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# Online adaptive equivalent consumption minimization strategy for fuel cell hybrid electric vehicle considering power sources degradation

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## Abstract

The aim of this paper is to present an on-line adaptive equivalent consumption minimum strategy (AECMS) for fuel cell hybrid electric vehicle powered by fuel cell, battery and supercapacitor. In order to design the AECMS, an equivalent consumption minimum strategy (ECMS) without considering power sources degradation is firstly designed to decrease hydrogen consumption and degradation of power sources, which chooses fuel cell as the main power source to supply steady power, battery as the main energy storage source to buffer energy demand by vehicle and supercapacitor as the peak power supplier. A testbench is built to validate the developed ECMS. By contrastive experimental tests on ECMS, Rule Based Control Strategy (RBCS) and a Hybrid ECMS Operating mode control Strategy (HEOS) through the built test bench, hydrogen consumption of ECMS decreases 2.16% and 1.47% respectively and it also has the smoothest fuel cell current. Along with the degradation of fuel cell and battery, the charge sustenance objective of battery cannot be reached. Therefore, AECMS is finally designed to adjust equivalent factors and fuel cell dynamic current change rate along with the state of health (SOH) of fuel cell and battery, to make sure the charge sustenance of battery and prolong the lifetime of fuel cell. The method that on-line estimates their SOHs and the effects of their

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degradation on ECMS are also analyzed.

*Keywords:* Fuel cell hybrid electric vehicles (FCHEVs), Adaptive equivalent consumption minimum strategy (AECMS), Degradation of power sources, Experimental validation, Online estimation of SOH

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### **Nomenclature**

$F$	Faraday constant
$I_{FC}$	Fuel cell current
$m_{BA}$	Battery equivalent hydrogen consumption
$m_{FC}$	Fuel cell hydrogen consumption
$m_{SC}$	Supercapacitor equivalent hydrogen consumption
$N_{cell}$	The cell number of fuel cell stack
$P_{BA}$	Battery output power
$P_{FC}$	Fuel cell power
$P_{SC}$	Supercapacitor output power
$R$	Ideal gas constant
$SOH_{BA}$	Battery SOH
$SOH_{FC}$	Fuel cell SOH
$T$	The temperature of fuel cell
$V_{FC}$	Fuel cell voltage

## 1. Introduction

Global warming, air pollution and exhaustion of fossil fuel are gradually attracting people's concern on environmental problems and energy crisis. Fuel cell hybrid electric vehicle (FCHEV) chooses fuel cell and energy storage sources (ESSs) as the power sources, takes hydrogen energy as the fuel and has no air pollutants, which make it a good candidate to solve these problems [1]. Compared to other kinds of fuel cell, proton exchange membrane fuel cell (PEMFC) has many advantages such as high power density, high efficiency and lower operating temperature, which makes it suitable for vehicle application [2]. Fast dynamic power requirement on fuel cell increases its degradation and leads to its fast failure, therefore ESSs are needed to hold the high frequency parts of power to let PEMFC supply steady power. Lithium-ion battery on account of its high energy density can be used as the main energy storage source of FCHEV. Supercapacitor has high power density, which is the great choice of peaking power device. Therefore, Lithium-ion battery and supercapacitor are chosen as the hybrid storage system for the FCHEV in this paper.

An energy management strategy (EMS) is needed to split the power demand among different power sources based on their special characteristics. EMSs can be classified as rule based control strategy (RBCS) and optimization based control strategy [3]. RBCS, including deterministic rule based strategy (DRBS), frequency based strategy and fuzzy rule based strategy (FRBS), depends on a set of rules to determine the control actions at each sample time [4, 5]. Thermostat control strategy [6], load following strategy [7], PID control approach [8], operating mode control [9, 10], stiffness coefficient mode [11] and state machine strategy [12] all belong to DRBS and have ever been used as the EMS of FCHEV. DRBS relies on deterministic rules to control the split of power requirement among different power sources, which are designed according to the designer's experience. FRBS is similar to DRBS. But its rules are in form of "if-then" and its states are described by different membership functions [13, 14]. Frequency based strategy is designed based upon different dynamic characteris-



31 tics of power sources. Compared to the optimization strategy, RBCS is simple  
32 and can be easily applied into the reality, but optimal results are more hardly  
33 reached [15].

34 Optimization based control strategy including local optimization control  
35 strategy and global optimization control strategy uses analytical or numeri-  
36 cal optimization algorithms to seek for optimal results at a specified sampling  
37 time [16, 17]. Global optimization control strategy uses the whole drive cycle as  
38 the optimization time and the knowledge about the power demand in sampling  
39 time is known in advance. Minimizing cost function is reached through optimal  
40 algorithms like dynamic programming [18], genetic algorithmic [19], liner pro-  
41 gramming [20] and particle swarm programming [21]. High computation cost  
42 and relying on road knowledge discourage its on-line usage. Local optimiza-  
43 tion strategy calculates optimal results in instantaneous time to overcome these  
44 problems.

45 Equivalent consumption minimization strategy (ECMS), as one kind of local  
46 optimization strategy, is used for FCHEV to split the power among PEMFC,  
47 battery and supercapacitor. The vast majority of research in the literature  
48 using ECMS focus on architectures using only two sources. Moreover, literature  
49 papers which use the same architecture as the one described in this paper neglect  
50 the supercapacitor equivalent hydrogen consumption, which is taken as null  
51 [22, 23, 24]. It is not only counter to the aim of minimizing whole hydrogen  
52 consumption at every sample time but also increases the complication of EMS,  
53 because an additional control strategy is needed to calculate supercapacitor  
54 reference power. Therefore, an ECMS calculating whole hydrogen consumption  
55 of all three power sources is designed for the FCHEV in this paper and is  
56 validated by the experimental test bench [25].

57 During the lifetime of FCHEV, the power sources degradation appears. As  
58 the main power source of FCHEV, the output power of fuel cell at corresponding  
59 current decreases along with degradation and its maximum efficiency point of  
60 operation also changes. The capacity of battery decreases, and its resistance in-  
61 creases along with its degradation, which decreases the output voltage of battery

62 [26]. Besides, precise estimation of battery SOC is not possible without consid-  
63 ering its degradation. Therefore, fuel cell and battery ageing models are needed  
64 to monitor their state of health (SOH) to ensure the performance, safety, avail-  
65 ability and reliability of power sources [27]. Supercapacitor lifetime is far larger  
66 than the vehicle and its other power sources, so its degradation can be neglected  
67 [28]. Then degradation of power sources is defined as fuel cell degradation and  
68 battery degradation in the following parts of the paper.

69 Using control parameters of EMS with healthy power sources into the situa-  
70 tion of degraded power sources cannot reach optimal results for EMS and evenly  
71 cannot make sure the normal operation of the vehicle [29]. The equivalent fac-  
72 tor (EF) of ECMS decides the conversion from electrical energy of ESSs to the  
73 equivalent fuel consumption, which can ensure the optimal control solution can  
74 be reached with the aim of minimizing the whole fuel consumption [30]. The  
75 degradation of power sources also affects the optimal EF value of ECMS and  
76 its deviation may even lead to unacceptable operations of the vehicle except for  
77 suboptimal results [31]. Therefore, the effects of power sources degradation on  
78 ECMS are analyzed in the paper, which is rarely studied by researchers. A novel  
79 adaptive ECMS (AECMS) is designed through tuning equivalent factor (EF)  
80 based on power sources degradation states. The dynamic power change rate of  
81 fuel cell is also adjusted along with power sources degradation to decrease fuel  
82 cell degradation rate. The gap between health management of power sources and  
83 EMS is filled by integrating the prognostics into the developed AECMS.

84 This paper is organized as follows: section two describes the vehicle architec-  
85 ture and the model of the powertrain including fuel cell, battery, supercapacitor  
86 and DC/DC converters. In the third section, SOH estimation of power sources is  
87 explained. In the fourth part, ECMS strategy in the health state is designed and  
88 validated through experimental test bench. In the fifth part, the effects of fuel  
89 cell degradation and battery degradation on ECMS are analyzed respectively  
90 and new AECMS strategy is designed. Finally, conclusions are drawn.

91 **2. Vehicle models**

92 *2.1. Powertrain architecture*

93 The architecture of FCHEV is shown in Figure 1. Through the reaction  
 94 between hydrogen and oxygen, PEMFC as the main power source transforms  
 95 the chemical energy into electric energy and it is connected to DC bus through a  
 96 unidirectional DC/DC converter. Supercapacitor as the peak power supplier is  
 97 connected to DC bus via a bidirectional DC/DC converter. Lithium-ion battery  
 98 as the main energy storage source is directly connected to DC bus to hold the  
 99 DC bus voltage.

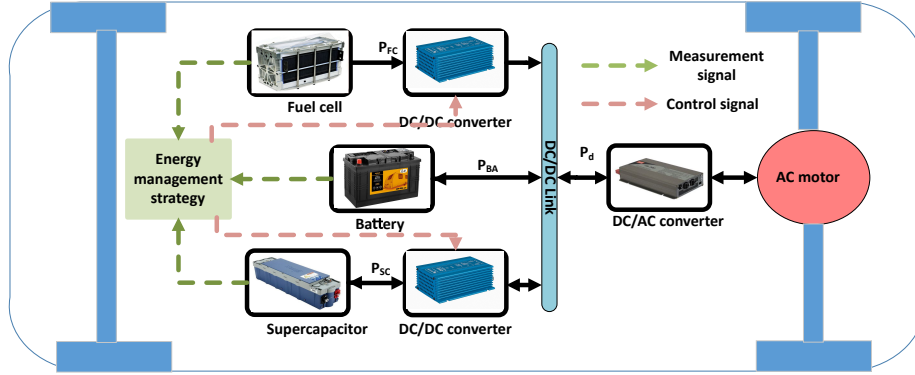


Figure 1: Powertrain architecture

100 The vehicle propulsion power at wheel can be calculated through the longi-  
 101 tudinal dynamics of a road vehicle as equation (1) [32]:

$$P_{cycle}(t) = v \left( m_v(t) \frac{d}{dt} v(t) + F_a(t) + F_r(t) + F_g(t) \right) \quad (1)$$

102 where  $P_{cycle}$  is the power demand from drive cycle,  $F_r$  the rolling friction,  
 103  $F_a$  the aerodynamic friction, and  $F_g$  the force caused by gravity when driving  
 104 on slope road:

$$F_a = \frac{1}{2} \rho A C_x v^2 \quad (2)$$

$$F_r = m_v C_r g \cos(\alpha) \quad (3)$$

$$F_g = m_v g \sin(\alpha) \quad (4)$$

105 where variable  $v$  is the speed of the vehicle,  $\rho$  is the air density,  $m_v$  the  
 106 vehicle mass,  $A$  front surface of the vehicle,  $g$  gravitational acceleration,  $C_x$  the  
 107 drag coefficient,  $C_r$  the aerodynamic drag coefficient, and  $\alpha$  the angle defining  
 108 the slope of the road.

109 The power demand  $P_{demand}$  on the DC bus should be met by three power  
 110 sources and its value can be calculated through equation (5).

$$P_{demand} = \frac{P_{cycle}}{\eta_{DC/AC} * \eta_{motor}} \quad (5)$$

111 where  $\eta_{DC/AC}$  is converter efficiency of DC/AC connected to a motor,  $\eta_{motor}$   
 112 is motor efficiency.

### 113 2.2. Fuel cell stack model

114 1.2 kW PEMFC is chosen as the main power source of FCHEV. The fuel cell  
 115 stack voltage can be described as equation (6) [33].

$$E = N_{cell} * (E_{rev} - E_{act} - E_{ohm} - E_{con}) \quad (6)$$

116 where  $E_{rev}$  is the thermodynamic reversible potential,  $E_{act}$  the activation  
 117 losses,  $E_{ohm}$  the ohmic losses,  $E_{con}$  the concentration losses.

