

# **Experimental investigation on laminar burning velocities of ammonia/hydrogen/air mixtures at elevated temperatures**

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 The present study introduces new laminar burning velocity data for ammonia/hydrogen/air mixtures measured by means of the outwardly propagating spherical flame method at atmospheric pressure, for previously unseen unburned gas temperatures ranging from 298 to 473 K, hydrogen fractions ranging from 0 vol.% to 60 vol.% in the fuel and equivalence ratios in the range [0.8 – 1.4]. Results show increasing velocities with increasing hydrogen fraction and temperature, with maximum values obtained for rich mixtures near stoichiometry. The new experimental dataset is compared to dedicated laminar burning velocity correlations from the literature and to simulations using detailed kinetic mechanisms. The ammonia/air correlation presents a good agreement with measurements over the whole range of experimental conditions. The ammonia/hydrogen/air correlation captures the effect of the initial temperature satisfactorily for equivalence ratios below 1.3 and hydrogen fractions below 50 vol.% in the fuel, but discrepancies are observed in other conditions. The effect of hydrogen addition is reproduced satisfactorily for hydrogen fractions between 20 and 40 vol.% in the fuel, but discrepancies are observed for rich mixtures

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Abbreviations. GD : Goldmann and Dinkelacker. LBV : Laminar Burning Velocity. OPSF: Outwardly propagating spherical flame. PDF: Probability density function. RES: Renewable Energy Sources. NTP: Normal Temperature and Pressure (298 K, 0.1 MPa)

 below 20 vol.% hydrogen and for all mixtures containing 50 vol.% hydrogen and more. An optimization of both correlations is proposed thanks to the experimental data obtained, but only with partial improvement of the ammonia/hydrogen/air correlation. State-of-the-art detailed kinetic reaction mechanisms yield values in close agreement with the present experiments. They could thus be used along with additional experimental data from different techniques to develop more accurate correlations for time-effective laminar burning 30 velocity estimates of  $NH<sub>3</sub>/H<sub>2</sub>/air$  mixtures.

## **Keywords**

Ammonia, Hydrogen, Laminar Burning Velocity, Elevated temperature, Spherical vessel, Sustainable fuel

## **1. Introduction**

 In response to growing concerns regarding climate change, a majority of governments have agreed on common objectives to mitigate this phenomenon [1]. A recent report by the Intergovernmental Panel on Climate Change evaluated the possibilities and implications of a global warming limitation of 1.5°C above pre-industrial levels, that would alleviate the negative effects of climate change [2]. The report indicated that fossil-free Renewable Energy Sources (RES) should supply 52 to 67% of the total primary energy demand by 2050 in order to sufficiently cut carbon-based greenhouse gas emissions and thus ensure the warming limitation.

 This transition will rely on diverse, mostly intermittent RES, such as wind or solar photovoltaic, and will thus require flexibility and grid-balancing strategies, as well as safe and efficient transport and storage. Power-to-Fuel strategies are promising options, in which excess electricity is used to produce hydrogen- based synthetic fuels in gaseous or liquid form. Those so-called electrofuels [3] present a high energy density (> 4-5 MJ/l), are stable in time (seasonal or long-term storage), can be transported over long distances and their production can be carbon-neutral. In spite of being already and increasingly recognized 47 as a fuel, molecular hydrogen  $(H<sub>2</sub>)$  presents major drawbacks caused by its high volatility and flammability, including the need for a tailored infrastructure and the associated storage and transport costs and safety issues.

[Tapez ici] Ammonia (NH3) has received recent interest as a carbon-free electrofuel [4,5] with a relatively high energy density (13 MJ/l), as it can be stored in liquid form under 1.1 MPa at 300 K and its lower heating  value reaches 18.8 MJ/kg. Additionally, ammonia is already transported and stored safely at industrial scale, which makes it a promising complementary alternative to molecular hydrogen. Current annual production reaches 180 Mt worldwide, mostly from the Haber-Bosch process using steam methane reforming as the 55 main  $H_2$  source. However, renewable hydrogen from electrolysis could be used instead while ensuring a 56 satisfactory stability of the Haber-Bosch process [6]. Grinberg Dana et al. showed that  $NH_3$  exhibits the highest Power-to-Fuel-to-Power ratio when compared with methane, methanol and dimethyl-ether [7]. However, the high corrosiveness and toxicity of ammonia require thorough safety precautions, especially when final users are exposed.

 Following these considerations, several studies focused on ammonia combustion, addressing many of 61 the remaining challenges regarding  $NH_3$  fundamental combustion properties, chemical kinetics modeling or combustion in gas turbines and internal combustion engines as single or dual fuel [8,9]. A major drawback of NH<sup>3</sup> as a fuel is its very low combustion intensity, as illustrated by its Laminar Burning Velocity (LBV), which is one order of magnitude smaller than that of conventional hydrocarbons in atmospheric conditions  $\left[10-17\right]$ . This represents a challenge for NH<sub>3</sub> as a fuel in practical combustion systems, but also for laminar flame experiments themselves, as noted by Pfahl et al. [12], Takizawa et al. [14] and Hayakawa et al. [15]. Indeed, as a function of the mixture composition, ignition energies have to be significantly higher than in the case of conventional hydrocarbons. Moreover, as the LBV for NH3/air mixtures is very slow, the buoyancy effect can cause an outwardly propagating spherical flame (OPSF) to propagate upward as well as outward, thus losing its spherical shape and compromising the measurement.

[Tapez ici] 71 Several experimental studies have considered enhancing the combustion by seeding  $NH_3$  with  $H_2$ , which could conveniently be obtained from ammonia decomposition, leading to a significant increase in the LBV 73 and extending the flammability ranges. Lee et al. investigated the combustion properties of  $NH<sub>3</sub>/H<sub>2</sub>/air$  premixed laminar OPSFs as a function of hydrogen fractions in the fuel blend, for several equivalence ratios from fuel-lean to fuel-rich at Normal Temperature and Pressure (NTP), i.e. 298K, 0.1 MPa [18,19]. Li et al., 76 by using the Bunsen burner method, provided LBV measurements for various  $NH<sub>3</sub>/H<sub>2</sub>/air$  mixtures at NTP as a function of equivalence ratios [20]. Ichikawa et al. investigated the LBV and Markstein length of several NH3/H2/air stoichiometric mixtures by means of the OPSF method also at 298 K but for an initial 79 pressure ranging from 0.1 to 0.5 MPa [21]. Han et al. reported LBV measurements obtained by means of the 80 heat flux method, with NH<sub>3</sub> blended with H<sub>2</sub>, CO or CH<sub>4</sub> as fuels and air as the oxidizer at NTP [22]. Kumar 81 and Meyer conducted Bunsen burner experiments for different  $NH<sub>3</sub>/H<sub>2</sub>$  blends at NTP, but their results 82 contradict those of the previously mentioned studies [23].

