluated by measurements of the instantaneous radial profiles of air velocity and temperature. Although hot-wires have a
 446 bandwidth frequency up to 10 kHz, which is high enough to measure the turbulent variation of the air velocity, the
 447 thermocouples are not adapted to measure turbulent temperature fluctuations because of their long response time.
 448 Consequently, the term $\{\langle u'T' \rangle\}_\Sigma$ cannot be computed relevantly from the instantaneous experimental measurements. By
 449 exploiting the Reynolds analogy between momentum and heat transport equations, these limitations may be overcome. From
 450 the 2D momentum balance equation for an incompressible pulsating turbulent pipe flow, in which the phase and time averaging
 451 operators are applied, the following equation is obtained:

$$\rho \left(\underbrace{\frac{\partial \{\bar{u}\bar{u}\}_\Sigma}{\partial x}}_A + \underbrace{\frac{\partial \{\tilde{u}\tilde{u}\}_\Sigma}{\partial x}}_B + \underbrace{\frac{\partial \{\langle u'u' \rangle\}_\Sigma}{\partial x}}_C \right) = \mu \left(\frac{\partial^2 \{\bar{u}\}_\Sigma}{\partial x^2} \right) + \underbrace{\frac{\partial \bar{p}}{\partial x}}_E \quad (16)$$

452 It can be observed that the terms A , B and C , which correspond to the different components of the axial momentum of the flow,
 453 have the same mathematical form as the 3 terms in the left-hand side of Eq. 9 and differ only from these terms by a
 454 multiplicative coefficient which corresponds to the Prandtl number. The terms D and E correspond, respectively, to the viscous
 455 dissipation and to the pressure losses due to the viscous effect with the wall of the pipe. For a turbulent flow with a Prandtl
 456 number around 1, which means that the diffusion of momentum and thermal energy is similar, the Reynolds analogy implies
 457 that if a given flow field can be determined, the heat transfer characteristics can be found. In the experimental configuration
 458 of this work, the Prandtl number is about 0.69 (and is slightly dependent on temperature) so that the Reynolds analogy is
 459 commonly adopted in these conditions. Consequently, the terms in the left-hand side of Eq. 9 and Eq. 16 must exhibit a similar
 460 behavior. However, the limitations of the velocity and temperature measurements mean that the terms in Eq. 16 cannot be
 461 computed directly from the experimental measurements for a hot turbulent pulsating flow. Since the hot-wire signal
 462 compensation for the temperature effect was applied only to the phase-averaged signals it was not possible to extract the
 463 turbulent component of the air axial velocity from hot-wire signals. To overcome these measurement limitations, experiments
 464 were conducted for a cold turbulent pulsating flow. Considering temperature as a passive scalar field transported by the
 465 turbulent flow, it can be assumed that a cold pulsating flow has a velocity field behavior similar to that of a hot pulsating flow
 466 with the same time-averaged Reynolds number. The same time-averaged Reynolds number implies that the time-averaged

467 turbulent contribution to the momentum transport should be the same in hot and cold flow conditions. Using these assumptions,
 468 experiments were conducted for a pulsating turbulent pipe flow with a centerline time-averaged temperature of 25°C. In these
 469 experimental conditions, temperature variations due to the flow compression are negligible and do not disturb the hot-wire
 470 signal. Consequently, instantaneous hot-wire signals were not compensated for the temperature impact, and were directly
 471 decomposed according to Eq. 3 to allow the calculation of the turbulent component of the air axial velocity. To account for the
 472 decrease in the sound velocity because of the decrease in the flow temperature, the total pipe length was reduced so as to have
 473 the same resonance frequencies for the cold pulsating flow as for the hot pulsating flow. The computed terms A , B and C of
 474 Eq. 16 are reported in Table 2. Results show that in the case of resonant flow the time-averaged product of the oscillating
 475 component of the air axial velocity $\{\tilde{u}\tilde{u}\}_\Sigma$ increases. This behavior is in agreement with the variation of the term $\overline{\{\tilde{u}\tilde{T}\}}_\Sigma$, as
 476 previously shown. The turbulent term $\{\langle u'u' \rangle\}_\Sigma$ also shows a local maximum corresponding to the fluid resonant state,
 477 suggesting that pulsation frequency impacts the turbulent properties of the flow. Nevertheless, because of its lower order of
 478 magnitude than the term $\{\tilde{u}\tilde{u}\}_\Sigma$, its contribution to the momentum transport equation can be neglected in comparison to that
 479 of the term $\{\tilde{u}\tilde{u}\}_\Sigma$. These results can be similarly applied to the heat flux transport equation: the oscillating heat flux advection
 480 plays the main role in the convective heat transfer enhancement compared to the effect of the turbulent heat flux advection on
 481 heat transfer.

