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Vinayak Jagannathrao Kalas, Alain Vissiere, Thierry Roux, Olivier Company, Sébastien Krut, et al.. A New Efficient Stiffness Evaluation Method to Improve Accuracy of Hexapods. International Design Engineering Technical Conferences and Computers and Information in Engineering (IDETC/CIE 2018), Aug 2018, Quebec City, Canada. pp.V05AT07A045, 10.1115/DETC2018-85986. hal-02025724

# HAL Id: hal-02025724 https://hal.science/hal-02025724v1

Submitted on 2 Mar 2021

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# DETC2018-85986

# A NEW EFFICIENT STIFFNESS EVALUATION METHOD TO IMPROVE ACCURACY OF HEXAPODS

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## ABSTRACT

Structural compliance of hexapod positioners limits their positioning accuracy. Taking a step towards solving this problem, this paper proposes a new efficient method to evaluate the stiffness of hexapods in order to predict and correct their positioning error due to compliance. The proposed method can be used to predict the six degree of freedom deflection of the platform under load. This method uses a simple lumped stiffness parameter model whose parameters can be estimated using the identification technique presented in this paper. An experimental study with micrometer level measurements performed on a hexapod based micro-positioning system is used to assess the efficiency of the presented method.

## INTRODUCTION

Hexapods, also called Gough-Stewart platforms, are increasingly used for applications demanding high accuracy six degree of freedom (DOF) positioning. Some general areas of applications include mirror positioning in telescopes [1], posidemanding high accuracy positioning [3]. One of the biggest advantages of hexapods is their high inherent stiffness [4]. Hence, in most applications, hexapod compliance doesn't considerably degrade the accuracy of positioning. However, as accuracy requirements become more stringent, quantifying the hexapod's compliance error becomes crucial in order to correct it. This work forms a part of a larger project, Posilab [5], that is focused on improving the performance of the current state of the art hexapod positioners. The work presented in this paper focusses on solving the problem of platform deflection due to hexapod compliance. The developed solution needed to be applicable to existing hexapods. Hence, redesigning the hexapod was not a feasible option. It was, therefore, necessary for the solution developed in this work to be applicable in a position compensation framework. This can be achieved by using a model-based approach to estimate hexapod compliance. This approach would encompass choosing a suitable stiffness parameter model followed by identification of these parameters using appropriate measurements [6].

tioning components in synchrotrons [2] and research applications

Stiffness modelling of parallel robots has acquired vast attention in research leading to development of very simple stiff-

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ness matrix [7] to an advanced one [8]. To be included in a position compensation framework, it is crucial to accurately predict platform deflection. This needs accurate prediction of the parameters of the stiffness model. This problem has been mainly approached in research in the following directions: (a) analytical estimation of stiffness model parameters, this option has been widely addressed in literature [9–16], and (b) stiffness parameter identification using measurements, which is a relatively new approach [6,17–23]. In many cases, analytically estimated stiffness parameters do not lead to accurate deflection predictions of endeffector/platform. Also, the latter option has the advantage of being analytically simple and less time-consuming as it doesn't involve cumbersome calculations. These reasons make option (b) very attractive for implementation in an industrial framework and hence, this method is suitable for our application.

The accuracy of robot model parameters identified using measurements is largely dependent on the robot poses used, external forces used and the number of measurements performed. Therefore, the identification procedure needs to be optimized to obtain best set of parameters. The concept of parameter observability addresses this issue. This has been discussed in detail in the 'Stiffness identification method' section.

Stiffness identification technique has been applied very well to serial robots [6], [18–21], [24]. One of the most advanced stiffness identification methods for serial robots has been presented by Dumas et al. [21]. In this work, the observability of parameters was defined based on an advanced stiffness model. Klimchik et al. [18] also presented a robust stiffness identification method that used a simple stiffness model. They optimized their identification to improve accuracy at the intended operational configuration of a 6-R serial robot.

Stiffness identification has not been used much for evaluating stiffness of parallel robots yet. Carbone et al. [22] evaluated the stiffness of a 3-DOF parallel manipulator CaPaMan by measuring displacements of the platform under load. However, this method was not optimized to obtain the best set of parameters. Bonnemains et al. [23] evaluated the stiffness of a 3-DOF tricept parallel robot. This work, too, did not address parameter observability satisfactorily. No work so far, to the best of authors' knowledge, has demonstrated a complete and robust stiffness identification method for parallel robots. Also, no stiffness identification method has been presented for a hexapod system.

