



Experimental study of 48600 Carbons fabrics behavior using marks tracking technique method

Mondher Nasri, Christian Garnier, Fethi Abbassi, Romain Brault, Olivier Dalverny, Ali Zghal

► To cite this version:

Mondher Nasri, Christian Garnier, Fethi Abbassi, Romain Brault, Olivier Dalverny, et al.. Experimental study of 48600 Carbons fabrics behavior using marks tracking technique method. 6th International Symposium on Aircraft Materials, ACMA 2016, May 2016, Agadir, Morocco. pp.1-8. hal-02359870

HAL Id: hal-02359870

<https://hal.science/hal-02359870>

Submitted on 12 Nov 2019

HAL is a multi-disciplinary open access archive for the deposit and dissemination of scientific research documents, whether they are published or not. The documents may come from teaching and research institutions in France or abroad, or from public or private research centers.

L'archive ouverte pluridisciplinaire **HAL**, est destinée au dépôt et à la diffusion de documents scientifiques de niveau recherche, publiés ou non, émanant des établissements d'enseignement et de recherche français ou étrangers, des laboratoires publics ou privés.



Open Archive Toulouse Archive Ouverte (OATAO)

OATAO is an open access repository that collects the work of some Toulouse researchers and makes it freely available over the web where possible.

This is an author's version published in: <http://oatao.univ-toulouse.fr/23302>

Official URL:

To cite this version:

Nasri, Mondher and Garnier, Christian and Abbassi, Fethi and Brault, Romain and Dalverny, Olivier and Zghal, Ali
Experimental study of 48600 Carbons fabrics behavior using marks tracking technique method. (2016) In: 6th
International Symposium on Aircraft Materials, ACMA 2016, 11 May 2016 - 13 May 2016 (Agadir, Morocco).

Any correspondence concerning this service should be sent to the repository administrator:

tech-oatao@listes-diff.inp-toulouse.fr

Experimental study of 48600 Carbons fabrics behavior using marks tracking technique method

M. Nasri[†], C. Garnier^{*}, F. Abbassi^{†‡}, R. Braut^{**}, O. Dalverny^{*}, A. Zghal[†]

[†]URMSSDT, ENSIT, University of Tunis, 5 Av. Taha Hussein, BP 56, Bâb Manara, 1008 Tunisia

^{*}University of Toulouse, INP-ENIT/LGP, 47 Av. d'Azereix, 65016 Tarbes, France.

[‡]Mechanical Engineering Department, College of Engineering, Dhofar University, PO Box 2509, Postal Code 211 Salalah, Sultanate of Oman

^{**}Nimitech Innovation, 1 Av. des victimes du 11 juin 1944, 65200 Bagnères-de-Bigorre, France

Abstract

Lightweight and energy saving are the main challenges in the aircraft industry production, that explain the increase of composite demand and the diversity of its applications. The investigation of the shear behavior and stiffness of technical woven fabric are essential to guarantee the performance of the final product. In case of forming process (for example RTM process), the in-plane deformability of the woven fabric is necessary for forming without creating defects. The change of the fiber orientation (warp and weft) have a significant impact on final mechanical properties. In this study, the use of marks tracking technique allow the determination of the rigidity of 48600 C 1300 carbon fabrics, and allow calculation of their shear angle, lock angle during tensile and bias-extension tests.

