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Component- and System-Level Degradation Modeling of Digital Instrumentation and Control Systems Based on a Multi-State Physics Modeling Approach

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Abstract: A system-level degradation modeling is proposed for the reliability assessment of digital Instrumentation and Control (I&C) systems in Nuclear Power Plants (NPPs). At the component level, we focus on the reliability assessment of a Resistance Temperature Detector (RTD), which is an important digital I&C component used to guarantee the safe operation of NPPs. A Multi-State Physics Model (MSPM) is built to describe this component degradation progression towards failure and Monte Carlo (MC) simulation is used to estimate the probability of sojourn in any of the previously defined degradation states, by accounting for both stochastic and deterministic processes that affect the degradation progression.

The MC simulation relies on an integrated modeling of stochastic processes with deterministic aging of components that results to be fundamental for estimating the joint cumulative probability distribution of finding the component in any of the possible degradation states.

The results of the application of the proposed degradation model to a digital I&C system of literature are compared with the results obtained by a Markov Chain Model (MCM). The integrated stochastic-deterministic process here proposed to drive the MC simulation is viable to integrate component-level models into a system-level model that would consider inter-system or/and inter-component dependencies and uncertainties. **Keywords:** Multi-State Physics Modeling; Digital I&C System; System-Level Model; Component-Level Model; Degradation State Probability.

ABBREVIATIONS

PSA	Probabilistic Safety Analysis
NPP	Nuclear Power Plant
I&C	Instrumentation and Control
MSPM	Multi-State Physics Model(ing)
MCM	Markov Chain Model
MC	Monte Carlo
RTD	Resistance Temperature Detector
RPS	Reactor Protection System
BPL	Bistable Processor Logic
LCL	Local Coincidence Logic
PTS	Partial Tripping Signal
ESS	Emergency Shutdown Signal
RTB	Reactor Trip Breaker
CDF	Cumulative Distribution Function
PDF	Probability Density Function

NOTATIONS

γ_Y	Failure threshold
Y	Physical variable
G	Limit-state function
\bar{X}	Vector of system parameters
S	Safety domain
F	Failure domain
∂F	Failure boundary
t	Time
$\bar{\delta}$	Vector of physical parameters
$\bar{\Phi}_m$	Vector of m -dimensional manufacturing features
$\bar{\Theta}_k$	Vector of k -dimensional stochastic parameters
\bar{B}_l	Vector of l -dimensional external parameters
$\varepsilon(t)$	Error term
$\bar{P}(t, \bar{\delta})$	State probability vector obtained from MSPM
$p_j(t, \bar{\delta})$	State probability of state j in MSPM
$M^i + 1$	Number of states in i -th component/module-level MSPM
C_j^i	Degradation state j of component (module) i
$\lambda_{(j,k)}^i(t, \bar{\delta})$	Failure rate of component (module) i from state C_j^i to C_k^i
$\mu_{(j,k)}^i(t, \bar{\delta})$	Repair rate of component (module) i from state C_k^i to C_j^i
σ	RTD measurement accuracy
τ	RTD response time
δ	RTD air gap size
α_t	Scale factor
$P_S(t, \delta)$	CDF of the RTD New-to-drift failure mode
δ_0	Initial air gap size
dt	Time interval
$d\delta_t$	Noise of air gap size
t_m	Mission time
N_M	Simulation times
$p_S(t \delta)$	Conditional PDF given air gap size interval
$P_S(t \delta)$	Conditional CDF given air gap size interval
$\lambda_S(t \delta)$	Conditional failure rate given air gap size interval
$R_S(t \delta)$	Conditional reliability of RTD given air gap size interval
λ_S	Failure rate in RTD-MCM
μ_S	Repair rate in RTD-MCM

$P_S(t)$	Unreliability obtained from RTD-MCM
$N+1$	Number of layers in system-level MSPM
L^l	Layer l in system-level MSPM
M^l+1	Number of degradation states of layer l
L_m^l	Degradation state m of layer l
$\mu_{L_m^l \rightarrow L^0}(t, \bar{\delta})$	Repair rate from state L_m^l to state L^0
$\lambda_{L_m^l \rightarrow L_n^\omega}(t, \bar{\delta})$	Failure rate from state L_m^l to state L_n^ω
$\lambda_{L_m^l \rightarrow L^N}(t, \bar{\delta})$	Failure rate from state L_m^l to system failure state L^N
λ_B	BPL failure rate
λ_L	LCL failure rate
β	Common cause factor
λ_{BC}	BPL common cause failure rate
λ_{LC}	LCL common cause failure rate
λ_R	RTB failure rate
$P(t \delta)$	RPS unreliability obtained from RPS-MSPM
$P(t)$	RPS unreliability obtained from RPS-MCM

1. Introduction

In support to the implementation of risk-informed decision-making approaches, Probabilistic Safety Analysis (PSA) of modernizing Nuclear Power Plants (NPPs) demands for detailed dynamic models of digital Instrumentation and Control (I&C) systems that can adequately represent digital components failure modes and quantify their contribution to the overall risk of the NPPs (Aldemir et al., 2007; Aldemir et al., 2006).

To this aim, dynamic methods are being increasingly integrated into existing PSA frameworks for digital I&C systems reliability assessment, such as: Dynamic Flowgraph Methodology (DFM) (Guarro et al., 2012; Aldemir et al., 2006; Aldemir et al., 2009), Markov/cell-to-cell mapping technique (CCMT) (Aldemir et al., 2006; Aldemir et al., 2009; Zhou et al., 2014), Petri Net (Lee et al., 2006; Kim and Kim, 2014), Bayesian Networks (Boudali and Dugan, 2006; Broy et al., 2011), Dynamic Fault Tree (DFT) (Dehlinger and Dugan, 2008), Dynamic Event Tree (DET) (Bucci et al., 2006) and Fuzzy C-Means (FCM) clustering method (Di Maio et al., 2011; Zio and Di Maio, 2009). On a system level, these methods can be used to tackle the twofold purpose of PSA: on one side, the identification of the system failure domain and, on the other side,

the quantification of the system failure probability.

With respect to the latter, given a failure threshold γ_Y not to be exceeded by a safety-relevant physical variable Y during the system operation, a limit-state function G can be defined as:

$$G = G(\bar{X}, \gamma_Y) = Y(\bar{X}) - \gamma_Y \quad (1)$$

where $\bar{X} = \{X_1, X_2, \dots, X_n\}$ defines the system parameters and operational conditions. This leads to the definition of a system safety domain $S = \{\bar{X}: G(\bar{X}, \gamma_Y) < 0\}$ and of a system failure domain $F = \{\bar{X}: G(\bar{X}, \gamma_Y) > 0\}$, that are partitioned by a system failure boundary $\partial F = G(\bar{X}, \gamma_Y) = 0$, for a given γ_Y .

The identification of the failure domain F is crucial especially when the system dynamics is complex and its component reliability assessment cannot be described by a Boolean, discrete and abrupt physics of failure, but rather by a multi-valued, and continuous degradation model as it is for digital I&C systems (Li et al., 2012; Lin et al., 2015; Lisnianski and Levitin, 2003). The biggest challenge to be overcome for devising realistic and effective degradation models consists in the collection of component reliability data that are, often, affected by multiple and competing failure modes that are difficult to be untangled and reduced to a single-lumped failure criterion analysis that would leverage the degradation modeling task. To avoid simplification and overlooking of failure interdependencies, we propose to resort to a Multi-State Physics Modeling (MSPM) approach at the component level, which can be easily upscaled for system-level degradation modeling. The MSPM approach is based on the structure of Markov (or semi-Markov) modeling for the quantification of components reliability measures (Unwin et al., 2011; Unwin et al., 2012; Rocco and Zio, 2013; Fleming et al., 2010). Recently, the MSPM approach has been proposed for modeling nuclear component degradation by accounting for both the effects of stochastic parameters affecting the degradation and the environmental parameters with their uncertainties (Lin et al., 2015; Di Maio et al., 2015).