| Frequency [Hz] | $\{\overline{u\overline{u}}\}_\Sigma$ [m ² /s ²] | $\{\tilde{u}\tilde{u}\}_\Sigma$ [m ² /s ²] | $\{\langle u'u' \rangle\}_\Sigma$ [m ² /s ²] |
|----------------|---|---|---|
| 10.4 | 245.7 | 375.8 | 6.4 |
| 12.4 | 242.3 | 293.4 | 3.6 |
| 20.0 | 211.3 | 1291.8 | 22.1 |
| 22.1 | 231.3 | 1893.5 | 40.3 |
| 30.0 | 214.6 | 210.2 | 2.6 |

482 Table 2 : Terms A , B , C of Eq. 16

483 4.4. Towards a Nusselt correlation for pulsating turbulent pipe flows

484 The quasi-steady assumption is generally adopted to solve analytically or numerically the heat transfer problem for a pulsating
 485 flow. Due to the fact that the flow is assumed to behave as if it were steady at the instantaneous velocity, the instantaneous
 486 local heat transfer is assumed to balance the variation of the fluid local properties at any time in the cycle. However, this
 487 assumption implies that the flow reaches a fully developed equilibrium within a time much less than the cycle time and is thus
 488 only valid for flows with low frequencies or small oscillation amplitudes. Fig. 12 shows, for three frequencies, the velocity
 489 boundary layer thickness δ , calculated according Eq. 17 and nondimensionalized by the pipe radius.

$$\delta = \int_0^{R_{in}} \left(1 - \frac{\langle u_r \rangle}{\langle u_{centerline} \rangle} \right) dr \quad (17)$$

490 where $\langle u_r \rangle$ represents the axial component of the phase-averaged air velocity, which is function of the distance from the
 491 wall. It is shown that the dimensionless velocity boundary layer thickness reaches a value of 1 for several phases. For these
 492 phase angles, $\langle u_{centerline} \rangle$ is equal to zero and $\langle u_r \rangle / \langle u_{centerline} \rangle$ tends to infinity, which corresponds to flow reversal
 493 phases. Results also show that, for 10 Hz and 22.5 Hz, the velocity boundary layer thickness strongly fluctuates during the
 494 cycle. This is characteristic of flow structures that are not in equilibrium near the wall. As the thermal energy is transported by
 495 the velocity field, also the heat transfer is not in equilibrium. Consequently, the quasi-steady approach is not suitable for these
 496 two pulsation frequencies. With regard to the 30 Hz case, even if the quasi-steady assumption is still not valid, the fluctuation
 497 of the velocity boundary layer thickness is decreased compared to the two other frequencies.

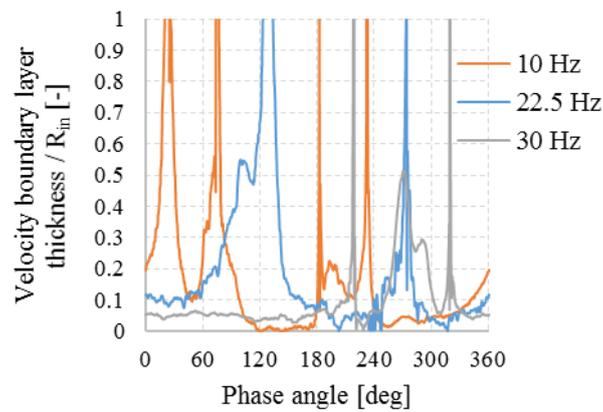


Figure 12: Velocity boundary layer thickness on the pipe
 internal radius, axial position B-B', $L = 5.91m$, $Re = 30000$