1. Introduction

In recent years, composite materials have been used increasingly in industries. Their technical specifications, as the ratio of strength /weight, corrosion resistance, and mechanical resistance will lead to major advances e.g. the decreasing of the weight of structures or the vibration damping. In the aerospace industry, the composite materials are used extensively. The forming process of woven composites can be classified under two categories: liquid composite molding (LCM) process and using process of semi-finished products (prepregs, compounds). The choice of the manufacturing process in the industry is done by a high speed of production and a high quality of the final product. Several compression molding processes exist (cold compression, hot pressing) [1]. Double curvature parts of polymer matrix composites reinforced with woven fabric can be elaborated using resin transfer molding (RTM) process, in which firstly, the dry woven reinforcements are formed and secondly, the resin is injected to obtain the final shape [2, 3]. This method has several disadvantages according to different points of view: difficulty to control the final shape and the effective mechanical properties, high costs implementation, low rates production. Some others problems are related to the preform deformation draped: tensions fibers between longitudinal and transverse direction, fiber misorientations, wrinkling or local buckles [5]. The quality of the preformed shape can be analyzed by the estimation of the woven elongation, the shear angle and the final orientation of fibers. In order to understand the behavior of the carbon fabrics during preforming process, it is strongly recommended to carry out some basic mechanical tests. These tests lead to the determination of the physical properties of dry reinforcements (stiffness modulus, shear modulus, Poisson's ratio, locking angle), which will be introduced in the forming simulations. In this context, we propose to study the 48600 carbon fabrics. Two mechanical tests are performed: a tensile test and a bias test. During this

tests, an advanced measurement method called ‘marks tracking technique’ is used to predict the displacement of yarns and their blocking angle. This optical method consists in tracking some specific points stated on the front face of sample. The recorded displacement of the points allows the determination of the shear angle and the strain of the samples. A comparison between measured and theoretical shear angles was done. The low shear rigidity of the 48600 fabric has been evaluated in the results of the bias tests [6].

2. Characterization methodologies of woven fabric

To characterize the shear behavior of the woven, several investigations can be conducted from various tests. The KES-F test (Kawabata Evaluation System for Fabrics), uses a woven specimen size of 200 mm x 50 mm. The sample is maintained by two opposed stop and a shear load is applied at constant speed [7]. The “picture frame” test is a metal square articulated with four arms, developed for woven textiles. This test consists in imposing a pin-joint restriction on the shear specimen in order to produce homogenous kinematics in the specimen. However, it is extremely sensitive to specimen misalignment issues [8, 7]. The “bias test” or “bias extension test” is used to characterize the shear properties of the fabrics. The shear solicitation is pure on the specimen but non-homogenous along it [9]. For this test, the specimen is relatively insensitive to misalignment. It is simple to implement, but some precautions are taken during the preparing of the specimen, such as ensuring the correct orientation of the yarns (figure 1). The advantage of bias test relative to the picture frame is the absence of tensile stress on yarns in central of the specimen [10].

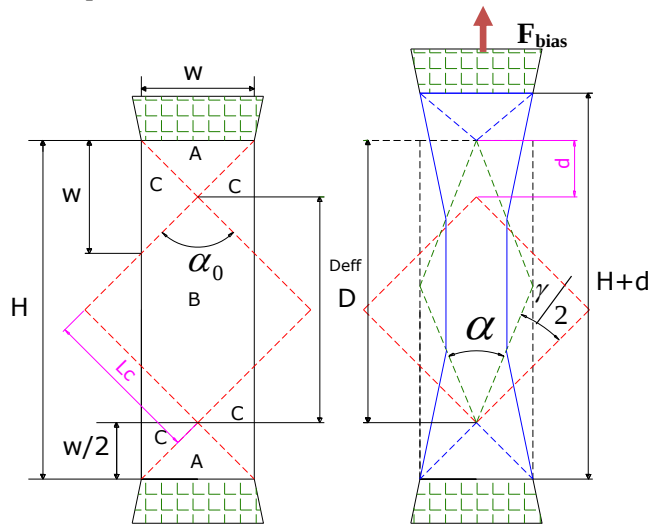


Figure 1: Schematics shapes of the bias extension test (undeformed and deformed)

During bias extension tests, the angle α , between warp and weft, gradually decreases. Such case leads to a compaction of yarns followed by a locking until it reaches a threshold of shear angle denoted γ_{lock} [8]. The shear locking angle is defined as a limit angle before creation of defects. After this threshold is reached, the formation of local wrinkles begins, producing a defect on shaping composite parts [4]. In this test, three regions, called A, B and C (Figure 1), can be clearly identified [7]. The first area ‘A’ corresponds to the attachment areas of the specimen without shear. The second one ‘B’, located in the center of the specimen corresponds to the pure sheared area. Finally, several halves shear zones correspond to portions containing warp and weft yarns of which both ends have free (partially sheared areas), as presented in Figure 1, (Zone C).