118 The thermodynamic reversible potential can be calculated as equation (7)  
 119 [34].

$$E_{rev} = E_0 - 0.85e^{-3}(T - T_c) + \frac{RT}{2F} \ln(\sqrt{P_{O_2}} P_{H_2}) \quad (7)$$

120 where  $E_0$  is the reversible nearest potential for a single cell,  $T$  is the tem-  
 121 perature of the cell,  $T_c$  is the temperature correction offset,  $P_{O_2}$  and  $P_{H_2}$  are  
 122 partial pressure of oxygen and hydrogen.

123 The dynamic activation losses  $E_{act}$  can be described as equation (8).

$$\frac{dE_{act}}{dt} = \frac{I_{FC}}{C_{dl}} \left( 1 - \frac{E_{act}}{\eta_{act}} \right) \quad (8)$$

124 where  $C_{dl}$  is the single-fuel cell double-layer capacitance.

125 The static activation losses  $\eta_{act}$  can be simplified as equation (9), when  $\eta_{act}$   
126 is large.

$$\eta_{act} = \frac{RT}{2\alpha F} \ln \left( \frac{I_{FC}}{I_0} \right) \quad (9)$$

127 where  $I_0$  is the exchange current density,  $\alpha$  is the symmetry factor.

128 The ohmic losses  $E_{ohm}$  can be obtained through equation (10).

$$E_{ohm} = I_{FC} * R_{FC} \quad (10)$$

129 where  $R_{FC}$  is the internal resistance.

130 Concentration losses  $E_{con}$  can be calculated as equation (11).

$$E_{con} = -B * \ln \left( 1 - \frac{I_{FC}}{I_{max}} \right) \quad (11)$$

131 where B is an empirical constant.  $I_{max}$  is the maximum allowed current.

132 According to the above function, the relation between fuel cell voltage and  
133 current can be defined.

134 Fuel cell transforms the chemical energy into electrical energy through the  
135 reaction between hydrogen and oxygen [35]. The theoretical efficiency of energy

136 conversion is defined as the ratio between the useful energy output and the en-

137 ergy input. The energy output of fuel cell is the output electrical energy and

138 the energy input is the energy contained in the mass of hydrogen supplied [36].

139 Besides that, some auxiliary equipment is needed to make sure the normal oper-

140 ation of fuel cell system, like an air compressor, cooling fan and control border.

141 The energy consumed by auxiliary equipment belongs to auxiliary efficiency.

142 The whole fuel cell system efficiency can be defined as equation (12).

$$\eta_{FCS} = \eta_{TH} * \eta_{AUX} = \frac{V_{FC}}{1.254} \left( \frac{P_{FC} - P_{AUX}}{P_{FC}} \right) \quad (12)$$

143 where  $\eta_{TH}$  is the theoretical efficiency,  $\eta_{AUX}$  is auxiliary efficiency,  $P_{AUX}$   
 144 auxiliary power.

145 The air compressor system and cooling system of fuel cell system are the  
 146 main auxiliary equipment for this fuel cell system [37], which are built based on  
 147 the experimental test. Power demand by the compressor is shown as equation  
 148 (13).

$$P_{cp} = \frac{C_P T_{air}}{\eta_{mec} \eta_{mot}} \left( \left( \frac{P_{out}}{P_{in}} \right)^{\frac{\gamma-1}{\gamma}} - 1 \right) F_{cp} \quad (13)$$

149 where  $P_{cp}$  is air compressor power,  $\eta_{mec}$  represents the compressor mechan-  
 150 ical efficiency,  $C_P$  heat capacity of air,  $T_{air}$  inlet air temperature,  $\eta_{mot}$  is the  
 151 efficiency of the compressor motor,  $P_{in}$  and  $P_{out}$  are input and output air pres-  
 152 sures respectively,  $\gamma$  is the ratio of the specific heat of air.

153  $F_{cp}$  is the compressor air flow rate is decided by fuel cell output current as  
 154 equation (14).

$$F_{cp} = S * M_{air} \frac{N_{cell} * I_{FC}}{4X_{O_2} * F} \quad (14)$$

155 where  $S$  is the stoichiometric ratio,  $M_{air}$  is the number of air moles,  $X_{O_2}$   
 156 oxygen molar fraction.

157 From the experimental measurement, the speed of controlled cooling fan is  
 158 divided into a constant speed zone and regulated zone, depending on the stack  
 159 temperature, as shown in the Table 1 [38].

160 It should be noticed the DC/DC converter connected to fuel cell system  
 161 affects the output power of fuel cell system on DC bus. So, its efficiency is also  
 162 included in fuel cell system. So its efficiency can be calculated as Figure 2.

163 It can be observed that there is a maximum efficiency point 42.83% at fuel  
 164 cell current 9.5A. The high efficiency zone is defined from 4.5A to 20A in  
 165 red color, where the value exceeds 40%. In order to reduce the final hydrogen  
 166 consumption, fuel cell should be operated to seek maximum efficiency point at  
 167 this zone. When fuel cell requirement is lower than 4.5A, the fuel cell is turned  
 168 off to save hydrogen.

Table 1: Cooling fan speed and stack temperature

<b>In nonregulated zone</b>	
Stack temperature range	Fan speed (%)
to 50.5 °C	35
From 50.5 °C to 53.5 °C	36
From 53.5 °C to 55.5 °C	37
From 55.5 °C to 58.5 °C	38
From 58.5 °C to 60.5 °C	39
From 60.5 °C to 63.5 °C	40
From 63.5 °C to 65.5 °C	41
From 65.5 °C to 67.5 °C	42
<b>In regulated zone</b>	
Fan speed regulator start temperature: 67.5 °C	
Fan speed regulator stop temperature: 65.0 °C	

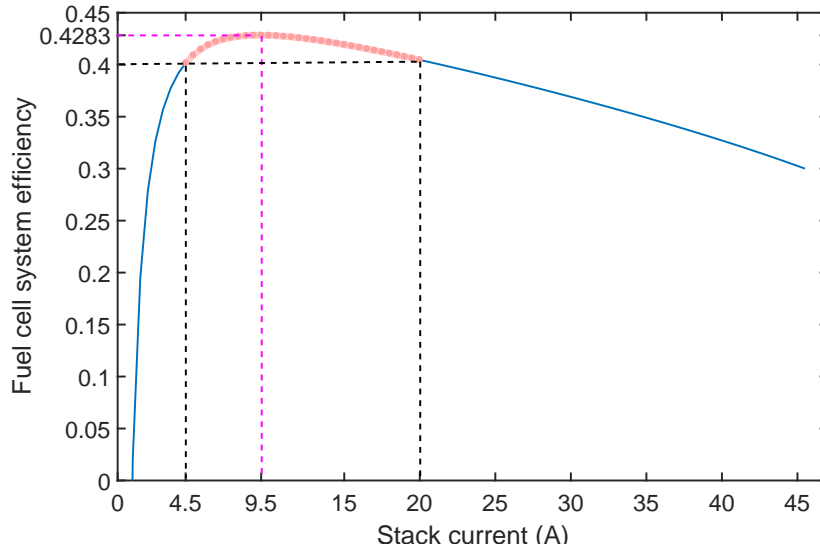


Figure 2: Fuel cell system efficiency curve along with stack current

169 The hydrogen consumption rate can be defined by fuel cell current as the  
 170 following equation (15) [39]:

$$m_{H_2} = \int_0^t \frac{M_{H_2} N_{cell}}{2F} I_{FC}(t) dt \quad (15)$$

171 where  $m_{H_2}$  is the hydrogen mass rate,  $M_{H_2}$  the hydrogen molar mass.

### 172 2.3. Battery model

173 An electrochemistry-based Lithium-ion battery model is used in this paper.  
 174 Compared to the empirical model, multi-physical model and equivalent circuit  
 175 battery model, the new model not only can make sure the low cost of compu-  
 176 tation but also meets the high accuracy requirement [40].

177 Li-ion battery is composed of a positive electrode, a negative electrode and  
 178 an electrolyte. During discharge, lithium ions ( $Li^+$ ) de-insert from the negative  
 179 electrode consisting of lithiated carbon( $Li_xC$ ), diffuse through the separator  
 180 consisting of electrolyte towards the positive electrode and intercalate in the  
 181 positive electrode consisting of lithium cobalt oxide ( $Li_xCoO_2$ ) [41]. The charg-  
 182 ing process is the reverse of discharging process. The voltage terms of the  
 183 battery are summarized in Figure 3. The overall voltage of battery  $V(t)$  is the  
 184 difference between the positive current collector potential and negative current  
 185 collector potential, and resistance losses at the current collectors are taken as  
 186 zero [42]. According to Figure 3, the battery voltage can be expressed as the  
 187 equation (16)

$$V(t) = V_{U,p} - V_{U,n} - V_{S,p} - V_{S,n} - V_e - V_{O,n} - V_{O,p} \quad (16)$$

188 where  $V_{U,p}$ ,  $V_{U,n}$  are equilibrium potentials at the positive current collector  
 189 and the negative current collector respectively,  $V_{O,n}$  and  $V_{O,p}$  are surface over-  
 190 potentials due to charge transfer resistance at the positive and negative current  
 191 collector,  $V_{S,p}$ ,  $V_{S,n}$  and  $V_e$  are the voltage drop due to solid phase ohmic resis-  
 192 tance at the positive and negative current collector and the electrolyte ohmic  
 193 resistance respectively. Each of these voltages is described in detail in the fol-  
 194 lowing parts.



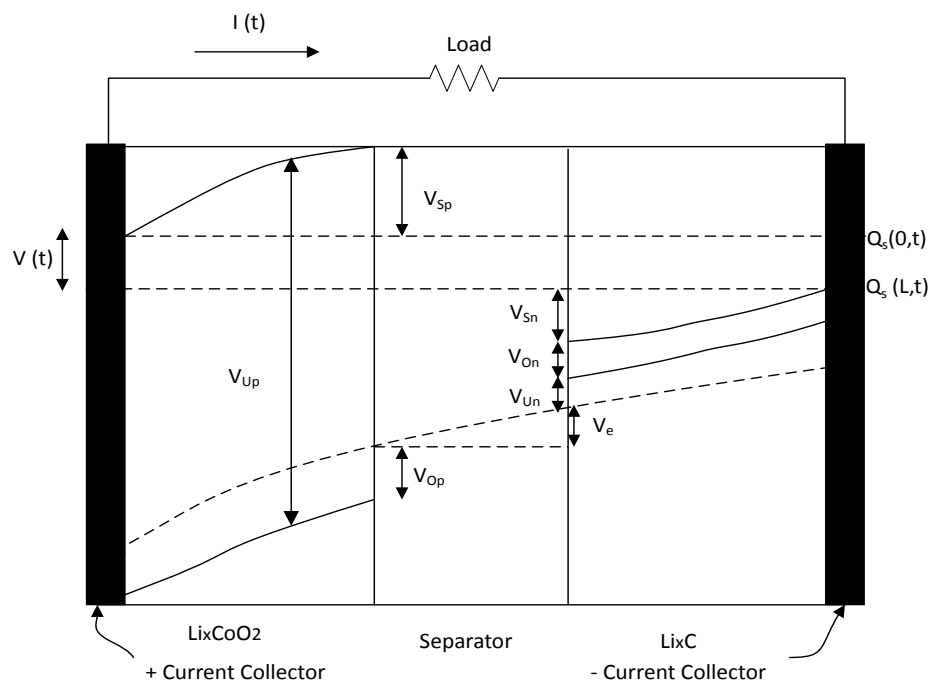


Figure 3: Battery model

195 Equilibrium potential can be calculated as the Nernst equation (17):

$$V_{U,i} = U_0 + \frac{RT}{nF} \ln \left( \frac{1 - x_i}{x_i} \right) + V_{act,i} \quad (17)$$

196 where  $i$  stands for the electrode ( $n$  for negative and  $p$  for positive),  $U_0$   
 197 is reference voltage,  $T$  is electrode temperature,  $n$  the number of electrons  
 198 transferred in the reaction,  $V_{act,i}$  the activity correction term (0 in the ideal  
 199 condition), which can be defined as equation (18) [43].

$$V_{act,i} = \frac{1}{nF} \left( \sum_{k=0}^{N_i} A_{i,k} \left( (2x_i - 1)^{k+1} - \frac{2x_i k (1 - x_i)}{(2x_i - 1)^{1-k}} \right) \right) \quad (18)$$

200  $x_i$  can be defined as equation (19).

$$x_i = \frac{q_i}{q_{max}} \quad (19)$$

201 where  $q_i$  the amount of Li ions in electrode  $i$ ,  $q_{max} = q_p + q_n$ , the total  
 202 amount of Li ions.  $x_p + x_n = 1$ , when fully charged  $x_p = 0.4$  and  $x_n = 0.6$ .  
 203 When fully discharged,  $x_p = 1$  and  $x_n = 0$ .

