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Bojan Mavkov, Emmanuel Witrant, Christophe Prieur, E Maljaars, F Felici, et al.. Experimental validation of a Lyapunov-based controller for the plasma safety factor and plasma pressure in the TCV tokamak. Nuclear Fusion, 2018, 58 (5), 10.1088/1741-4326/aab16a . hal-01815814

HAL Id: hal-01815814

<https://hal.science/hal-01815814>

Submitted on 14 Jun 2018

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Experimental validation of a Lyapunov-based controller for the plasma safety factor and plasma pressure in the TCV tokamak

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Abstract. In this paper model-based closed-loop algorithms are derived for distributed control of the inverse of the safety factor profile and the plasma pressure parameter β of the TCV tokamak. The simultaneous control of the two plasma quantities is performed by combining two different control methods. The control design of the plasma safety factor is based on an infinite-dimensional setting using Lyapunov analysis for partial differential equations, while the control of the plasma pressure parameter is designed using control techniques for single-input and single-output systems. The performance and robustness of the proposed controller is analyzed in simulations using the fast plasma transport simulator RAPTOR. The control is then implemented and tested in experiments in TCV L-mode discharges using the RAPTOR model predicted estimates for the q -profile. The distributed control in TCV is performed using one co-current and one counter-current Electron Cyclotron Heating actuation.

Keywords: TCV, safety factor, plasma pressure, control, RAPTOR

Submitted to: *Nucl. Fusion*

1. Introduction

One of the main challenges in tokamak plasma control is to achieve the advanced tokamak steady-state scenarios [1]. These scenarios are characterized by a high fusion gain, good plasma confinement and MHD stability. Real-time simultaneous control of several radially distributed magnetic and kinetic plasma parameters is one of the main challenges in the control of hybrid and advanced tokamak scenarios [2]. In this work we are focusing on developing algorithms for simultaneous closed-loop control of the safety factor profile q (and its inverse ι) and the plasma pressure or stored energy. Using feedback control in the real-time control system of a tokamak provides

a better disturbance rejection and less dependency on model uncertainties than using open-loop control only. Feedback control thus improves the performance of reference profile tracking and the robustness of the plasma with respect to MHD instabilities.

Feedback control algorithms for the safety factor can be developed using finite or infinite dimensional feedback control theory. The finite-dimensional control model is obtained by spatial discretization of an infinite-dimensional Partial Differential Equation describing the plasma state. This approach is called an early lumping approach, in which the PDEs are approximated (lumped) first as a set of Ordinary Differential Equations (ODEs) and the control is designed based on this set. Control algorithms for the safety factor profile based on a Multiple Inputs Multiple Outputs (MIMO) approach using different finite dimensional models are developed in previous works [3–9]. There are several works already dedicated to the simultaneous control of the safety factor profile and the pressure parameters. In [10] successful closed-loop experiments are performed for the simultaneous control of the relative internal poloidal flux profile Ψ_r together with the normalized pressure parameter, β_N . In that work the control-oriented model is obtained from experimental data using a generic two time-scales method. Simultaneous control of the q -profile and β_N using first-principles-driven physics-based model is developed for DIII-D H-mode scenarios in [11] and for ITER H-mode scenarios in [12, 13]. The control models used in these works are spatially discretized.

An infinite-dimensional control system is a dynamical system whose state lies in an infinite dimensional vector space, typically the state space of a PDE. In this control approach, the distributed-parameter nature of the system is kept as long as possible in the control design (late lumping approach). This control approach allows us to design feedback control strategies for the full profile throughout the plasma radius.

Control algorithms for the safety factor profile based on infinite dimensional control theory using Lyapunov techniques are already used in several works. Lyapunov functions are energy-like functions that can be used to evaluate the stability of a dynamical system and to design stabilizing controllers. In [14], a strict Lyapunov control function for the diffusion equation of the poloidal magnetic flux gradient is computed. In [15], a Lyapunov-based control strategy using sum-of-squares polynomials is used to maximize the bootstrap current considering a resistivity bounded between two extremal profiles. In [16] a proportional integral (PI) controller is developed for $\mathcal{H}\infty$ stabilization of the spatial distribution of the internal poloidal flux profile Ψ_r . All these algorithms were tested using only plasma simulators and they were based on the configuration of the Tore Supra tokamak. Tore Supra has circular cross-section and the magnetic flux equation is simplified using cylindrical coordinates. More results using distributed and boundary infinite dimensional control based on IDA-PBC (Interconnexion Damping Assignment- Passive Based Control) design are presented in [17].

The control model is a coupled system of a 1D resistive diffusion equation (ι control) and a nonlinear ordinary differential equation (β control). Enabled by the different time scales of the kinetic and magnetic parameters, the control in this work is synthesized by designing separate control algorithms for each controlled variable. The safety factor profile is controlled using the poloidal magnetic flux radial gradient z as a state variable. This parameter is closely related to the safety factor. Our q -profile controller is based on an infinite dimensional setting using Lyapunov techniques. We extend the control computation and the stability analysis presented in [14, 18] to a more general case without using the cylindrical approximation. The control Lyapunov

function presented in these works is analogous with the proportional action presented here. Moreover, for successful tracking of the desired reference profiles in the real TCV experiments, an integral action is added in the control design. We have also constrained the shape of the weights in the Lyapunov function to highlight the different spatial locations at which we want to emphasize our control action. The control of the plasma pressure parameter is designed separately using control techniques for Single-Input and Single-Output (SISO) systems.

The control strategy is applied to a controller test environment on TCV L-mode discharges, described in details in [19]. The purpose of this environment is to facilitate offline development and testing of controllers as well as experimental implementation and evaluation on TCV. The fast plasma transport simulator RAPTOR [20, 21] and various algorithms reconstruct the plasma equilibrium and plasma profiles by merging the available measurements with model-based predictions. It estimates the plasma states at many locations, that are provided to the controllers. RAPTOR can also be used as a nonlinear plasma simulator for closed loop control when it is run off-line outside the TCV control system [22]. This enables to test the control algorithms performance and robustness before they are implemented on real TCV experiments.

The paper is organized as follows. In Section 2 the control models and control problems are defined. In Section 3 the control algorithms are presented. The results from the RAPTOR simulations are presented in Section 4 and the results from the TCV experiments are presented in Section 5. Some concluding remarks are given in Section 6.

2. Control model and experimental settings

2.1. Safety factor control model

One of the key parameters to analyze the plasma stability and performance is the safety factor q . Another parameter that can be used for the control is the inverse of the safety factor $\iota = 1/q$, which can be considered as a more natural control variable since it is proportional to the spatial derivative of the poloidal magnetic flux, whose evolution is governed by a parabolic equation. The corresponding model is an approximate 1D diffusion equation as in [23] assuming constant toroidal magnetic flux at the plasma boundary:

$$\frac{\partial \psi}{\partial t} = \frac{\eta_{\parallel} F^2}{16\pi^2 \mu_0 \Phi_b^2 \rho} \frac{\partial}{\partial \rho} \left(\frac{g_2 g_3}{\rho} \frac{\partial \psi}{\partial \rho} \right) - \frac{\eta_{\parallel} B_0}{2\Phi_b \rho} V' \left(j_{aux}(u) + j_{bs} \right) \quad (1)$$

where ρ is the normalized square-root toroidal flux, defined as: $\rho = \sqrt{\frac{\Phi}{\Phi_b}}$ (Φ being the toroidal magnetic flux and Φ_b its value at the last closed magnetic surface), B_0 is the magnetic field at the major plasma radius $R = R_0$, μ_0 is the permeability of vacuum, V is the plasma volume and $V' = \frac{\partial V}{\partial \rho}$ is the spatial derivative of the plasma volume, and η_{\parallel} is the parallel electrical neoclassical resistivity of the plasma. The non-inductive current density $j_{ni} = j_{aux} + j_{bs}$ is obtained by combining the auxiliary sources of current density $j_{aux}(u)$ (heating and current drive radio-frequency (RF) systems) and the bootstrap current density j_{bs} . The vector u contains the actuator inputs represented by the power delivered by the Electron-Cyclotron (EC) clusters $[P_{EC,1}(t); P_{EC,2}(t)]$ used in the RAPTOR simulations and in the TCV experiments.

The coefficients F, g_2, g_3 and V' are defined as [24, 25]:

$$F = RB_\phi; \quad g_2 = \left(\frac{\partial V}{\partial \psi} \right) \left\langle \left| \frac{\nabla \psi}{R^2} \right| \right\rangle; \quad g_3 = \left\langle \frac{1}{R^2} \right\rangle; \quad V' = \frac{\partial V}{\partial \rho} \quad (2)$$

where $\langle \cdot \rangle$ denotes a flux surface average, R is the distance from the vertical axis and B_ϕ is the toroidal magnetic field.

In this configuration, the Electron Cyclotron Current Drive System (ECCD) is represented by co-current (with respect to the ohmic plasma current) and counter-current deposition. The gyrotrons in TCV are grouped in clusters. The power delivered by the first cluster (with input power $P_{EC,1}(t)$) consists of one gyrotron with counter-current drive. The second cluster (with input power $P_{EC,2}(t)$) consists of two gyrotrons that are used for co-current drive. A set of coils generating a poloidal magnetic field are used for feedback control of the plasma vertical and horizontal position in the poloidal plane. The position of the magnetic field is chosen such that the EC power deposition is absorbed on the magnetic axis. The total auxiliary current density model is computed as the sum of the current densities induced by each ECCD antenna and the bootstrap current density: $j_{ni,mod} = \sum_1^{n_{aux}} j_{aux,i} + j_{bs}$, where $n_{aux} = 2$ denotes the number of EC antennas used in our case. Each individual auxiliary current drive is modeled by the scaling law presented in [22]:

$$j_{aux,i}(\rho, t) = \frac{T_e}{n_e} j_{dis,i}(\rho) P_{EC,i}(t) \quad (3)$$

which is the product of the weighted Gaussian distributions $j_{dis,i}(\rho)$ (given in [20]) with the input powers $P_{EC,i}(t)$ and the current-drive efficiency $\left(\frac{T_e}{n_e}\right)$. The nominal current deposition width $\omega_{dep} = 0.35$ and the nominal location of the peak of the deposition $\rho_{dep} = 0$ are used for both clusters. The reference values of $j_{dis}(\rho)$ and the relative power density profile $Q_{dis}(\rho)$ for the two EC clusters are shown in Fig. 1.

The bootstrap current profile j_{bs} and the neoclassical resistivity $\eta_{||}$ models are the one

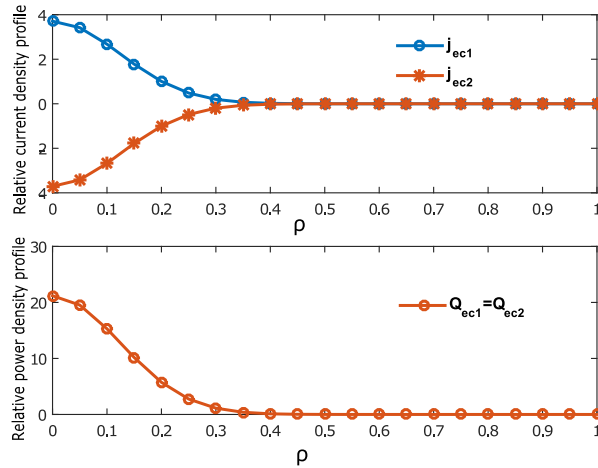


Figure 1: The relative current density profile $j_{dis,ec}$ ($10^{20} \frac{A}{m^5 \text{ keV W}}$) and relative power density profile $Q_{dis,ec}$ (m^{-3}) for the TCV configuration.

used in the RAPTOR code and are based on the Sauter-Angioni equations [26, 27].