Using a simple kinematic analysis, in the central area, the theoretical shear angle of zone B can be calculated [11]:

$$\gamma = \frac{\Pi}{2} - \alpha = \frac{\Pi}{2} - 2 \arccos \left(\frac{D+d}{\sqrt{2}D} \right) \quad (1)$$

Where D is the initial diagonal length of the zone B and d is the displacement of the mobile clamps of the tensile machine. This expression is only valid with the following hypotheses: warp and weft yarns rotate without slipping and the strain of yarns is homogeneous partially. The shear angle can be calculated in the central region using a fictive shear frame, according to the extension of the test specimen, and its initial dimensions [10]. Hence, the shear angle is calculated by the following formulae:

$$\gamma = \frac{\Pi}{2} - \alpha = \frac{\Pi}{2} - 2 \cos^{-1} \left(\frac{\sqrt{2}}{2} \left(1 + \frac{d}{H-W} \right) \right) \quad (2)$$

3. Experimental setup

3.1. Material

To determine the behavior of woven carbon reinforcements, tensile and bias extension tests were carried out following three directions (0°, 45° and 90°). In this work, the used carbon fabric is the HexForce 48600 1300 C [6]. This fabric is a twill 2 × 2 weave with a 0.62 thickness ply. This fabric is considered to be balanced (50% of yarns longitudinally and transversally). The fabric properties are presented in the Table 1.

Table1: Properties of the carbon fabric 48600 C 1300

Designation	Weave style	Type of yarns	Nominal weight	Nominal construction
48600 C 1300	Twill 2×2	T700SC 12K 50C	600 g/m ²	3.7 yarns/cm

3.2. Sample design

In the literature, there is no standard form of specimens for fabric characterization tests [8, 10]. Several studies specify the choice of specimen dimension in order to get a pure shear zone in the center of specimen: the ratio λ between the length and the width of the specimen should not be smaller than 2 [10, 7]. In this study, this ratio is 3. The geometrical details are shown in figure 2.

