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MRAS-based Sensorless Speed Integral Backstepping Control for Induction Machine, using a Flux Backstepping Observer

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ABSTRACT

This paper presents a study concerning a sensorless vector control of an induction machine fed by a voltage source inverter. The aim is to provide a scheme to control the speed and the rotor flux using a sensorless integral backstepping control approach. The rotor speed estimation is done by an observer using the model reference adaptive system (MRAS) technique whereas the nonlinear backstepping observer is used to get the rotor flux. The main objective is to achieve a robust control, adaptive and efficient, which will allow us to test and evaluate the performance of the proposed observer, combined with a sensorless control of the induction machine. Tests and validation are done using numerical simulations with MATLAB/SIMULINK-PSB (Power System Block set) toolbox. The results show good performance in terms of robustness regarding machine parameter variations and show the excellent quality of the control law associated with the observer, despite the observability problems when the machine operates at low speed.

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1. INTRODUCTION

In the recent decades, the variable speed electric drives developed rapidly and aroused significant interest in most sectors of industrial activity implementing the driving force, due to technological advances, but also methodological [1]. The direct current machine has long been the best source of electro mechanical variable speed because of its ease of control. This machine, with its two coils to orthogonal fields, presents a natural decoupling between stator flux and electromagnetic torque [2]. However, its major disadvantage is the existence of the "collector-brush" device which is the origin of his employment limitations. This mechanical device is poorly tolerated in some applications for its high price, its frequent and costly maintenance, but also because of the presence of electrical arcs prohibiting use in explosive atmospheres. Due to these constraints, research was carried for other solutions, exploiting the AC machines and especially the squirrel cage induction motors (IM) [3]. The IM has certain advantages related to its simple design, robustness, low cost and maintenance. Also, it is widely used in many fields such as aerospace, rail and marine traction, transportation systems and in a multitude of industrial applications (machine tools, driving machines, pumping systems, ventilation and lifting [4].

However, the use of the induction machine for variable speed drives raises a couple of difficulties, which leads to a complex control compared to that of another motor. Indeed, this type of motor is characterized by a multivariable and nonlinear mathematical model, with a strong coupling between the control variables of the magnetic flux and the electromagnetic torque. It is not possible to control a motor

output variable (torque, speed or position) by varying a single input variable (current or voltage). Moreover, some of its state variables are not accessible to measurement and the motor parameters are subject to variations in time [5], [6]. The implementation of these control methods were only possible through the development of digital signal controllers, microcontrollers, static converters. Also, several techniques have been proposed to obtain for the induction machine, the high-performance control devices, and experience its qualities and benefits in the field of variable speed [7].

Among the controls, we distinguish the vector control based on dynamic models, using the Park transformation to solve the torque-flux decoupling issue. The combination of this technique with a modern automatic approaches based on nonlinear control algorithms such as the Backstepping control contribute to solve the robustness problem (the change of load torque and machine parameters). However, it requires the installation of an incremental encoder to measure the rotor speed [8-9]. Using the encoder leads to additional cost which can be higher than the machine one, for low power application. In addition, there must be an additional space for installing the encoder. The system reliability decreases because of this fragile device. Because of all these inconvenient, one tried to exclude the use the incremental encoder and research on the sensorless control of induction machines began. Several strategies were proposed in the literature to achieve this objective [10-11]. Much of the proposed methods are based on observers and estimators in open loop. They largely depend on the model of the induction machine. For this reason, these techniques fail to replace the incremental encoder in the field of low speeds. Further research based on the combination of modern techniques and theories including adaptive laws. These techniques promise to give better results, even in the low-speed including stopped.

In this paper, we chose to use a nonlinear integral backstepping control design scheme, for speed control and we propose a new combination for the realization of robust observers, by using the model reference adaptive system (MRAS) to estimate the rotor speed and the Backstepping observer to estimate the rotor flux.

2. INDUCTION MOTOR MODEL AND VECTOR CONTROL

2.1 Model of IM in (d-q) Reference Frame

The mathematical model of the induction machine in the (d-q) synchronously rotating frame is given by the following nonlinear equations [12-13]:

$$\begin{aligned} \dot{i}_{sd} &= a_1 i_{sd} + \omega_s i_{sq} + a_2 \phi_{rd} + a_3 \omega_r \phi_{rq} + b v_{sd} \\ \dot{i}_{sq} &= -\omega_s i_{sd} + a_1 i_{sq} - a_3 \phi_{rd} \omega_r + a_2 \phi_{rq} + b v_{sq} \\ \dot{\phi}_{rd} &= a_4 i_{sd} - a_5 \phi_{rd} + (\omega_s - \omega_r) \phi_{rq} \\ \dot{\phi}_{rq} &= a_4 i_{sq} - (\omega_s - \omega_r) \phi_{rd} - a_5 \phi_{rq} \\ \dot{\omega}_r &= a_6 (i_{sq} \phi_{rd} - i_{sd} \phi_{rq}) - a_7 \omega_r - a_8 T_l \end{aligned} \quad (1)$$

With

$$\begin{aligned} a_1 &= -b(R_s + \frac{L_m^2}{L_r} R_r), \quad a_2 = \frac{L_m R_r}{(\sigma L_s L_r)}, \quad a_3 = -\frac{L_m}{(\sigma L_s L_r)} \\ a_4 &= \frac{L_m R_r}{L_r}, \quad a_5 = \frac{R_r}{L_r}, \quad a_6 = \frac{p^2 L_m}{J L_r}, \quad a_7 = \frac{f_c}{J}, \quad a_8 = \frac{p}{J} \\ \sigma &= 1 - \frac{L_m^2}{L_s L_r}, \quad b = \frac{1}{\sigma L_s} \end{aligned}$$