The model of the bootstrap current density can be used in a simplified version under the assumption of a tight coupling between the electron and ion species. With this assumption the electron and the ion densities are considered to be equal $n_e = n_i$ and $\frac{\partial \ln(T_i)}{\partial \Phi} = \frac{\partial \ln(T_e)}{\partial \Phi}$, where T_i is the ion temperature. The approximated model is derived as :

$$j_{bs} = \frac{k_{bs}}{\partial \Psi / \partial \rho} \left[\mathcal{L}_{31} \frac{\partial n_e}{\partial \rho} T_e + \left(\mathcal{L}_{31} + R_{pe} \mathcal{L}_{32} + (1 - R_{pe}) \mathcal{L}_{34} \right) \frac{\partial T_e}{\partial \rho} n_e \right] \quad (4)$$

where $k_{bs}, \mathcal{L}_{31}, \mathcal{L}_{32}, \mathcal{L}_{34}$ depend on magnetic configuration of the plasma equilibrium and $R_{pe} = p_e/p$ is the ratio between electrons and total pressures.

The neoclassical electrical resistivity model in RAPTOR is calculated as:

$$\eta_{\parallel} = 1/\sigma_{\parallel} = 1/(c_{neo}(\rho)\sigma_{Spitzer}) \quad (5)$$

where σ_{\parallel} is neoclassical electrical conductivity and $\sigma_{Spitzer} \propto T_e^{3/2}$ e is the Spitzer conductivity. The parameter c_{neo} is the neoclassical correction, which depends on geometric effects as well as collisionality. These two plasma parameters are time-varying and highly dependent on the electron and ion temperatures and densities.

The controlled state is selected to be the magnetic flux gradient $z = \frac{\partial \Psi}{\partial \rho}$. The variables of main interest for the control (q , ι and current profiles) can be easily related to the magnetic flux gradient $\iota = \frac{1}{q} = \frac{1}{2\Phi_b \rho} z$. Differentiating (1) in space and considering the previous change of variables, as in [28], yields:

$$\frac{\partial z}{\partial t} = \frac{\partial}{\partial \rho} \left(\frac{a_1(\rho)}{\rho} \frac{\partial}{\partial \rho} (a_2(\rho)z) \right) - \frac{\partial}{\partial \rho} \left(a_3(\rho) (j_{aux}(u) + j_{bs}) \right) \quad (6)$$

with boundary conditions:

$$\begin{aligned} z(0, t) &= 0, \quad \forall t \geq 0 \\ z(1, t) &= a_4(1)I_p(t), \quad \forall t \geq 0 \end{aligned} \quad (7)$$

and initial condition:

$$z(\rho, 0) = z_0(\rho), \quad \forall \rho \in [0, 1] \quad (8)$$

where $a_1(\rho) = \frac{\eta_{\parallel} F^2}{16\pi^2 \mu_0 \Phi_b^2}$, $a_2(\rho) = \frac{g_2 g_3}{\rho}$, $a_3(\rho) = \frac{\eta_{\parallel} B_0}{2\Phi_b \rho} V'$, $a_4(\rho) = \frac{\rho}{g_2 g_3} \frac{16\pi^3 \mu_0 \Phi_b}{F}$ and I_p is the total plasma current.

For control purposes we linearize the model around a given equilibrium profile \bar{z} . This equilibrium is calculated by taking the values of the plasma parameters when the inputs (\bar{u}, \bar{I}_p) are applied sufficiently long for the system to reach a steady state:

$$0 = \frac{\partial}{\partial \rho} \left(\frac{a_1(\rho)}{\rho} \frac{\partial}{\partial \rho} (a_2(\rho)\bar{z}) \right) - \frac{\partial}{\partial \rho} \left(a_3(\rho) (\bar{j}_{aux}(\bar{u}) + \bar{j}_{bs}) \right) \quad (9)$$

with boundary conditions: $\bar{z}(0, t) = 0$, $\bar{z}(1, t) = a_4(1)\bar{I}_p(t)$. The linearized model is derived around the steady state by substituting, in the reference model, $z = \bar{z} + \tilde{z}$, $u = \bar{u} + \tilde{u}$ and $I_p = \bar{I}_p + \tilde{I}_p$ using Taylor series with first order approximation. Here $(\bar{z}, \bar{u}, \bar{I}_p)$ denotes the equilibrium of the system and $(\tilde{z}, \tilde{u}, \tilde{I}_p)$ denotes the variations around this equilibrium. The simplified linearized model obtained from (6) is derived as:

$$\frac{\partial \tilde{z}}{\partial t} = \frac{\partial}{\partial \rho} \left(\frac{a_1(\rho)}{\rho} \frac{\partial}{\partial \rho} (a_2(\rho)\tilde{z}) \right) - \frac{\partial}{\partial \rho} \left(a_3(\rho) (\tilde{j}_{aux}(\tilde{u}) + \tilde{j}_{bs}) \right) \quad (10)$$

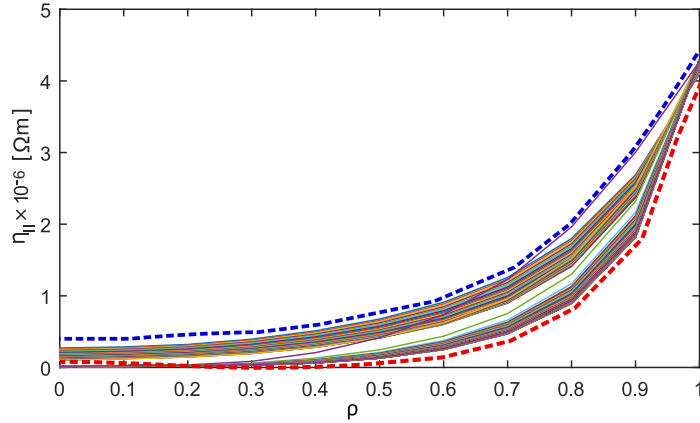


Figure 2: η_{\parallel} parameter ranges with its minimum (red dashed line) and its maximum value (blue dashed line).

with boundary conditions:

$$\begin{aligned}\tilde{z}(0, t) &= 0, \forall t \geq 0 \\ \tilde{z}(1, t) &= a_4(1)\tilde{I}_p(t), \forall t \geq 0\end{aligned}\quad (11)$$

and initial condition:

$$\tilde{z}(\rho, t_0) = \tilde{z}_0, \forall \rho \in [0, 1] \quad (12)$$

For the control application, the perturbed bootstrap current $\tilde{j}_{bs}(\rho, t) = j_{bs} - \bar{j}_{bs}$ can be considered as an external disturbance to be attenuated/compensated (since $I_{bs}/I_p \leq 10\%$ in L-mode plasmas). When the perturbed bootstrap current is relatively large and cannot be neglected, it should be included in the stability analysis. Infinite dimensional feedback control strategies, where the bootstrap current is included in the control design and stability analysis can be found in [15, 29]. The coefficient $\bar{\eta}_{\parallel}$ can be considered as an uncertain parameter that varies in a given operational range:

$$\bar{\eta}_{\parallel}(\rho, t) = \omega_{\eta}(t)\eta_{\parallel, \min}(\rho) + (1 - \omega_{\eta}(t))\eta_{\parallel, \max}(\rho), \text{ for all } t \geq 0 \quad (13)$$

where $\omega_{\eta}(t) \in [0, 1]$ and $\eta_{\parallel, \min}$ and $\eta_{\parallel, \max}$ are the minimum and the maximum values of η_{\parallel} , respectively, presented in Fig. 2. The bounds of the resistivity η_{\parallel} are calculated using the model (5), by modulating the ECCD clusters with different power intensities in the RAPTOR simulations. From the plot in Fig. 2 we can see that the variations of η_{\parallel} are not significant in the operation range that is considered in our experiments. We thus assume that the η_{\parallel} profile scales linearly between the two outer bounds ($\eta_{\parallel, \min}$ and $\eta_{\parallel, \max}$). While this uncertainty is not crucial for the control of a L-mode plasma, it may become important in other plasma operation modes in which the plasma resistivity changes more dramatically.

2.2. Plasma pressure and thermal energy control model

The plasma parameter β is defined as the ratio between the volume-averaged pressure and the magnetic pressure. This parameter is of prime interest to analyze MHD stability in tokamaks. There are several definitions of this parameter. Following [30], the most commonly used one is:

$$\beta = \frac{\langle p \rangle_V}{B_0^2/(2\mu_0)} = \frac{2}{3} \frac{W_{th}/V}{B_0^2/(2\mu_0)} \quad (14)$$

where W_{th} is the stored thermal energy in the plasma and $\langle \cdot \rangle_V$ denotes the volume-average operation $1/V \int_V (\cdot) dV$. The energy stored in the plasma is presented by [28]:

$$W_{th} = \frac{3}{2} \int_V (n_e T_e + n_i T_i) dV \quad (15)$$

where T_i is the ions temperature profile and n_i is the ions density profile. Assuming that the plasma volume does not change in time, the approximate energy balance equation is given by:

$$\frac{dW_{th}}{dt} = -\frac{W_{th}}{\tau_{th}} + P_{tot}(u) \quad (16)$$

where τ_{th} is the global energy confinement time which can be modelled using a scaling law [31, 32] such as $\tau_{th} = 0.14 I_p^{0.91} B_0^{-0.13} n_e^{-0.77} P_{tot}^{-0.75}$, and the total power $P_{tot}(u)$ can be derived as a sum of several contributions. The main sources of electrons heating are the auxiliary radio frequency (RF) heating sources. Other sources are the ohmic heating power, electron-ion power loss and radiation power loss. All these terms depend on many plasma parameters (temperature, q , etc.) as well as their radial derivative (e.g. temperature gradients, magnetic shear, etc.).

The simplified linearized model of (16) around the steady state $(\bar{W}_{th}, \bar{u}, \bar{I}_p)$ is derived using Taylor series with first order approximation considering the model of τ_{th} given in [31]:

$$\begin{aligned} \frac{d\tilde{W}_{th}}{dt} &= -\frac{\tilde{W}_{th}}{\bar{\tau}_{th}} + k_p \tilde{P}_{tot}(\tilde{u}) + k_{I_p} \tilde{I}_p + \omega \\ \tilde{\beta} &= \frac{2}{3} \frac{\tilde{W}_{th}/V}{B_0^2/(2\mu_0)} \end{aligned} \quad (17)$$

where $k_{I_p} = -\frac{\partial(\bar{W}_{th}/\bar{\tau}_{th})}{\partial \bar{I}_p} + \frac{\partial \bar{P}_{tot}}{\partial \bar{I}_p}$, $k_p = -\frac{\partial(\bar{W}_{th}/\bar{\tau}_{th})}{\partial \bar{P}_{tot}} + 1$, and ω contains the variations of τ_{th} that depend on plasma parameters other than P_{tot} and I_p (considered as disturbances in the system, i.e. the variation of the ohmic power). The parameter $\bar{\tau}_{th}$ is assumed to be bounded as $\bar{\tau}_{min} \leq \bar{\tau}_{th} \leq \bar{\tau}_{max}$. In our simplified control model, the total heating power is calculated as the sum of the two main sources of heating $\tilde{P}_{tot}(\tilde{u}) = \tilde{P}_{EC,1}(t) + \tilde{P}_{EC,2}(t)$.

Although there is a coupling between the dynamics of these two states (\tilde{z} and \tilde{W}_{th}), the control actions are computed separately. Variations of the other plasma parameters are considered as disturbances that need to be handled by the robustness margin of the controller. The ion and electron temperatures and densities are considered as uncertainties for ι (leading to the bounds on $\eta_{||}$ in Fig. 2) or volume-averaged for β . The profiles distributions and couplings, e.g between q and T_e , can be considered with more complex PDE controllers as proposed in [29] but we focus here on the control design motivated by the experimental setup.

3. Feedback control design

In this section we present the control algorithms that were implemented in the TCV experimental session on feedback controllers testing in 2016/2017, in which several different controllers were implemented [19, 33, 34]. In these TCV experiments only the powers of the two (ECH/ECCD) antennas are used as in-domain control actuators. The control algorithms are designed separately for ι and β , and a composite control of two signals, u_ι for the control of ι and u_β for control of β , is sent to the TCV actuators. For the control of the ι profile a proportional-integral infinite dimensional state feedback control is applied and the control of the β parameter is tuned using classical control methods for ODEs. An anti-windup compensator is added to the control design to avoid the drawbacks from the actuators saturation in the system (discussed in details in Appendix C). The complete control scheme is presented as a block diagram in Fig. 3.