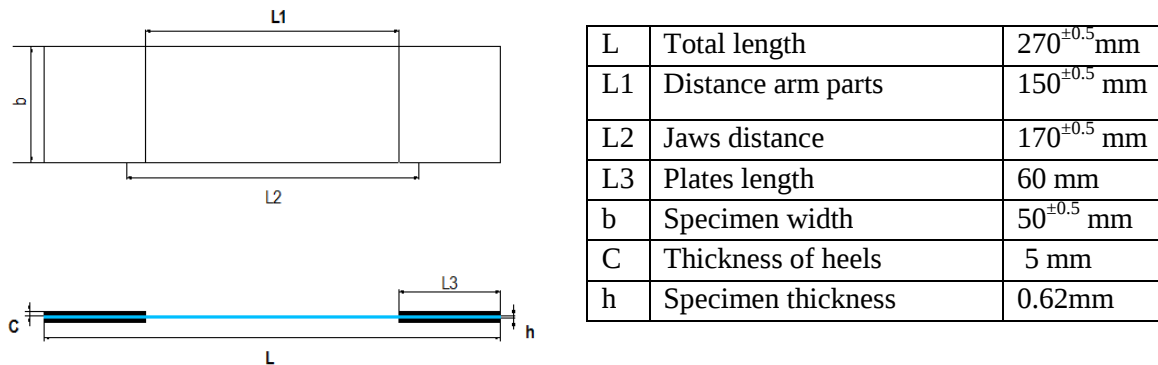


Figure 2: Geometry of fabric specimen

3.3. Displacement measurement by marks tracking method

The fundamental principle of marks tracking method is to mark previously dry reinforcements with painted points at strategic locations, as illustrated in Figure 3. This procedure allows the determination of the position of each markers for every configurations, and calculates the displacements of these points between initial and deformed configurations. The objective of such an apparatus is to follow, in real time, the behavior of yarns by tracking markers. Two CCD cameras are positioned to measure and systematically register two configurations, initial and final of yarns. In fact, the calculation of the 3D marker positions is done from a standard network of markers whose positions of the left camera are known (x_g, y_g, z_g) in the reference right camera (X_d, Y_d, Z_d) . The measurement of positions during loading is performed by a stereo-image correlation system.

The used DIC system is the GOM ATOS system with the software Aramis. The numerical image acquisition is 1392 x 1040 pixels and the resolution of the measure of surface displacement is less than 15 μ m. Images are recorded during the test and analysed by the software Aramis, developed by GOM, in order to determine the full field displacements of markers [12]. The focal length of objectives used is 35mm for the tests at 0°/90° and 12 mm for the tests at 45°.

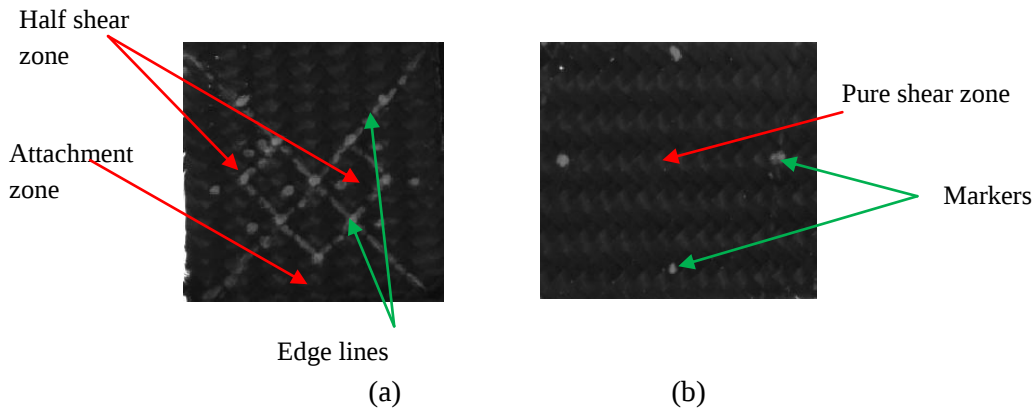


Figure 3: Marking of Woven 48600 C1300

4. Mechanical Tests

4.1. Test 0°/90°

Tensile tests along warp and weft directions (0° / 90°) are carried out on samples presented in chapter 3.2. In order to avoid slipping between the specimen and the jaws, we have used dimensions steel plates for clamping (60 mm × 50 mm). The tests are performed at a test speed of 10 mm/min on an Instron 550R testing machine (figure 5). The load cell is 100 kN, and the controlled room temperature is 22 ° C. The fabric is placed without any pre-loading; the displacement and load are set to “0” as references of the test.

4.2. Bias-extension test

Bias tests are used to characterize the shear mechanisms of fabric reinforcement. Considering the low values of the forces involved during this test, a 500 N load cell is used on the Instron

550R tensile machine. These tests are carried at room temperature (22°C) at a test speed of 10 mm/min. and load–displacement diagrams are recorded. The sample length/width ratio λ equals to 3 and the initial angle between warp and weft is 90°.