The method to be developed for the application concerned to this paper had to be directly applied in the industry. It was, therefore, very important for the stiffness evaluation method to be simple while being accurate enough. It is also desirable to have a less time consuming method in an industrial framework. These factors facilitate the use of a simple stiffness matrix as presented in [7].

This paper presents a method to identify the components of the Cartesian stiffness matrix of hexapods by means of measurements. This method only requires a small number of tests to be carried out. The method uses a simple lumped stiffness parameter model that uses only one spring per leg to model the overall stiffness. An observability index is then used to find the best set of poses and forces for parameter identification. The choice of this index is also discussed. Finally, the efficiency of the proposed method is validated by means of experiments: the predicted 6-DOF deflection of platform under load is compared to the measured values. The experimental study is performed on a micro-positioning system developed by Symétrie [25], a manufacturer of high-precision hexapod positioners. The analysis in this paper is demonstrated using a spherical-prismatic-spherical (SPS) hexapod positioner. However, this method is also valid for other sorts of hexapods: universal-prismatic-universal (UPU), UPS and SPU.

This paper is organised as follows: Firstly, the kinematics and stiffness model of a general SPS hexapod is shown. The method for stiffness identification is then presented. This is followed by a brief description of the measurement technique used to measure the 6-DOF pose parameters and displacement of the hexapod. The implementation and experimental validation of the stiffness identification method is then presented. The results and the intended future work are then discussed followed by conclusions on the work presented in this paper.

#### KINEMATICS AND STIFFNESS MODEL

Fig. 1 shows the schematic of a general SPS hexapod. Each leg consists of a SPS chain that connects to the base at points  $A_i$  and to the platform at points  $B_i$ . Two frames of references are defined for the sake of analysis, the fixed frame of reference  $R_f$  and the mobile frame of reference  $R_m$  (fixed to the platform).  $R_f$  and  $R_m$  align exactly with each other when the platform is in nominal position. The connection points  $A_i$  and  $B_i$  lie on circles with centers  $O_b$  and  $O_m$ , respectively.

The platform pose, given by *X*, can be written as,

$$X = [a, b, c, \alpha, \beta, \gamma]^T \tag{1}$$

The first three elements of *X* represent the translation along X, Y and Z axes of the mobile frame of reference,  $R_m$ .  $\alpha$ ,  $\beta$  and  $\gamma$  are the angular rotations of the platform around X, Y and Z axes of the same frame. The angles are defined as per X-Y-Z extrinsic Euler convention. *q* contains the lengths of six prismatic links.

$$q = [q_1, q_2, q_3, q_4, q_5, q_6]^T$$
(2)

A simple lumped stiffness model (as shown in Fig. 2) is used to model the static stiffness characteristics of this mechanism. One spring is used to model the stiffness of each leg ( $k_{i=1 \text{ to } 6}$ ). This model assumes that the actuators are the main contributors of

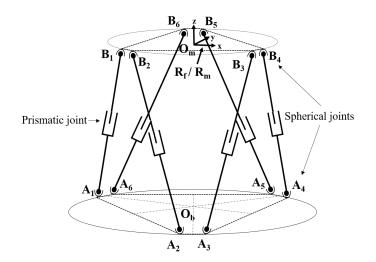


FIGURE 1. Schematic of a SPS hexapod

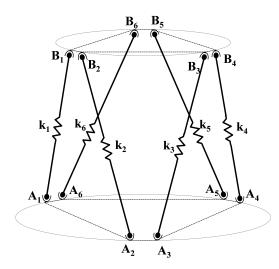


FIGURE 2. Lumped stiffness model of the SPS hexapod

compliance and that the rest of the structure is rigid. The joints are assumed to be frictionless. Also, the legs of the hexapod are assumed massless and considered to be subjected to axial loads only.