2.2 Direct rotor flux orientation (DRFO)

The aim of vector control is to let the induction machine effectively respond to reference variations (rotor position, torque, speed) in a wide range of operation and for applications requiring high dynamic performance. It is based on the instantaneous torque control and a suitable choice of reference (d, q) of Park, so that the d-axis is aligned with the rotor flux vector, to obtain the decoupling between the electromagnetic torque and rotor flux. This alignment imposes the following condition: [14-15]:

$$\phi_r = \phi_{rd} \text{ and } \phi_{rq} = 0 \quad (2)$$

This choice allows to control the torque via the current i_{sq} and the rotor flux through the current i_{sd} .

3. INTEGRAL BACKSTEPPING CONTROL

3.1 Principle of the integral backstepping control

The Backstepping technique is a relatively new control method of nonlinear systems. The principle of this approach is to stabilize at the beginning the first sub system by a stabilizing function known via a chosen Lyapunov function, then to add to its error an integrator. The same procedure for the successive subsystems to finally reach a global Lyapunov function giving the overall control law that stabilizes the system [16-17]. The combination of technical Backstepping with vector control of the induction machine gives the better performance of the control law [18-19].

3.2 The integral Backstepping control for IM and rotor flux estimator

The objective to design an integral backstepping control, allowing to ensure the tracking of rotor speed and rotor flux for the machine [20-21], we define the errors e_{ω_r} and $e_{\varphi_{rd}}$ respectively representing the error between the actual speed ω_r and the reference speed ω_r^* and the error between the flux module φ_{dr} and its reference φ_{dr}^* .

a) First step "speed and flux loop"

$$\begin{aligned} e_{\omega_r} &= \omega_r^* - \omega_r + Z_{\omega_r} \\ e_{\varphi_{rd}} &= \varphi_{rd}^* - \varphi_{rd} + Z_{\varphi_{rd}} \end{aligned} \quad (3)$$

With:

$$\begin{aligned} Z_{\varphi_{rd}} &= k'_{\varphi_{rd}} \int_0^t (\varphi_{rd}^* - \varphi_{rd}) dt \\ Z_{\omega_r} &= k'_{\omega_r} \int_0^t (\omega_r^* - \omega_r) dt \end{aligned}$$

Then the dynamical error equations are given by:

$$\begin{aligned} \dot{e}_{\omega_r} &= \dot{\omega}_r^* - \dot{\omega}_r + \dot{Z}_{\omega_r} \\ \dot{e}_{\varphi_{rd}} &= \dot{\varphi}_{rd}^* - \dot{\varphi}_{rd} + \dot{Z}_{\varphi_{rd}} \end{aligned} \quad (4)$$

With:

$$\begin{aligned} \dot{Z}_{\varphi_{rd}} &= k'_{\varphi_{rd}} (\varphi_{rd}^* - \varphi_{rd}) \\ \dot{Z}_{\omega_r} &= k'_{\omega_r} (\omega_r^* - \omega_r) \end{aligned}$$

Then the error dynamical equations becomes:

$$\begin{aligned} \dot{e}_{\omega_r} &= \dot{\omega}_r^* - a_6 \cdot (i_{sq} \cdot \varphi_{rd}) + a_7 \cdot \omega_r + a_8 \cdot T_l + \dot{Z}_{\omega_r} \\ \dot{e}_{\varphi_{dr}} &= \dot{\varphi}_{rd}^* - a_4 \cdot i_{sd} + a_5 \cdot \varphi_{rd} + \dot{Z}_{\varphi_{dr}} \end{aligned} \quad (5)$$

We require that $\dot{e}_{\omega_r} = -k_{\omega_r} e_{\omega_r}$ and $\dot{e}_{\varphi_{dr}} = -k_{\varphi_{dr}} e_{\varphi_{dr}}$, where k_{ω_r} and $k_{\varphi_{dr}}$ are positives constants.

We can use the Lyapunov function:

$$v = \frac{1}{2} (e_{\omega_r}^2 + e_{\varphi_{dr}}^2) \quad (6)$$

Its derivative is:

$$\begin{aligned}\dot{v} &= e_{\omega_r} \dot{e}_{\omega_r} + e_{\varphi_{rd}} \dot{e}_{\varphi_{rd}} \\ \dot{v} &= e_{\omega_r} (\dot{\omega}_r^* - a_6 (\dot{i}_{sq} \varphi_{rd}) + a_7 \omega_r + a_8 T_l + \dot{z}_{\omega_r}) + e_{\varphi_{rd}} (\dot{\varphi}_{rd}^* - a_4 \dot{i}_{sd} + a_5 \varphi_{rd} + \dot{z}_{\varphi_{rd}})\end{aligned}\quad (7)$$

The aim of tracking is fulfilled ($\dot{v} < 0$) choosing the virtual control that represents the stabilization function as follows:

$$\begin{aligned}\hat{i}_{sq}^* &= \frac{1}{\varphi_{rd} a_6} [k_{\omega_r} e_{\omega_r} + \dot{\omega}_r^* + a_7 \omega_r + a_8 T_l + \dot{z}_{\omega_r}] \\ \hat{i}_{sd}^* &= \frac{1}{a_4} [k_{\varphi_{rd}} e_{\varphi_{rd}} + \dot{\varphi}_{rd}^* + a_5 \varphi_{rd} + \dot{z}_{\varphi_{rd}}]\end{aligned}\quad (8)$$

The derivative of the Lyapunov function becomes:

$$\dot{v} = -k_{\omega_r} e_{\omega_r}^2 - k_{\varphi_{rd}} e_{\varphi_{rd}}^2 < 0$$

So, the errors in (3) converge to zero.