83 All these studies reported that the addition of hydrogen to an NH<sub>3</sub> blend significantly increased the LBV with a maximum value around an equivalence ratio of 1.1. However, only partial agreement is found 85 between the different literature sources and the LBV dataset for  $NH<sub>3</sub>/H<sub>2</sub>/air$  flames remains significantly limited, especially at temperatures above 300 K and pressures above 0.1 MPa. Moreover, the comparisons between LBV measurements and numerical simulation results found in the literature show a remaining potential for the improvement of the chemical kinetic mechanisms. For instance, Ichikawa et al. [21] showed only qualitative agreement between their experimental LBVs and those obtained with the mechanisms of Miller et al. [24], Lindstedt et al. [25], Tian et al. [26] and Konnov [27], just as Han et al. [22] exhibited discrepancies between their experiments and the mechanisms of Okafor et al. [28] among others. Recently, 92 Cavaliere et al. [29] conducted a survey on ten chemical kinetic mechanisms including  $NH<sub>3</sub>/H<sub>2</sub>$  chemistry and selected those of Okafor et al. [28], Mathieu and Petersen [30] and Otomo et al. [31] to be reduced, so as to decrease their computational cost. However, they still report a significant scatter in the results obtained by 95 the different mechanisms and see potential for further improvement in the  $NH_3$  and  $NH_3/H_2$  sub-mechanisms.

 As the use of chemical kinetic mechanisms in simulations remains very computationally intensive, Goldmann and Dinkelacker proposed semi-empirical correlations, called GD correlations in the following, 99 for the LBV of NH<sub>3</sub>/air, NH<sub>3</sub>/H<sub>2</sub>/air and NH<sub>3</sub>/H<sub>2</sub>/N<sub>2</sub>/air mixtures [32]. These semi-empirical correlations were based on the correlation by Metghalchi and Keck [33], by considering the LBV dataset available and estimates obtained thanks to the detailed ammonia oxidation mechanism of Mathieu and Petersen [30]. The correlations are simple, making them very useful to be included in computationally intensive CFD simulations. The LBV is given as a function of the global equivalence ratio, the hydrogen amount, the nitrogen ratio (to simulate dilution) and the unburned temperature and pressure, following the form in Eq. 1:

<span id="page-4-0"></span>
$$
s_{\mathrm{u}}^{0}=s_{\mathrm{u,ref}}^{0}T_{\mathrm{n}}^{\alpha}p_{\mathrm{n}}^{\beta}\gamma\kappa,\#(1)
$$

105 where  $s_{\text{u,ref}}^0$  is a reference velocity depending only on the fuel mixture composition and the equivalence 106 ratio,  $T_n$  and  $P_n$  the normalized unburned gas temperature and pressure respectively,  $\gamma$  a factor to model the 107 effect of nitrogen dilution (when relevant) and  $\kappa$  a correction factor to fit the correlations to the experimental values from the literature. Since literature data included only measurements around 300 K and 109 pressure up to 0.5 MPa, the  $\kappa$  factor was determined by considering only these conditions in the GD correlations. The proposed correlations were found to agree very well with the experimental data available in the literature, as well as with results from kinetics simulations using the detailed reaction mechanism of Mathieu and Petersen from which they were derived. Since those simulation results showed no abrupt behavior and an improved accuracy when increasing the pressure, Goldmann and Dinkelacker concluded that the mechanism could be extrapolated to pressures above its validation range, and therefore made the same assumption for their correlations.

 However, this conclusion remains uncertain, as is the accuracy of the reaction mechanisms and LBV 117 correlations for NH<sub>3</sub>/air and NH<sub>3</sub>/H<sub>2</sub>/air mixtures at elevated temperature (above 300 K), due to the lack of experimental literature data under such conditions for validation purposes. The objective of the present study 119 was to partially fill this lack of data by introducing new LBV measurements of  $NH<sub>3</sub>/air$  and  $NH<sub>3</sub>/H<sub>2</sub>/air$  outwardly propagating spherical flames at 0.1 MPa of pressure for unburned gas temperatures up to 473 K. The GD correlations are also discussed with respect to the new experimental data and an optimization attempt is presented.

## **2. Experimental and numerical methods**

## *2.1. Experimental set-up*

[Tapez ici] 125 We carried out the experiments in a 4.2 dm<sup>3</sup> stainless steel spherical vessel that can be heated up to 473 K by an incorporated resistive coil. A type-K thermocouple and a piezoelectric pressure transducer were used to 127 monitor the temperature  $T_u$  and pressure  $P_u$  inside the chamber, respectively. The filling procedure was already described in a previous study on a similar set-up [34] and is thus only briefly summarized here. A vacuum pump is used to empty the vessel to a residual pressure of less than 1 kPa and the preheated reactive gases are then introduced thanks to Brooks 5850S thermal mass flowmeters, while being stirred by a fan to 131 ensure a homogeneous mixture. Bottled gases are used, including synthetic air with  $20.9\% \pm 0.2\%$  oxygen

132 and a 99.999% purity, hydrogen with a 99.999% purity and ammonia with a 99.98% purity. After the intake,

133 a quiescent phase of 10 s is set in order to avoid any undesired fluid motion. Then, a discharge energy is 134 delivered for ignition at the center of the chamber thanks to two 1 mm tungsten electrodes. The spark gap

135 can be adjusted from 1 to 3 mm to favour ignition under lean conditions with a high ammonia content.

## 136 *2.2. Experimental conditions*

137 The global stoichiometric combustion reaction of  $NH<sub>3</sub>/H<sub>2</sub>/air$  is:

$$
(1 - x_{\text{H}_2})\text{NH}_3 + x_{\text{H}_2}\text{H}_2 + \frac{3 - x_{\text{H}_2}}{4} (0_2 + 3.76\text{N}_2) \rightarrow
$$

$$
\left(\frac{3 - x_{\text{H}_2}}{2}\right)\text{H}_2\text{O} + \left(\frac{13.28 - 5.76x_{\text{H}_2}}{4}\right)\text{N}_2 \neq (2),
$$

138 with  $x_{H_2}$ , the hydrogen molar fraction in the fuel mixture. The global equivalence ratio  $\phi$  is defined as:

$$
\phi = \frac{\frac{X_{\text{H}_2} + X_{\text{NH}_3}}{X_{\text{air}}}}{\left(\frac{X_{\text{H}_2} + X_{\text{NH}_3}}{X_{\text{air}}}\right)_{\text{st}}}, \#(3)
$$

139 where  $X_s$  represents the molar fraction of the species  $s$  in the reactive mixture. The experimental conditions 140 are summarized in Table 1. In some boundary cases, such as fuel-lean (resp. fuel-rich) mixtures with  $\phi \leq 0.9$ 141 (resp.  $\phi \ge 1.2$ ) and a small hydrogen fraction, the mixture ignition fails to induce flame propagation or 142 buoyancy instability phenomena alter the flame propagation too much to extract meaningful data and such 143 cases are thus left aside.

## 144 **Table 1.** Experimental conditions



145

## 146 *2.3. Image processing*

 In the cases with sucessful laminar flame propagation, double Schlieren images of the flame are recorded from two orthogonal angles through two pairs of opposite quartz windows (70 mm diameter) by a Phantom v1210 high-speed CMOS camera, as fully described in [35]. The frame acquisition rate of the camera is adjusted with respect to the propagation velocity of the flame up to 18000 fps, in order to maximize the

 number of usable images. Following recommendations by Huo et al. [36], a minimum number of 30 images is considered for post-processing. The double Schlieren configuration allows the detection of flame instabilities, but only one view is used to extract the LBV. An example of a flame image is shown in Figure 154 1.