The relationship between the platform velocity vector and the joint velocity vector is given by,

$$\dot{q} = J^{-1}V \tag{3}$$

Here, V contains the components of velocity of the platform. J is the jacobian matrix of the hexapod which is evaluated as per method outlined in [26]. Now, the relationship between differ-

ential leg lengths and the differential platform pose ( $\Delta q$  and  $\Delta X$ ) can be written as,

$$\Delta q = \widetilde{J}^{-1} \Delta X \tag{4}$$

Here,  $\widetilde{J}^{-1}$  is given by,

$$\widetilde{J}^{1} = J^{-1} \begin{bmatrix} I & 0\\ 0 & M \end{bmatrix}$$
(5)

In equation (5), *I* is a  $3 \times 3$  identity matrix and matrix *M* is given by,

$$M = R \times \begin{bmatrix} 1 & 0 & -s_{\beta} \\ 0 & c_{\alpha} & c_{\beta} . s_{\alpha} \\ 0 & -s_{\alpha} & c_{\beta} . c_{\alpha} \end{bmatrix}$$
(6)

Here,  $c_{\bullet} = cos(\bullet)$  and  $s_{\bullet} = sin(\bullet)$ . *R* in equation (6) is the rotation matrix given by,

$$R = \begin{bmatrix} c_{\gamma}.c_{\beta} & -s_{\gamma}.c_{\alpha} + c_{\gamma}.s_{\beta}.s_{\alpha} & s_{\gamma}.s_{\alpha} + c_{\gamma}.s_{\beta}.c_{\alpha} \\ s_{\gamma}.c_{\beta} & c_{\gamma}.c_{\alpha} + s_{\gamma}.s_{\beta}.s_{\alpha} & -c_{\gamma}.s_{\alpha} + s_{\gamma}.s_{\beta}.c_{\alpha} \\ -s_{\beta} & c_{\beta}.s_{\alpha} & c_{\beta}.c_{\alpha} \end{bmatrix}$$
(7)

The operation shown in equation (5) is necessary because  $\tilde{J}$  relates the derivatives of components of X to the derivatives of components of q. Hence,  $J^{-1}$  from equation (3) needs to be altered to take into account the relationship between angular velocities and angle derivatives as per X-Y-Z extrinsic Euler convention as shown in [27]. Now, the Cartesian stiffness matrix  $K_C$  of this parallel system can be expressed as:

$$K_C = J^{-T} K \widetilde{J}^{-1} \tag{8}$$

Matrix K is diagonal matrix with leg stiffnesses forming its diagonal elements as shown in equation (9).

$$K = \begin{bmatrix} k_1 & 0 & 0 & 0 & 0 & 0 \\ 0 & k_2 & 0 & 0 & 0 & 0 \\ 0 & 0 & k_3 & 0 & 0 & 0 \\ 0 & 0 & 0 & k_4 & 0 & 0 \\ 0 & 0 & 0 & 0 & k_5 & 0 \\ 0 & 0 & 0 & 0 & 0 & k_6 \end{bmatrix}$$
(9)

The matrix  $K_C$  relates the  $6 \times 1$  wrench vector  $\Delta F$  applied at the platform and the  $6 \times 1$  platform displacement vector  $\Delta X$  as,

$$\Delta F = K_C \Delta X \tag{10}$$

Equation (10) can be reorganized to get the following equation:

$$Ax = \Delta X \tag{11}$$

Here, *x* is a  $6 \times 1$  vector that contains joint compliances  $1/k_i$ .

$$x = [1/k_1, 1/k_2, 1/k_3, 1/k_4, 1/k_5, 1/k_6]^T$$
(12)

The  $6 \times 6$  matrix A is a function of  $\Delta F$ ,  $\tilde{J}$  and J. The elements of A are given by,

$$A_{ij} = \widetilde{J}_{ij} \left( \sum_{k=1}^{6} J_{kj} \Delta F_k \right)$$
(13)

Here, i, j and k are the subscripts for the row and column of respective matrices.  $\Delta F_{k=1 \text{ to } 6}$  refer to the components of wrench applied to the platform.

# STIFFNESS IDENTIFICATION METHOD Method description

When we substitute the values of estimated lumped stiffnesses ( $k_{i=1 \text{ to } 6}$ ), wrench  $\Delta F$ , hexapod jacobian *J* and  $\tilde{J}$  in equation (11), we get the expected platform deflection  $\Delta X$ . The same equation can be used to identify lumped stiffness values when we know everything in equation (11) except *x*. This implies loading the platform, measuring the platform deflection and solving equation (11). If we increase the number of poses to be measured, we get a  $6n \times 1 x$  vector and a  $6n \times 6 A$  matrix, where *n* is the number of poses.