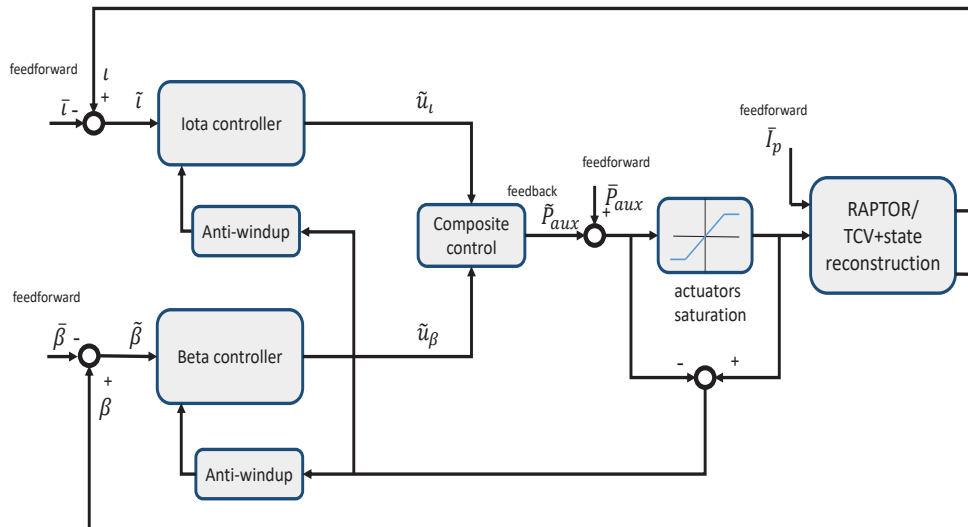


Figure 3: Control scheme

3.1. Distributed control of ι

The feedback control of the ι profile is achieved with an infinite dimensional PI controller. The auxiliary heating powers are set to obtain the desired profile of \tilde{j}_{ni} . The feedback control is applied using the real-time estimate of \tilde{z} , which represents the error between the z profile and the desired reference profile \bar{z} . The feed-forward signal corresponding to this stationary profile is calculated by (9) and is set with \bar{u} and \bar{I}_p . In the simulation results the feedforward term \bar{z} is directly obtained from RAPTOR as steady-state (along with \bar{u} and \bar{I}_p), while in the TCV experiments the feedforward term is calculated by numerically solving \bar{u} and \bar{I}_p from the equality (9) with the given values of the desired target profile \bar{z} .

In Lyapunov-based control, the control law is designed such that closed-loop stability can be guaranteed using Lyapunov stability analysis. Lyapunov functions are energy-like functions and by examining their time derivative we can conclude that the closed-loop system is stable or asymptotically stable [35]. The asymptotic stability of the closed-loop system guarantees that the system converges to the desired equilibrium. The stability analysis of the closed-loop system will be discussed in Section 3.2. To obtain an appropriate closed-loop system description, the control law is written in a specific form. For this purpose, the feedback-control of the z -profile is determined such that the control parameter $\tilde{j}_{ni,fb}$ is calculated as:

$$\tilde{j}_{ni,fb}(\tilde{u}_\iota, \rho, t) = \frac{1}{a_3(\rho)} \left(\int_0^\rho \alpha_p(r) \tilde{z}(r, t) dr - \int_0^\rho \mathcal{I}_\iota(r, t) dr \right) \quad (18)$$

where the proportional action is represented by weighted spatial integral of the error on the z profile (\tilde{z}) and the integral action of the control is calculated as:

$$\frac{\partial \mathcal{I}_\iota(\rho, t)}{\partial t} = -\alpha_I(\rho) \tilde{z}(\rho, t) - \lambda(t) \mathcal{I}_\iota(\rho, t) \quad (19)$$

where \tilde{u}_ι stands for the feedback control signal for the ι parameter, \mathcal{I}_ι is the integral action of the controller, and $\alpha_p(\rho)$ and $\alpha_I(\rho)$ correspond to the proportional and the integral gain of the control, respectively. The implementation of the feedback control is discussed in Section 3.4.

The parameter $\lambda(t) \in [0, \lambda_{\max}]$ (for a given λ_{\max}) in the integral action is called a “forgetting factor” for the integrator [4, 36]. This parameter is used to cancel high overshoots by weighting down past accumulated errors when ι is far from the reference profile. This usually appears when the target profile is changed. This term vanishes in finite time to avoid a steady-state error ($\lambda(t) \rightarrow 0$ when $t \rightarrow \infty$) and is activated at the beginning of the control or at the time instants when there are sudden (significant) changes of the reference profile. The value of this parameter is selected as $\lambda(t) = \kappa e^{-\sigma t^2}$, where κ and σ are chosen constant parameters. The values of these parameters are manually tuned to obtain the best performance of the controller. The resulting closed-loop system (10), when the feedback control (18) applies, has the dynamics:

$$\begin{cases} \frac{\partial \tilde{z}}{\partial t} = \frac{\partial}{\partial \rho} \left(\frac{a_1(\rho)}{\rho} \frac{\partial}{\partial \rho} (a_2(\rho) \tilde{z}) \right) - \alpha_p(\rho) \tilde{z} + \mathcal{I}_\iota \\ \frac{\partial \mathcal{I}}{\partial t} = -\alpha_I(\rho) \tilde{z} - \lambda(t) \mathcal{I}_\iota \end{cases} \quad (20)$$

In this TCV experimental setup I_p is feedback controlled via the inductive voltage and its value is set to the desired reference value $I_p = \bar{I}_p$. This leads to the following boundary conditions:

$$\begin{aligned} \tilde{z}(0, t) &= 0, \quad \forall t \geq 0 \\ \tilde{z}(1, t) &= 0, \quad \forall t \geq 0 \end{aligned} \quad (21)$$

The stability analysis is crucial for the feedback control design and they are used to prove that the closed loop system converges to the desired equilibrium point. This means that the parameters α_p and α_I of the controller should be chosen such that the closed-loop system is asymptotically stable.

3.2. Stability of the closed loop system

The stability analysis for the closed-loop system (20) is performed using Lyapunov analysis for infinite-dimensional systems. More results on Lyapunov theory for infinite dimensional systems can be found in [37, 38]. To compute the stability of the closed-loop system (20), the following candidate Lyapunov function is chosen:

$$\mathcal{V} = \mathcal{V}_z + \mathcal{V}_I \quad (22)$$

where

$$\mathcal{V}_z = \frac{1}{2} \int_0^1 \rho^2 p_z(\rho) \tilde{z}^2 d\rho \quad (23)$$

and

$$\mathcal{V}_I = \frac{1}{2} \int_0^1 \rho^2 p_I(\rho) \mathcal{I}_v^2 d\rho \quad (24)$$

The candidate Lyapunov function is chosen as a weighted $L^2(0, 1)$ norm ‡, where $p_z(\rho) > 0$ and $p_I(\rho) > 0$ for $\rho \in [0, 1]$ are the weighting polynomial functions, and it is multiplied by the term ρ^2 to handle the singularity at $\rho = 0$.

Theorem 1. *Suppose that for a given positive value γ_1 and a time-varying positive number $\gamma_2(t)$, there exist polynomials $p_z, p_I, k_z = \alpha_p p_z, k_I = \alpha_I p_I$ such that $p_z(\rho) > 0, p_I(\rho) > 0, k_z > 0$ and $k_I > 0$ for all $\rho \in [0, 1]$. If the following inequality is verified:*

$$A(\rho) + A_1(\rho, t) \leq 0 \quad (25)$$

where

$$A(\rho) = \begin{bmatrix} c_{1,1}(\rho) & c_{1,2}(\rho) & c_{1,3}(\rho) \\ 0 & c_{2,2}(\rho) & 0 \\ c_{1,3}(\rho) & 0 & c_{3,3}(\rho) \end{bmatrix} \quad (26)$$

$$\begin{aligned} c_{1,1} &= -\left(2p_z(\rho) + \rho p'_z(\rho)\right) a_1(\rho) \frac{\partial(a_2(\rho))}{\partial \rho} - k_z(\rho) \rho^2 \\ c_{2,2} &= -\rho p_z(\rho) a_1(\rho) a_2(\rho) \\ c_{1,2} &= -\frac{1}{2} \left(2p_z(\rho) a_2(\rho) + \rho p'_z(\rho) \frac{\partial a_2(\rho)}{\partial \rho} + \rho p'_z(\rho) a_2(\rho)\right) a_1(\rho) \\ c_{1,3} &= \frac{1}{2} \rho^2 \left(p_z(\rho) - k_I(\rho)\right) \\ c_{3,3} &= -\rho^2 p_I(\rho) \lambda(t) \end{aligned} \quad (27)$$

and

$$A_1(\rho, t) = \frac{1}{2} \begin{bmatrix} \gamma_1 p_z(\rho) \rho^2 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & \gamma_2(t) p_I(\rho) \rho^2 \end{bmatrix} \quad (28)$$

then the time derivative $\dot{\mathcal{V}}$ of \mathcal{V} defined in (22) along the solutions to (20) and (21) verifies:

$$\dot{\mathcal{V}} \leq -\frac{1}{2} \gamma_1 \int_0^1 \rho^2 p_z(\rho) \tilde{z}^2 d\rho - \frac{1}{2} \gamma_2(t) \int_0^1 \rho^2 p_I(\rho) \mathcal{I}_v^2 d\rho \leq -\gamma(t) \mathcal{V} \quad (29)$$

‡ The L^2 norm of ξ on a domain Ω is denoted as $\|\xi\|_{L^2} = \left(\int_{\Omega} \xi^2 d\Omega\right)^{\frac{1}{2}}$

where

$$\gamma(t) = \min_{t \in [0, T]} (\gamma_1, \gamma_2(t)) \quad (30)$$

The proof of this theorem is given in [Appendix A](#).

The numerical computation of the the polynomial functions $p_z(\rho)$, $p_I(\rho)$, $k_z(\rho)$ and $k_I(\rho)$ is presented in [Appendix B](#). Once the solution of the polynomial functions is found, the controller gains are calculated as $\alpha_p = k_z/p_z$, $\alpha_I = k_I/p_I$.

The stability of our coupled system of linearized equations is established with [Theorem 1](#). The parameters γ_1 and $\gamma_2(t)$ determine the convergence rate of our closed-loop system. By changing these parameters we can obtain the desired values of $\alpha_z(\rho)$ and $\alpha_I(\rho)$ to obtain best performance of the controller. The parameter $\gamma_2(t) = \kappa_\gamma e^{-\sigma t^2}$ is selected to be time-varying according to the time-varying parameter $\lambda(t)$. Here κ_γ is a tuning parameter and the term $e^{-\sigma t^2}$ is the same one that is used to define $\lambda(t)$.

Another important property is to estimate how quickly the solutions converge to the desired equilibrium state.

Corollary 1. *Under the conditions of [Theorem 1](#) and using the definition of $\gamma(t)$, the closed-loop system [\(20\)](#) with boundary condition [\(21\)](#) and initial conditions [\(12\)](#) is globally exponentially stable. The convergence rate of the system satisfies:*

$$\mathcal{V}(\tilde{z}, \mathcal{I}_t) \leq e^{-\int_0^t \gamma(r) dr} \mathcal{V}(\tilde{z}_0, \mathcal{I}_{t,0}) \quad (31)$$

where $\gamma(t)$ is given in [\(30\)](#).