In this study, a component-level MSPM model for a digital I&C system is developed by integrating in the model both the stochastic and the deterministic processes that affect component degradation. The physical variable Y to be considered for the failure domain F identification is given in Eq. (2) (Kaiser and Gebraeel, 2009):

$$Y = Y(\bar{X}) = f(t, \bar{\delta}) + \varepsilon(t) = f(t, \bar{\Phi}_m, \bar{\Theta}_k, \bar{B}_l) + \varepsilon(t) \quad (2)$$

where t is the deterministic aging time, $\bar{\delta}$ is a collection of physical parameters affecting the degradation process that can be seen as composed by $\bar{\Phi}_m = \{\varphi_1, \dots, \varphi_m\}$ which is a vector of m -dimensional manufacturing features that affect the degradation (e.g., burn-in, contamination, etc.), $\bar{\Theta}_k = \{\theta_1, \dots, \theta_k\}$ which is a vector of k -dimensional stochastic parameters that account for the components variability (e.g., nominal frequency stability, calibration error after maintenance, etc.), $\bar{B}_l = \{\beta_1, \dots, \beta_l\}$ which is a vector of l -dimensional external parameters that capture the variability of time-varying operating and environmental conditions (e.g., temperature, flux, etc.), and $\varepsilon(t)$ that is an error term that captures noise and disturbances. In principle, a component response surface to any possible different setting of degradation features (stochastic and external parameters, and error terms) can be built (with infinite computational resources) such that the safety domain S can be partitioned from the failure domain F by setting a failure threshold γ_Y .

In this work, a Monte Carlo (MC) simulation is used to estimate the transition probabilities among the degradation states of MSPM and drive, by random walks, the stochastic process of the evolution of the air gap size in time and the deterministic evolution of the component aging on the response surface for the identification of the limit surface of the drift event of a Resistance Temperature Detector (RTD) that is embedded into a digital I&C system of a NPP.

Finally, as for traditional PSA (where system-level models are developed by combining or replacing subsystem or component models in the overall structure of a Fault Tree (FT) or an Event Tree (ET) (Aldemir et al., 2007; Aldemir et al., 2009; Gulati and Dugan, 1997)), the system failure probability of the digital I&C system is quantified by upscaling the component-level MSPM into a system-level model that considers the inter-system or/and inter-component dependencies and the aleatory or/and epistemic uncertainties affecting each component behavior.

The rest of paper is as follows. Section 2 presents the illustration of component-level MSPM for a digital I&C component (e.g., sensor, reactor trip breaker, CPU, etc.)

and the comparison of the results obtained by the Markov Chain method. Section 3 presents the system-level MSPM framework of a digital I&C system in a NPP. In section 4 conclusions are drawn.

2. The Component-Level MSPMs

In this study, the digital I&C System of reference, whose reliability assessment is required for the failure domain identification and the related risk quantification of a NPP, is a typical NPP Reactor Protection System (RPS). Its objective is to trigger reactor emergency shutdown as soon as an anomaly is detected in the measurements of some relevant signals (i.e., primary coolant pressure, temperature, etc.). As shown in Fig. 1, this digital I&C system is composed of two redundant channels (A and B). Each channel consists of one signal sensor (S-A and S-B), one Bistable Processor Logic (BPL) subsystem (BPL-A and BPL-B), and one Local Coincidence Logic (LCL) subsystem (LCL-A and LCL-B).

An independent signal sensor and an independent BPL is installed in each channel. If any of the two redundant measured signals exceeds a safety threshold value, a Partial Tripping Signal (PTS) is measured from the corresponding BPL, e.g. a PTS from BPL-A is measured because S-A exceeds the tripping value. The signal processing proceeds only if both channels produce the PTS: each PTS from a BPL is sent to both LCL-A and LCL-B, which process information by an “AND” gate. In other words, an Emergency Shutdown Signal (ESS) is produced only when receiving two PTSs from different BPLs; ESSs, then, activates the Reactor Trip Breaker (RTB), when at least one ESS is triggered, i.e., the information is processed by an “OR” gate. Once the RTB is activated, the power supply system and Control Rod Drive Mechanism (CRDM) which are connected with the RTB come into use to control the power of the reactor.

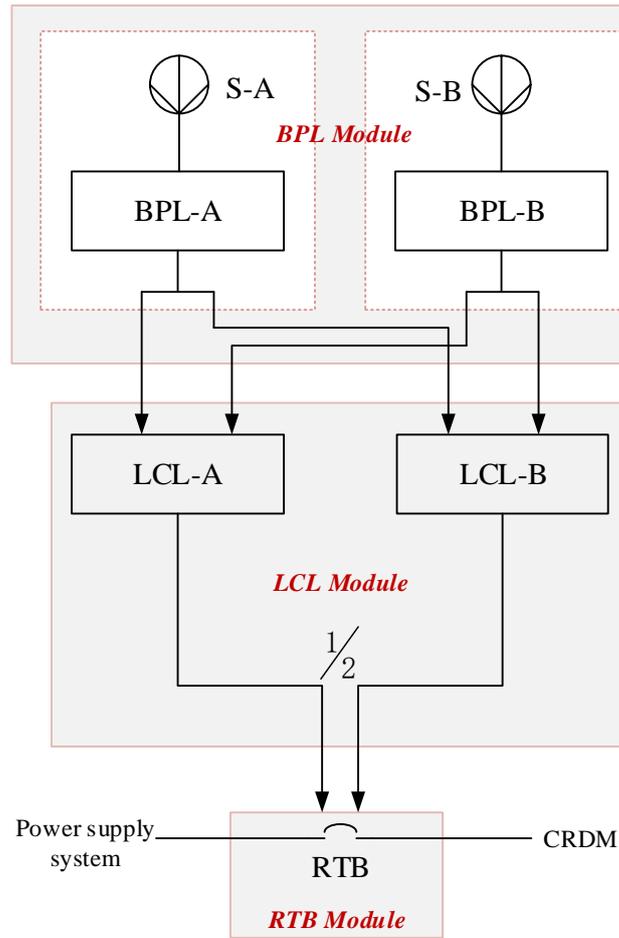


Fig. 1 A typical RPS (Wang et al., 2015)

The system can be decomposed into modules to reduce the complexity of system-level modeling based on the functions the embedded components are devised for and the failure effects they produce on the system. According to the RPS scheme of Fig. 1, the RPS modules are identified:

- The BPL Module, consists of two groups of components: sensor and BPL (i.e., “S-A and BPL-A” and “S-B and BPL-B”); these components are connected in series and their failure effects on the system can be combined.
- The LCL Module consists of the two LCLs (i.e., LCL-A and LCL-B). Since the ESS is triggered only when both LCLs simultaneously receive two PTSs from the two BPLs, this module is highly dependent of the BPL module.
- The RTB Module.

Without loss of generality, let us consider and build the component-level MSPM of the module “S-A and BPL-A”. It is worth mentioning that, for a comprehensive and

exhaustive analysis, a component-level MSPM of each module shown in Fig. 1 should be built with great detail and embedded into the system-level MSPM, as we shall see in Section 3.