156 **Figure 1.** Double Schlieren view at time t of a stable NH<sub>3</sub>/H<sub>2</sub>/air flame at 0.1 MPa and 473 K.

 The radii of the spherical flames are extracted using an image postprocessing algorithm with background 158 substraction. The range of the flame radius  $R_f$  used for the analysis is manually kept between roughly 6.5 and 25 mm in order to eliminate the spark ignition and wall-pressure confinement effects. The pressure measured in the vessel remains constant over the whole analysis range. The stretched laminar flame 161 propagation velocity  $s_b$  is calculated as a first-order gradient  $s_b = dR_f/dt$ , since the burned gas is assumed 162 to be quiescent. Assuming that the flame is adiabatic and the propagation quasi-steady,  $s_h$  is then extrapolated to zero stretch using the nonlinear Equation 4 proposed by Kelley and Law [37], based on an asymptotic analysis by Ronney and Sivashinsky [38] and validated by Halter et al. for methane and isooctane/air flames [39]:

$$
\left(\frac{s_{\rm b}}{s_{\rm b}^0}\right)^2 \ln \left(\frac{s_{\rm b}}{s_{\rm b}^0}\right)^2 = -\frac{2L_{\rm b}K}{s_{\rm b}^0}.\,\#(4)
$$

166 The flame stretch K is calculated according to  $K = 2/R_f * dR_f/dt$  for a spherical flame,  $L_b$  is the Markstein 167 length and  $s_b^0$  the unstretched flame propagation velocity of the burned state, respectively.

168 The laminar burning velocity is finally calculated from the continuity equation through the flame 169 surface,  $s_u^0 = \rho_b / \rho_u * s_b^0$ , where burned and unburned gas densities are calculated from equilibrium 170 calculations.

## [Tapez ici] 171 *2.4. Uncertainty considerations*

155

172 The validation of kinetic mechanisms requires accurate measurement data including quantitative uncertainty 173 ranges. In the present study, a method based on the work of Moffat [40] and developed by Brequigny et al. 174 [41] was implemented for uncertainty quantification.

175 The experimental errors are of two kind and described as experimental hardware errors  $(\Delta s_u^0)_{P,T,X_s}$ , 176 reflecting the accuracy of the initial temperature, pressure and mixture composition monitoring, and imaging 177 errors  $(\Delta s_u^0)_{\text{imaging}}$ , both from the imaging technique itself and the processing.

178 The temperature and pressure error terms are determined by using the correlations by Goldmann and 179 Dinkelacker [32] in Eq. 1, as  $|\alpha| \cdot \Delta T/T$  and  $|\beta| \cdot \Delta P/P$  respectively. The exponents  $\alpha$  and  $\beta$ , which are 180 functions  $\phi$  and  $x_{H_2}$  are calculated for each test condition. In most cases, the combined error from those two 181 terms is significantly smaller than  $\pm 2\%$ , but can be higher in a few cases at 298 K initial temperature, due to 182 the difficulty of maintaining the vessel temperature after several combustion tests, without exceeding  $\pm 8\%$ .

183 The uncertainties on the mixture composition are due to the accuracy of the mass flow meters (1% of the 184 full scale) and propagate on the LBV through  $\phi$  and  $x_{H_2}$ . Depending on the representation of  $s_u^0$  chosen in 185 this article, either  $\phi$  or  $x_{H_2}$  can be a variable, while the other is a fixed parameter. While the uncertainty on 186 the variable must be indicated with horizontal error bars, that on the fixed parameter(s) must be propagated 187 in the depicted LBV. Since the LBV dependence on those parameters in the GD correlations is complex, the 188 propagation of the uncertainty could hardly be formulated analytically as for the temperature and pressure 189 errors. It was thus estimated by means of a Monte Carlo method as follows. The set point value of the fixed 190 parameter ( $\phi$  or  $x_{H_2}$ ) is considered as the mean value of a normal probability density function (PDF), whose 191 standard deviation is given by the uncertainty on the parameter. During a great number of  $N = 10000$ 192 iterations, a random value is taken for the parameter following that PDF. The LBV is then calculated by 193 means of the corresponding GD correlation for each iteration, all other parameters and variables remaining 194 unchanged. For each initial condition, a normal PDF for the LBV is obtained this way, of which the standard 195 deviation is considered to be the propagated uncertainty of  $\phi$  or  $x_{H_2}$  on the LBV. The empirical correction 196 factor  $\kappa$  of the GD correlations (Eq. 1) was applied only when  $x_{H_2} < 0.5$ , as it degrades strongly the 197 prediction capability of the  $\phi$  and  $x_{H_2}$  dependences at higher hydrogen fuel fractions, as will be seen in  Section 3. Overall, the LBV uncertainty resulting from mixture composition errors is thought to be conservative.

 Another uncertainty source associated with the OPSF technique is radiation-induced uncertainty. Yu et al. showed that radiative losses affect the LBV of OPSFs by reducing the flame temperature and by inducing an inward flow in the burned gas due to radiation cooling [42]. They proposed a fuel independent correlation for the determination of the radiation-induced relative error on the LBV in such flames, which is a decreasing function of the LBV and depends on *T*<sup>u</sup> linearly. However, that correlation was only validated numerically in the case of different hydrocarbon fuels and syngas, and it is unclear whether it is applicable 206 in our case, especially due to the absence of  $CO_2$  in the burned gas of NH<sub>3</sub>/H<sub>2</sub>/air flames. A recent numerical study by Nakamura and Shindo [43] showed a significant impact of radiative heat losses on the LBV of NH3/air flames using simulations of 1D freely propagating premixed flames with their own reaction mechanism [44]. Under NTP conditions, the relative error increased away from stoichiometry and reached 210 about 13 % for  $\phi = 0.8$ , 3 % for  $\phi = 1.0$  and about 8 % for  $\phi = 1.4$ , corresponding to absolute errors of a few millimeters per second. However, those results do not take into account the spherical geometry of the presently studied flames and depend on the accuracy of the reaction mechanism that was used. Those 213 considerations are summarized in Figure 2 for NH<sub>3</sub>/air flames at NTP, which are the most radiation-affected conditions due to very low LBVs. The simulation results of Nakamura and Shindo are plotted together with similar simulations that we carried out in ChemkinPro [45] (see Sec. 2.5) with the absorption coefficients in [43] and the detailed reaction mechanism of Otomo et al. [31], as well as results obtained by applying the correlation of Yu et al. to present experimental data. Figure 2 highlights the dependence of the radiation- induced error on the LBV, and shows good agreement between the different estimates. Therefore, the correlation of Yu et al. was used presently as the best estimate for the radiation-induced uncertainty,  $(\Delta s_u^0)$ <sub>radiation</sub>, added only in the positive uncertainty  $+ U_{g}^{95\%}$ , and decreasing with increasing LBV down to less than 1% for fast flames. The same estimation method was recently used by Mei et al. [17].





223 **Figure 2.** Relative radiation-induced uncertainty on the LBV of NH3/air flames at NTP.