Proof. Using separation of variables and integrating the inequality [\(29\)](#) over time give the result. \square

3.3. Control of β

The control synthesis for β is developed using only the auxiliary heating sources as the system input, given by $\tilde{P}_{\text{tot}}(\tilde{u}_\beta)$. The value of the plasma current during the experiments is considered as: $I_p(t) = \bar{I}_p$. The closed-loop control algorithm of β is developed using classical methods for designing Proportional–Integral–Derivative (PID) controllers. There are many works dedicated to optimal tuning of the parameters of PID controllers. Such methods can be found in [\[39, 40\]](#). In this section the simple rule for PID tuning presented in [\[41\]](#) is used. This rule is called Simple Internal Model Control (SIMC). In our application only Proportional (P) and Integral (I) components are applied, whereas the Derivative (D) component is not necessary to be used when the dynamic of the system is represented by first order ODE. In the TCV control system there can be several sources of time delays [\[19\]](#) (e.g. signal routing delays and reconstruction delays). The effective delay in the TCV experiments can be up to $\theta \approx 5$ ms and this is taken into consideration when the feedback control is designed. The transfer function of the derived PI controller is chosen in cascade form as:

$$c(s) = \frac{\tilde{P}_{\text{tot}}(s)}{\tilde{\beta}(s)} = -K_c \frac{\tau_I s + 1}{\tau_I s} \quad (32)$$

or presented in time domain:

$$\tilde{P}_{\text{tot}}(\tilde{u}_\beta(t)) = -K_c \tilde{\beta}(t) - \int_0^1 \frac{K_c}{\tau_I} \tilde{\beta}(\sigma) d\sigma = -K_{\beta,p} \tilde{\beta}(t) - \int_0^t K_{\beta,i} \tilde{\beta}(\sigma) d\sigma \quad (33)$$

where $K_{\beta,p}$ is the proportional gain and $K_{\beta,i}$ is the integral gain of the controller. The coefficients of the controller are computed as follows:

$$\begin{aligned} K_c &= \frac{1}{k_\beta} \frac{\bar{\tau}_{th}}{\tau_c + \theta} = \frac{1}{k'_\beta} \frac{1}{\tau_c + \theta} \\ \tau_I &= \min\left\{\bar{\tau}_{th}, \frac{4}{k'_\beta K_c}\right\} = \min\{\bar{\tau}_{th}, 4(\tau_c + \theta)\} \end{aligned} \quad (34)$$

where $k_\beta = k_p \bar{\tau}_{th} \frac{\beta}{W_{th}} = \frac{2k_p \bar{\tau}_{th}}{3VB_0^2/(2\mu_0)}$ and $k'_\beta = \frac{k_\beta}{\bar{\tau}_{th}}$. The SIMC PI-rule has one tuning parameter τ_c which can be used as a trade off between performance ("tight" control) and robustness ("smooth" control). Moreover, the value of this parameter can be changed according to the control requirements. For a robust performance of the tracking, this parameter should be $\tau_c \geq \theta$. If this parameter is decreased the convergence speed and the disturbance rejection are improved. For better stability, robustness and small input usage the value of this parameter should be increased. The optimal tuning of this parameter and the performance of the PI SIMC method is discussed in [42].

3.4. Control implementation

As stated in the previous section, only the auxiliary powers are used as controlled inputs in the system. This leads to limited shapes of the generated current density profiles due to the limited degrees of freedom of the actuators. An optimization problem should be formulated to fit the desired current density control calculated in (18) with the achievable current density profile. The control is solved such that an optimization problem is formulated to deal with the simultaneous control of several parameters. The control inputs $\tilde{u} = [\tilde{P}_{EC,1}, \tilde{P}_{EC,2}]$ are calculated using an optimization algorithm that solves:

$$\arg \min_{\tilde{u}} \omega_1 f_z(\tilde{u}_\iota) + \omega_2 f_\beta(u_\beta) \quad (35)$$

subject to

$$\begin{aligned} \tilde{P}_{1,\min} &\leq \tilde{P}_{EC,1} \leq \tilde{P}_{1,\max} \\ \tilde{P}_{2,\min} &\leq \tilde{P}_{EC,2} \leq \tilde{P}_{2,\max} \end{aligned}$$

where:

$$\begin{aligned} f_z(\tilde{u}_\iota) &= \int_0^1 \frac{1}{\sigma_j^2} \left(\tilde{j}_{ni,fb} - \tilde{j}_{ni,mod}(\tilde{u}_\iota) \right)^2 d\rho \\ f_\beta(\tilde{u}_\iota) &= \frac{1}{\tilde{P}_{tot,fb}^2} \left(\tilde{P}_{tot,fb} - \tilde{P}_{tot}(\tilde{u}_\beta) \right)^2 \end{aligned} \quad (36)$$

$\tilde{j}_{ni,fb}$ is the desired feedback control calculated in (18) and $\tilde{j}_{ni,mod}(\tilde{u}_\iota)$ is the current profile that can be achieved by the available actuators in this experimental setup. Here $\tilde{P}_{tot,fb}$ is the desired feedback control calculated in (34) and $\tilde{P}_{tot}(\tilde{u}_\beta)$ is total power that can be achieved by the available actuators. The parameters required to calculate the desired profile $\tilde{j}_{ni,mod}$ are provided in real-time by the RAPTOR observer in the TCV experiments. The cost function f_z is normalized by multiplying with the

inverse of the square of the variance in the errors σ_j , that is common technique used in weighted least squares problems, and f_β is normalized by multiplying with the inverse of $\tilde{P}_{tot,fb}$.

The parameters ω_1 and ω_2 are used as scaling parameters in the optimization. They can be used to tune the priority of the control of one of the plasma parameters. The nominal values of these parameter are set as $\omega_1 = \omega_2 = 0.5$ to give equal priority to both plasma parameters. These weighting parameters can be changed according to the different control objectives (e.g. $\omega_1 = 0$, $\omega_2 = 1$ to yield only β control).

In (35), $\tilde{P}_{tot}(\tilde{u}_\beta)$ is the desired control of the β component calculated in (33).

Remark 1. *The global exponential stability of the nominal system for the evolution of z ($\tilde{u} = 0$ when $u = \bar{u}$) is proven in several previous works [14, 18, 29, 43]. In this work, when closing the loop (20), the contribution of the feedback control ($-\alpha_p(\rho)\tilde{z} + \mathcal{I}_\iota$) accelerates the convergence to the desired reference. The limitations of the achievable shapes for the auxiliary current deposit can reduce the desired convergence rate. In the worst case, $\tilde{u} = 0$ is always a feasible state that results to globally exponentially stable system.*

3.5. Performance criterion

The feedback control performance is evaluated based on a weighted norm of the error between the reference and achieved values of the plasma parameters of interest. This performance measurement can also be used to compare the efficiency of this controller with other controllers that are using different control strategies and that are implemented in the same environment. The performance measures, that are used to measure the efficiency of the feedback control, are given by $J_\beta(t)$ to measure the performance of β feedback control and $J_\iota(t)$ for the performance measurement of ι . These functions are defined in [19] as:

$$\begin{aligned} J_\beta(t) &= \frac{(\beta(t) - \bar{\beta}(t))^2}{\bar{\beta}(t)^2} \\ J_\iota(t) &= \sum_{i=1}^{n_p} W(\rho_i) \frac{(\iota(x_i, t) - \bar{\iota}(\rho_i, t))^2}{\bar{\iota}(\rho_i, t)^2} \end{aligned} \quad (37)$$

In other words, $J_\iota(t)$ is integral of weighted ι tracking error, where the spatial weighting $W(\rho)$ is defined in [19] and is used to highlight the part of the region of interest to measure the performance of the feedback control. In these control scenarios more emphasis is given to the region closer to the plasma centre. The total weighting norm for the performance measurement of the combined control of the two parameters is given by $J_{tot}(t) = v_\beta J_\beta(t) + v_\iota J_\iota(t)$, where the weights v_β and v_ι are used to highlight the relative importance of the controlled plasma parameter of interest.

4. Simulation results

In the first phase, RAPTOR is run outside the TCV control environment as a simulator in stand-alone situation. The aim of this simulation environment is to have extensive tests of the feedback control performances before being applied in the real TCV experiments. Several control scenarios are considered to examine the robustness and performance of the controller. In these simulations the parameters of the RAPTOR

simulator are chosen to be the same with the one of the experimental setup in TCV. The configuration of the reference values of the current and power distributions are the same as in the TCV experiments. In this configuration, the flat-top plasma current is set to a constant $I_p = 120$ kA, while the EC clusters are used as control inputs. One P_{ec1} (counter-ECCD) and one P_{ec2} (co-ECCD) clusters are used in the TCV experiments. In these RAPTOR simulations, P_{ec1} has an allowed power range of 100 – 750 kW, and P_{ec2} has 360 – 900 kW. The RAPTOR simulator solves the non-linear coupled transport of T_e and Ψ represented by two coupled PDEs. The sawteeth effect is not included in this model. The n_e profile is prescribed in the RAPTOR simulations, and in real experiments, it can be time-varying and provided by a profile reconstruction code. In the current version of the RAPTOR code, the temperature of the ions T_i is estimated using pre-set ratio $f_{Ti}(\rho) = T_e/T_i$ and n_i is estimated using the assumption $n_e \approx n_i$.

The feedback control algorithm for the ι profile has available reconstructions obtained in $n_p = 11$ equidistant ρ -grid points. To implement the control given by the optimization problem (35), we apply numerical integration of (36). The simplified model used to represent the ECCD current and power distributions permits the optimization problem (35) to be solved as a constrained linear least square problem. The extracted feedforward stationary values of the inputs that correspond to the desired reference profiles (\bar{P}_{ec1} , \bar{P}_{ec2} and \bar{I}_p) are calculated solving the static equation (9) and added to the feedback control. In RAPTOR simulations we can generate the target profiles when the corresponding feedforward input signals are applied sufficiently long to bring the plasma in a steady state. Using these references in the control simulations we can achieve a perfect tracking of the target profile in a steady state. However, this is not the case in most of the real experiments where it is difficult to achieve a perfect tracking of the target profile due to the limited degrees of freedom of the available actuators.

The PI feedback control of the β parameter is calculated using a fixed values for the global energy confinement time $\tau_{th} = 4.4$ ms and the possible effective time delay is taken also into consideration. The optimal value of the free parameter is fixed at $\tau_c = 0.1$ after tuning, using simulation tests and choosing the best tracking performance. The ι controller parameter profiles, $\alpha_p(\rho)$ and $\alpha_I(\rho)$, that were used in these simulations are presented in Fig. 5a and Fig. 5b. The values of the control parameters were tuned such that they satisfy the stability criterion posed by Theorem 1. Because in these simulations we perform simultaneous control of the both parameters, the weighting functions are selected as $\omega_1 = \omega_2 = 0.5$.

Fig. 4 presents the results from the closed-loop RAPTOR simulation where several different reference points are extracted and the control is tested to track these different points. In this simulations the control is activated at $t = 0.1$ s. From the simulation results we can see that the closed-loop system successful tracks the multiple targets for both plasma parameters. In these plots we presented the performance of the control using different gains for the ι control, while the β controller gain was the same for both simulations. The second control scenario (control scenario 2) has increased control gains $\alpha_p(\rho)$ and $\alpha_I(\rho)$ in comparison with the first control scenario (control scenario 1). The gains of the first control scenario 1 are given in Fig. 5a and the gains of the control scenario 2 are given in Fig. 5b. From these plots we can see the effect of these gains in the control performance. The features of these terms are very similar with the one used in the control of SISO systems. Increasing the value of the proportional gain increases the setting time and decreases the steady state error

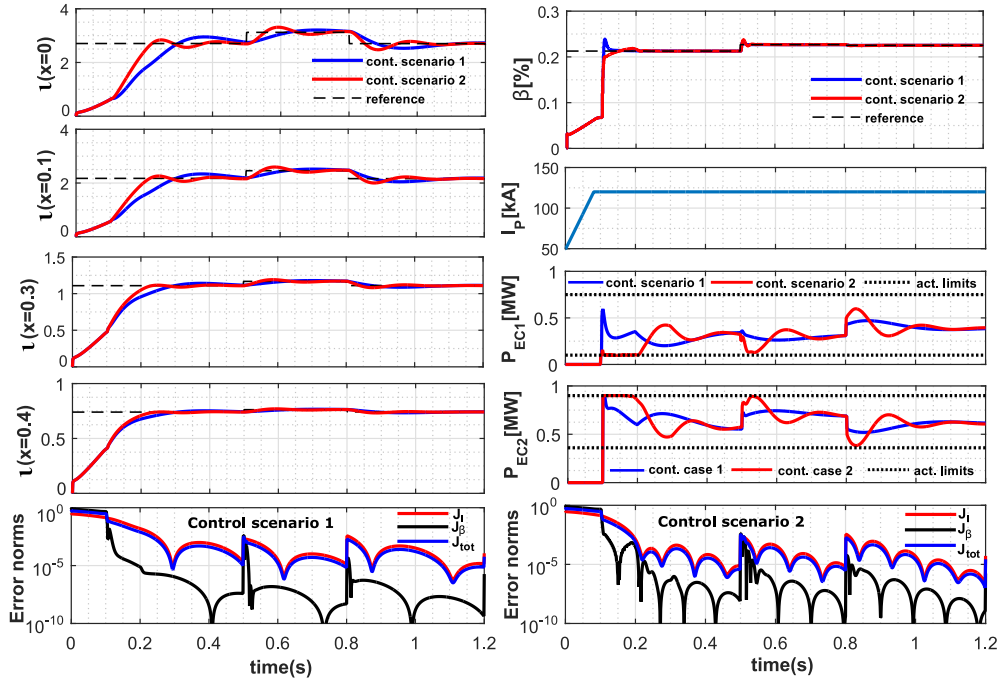


Figure 4: Evolution of ι , β , P_{aux} , I_p and the error norms in a closed-loop RAPTOR simulation.

while increasing the integral action eliminates the offset more efficiently. If these gains are too large, they can introduce oscillations and high sensitivity to high frequency disturbances.