204 The equilibrium voltage is directly decided by the amount of charge in the  
 205 electrodes. Each electrode can be divided into surface layer (subscript  $s$ ) and  
 206 buck layer (subscript  $b$ ). So the relationship can be determined by:

$$q_p = q_{s,p} + q_{b,p} \quad (20)$$

$$q_n = q_{s,n} + q_{b,n} \quad (21)$$

$$q_{max} = q_{s,n} + q_{b,n} + q_{s,p} + q_{b,p} \quad (22)$$

207 In the buck layer, the concentration of Li ion is nearly even but inside the  
 208 surface layer, the concentration changes drastically. The concentrations of Li  
 209 ions in the surface layers are described as:

$$c_{b,i} = \frac{q_{b,i}}{v_{b,i}} \quad (23)$$

$$c_{s,i} = \frac{q_{s,i}}{v_{s,i}} \quad (24)$$

210 where  $q_{s,i}$  and  $q_{b,i}$  are the charge at different layers,  $v_{b,i}$  and  $v_{s,i}$  are the  
 211 volume of layers. The diffusion rate from the bulk to the surface is:

$$q_{bs,i}' = \frac{c_{b,i} - c_{s,i}}{D} \quad (25)$$

212 Where D is the diffusion constant. So the charge variables are calculated as:

$$q_{s,p}' = i_{app} + q_{bs,p}' \quad (26)$$

$$q_{b,p}' = i_{app} + q_{bs,p}' - i_{app} \quad (27)$$

$$q_{b,n}' = i_{app} + q_{bs,n}' - i_{app} \quad (28)$$

$$q_{s,n}' = -i_{app} + q_{bs,n}' \quad (29)$$

213 where  $i_{app}$  is the applied electric current, the mole fraction in the surface  
 214 and buck can be calculated based on the charge:

$$x_i = \frac{q_i}{q_{max}} \quad (30)$$

$$x_{s,i} = \frac{q_{s,i}}{q_{s,i,max}} \quad (31)$$

$$x_{n,i} = \frac{q_{n,i}}{q_{n,i,max}} \quad (32)$$

215 According to above function the value of  $V_{U,p}$  and  $V_{U,n}$  can be calculated.  
 216 Regarding  $V_{S,p}$ ,  $V_{S,n}$ ,  $V_e$ , they can be categories as the Ohmic potential:

$$\begin{aligned} V_r &= V_{S,p} + V_{S,n} + V_e \\ &= i_{app}(R_{S,p} + R_{S,n} + R_e) \\ &= i_{app}R \end{aligned} \quad (33)$$

217 The surface potentials  $V_{O,n}$  and  $V_{O,p}$  are due to charge transfer resistance  
 218 and solid-electrolyte interface (SEI) kinetics can be calculated as the simplified  
 219 Butler-Volmer equation (34).

$$V_{O,i} = \frac{RT}{F\alpha} \arcsin\left(\frac{J_i}{2J_{i0}}\right) \quad (34)$$

220 where  $\alpha$  is the symmetry factor,  $J_i$  is the current density,  $J_{i0}$  is the exchange  
 221 current density.

222 Now that, all potentials at equation (16) are defined. Regarding the battery  
 223 dynamic changes can be calculated as:

$$V(t) = V_{U,p} - V_{U,n} - V_r' - V_{O,n}' - V_{O,p}' \quad (35)$$

$$V_r'' = \frac{V_r - V_r'}{\tau_r} \quad (36)$$

$$V_{o,i}'' = \frac{V_{o,i} - V_{o,i}'}{\tau_{o,i}} \quad (37)$$

224 where  $\tau$  are empirical time constants.

225 The SOC of the battery can be calculated according to the charge on the  
 226 buck and surface layer and the whole amount of charge, scaled from 0 to 1 as  
 227 equation (38).

$$SOC = \frac{q_n}{0.6q_{max}} \quad (38)$$

#### 228 2.4. Supercapacitor model

229 The equivalent circuit model of the supercapacitor is simplified into an equiv-  
 230 alent capacitor connected with a resistor in series [44]. The capacitor accounts  
 231 for the canonical capacitance effect of supercapacitor, while the series resistor  
 232 represents the overall ohmic losses [45]. Compared to rechargeable battery,  
 233 supercapacitor has unique electrostatic energy storage characteristic that the  
 234 supercapacitor SOC is directly related to its terminal voltage, so its SOC and  
 235 output current can be calculated through equation (39)(40).

$$SOC = \frac{V_{max} - V_t}{V_{max} - V_{min}} \quad (39)$$

$$I = \frac{V_{oc} - \sqrt{V_{oc}^2 - 4RP}}{2R} \quad (40)$$

236 where  $V_{max}$  is maximum output voltage,  $V_{min}$  output minimum voltage,  $V_t$   
 237 is voltage at sample time  $t$ ,  $R$  equivalent resistance,  $P$  the output power of  
 238 supercapacitor.

239 2.5. DC/DC converter model

240 As described in Section 2.1, a DC/DC boost converter and a DC/DC buck/boost  
 241 converter are connected to the fuel cell and supercapacitor respectively. Each  
 242 converter has two IGBT transistors controlled by two complementary pulse  
 243 width modulation (PWM) signals which are calculated according to fuel cell  
 244 and supercapacitor reference currents as Figure 4. The same topology is applied  
 245 to design the DC/DC converter between fuel cell and DC-bus. Nevertheless, in  
 246 this case, only one IGBT transistor is used (s2), the other one is always set to  
 247 off mode.

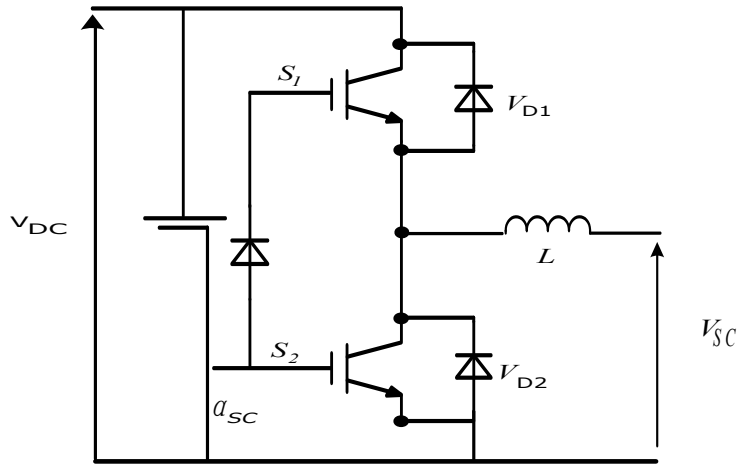


Figure 4: 2-quadrant DC/DC buck/boost converter for supercapacitor

248 The relationship between input power and output power of two converters  
 249 is shown as equation (41).

$$I_{out} = \eta_{conv} \frac{P_{in}}{U_{out}} \quad (41)$$

250 where  $P_{in}$  is input power,  $U_{out}$  is output voltage and  $\eta_{conv}$  is DC/DC con-  
 251 verter efficiency.

### 252 3. Estimation of power sources SOHs

253 The degradation of power sources affects the normal operation of EMS. In  
254 this section, the method used to determine both fuel cell and battery SOHs is  
255 presented. Model based prognostic approach is used to assess the power sources  
256 degradation. The main advantage of using this method is the small amount  
257 of data required and its on-line implementation ability. Joint state parameter  
258 estimation is the main prognostic problem in this paper and the health of  
259 power sources is determined based on the estimated state parameters [46]. The  
260 prognostic model has strong non-linearity, so a non-linear filter is required. The  
261 unscented Kalman filter (UKF) has not only higher accuracy than extended  
262 Kalman filter but also lower computational cost than particle filter [47]. There-  
263 fore in this paper, UKF is chosen to estimate the battery SOH and fuel cell  
264 SOH. The basic framework for the UKF estimation of the state of a discrete-  
265 time non-linear dynamic system as following equations [48]:

$$x_{k+1} = F(x_k, u_k, v_k) \quad (42)$$

$$y_k = H(x_k, n_k) \quad (43)$$

266 where  $x_k$  is the unobserved state of the system,  $u_k$  is the input,  $v_k$  is the process  
267 noise,  $y_k$  is the observed measurement signal,  $n_k$  the observed noise.

268 The general architecture of the prognostic approach to estimate fuel cell  
269 and battery SOHs using UKF is shown in Figure 5 [49]. Through the input  
270 reference currents of fuel cell and battery ( $u_k$ ) and the real output voltages ( $y_k$ )  
271 measured by sensors, UKF is used to estimate and modify the unobserved state  
272 parameters ( $x_k$ ) of their degradation model. With the right estimated state  
273 parameters, their SOHs can be determined.

#### 274 3.1. Battery degradation and on-line state of health estimation

275 As described in section 2.3, battery model parameters  $q_{s,n}$ ,  $q_{b,n}$ ,  $q_{s,p}$ ,  $q_{b,p}$ ,  
276  $V_{o,p}'$ ,  $V_{o,n}'$ ,  $V_r'$  are taken as states variables  $x$ , and battery output voltage  $V$   
277 as the output variable  $y$ . In order to save time and reduce computational cost,

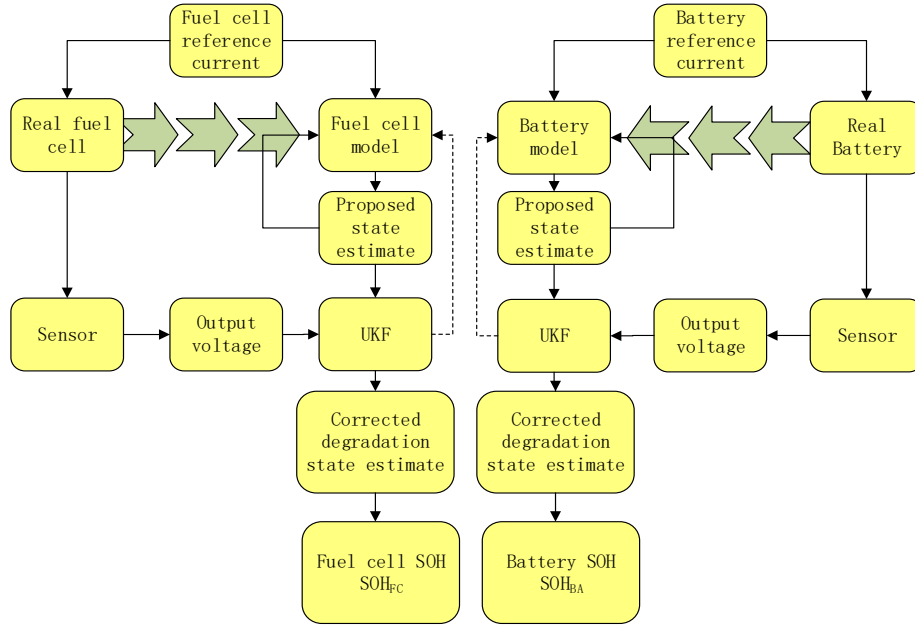


Figure 5: The general architecture of prognostic approach for estimation of power sources SOHs

278 one single battery is tested under a random sequence of charging (negative)  
 279 and discharging (positive) currents among (-4.5A, -3.75A, -3A, -2.25A, -1.5A,  
 280 -0.75A, 0.75A, 1.5A, 2.25A, 3A, 3.75A, 4.5A) [50]. This type of charging and  
 281 discharging operation is referred to here as a random walk (RW) operation. The  
 282 fitting results of battery model under several RWs are shown as Figure 6

283 It can be observed that the battery model can well fit the measurement  
 284 of battery in the health state. Along with battery degradation, some physical  
 285 ageing phenomena can be observed such as SEI layer growth, lithium corrosion,  
 286 lithium loss of lithium plating and changes in diffusion property [51]. The fitting  
 287 result of the same battery model under health state for an aged battery can be  
 288 seen in Figure 7.