224

225 The repeatability of the tests is assessed by conducting a minimum number  $n$  of three measurements for each initial condition. The standard deviation proves the repeatability of the measurements and is used to calculate the statistical error in the form of a 95% confidence interval by means of a Student's t-distribution, as in Eq. 5:

$$
\left(\frac{\Delta s_{\rm u}^0}{s_{\rm u}^0}\right)_{\rm statistical} = t \frac{\sigma_{s_{\rm u}^0}}{\sqrt{n}} \#(5)
$$

229 with t, the value of the Student's density function ( $t = 3.182$  for  $n = 3$ ). The statistical error exceeds 230 10% in a small number of cases, generally under very lean or rich conditions, where the flame propagation is 231 mostly affected by perturbations. In general, this value is below 5% and down to 0.1%.

232 The experimental error terms are finally combined with the imaging error, the radiation error and the 233 statistical error obtained by repeating several identical measurements to calculate the overall uncertainty, 234  $U_{s,0}^{95\%}$ , given in Eq. 6:

$$
U_{s_{\mathrm{u}}^{0}}^{95\%} = \sqrt{(\Delta s_{\mathrm{u}}^{0})_{T}^{2} + (\Delta s_{\mathrm{u}}^{0})_{P}^{2} + (\Delta s_{\mathrm{u}}^{0})_{X_{\mathrm{s}}}^{2} + (\Delta s_{\mathrm{u}}^{0})_{\mathrm{imaging}}^{2} + (\Delta s_{\mathrm{u}}^{0})_{\mathrm{radiation}}^{2} + (\Delta s_{\mathrm{u}}^{0})_{\mathrm{statistical}}^{2} \#(6)}
$$

235 Therefore, all the data are presented in the following figures with the average LBV values along with the 236 overall uncertainty,  $\overline{s_u^0}$   $\pm U_{s_0}^{95\%}$ , meaning that the error bars can be assimilated to 95% confidence intervals. 237 All the uncertainties are reported in the Supplementary Material.

## 238 *2.5. Numerical modelling*

 LBV simulations were performed with Ansys ChemkinPro [45] by using three detailed kinetic mechanisms 240 for  $NH<sub>3</sub>/H<sub>2</sub>/air$  combustion [30,31,44]. Otomo's model [31] was validated against experimental results 241 including LBVs obtained in various  $NH<sub>3</sub>/H<sub>2</sub>/A$ ir mixture at ambient temperature and 0.1 MPa. Nakamura's 242 model [44] was validated against  $NH<sub>3</sub>/air$  weak flames in micro flow reactor, including species profiles 243 measurements. Both mechanisms reveal the importance of intermediate species such as NH<sub>2</sub>, HNO and N2Hx. The third one by Mathieu and Petersen [30] was mostly validated against ignition delay times and was selected by Goldman and Dinkelacker [32] to fit their semi-empirical correlations.

 The intercomparison between the experimental values and results from full kinetic computations was limited to 2 cases, but in the entire range of equivalence ratio [0.8-1.4]. First, the unburned temperature is set to 248 473K, and  $x_{H_2}$  is varied between 0 and 0.6. Second, the temperature varies in the entire range of the 249 experiments, but  $x_{H_2}$  is fixed equal to 0.6. Those are the conditions of maximal discrepancy between the GD correlations and the present experiments, as it will be seen in Sec. 3.

## **3. Results and discussion**

 This section presents the main experimental results and compares them to the LBV values estimated by means of the detailed kinetic mechanisms in Sec. 2.5 and the GD correlation. The experimental corrective 254 factor  $\kappa$  defined in Section 1 is always applied in the GD correlations with the values given in [30], unless otherwise stated. The extensive dataset obtained during the present study is available in the Supplementary Material.

## *3.1. Effect of the equivalence ratio*

[Tapez ici] Figure 3 shows a comparison of the present LBV measurements with experimental literature data and 259 the GD correlation for NH<sub>3</sub>/air flames under NTP. The very low LBVs result in a significant scatter in the experimental data, due to technique-dependent uncertainties that are exacerbated by the instability phenomena already described in Sec. 1 and Sec. 2.2 for the OPSF technique. While the data for stoichiometric flames show a good general consistency across the studies, lean and rich flames exhibit differences between the measurements up to more than 2 cm/s. While this discrepancy might seem reasonable in other cases, it represents presently an error of more than 50%, due to the very low LBVs. Present measurements are in the middle of the literature scatter, including very recent measurements with

- 266 different techniques by Han et al. [22] and Mei et al. [17]. Error bars are asymmetric due to the positive-only
- 267 radiation-induced uncertainty, which is especially significant at low LBVs.

268



269

270 **Figure 3.** Laminar burning velocities of NH3/air flames under atmospheric conditions. Symbols: 271 experiments. Line: GD correlation at 0.1 MPa, 298 K [32].

[Tapez ici] 272 Figure 4 shows the variation of the measured LBV as a function of the equivalence ratio over the whole 273 temperature range for  $NH_3/air$  and  $NH_3/H_2/air$  flames, along with experimental data from the literature close 274 to 298 K and 0.1 MPa and GD correlations values. For  $NH<sub>3</sub>/H<sub>2</sub>/air$  flames, present measurements at 298 K 275 are in reasonable agreement with the experimental literature data. For  $x_{H_2} = 0.05$  (Fig. 4b), good agreement 276 is found with Ichikawa et al. [21] at stoichiometry, while a small discrepancy is observed with Han et al. 277 [22]. For  $x_{H_2} = 0.3$  (Fig. 4c), the present measurements partially agree with those of Lee et al. [19] and a 278 very good agreement is found with Han et al. at stoichiometry. For  $x_{H_2} = 0.4$  (Fig. 4d), our values agree with 279 those of Han et al. for equivalence ratios between 0.8 and 1.2, and disagree for  $\phi = 1.3$  and 1.4, while a 280 slight mismatch is observed with a value of Ichikawa et al. at stoichiometry. For  $x_{H_2} = 0.5$  (Fig. 4e), very 281 good agreement is found with the data of Li et al. [20], except for  $\phi = 1.4$  and partial agreement is observed

282 with the data of Lee and coworkers. For  $x_{H_2} = 0.6$  (Fig. 4f), present values agree with those of Li et al. only 283 below stoichiometry, and a good agreement is found at stoichiometry with Ichikawa and coworkers. Overall, 284 a comparative assessment of the uncertainties relative to LBV measurement using different methods would 285 be of interest in the case of  $NH<sub>3</sub>/H<sub>2</sub>/air$  flames, along with additional measurements with various techniques, 286 in order to evaluate the reliability of the global dataset. This is however beyond the scope of the present 287 work.

288 The measured LBVs present a classical bell shape as a function of the equivalence ratio, with a 289 maximum near  $\phi = 1.1$  for all temperatures and hydrogen fractions up to  $x_{\text{H}_2} = 0.5$ . However, for mixtures 290 with  $x_{\text{H}_2} = 0.6$ , the variation in the experimental LBV as  $\phi$  varies becomes smaller around the maximum 291 due to the effect of hydrogen, resulting in a relatively "flatter" shape of the data distribution and a slight shift 292 of the maximum towards  $\phi = 1.2$ , as illustrated in Fig. 4f. At  $T_u = 373$  K for instance, this results in a 293 relative increase of about 30 % of the LBV between  $\phi = 0.9$  and  $\phi = 1.1$ , and a decrease of 18% between  $\phi$ 294 = 1.1 and  $\phi$  = 1.3 for  $x_{\text{H}_2}$  = 0.3, while the relative increase and decrease reach only 13 % and 5 % 295 respectively for  $x_{\text{H}_2} = 0.6$ .