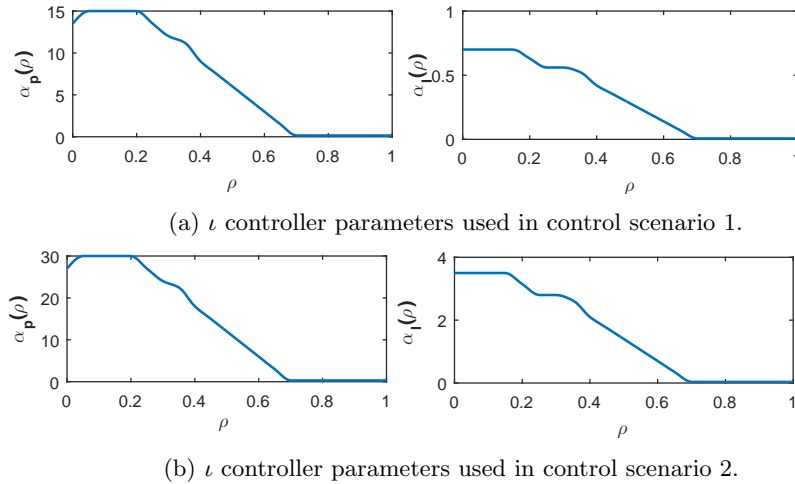


Figure 5: Values of controller gains α_p and α_I used in the RAPTOR simulations.

In Fig. 6, the ι profile of the closed-loop system from another RAPTOR simu-

lation is presented to illustrate the effect of using the parameter $\lambda(t)$. In the plots we are presenting two simulations using same parameters besides different $\lambda(t)$. In the first simulation the feedback control is activated and the forgetting factor $\lambda(t)$ is added to the integral action at $t = 0.1$ s. In these RAPTOR simulations the components that define $\lambda(t)$ are set to $\kappa = 0.08$ and $\sigma = 10^2$. This parameter was active only during the transition period when the control was activated at 0.1 s and was set to zero afterwards to avoid steady state error. In the second simulation only pure integrator ($\lambda(t) = 0$) is used. In the transition intervals we can have an overshoot in the closed-loop tracking as a result of the integral action. From the plots we can conclude that by adding $\lambda(t)$ we improve the performance of the integral action by reducing these overshoots.

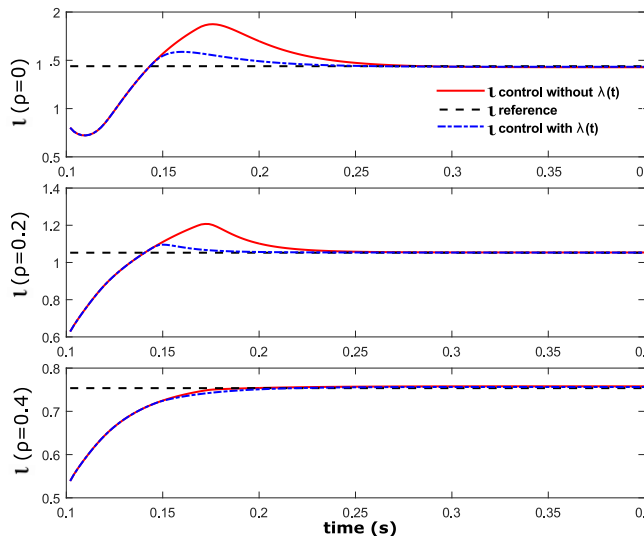


Figure 6: Response of ι with a pure integrator (red line) compared with the response when a forgetting factor $\lambda(t)$ is added to the integral action (blue dashed line) at several discrete locations.

4.1. Controller performance with time delays

Transportation of the control signals and their transition in the control systems can generate delays. Sometimes the effect of these delays cannot be neglected. For this reason, the performance of the control algorithm should be tested and adapted to these delays, which can be present when the control is applied on the real TCV control system. The control performance is tested by adding a 5 ms time delay in the plasma states from RAPTOR that enters in the controller, based on the possible time delays in the control environment.

The performance of the feedback control of the slow component ι are presented in Fig. 7. The time delays of small scales do not affect significantly the control of this component. In the case of a larger time delay of 100 ms and higher that can affect the performance, some extra tuning of the control parameters should be considered as discussed in [18]. The time delays in the system have more impact on the performance

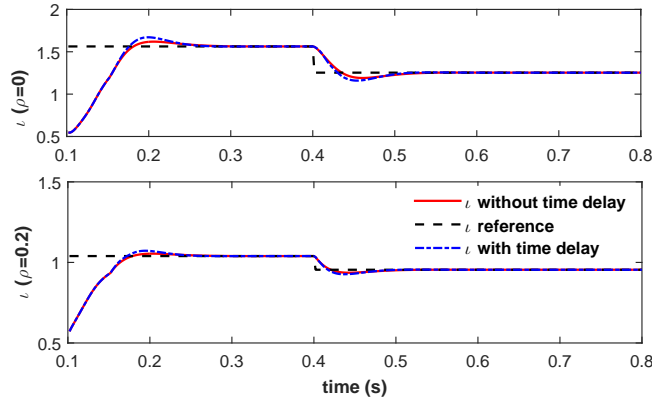
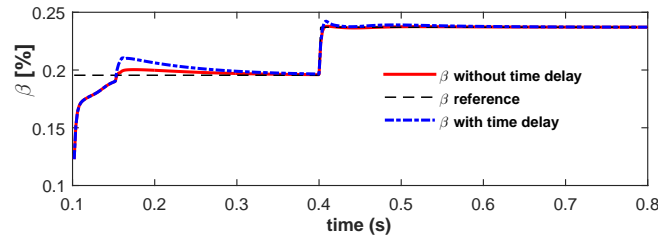
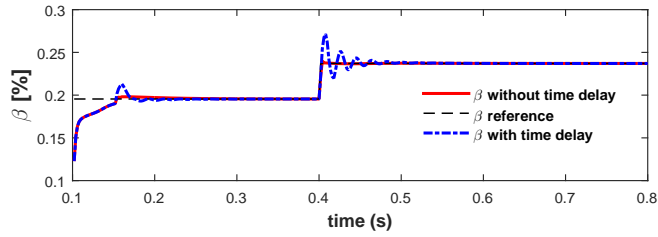


Figure 7: Tracking evolution of l in the presence of a time delay in the feedback loop.



(a) β tracking with $\tau_c = 0.1$.



(b) β tracking with $\tau_c = 0.01$.

Figure 8: Tracking evolution of β in a presence of a time delay of 5 ms in the feedback loop.

of the fast varying β parameter than on l . The results of control for β with $\tau_c = 0.1$ and $\tau_c = 0.01$ are presented in Fig. 8a and Fig. 8b, respectively. The robustness is increased and the convergence speed of the response is decreased by using an increased value of $\tau_c = 0.1$. In this case the high overshoots and oscillations are effectively attenuated even when a time delay of 5 ms is present. In the case when $\tau_c = 0.01$ is used the speed of the convergence is improved. On the other hand, when the time delay is introduced, there are higher overshoots and oscillations in the closed-loop tracking.

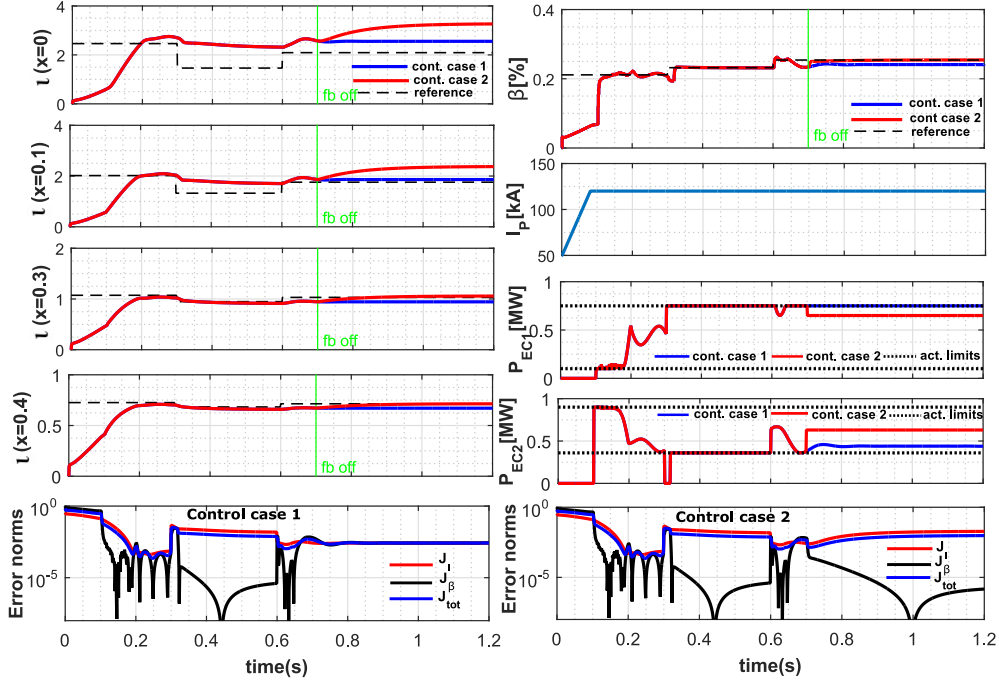


Figure 9: Evolution of ι , β , P_{aux} , I_p and the error norms from the RAPTOR simulation with changed deposition location of the actuator P_{ec2} . In control case 2 the feedback component is switched off at $t = 0.7$ s and only feedforward control is used.

4.2. Change of the deposition location and width of P_{ec2}

The control performance is tested in a scenario where the deposition location and width of the ECCD clusters are different than the ones given by the nominal model. For this purpose, the deposition in the second ECCD cluster (P_{ec2}) is changed such that the width is increased by 12.5% ($\omega_{dep} = 0.40$) and the location of the peak of the deposition is $\rho_{dep} = 0.05$. The desired reference profile is no longer reachable using the available inputs because of this perturbation that is introduced in the system. The simulation results for the tracking of ι profile and β are presented in Fig. 9. In these simulations we considered two control cases. In the first scenario the control is activated at $t = 0.1$ s (control case 1) and the control is active until the end of the simulation. While in the second scenario (control case 2) the feedback control was activated in the time window $t = 0.1 - 0.7$ s. The feedback control is deactivated after 0.7 s to compare the performance of the feedback control with the case where only the feedforward inputs are used. From Fig. 9, it can be clearly concluded that the feedback control improves the control performance of the ι profile by reducing the offset. In both cases a good tracking of β is still achieved because this parameter depends less on the deposition location and width of the ECCD clusters. Also the results reveal that q_0 is very sensitive to actuator settings.