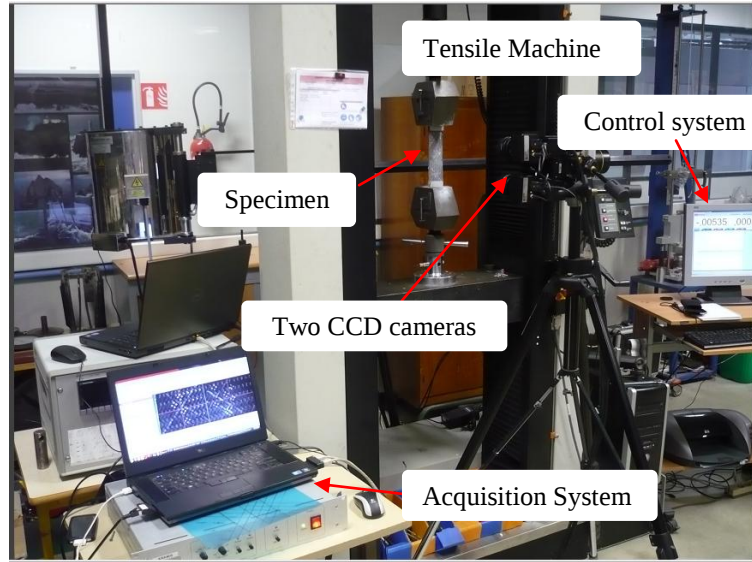


Figure 4: Equipment test

5. Results and discussion

5.1 Tensile tests results

The figure 5 shows the typical load–displacement curve for the 0° tensile test, which shows the variation of the force depending on the displacement of the clamping jaws. This curve is composed of three distinctive parts. In fact, the first one is nonlinear until displacement of 1.7 mm. This stage corresponds to the setting up of the woven, reducing the crimp phenomenon in this direction until the yarn is in tension. At the yarn strain beginning, the stiffness is non-linear and is increasing until a linear trend. The second part is linear and corresponds to the tension of the fabric until maximum load (around 9200 N). The last part is the progressive deterioration of specimen until full unweaving of the fabric. The stiffness of the fabric K is calculated using the mean slope in part 2. In this case, $K_{\text{fabric}0^\circ} = 4897 \text{ N/mm}$.

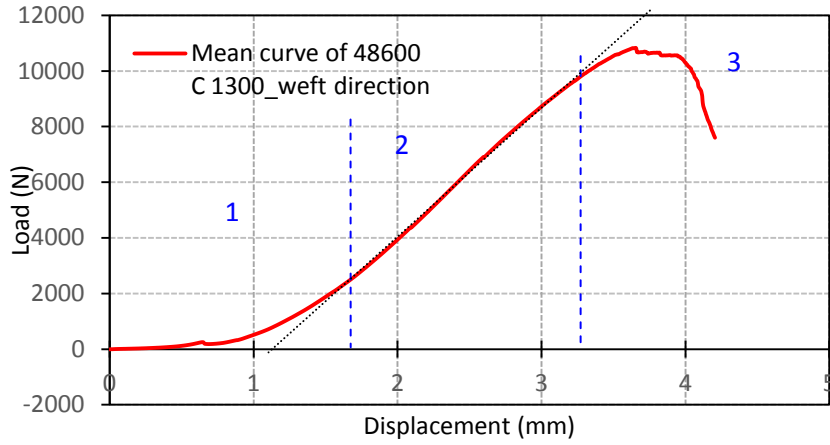


Figure 5 : : Results of tensile test of 48600 C 1300 in weft direction

5.2 Results of the bias extension

The response of the global fabric deformation 48600 C 1300 in the 45° direction is shown in Figure 6. It is observed that the load recorded is very small. In fact, this load increases to a maximum value of 5.5 N, and then decreases progressively. As for the resultant displacement, it reaches 50 mm at failure. Figure 7 shows the evolution of the measured shear angle during the test versus the displacement in the central area of specimen. This measured shear angle is compared with the theoretical values calculated with Eq. (2). A good correlation is observed between measured and calculated angle until 23°. After this value, a deviation between the theoretical shear angle and measured shear angle is observed, as a result of slipping at the frontier between the three areas of the specimen. In order to validate the tests, the evolution of shear angles at seven positions, in three zones, are measured, as illustrated in figure 8. It can be seen that the distribution of shear angles α_i and β_i measured in zone B is uniform, similar observation of which can be found in [13]. Therefore, the slip appearance at the interface between the zones allows stopping the increase of shear in the central area.