Fig. 3 shows the procedure for stiffness identification. It consists of the following steps:

- 1. Define experimental constraints that restrict the identification process. For example: workspace limits, loading limitations, etc.
- 2. Choose the number of poses and the number of measurements per pose. More number of measurements per pose help to reduce the uncertainty of identified parameters at a given pose due to measurement errors.
- 3. Identify the best set of poses (considering workspace constraints) and corresponding external forces (considering loading constraints) to identify the stiffness parameters. The next sub-section elaborates on the need and method for choosing the optimal set of poses and forces.

- 4. Load the platform suitably at identified best poses and measure the 6-DOF platform deflection.
- 5. Solve equation (11) to obtain the vector x. This can be accomplished by minimizing the Euclidean norm of  $(Ax \Delta X)$ . The MATLAB function *lsqminnorm* can be used for this purpose.

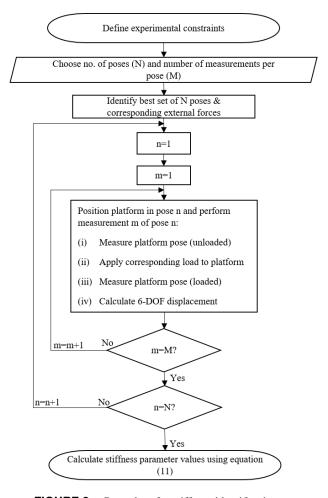


FIGURE 3. Procedure for stiffness identification

#### Algorithm to identify optimal set of poses and forces

The idea behind having measurements at more number of poses is the same as that for robot geometric calibration. In geometric calibration, measurements of large number of poses spread across the robot's workspace are used. These measurements are then used to get the best geometric parameter set that minimizes the error between commanded and attained robot pose throughout the workspace. Large number of poses are needed for this to reduce the influence of measurement errors. However, practicality concerns demand reduction of these measurements. It has been shown that choosing the right set of measurement poses improves identification accuracy [28].

In our application, matrix A decides the quality of identification of parameters. This is because it relates the measurements done to the identified parameters. Since A is a function of X and  $\Delta F$ , the best parameters can be identified by judicious choice of robot poses X and platform forces  $\Delta F$ . Hence, the parameters need to be optimized for all reachable poses and allowable forces.

Parameter identification observability is a well-researched topic in robot calibration literature. However, stiffness parameter identification observability is a relatively new area of research. Klimchik et al. [18] have presented a method to optimize the pose set for stiffness parameter identification to increase the accuracy of displacement prediction at just one particular pose. This is not suitable for us since we seek to increase the accuracy of displacement prediction throughout the workspace of the hexapod.

Many observability indices can be found in the calibration literature:  $O_1$  [28],  $O_2$  [29],  $O_3$  [30],  $O_4$  [31] and  $O_5$  [32]. The reliability and performance of these observability indices have been studied very well too [32–37]. There is no absolute consensus on the best observability index and also their performances have been noted to depend on the type of robot being calibrated [36]. Joubair et al. [34] have also shown that all these observability indices yield good result when the measurement noise is low.

The observability indices also depend on the properties of robot models. Matrix scaling needs to be employed when parameters for different variables with different units need to be identified in calibration. Also, different scaling approaches produce different results. Only  $O_1$  is suitable for optimization of parameter estimation with unscaled robot models. The mathematical proof for this can be found in [32]. If this stiffness identification technique is to be later included in a complete calibration technique that also includes geometric parameters, the issue of matrix scaling can be more complex. Therefore, to avoid complications related to matrix scaling, it is advantageous to select  $O_1$  as the observability index.

For stiffness parameter identification using equation (11),  $O_1$  is obtained by singular value decomposition of matrix A. It can be calculated as shown in [28],

$$O_1 = \frac{(\sigma_1 \sigma_2 \sigma_3 \dots \sigma_p)^{1/p}}{\sqrt{N}} \tag{14}$$

Here, *p* is the number of parameters to be identified which in our case is 6 and *N* is the number of poses.  $\sigma_{1 \text{ to } p}$  are the singular values of matrix *A*. The best set of poses and corresponding external forces maximize the index  $O_1$ . An optimization routine then needs to be used to find the best set of poses and external

forces that maximize  $O_1$ .