The state φ_{rd} is unknown, so we need to use his estimation in (8), so the new expression:

$$\begin{aligned}\hat{i}_{sq}^* &= \frac{1}{\hat{\varphi}_{rd} a_6} [k_{\omega_r} e_{\omega_r} + \dot{\omega}_r^* + a_7 \omega_r + a_8 T_l + \dot{z}_{\omega_r}] \\ \hat{i}_{sd}^* &= \frac{1}{a_4} [k_{\varphi_{rd}} e_{\varphi_{rd}} + \dot{\varphi}_{rd}^* + a_5 \hat{\varphi}_{rd} + \dot{z}_{\varphi_{rd}}]\end{aligned}\quad (9)$$

b) Second step "currents loop"

Once virtuals inputs are calculated, we define the error on the currents as follows:

$$\begin{aligned}e_{i_{sq}} &= \hat{i}_{sq}^* - i_{sq} + Z_{i_{sq}} \\ e_{i_{sd}} &= \hat{i}_{sd}^* - i_{sd} + Z_{i_{sd}}\end{aligned}\quad (10)$$

With:

$$\begin{aligned}Z_{i_{sq}} &= k'_{i_{sq}} \int_0^t (i_{sq}^* - i_{sq}) dt \\ Z_{i_{sd}} &= k'_{i_{sd}} \int_0^t (i_{sd}^* - i_{sd}) dt\end{aligned}$$

Then, the error dynamical equations are expressed by:

$$\begin{aligned}e_{i_{sq}} &= \frac{1}{\hat{\varphi}_{rd} a_6} [k_{\omega_r} e_{\omega_r} + \dot{\omega}_r^* + a_7 \omega_r + a_8 T_l + \dot{z}_{\omega_r}] - \dot{i}_{sq} + Z_{i_{sq}} \\ e_{i_{sd}} &= \frac{1}{a_4} [k_{\varphi_{rd}} e_{\varphi_{rd}} + \dot{\varphi}_{rd}^* + a_5 \hat{\varphi}_{rd} + \dot{z}_{\varphi_{rd}}] - \dot{i}_{sd} + Z_{i_{sd}}\end{aligned}$$

Then, the error equations (5) can be expressed as:

$$\begin{aligned}\dot{e}_{\omega_r} &= -k_{\omega_r} e_{\omega_r} + a_6 \hat{\varphi}_{rd} e_{i_{sq}} - a_6 \hat{\varphi}_{rd} Z_{i_{sq}} \\ \dot{e}_{\varphi_{rd}} &= -k_{\varphi_{rd}} e_{\varphi_{rd}} + a_4 e_{i_{sd}} - a_5 \hat{\varphi}_{rd} - a_4 Z_{i_{sd}}\end{aligned}\quad (11)$$

Where the flux estimation error is:

$$\tilde{\varphi}_{rd} = \hat{\varphi}_{rd} - \varphi_{rd}$$

As well, the dynamical equations for the error $e_{i_{sq}}$, $e_{i_{sd}}$ can be expressed as:

$$\begin{aligned} \dot{e}_{i_{sq}} &= \hat{i}_{sq}^* - \dot{i}_{sq} + \dot{z}_{i_{sq}} \\ \dot{e}_{i_{sd}} &= \psi_{i_{sq}} - \lambda_{i_{sq}} - b v_{sq} + \dot{z}_{i_{sq}} \end{aligned} \quad (12)$$

And

$$\begin{aligned} \dot{e}_{i_{sd}} &= \hat{i}_{sd}^* - \dot{i}_{sd} + \dot{z}_{i_{sd}} \\ \dot{e}_{i_{sd}} &= \psi_{i_{sd}} - \lambda_{i_{sd}} - \frac{k_{\varphi_{rd}}}{L_m} \tilde{\varphi}_{rd} - b v_{sd} + \dot{z}_{i_{sd}} \end{aligned} \quad (13)$$

Where:

$$\begin{aligned} \psi_{i_{sq}} &= \frac{1}{\hat{\varphi}_{rd} \cdot a_6} [k_{\omega_r} e_{\omega_r} + \dot{\omega}_r^* + a_7 \cdot \omega_r + a_8 \cdot T_1 + \dot{z}_{\omega_r}] + \frac{1}{\hat{\varphi}_{rd} \cdot a_6} [k_{\omega_r} \dot{e}_{\omega_r} + \ddot{\omega}_r^* + a_7 \cdot \dot{\omega}_r + a_8 \cdot \dot{T}_1 + \ddot{z}_{\omega_r}] \\ \psi_{i_{sd}} &= \frac{1}{a_4} [k_{\varphi_{rd}} \dot{e}_{\varphi_{rd}} + \ddot{\varphi}_{rd}^* + a_5 \hat{\varphi}_{rd} + \ddot{z}_{\varphi_{rd}}] \\ \lambda_{i_{sq}} &= -\omega_s \cdot i_{sd} + a_1 \cdot i_{sq} - a_3 \cdot \varphi_{rd} \cdot \omega_r + a_2 \cdot \varphi_{rq} \\ \lambda_{i_{sd}} &= a_1 \cdot i_{sd} + \omega_s \cdot i_{sq} + a_2 \cdot \varphi_{rd} + a_3 \omega_r \cdot \varphi_{rq} \end{aligned}$$