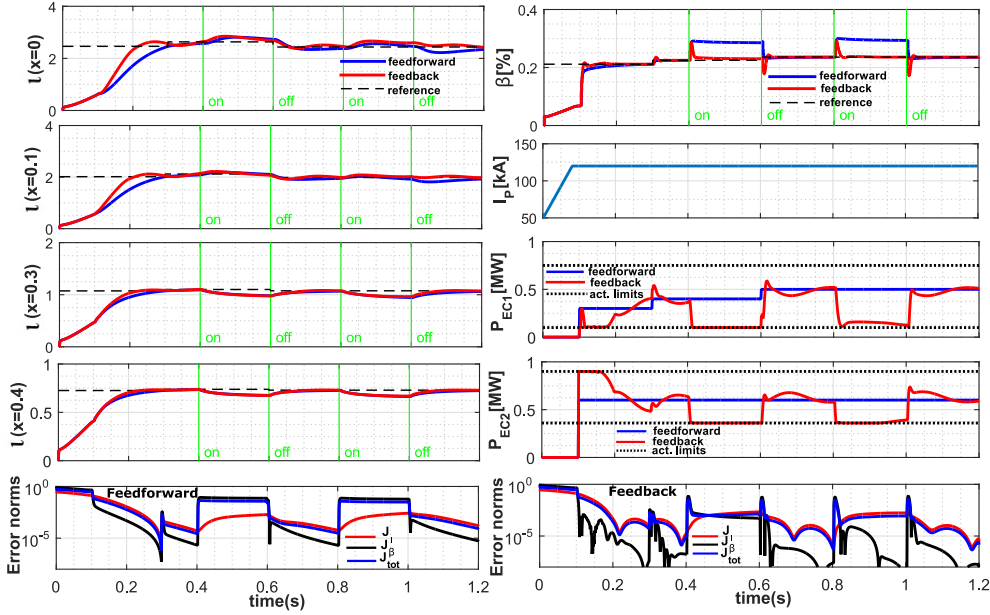


Figure 10: Evolution of l , β , P_{aux} , I_p and the error norms from RAPTOR simulation with added NBI heating as a source of disturbance. The input power is added at the time intervals $t \in [0.4, 0.6]$ s and $t \in [0.8, 1]$ s with a power of 1 MW.

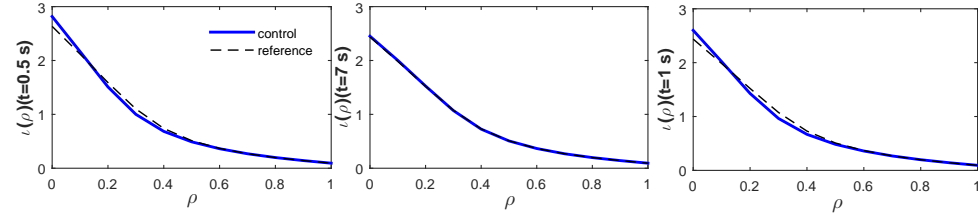


Figure 11: Profile of l at $t = 0.5$ s, 0.7 s and 1 s from the RAPTOR simulation with added NBI heating as a source of disturbance.

4.3. Disturbance attenuation

The sensitivity of the control algorithm to disturbances and their rejection is one of the most important features that should be tested before the controller is applied. In a complex system like the tokamak, where the model of the system depends on numerous parameters, there can exist many of sources of disturbances. To test the disturbance attenuation of the controller, an additional Neutral Beam Injection (NBI) with power and co-current drive was added to the simulations to act as a source of external disturbances. The model for the NBI that is used in RAPTOR simulator is simple and uses Gaussian distribution for the current density and for the heating power density. However, the accuracy of the model for these simulations is less important, since the main goal of this NBI is to act only as a disturbance source. The control of the system is activated at $t = 0.1$ s and the disturbing source was activated during

the time intervals $t \in [0.4, 0.6]$ s and $t \in [0.8, 1]$ s with a power of 1 MW. The results from the RAPTOR simulation are presented in Fig. 10. When the NBI heating is added the plasma parameters deviate from the target values and the feedback control algorithm tends to bring the plasma parameters to the desired references. In the plots we included the results from two simulations. In the first case only the feedforward inputs were used, while in the second case the feedback controls were added to the feedforward signals. From the results we can see that the feedback action successfully manages to attenuate the disturbance that comes from the NBI source. We can also confirm that the feedback control accelerates the convergence rate to the target state. Resulting ι -profile for this simulation at several time instants is presented in Fig. 11.

5. TCV experiments

After the extensive tests of the control algorithm using the RAPTOR simulator, the control algorithm is transferred and implemented in the TCV control system. The control environment [19] permits the MATLAB software for the controller to be used directly in the experiments.

The RAPTOR code is incorporated as an observer in the TCV control environment, which provides the essential real-time estimates of the T_e and Ψ profiles in several ρ -grid points. The T_e and Ψ profiles are reconstructed by the RAPTOR-observer using an extended Kalman filter scheme [21, 44]. These profile estimates are necessary to calculate the estimates of $\iota \propto \frac{1}{\rho} \frac{\partial \Psi}{\partial \rho}$ profile and $\beta \propto \int_0^1 V' n_e T_e d\rho$ for our control purposes. The TCV diagnostic system lacks measurements of the internal magnetic field in the plasma region. Thus, the estimated ι profile is calculated using only the RAPTOR model predictions constrained by the plasma geometry and experimental total plasma current measurement. The effect of sawteeth was not considered in the model, which results to a physically unrealistic estimate of ι profile that is much larger than 1 in many cases. However, this drawback does not affect the test of the performance of the presented control algorithm because the estimated and the reference profiles were generated using the same model. More details about the accuracy of the estimates and comparison of the profiles with off-line profile reconstructions using LIQUE [45] and ASTRA [25] can be found in the paper dedicated to the description of the control environment [19].

The volume-averaged density in the TCV experiments is calculated in real-time using the code described in [46]. The electron density is controlled using a robust density controller [47] and this control is executed in parallel with the profile control in the TCV control system. The density is chosen low ($n_e(\rho = 0) = 2 \cdot 10^{19} \text{m}^{-3}$) in order to maximize the ECCD current. See [19, 48, 49] for more details on the TCV control system.

In the TCV experiments the plasma current I_p is controlled by the inductive voltage from external coils, and can be adjusted in real-time. The value of this parameter is set to the desired value and is not included in our feedback control synthesis. In these experiments the cluster P_{ec1} has time-varying power limits. The range of this cluster is 100 – 750 kW, with an extra restriction that the minimal allowed power increases to 550 kW if $P_{ec1} < 500$ kW for more than 0.47 s. The power of cluster P_{ec2} is in the range of 360 – 900 kW.

RAPTOR observer provides the estimate of the ι profile at 11 equidistant ρ -grid points in the range of $\rho \in [0 : 1]$ for the control purposes. For the control synthesis a linear model was used that was extracted by applying static inputs $I_p = 120$ kA

and $P_{ec1} = P_{ec2} = 500$ kW. The feedforward signal of the control is calculated in real time using the lumped representation of the linearized model of (9) at several ρ -grid points. The corresponding signals were calculated as variations around this model. The small offset that can appear as a result of the nonlinear nature of the system is compensated easily by the integral action of the control. In the TCV experiments the optimization problem (35) is solved as a linear least square problem, without including the limits of the actuators. This permits easy implementation using only matrix multiplication and inversion, without significantly affecting the performance. The anti-windup compensator that is implemented in the TCV control system is updated with the real time actuator limits which are crucial for its performance. The simplicity of the control algorithm presented in this paper allows it to be executed in a short time and easily applicable within the TCV control system.

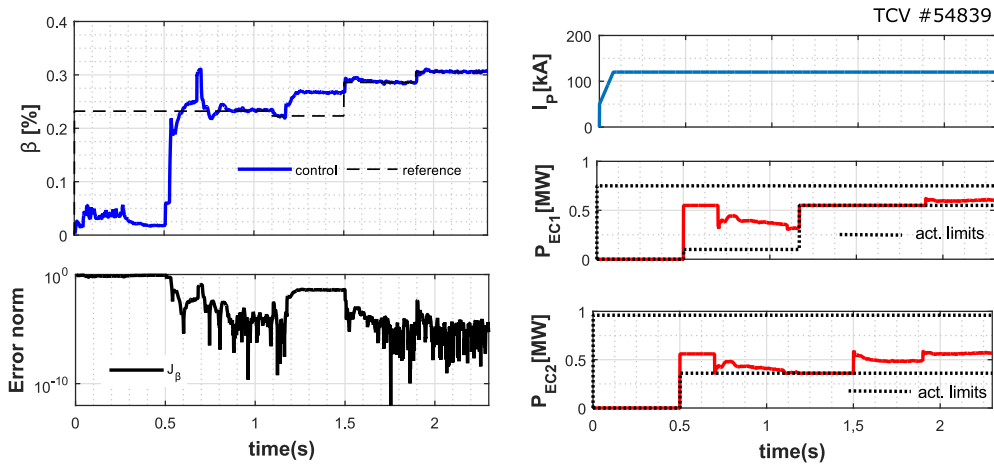


Figure 12: Evolution of β , P_{aux} , I_p and the error norm in shot # 54839. Two ECCD clusters are used as control inputs and the value of I_p is fixed.

The control strategy is tested in several TCV shots where different sets of control parameters are used. In the shot # 54839 only control of the pressure parameter β is considered. In these experiments the same β controller gains were used as in the simulations. In this experiment the controller should track several targets for β . The results from this shot are presented in Fig. 12. From the RAPTOR observer outputs we can see that the β controller successfully tracks the different targets. In the case where only β control is active, the total power is equally shared between the two clusters, thus the powers delivered by the two inputs are equal until one of the clusters saturates. Because of the variable power limitations of the first cluster P_{ec1} , after around 1.2 s the minimum limit of this actuator rises to 550 kW and the target state is no longer reachable. At 1.5 s and 2 s the target states were increased and they were successfully reached by the feedback controller.

The control of the ι profile is applied in shot # 54838. The values of the controller parameters $\alpha_p(\rho)$ and $\alpha_I(\rho)$ used in this shot are presented in Fig. 15a, and $\lambda(t) = 0$ was used. Since we have only two degrees of freedom given by the power delivered by the two actuators and several controlled points of the ι profile, the target profile should be chosen *a priori* such that it can be achieved with a limited number of actuators. From the plots shown in Fig. 13 we can see that the feedback control manages to reach

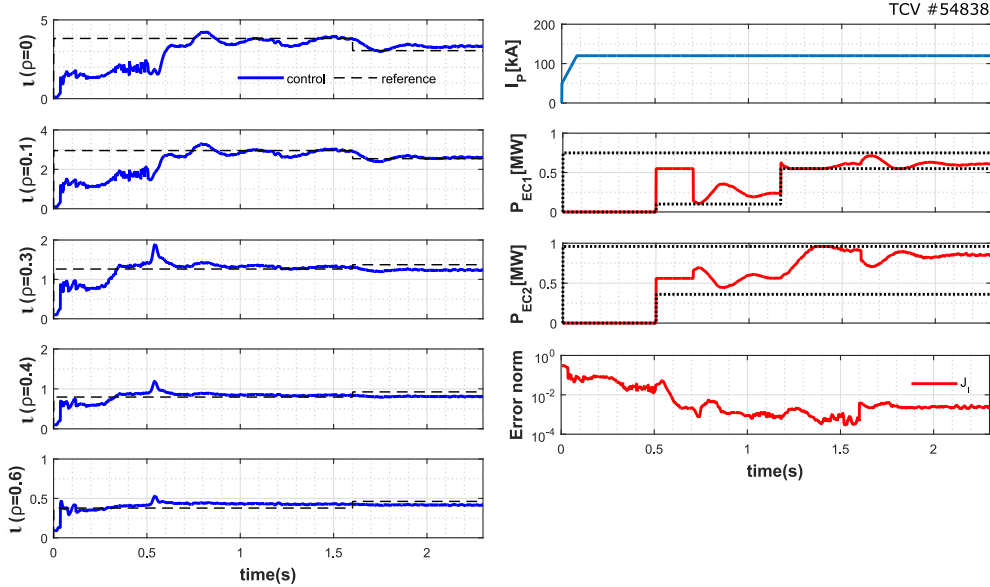


Figure 13: Evolution of ι , P_{aux} , I_p and the error norm in shot # 54838. Two ECCD clusters are used as control inputs and the value of I_p is fixed.