### **MEASUREMENT TECHNIQUE**

This section elaborates on the measurement technique used to measure the hexapod pose parameters in the experiments presented in this paper. The measurement setup is as shown in Fig. 4. The measurement system consists of a LK-METRIS coordinate measuring machine (CMM) with RENISHAW SP25M scanning probe. Precision balls are glued to the hexapod for measurement of poses using the CMM. The procedure can be divided into two steps: (i) identifying the coordinate frame fixed to the mobile platform, and (ii) measurement of poses using precision balls. The interest here is to measure the 6-DOF parameters, *X*, of the platform pose by measuring the positions of precision balls in space.

The identification of reference frame fixed to the mobile platform is done in the manner similar to the one described in [38]. The reference holes on the hexapod are used for this purpose. These reference holes are machined into the platform of the hexapod in the production phase. This is followed by measuring the positions of three precision balls with reference to the mobile frame. These three balls' positions can then be used as three points in space to identify the mobile frame.

The measurement technique can be demonstrated using Fig. 5. Let Pose 1 be the pose in which the precision balls are identified in the mobile frame. Let  $P_{A1}$ ,  $P_{B1}$  and  $P_{C1}$  be the positions of the three precision balls associated with this pose and expressed in the mobile frame of reference. The pose vector  $X_1$  associated with this pose is  $[0,0,0,0,0,0]^T$ . Let's call this measurement M1. Now, when the platform moves to Pose 2, we carry out measurement M2 of the precision balls' positions:  $P_{A2}$ ,  $P_{B2}$  and  $P_{C2}$ . The precision balls' positions of measurement M2 are also expressed in mobile frame. A least square fitting algorithm is then used to superimpose the relative positions of the balls measured in M1on to the ball positions measured in M2. The mobile frame associated with Pose 2 can then be easily identified since the pose of mobile frame relative to precision balls' positions ( $P_{A1}$ ,  $P_{B1}$  and  $P_{C1}$ ) of Pose 1 is known. We now obtain Pose 2 (X<sub>2</sub>) with respect to Pose 1. Subsequent poses can then be measured with respect to Pose 1. All this is done automatically using dedicated software tools developed by Symétrie. The 6-DOF deflection between the poses can be obtained by subtracting  $X_1$  from  $X_2$ .

# METHOD IMPLEMENTATION AND EXPERIMENTAL VALIDATION

This section elaborates on how the stiffness identification method was used to evaluate stiffness of a hexapod positioner from Symétrie. The product details cannot be disclosed due to confidentiality. The test setup is as shown in Fig. 6.

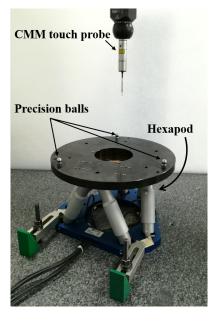


FIGURE 4. Measurement setup

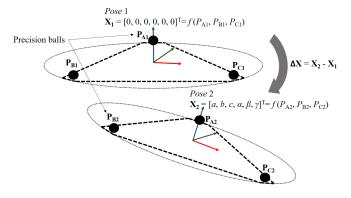


FIGURE 5. Pose measurement using precision balls

The experimental constraints included: (i) the allowable workspace of the robot, and (ii) the loading constraint. Concerning the workspace constraints for identification, the platform could not be rotated around its X and Y axes in addition to allowable workspace of the hexapod. This constraint was incorporated to make sure that the mounted mass doesn't slide off the platform. The loading constraint in this case was that the platform could only be loaded by placing mass on it. The load was, therefore, along the Z axis of the platform.

Now, the number of poses and number of measurements per pose had to be chosen. For any system of equations, the number of equations should at least be equal to the number of unknowns. In this case, we have 6 unknowns and each pose measurement gives us six equations (6-DOF measurement). Hence, just one measurement is analytically enough to identify all parameters.