Now we extend the Lyapunov function in (6) to include the new states variables $e_{i_{sq}}$, $e_{i_{sd}}$ and the parameter estimation errors $\tilde{\varphi}_{rd}$ as:

$$v = \frac{1}{2} (e_{\omega_r}^2 + e_{\varphi_{rd}}^2 + e_{i_{sq}}^2 + e_{i_{sd}}^2 + \frac{1}{\gamma} \tilde{\varphi}_{rd}^2) \quad (14)$$

Where γ is a positive constant known as the adaptive gain. We use the energy function v , to obtain both the control and estimation algorithms.

The derivative of the above equation is:

$$\dot{v} = e_{\omega_r} \dot{e}_{\omega_r} + e_{\varphi_{rd}} \dot{e}_{\varphi_{rd}} + e_{i_{sq}} \dot{e}_{i_{sq}} + e_{i_{sd}} \dot{e}_{i_{sd}} + \frac{1}{\gamma} \tilde{\varphi}_{rd} \dot{\tilde{\varphi}}_{rd} \quad (15)$$

With:

$$\begin{aligned} \dot{\tilde{\varphi}}_{rd} &= \dot{\hat{\varphi}}_{rd} - \dot{\varphi}_{rd} \\ \dot{\hat{\varphi}}_{rd} &= \hat{\varphi}_{rd} - (a_4 i_{sd} - a_5 \varphi_{rd}) \end{aligned}$$

By replacing systems (11), (12) and (13) in (15) equation is obtained:

$$\begin{aligned} \dot{v} = & -k_{\omega} e_{\omega}^2 - k_{\psi_{rd}} e_{\psi_{rd}}^2 - k_{i_{sq}} e_{i_{sq}}^2 - k_{i_{sd}} e_{i_{sd}}^2 + \frac{1}{\gamma} \dot{\hat{\phi}}_{rd} + e_{i_{sq}} (a_6 \hat{\phi}_{rd} e_{\omega} + k_{i_{sq}} e_{i_{sq}} + \psi_{i_{sq}} - \lambda_{i_{sq}} - b v_{sq} + \dot{z}_{i_{sq}}) \\ & + e_{i_{sd}} (a_4 e_{\omega} + k_{i_{sd}} e_{i_{sd}} + \psi_{i_{sd}} - \lambda_{i_{sd}} - b v_{sd} + \dot{z}_{i_{sd}}) \\ & + \hat{\phi}_{rd} [-a_5 e_{\omega} - \frac{k_{\phi_{rd}}}{L_{m5}} e_{i_{sd}} + \frac{1}{\gamma} (\dot{\hat{\phi}}_{rd} - a_4 i_{sd} + a_5 \hat{\phi}_{rd})] \end{aligned}$$

For the condition that $\dot{v} < 0$, the control voltages v_{sd} and v_{sq} as follows:

$$\begin{aligned} v_{sd} &= \sigma L_s (k_{i_{sd}} e_{i_{sd}} + \psi_{i_{sd}} - \lambda_{i_{sd}} - b v_{sd} + \dot{z}_{i_{sd}}) \\ v_{sq} &= \sigma L_s (k_{i_{sq}} e_{i_{sq}} + \psi_{i_{sq}} - \lambda_{i_{sq}} - b v_{sq} + \dot{z}_{i_{sq}}) \end{aligned} \tag{16}$$

And the adaptation law of rotor flux $\hat{\phi}_{rd}$ given by:

$$\dot{\hat{\phi}}_{rd} = -\gamma (-a_5 e_{\omega} - \frac{k_{\phi_{rd}}}{L_{m5}} e_{i_{sd}}) + a_4 i_{sd} - a_5 \hat{\phi}_{rd} \tag{17}$$

4. ROTOR SPEED MRAS OBSERVER

4.1 Principle of MRAS observer

Currently there are many techniques for achieving speed estimator, but the most used is certainly the approach using an Model Reference Adaptive System (MRAS), this for its easier implementation and its good performance in terms of precision and stability [22].

The technique of MRAS based on the rotor flux for estimating the speed of the induction machine, originally developed by C. Schauder [23], is certainly the most popular strategy. The principle of the speed estimation by a MRAS is based on the comparison of rotor flux outputs of two estimators using different formulations. The first estimator is based on the stator equations defining a voltage model, called the reference model and the second, named adjustable or adaptive model is described by the rotor equations, it defines a current model.

The estimated components of the rotor flux, for both models, are based only on the measurement of stator quantities. We use the reference voltages and the measured currents usually expressed in a stationary reference frame related to the stator (α - β) [5]. The structure of the adaptive observer is shown in Figure 1.