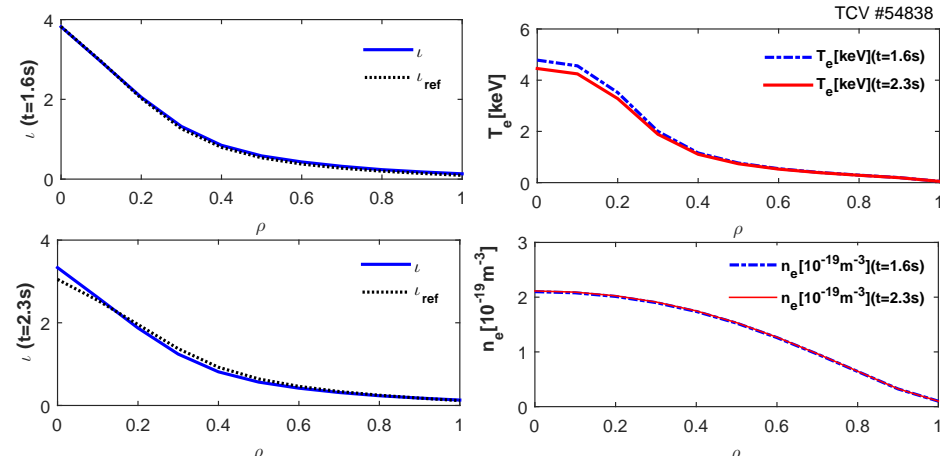


Figure 14: Profiles of ι , T_e and n_e in shot # 54838.

the desired target state after the activation of the control. In the ι profile plots we can observe certain oscillations that are coming from the increased integral action and also from the model uncertainties that are coming from different plasma phenomena. To reduce these oscillations in future experiments, the controller gains should be reduced and also techniques to filter the uncertainties can be applied. Because of the change of the minimum allowed power for the first cluster, at around 1.2 s the power P_{ec1} increases to 550 kW which results in an increase of P_{ec2} to compensate this change. At 1.6 s the target ι profile is changed and the feedback control aims at the new desired profile. The resulting and the target ι profile at $t = 1.6$ s and $t = 2.3$ s and

the temperature profile are presented in Fig. 14. From the plots we can see that not all the desired target states were reachable. This is because the reference parameters for the second target were generated by using high $I_p = 150$ kA. The feedback control applied using only the EC clusters gives more emphasis on the desired profile states closer to the center of the plasma. To achieve a better tracking of the points that are closer to the plasma edge, it is necessary to include also I_p as an boundary actuator.

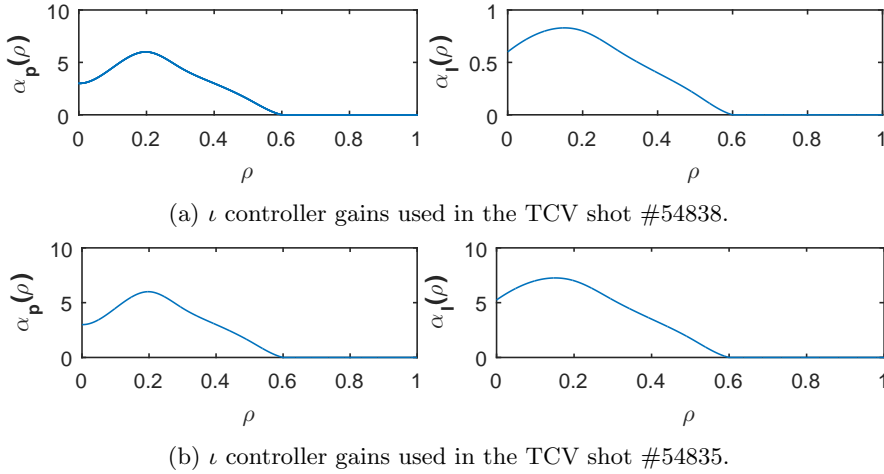


Figure 15: Values of controller gains α_p and α_I used in the TCV shots.

Simultaneous control of both ι and β is performed in shot # 54835. The values of the controller parameters $\alpha_p(\rho)$ and $\alpha_I(\rho)$ used in this shot are presented in Fig. 15b, and $\lambda(t) = 0$ was used. The results from this shot are presented in Fig. 16. The resulting and the target ι profile at $t = 1.6$ s and $t = 2.3$ s and the temperature profile are presented in Fig. 17. The control algorithm manages to achieve the desired ι profile and β targets at the beginning of the control. When the targets for both plasma quantities are changed at 1.6 s, these targets cannot be reached and the actuators are in saturation. As in the case of shot # 54838, the second target at 1.6 s were generated using $I_p = 150$ kA and they were not reachable using fixed $I_p = 120$ kA but are nevertheless relatively well reproduced within the actuator limits.

6. Conclusions and future work

In this paper we presented feedback control algorithms for simultaneous control of the plasma ι profile and the β pressure quantity. The control algorithms were successfully tested using simulations and real TCV experiments.

The results show a successful simultaneous tracking of ι and β using two EC clusters. The performance of the control using several control scenarios in presence of different uncertainties and disturbances was tested in RAPTOR simulations prior to the experiments. Finally, the feedback control algorithms were implemented on the TCV control system and tested in several shots.

While our algorithm is easy to be adopted to other in-domain actuators (auxiliary heating and current drive), it requires modifications or combination with other control methods to be extended to include boundary actuators. A boundary control algorithm

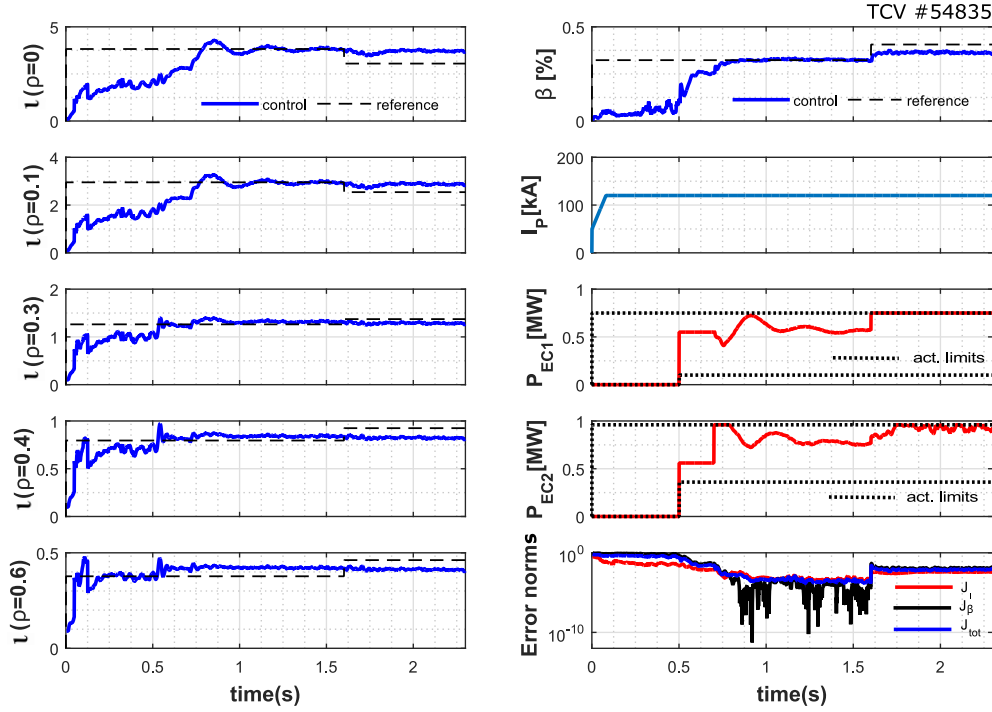


Figure 16: Evolution of l , β , P_{EC} , I_p and the error norms in shot # 54835. Two ECCD clusters are used as control inputs and the value of I_p is fixed.

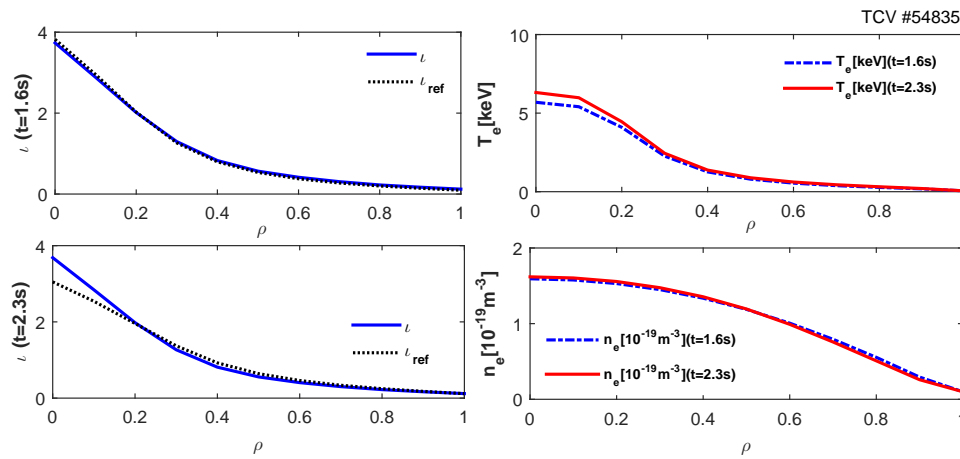


Figure 17: Profiles of l , T_e and n_e in shot # 54835.

applied to PDEs, that includes the boundary input I_p controlled directly by the loop voltage, could eventually improve the performance of the system [50, 51]. Direct application of these infinite-dimensional boundary control methods is difficult as a result of the complexity of the equation. However, a spatially discretized version of the backstepping approach that includes I_p as a control input is already applied for the control of the current profile and experimentally tested on DIII-D in L-mode [7] and in H-mode [52]. In [52] the boundary control method is extended to include in-domain control.

The unified framework for different control strategies if this TCV experimental campaign permits us to compare the performance of the different controllers that were implemented. One of the most successful control performances are done by using Model Predictive Control (MPC). MPC algorithms for the control of the plasma safety factor are developed in several works [19, 53, 54]. These algorithms are more complex and require implementation of quadratic programming solver that demanded bigger computational power. Successful testing of the MPC in this experimental campaign [19] shows very good handling of the actuator constraints that can be included explicitly in the MPC synthesis. A disturbance observer to estimate the slowly varying component is used to reduce the steady state error in this controller. This is equivalent to the objective of the integral action that is applied in our design. The disturbance observer gives a good transition between different target states and good tracking performance with reduced overshoots. Some of the techniques applied in the MPC controller can be used as an insight to be adopted to this control algorithm to improve the control performance. Qualitative comparison of the implemented MPC with the robust control [11, 13] and the IDA-PBC [5] can be found in [55].

The main advantages of the proposed algorithm in this paper is its simplicity to be implemented the TCV control system and the fast computational time. It can also incorporated easily some techniques (i.e anti-windup, proportional-integral term) that can improve the control performance. With the proposed Lyapunov approach we have stability margin that gives us some robustness to the uncertainties that can appear in the control design and implementation. The main drawbacks of this algorithm is that it has several parameters that must be tuned before it can be applied in real experiments. This requires very good simulator that reflects accurate the dynamics of the system. Moreover, the infinite-dimensional approach demands estimation of the plasma profiles in many grid points in order to be implemented, that is not always the case. In future works this algorithm should also be extended to include the dynamics and the terms that were neglected in our simplified control model, but that could influence the performance in other plasma operation mode.

Acknowledgments

This work has been carried out within the framework of the EUROfusion Consortium and has received funding from the Euratom research and training programme 2014-2018 under grant agreement No 633053. The views and opinions expressed herein do not necessarily reflect those of the European Commission. This work is also supported by the ANR project TORID contract number ANR-12-BS03-0008, Swiss National Science Foundation and the Netherlands Organisation for Scientific Research (NWO) via the Innovational Research Incentives Scheme.