Each component-level model is a MSPM where the dynamics of component degradation is represented by transitions among a finite number M of degradation states at any time instant t and any value of the affecting parameters $\bar{\delta}$ (Li et al., 2012; Di Maio et al., 2015); similarly to a Markov Chain Model (MCM), a state probability p is assigned to each degradation state, that is collectively represented by a state probability vector $\bar{P}(t, \bar{\delta}) = \{p_0(t, \bar{\delta}), p_1(t, \bar{\delta}), \dots, p_j(t, \bar{\delta}), \dots, p_M(t, \bar{\delta})\}$ for all the M states.

- In general terms, we model each i -th component with a graph with M^i+1 nodes, each of which identifies a state of the component degradation progression. Herein, “ C_0^i ” and “ C_M^i ” represent the “New” and “Failed” states of the i -th component in the system, whereas any other state “ C_j^i ”, $j = 1, 2, \dots, M^i$ is an intermediate degradation state, where the component is partially functioning.
- State “ C_0^i ” is selected as the initial state at time $t=0$, i.e., the component is “New” at $t=0$.
- $\lambda_{(j,k)}^i(t, \bar{\delta})$, $j = 1, 2, \dots, M^i, k = 1, 2, \dots, M^i$, and $j \neq k$ is the transition rate of the i -th component degradation model from state “ C_j^i ” to state “ C_k^i ”, with respect to the time and the degradation affecting factor vector $\bar{\delta}$.
- Components can be repaired even if not “Failed”, i.e., repair rate $\mu_{(j,k)}^i(t, \bar{\delta})$, $j = 1, 2, \dots, M^i, k = 1, 2, \dots, M^i$, and $j \neq k$ can be foreseen between the intermediate transition states “ C_j^i ” and “ C_k^i ”.

An example of MSPM of a generic single component i is sketched in Fig. 2, below.

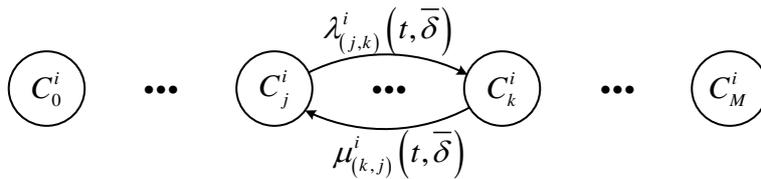


Fig. 2 MSPM degradation model of single component i

2.1 The Sensor MSPM

Different types of sensors are employed for measuring temperatures, pressures, levels and flows in NPP safety systems, such as Resistance Temperature Detectors (RTDs) (Montalvo et al., 2014; Hashemian, 2011; Yun et al., 2012), pressure transmitters (Hashemian, 2011), eddy current sensors (García-Martín et al., 2011; Uchanin and Najda, 2011) and optical fiber sensors (Ferdinand et al., 2013). Without loss of generality, we assume that the S-A and S-B of the digital I&C system of Fig. 1 are RTDs. We focus on RTDs because they are critical components, whose effectiveness in promptly detecting anomalous temperature changes greatly support plant operators in the monitoring of NPP operational conditions and guide counteracting measure to avoid system failure. This is why RTDs must properly generate accurate and timely data (Baraldi et al., 2015). The NPP plant power level is, indeed, set based on the information gathered from RTDs: the better the performance of these process instrumentations in terms of measurement accuracy, the larger the power rate with enough margin from the system failure domain F (hence, the better the plant economics) (Yun et al., 2012).

To build the RTD-MSPM model (as sketched in Fig. 3), the sensor failure modes have to be identified (e.g., bias (Uren et al., 2015), drift (Uren et al., 2015; Garvey and Hines, 2006), performance degradation (Fernandez et al., 2015), freezing (Boskovic and Mehra, 2002), and calibration error (Castello et al., 2014)) and assumptions are made for the subsequent quantitative analysis:

- Six degradation states, besides the sensor functioning state (C_0^{RTD}), are set and organized into RTD-MSPM model: C_1^{RTD} corresponds to Bias, C_2^{RTD} corresponds to Drift, C_3^{RTD} corresponds to (performance) Degradation, C_4^{RTD} corresponds to Freezing, C_5^{RTD} corresponds to Calibration Error and C_6^{RTD} corresponds to Complete Failure. Transitions can occur between state C_0^{RTD} and any of the other degradation states, with transition rate $\lambda_{(0,j)}^{RTD}(t, \bar{\delta})$ (here j

= 1, 2, 3, 4, 5, 6) and between degradation states with transition rate $\lambda_{(j,k)}^{RTD}(t, \bar{\delta})$ (here $j, k = 1, 2, 3, 4, 5, 6$ and $j \neq k$). It is assumed that repair is just carried out from each degradation state to state C_0^{RTD} , with repair rate $\mu_{(j,0)}^{RTD}(t, \bar{\delta})$.

- The initial state at time $t=0$ is sensor functioning state C_0^{RTD} .

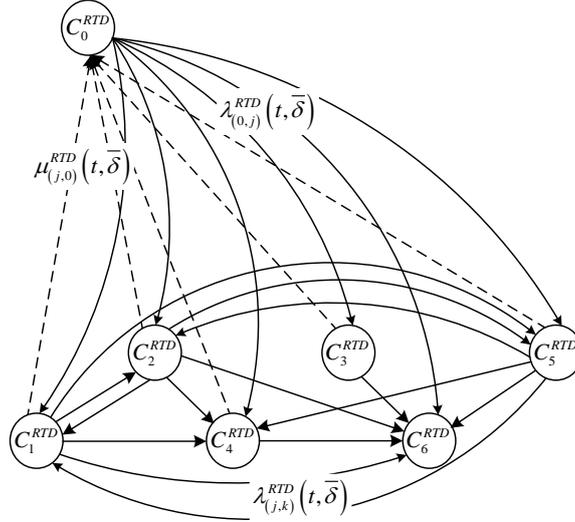


Fig. 3 Schematic diagram of the RTD-MSPM

The RTD-MSPM is further simplified as in Fig. 4, where C_0^{RTD} is the RTD functioning state and C_2^{RTD} is the RTD failure degradation state due to drift with failure rate $\lambda_{(0,2)}^{RTD}(t, \bar{\delta})$ and repair rate $\mu_{(2,0)}^{RTD}(t, \bar{\delta})$. This simplification is due to the experimental evidence that the main failure mode to be considered for RTDs is drift (Balaban et al., 2009). Overtime exposure to high temperatures can cause a drift in the measurements up to several degrees per year; shock and vibration generate strain in resistive wires of the RTDs and change their characteristics. In what follows, the quantitative analysis on the RTD-MSPM is done.

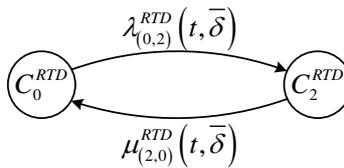


Fig. 4 Drift case of RTD-MSPM model

2.2 Estimation of the RTD-MSPM Parameters

RTDs contain a sensing element whose resistance changes with fluid temperature. Well-type RTDs are assembled into thermowells containing the sensing elements, sheaths and insulation materials, whereas wet-type RTDs are designed for direct immersion into the fluid (Yun et al., 2012). Two types of RTDs are shown in Fig. 5 (Hashemian, 2011).