498 Whereas for a stationary turbulent pipe flow it has been largely demonstrated that the Nusselt number is a function of the
 499 Gz_T , Re and Pr numbers, for an unsteady convection process the Vaschy-Buckingham Π analysis applied in low-speed laminar
 500 flow conditions in a pipe shows that the time-averaged Nusselt number may depend on as many as seven dimensionless
 501 parameters [28]. Nevertheless, some of these parameters are not affected by the flow pulsation variation. As the time-averaged
 502 mass flow rate was kept constant for all experiments, the time-averaged Reynolds number is considered constant. In the same
 503 way, the Prandtl number for the air is slightly impacted by the temperature variations and is here assumed constant. The Graetz
 504 number, defined as $Gz_T = D/L_d Re Pr$, is useful in determining the thermally developing flow entrance length L_d for the
 505 description of entrance effects on laminar flow heat transfer. As it is applicable mainly to transient heat conduction in a laminar
 506 pipe flow, Gz_T is not a relevant parameter in this study, where the flow is in turbulent conditions, to describe the pulsation
 507 effect on heat transfer. Acoustic streaming, characterized by the parameter $U_0/\omega D$, is described by Jackson et al. [29] as a
 508 phenomenon that appears in resonant pulsating flows and generates a secondary time-averaged velocity component. The flow
 509 generated by acoustic streaming is characterized by large longitudinal counter-rotating vortices that are one quarter the length
 510 of the acoustic wavelength and occur at half-wavelength intervals. In the quarter wavelength between these vortical structures,
 511 acoustic streaming causes a velocity component towards the wall of the pipe, increasing the velocity near the wall. As shown
 512 by Lei et al. [30], axial velocity increases rapidly from zero at the wall to the maximum at the thickness of the acoustic layer

513 and decreases slowly with the further increase of the distance from the wall. Stagnation points, corresponding to a node, are
514 located at equally spaced intervals equal to one quarter of the pressure wavelength. In the present study, the flow was not
515 always in resonant state, which means that acoustic streaming was not generated at all the investigated pulsation frequencies.
516 When the flow was excited at a pulsation frequency equals to one of the resonant mode frequencies, e.g. 22.5 Hz, experimental
517 profiles of the time-averaged axial component of the air velocity were analyzed using the hot-wire anemometer measurements.
518 As the velocity profiles were measured at a fixed axial position of the test section (see Fig. 2), it was verified that this position
519 was not located on a node of the acoustic wave (at the pulsation frequency of 22.5 Hz, assuming a time-averaged temperature
520 of 150°C, the pressure wavelength assumes a value of about 18.3 m). Experimental profiles of the time-averaged axial
521 component of the air velocity showed that the axial velocity was always positive. This means that no recirculation cell was
522 formed in flow resonant state and that the acoustic streaming effect can be neglected in the Nusselt number correlation analysis.
523 In Fig. 13, the relative spatial and time-averaged Nusselt number is plotted versus the remaining relevant parameters: the
524 grouping $c/\omega D$, which is a measure of the wavelength of the applied pressure oscillation to the relevant streamwise length
525 scale (c is the velocity of sound) ; the Womersley number, defined as $Wo = (D/2)\sqrt{\rho\omega/\mu}$, which is a ratio of the channel
526 height to the Stokes boundary-layer thickness, is a dimensionless expression of the pulsatile flow frequency in relation to
527 viscous effects ; the amplitude ratio $\Delta\langle u \rangle/\bar{u}$, which is a measure of the amplitude of the velocity waveform. These
528 parameters are calculated from the experiments for all the investigated pulsation frequencies lower than 30 Hz. As previously
529 stated, results were computed for the experiments without pipe cooling by dry ice.