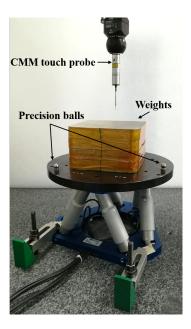
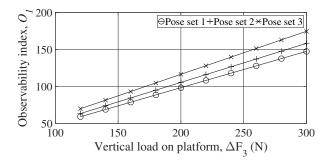


FIGURE 6. Experimental setup

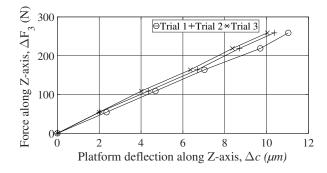
For efficiency, it is also desirable to keep the number of poses for identification less. Hence, compromising between efficiency and accuracy, 3 poses and 3 measurements per pose were chosen. This choice would lead to a total of 9 measurements for identification. Given the nature of measurements in our case, this result in 54 equations for 6 unknowns.

These choices and constraints lead to 13 design variables for the optimization: twelve pose variables and one force variable. Concerning the load variable,  $\Delta F_3$ , further analysis revealed that the index  $O_1$  increased with increased magnitude of  $\Delta F_3$  for any random set of poses (Fig. 7). It was also observed that the hexapod has linear stiffness characteristics. Fig. 8 shows the results of an experiment that shows the hexapod's linear stiffness behavior. In this case, the hexapod platform was loaded only along its Z-axis at pose  $[0,0,0,0,0,0]^T$ . Due to these reasons, a choice was made to use just one load vector of maximum possible magnitude to induce maximum platform deflection. This was 34.5 kg in the given case. For displacement calculation, the reference condition was a pre-load of 12.2 kg. This was necessary to suppress the play in actuators. Hence, the effective load for which the platform deflection was measured was 22.3 kg along the Z-axis of the platform. It must be noted that in cases where stiffness is considerably dependent on applied force, more loading conditions will have to be considered for this optimization.

Since the force variable was already chosen, the number of design variables reduced to 12. MATLAB's *fmincon* function was then used to find the best set of 3 configurations that gave the highest  $O_1$ . These results are very sensitive to the starting point supplied to the optimization algorithm. To tackle this, par-



**FIGURE 7**. Trend of  $O_1$  with respect to  $\Delta F_3$ 



**FIGURE 8**. Plot showing linear stiffness behavior of hexapod under study

allel computing was used along with large number of starting points to find the best solution. The best pose set for identification generated by this optimization is shown in table 1. The hexapod was loaded in these poses and the platform deflections were measured. Equation (11) was then solved to get the stiffness parameters which are tabulated in table 2.

TABLE 1. Best pose set for stiffness identification

	a(mm)	b(mm)	c(mm)	$\alpha(deg)$	$\beta(deg)$	γ(deg)
Pose1	2.616	0.832	-0.666	0	0	15.756
Pose2	10.017	-3.434	-0.764	0	0	-11.007
Pose3	-7.784	-3.210	1.760	0	0	-11.266

The efficiency of this stiffness identification method was then validated using platform deflection measurements done at different poses along the X and Y axes of the hexapod. The detailed description of poses is tabulated in table 3. The load used was same as that for stiffness identification. Figures 9 and 10 show the comparison between the predicted and measured 6-DOF deflections of the platform at different poses along the X

 TABLE 2.
 Identified stiffness parameters

Stiffness parameter value $(N/\mu m)$					
$k_1$	<i>k</i> <sub>2</sub>	<i>k</i> <sub>3</sub>	$k_4$	<i>k</i> 5	<i>k</i> <sub>6</sub>
4.1981	3.7898	2.7503	4.3200	3.6716	3.7288

and Y axes of the hexapod, respectively. Table 4 shows the RMS values of error in prediction of the 6-DOF deflections.

TABLE 3. Pose set for experimental validation

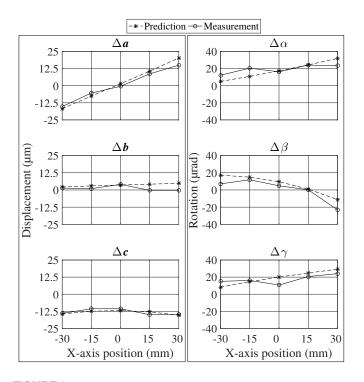
Pose along	Pose parameters						
1 Ose along	a(mm)	b(mm)	c(mm)	$\alpha(deg)$	$\beta(deg)$	$\gamma(deg)$	
	-30						
	-15						
X-axis	0	0	0	0	0	0	
	15						
	30						
		-30					
		-15					
Y-axis	0	0	0	0	0	0	
		15					
		30					