296 The NH<sub>3</sub>/air GD correlation reproduces well the shape of the LBV as a function of  $\phi$ , even at elevated 297 temperatures, as shown in Fig. 4a. The shape is also relatively well reproduced by the  $NH<sub>3</sub>/H<sub>2</sub>/air GD$ 298 correlation for  $0.2 \le x_{\text{H}_2} \le 0.4$ , as illustrated in Fig. 4c and 4d for  $x_{\text{H}_2} = 0.3$  and 0.4. Those are cases where 299 the effect of the equivalence ratio in the GD correlation was validated against literature measurements at 300 NTP conditions. In the cases for which limited experimental data were available or only data that do not 301 agree with the present results, the  $NH<sub>3</sub>/H<sub>2</sub>/air$  GD correlation fails to reproduce accurately the shape of the 302 present measurements. In particular for  $x_{H_2} = 0.05$ , Fig. 4b shows that the correlation underestimates the 303 LBVs for rich mixtures at all temperatures. The correlation values are not symmetric with respect to the 304 maximum LBV, which is underestimated itself both in its magnitude and equivalence ratio of occurrence as 305 compared to the experiments. Fig. 4f shows that when  $x_{H_2} = 0.6$  the estimated shape is symmetric, but does 306 not accurately reproduce the much "flatter" shape of the present experimental points, especially at higher

temperatures. Thus, the availability of reliable experimental LBV data seems to be critical for the behavior





310 **Figure 4.** Laminar burning velocities of NH<sub>3</sub>/air and NH<sub>3</sub>/H<sub>2</sub>/air mixtures at  $P_u = 0.1$  MPa.



*3.2. Effects of hydrogen enrichment*

 The influence of the fuel hydrogen fraction on the LBV is presented in Fig. 5 at NTP conditions. For the sake of readability, only three equivalence ratios are depicted, along with corresponding LBV data from the literature and GD correlation values. Note that the Bunsen burner study of Li et al. [20] did not always provide data at the exact equivalence ratios that are depicted, so the closest values were plotted. The LBV exhibits an exponential increasing trend with volumetric hydrogen addition in the fuel. Present LBV measurements agree well with available literature data for a large majority of cases, as well as with GD 319 correlation values. However, significant discrepancies are noticed for  $x_{H_2} > 0.5$  and  $\phi \ge 1.0$ , as noted in Sec. 3.1. In those conditions, the present data agree with the value of Ichikawa et al. [21], but disagree with the data of Li et al. [20] and the correlation. The good agreement between the correlation and Li et al.'s data 322 is explainable by the high relative weight of that experimental dataset in the  $NH<sub>3</sub>/H<sub>2</sub>/air$  LBV literature at the time of establishing the correlation. The latter was thus mainly fitted on that dataset, through the correction 324 factor  $\kappa$ . Therefore, the accuracy of the measurements at high hydrogen fractions in the fuel might be questioned, with respect to the chosen experimental method. The OPSF method used by Ichikawa et al. and 326 the present authors yields closely agreeing values at  $\phi = 1.0$ . Under NTP conditions, the GD correlation 327 slightly underestimates present LBVs when  $\phi \ge 1.0$  and  $0.05 \le x_{\text{H}_2} \le 0.2$ , agrees well with all 328 measurements for  $0.2 < x_{\text{H}_2} \le 0.5$ , and significantly overestimates LBVs for  $x_{\text{H}_2} > 0.5$  and  $\phi \ge 1.0$ . The data of Lee et al. [19] show slight discrepancies with the other data and the correlation but remain in the trend. However, the data of Kumar et al. [23] are off the trend and will thus be left aside in the considerations of Sec. 3.4, as it was originally the case when the GD correlation was developed in [32].



**Figure 5.** Laminar burning velocities of NH<sub>3</sub>/H<sub>2</sub>/air mixtures at  $T_u = 298$  K and  $P_u = 0.1$  MPa. Symbols: 334 experiments; vertical error bars are 95% confidence intervals. Lines: GD correlation [32].

 The behavior of the GD correlation with respect to the hydrogen fraction is confirmed by LBV 336 measurements at higher temperatures, as shown in Figure 6 in the case  $\phi = 1.1$ , where the maximum LBV value is reached in most conditions. The measured and estimated values both exhibit exponential increasing trends as a function of the fuel hydrogen fraction for all temperatures, but with different slopes when depicted in a logarithmic scale. A general underestimate of the present data by the GD correlation is observed for low hydrogen fractions, while the LBVs at high hydrogen fractions are largely overestimated. 341 However, the correlation is in close agreement with data of Han et al. [22] up to 40% H<sub>2</sub> at  $T_u = 298$  K.

342 It should be noted that only the correlation for  $NH<sub>3</sub>/H<sub>2</sub>/air$  mixtures was plotted here. The NH<sub>3</sub>/air 343 correlation agrees much better with experimental values for  $x_{H_2} = 0$ , as previously seen in Fig. 4a. So, a 344 lack of continuity is observed between the NH<sub>3</sub>/H<sub>2</sub>/air and NH<sub>3</sub>/air correlations when the hydrogen fraction 345 becomes very small. Again, this is likely a consequence of the limited availability of accurate measurement 346 data for correlation fitting at the time it was established. Since most of the previously available data were 347 within the range  $0.3 \le x_{\text{H}_2} \le 0.6$ , the best correlation prediction is found here. It should be noted here that 348 the mechanism by Mathieu and Petersen [30], used also to establish the correlation, is found to 349 underestimate, respectively overestimate, the LBV of mixtures with low, respectively high hydrogen content

350 [32].

332



351

352 **Figure 6.** Laminar burning velocities of NH<sub>3</sub>/H<sub>2</sub>/air mixtures at  $P_u = 0.1$  MPa and  $\phi = 1.1$ . Filled symbols: 353 present measurements. Hollow triangles: measurements of Han et al. [22]. Lines: GD correlation [32].

354 This is partly confirmed by Figure 7 that shows the evolution of the LBV with  $x_{H_2}$  at 473 K. Present 355 measurements are compared with GD correlation values and simulations results using the detailed kinetic 356 mechanisms introduced in Sec. 2.5. In such conditions, the mechanism of Mathieu and Petersen 357 underestimates the experimental LBVs when  $\phi = 0.8$ , when  $\phi = 1.0$  and  $x_{H_2} < 0.5$  and when  $\phi = 1.2$  and 358  $x_{\text{H}_2}$  < 0.5. The mechanism of Otomo et al. is found to slightly underestimate the LBV in most cases, 359 especially for intermediate hydrogen fuel fractions. The mechanism of Nakamura et al. shows the best 360 overall agreement with the present data at elevated temperature as a function of the hydrogen fuel fraction. 361 The GD correlation estimates diverge significantly from both the experimental and kinetic modelling trends 362 when  $x_{H_2}$  increases, even though it is based on the Mathieu and Petersen mechanism. This is probably a 363 consequence of the scarcity and accuracy of the available experimental dataset at high hydrogen fractions 364 used to fit the correlation in the first place, through the correction factor  $\kappa$  that is applied here. The behavior 365 of the correlation without applying  $\kappa$  is much closer to the Mathieu and Petersen's mechanism (no plotted 366 here for readability), and yields thus accurate estimates for  $x_{H_2} \ge 0.5$ .