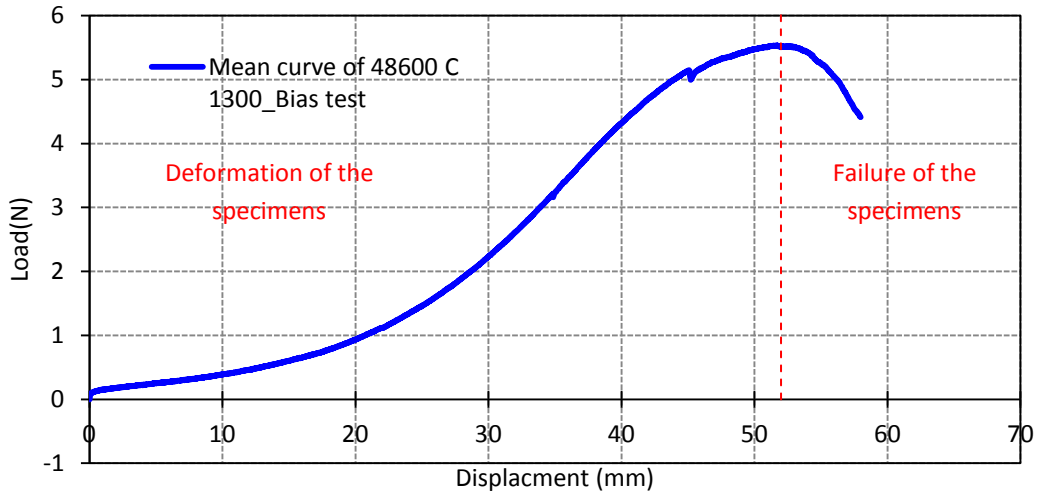


Figure 6: Results of bias test of 48600 C 1300

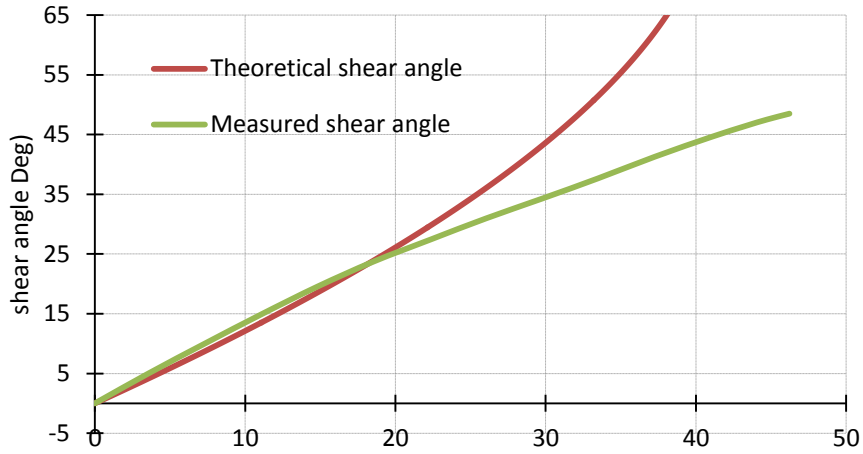


Figure 7: Theoretical shear angles and measured of 48600 C 1300

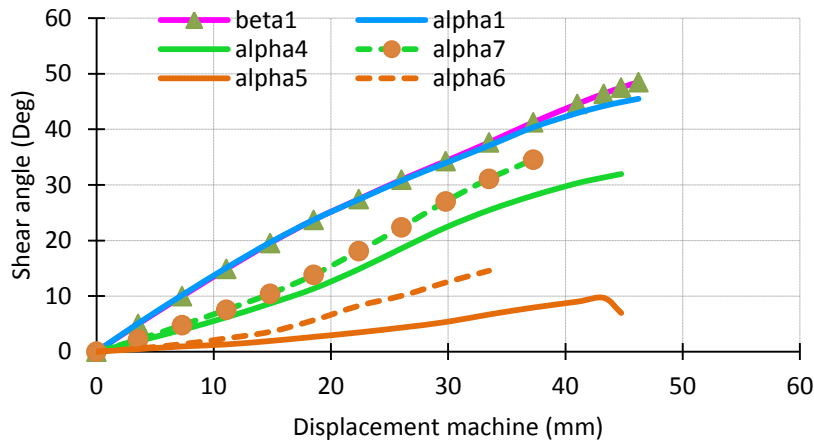


Figure 8: Evolution of the shear angle during bias test



6. Conclusion

This paper presents an experimental study of mechanical characterization of fabric 48600 C 1300. Uniaxial tensile tests and bias-extension tests are carried out with the application of the marker tracking method. This technique has two interests: the visualization of the deformation of the yarns and the tracking of different markers during tests. This experimental setup allow the determination of the displacement fields and the calculation of fabric stiffness and shear angle. This experimental setup is validated by the calculation and the correlation of different shear angle in each area. Measured shear angle in area C is the half the shear angle in area B and the shear angle in area A is very low. These measurements validate the hypothesis of the bias tests.

References

- [1] C.Garnier. Etude du comportement dynamique des structures composites réalisées par LRI : application à l'impact et à la fatigue, PhD ,NP,Toulouse,2011.
- [2] D.Soulat, S. Allaoui, and S. Chatel, J. Material Forming.Experimental device for the preforming step of the RTM process.2(2009) 181-184.
- [3] J.LAUNAY, A.V. Duong, G.Hivet, V.VALLE, and S. ALLAOUI. La mesure des déformations pour l'étude du comportement et la mise en forme des renforts de composites. 18ème Congrès Français de Mécanique (Grenoble), 2007.