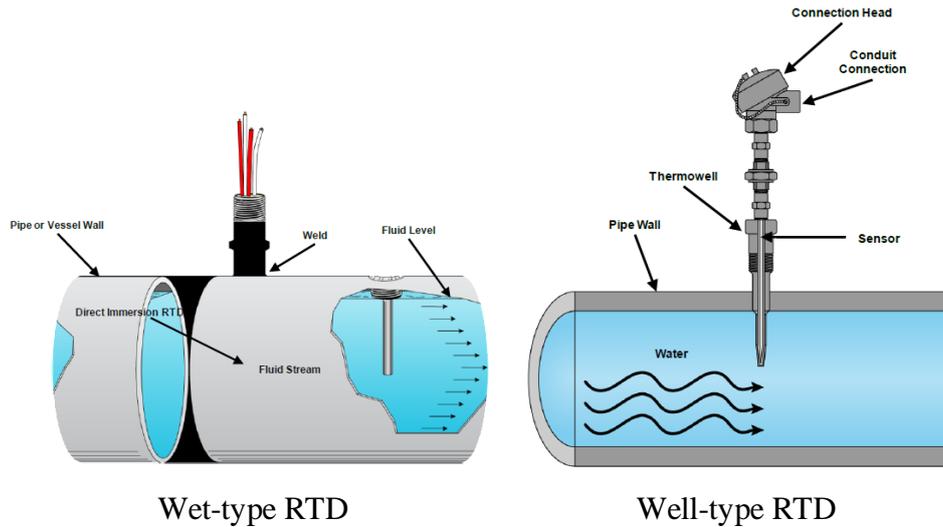


Fig. 5 Two types of RTDs (Hashemian, 2011)

The performance of RTDs is characterized by their measurement accuracy σ and response time τ . The former characterizes the RTD accuracy in measuring a sudden temperature change (Wei et al., 2013), whereas the latter is a pivotal indicator that measures how quickly the RTD responds to a sudden and significant temperature change (Yun et al., 2012). On one hand, being τ the time the RTD needs for reaching 63.2% of a sudden temperature change, τ is also representative of σ ; on the other hand, since some failure modes might affect the fluid temperature and some others such as bias, freezing, loss of signal showing constant measured values even if the RTD is degraded and, thus, τ cannot be considered as performance metric, both τ and σ are to be simultaneously considered for defining the RTD failure domain.

Under normal reactor operation conditions, effects of intrinsic properties (intrinsic shape and material properties, for example) are negligible on its own performance because they cannot be altered once the RTD is manufactured, such as τ and ε (Hashemian, 1994). Instead, component aging t and uncertainty of air gap size δ

between the bottom of the thermowell and the RTD sensing tip due to contamination, metallurgical changes, moisture or dirt entering, mechanical shock, etc. are more likely to cause RTD drift (Hashemian, 2011; Swanson, 2007).

Aging can affect RTD performance with different degradation modes; for example, the sensing element resistance increases under tensile stress and decreases with compression stress that varies with time t , resulting in off-calibration, increase in τ , reduced insulation resistance, erratic output, wiring problems, etc. Among these, sensor off-calibration and τ changes are the most relevant features to be monitored (Hashemian, 2011).

Moreover, we assume τ to heavily depend on air gap size δ between the RTD sensing tip and the bottom of the thermowell (Hashemian, 2013), even though also debris, dirt, and metal shavings entering the thermowell during installation, and/or moisture entering the insulation material during operation can, also cause an off-design of air gap that prevents the RTD from reaching the very bottom of its thermowell. Moreover, RTD movement in the thermowell due to vibration, thermal, or mechanical shock can cause the RTD sensing tip to displace away from the bottom of the thermowell (Hashemian, 2011) and calibration drift (Yun et al., 2012).

In conclusion, in what follows, τ is identified as the RTD drift physical variable Y being degrading due to t and δ , as in Eq. (3):

$$Y = \tau(t, \delta) \quad (3)$$

Given a RTD failure threshold γ_Y not to be exceeded by $\tau(t, \delta)$ during operation, then, its limit-state function can be formulated as:

$$G = Y - \gamma_Y = \tau(t, \delta) - \gamma_Y \quad (4)$$

To partition the RTD safety domain S from its failure domain F , we firstly build a physical mathematical relationship between τ , t and δ based on the experimental data listed in Table 1 and 2 (Hashemian, 2011; Yun et al., 2012).

Table 1 Experimental data for τ at fixed t and $\delta=0$ (Yun et al., 2012)

Aging Time t [yr]	0	2	4	5	6
Response Time τ [s]	2.1	4.4	4.8	5.0	5.2
Variance	1.67	0.77	0.72	0.77	0.67

Table 2 Experimental data for τ at $t=0$ and fixed δ (Hashemian, 2011; Yun et al., 2012)

Air gap size δ [mm]	0	0	0.2	0.4	0.5	0.6	0.8	1.0	1.0	1.5	2.0
Response time t [s]	0.9	3.3	4.1	5.0	2.94	5.9	6.5	3.33	7.5	3.48	3.58

Fig. 6 shows the trend of τ as long as t increases, independently from the air gap contamination (i.e., $\delta = 0$) when data in Table 1 are used as interpolation data. Fig. 7 shows the trend of τ as a function of δ , when the RTD is new and data in Table 2 are used as interpolation data (i.e., aging $t=0$) (Yun et al., 2012).

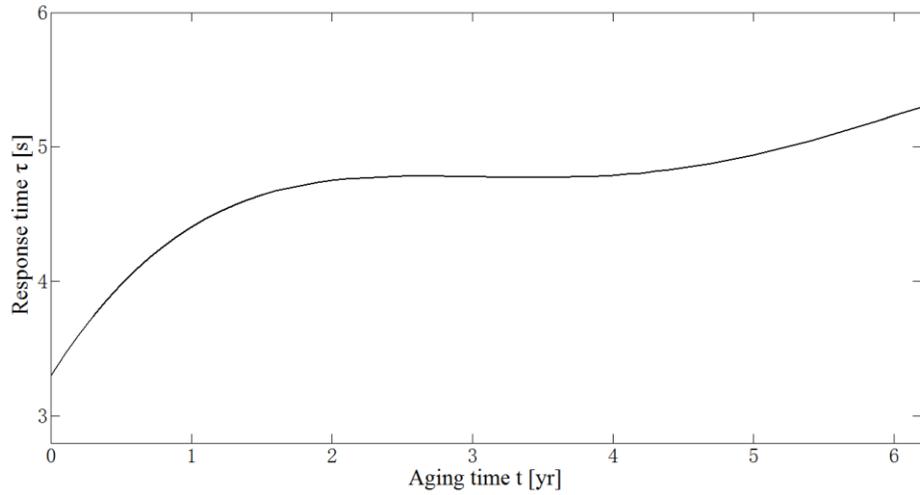


Fig. 6 $\tau(t, 0)$

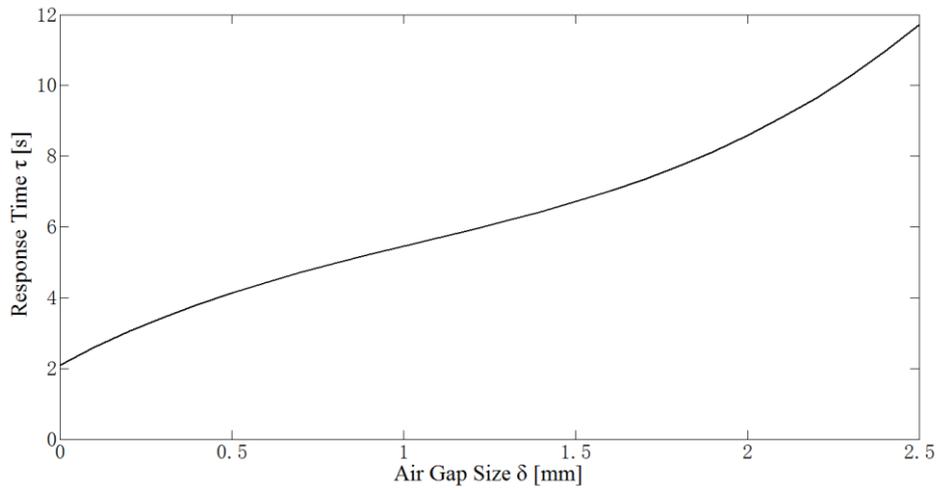


Fig. 7 $\tau(0, \delta)$

The function $\tau(t, \delta)$ as it is plotted in Fig. 8 is obtained by resorting to Eq. (5):

$$\tau(t, \delta) = \alpha_t \cdot \tau(t - 1, \delta) \quad (5)$$

where $\tau(0, \delta)$ is the curve of Fig. 7 and the factor α_t accounts for the changes of response time τ with the increase of t , by scaling the $\tau(t, 0)$ using the scale factor α_t :

$$\alpha_t = \frac{\tau(t)}{\tau(t-1)} \quad (6)$$

where, $\alpha_1 = \tau(1)/\tau(0)$. Table 3 reports the estimate of α_t for six discrete aging times t .