289 It can be observed that the error between estimated battery voltage through  
 290 the model and real measured voltage is large. So some parameters of battery  
 291 model should be changed along with battery ageing. For the electro-chemistry

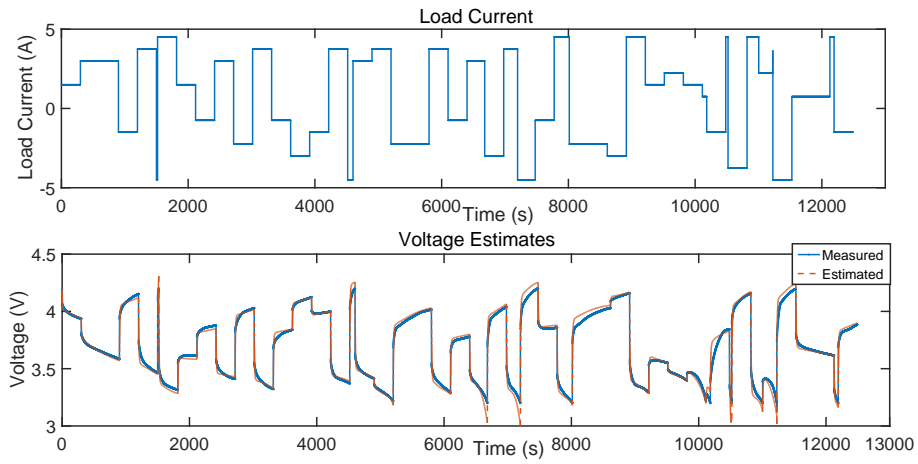


Figure 6: Battery model fitting for new battery

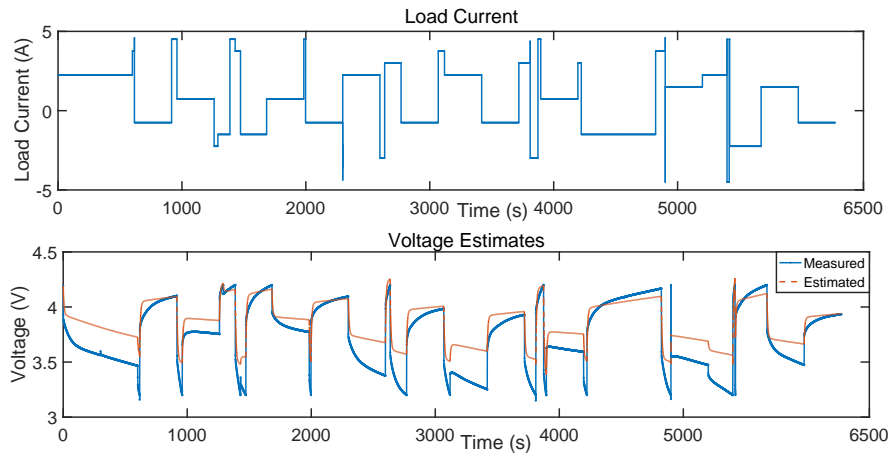


Figure 7: Battery model fit for aged battery



292 based model, the maximum charge  $q_{max}$ , which stands for the loss of active Li  
 293 ions due to degradation, the internal resistance  $R$  representing SEI layer growth  
 294 and the diffusion constant  $D$  are selected as the ageing parameters of battery  
 295 ageing mode. Along with the degradation,  $q_{max}$  decreases,  $R$  and  $D$  increase.  
 296 The state of health of battery can be calculated based on  $q_{max}$ , as equation  
 297 (44).

$$SOH_{BA} = \frac{q_{max}^{int} - q(t)}{q_{max}^{int} - q_{max}^{min}} \quad (44)$$

298 where  $q_{max}^{int}$  battery initial maximum charge,  $q_{max}^{min}$  is the 50% of the  $q_{max}^{int}$ ,  
 299 which is the threshold of battery end of life,  $q(t)$  is battery maximum charge at  
 300 time  $t$ .

301 The ageing parameters ( $q_{max}$ ,  $R$ ,  $D$ ) change in time as a function of usage.  
 302 The dynamic change rate can be described as:

$$q_{max}' = w_q |i_{app}| \quad (45)$$

$$R' = w_R |i_{app}| \quad (46)$$

$$D' = w_D |i_{app}| \quad (47)$$

303 where  $i_{app}$  is the applied current,  $w_q$ ,  $w_R$  and  $w_D$  are the ageing rate pa-  
 304 rameters.

305 The experimental test of random charge/discharge sequences of battery can  
 306 well simulate battery operations in the vehicle due to its randomness. Therefore,  
 307 a 3140 minutes experimental test data is used to verify the validation of bat-  
 308 tery degradation model. The modification of battery degradation parameters is  
 309 determined by UKF based on the on-line measured battery voltages from the  
 310 3140 minutes experiment test. The measured and estimated battery voltages,  
 311 estimated degradation parameters  $q_{max}$  and its SOH along with time are shown  
 312 in Figure 8 and Figure 9.

313 With the added degradation state parameters, the battery voltage estimated  
 314 by UKF and its degradation model can match well with the measured volt-

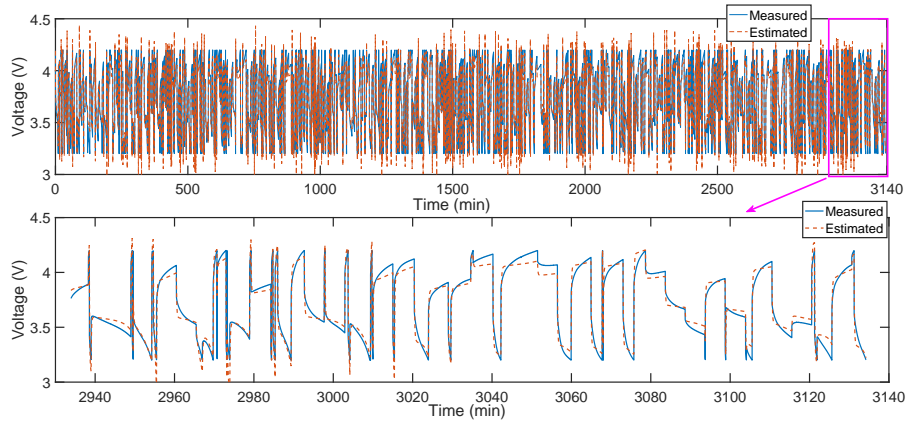


Figure 8: Battery degradation model fit for aged battery

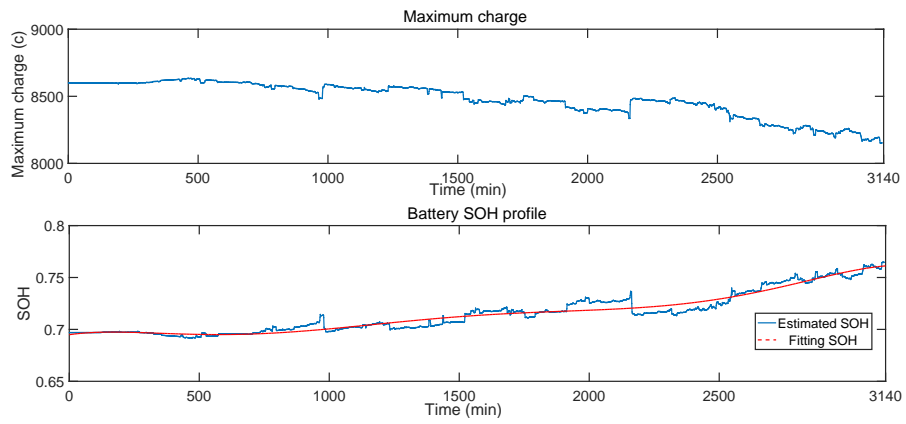


Figure 9: On-line UKF estimates of the degradation state parameter and SOH

315 age over the random charge/discharge sequences. Some errors are still present  
 316 which are due to temperature effects. The internal battery temperature changes  
 317 over time are not considered in the degradation model. The change of battery  
 318 charge/discharge currents leads to the variation of battery degradation state,  
 319 which results in the fluctuation of the decreasing trend of  $q_{max}$  in Figure 9.  
 320 According to the estimated  $q_{max}$  value, battery SOH can be calculated through  
 321 equation (44) and its range is (0,1). Even though there are some fluctuations  
 322 of SOH, the whole trend of battery, which is shown as the line in red color,  
 323 increases along with time meaning the increase of battery degradation. Mean-  
 324 while, the battery SOC also can be precisely estimated with decreasing  $q_{max}$   
 325 along with time based on equation (38).

### 326 3.2. Fuel cell degradation and on-line state of health estimation

327 Upon long term operation of PEMFC, its main components including mem-  
 328 brane, electrodes, bipolar plates, gas diffusion layers, and sealing gaskets would  
 329 undergo mechanical, chemical and electrochemical degradation changes resulting  
 330 in the decrease of fuel cell performance [52]. The presented fuel cell degradation  
 331 model is designed based on the fuel cell model of Section 2.2. The sample time  
 332 of fuel cell ageing process is much larger than the one of fuel cell dynamic model.  
 333 So all the dynamic changes of fuel cell are neglected. The static voltage of the  
 334 PEMFC can be described as equation (48) [53].

$$E = N_{cell} * \left( E_{rev} - A * \ln \left( \frac{I_{FC}}{I_0} \right) - R * I_{FC} - B * \ln \left( 1 - \frac{I_{FC}}{I_{max}} \right) \right) \quad (48)$$

335 According to the previous study of [54, 55], the resistance  $R$  and maximum  
 336 current  $I_{max}$  have large variations along with fuel cell degradation, which are  
 337 chosen as the degradation state parameters. Their variations with time can be  
 338 described as [56]:

$$R(t) = R_0(1 + \alpha(t)) \quad (49)$$

$$I_{max}(t) = I_{max0}(1 - \alpha(t)) \quad (50)$$

$$\alpha(t) = \beta * t \quad (51)$$

339 where  $R_0$  and  $I_{max0}$  are the initial values,  $\alpha$  and  $\beta$  represent degradation  
 340 variance and degradation rate along with time. The fuel cell SOH can be defined  
 341 as equation (52).

$$SOH_{FC} = \frac{\alpha(t) - \alpha_{min}}{\alpha_{max} - \alpha_{min}} \quad (52)$$

342 The estimation of  $SOH_{FC}$  relies on the precise estimation of  $\alpha$  and  $\beta$  through  
 343 the following discrete non-linear system as equation (53), (54).

$$x_{k+1} = A * x_k + w_k \quad (53)$$

$$y_k = g(x_k, u_k) + v_k \quad (54)$$

344 where  $x_k = [\alpha, \beta]^T$  is the UKF state variable,  $A = [1, T; 0, 1]$ ,  $y_k$  is the  
 345 fuel cell voltage,  $w_k$  and  $v_k$  are process and observation noises,  $u_k$  is the input  
 346 current load,  $g(x_k, u_k)$  is described as (48).

347 In order to verify the validation of fuel cell degradation model, a PEMFC  
 348 experimental degradation voltage data is used, which are achieved through a  
 349 400 h experimental degradation test on a 1.2 kW commercial Ballard NEXA  
 350 PEM fuel cell stack [57]. The fuel cell output current is kept as 12A for the  
 351 whole experimental period. Fuel cell degradation model is used to estimate the  
 352 fuel cell voltage along with the experimental test. The variation of degradation  
 353 parameters of fuel cell degradation model is determined by UKF based on the  
 354 on-line measured fuel cell voltage through voltage sensor. The measured and  
 355 estimated fuel cell voltages, estimated degradation parameters  $\alpha$  and its SOH  
 356 along with time are shown in Figure 10.