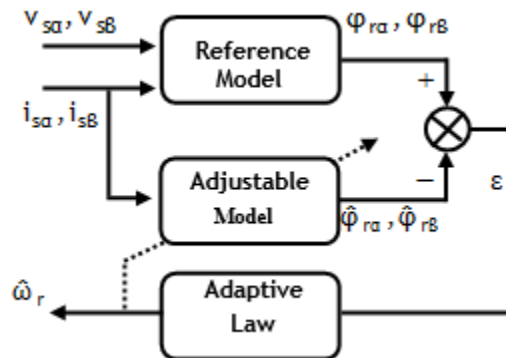


Figure 1. Block diagram of MRAS observer

4.2 Estimation of rotor speed by MRAS observer

The equations of the reference model are given by:

$$\begin{cases} \dot{\hat{\phi}}_{r\alpha} = \frac{L_r}{L_m} (v_{s\alpha} - R_s i_{s\alpha} - \sigma L_s \dot{i}_{s\alpha}) \\ \dot{\hat{\phi}}_{r\beta} = \frac{L_r}{L_m} (v_{s\beta} - R_s i_{s\beta} - \sigma L_s \dot{i}_{s\beta}) \end{cases} \quad (18)$$

The equations of the adjustable model are given by:

$$\begin{cases} \dot{\hat{\phi}}_{r\alpha} = -\frac{1}{T_r} \hat{\phi}_{r\alpha} - \omega_r \hat{\phi}_{r\beta} + \frac{L_m}{T_r} i_{s\alpha} \\ \dot{\hat{\phi}}_{r\beta} = -\frac{1}{T_r} \hat{\phi}_{r\beta} + \omega_r \hat{\phi}_{r\alpha} + \frac{L_m}{T_r} i_{s\beta} \end{cases} \quad (19)$$

With:

$$T_r = \frac{L_r}{R_r}$$

We notice that the rotor speed is only present in equations (19) and not in equations (18). The adaptation algorithm is chosen to make the adjustable model converge toward the reference model by minimizing the error. For this, the algorithm parameters are defined according to the criterion of hyperstability of Popov. [23]

The error between the states of the two models can be expressed in matrix form by:

$$\begin{bmatrix} \varepsilon_\alpha \\ \varepsilon_\beta \end{bmatrix} = \begin{bmatrix} \phi_{r\alpha} - \hat{\phi}_{r\alpha} \\ \phi_{r\beta} - \hat{\phi}_{r\beta} \end{bmatrix} \quad (20)$$

And from equations (18) and (19) its dynamics are given by:

$$\dot{\varepsilon} = A\varepsilon - W \quad (21)$$

With:

$$\dot{\varepsilon} = \begin{bmatrix} \dot{\varepsilon}_\alpha \\ \dot{\varepsilon}_\beta \end{bmatrix}, A = \begin{bmatrix} -\frac{1}{T_r} - \omega_r \\ \omega_r - \frac{1}{T_r} \end{bmatrix}, \varepsilon = \begin{bmatrix} \varepsilon_\alpha \\ \varepsilon_\beta \end{bmatrix}, W = (\omega_r - \hat{\omega}_r) \begin{bmatrix} -\hat{\phi}_{r\alpha} \\ \hat{\phi}_{r\beta} \end{bmatrix}$$

Schauder [24] offers an adaptation law that meets the criterion of Popov and given by equation:

$$\hat{\omega}_r = \chi_2(\varepsilon) + \int_0^t \chi_1(\varepsilon) d\tau \quad (22)$$

The criterion of Popov requires satisfaction of the following inequality [23]:

$$\int_0^t \varepsilon^T W dt \geq -\mu^2 \quad (23)$$

With μ is a positive constant.

Replacing ε and \hat{W} by their values, Equation (23) becomes:

$$\int_0^t \left\{ \left[\varepsilon_\alpha \hat{\varphi}_{r\beta} - \varepsilon_\beta \hat{\varphi}_{r\alpha} \right] \left[\omega_r - \chi_2(\varepsilon) - \int_0^t \chi_1(\varepsilon) dt \right] \right\} \geq -\mu^2 \tag{24}$$

We can solve the equation (24) using the following relationship:

$$\int_0^t k \left(\frac{d}{dt} f(t) \right) f(t) dt \geq -\frac{1}{2} k f(0)^2 \tag{25}$$

The use of the above expression provides the following functions:

$$\begin{aligned} \chi_1 &= k_i (\varepsilon_\beta \hat{\varphi}_{r\alpha} - \varepsilon_\alpha \hat{\varphi}_{r\beta}) = k_i (\varphi_{r\beta} \hat{\varphi}_{r\alpha} - \varphi_{r\alpha} \hat{\varphi}_{r\beta}) \\ \chi_2 &= k_p (\varepsilon_\beta \hat{\varphi}_{r\alpha} - \varepsilon_\alpha \hat{\varphi}_{r\beta}) = k_p (\varphi_{r\beta} \hat{\varphi}_{r\alpha} - \varphi_{r\alpha} \hat{\varphi}_{r\beta}) \end{aligned} \tag{26}$$

Replacing the equations (26) in equation (22) there was obtained:

$$\hat{\omega}_r = k_p (\varphi_{r\beta} \hat{\varphi}_{r\alpha} - \varphi_{r\alpha} \hat{\varphi}_{r\beta}) + k_i \int (\varphi_{r\beta} \hat{\varphi}_{r\alpha} - \varphi_{r\alpha} \hat{\varphi}_{r\beta}) dt \tag{27}$$

k_p and k_i are the PI controller gains of the adaptation mechanism.