Table 3 Estimate of scale factors at six discrete aging times

Aging time t [yr]	0	1	2	3	4	5	6
Response time τ [s]	2.1040	3.5750	4.3818	4.7367	4.8525	4.9415	5.2163
Weighting factor α_t	/	1.70	1.23	1.08	1.02	1.02	1.06

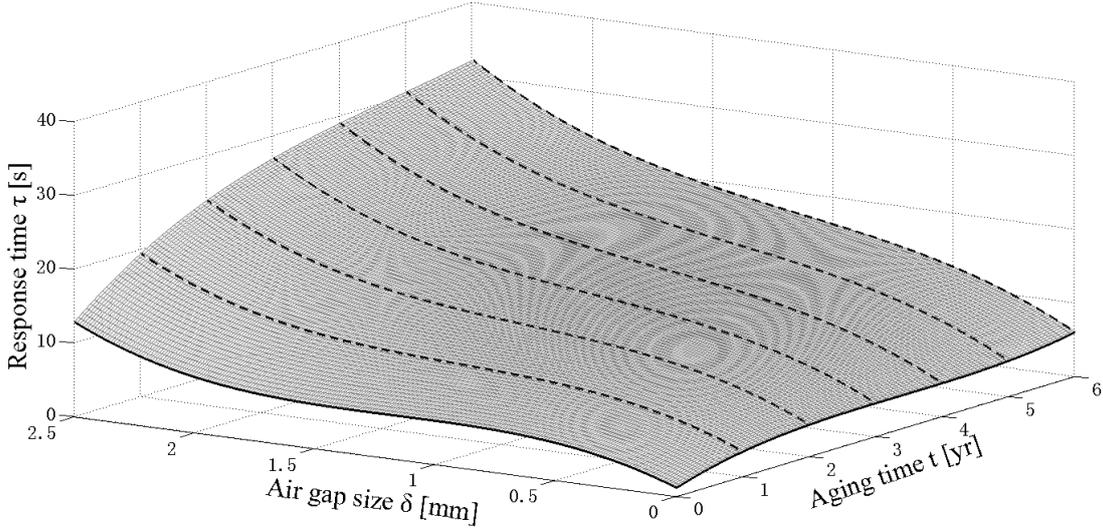


Fig. 8 $\tau(t, \delta)$

As mentioned in (Hashemian, 2011), the τ of a well-type RTD usually ranges in [4s, 8s]; hence, the RTD failure threshold γ_Y is here set equal to 8s. The Cumulative Distribution Function (CDF) $P_S(t, \delta)$ of the RTD new-to-drift-failure mode that can account for the stochasticity of the process and of the uncertainties affecting the degradation (for example, the initial air gap size δ_0 and the noise affecting the air gap size δ_t due to the vibration) can be found by running N_M Monte Carlo simulations, as follows (see Table 4 for the list of parameters):

- For each trial, at the initial time $t=0$, we sample the value of δ_0 from the uniform distribution $U(0,1)$ as initial air gap size.
- At each t that increases with the time step dt , the value $d\delta_t$ is sampled from a normal distribution $N(0,0.025t)$; thus, $\delta = \delta_0 + d\delta_t$.
- At each t within the mission time $[t_0, t_m]$, τ is estimated using the curve $\tau(t, \delta)$ of Fig. 8. If the value of τ exceeds the threshold γ_Y , the RTD is assumed to

fail at time t with air gap size δ .

Table 4 Setting of Parameters for Monte Carlo Simulation

Parameter	Description	Unit	Value	Distribution
N_M	Simulation Times	/	10^4	/
γ_Y	Response Time Threshold	s	8	/
dt	Time interval	yr	1/100	
t_0	Initial time	yr	0	/
t_m	Mission time	yr	6	/
δ_0	Initial Air Gap Size	mm	/	U(0, 1)
$d\delta_t$	Noise of Air Gap Size	mm	/	N(0, 0.025t)

Pictorially, we can show the evolution of $\tau(t, \delta)$ for each trial, as sketched in Fig. 9: for a sampled δ_0 (equal to 0.12mm), the air gap size oscillates during the RTD life around δ_0 (see in Fig. 9(a)); on the other hand, the response time τ , stochastically changes with the increase of time t (in Fig. 9(b)) and, thus, the transition between states C_0^{RTD} and C_2^{RTD} (see Fig. 4) of drift failure mode is determined when the response time τ reaches the failure threshold γ_Y , as shown in Fig. 9(c).

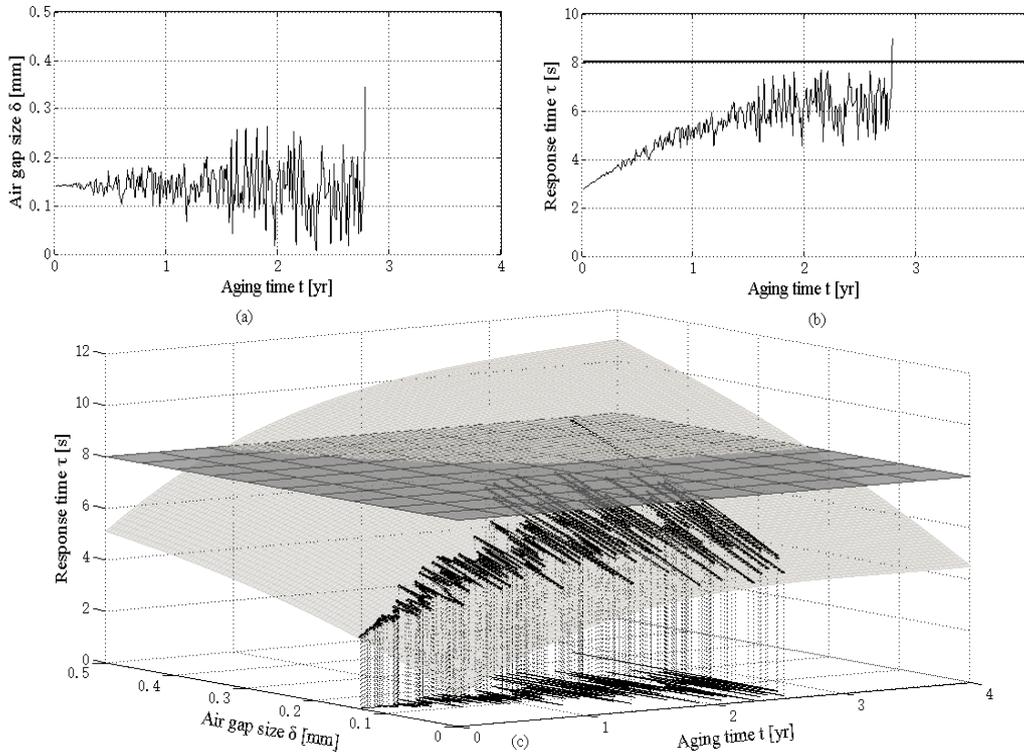


Fig. 9 One trial of MC simulation: (a) the stochastic path of air gap size δ changing with the aging t ; (b) the evolution of response time τ with the aging t ; (c) the simulated path response time τ with respect to δ and t on the safety domain of the fitting curved surface