357 It can be observed from Figure 10, the estimated fuel cell voltages through  
 358 UKF accurately fit with the measured voltages from the experimental data,

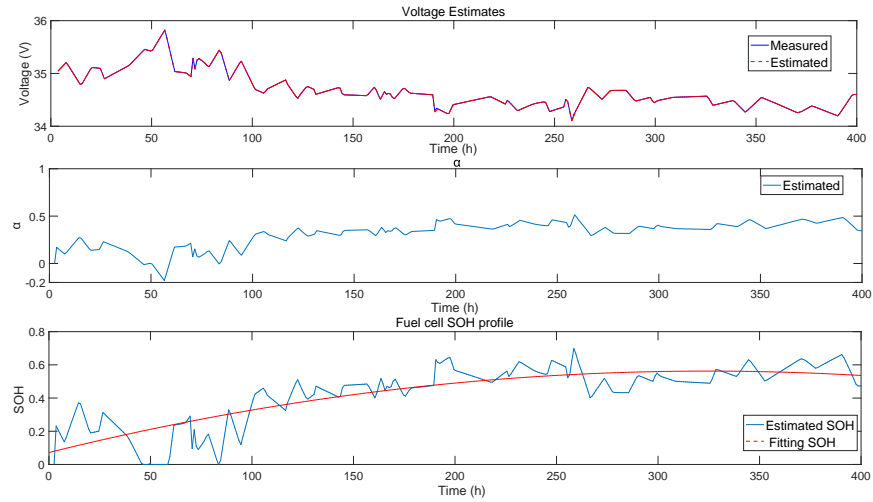


Figure 10: Estimation of the stack voltage, degradation parameter  $\alpha$  and SOH

359 which can prove the accuracy of the online estimation model. The negative  
 360 degradation parameter  $\alpha$  and its fluctuations are due to the reversal of fuel cell  
 361 degradation. When  $\alpha$  is less than 0, its value is set as zero. With the estimated  
 362 value of  $\alpha$ , the fuel cell SOH can be calculated through equation (52) and its  
 363 range is (0,1). It also can be observed that the general trend of fuel cell SOH  
 364 in red color increases along with time.

#### 365 4. Energy management strategy

##### 366 4.1. Equivalent consumption minimization strategy

367 In order to minimize hydrogen consumption and prolong the fuel cell life-  
 368 time, an ECMS is designed. The core idea of ECMS is transforming the electric  
 369 consumption from battery and supercapacitor into equivalent hydrogen con-  
 370 sumption and minimizing the sum of the equivalent hydrogen consumption and  
 371 direct hydrogen consumption from fuel cell [58]. Meanwhile, some constraints  
 372 are added to make sure the normal operation of power sources. Its optimal  
 373 equation can be described as the following equation (55):

$$\begin{aligned}
\min f_w(t) &= m_{FC}(t) + m_{BA}(t) + m_{SC}(t) \\
&\left\{ \begin{array}{l} I_{FC}^{min} \leq I_{FC} \leq I_{FC}^{max} \\ I_{SC}^{min} \leq I_{SC} \leq I_{SC}^{max} \\ -dI_{FC} \leq \frac{I_{FC}(t) - I_{FC}(t-1)}{T} \leq dI_{FC} \end{array} \right. \quad (55)
\end{aligned}$$

374 where  $f_w(t)$  represents whole hydrogen consumption at sample time  $t$ ,  $m_{FC}(t)$   
375 is fuel cell hydrogen consumption,  $m_{BA}(t)$  is battery equivalent hydrogen con-  
376 sumption,  $m_{SC}(t)$  represents supercapacitor equivalent hydrogen consumption,  
377  $I_{FC}$  and  $I_{SC}$  represents the current of fuel cell and supercapacitor respectively,  
378  $I_{SC}^{min}$  is minimum current (-18A),  $I_{SC}^{max}$  are maximum current (18A),  $I_{FC}(t)$  and  
379  $I_{FC}(t-1)$  is fuel cell current at time  $t$  and at time  $t-1$ ,  $dI_{FC}$  is fuel cell maxi-  
380 mum current change rate (1A/S). In case fuel cell works at low efficiency zone  
381 at very low current and frequent on/off cycle, the minimum current of fuel cell  
382  $I_{FC}^{min}$  is set as 4.5A, less than that value, fuel cell is shut off. The maximum  
383 current  $I_{FC}^{max}$  is its normal current (47A).

384 In order to let fuel cell seek for its maximum efficiency point in its high  
385 efficiency zone and keep charge sustenance of ESSs which means that let the  
386 terminal battery and supercapacitor SOC be equal or close to their initial values,  
387 some corresponding penalty coefficients are added into the objective function of  
388 equation (55) as shown in equation (56).

$$\begin{aligned}
f_w(t) &= K_{FC}m_{FC}(t) + K_{BA}m_{BA}(t) + K_{SC}m_{SC}(t) \\
&= K_{FC}m_{FC}(t) + K_{BA}\lambda_{BA}P_{BA}(t) + K_{SC}\lambda_{SC}P_{SC}(t) \quad (56)
\end{aligned}$$

389 where  $K_{BA}$  and  $K_{SC}$  are penalty coefficients which limit battery and super-  
390 capacitor SOC range and variation between final SOC and initial SOC,  $K_{FC}$  is  
391 the fuel cell efficiency penalty coefficient,  $\lambda_{BA}$  and  $\lambda_{SC}$  are the corresponding  
392 equivalent factors,  $P_{BA}(t)$  and  $P_{SC}(t)$  are the battery power and supercapacitor  
393 power.

394 Fuel cell efficiency penalty coefficient  $K_{FC}$  is described as equation (57):

$$K_{FC} = \begin{cases} (1 - 2 * \frac{\eta - \eta_{opt}}{\eta_{max} - \eta_{min}})^2 & \eta \geq 0.4 \\ (1 - 2 * \frac{\eta - \eta_{opt}}{\eta_{max} - \eta_{min}})^4 & \eta < 0.4 \end{cases} \quad (57)$$

395 where  $\eta$  is the instantaneous efficiency,  $\eta_{opt}$  is optimal efficiency (0.4283),  
 396  $\eta_{max}$  the maximum efficiency (0.4283),  $\eta_{min}$  the minimum efficiency (0). When  
 397 fuel cell system efficiency is below than 0.4, a large penalty value  $K_{FC}$  is calcu-  
 398 lated to shut fuel cell down or operate fuel cell to meet power demand by drive  
 399 cycle based on battery and supercapacitor SOC values. The orders of  $K_{FC}$  for  
 400 two conditions are 2 and 4 respectively which are decided by drive cycle power  
 401 demand. Seeking maximum efficiency point and restricting high efficiency zone  
 402 ( efficiency above 0.4 ) are controlled through  $K_{FC}$ .

403 Battery SOC penalty coefficient  $K_{BA}$  is defined as equation (58):

$$K_{BA} = \begin{cases} (1 - \frac{2*(u - B_{int})}{B_{max} - B_{min}})^4 & B_{min} \leq u \leq B_{max} \\ (1 - \frac{2*(u - B_{int})}{B_{max} - B_{min}})^{20} & u < B_{min}, u > B_{max} \end{cases} \quad (58)$$

404 where  $u$  is the instantaneous battery SOC,  $B_{int}$  is battery initial SOC,  $B_{max}$   
 405 the maximum SOC,  $B_{min}$  the minimum SOC.  $K_{BA}$  operates the battery SOC  
 406 to return back to its initial SOC. When battery SOC reaches  $B_{min}$  or  $B_{max}$ ,  
 407 high  $K_{BA}$  value is defined as the punish factor to avoid the battery continues  
 408 to discharge and charge respectively.

409 Supercapacitor penalty coefficient  $K_{SC}$  is composed of SOC coefficient  $S_{eff}$   
 410 and peak power coefficient  $S_{peak}$ .  $S_{eff}$  is similar to  $K_{BA}$  to limit supercapaci-  
 411 tor SOC value at reasonable range.  $S_{peak}$  is used to let supercapacitor supply  
 412 peak power. In order to avoid the frequent fuel cell on/off cycles and frequent  
 413 supercapacitor charge/discharge cycles due to supercapacitor SOC large ampli-  
 414 tude changes in short time, supercapacitor SOC is equivalent to battery SOC

415 to define  $S_{eff}$ .  $K_{SC}$ ,  $S_{eff}$  and  $S_{peak}$  can be defined as following equations  
 416 respectively:

$$K_{SC} = S_{eff} * S_{peak} \quad (59)$$

$$S_{eff} = \begin{cases} (1 - 2 \frac{ax+b-S_{opt}}{S_{max}-S_{min}})^2 & S_{min} \leq x \leq S_{max} \\ (1 - 2 \frac{ax+b-S_{opt}}{S_{max}-S_{min}})^{20} & x < S_{min}, x > S_{max} \end{cases} \quad (60)$$

$$S_{peak} = \begin{cases} 1 & 0 \leq I_{load} \leq 30 \\ -0.01 * I_{load} + 1 & I_{load} < 0, I_{load} > 30 \end{cases} \quad (61)$$

417 where  $x$  is the instantaneous supercapacitor SOC,  $S_{opt}$  is optimal SOC,  $S_{max}$   
 418 the maximum SOC,  $S_{min}$  the minimum SOC,  $I_{load}$  is current demand on the  
 419 DC bus,  $a$  and  $b$  are the transform coefficients from supercapacitor SOC to  
 420 equivalent battery SOC and their values are decided by battery minimum SOC  
 421 and maximum SOC.

422 After defining all the parameters in the equation (55), the ECMS strategy  
 423 is transformed into a nonlinear constrained optimization problem. If this strat-  
 424 egy can be used online is decided by computation cost and storage memory  
 425 requirement. In this paper, sequential based programming (SQP) approach is  
 426 programmed in C language to solve this problem. The SQP method generates  
 427 steps by solving quadratic subproblems and it can be used both in line search  
 428 and trust-region frameworks [59]. The SQP algorithm belongs to a local op-  
 429 timization algorithm that is very useful for solving problems with significant  
 430 nonlinearities [60].

#### 431 4.2. Experiment implementation and validation

432 In order to verify experimentally and evaluate the performances of EMS, a  
 433 test bench is designed. It includes the main components described at Section  
 434 2 such as PEMFC, battery, supercapacitor, DC/DC converters and some other  
 435 equipment like measurement sensors, power supply, electric load, MicroAutoBox



436 and PC. The parameters of the vehicle and components in the test bench are  
 shown in Table 2.

Table 2: The parameters of the vehicle and components in the test bench

Vehicle	Mass	530 kg	Front surface	2.56 $m^2$
	Drag coefficient	0.8	Rolling coefficient	0.02
PEMFC	Manufacturer	The Ballard NEXATM	Rated power	1200 W
	Rated current	46 A	Rated voltage	26 V
	Voltage range	22-50 V		
Battery	Manufacturer	Yuasa Battery	Capacity	90 Ah
	Numbers	4 series	Nominal Voltage	12 V
Supercapacitor	Manufacturer	Maxwell Technologies	Rated Capacitance	58 F
	Numbers	2 series*2 parallel	Rated Voltage	16 V
Experiment conditions	Ambient temperature	25.18 °C	Input hydrogen pressure	10.24 Barg

437

438 Some sensors are used to measure relevant current and voltage values of  
 439 power sources. The vehicle used in this paper is a light duty vehicle and is  
 440 designed for the postal delivery service in the university campus or city com-  
 441 munities. A postal delivery mission is based on a lot of start/stop sequences,  
 442 making the mean value of speed over time very small. In order to reproduce  
 443 the behavior of such application, WVUCITY drive cycle, New York Bus drive  
 444 cycle and LA92 drive cycle are chosen for simulation purposes due to their sim-  
 445 ilarities with the required average speed and start/stop numbers. Power supply  
 446 and electric load are used to supply negative power and positive power of drive

447 cycle respectively. MicroAutoBox II from dSPACE serves as the control unit.  
 448 The ECMS is downloaded into this control unit. According to all gathered control  
 449 signals needed by EMS, the fuel cell reference current and supercapacitor  
 450 reference current are calculated respectively by the control unit. Two classical  
 451 PI controllers are applied to adjust duty cycles of PWM signals to control the  
 452 fuel cell and supercapacitor output currents to track the reference currents as  
 453 Figure 11. Afterward, 20KHz PWM signal produced by MicroAutoBox II is  
 454 transmitted into DC/DC converter and controls its normal operation.