The flux position and the Park angle is determined by:

$$\hat{\theta}_s = \arctg \left(\frac{\hat{\varphi}_{r\beta}}{\hat{\varphi}_{r\alpha}} \right) \tag{28}$$

The integral Backstepping control with the MRAS-Backstepping observer combination to estimate the rotor flux and the rotor speed for induction machine can be presented by the block diagram of Figure 2.

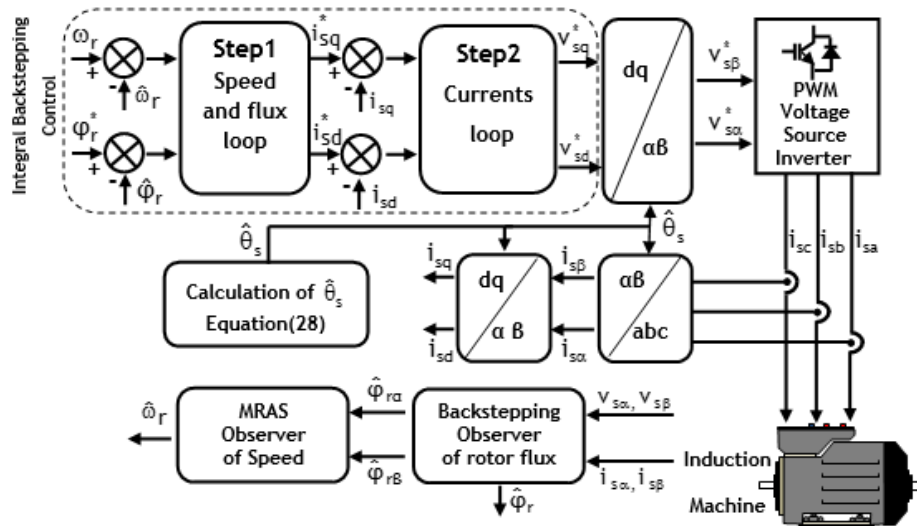


Figure 2. Block diagram of Sensorless integral Backstepping control scheme associated with MRAS-Backstepping observer