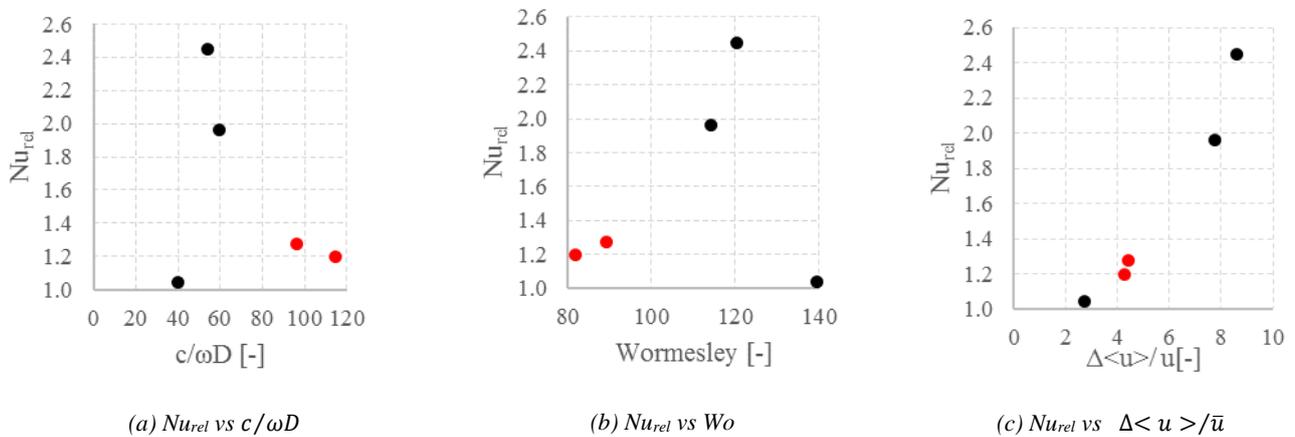


Figure 13: Relative time-averaged Nusselt number as a function of dimensionless parameters

530 The results in Fig. 13a show the dependence of the relative Nusselt number on the term $c/\omega D$: an increase in the pulsation
531 frequency (corresponding to a decrease in $c/\omega D$ since the speed of sound can be considered constant, as the air temperature
532 at the inlet of the test-section is maintained constant for all experiments) leads to an increase in Nu_{rel} till a maximum
533 corresponding to the second resonance mode frequency. This local maximum is followed by a decrease in the heat transfer
534 enhancement when the pulsation frequency is increased. In Fig. 13b, Nu_{rel} is plotted versus the Womersley number. According
535 to the definitions of the Womersley number and of the Stokes layer, an increase in the pulsation frequency leads to a thinning

536 of the viscous fluid boundary layer for a pulsating flow, which means that the velocity field is affected in a more restricted
537 zone near the wall with the increase in Wo . However, in terms of heat transfer enhancement, an optimal value of the Womersley
538 number exists for the maximization of the relative Nusselt number. This result indicates that the transport of energy and the
539 transport of the momentum are differently impacted by the variation of the Womersley number. This result agrees with the
540 experimental results in Ref. [31], where a local maximum in the heat transfer enhancement was found for a particular value of
541 Wo . In the present study, this optimal value of Wo corresponds to a pulsation frequency identified as a resonance frequency of
542 the system (the second resonance mode). The monotonic dependence of the relative Nusselt number on velocity amplitude
543 ratio $\Delta\langle u \rangle/\bar{u}$ is shown in Fig. 13c. For a constant time-averaged velocity, the relative Nusselt number increases with the
544 velocity oscillation amplitude. When the velocity oscillation amplitude is relatively low (for an amplitude ratio roughly less
545 than 4), the heat transfer is moderately enhanced, while Nu_{rel} increases rapidly when $\Delta\langle u \rangle/\bar{u} > 4$. A similar behavior was
546 found by Wang et al. [32] in their numerical study. Fig. 13c also shows that relatively constant values of Nu_{rel} are obtained for
547 relatively constant values of $\Delta\langle u \rangle/\bar{u}$, as depicted by the red markers. Results corresponding to these two same experiments
548 are also marked by the red points on Figs. 13a and 13b, and it can be noted that a slight variation in the relative Nusselt number
549 corresponds to significant variations in $c/\omega D$ and of Wo . These results are consistent with the developed analytical formulation
550 combined with the experimental measurements that have shown that a large oscillating velocity component is the most
551 important mechanism in the heat transfer enhancement. This mechanism is described by the amplitude of the velocity
552 waveform and, consequently, the characteristic term that should primarily be taken into account in a correlation to predict heat
553 transfers in the case of a pulsating flow is $\Delta\langle u \rangle/\bar{u}$. Nevertheless, even if Fig. 11 has shown that this term is directly linked
554 to the oscillating heat flux advection, this approach does not identify the physical mechanisms involved in the heat transfer
555 enhancement. This velocity waveform parameter is representative of the adopted 1D approach that does not quantify the energy
556 transport mechanisms in the radial direction that are integrated into the average over the cross section. For $\Delta\langle u \rangle/\bar{u}$ values
557 higher than 1, a periodic flow reversal initiates downstream and rapidly propagates upstream. As reversed fluid moves
558 upstream, it encounters forward-moving fluid and a radial velocity must be established to preserve continuity. The occurrence
559 of these mass ejections may increase the radial transport of energy and should be evidenced with a 2D analysis.