Appendix A. Proof of Theorem 1

Proof. The time-derivative of the Lyapunov function is:

$$\dot{V} = \dot{V}_z + \dot{V}_I \quad (\text{A.1})$$

where

$$\begin{aligned} \dot{V}_z(t) &= \int_0^1 \rho^2 p_z(\rho) \frac{\partial \tilde{z}}{\partial t} \tilde{z} d\rho \\ &= \int_0^1 \rho^2 p_z(\rho) \frac{\partial}{\partial \rho} \left(\frac{a_1(\rho)}{\rho} \frac{\partial}{\partial \rho} (a_2(\rho) \tilde{z}) \right) \tilde{z} d\rho \\ &\quad - \int_0^1 \rho^2 p_z(\rho) \alpha_p(\rho) \tilde{z}^2 d\rho + \int_0^1 \rho^2 p_z(\rho) \tilde{z} \mathcal{I}_l d\rho \end{aligned} \quad (\text{A.2})$$

Applying integration by parts in $\dot{V}_z(t)$ and defining $k_z(\rho) = \alpha_p(\rho) p_z(\rho)$ we get:

$$\begin{aligned} \dot{V}_z(t) &= \rho p_z(\rho) \tilde{z} a_1(\rho) \frac{\partial}{\partial \rho} (a_2(\rho) \tilde{z}) \Big|_0^1 - \int_0^1 \rho p_z(\rho) a_1(\rho) a_2(\rho) \left(\frac{\partial \tilde{z}}{\partial \rho} \right)^2 d\rho \\ &\quad - \int_0^1 \left(2p_z(\rho) + \rho p'_z(\rho) \right) a_1(\rho) \frac{\partial a_2(\rho)}{\partial \rho} \tilde{z}^2 d\rho \\ &\quad - \int_0^1 \left(2p_z(\rho) a_2(\rho) + \rho p_z(\rho) \frac{\partial a_2(\rho)}{\partial \rho} + \rho p'_z(\rho) a_2(\rho) \right) a_1(\rho) \tilde{z} \frac{\partial \tilde{z}}{\partial \rho} d\rho \\ &\quad - \int_0^1 \rho^2 k_z(\rho) \tilde{z}^2 d\rho + \int_0^1 \rho^2 p_z(\rho) \tilde{z} \mathcal{I}_l d\rho \end{aligned} \quad (\text{A.3})$$

Calculating the time derivative of $\dot{V}_I = \int_0^1 \rho^2 p_I(\rho) \frac{\partial \mathcal{I}_l}{\partial t} \mathcal{I}_l d\rho$ and defining $k_I(\rho) = \alpha_I(\rho) p_I(\rho)$, we get:

$$\dot{V}_I(t) = - \int_0^1 \rho^2 k_I(\rho) \tilde{z} \mathcal{I}_l d\rho - \int_0^1 \rho^2 p_I(\rho) \lambda(t) \mathcal{I}_l^2 d\rho \quad (\text{A.4})$$

Combining the equations (A.3) - (A.4), \dot{V} can be presented as:

$$\dot{V}(t) = \int_0^1 \begin{bmatrix} \tilde{z} \\ \frac{\partial \tilde{z}}{\partial \rho} \\ \mathcal{I}_l \end{bmatrix}^T \begin{bmatrix} c_{1,1}(\rho) & c_{1,2}(\rho) & c_{1,3}(\rho) \\ 0 & c_{2,2}(\rho) & 0 \\ c_{1,3}(\rho) & 0 & c_{3,3}(\rho) \end{bmatrix} \begin{bmatrix} \tilde{z} \\ \frac{\partial \tilde{z}}{\partial \rho} \\ \mathcal{I}_l \end{bmatrix} \quad (\text{A.5})$$

where $c_{1,1}, c_{2,2}, c_{1,2}, c_{1,3}, c_{3,3}$ are defined in the statement of Theorem 1. Considering (25), we get $\dot{V} \leq -\frac{1}{2}\gamma_1 \int_0^1 \rho^2 p_z(\rho) \tilde{z}^2 d\rho - \frac{1}{2}\gamma_2(t) \int_0^1 \rho^2 p_I(\rho) \mathcal{I}_l^2 d\rho$. This concludes the proof of Theorem 1. \square

Appendix B. Computation of the Lyapunov function

The exponential stability of the system is verified by solving inequality (25) in Theorem 1. The solution of this inequality is found by transforming the inequality in

to an Linear Matrix Inequality (LMI) problem by representing the weights of the Lyapunov functions, $p_z(\rho)$, $p_I(\rho)$ and the control gains $k_z(\rho), k_I(\rho)$ as Legendre polynomials. Legendre polynomials are orthogonal in the range $[-1, 1]$, and the polynomials $p_z(\rho)$, $p_I(\rho)$, $k_z(\rho)$ and $k_I(\rho)$ may be expanded in this interval in terms of them as [56]. The $p_z(\rho)$ can be presented as:

$$p_z(\rho) = \sum_{i=0}^{N_1} c_{1,i} P_i(\rho)$$

where $c_{1,1}, \dots, c_{1,N_1}$ are constant that need to be found, $P_i(x)$ is i -th order Legendre polynomial and N_1 is the order of the Legendre polynomial for p_z . The same representation is applied for $p_I(\rho)$, $k_z(\rho)$ and $k_I(\rho)$.

Sampling the interval $[0, 1]$ and representing $p_z(\rho)$, $p_I(\rho)$, $k_z(\rho)$ and $k_I(\rho)$ as a sum of Legendre polynomials permits us to formulate the following LMI problem:

For given parameters γ_1 and γ_2 , find $p_z(\rho), p_I(\rho), k_z(\rho)$ and $k_I(\rho)$ such that:

- $A(\rho) + A_1(\rho) \leq 0, \forall \rho \in [0, 1]$;
- $\alpha_l W_p(\rho) < p_z(\rho) \leq \alpha_h W_p(\rho)$ and $\alpha_l W_p(\rho) < p_I(\rho) \leq \alpha_h W_p(\rho), \forall \rho \in [0, 1]$;
- $k_z(\rho) > 0$ and $k_I(\rho) > 0$;
- $\eta_{\parallel}(\rho) = \omega_{\eta} \eta_{\parallel, \min}(\rho) + (1 - \omega_{\eta}) \eta_{\parallel, \max}(\rho), \forall \omega_{\eta} \in [0, 1]$.

The LMI is solved to guarantee the stability of the closed-loop system for different values of $\lambda_{max} \geq \lambda(t) > 0$. The LMI solver YALMIP [57] is used to find the unknown constant polynomial coefficients in $p_z(\rho), p_I(\rho), k_z(\rho)$ and $k_I(\rho)$.

The shape of the weighting polynomials of the Lyapunov function $p_z(\rho)$ and $p_I(\rho)$ are constrained by the lower limit curve $\alpha_l W_p(\rho) > 0$ and the upper limit curve $\alpha_h W_p(\rho) > 0$, where the weighting function $W_p(\rho)$ is chosen such that it highlights the area of ρ which is the most important for the control performance. In these experiments the shape of $W_p(\rho)$ is chosen according to the weighting function $W(\rho)$ defined in the performance criteria (37). The plots of the solutions of the polynomials p_z and p_i in a case where these constraints are applied are presented in Fig. B1.

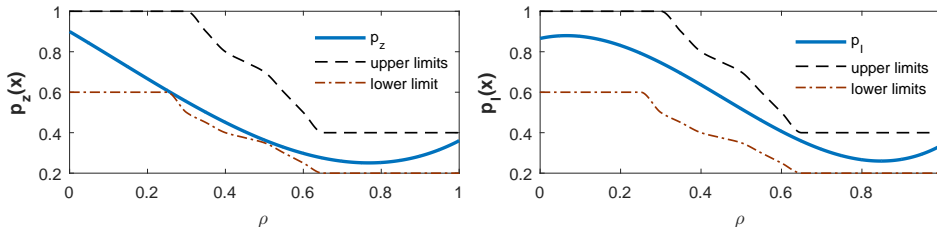


Figure B1: Plots of the solution p_z and p_I and the constrained .

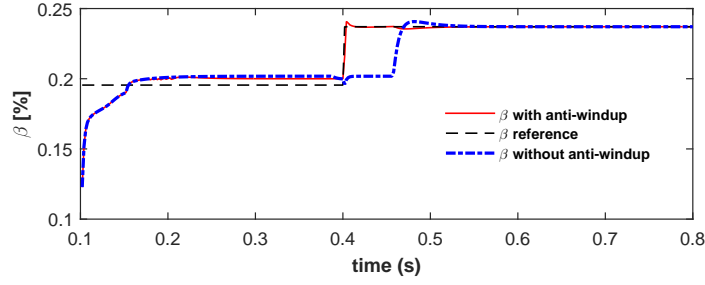


Figure C1: RAPTOR simulations comparing the performance of the control of β without (blue dash line) and with (red line) an anti-windup scheme.

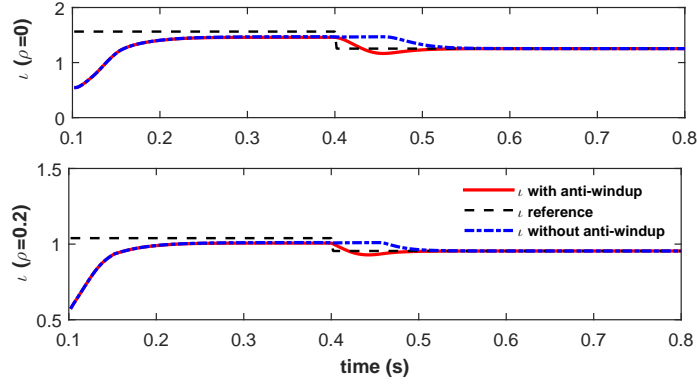


Figure C2: RAPTOR simulations comparing the performance of the control of l with (blue dash line) and without (red line) an anti-windup scheme.

Appendix C. Anti-windup implementation

The powers of auxiliary antennas used in the tokamaks are subject to saturation. The saturation for the i -th actuator can be presented as:

$$\text{sat}(P_{EC,i}) = \begin{cases} P_{EC,i,\min} & \text{if } P_{EC,i} < P_{EC,i,\min} \\ P_{EC,i} & \text{if } P_{EC,i,\min} \leq P_{EC,i} \leq P_{EC,i,\max} \\ P_{EC,i,\max} & \text{if } P_{EC,i} \geq P_{EC,i,\max} \end{cases} \quad (\text{C.1})$$

Integral windup can occur in loops where the actuator saturates and the controller has integral action. It occurs when the integral terms accumulate a significant error as a result of saturated integrators when the desired output is not achievable. This can lead to large overshoots and undesirable transients. To avoid windup, an extra feedback path is provided in the controller by comparing the difference between the output of the controller and the actuator output. Stability and anti-windup techniques for systems with saturated actuators can be found in [58, 59].

We applied anti-windup compensation by using the conditional integration (CI) technique that is presented in [60, 61]. In the case where the saturation levels are

known and where digital controller is used, this method can be effectively and easily implemented in the control. In this technique, the integral term is increased only when certain conditions are satisfied; otherwise it is kept constant. In the case of the auxiliary heating sources in tokamak plasma, the CI is applied such that the integral action is suspended when the actuators powers are saturated. In our case, the CI is calculated for each actuator separately. For the z component, the integral term $\tilde{\mathcal{I}}_z$ is frozen when the actuators saturate as:

$$\frac{\partial \tilde{\mathcal{I}}_z}{\partial t} = \begin{cases} 0 & \text{if } \tilde{u} \text{ saturates} \\ -\alpha_I(\rho)\tilde{z} - \lambda(t)\mathcal{I}_z & \text{otherwise} \end{cases} \quad (\text{C.2})$$

In the case of the controller of β , the integral term $\tilde{\mathcal{I}}_\beta$ is frozen when the actuators are being driven into saturation:

$$\frac{d\tilde{\mathcal{I}}_\beta}{dt} = \begin{cases} 0 & \text{if } \tilde{u} \text{ saturates} \\ K_{\beta,i}\tilde{\beta} & \text{otherwise} \end{cases} \quad (\text{C.3})$$

The integration stops only when all the actuators are in saturation. Fig. C1 and Fig. C2 present the plots of β and ι -profile from a closed-loop RAPTOR simulation. In the time interval $t \leq 0.4$ s, the reference values of the outputs were selected such that the system actuators are forced to reach a saturation level. In this interval the integrator continues to accumulate the error if no anti-windup scheme is applied. After $t = 0.4$ s the target values are changed and the performance of the anti-windup technique can be tested. In the case when the anti-windup technique is not applied, there are delays in the controllers response to the change of the reference signal. From these results it can be concluded that the simple CI technique successfully solves the controller windup problem.

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