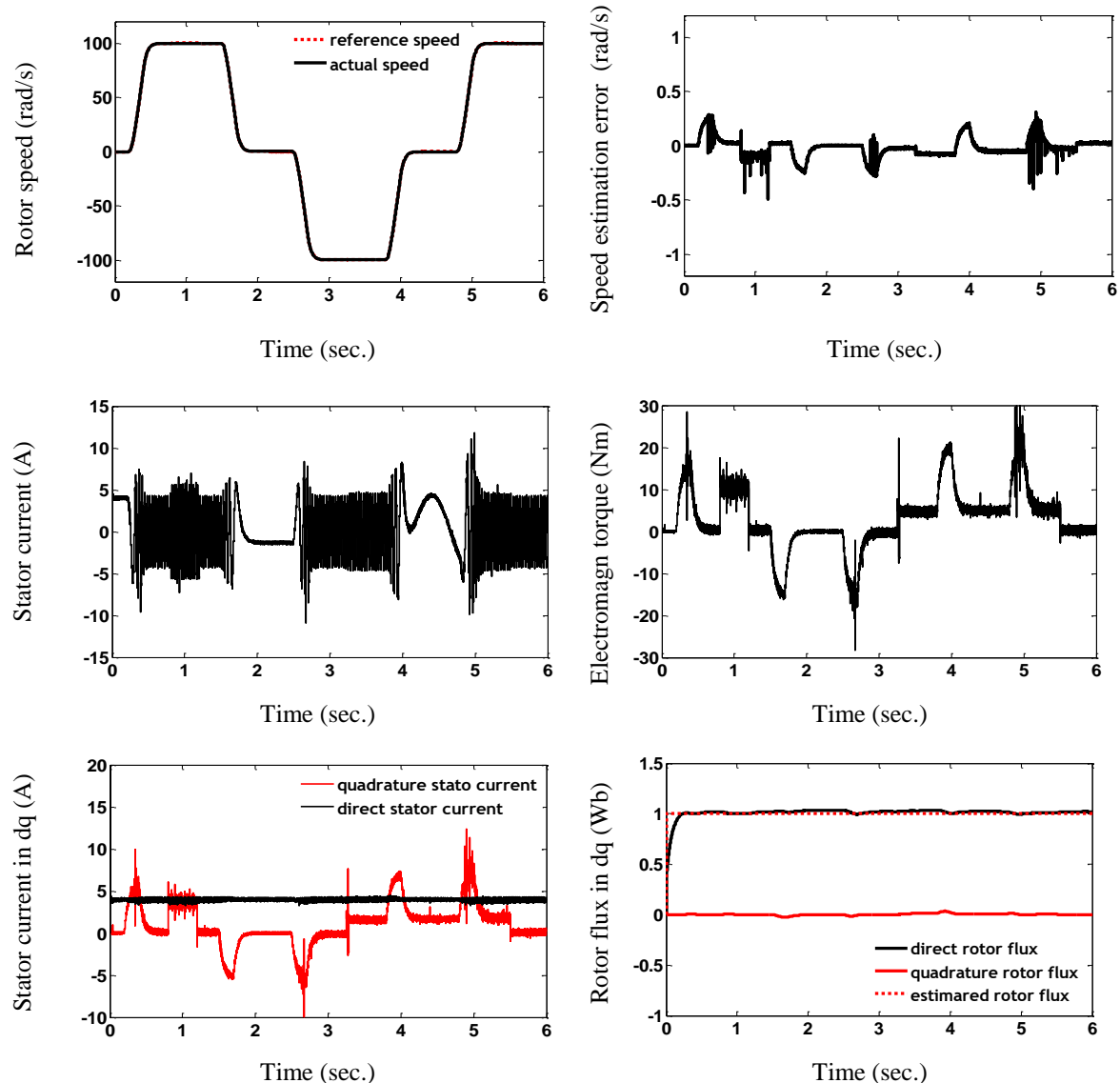
5. SIMULATION RESULTS AND INTERPRETATIONS

5.1 Simulation results

To evaluate the performance of the control and the two observers proposed in this study, two Benchmarks are proposed by [3]. The first Benchmark presents rapid transitions and operating areas at large and low speeds, the second Benchmark consists of low and very low speed operation and inversion. The objective is to impose reference trajectories taking into account the problems of observability of induction machine at very low speed. Recall that the sufficient condition of observability loss corresponds to having simultaneously constant mechanical speed and zero stator pulsation [25].

The results presented below were obtained by simulation tests conducted for sensorless integral Backstepping control scheme associated with the MRAS and Backstepping observer, on the profiles imposed by Benchmarks 1 and 2.

a) Benchmarks "1"



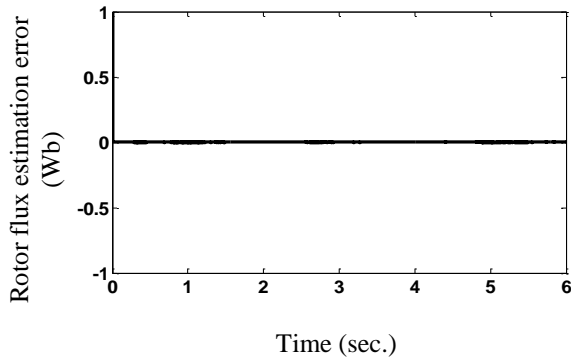
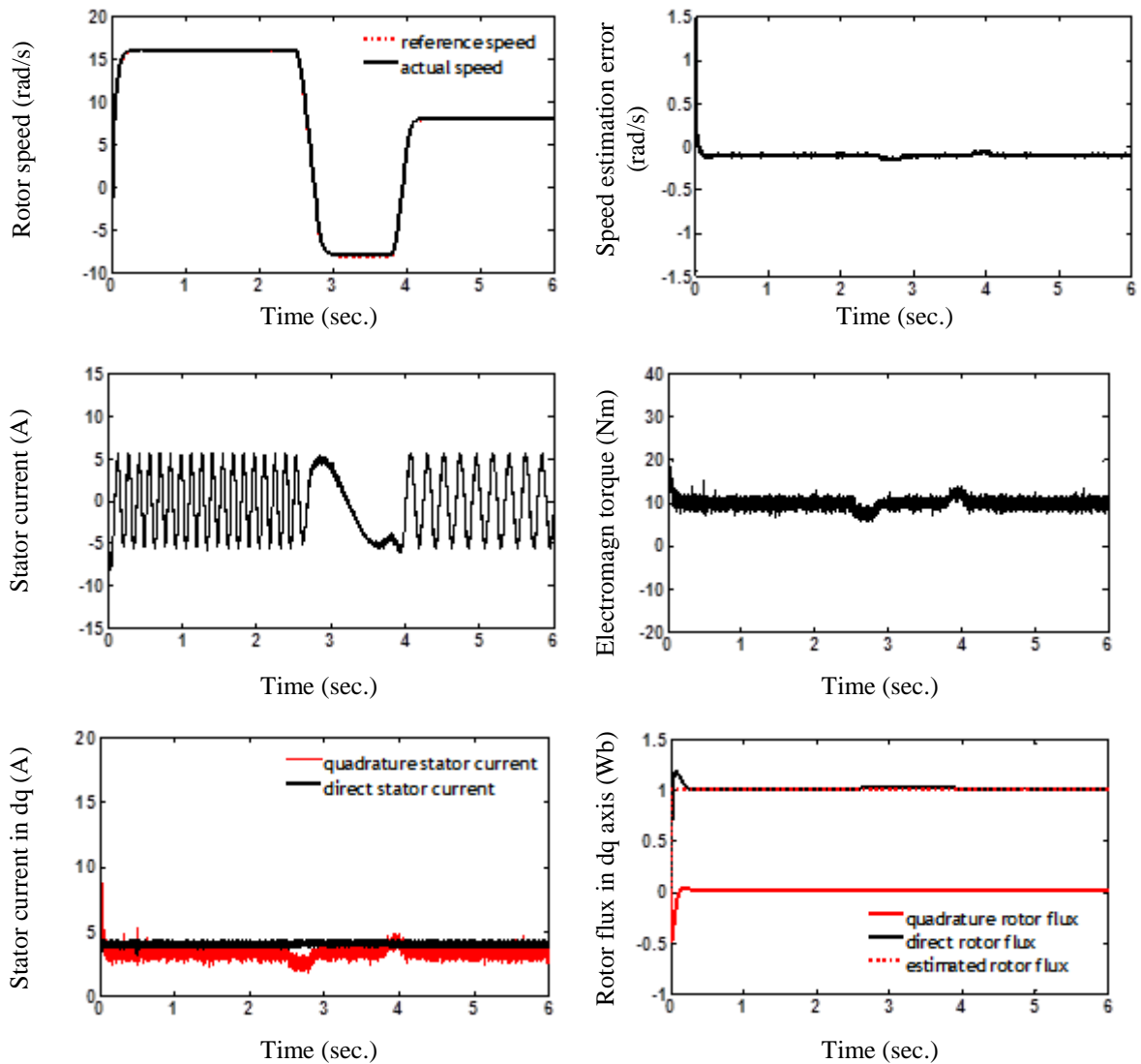