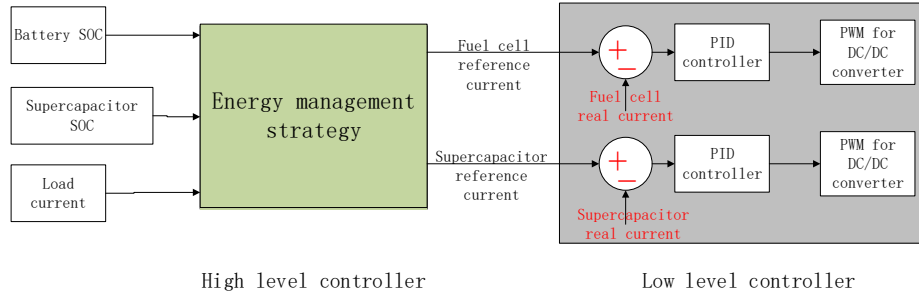


Figure 11: Control system of testbench

455 Human machine interface in the personal computer is designed under the  
 456 software ControlDesk 4.2 to monitor the predefined variable's instantaneous  
 457 value, tune control parameters of the whole control system and record all real  
 458 time variable's value in order to analyze the final experiment results. The ar-  
 459 chitecture of the test bench is shown in Figure 12.

460 The experiment and simulation results are shown in Figure 13 including cur-  
 461 rents of three power sources, SOC's of two energy storage sources as well as fuel  
 462 cell efficiency penalty coefficient  $K_{FC}$ , battery SOC penalty coefficient  $K_{BA}$  and  
 463 supercapacitor penalty coefficient  $K_{SC}$ . In order to facilitate the explanation  
 464 of the operation process of ECMS strategy, the magnifying simulation results:  
 465 currents of three power sources and their corresponding penalty coefficients from  
 466 time 500s to 700s are shown in Figure 14.

467 It can be observed from Figure 13 that the load currents supplied by a  
 468 power supply and an electrical load fit well with the simulation current profile

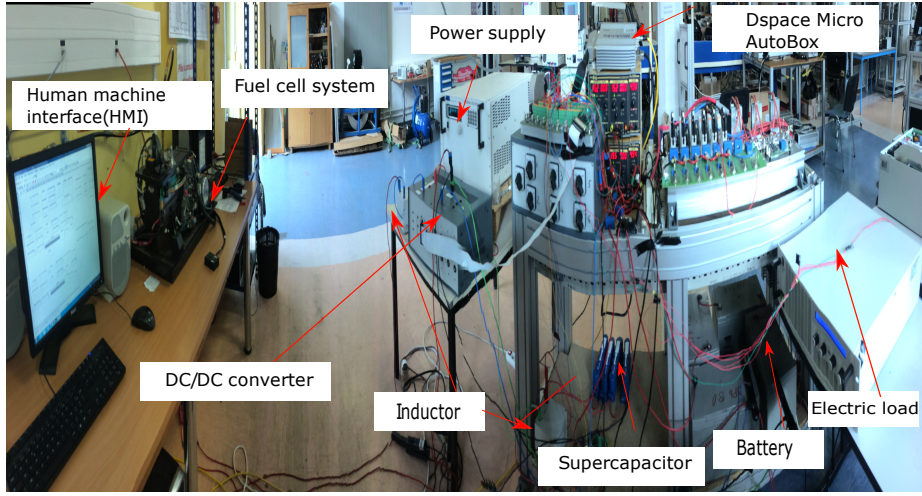


Figure 12: Test bench architecture

469 of the drive cycle. Fuel cell starts at 440s and 456s for simulation results and  
 470 experimental results, respectively. Experiment results of fuel cell current and  
 471 battery current are similar to simulation results. The final simulation result of  
 472 battery SOC is 0.792 similar to the experiment value 0.7915. Supercapacitor  
 473 current is sensitive to the power requirement of the drive cycle, fuel cell current  
 474 and battery SOC, which explains the difference between simulation and exper-  
 475 imental results on supercapacitor current. Considering the low energy density  
 476 of supercapacitor and its role as peak power supplier, this difference can be  
 477 neglected.

478 In Figure 14, battery SOC is 0.7967 at simulation time 500s. With the  
 479 discharge of the battery, its penalty coefficient  $K_{BA}$  increases. The objective  
 480 function of ECMS leads to decrease the battery current to prevent the reduc-  
 481 tion of battery SOC when the SQP algorithm solves the minimizing objective  
 482 function of the following step. Fuel cell plays a bigger role in supplying power  
 483 requirement of drive cycle along with the increase of  $K_{BA}$ . Meanwhile, its ef-  
 484 ficiency decreases and  $K_{FC}$  decreases along with the increase of its current.  
 485 Therefore, the difference between the actual current and the maximum efficient  
 486 one (9.5 A) is enlarged.  $K_{FC}$  as a penalty coefficients increases to operate fuel

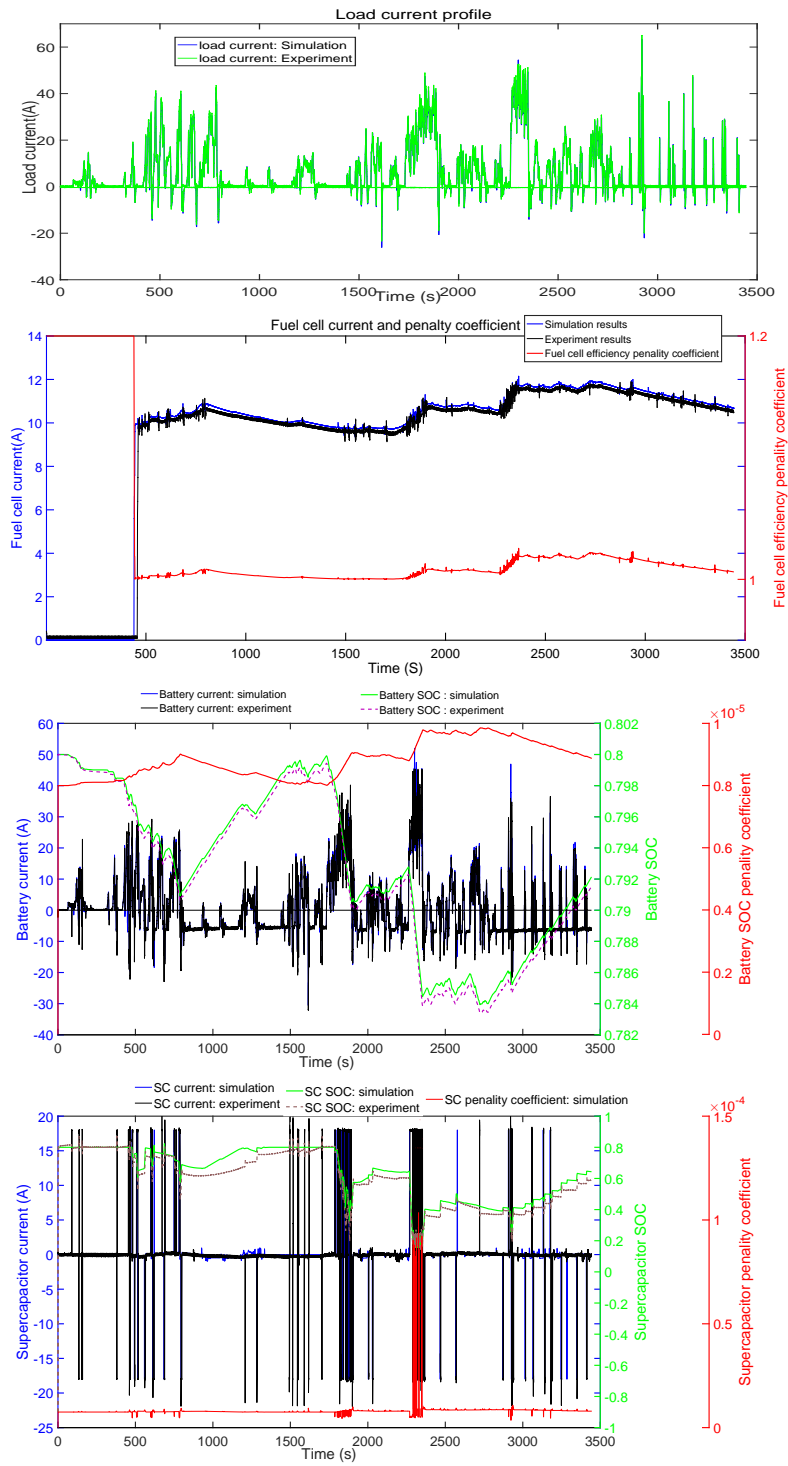
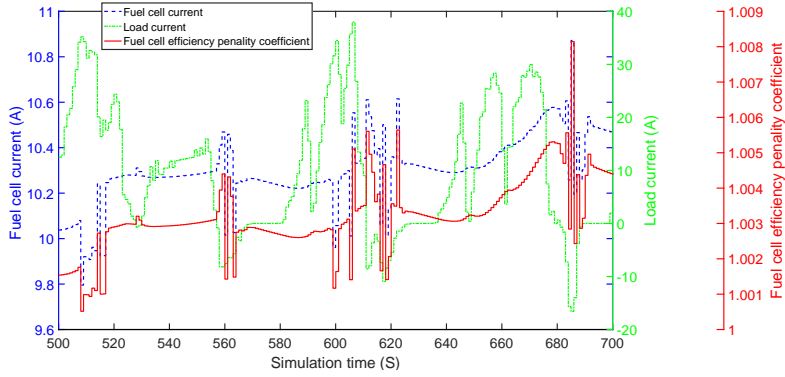
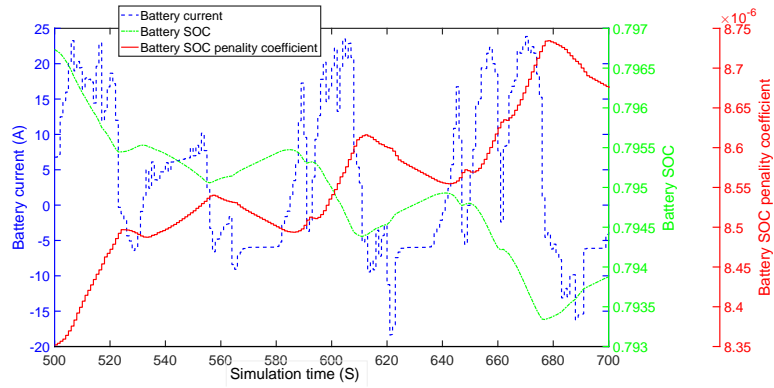


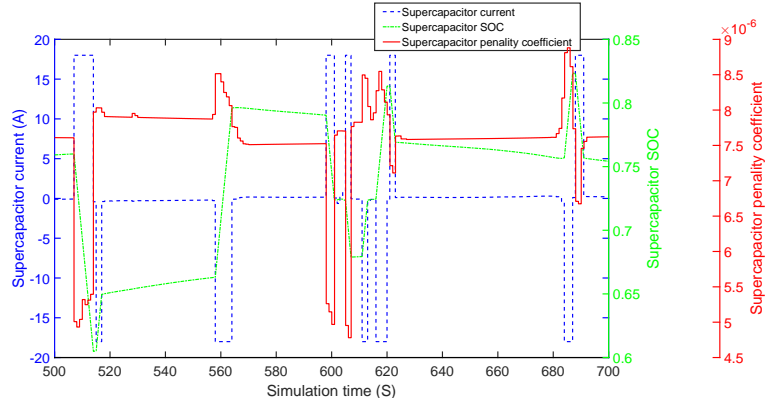
Figure 13: Comparative results between experiment and simulation



(a) Fuel cell current, load current and penalty coefficient



(b) Battery current, battery SOC and battery penalty coefficient



(c) Supercapacitor current, supercapacitor SOC and supercapacitor penalty coefficient

Figure 14: Operations of three power sources

487 cell to seek for maximum efficiency point. When  $K_{BA}$  and  $K_{FC}$  both increase  
488 to decrease battery current and fuel cell current, their reductions rely on the  
489 increasing degree of their penalty coefficients in the objective function. It should  
490 be mentioned that the output currents of fuel cell and battery on DC bus should  
491 be equal to the load current, which may increase both currents even though  
492  $K_{FC}$  and  $K_{BA}$  both increase. Supercapacitor is operated as the battery. The  
493 only difference is peak power coefficient  $S_{peak}$ . When the load current is larger  
494 than 30A or lower than 0A,  $S_{peak}$  is lower than 1 and supercapacitor penalty  
495 coefficient  $K_{SC}$  gets smaller than battery penalty coefficient  $K_{BA}$ , leading to  
496 discharging firstly in larger current than 30A and charging firstly when the load  
497 current is negative.