After the N_M trials of MC simulations have been run, the conditional Probability Density Function (PDF) $p_S(t|\delta)$ and conditional Cumulative Distribution Function (CDF) $P_S(t|\delta)$ of the RTD New-to-drift transition of Fig. 4 can be empirically built (shown in Fig. 10 and 11, respectively) and used to calculate the conditional failure rate $\lambda_S(t|\delta)$:

$$F_S(t < T < t + \Delta t | (T > t, \delta)) = \frac{F_S(t < T < t + \Delta t | \delta)}{F_S(T > t | \delta)} = \frac{F_S(T < t + \Delta t | \delta) - F_S(T < t | \delta)}{1 - F_S(T < t | \delta)} = \frac{P_S(t + \Delta t | \delta) - P_S(t | \delta)}{1 - P_S(t | \delta)} \quad (7)$$

$$\lambda_S(t | \delta) = \lim_{\Delta t \rightarrow 0} \frac{F_S(t < T < t + \Delta t | (T > t, \delta))}{\Delta t} = \lim_{\Delta t \rightarrow 0} \frac{P_S(t + \Delta t | \delta) - P_S(t | \delta)}{\Delta t} \cdot \frac{1}{1 - P_S(t | \delta)} = \frac{p_S(t | \delta)}{R_S(t | \delta)} \quad (8)$$

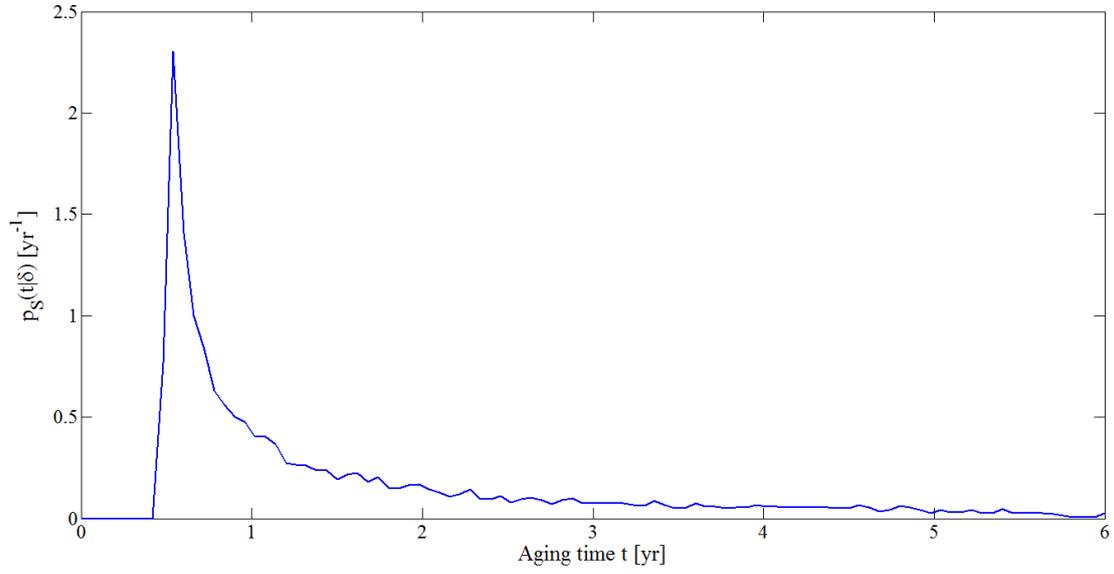


Fig. 10 Conditional probability density function of RTD New-to-drift failure mode

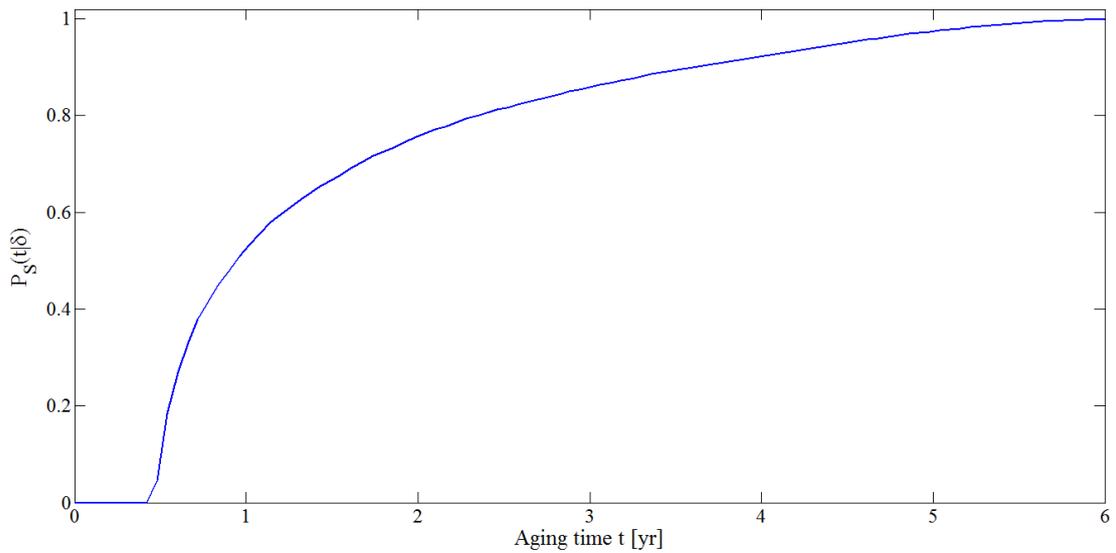


Fig. 11 Conditional cumulative distribution function of RTD New-to-drift failure mode

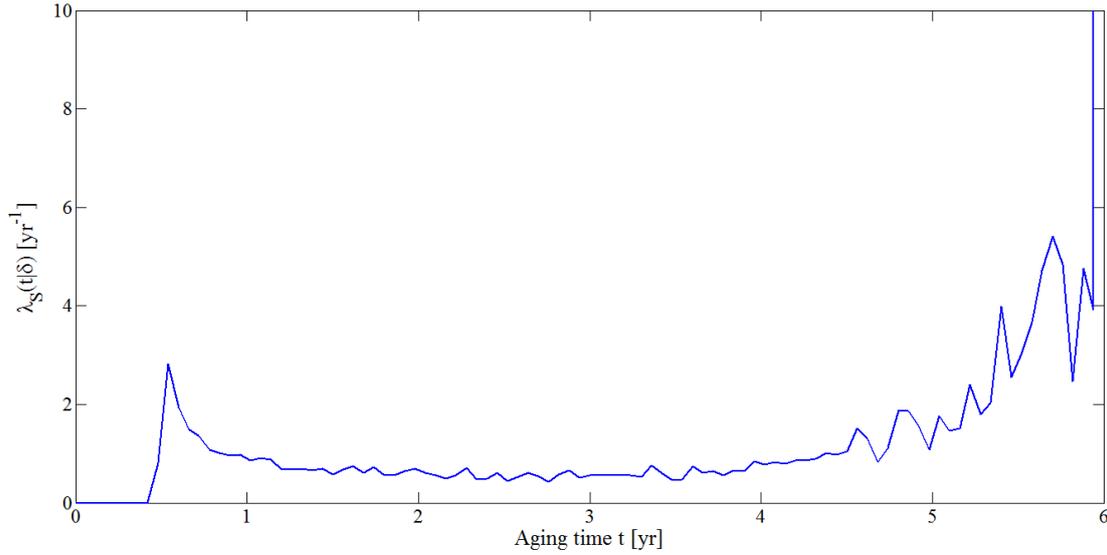


Fig. 12 Conditional failure rate of RTD New-to-drift failure mode

It is worth mentioning that the conditional CDF $P_S(t|\delta)$ of Fig. 11 describing the uncertain timing of RTD New-to-drift failure mode shows a sharp increase in $[0.5, 1.5]$ yr, after which it starts to level off to reach $P_S(t|\delta)$ at 5.8yr. Therefore, the failure rate $\lambda_S(t|\delta)$ of Fig. 12 shows the typical infant mortality and wear out periods, and tends to be constant in the useful life, which coincides with a general bath-tub curve, but with non-constant values along life.