Figure 3. Performance in Benchmarks 1 of the proposed observer during starting operation with load change

b) Benchmarks "2"



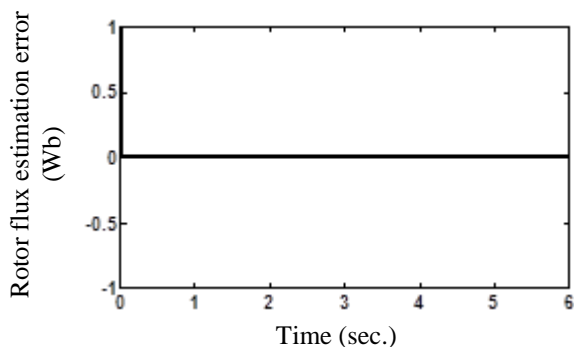


Figure 4. Performance in Benchmarks 2 of the proposed observer at low speed region

5.2 Interpretations

Benchmark 1 defines the reference trajectories for speed, rotor flux and load torque, as shown in the figure 3. At the beginning, reference speed is imposed zero, to enable the flux to reach its nominal value of 1Wb which is then kept constant. At $t=0,2$ s, the rotor speed is increased to 100 rad/s, according to an acceleration ramp, and remains constant up to 1,5 s. Then it is reduced, reaching a value of zero at 1,7 s, the machine is then maintained at standstill until 2,5 s. This first phase is used to test the behavior of the sensorless control for speed variations in quick transitions, and for a first critical area of operation, between 1,7s and 2,5s, at zero speed and without mechanical load. The application and removal of a torque load at the moments 0,8s and 1,2s will also assess the impact of this type of disturbance. Then, between 2,5 s and 4,8 s, a quasi-symmetric velocity profile is imposed in the opposite rotational direction, defining a second area of critical operating at a rate of -3,25 rad/s between 4 s and 4,8 s. Finally, the machine speed is again increased to 100 rad/s, to be there maintained up to 6 s. During the second critical phase, a torque load equal to 50% of the nominal value is applied at time 3,25 s and canceled at 5,5s, with the aim of analyzing the behavior of the training in an unobservable area, but this time in the presence of a mechanical load. The objective of Benchmark 2 is to evaluate the performance strategies of sensorless control at low and very low speed sensor in conditions of unobservable area of the machine and to highlight the influence of the load torque on the behavior of the control.

In figure 3, we notice that the estimated speed perfectly follows the machine speed with good behavior in terms of tracking and disturbance rejection. The tracking error is minimal never exceed 1 rad/s during transient. The estimated speed converges quickly and accurately to the actual machine speed and the estimation error is also small, but presents some oscillations between 1,25 rad/s to 1,40 rad/s, during motor start. When going from 100 to 0 rad/sec, then -100 to 0 rad/s, this error is 0,3 and 0,4 rad/s, while in the two critical areas the maximum values are 0,050 and -0,040 rad/s for steady state. The decoupling is adequately maintained. However, we observe small oscillations during transients and very low orientation errors in the two critical areas.

Figure 4 shows clearly that the estimated rotor speed follows with excellent precision the actual machine speed. The estimation error is very small at low and very low speed steady. When reversing the direction of rotation from -8 rad/s to 8 rad/s, the estimation shows some oscillations with a maximum error of 0,4 rad/s, which is acceptable.

6. CONCLUSION

In this paper, a new sensorless drive system approach is presented, coupled with a direct vector control for induction machine. The proposed method consists on the use of integral backstepping control with a nonlinear observer of rotor flux and adaptive observer for estimating rotor speed.

The results show excellent estimation under different operating conditions, especially at low speeds. In addition, it shows robustness against external disturbances and changes in machine parameters. This study shows that it is possible to realize an association of two observers for sensorless control based on a nonlinear approach with a high level of performance.

7. NOMENCLATURE

T_{em}, T_L : Electromagnetic and load torque.

i_{sd}, i_{sq} : Stator currents.

Φ_{rd}, Φ_{rq} : Rotor flux.

ω_m : Electrical angular.

σ : Total leakage factor.

R_s and R_r : Stator and rotor resistances.

L_s and L_r : Stator and rotor inductances.

L_m : Mutual inductance.

p : Number of pole pairs.

J : Moment of inertia.

f_c : Friction coefficient.

8. APPENDIX

Table 1: Nominal parameters of the induction machine used:

Rated power	1.5kW
Rated voltage	220V
Rated speed	1428rpm
Nominal frequency	50Hz
Rated stator current	3.64 A
Stator resistance	4.85 Ω
Rotor resistance	3.805 Ω
Cyclic stator inductance	0.274H
Cyclic rotor inductance	0.274H
Mutual inductance	0.258H
Number of pole pairs	2
Moment of inertia	0.031kg/m ²
Friction coefficient	0.00114Nm.s/rd

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