- [4] C.TEPHANY, Analyse de la formabilité de renforts composites à base de fibres naturelles, PhD thesis of INSA,Lyon,2014.
- [5] P.Boisse, M.Borr, K.Buet,and A.Cherouat,J. Composites Part B: Engineering.Finite element simulations of textile composite forming including the biaxial fabric behaviour.28(1997)453-464.
- [6] Hexcel, Hexflow® 48600 C 1300, 2015.
- [7] S. DRIDI,Essais de caractérisation des structures tissées, PhD de école centrale de Lyon 2010.
- [9] X. Q. Peng , and J. Cao. J. Composites Part A: Applied Science and Manufacturing, Acontinuum mechanics-based non-orthogonal constitutive model for woven composite fabrics.36(2005) 859-874.
- [10] E. De Luycker, F. Morestin, P. Boisse, and D. Marsal ,J. composite structures Simulation of 3D interlock composite performing.88(2009). 615-623.
- [11] J. Cao, R. Akkerman, P. Boisse, J. Chen, H.S.Cheng, E.F.De Graaf, J. Composites Part A,Applied Science and Manufacturing Characterization of mechanical behavior of woven fabrics: experimental methods and benchmark results.39(2008) 1037-1053.
- [12] C. Goidescu, H. Weleman, C. Garnier, M. Fazzini, R. Brault ,J. Composites Part B: Engineering, Damage investigation in CFRP composites usingfull-field measurement techniques: combination of digitalimage stereo-correlation, infrared thermography andX-ray tomography.48(2013) 95-105.
- [13] L.Li, Y. Zhao, Y. Chen, J. Yang, and Y. Duan, J. Materials and Design. In-plane shear investigation of biaxial carbon non-crimp fabrics with experimental tests and finite element modeling. 63(2014) 757-765.