498 In order to prove the superiority of the new designed ECMS, A RBCS is de-  
499 signed and it is divided into two parts: load following control strategy (LFCS)  
500 as the first part to decide fuel cell reference current, and, operating mode con-  
501 trol strategy (OMCS) as the second part to calculate supercapacitor current.  
502 Regarding LFCS, fuel cell is set to work in high efficiency zone and its reference  
503 current is decided by the current demand of drive cycle and battery SOC value  
504 as equation (62). The on/off cycle of fuel cell is also controlled by the LFCS  
505 through the added control parameters: fuel cell minimum off time, which means  
506 since the fuel cell was last on, the restart should not be less than this time, and  
507 fuel cell minimum on time to avoid frequent on/off cycles. The charge suste-  
508 nance of the battery is also sought through adjusting the fuel cell current based  
509 on battery SOC.

$$I_{fc} = \begin{cases} I_{min} & I_{ch} + I_d < I_{min} \\ I_{ch}(baSOC) + I_d & I_{min} \leq I_{ch} + I_d \leq I_{max} \\ I_{max} & I_{ch} + I_d > I_{max} \end{cases} \quad (62)$$

510 when fuel cell current is decided by the LFCS, the supercapacitor current

511 is decided by supercapacitor SOC and the difference between fuel cell reference  
 512 current and power demand of drive cycle. The flow chart of OMCS to determine  
 513 supercapacitor current is shown as figure 15. When fuel cell current and bat-  
 514 tery current are decided, battery current is passively decided by the difference  
 515 between current demand, fuel cell current and supercapacitor current on DC  
 516 bus.

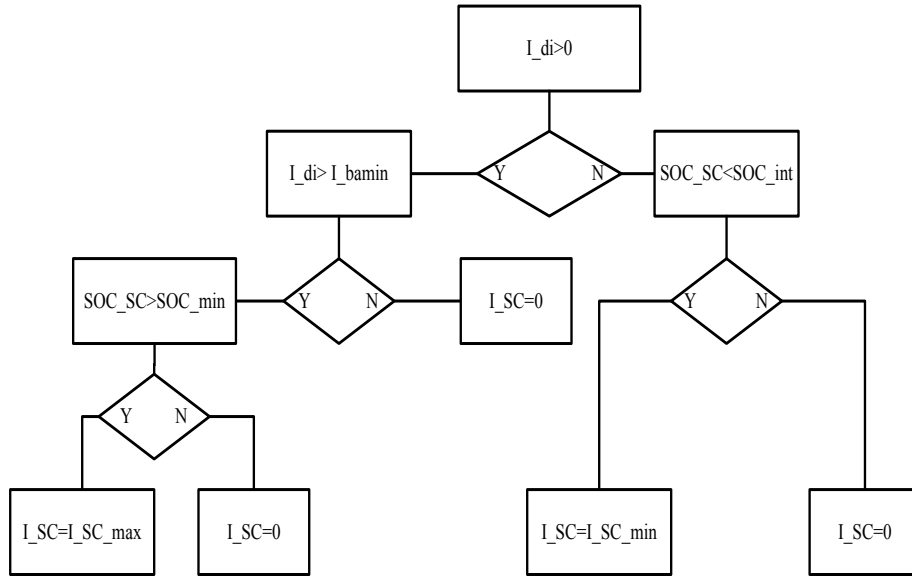


Figure 15: Flowchart of OMCS

517 Due to low energy density and the role of supercapacitor as peak power sup-  
 518 plier, many research on ECMS simplifies the equivalent hydrogen consumption  
 519 of supercapacitor into null. To prove that the simplification of ECMS cannot  
 520 reach the optimal resolution, an HEOS is designed. The first part of HEOS is  
 521 simplified ECMS to calculate fuel cell current and the second part is OMCS to  
 522 calculate supercapacitor current. Three control strategies are tested using the  
 523 described test bench and experiment results are shown as Table 3.

524 According to Table 3, final battery SOC<sub>s</sub> of all three strategies are above  
 525 the 0.79, which is defined as the criteria value that whether the battery satisfies  
 526 charge-sustenance requirement. Hydrogen consumption by fuel cell at the end

Table 3: Experiment results

EMSS	ECMS	RBCS	HEOS
Initial supercapacitor and battery SOC	0.8	0.8	0.8
Final supercapacitor SOC	0.5865	0.8304	0.8199
Final battery SOC	0.7915	0.7946	0.7906
First hydrogen consumption (L)	194.07	210.35	194.77
Final hydrogen consumption (L)	225.06	230.02	228.42

527 of drive cycle is defined as the first hydrogen consumption. Regarding different  
528 battery final SOC values at the end of the drive cycle, fuel cell is continued to  
529 operate at maximum efficiency point to charge the battery until its final SOC  
530 reaches to 0.8. Meanwhile, the supercapacitor is charged or discharged to let  
531 its final SOC be equal to 0.8. When all SOC of energy storage sources are  
532 0.8, the fuel cell stops working, the hydrogen consumption is defined as the  
533 final hydrogen consumption. It can be observed that the ECMS has least final  
534 hydrogen consumption, which proves its superiority.

## 535 5. Power sources degradation and AECMS

### 536 5.1. The effect of power sources degradation on ECMS

537 Along with fuel cell degradation, the fuel cell output voltage decreases, but  
538 auxiliary power doesn't change, so efficiency decreases. When fuel cell is fully  
539 degraded, that its SOH is equal to 1, the efficiency changes as Figure 16.

540 It can be observed that the efficiency of the aged fuel cell is all less than  
541 the new one. The maximum efficiency decreases from 42.83% to 41.37%, but  
542 the maximum efficiency point occurs at the same current point 9.5A. So, the  
543 parameter  $\eta_{opt}$  and  $\eta_{max}$  of  $K_{FC}$  should tune with fuel cell degradation degree.

544 In order to analyze the effects of power sources degradation on the designed  
545 ECMS, four degradation conditions of fuel cell and battery are considered as  
546 Table 4. The simulation results for four conditions are shown as Figure 17, in or-  
547 der to be convenient to describe different conditions, they are defined as FHBH,



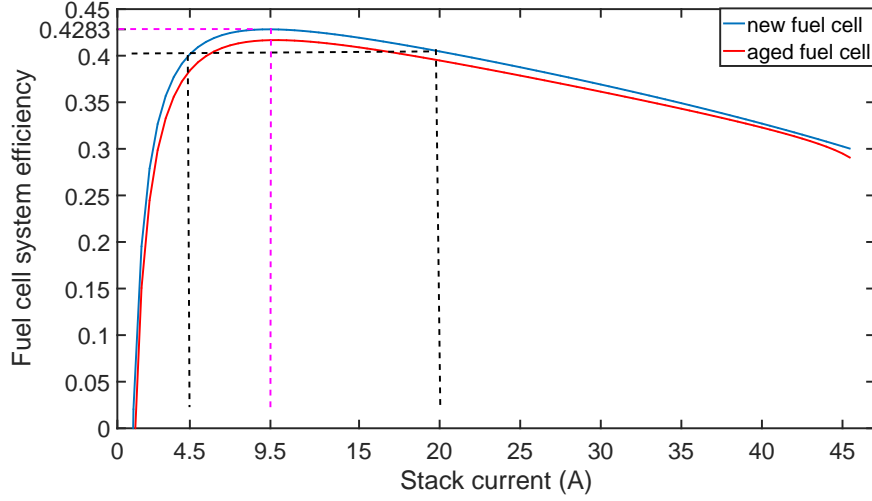


Figure 16: Comparative results of fuel cell system efficiency between the new fuel cell and aged one

548 FDBH, FHBD and FDBD respectively. The simulation results are shown as  
 549 Table 5. Every condition is analyzed respectively in detail.

Table 4: Four degradation conditions of power sources

Condition	$SOH_{FC}$	$SOH_{BA}$
FHBH	0	0
FHBD	0	1
FDBH	1	0
FDBD	1	1

550 From Figure 17 and Table 5, it can be observed that the fuel cell of FHBH  
 551 starts at 443s and it fluctuates around the maximum efficiency point due to  
 552 battery and supercapacitor SOC variations. The battery final SOC is almost  
 553 equal to its initial value. FHBH has the least maximum battery SOC variation  
 554 which is defined as the difference between maximum battery SOC and minimum  
 555 battery SOC during the whole drive cycle period.

556 Compared to FHBH, the fuel cell of FHBD starts earlier, the battery final

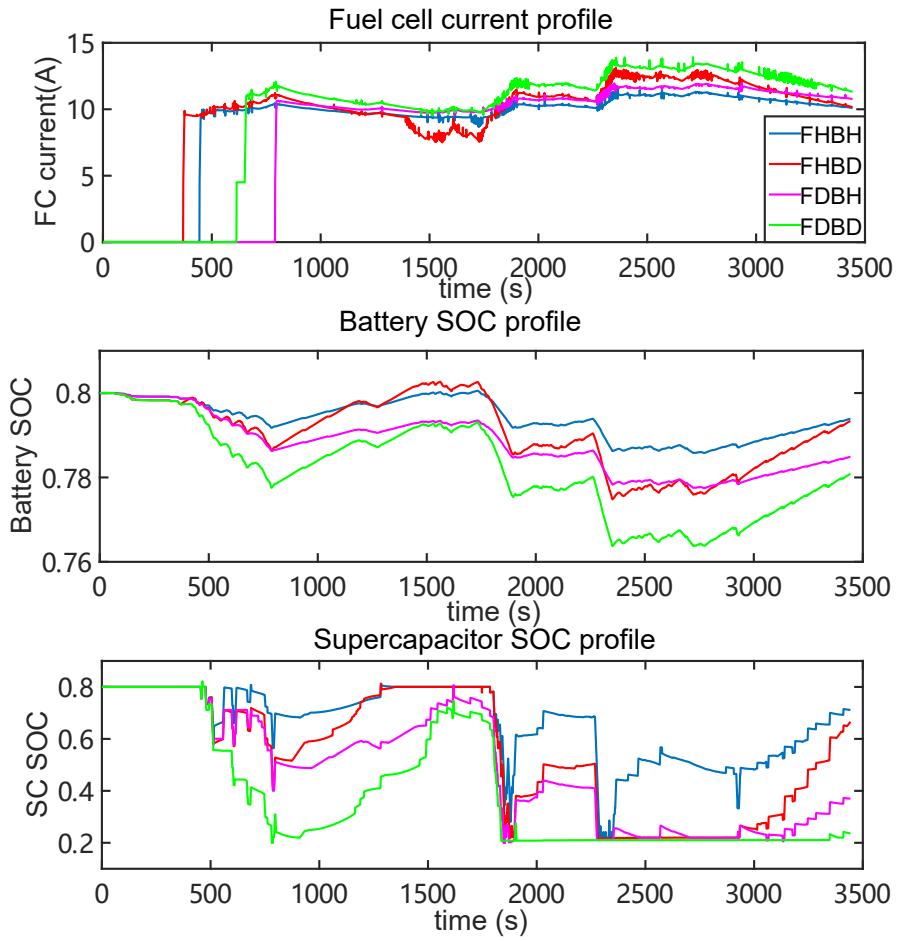


Figure 17: Comparative results of ECMS under four power sources degradation conditions

Table 5: Simulation results of ECMS for four conditions

	FHBH	FDBH	FHBD	FDBD
hydrogen consumption (g)	15.1	14.0	16.0	16.1
battery initial SOC	0.8	0.8	0.8	0.8
battery final SOC	0.7939	0.7849	0.7934	0.7809
battery SOC variation	0.0147	0.0225	0.0279	0.363
supercapacitor initial SOC	0.8	0.8	0.8	0.8
supercapacitor final SOC	0.7115	0.3699	0.6661	0.2363
supercapacitor SOC variation	0.5884	0.6072	0.5969	0.6223

557 SOC is almost equal to FHBH. But due to the decrease of battery capacity  
 558 and the increase of battery resistance along with its degradation, the battery  
 559 SOC variation for the whole drive cycle is larger than FHBH, meanwhile, this  
 560 also leads to larger fluctuation amplitude of fuel cell current. Final hydrogen  
 561 consumption of FHBD is 16g increasing 5.625% than FHBH.