2.3 Comparison with Markov Chain Model

The results shown in Section 2.2 are compared with those of a MCM to describe a component failure in a binary-state modeling framework and neglecting any physical modeling of component degradation features. The equivalent of the RTD-MSPM of Fig. 4 is shown in the RTD-MCM of Fig. 13, where states “0” and “1” represent no-failure state and failure state respectively, and the transition rates between them λ_S and μ_S are constant values (λ_S is equal to $1E-4/hr$ and μ_S to 0 (US: EPRI, 2008)). Therefore, the RTD unreliability (i.e., failure state probability) $P_S(t)$ can be calculated by means of MCM quantitative analysis and is plotted in Fig. 14.

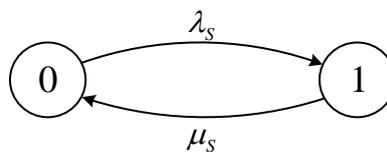


Fig. 13 Schematic diagram of RTD-MCM (without repair)

Fig. 14 shows the results obtained from MSPM compared with those from MCM. The $P_S(t|\delta)$ (obtained by MSPM) is close to the $P_S(t)$ (obtained by MCM), which, however, overestimates the unreliability of the RTD. The degradation process modeling in the MSPM is more realistic than in MCM, and the latter is not able to guarantee the drift onset beyond a threshold value (especially at low aging).

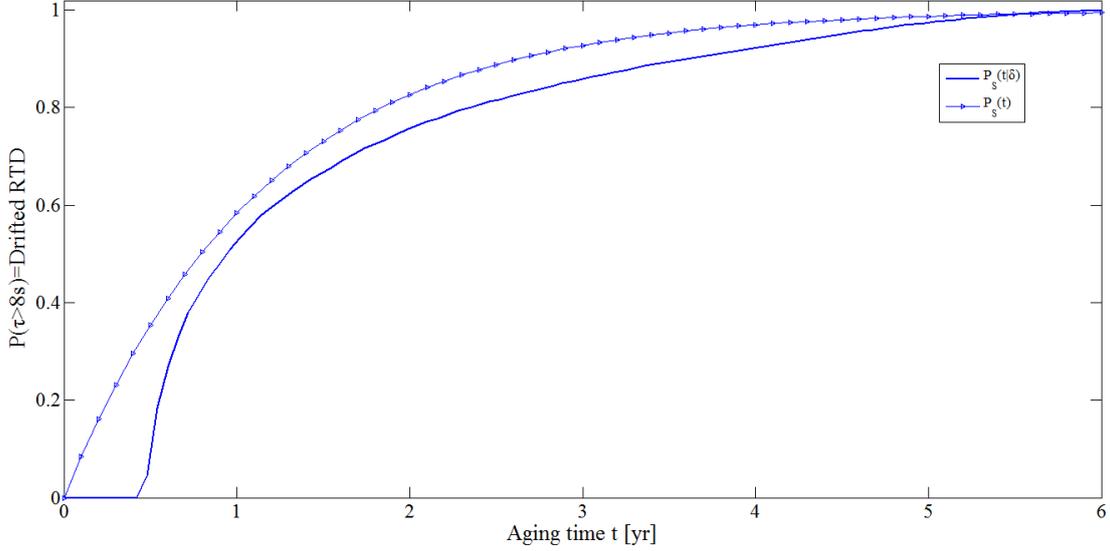


Fig. 14 Failure state probability obtained from MSPM and MCM approaches

3. The System-Level MSPM for a Digital I&C System

In Section 2 we have shown how it is possible to resort to a MSPM framework to build a realistic model of a component/module that is embedded into a system. Now, the general framework of a MSPM approach is presented, when all the components of a Digital I&C system are considered. As shown in Fig. 15, we propose a modular scheme that integrates the component-level and/or subsystem-level models into a system-level structure (Gulati and Dugan, 1997; Wang et al., 2015). Attention should be paid to the definition of system failure modes and to the identification of the component degradation states with their dependencies and the uncertainties, as well as on the inter-component dependencies by a qualitative and quantitative screening of the system behavior (i.e., by Failure Modes and Effects Analysis (FMEA) (Zio, 2007)). In this work, the components degradation process, their failure modes and the overall system behavior is modelled by implementing physics model at both component level or/and subsystem level, which consider inter-component and intra-components

dependencies and uncertainties.

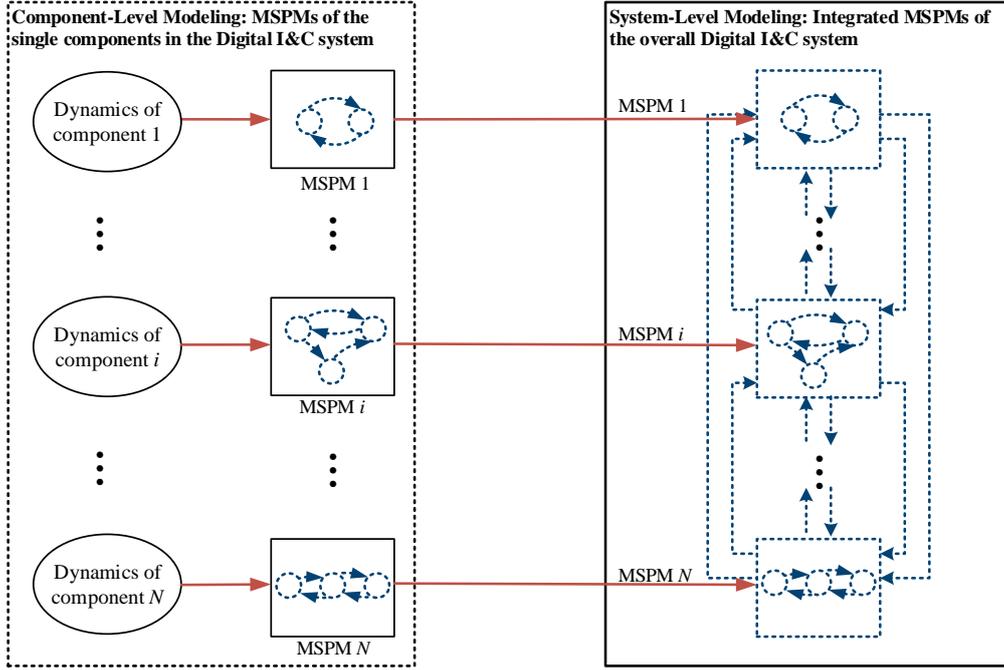


Fig. 15 General framework of a component- and system-level MSPM-based approach for a digital I&C systems