368 **Figure 7.** Laminar burning velocities of NH<sub>3</sub>/H<sub>2</sub>/air mixtures at  $P_u = 0.1$  MPa and  $T_u = 473$  K. Symbols: 369 experiments. Lines: numerical models.

## 370 *3.3. Effects of the temperature increase*

 Increasing the unburned gas temperature leads to an increase in the LBV and allows a more stable flame propagation in most cases. The respective influences of the equivalence ratio and hydrogen fraction on the LBV remain qualitatively unchanged when the temperature is increased: the experimental data distributions keep similar bell shapes in Fig. 4 and the slopes of the different distributions remain approximately parallel in Fig. 6 with increasing temperature.

376 The influence of the unburned gas temperature on the LBV is well estimated by the NH<sub>3</sub>/air GD correlation 377 (Fig. 4a), as well as by the NH<sub>3</sub>/H<sub>2</sub>/air GD correlation when  $x_{H_2} = 0.05$  for lean mixtures (Fig. 4b) and 378 when  $0.2 \le x_{H_2} \le 0.4$  (Figs. 4c and 4d). In order to isolate the temperature effect, the values of the 379 temperature exponent,  $\alpha$ , assuming an exponential temperature dependence of the LBV as in Eq. 1 are 380 depicted in Figure 8 as a function of the H<sub>2</sub> fraction. The experimental values of  $\alpha$ , as well as the associated 381 uncertainties were estimated by means of a Monte Carlo method to allow for the propagation of the 382 uncertainties on the LBV measurements as described in Sec. 2.4. For each test point and each iteration,  $\alpha$  is 383 determined as the slope of  $ln(s_u^0) = f(ln(T_u))$  by means of a least-squares linear regression, where  $s_u^0$  is 384 randomly chosen from a normal PDF with the average measured LBV as mean value and the global positive 385 LBV uncertainty as standard deviation. The mean value and standard deviation of the resulting  $\alpha$ -

386 distribution are taken as the best guess and uncertainty for  $\alpha$ . Results show satisfactory agreement with the values given by the GD correlations, as a function of both the equivalence ratio and the hydrogen fraction, indicating the ability of the correlation to accurately estimate the temperature effect in most cases. However, no agreement is found between the experimental and estimated temperature behavior when the fuel contains  $50\%$  H<sub>2</sub> or more.





392 **Figure 8.** Temperature dependence of the LBV.  $\alpha$ : temperature exponent as in Eq. 1.

 This is emphasized in Figure 9, where the experimental LBVs are shown as a function of the initial gas 394 temperature along with the GD correlation and results from kinetic simulations for  $x_{H_2} = 0.6$ . Again, the estimations by the GD correlation diverge away from the experimental trend and the kinetic model estimations, in particular those obtained with Mathieu and Petersen's mechanism for the same reasons mentioned in Sec. 3.2. While the different kinetic reaction mechanisms tested here show similar trends with respect to the temperature effect on the LBV, their estimations remain scattered and do not completely agree with the present measurements. While the mechanism of Otomo et al. generally underestimates the LBVs (as emphasized in the Supplementary Material), the mechanisms of Mathieu and Petersen agrees well, while the one of Nakamura et al. slightly overestimates the experimental data for rich mixtures. It should be remembered here that the error bars shown in Fig. 9 are considered to be conservative due to the use of the 403 GD correlation to estimate the uncertainty caused by the mixture composition error as explained in Sec. 2.4.





## 407 *3.4. Correlation optimization*

408 In an attempt to optimize the correlations proposed by Goldmann and Dinkelacker [32], the new 409 experimental dataset was used to adjust the coefficient correlations for  $P_u \le 0.5$  MPa and  $T_u \le 700$  K. To that 410 end, the present LBV experimental dataset was consolidated with the literature data [10,11,20–22,12–19] 411 and used to fit the correction factor  $\kappa$  by means of a least-squares algorithm. The form of the correction 412 factor was extended by introducing two new coefficients  $k_5$  and  $k_6$  to take into account the temperature 413 dependence, analogously to the pressure dependence:  $\kappa = k_1 k_2^{p_n} p_n^{k_3} (\frac{1}{\phi})^{k_4} k_5^{r_n} T_n^{k_6}$ . In each case, the 414 accuracy of the obtained correlation is assessed by the mean absolute percentage error  $M_{\phi}$ , the maximum 415 absolute percentage error  $M_{\text{max}}$ , the minimum absolute percentage error  $M_{\text{min}}$  and the coefficient of 416 determination  $R^2$  over the considered dataset of size N, as in [32]. Different comparisons were conducted 417 and are summarized for the NH<sub>3</sub>/air and NH<sub>3</sub>/H<sub>2</sub>/air correlations in Table 2 and Table 3, respectively.

418 Unsurprisingly, the GD correlation compares well with the consolidated experimental dataset for  $419$  NH<sub>3</sub>/air mixtures as seen in Table 2, since a satisfactory agreement was found with the new experimental 420 data at elevated temperatures in Sec. 3.1 and 3.3. The extended correlation with the new coefficients of 421 Table 4 presents a slightly worsened mean absolute percentage error but a better coefficient of determination

422 than the GD correlation.

**Table 2.** Accuracy of the NH<sub>3</sub>/air correlation against experiments for  $P_u \le 0.5$  MPa.



424

425 However, Table 3 shows that the new consolidated dataset does not compare well with the GD 426 correlation for NH<sub>3</sub>/H<sub>2</sub>/air mixtures, since  $M_{\phi}$  increases from 8.26 % to 11.56 % and  $R^2$  drops from 0.977 to 427 0.867. This was expected following the discrepancies observed in Sec. 3.1, 3.2 and 3.3 between the GD 428 correlation and the present experimental values, especially at high hydrogen fractions. For this reason, a 429 comparison between the GD correlation and the experimental dataset for  $0 < x_{H_2} \le 0.4$  was attempted. The 430 original correlation proved to be slightly more accurate in that case, though not reaching the accuracy shown 431 against the former experimental dataset. Thus, the present attempts to fit the new correlation on the new 432 consolidated dataset resulted in a poorer accuracy when considering the entire  $H_2$  range. Considering that the 433 original correlation behaves differently for small and large hydrogen fractions, it was decided to apply the 434 optimization only in the range  $0 < x_{\text{H}_2} \le 0.4$ , leading to an improvement in accuracy of the correlation 435 against the consolidated dataset, as seen in Table 3.

436 **Table 3.** Accuracy of the NH<sub>3</sub>/H<sub>2</sub>/air correlation against experiments for  $P_u \le 0.5$  MPa.