TABLE 4.	Error in deflection	prediction	(RMS	values)
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$\epsilon_{\Delta a}$	2.7 μm
$\epsilon_{\Delta b}$	3.1 μm
$\epsilon_{\Delta c}$	2.2 μm
$\epsilon_{\Delta lpha}$	6.4 μ <i>rad</i>
$\epsilon_{\Delta\beta}$	8.3 µ <i>rad</i>
$\epsilon_{\Delta\gamma}$	8.8 µ <i>rad</i>

# **DISCUSSION AND FUTURE WORK**

Validation results show that the predicted and measured deflections of the platform are very close. As seen from table 4, the RMS values of prediction error is under 3.1  $\mu m$  for translational deflections and 8.8  $\mu rad$  for rotational deflections. This method



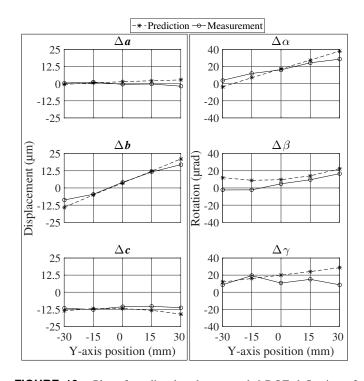
**FIGURE 9**. Plot of predicted and measured 6-DOF deflection of hexapod under load at poses along X-axis

will also reduce the positioning inaccuracy due to compliance of the hexapod to the same level. This is due to the fact that the model prediction accuracy is the only source of error in position compensation technique. This implies that the positioning error due to compliance deflection will reduce, for example, along Zaxis translational coordinate of the hexapod from 13.5  $\mu m$  (RMS value of measured  $\Delta c$ ) to 2.2  $\mu m$ .

The small difference that exists between the prediction and the measured values could be attributed to the following:

- 1. Uncertainty of measurement of the measurement system: All measurements have an uncertainty bound and this limits both stiffness parameter identification and the experimental validation.
- 2. Geometric parameter error: The geometric parameters of the hexapod used for analysis was not calibrated. This is a source of error in parameter identification and experimental validation. This is because of the use of an erroneous geometric model to identify stiffness parameters from the measurements and for predicting the displacements.
- Incomplete stiffness model: The simple stiffness model used cannot capture the complete compliance behavior of the hexapod. Consequently, there will be some residual error due to unmodelled compliance.

For future works, the method presented in this paper will, firstly, be validated for other loading conditions such as loading



**FIGURE 10**. Plot of predicted and measured 6-DOF deflection of hexapod under load at poses along Y-axis

along X and Y axes of the platform. It will also be validated for other hexapod mounting conditions in which some or all legs are subjected to tensile loading. These studies are of interest to validate the efficacy of the presented method to improve accuracy of positioning in applications such as the one shown in [39]. This method will then be extended to include geometric and thermal parameter models to further improve the positioning accuracy of the hexapod.

### CONCLUSION

This paper proposed a new efficient method to experimentally evaluate the stiffness of hexapods. This can be used to accurately predict and correct its positioning error due to structural compliance. The method employed a simple lumped stiffness parameter model which uses one spring per leg to mimic its compliance behavior. A stiffness identification procedure was outlined to estimate the parameters of the proposed stiffness model. This method required small number of measurements. An optimization based on an observability index was used to generate the best set of poses and external forces for stiffness parameter identification. The efficiency of the presented method was evaluated using an experimental study performed on a hexapod based micro-positioner. This was done by comparing the predicted 6-DOF deflections of the platform with the measured deflections. Results showed that the chosen model with identified parameters could predict and consequently allow to correct the platform deflections of the tested hexapod very well. The RMS values of prediction error in the validation experiments is under 3.1  $\mu m$  for translational deflections and 8.8  $\mu rad$  for rotational deflections. The method is shown to reduce the RMS of positioning error due to compliance deflection, for example, along the Z-axis translational coordinate of the hexapod from 13.5  $\mu m$  to 2.2  $\mu m$ .

## ACKNOWLEDGMENT

This work has been supported by Agence nationale de la recherche (ANR), France (Projet ANR-15-LCV3-0005). The authors hereby express their gratitude for the support. The authors would like to thank Marielle Baud from Symétrie for the help provided in using CMM. The authors would also like to acknowledge the valuable inputs from Olivier Lapierre, Gilles Diolez, Pierre Noire and Tristan Caritey of Symétrie.

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