560 5. CONCLUSIONS

561 The results of an experimental investigation of convective heat transfer in pulsating turbulent flow in a pipe, with large-
562 amplitude oscillations, have been reported. The experimental apparatus was designed to generate a turbulent pulsating hot air
563 flow representative of engine exhaust flow conditions. Particular attention was paid to the calculation of the time-averaged
564 convective heat transfer by developing a 1D analytical formulation of the heat transfer problem for a pulsating turbulent pipe
565 flow. This development evidenced that whenever a direct measurement of the total time-averaged convective heat transfer is
566 not available, instantaneous measurements of the air velocity and temperature are required to correctly compute the terms

567 linked to the oscillating component of the flow in the energy balance equation. Experimental results showed that, in the case
568 of a reversed pulsating flow, a contribution to the energy balance equation through the boundaries, which impacts the total
569 convective heat transfers, may occur. Flow pulsation was found to enhance heat transfers in the entire range of frequencies
570 investigated. In particular, it was experimentally observed that, when the flow is excited with a frequency equal to a resonance
571 mode of the system, a strong increase in heat transfers occurs. Instantaneous measurements of air velocity and temperature
572 demonstrated that the increase in the energy axial advection due to the oscillating component of the velocity is the major cause
573 of the heat transfer enhancement. This behavior was observed to be independent of the pipe length, and therefore independent
574 of the acoustic resonance modes of the pipe. These results suggest that flow pulsation may be used as an active method for
575 heat transfer enhancement. The relative Nusselt number was found to be directly linked to the ratio between the velocity
576 oscillation amplitude and the velocity time-averaged component, suggesting that this parameter is representative of the
577 predominant heat transfer enhancement mechanism and should be included in a convective coefficient correlation for the
578 modeling of heat transfers, not only in engine exhaust systems but also for all engineering applications where heat is transferred
579 under pulsating conditions.

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583 **REFERENCES**

- 584 [1] R. Host, P. Moilanen, M. Fried, B. Bogi, Exhaust System Thermal Management : A Process to Optimize Exhaust
585 Enthalpy for Cold Start Emissions Reduction, SAE Tech. Pap. (2017).
- 586 [2] C. Sprouse III, C. Depcik, Review of organic Rankine cycles for internal combustion engine exhaust waste heat
587 recovery, *Appl. Therm. Eng.* 51 (2013) 711–722.
- 588 [3] J.E. Dec, J.O. Keller, Pulse combustor tail-pipe heat-transfer dependence on frequency, amplitude, and mean flow rate,
589 *Combust. Flame.* 77 (1989) 359–374.
- 590 [4] J.E. Dec, J.O. Keller, V.S. Arpaci, Heat transfer enhancement in the oscillating turbulent flow of a pulse combustor
591 tail pipe, *Int. J. Heat Mass Transf.* 35 (1992) 2311–2325.
- 592 [5] J.E. Dec, J.O. Keller, Time-resolved Gas Temperatures in the Oscillating Turbulent Flow of a Pulse Combustor Tail
593 Pipe, *Combust. Flame.* 80 (1990) 358–370.
- 594 [6] J.E. Dec, J.O. Keller, I. Hongo, Time-Resolved Velocities and Turbulence in the Oscillating Flow of a Pulse
595 Combustor Tail Pipe, *Combust. Flame.* 83 (1991) 271–292.
- 596 [7] Y. Xu, M. Zhai, L. Guo, P. Dong, J. Chen, Z. Wang, Characteristics of the pulsating flow and heat transfer in an elbow
597 tailpipe of a self-excited Helmholtz pulse combustor, *Appl. Therm. Eng.* 108 (2016) 567–580.