437

438 The corresponding coefficients in Table 4 can be used for the correction factor  $\kappa$  in LBV calculations in 439 the range  $0 \le x_{\text{H}_2} \le 0.4$ , all other coefficients in the GD correlations remaining unchanged compared to 440 [32]. However, the new correlation estimates the consolidated dataset less accurately than the original GD 441 correlation estimates the original dataset. Ultimately, a better understanding of the role of NH<sub>3</sub> and H<sub>2</sub> 442 reactions in the detailed kinetic mechanisms should help in better predicting the laminar burning velocity, 443 especially for mixtures containing very low or very high hydrogen fractions. Refined mechanisms could 444 then be used to establish more reliable correlations for reduced computational intensity.



445 **Table 4.** New coefficients for the empirical correction factor with temperature dependence.



446

## 447 **4. Summary and conclusions**

 An extensive new experimental dataset of ammonia/air and ammonia/hydrogen/air laminar burning velocities was obtained at atmospheric pressure for equivalence ratios ranging from 0.8 to 1.4, hydrogen 450 fractions in the fuel from 0 % to 60 % and unburned gas temperatures from 298 to 473 K by means of the outwardly propagating spherical flame method. Experimental hardware, imaging, radiation-induced and statistical errors were taken into account. Results are in good agreement with previous measurements from the literature obtained with the same method and the heat flux method under NTP, but discrepancies are observed with measurements obtained with the Bunsen burner method for high hydrogen fractions. More than 260 new data points are presented, that significantly enrich the literature data, especially at high temperatures and hydrogen fuel fractions.

 A comparison of the present measurements with the LBV correlations recently developed by Goldmann and Dinkelacker, as well as with chemical kinetic simulations using state-of-the-art reaction mechanisms, led to the following summary:

# 460 1. The original NH<sub>3</sub>/air correlation agrees well with the present experimental data for all temperatures, thus validating its ability to accurately approximate the LBV of mixtures at higher temperatures.

- 463 2. The original  $NH<sub>3</sub>/H<sub>2</sub>/air$  correlation underestimates the LBV of rich mixtures with low hydrogen fractions, while it generally overestimates the LBV of mixtures with high hydrogen fractions, especially at elevated temperature and regardless of the equivalence ratio. The LBV of mixtures with intermediate hydrogen fractions are estimated satisfactorily in most cases. This is explained by the better accuracy and availability of experimental data in that range at the time the correlation was first developed.
- 469 3. The influence of the temperature is reproduced satisfactorily by the  $NH<sub>3</sub>/H<sub>2</sub>/air$  correlation, 470 except for mixtures with  $x_{\text{H}_2} \geq 0.5$ , where the influence of the temperature is again overestimated.
- 4. Current reaction mechanisms show satisfactory agreement with the present experimental data over the whole range of investigation, but a significant scatter remains between them.
- 5. An attempt to optimize the correlations by fitting them on the new consolidated experimental dataset through an experimental correction factor was successful for a limited set of conditions only.

 As a result, while the detailed kinetic mechanisms may still require some fine tuning to improve their 478 accuracy, the development of new LBV correlations for  $NH<sub>3</sub>/H<sub>2</sub>$  fuels requires careful validation based on fully validated reaction mechanisms and accurate experimental data. However, the present optimization proposed for the GD correlations should allow to conduct turbulent combustion simulations with satisfactory accuracy and low computational cost for hydrogen fractions in the fuel smaller than 50%.

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## **References**

- [1] UNFCCC. Paris Agreement. Paris: 2015.
- [2] Rogelj J, Shindell D, Jiang K, Fifita S, Forster P, Ginzburg V, et al. Mitigation pathways compatible with 1.5°C in the context of sustainable development. IPCC Spec Rep 2018:93–174.
- 489 [3] Tatin A, Bonin J, Robert M. A Case for Electrofuels. ACS Energy Lett 2016:1062–4.
- doi:10.1021/acsenergylett.6b00510.
- [4] Zamfirescu C, Dincer I. Using ammonia as a sustainable fuel. J Power Sources 2008;185:459–65. doi:10.1016/j.jpowsour.2008.02.097.
- [5] Giddey S, Badwal SPS, Munnings C, Dolan M. Ammonia as a Renewable Energy Transportation
- Media. ACS Sustain Chem Eng 2017;5:10231–9. doi:10.1021/acssuschemeng.7b02219.
- [6] Cheema II, Krewer U. Operating envelope of Haber–Bosch process design for power-to-ammonia. RSC Adv 2018;8:34926–36. doi:10.1039/C8RA06821F.
- [7] Grinberg Dana A, Elishav O, Bardow A, Shter GE, Grader GS. Nitrogen-Based Fuels: A Power-to-Fuel-to-Power Analysis. Angew Chemie - Int Ed 2016;55:8798–805. doi:10.1002/anie.201510618.
- [8] Valera-Medina A, Xiao H, Owen-Jones M, David WIF, Bowen PJ. Ammonia for power. Prog Energy Combust Sci 2018;69:63–102. doi:10.1016/j.pecs.2018.07.001.
- [9] Kobayashi H, Hayakawa A, Somarathne KDKA, Okafor EC. Science and technology of ammonia combustion. Proc Combust Inst 2019;37:109–33. doi:10.1016/j.proci.2018.09.029.
- [10] Zakaznov VF, Kursheva LA, Fedina ZI. Determination of normal flame velocity and critical diameter of flame extinction in ammonia-air mixture. Combust Explos Shock Waves 1978;14:710–3.
- doi:10.1007/BF00786097.
- [11] Ronney PD. Effect of Chemistry and Transport Properties on Near-Limit Flames at Microgravity.
- Combust Sci Technol 1988;59:123–41. doi:10.1080/00102208808947092.
- [12] Pfahl UJ, Ross MC, Shepherd JE, Pasamehmetoglu KO, Unal C. Flammability limits, ignition energy,
- and flame speeds in H2-CH4-NH3-N2O-O2-N2 mixtures. Combust Flame 2000;123:140–58.