562 Compared to the other three conditions, the fuel cell of FDBH starts at the  
 563 latest, which is due to the decrease of fuel cell efficiency and the increase of  
 564 EF. The lower battery SOC is needed to trigger the operation of fuel cell. The  
 565 fuel cell current of FDBH is larger than FHBH on the whole due to low battery  
 566 SOC. Final battery SOC FDBH is 0.7849 and the difference between the initial  
 567 and final value increases by 59.6% than FHBH, which cannot meet the battery  
 568 charge sustenance requirement.

569 FDBD has the largest battery and supercapacitor SOC variations and the  
 570 final SOC's are the least. FDBD also has the most hydrogen consumption and  
 571 fuel cell current is larger than the other three conditions when fuel cell is started.

572 Through the analysis of four degradation conditions, it can be concluded that  
 573 compared to the health state, the battery degradation leads to more variable  
 574 battery SOC, more fluctuation of fuel cell current around maximum efficiency  
 575 point, more hydrogen consumption and earlier fuel cell start-up, but final bat-  
 576 tery SOC is not affected. Fuel cell degradation makes its maximum efficiency

577 value decrease but corresponding fuel cell current doesn't change. The trigger  
578 time of fuel cell start-up is delayed and battery final SOC changes much to  
579 the initial SOC, which cannot be allowed for the charge sustenance of FCHEV.  
580 At the real situation, fuel cell and battery degrade together and the operation  
581 results of the vehicle are worse than single power source degradation.

### 582 5.2. Adaptive equivalent consumption minimization strategy

583 EF is an important parameter for ECMS. For an assigned known driving  
584 mission, a constant equivalent factor is enough to reach an optimal solution.  
585 But when fuel cell and battery degrade, the former constant equivalent factor  
586 cannot guarantee the charge sustenance of ESSs and optimal fuel consumption.  
587 In order to solve this problem, AECMS is proposed by adjusting EF according  
588 to SOHs of power sources [61]. The variation of equivalent factors is defined as  
589 equation (63)(64).

$$\lambda_{BA} = \lambda_{BA0} (1 + 0.193SOH_{FC}) (1 + 0.193SOH_{BA}) \quad (63)$$

$$\lambda_{SC} = \lambda_{SC0} (1 + 0.193SOH_{FC}) (1 + 0.193SOH_{BA}) \quad (64)$$

590 where  $\lambda_{BA0}$  and  $\lambda_{SC0}$  are initial battery and supercapacitor EFs.

591 The degradation of battery leads to the increase of fuel cell current variation  
592 in the whole drive cycle, which means the increase of fuel cell degradation rate.  
593 In order to increase fuel cell lifetime, the limitation on its dynamic change rate  
594 in equation (55) is adjusted according to battery SOH as equation (65).

$$dI = dI_0 * (1 - 0.5 * SOH_{BA}) \quad (65)$$

595 The architecture of the AECMS control system for FCHEV is shown in  
596 Figure 18. Based on the built degradation model of power sources and UKF,  
597 their SOHs are estimated. The equivalent factor and dynamic change rate of  
598 fuel cell are changed according to their SOHs to make sure the normal operation  
599 of the vehicle, increase the durability of power sources, and keep the stability  
600 of control system.

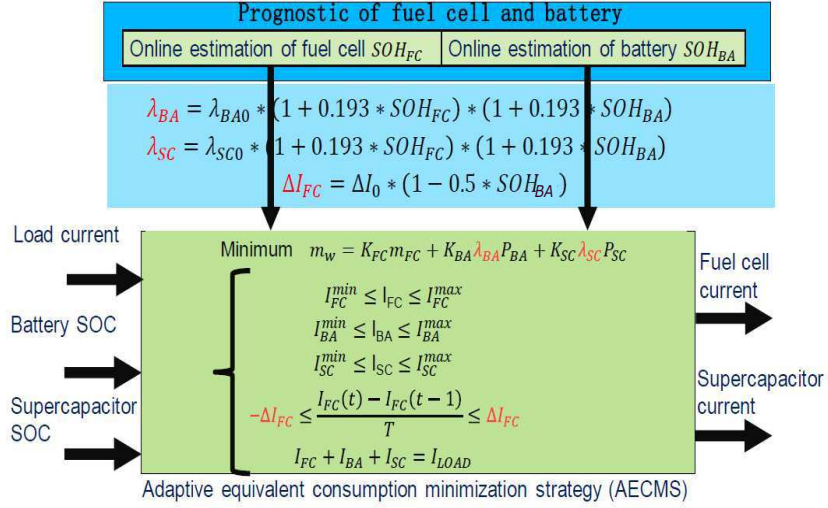


Figure 18: Control system of AECMS for FCHEV

601 The simulation results of AECMS are shown in Figure 19 and Table 6.

Table 6: Simulation results of AECMS for four conditions

	FHBH	FDBH	FHBD	FDBD
hydrogen consumption (g)	15.1	17.7	17.4	19.8
battery initial SOC	0.8	0.8	0.8	0.8
battery final SOC	0.7939	0.7953	0.8	0.8
battery SOC variation	0.0147	0.0157	0.0245	0.0258
supercapacitor initial SOC	0.8	0.8	0.8	0.8
supercapacitor final SOC	0.7113	0.7620	0.8	0.8
supercapacitor SOC variation	0.5884	0.5967	0.6183	0.6032

602 From Figure 19, it can be observed that the simulation results of the con-  
603 dition FHBH are same for ECMS and AECMS. For condition FDBH, the start  
604 time of fuel cell changes from the latest for ECMS to the earliest for AECMS and  
605 the fuel cell current is around maximum efficiency point. Regarding FHBD and  
606 FDBD, the start time is put off and fuel cell current gets larger. The increase of

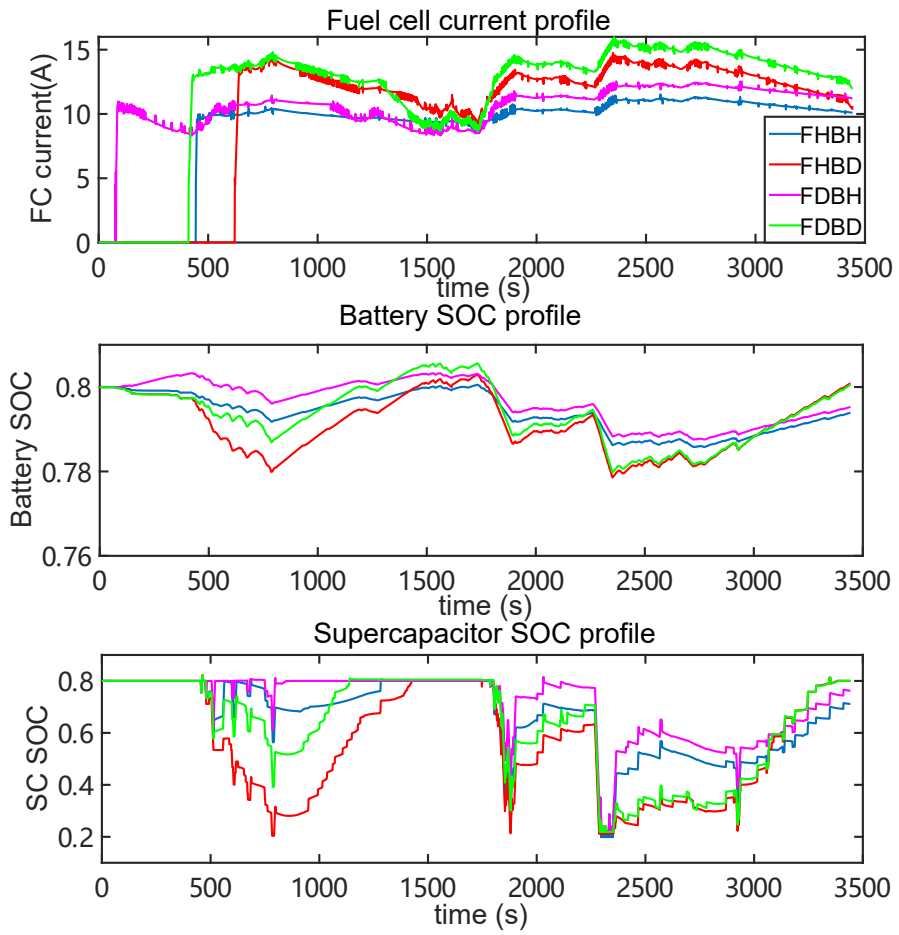


Figure 19: Comparative results of AECMS under four degradation conditions of power sources

607 adaptive equivalent factor along with power sources degradation advances the  
608 start of fuel cell. The decrease of dynamic change rate along with battery SOH  
609 delays the start of fuel cell and makes the increase of fuel cell current. Table  
610 6 shows that all battery final SOC's for four degradation conditions are above  
611 0.79 meaning that the charge sustaining requirement of energy storage sources  
612 is met.

613 Same to ECMS, AECMS has play the full potential of fuel cell, battery and  
614 supercapacitor: fuel cell supplies steady energy and seeks for the maximum effi-  
615 ciency in the high efficiency zone, battery is operated as the main energy buffer  
616 and its charge sustenance is kept at the end of drive cycle and supercapacitor  
617 supplies peak power. The designed AECMS can make sure the normal operation  
618 of a vehicle even though power sources have degraded seriously. It should be  
619 mentioned that when fuel cell and battery SOHs reach 1, it doesn't mean the  
620 fuel cell and battery cannot work anymore and it just shows that the probability  
621 of the failure of power sources is very high. In case of bringing serious damage  
622 to the vehicle, power sources should be replaced in advance.

## 623 **6. Conclusion**

624 In this paper, an ECMS strategy is designed for the FCHEV. Under the  
625 guidance of ECMS, three power sources full play their potential: Fuel cell as the  
626 main power source is operated to seek for the maximum efficiency point in the  
627 defined high efficiency zone, while the battery is taken as the main energy storage  
628 source to buffer energy demand by vehicle and the supercapacitor is dedicated to  
629 providing the peak power. Considering low energy density of supercapacitor, its  
630 equivalent hydrogen consumption is neglected by many ECMSs which leading  
631 to more hydrogen consumption and complex of the control system. This paper  
632 takes all three power sources into the objective function.

633 In order to prove the validity of the ECMS, an experimental test bench is  
634 built and two comparative control strategies: RBCS and HEOS are designed.  
635 The WVUCITY, New York Bus and LA92 drive cycles have been emulated on an

636 experimental test bench with the three above control strategies. The experiment  
637 results show that the proposed ECMS has the least hydrogen consumption and  
638 it offers the longest durability of fuel cell. Along with vehicle operation, fuel cell  
639 and battery degrade, which brings the crisis of optimization and reliability to  
640 the EMS. For the ECMS, the maximum efficiency point, and optimal EF value  
641 are changed. The aging models of the fuel cell and the battery are built, and the  
642 on-line prognostic approach is designed. According to the real-time SOH value  
643 of fuel cell and battery, the designed novel AECMS adjust the EFs and fuel cell  
644 dynamic current rate to make sure the normal operation of the vehicle. In the  
645 future, the optimal coefficients of the equivalent factor and dynamic change rate  
646 of fuel cell current will be studied.

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