- 598 [8] M. Zhai, X. Wang, T. Ge, Y. Zhang, P. Dong, F. Wang, G. Liu, Y. Huang, Heat transfer in valveless Helmholtz pulse
599 combustor straight and elbow tailpipes, *Int. J. Heat Mass Transf.* 91 (2015) 1018–1025.
- 600 [9] J.T. Patel, M.H. Attal, An Experimental Investigation of Heat Transfer Characteristics of Pulsating Flow in Pipe, *Int.*
601 *J. Curr. Eng. Technol.* 6 (2016) 1515–1521.
- 602 [10] A.E. Zohir, Heat Transfer Characteristics in a Heat Exchanger for Turbulent Pulsating Water Flow with Different
603 Amplitudes, *J. Am. Sci.* 8 (2012) 241–250.
- 604 [11] S.A.M. Said, Al-Farayedhi A., M. Habib, S.A. Gbadebo, A. Asghar, S. Al-Dini, Experimental Investigation of Heat
605 Transfer in Pulsating Turbulent Pipe Flow, *J. Heat Fluid Flow.* (2005) 54.
- 606 [12] S.V. Nishandar, R.H. Yadav, Experimental investigation of heat transfer characteristics of pulsating turbulent flow in
607 a pipe, *Int. Res. J. Eng. Technol.* 2 (2015) 487–492.
- 608 [13] E.A.M. Elshafei, Safwat Mohamed, H. Mansour, M. Sakr, Numerical study of heat transfer in pulsating turbulent air
609 flow, *J. Eng. Technol. Res.* 4 (2012) 89–97.
- 610 [14] H.R. Kharvani, F.I. Doshmanziari, A.E. Zohir, D. Jalali-Vahid, An experimental investigation of heat transfer in a
611 spiral-coil tube with pulsating turbulent water flow, *Heat Mass Transf.* (2015).
- 612 [15] H. Ghaedamini, P.S. Lee, C.J. Teo, Enhanced transport phenomenon in small scales using chaotic advection near
613 resonance, *Int. J. Heat Mass Transf.* 77 (2014) 802–808
- 614 [16] C. Wantha, Effect and heat transfer correlations of finned tube heat exchanger under unsteady pulsating flows, *Int. J.*
615 *Heat Mass Transf.* 99 (2016) 141–148.
- 616 [17] Z. Guo, H.J. Sung, Analysis of the Nusselt number in pulsating pipe flow, *Int. J. Heat Mass Transf.* 40 (1997) 2486–
617 2489.
- 618 [18] C. Guardiola, P. Olmeda, B. Pla, P. Bares, In-cylinder pressure based model for exhaust temperature estimation in
619 internal combustion engines, *Appl. Therm. Eng.* 115 (2017) 212–220.
- 620 [19] J.M. Luján, H. Climent, P. Olmeda, V.D. Jiménez, Heat transfer modeling in exhaust systems of high-performance
621 two-stroke engines, *Appl. Therm. Eng.* 69 (2014) 96–104.
- 622 [20] W.C. Reynolds, A.K.M.F. Hussain, The mechanics of an organized wave in turbulent shear flow. Part 3. Theoretical
623 models and comparisons with experiments, *J. Fluid Mech.* 54 (1972) 263–288.
- 624 [21] R.E. Johnk, T.J. Hanratty, Temperature profiles for turbulent flow of air in a pipe: I, The fully developed heat-transfer
625 region, *Chem. Eng. Sci.* 17 (1962) 867–879.
- 626 [22] K. Kar, S. Roberts, R. Stone, M. Oldfield, B. French, Instantaneous exhaust temperature measurements using
627 thermocouple compensation techniques, *SAE Tech. Pap.* (2004) 169–190.
- 628 [23] M. Tagawa, K. Kato, Y. Ohta, Response compensation of fine-wire temperature sensors, *Rev. Sci. Instrum.* 76 (2005).
- 629 [24] J.H. Lienhard, *A heat transfer Textbook*, Third Edit, Phlogiston Press, 2008.

- 630 [25] W.M. Rohsenow, J.R. Hartnett, Young I. Cho, Handbook of Heat Transfers, Third Edit, McGraw-Hill, 1998.
- 631 [26] L. Angeles, Convective heat and mass transfer, (2013).
- 632 [27] M. Faghri, K. Javdani, A. Faghri, Heat transfer with laminar pulsating flow in a pipe, Lett. Heat Mass Transf. 6 (1979)
633 259–270.
- 634 [28] S.P. Kearney, A.M. Jacobi, R.P. Lucht, Time-Resolved Thermal Boundary-Layer Structure in a Pulsatile Reversing
635 Channel Flow, J. Heat Transfer. 123 (2001) 655–664.
- 636 [29] T.W. Jackson, K.R. Purdy, Resonant Pulsating Flow and Convective Heat Transfer, J. Heat Transfer. (1965) 507–512.
- 637 [30] H. Lei, D. Henry, H. Benhadid, Acoustic force model for the fluid flow under standing waves, Appl. Acoust. 72 (2011)
638 754–759.
- 639 [31] H. Li, Y. Zhong, X. Zhang, K. Deng, H. Lin, L. Cai, Experimental Study of Convective Heat Transfer in Pulsating Air
640 Flow inside Circular Pipe, Int. Conf. Power Eng. (2007) 880–885.
- 641 [32] X. Wang, N. Zhang, Numerical analysis of heat transfer in pulsating turbulent flow in a pipe, Int. J. Heat Mass Transf.
642 48 (2005) 3957–3970.