- doi:10.1016/S0010-2180(00)00152-8.
- [13] Jabbour T, Clodic DF. Burning Velocity and Refrigerant Flammability Classification. Trans Am Soc Heat Refrig Air Cond Eng 2004;110:522–33.
- [14] Takizawa K, Takahashi A, Tokuhashi K, Kondo S, Sekiya A. Burning velocity measurements of nitrogen-containing compounds. J Hazard Mater 2008;155:144–52.
- doi:10.1016/j.jhazmat.2007.11.089.
- [15] Hayakawa A, Goto T, Mimoto R, Arakawa Y, Kudo T, Kobayashi H. Laminar burning velocity and Markstein length of ammonia/air premixed flames at various pressures. Fuel 2015;159:98–106. doi:10.1016/j.fuel.2015.06.070.
- [16] Davis SG, Pagliaro JL, Debold TF, van Wingerden M, van Wingerden K. Flammability and explosion
- characteristics of mildly flammable refrigerants. J Loss Prev Process Ind 2017;49:662–74.
- doi:10.1016/j.jlp.2017.05.019.
- [17] Mei B, Zhang X, Ma S, Cui M, Guo H, Cao Z, et al. Experimental and kinetic modeling investigation on the laminar flame propagation of ammonia under oxygen enrichment and elevated pressure
- conditions. Combust Flame 2019;210:236–46. doi:10.1016/j.combustflame.2019.08.033.
- [18] Lee JH, Lee SI, Kwon OC. Effects of ammonia substitution on hydrogen/air flame propagation and emissions. Int J Hydrogen Energy 2010;35:11332–41. doi:10.1016/j.ijhydene.2010.07.104.
- [19] Lee JH, Kim JH, Park JH, Kwon OC. Studies on properties of laminar premixed hydrogen-added
- 
- ammonia/air flames for hydrogen production. Int J Hydrogen Energy 2010;35:1054–64.
- doi:10.1016/j.ijhydene.2009.11.071.
- [20] Li J, Huang H, Kobayashi N, He Z, Nagai Y. Study on using hydrogen and ammonia as fuels:
- Combustion characteristics and NOx formation. Int J Energy Res 2014;38:1214–23.
- doi:10.1002/er.3141.
- [21] Ichikawa A, Hayakawa A, Kitagawa Y, Kunkuma Amila Somarathne KDD, Kudo T, Kobayashi H.
- Laminar burning velocity and Markstein length of ammonia/hydrogen/air premixed flames at elevated pressures. Int J Hydrogen Energy 2015;40:9570–8. doi:10.1016/j.ijhydene.2015.04.024.
- [Tapez ici] [22] Han X, Wang Z, Costa M, Sun Z, He Y, Cen K. Experimental and kinetic modeling study of laminar
- 537 burning velocities of NH3/air, NH3/H2/air, NH3/CO/air and NH3/CH4/air premixed flames.
- Combust Flame 2019;206:214–26. doi:10.1016/j.combustflame.2019.05.003.
- [23] Kumar P, Meyer TR. Experimental and modeling study of chemical-kinetics mechanisms for H2-
- NH3-air mixtures in laminar premixed jet flames. Fuel 2013;108:166–76.
- doi:10.1016/j.fuel.2012.06.103.
- [24] Miller JA, Smooke MD, Green RM, Kee RJ. Kinetic Modeling of the Oxidation of Ammonia in
- Flames. Combust Sci Technol 1983;34:149–76. doi:10.1080/00102208308923691.
- [25] Lindstedt RP, Lockwood FC, Selim MA. Detailed kinetic modelling of chemistry and temperature effects on ammonia oxidation. Combust Sci Technol 1994;99:253–76.
- doi:10.1080/00102209408935436.
- [26] Tian Z, Li Y, Zhang L, Glarborg P, Qi F. An experimental and kinetic modeling study of premixed
- NH3/CH4/O2/Ar flames at low pressure. Combust Flame 2009;156:1413–26.
- doi:10.1016/j.combustflame.2009.03.005.
- [27] Konnov AA. Implementation of the NCN pathway of prompt-NO formation in the detailed reaction mechanism. Combust Flame 2009;156:2093–105. doi:10.1016/j.combustflame.2009.03.016.
- [28] Okafor EC, Naito Y, Colson S, Ichikawa A, Kudo T, Hayakawa A, et al. Experimental and numerical
- study of the laminar burning velocity of CH4–NH3–air premixed flames. Combust Flame
- 2018;187:185–98. doi:10.1016/j.combustflame.2017.09.002.
- [29] Cavaliere R, Costa M, Bai XS, da Rocha RC, Costa M, Bai XS, et al. Chemical kinetic modelling of ammonia/hydrogen/air ignition, premixed flame propagation and NO emission. Fuel 2019;246:24–33. doi:10.1016/j.fuel.2019.02.102.
- 
- [30] Mathieu O, Petersen EL. Experimental and modeling study on the high-temperature oxidation of
- Ammonia and related NOx chemistry. Combust Flame 2015;162:554–70.
- doi:10.1016/j.combustflame.2014.08.022.
- [31] Otomo J, Koshi M, Mitsumori T, Iwasaki H, Yamada K. Chemical kinetic modeling of ammonia
- oxidation with improved reaction mechanism for ammonia/air and ammonia/hydrogen/air
- combustion. Int J Hydrogen Energy 2018;43:3004–14. doi:10.1016/j.ijhydene.2017.12.066.

- [32] Goldmann A, Dinkelacker F. Approximation of laminar flame characteristics on premixed
- ammonia/hydrogen/nitrogen/air mixtures at elevated temperatures and pressures. Fuel 2018;224:366– 78. doi:10.1016/j.fuel.2018.03.030.
- [33] Metghalchi M, Keck JC. Burning velocities of mixtures of air with methanol, isooctane, and indolene at high pressure and temperature. Combust Flame 1982;48:191–210. doi:10.1016/0010-
- 2180(82)90127-4.
- [34] Galmiche B, Halter F, Foucher F. Effects of high pressure, high temperature and dilution on laminar burning velocities and Markstein lengths of iso-octane/air mixtures. Combust Flame 2012;159:3286– 99. doi:10.1016/j.combustflame.2012.06.008.
- [35] Brequigny P, Endouard C, Mounaïm-Rousselle C, Foucher F. An experimental study on turbulent
- premixed expanding flames using simultaneously Schlieren and tomography techniques. Exp Therm
- Fluid Sci 2018;95:11–7. doi:10.1016/j.expthermflusci.2017.12.018.
- [36] Huo J, Yang S, Ren Z, Zhu D, Law CK. Uncertainty reduction in laminar flame speed extrapolation for expanding spherical flames. Combust Flame 2018;189:155–62.
- doi:10.1016/j.combustflame.2017.10.032.
- [37] Kelley AP, Law CK. Nonlinear effects in the extraction of laminar flame speeds from expanding
- spherical flames. Combust Flame 2009;156:1844–51. doi:10.1016/j.combustflame.2009.04.004.
- [38] Ronney PD, Sivashinsky GI. A Theoretical Study of Propagation and Extinction of Nonsteady
- Spherical Flame Fronts. SIAM J Appl Math 1989;49:1029–46. doi:https://doi.org/10.1137/0149062.
- [39] Halter F, Tahtouh T, Mounaïm-Rousselle C. Nonlinear effects of stretch on the flame front
- propagation. Combust Flame 2010;157:1825–32. doi:10.1016/j.combustflame.2010.05.013.
- [40] Moffat RJ. Describing the uncertainties in experimental results. Exp Therm Fluid Sci 1988;1:3–17. doi:10.1016/0894-1777(88)90043-X.
- [41] Brequigny P, Uesaka H, Sliti Z, Segawa D, Foucher F, Dayma G, et al. Uncertainty in measuring
- laminar burning velocity from expanding methane-air flames at low pressures. 11th Mediterr.
- Combust. Symp., Tenerife, Spain: 2019.
- [Tapez ici] [42] Yu H, Han W, Santner J, Gou X, Sohn CH, Ju Y, et al. Radiation-induced uncertainty in laminar
- flame speed measured from propagating spherical flames. Combust Flame 2014;161:2815–24.
- doi:10.1016/j.combustflame.2014.05.012.
- [43] Nakamura H, Shindo M. Effects of radiation heat loss on laminar premixed ammonia/air flames. Proc
- Combust Inst 2019;37:1741–8. doi:10.1016/j.proci.2018.06.138.
- [44] Nakamura H, Hasegawa S, Tezuka T. Kinetic modeling of ammonia/air weak flames in a micro flow
- reactor with a controlled temperature profile. Combust Flame 2017;185:16–27.
- doi:10.1016/j.combustflame.2017.06.021.
- [45] ANSYS Chemkin-Pro 2019.
-