With respect to the inter-component degradation physics, the following assumptions hold for each i -th component degradation MSPM. Provided that we are able to build, for each i -th component, its component-level MSPM, the system-level MSPM can be built, as shown in Fig. 16, where:

- Inter-components operating logic determines the layout of the $N+1$ layers, where layer “ L^0 ” and layer “ L^N ” are the “System functioning (L^0)” and “System failure (L^N)” states, respectively. Any intermediate layer “ L^l ”, $l = 1, \dots, N - 1$ represents the module or submodule operating logic, where the component can be grouped in subsystems for similarity of task, location, characteristics, etc. (Inter-components (e.g., cascading failures) and intra-components (e.g., components common cause failures) dependencies and uncertainties should be accounted for in detail for the sake of the model accuracy).
- L_m^l , $m = 1, 2, \dots, M^l$, is the system degradation state at progression level m , for the module failure mode l .
- All states L_m^l are repairable states with repair rate $\mu_{L_m^l \rightarrow L^0}(t, \bar{\delta})$ to L^0

(indicated by the dashed lines in Fig. 16) and failure rate $\lambda_{L_m^l \rightarrow L_n^\omega}(t, \bar{\delta})$, $l, \omega = 1, \dots, N-1, m = 1, 2, \dots, M^l$, and $n = 1, 2, \dots, M^\omega$ for the transition from state L_m^l to any other state L_n^ω belonging to any layer $\omega \neq l$ (indicated by solid lines in Fig. 16).

- States L_m^l lead to L^N (i.e., system failure due to different failure modes) with failure rate $\lambda_{L_m^l \rightarrow L^N}(t, \bar{\delta})$.

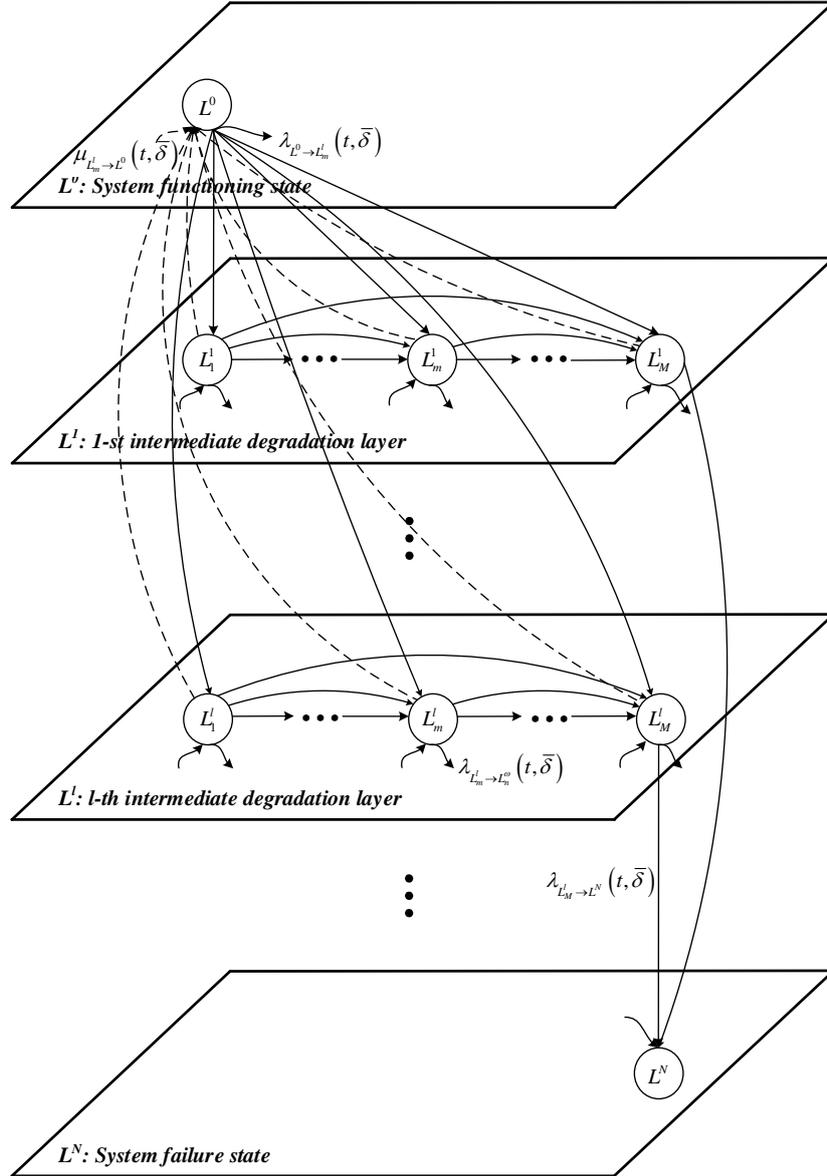


Fig. 16 The system-level MSPM degradation model

The construction of the system-level MSPM for the RPS of Fig. 1, thus, proceeds as follows:

- Identification of system modules: BPL, LCL and RTB (see section 2).
- Definition of the $(N+1) = 5$ layers.

The layers of the system-level MSPM of the RPS (see Fig. 17) are, therefore:

- Layer L^0 is the “RPS functioning (L^0)” state.
- Layer L^1 models the degradation of the system due to either the submodule “S-A and BPL-A” or the submodule “S-B and BPL-B” failure, since the two submodules perform the same function in the system and the logic of such that the system goes into failure if at least one submodule fails. Due attention has to be, therefore, paid to the modeling the logic that process the PTSs. In layer L^1 , states $L_1^1, \dots, L_m^1, \dots, L_M^1$ are states of the degradation of this module (shown in Fig. 17).
- Layer L^2 models the degradation of the LCL module. Due to the operational logic of this module, the system goes into failure state only if both LCL-A and LCL-B fail, namely, the LCL module goes into state L_{AND}^2 that is shown in Fig. 17. States $L_1^2, \dots, L_m^2, \dots, L_M^2$ are states of the degradation of LCL module.
- Layer L^3 models the degradation of the RTB, from which the shutdown signal is directly sent to the power supply system and to the CRDM. States $L_1^3, \dots, L_m^3, \dots, L_M^3$ are the states of degradation of the RTB.
- Layer L^4 is the “RPS failure (L^4)” state.

- Identification of possible cascading failures affecting more than one layer/module.
- Identification of possible common cause failures among components belonging to different layers/modules.
- Quantification of the uncertainties in the probabilities of transition between states and layers/modules.

For completeness sake, in Fig. 17, $\mu_{L_m^l \rightarrow L^0}(t, \bar{\delta})$ ($l = 1, 2, 3, 4$) is the repair transition rate from the state m of the l -th layer to the “RPS functioning (L^0)” state, as a function of time t and of the factors $\bar{\delta}$ affecting the degradation, $\lambda_{L^0 \rightarrow L_m^l}(t, \bar{\delta})$ is the

failure transition rate from the “RPS functioning (L^0)” state to the state m of the l -th layer, $\lambda_{L_m^l \rightarrow L_n^\omega}(t, \bar{\delta})$ ($\omega = 1, 2, 3, 4$) is the failure transition rate from the state m of the l -th layer to the state n of the ω -th layer, and $\lambda_{L_M^1 \rightarrow L^4}(t, \bar{\delta})$, $\lambda_{L_{AND}^2 \rightarrow L^4}(t, \bar{\delta})$, and $\lambda_{L_M^3 \rightarrow L^4}(t, \bar{\delta})$ are failure rates from the worst degradation states of each layer/module to the RPS failure state L^4 .

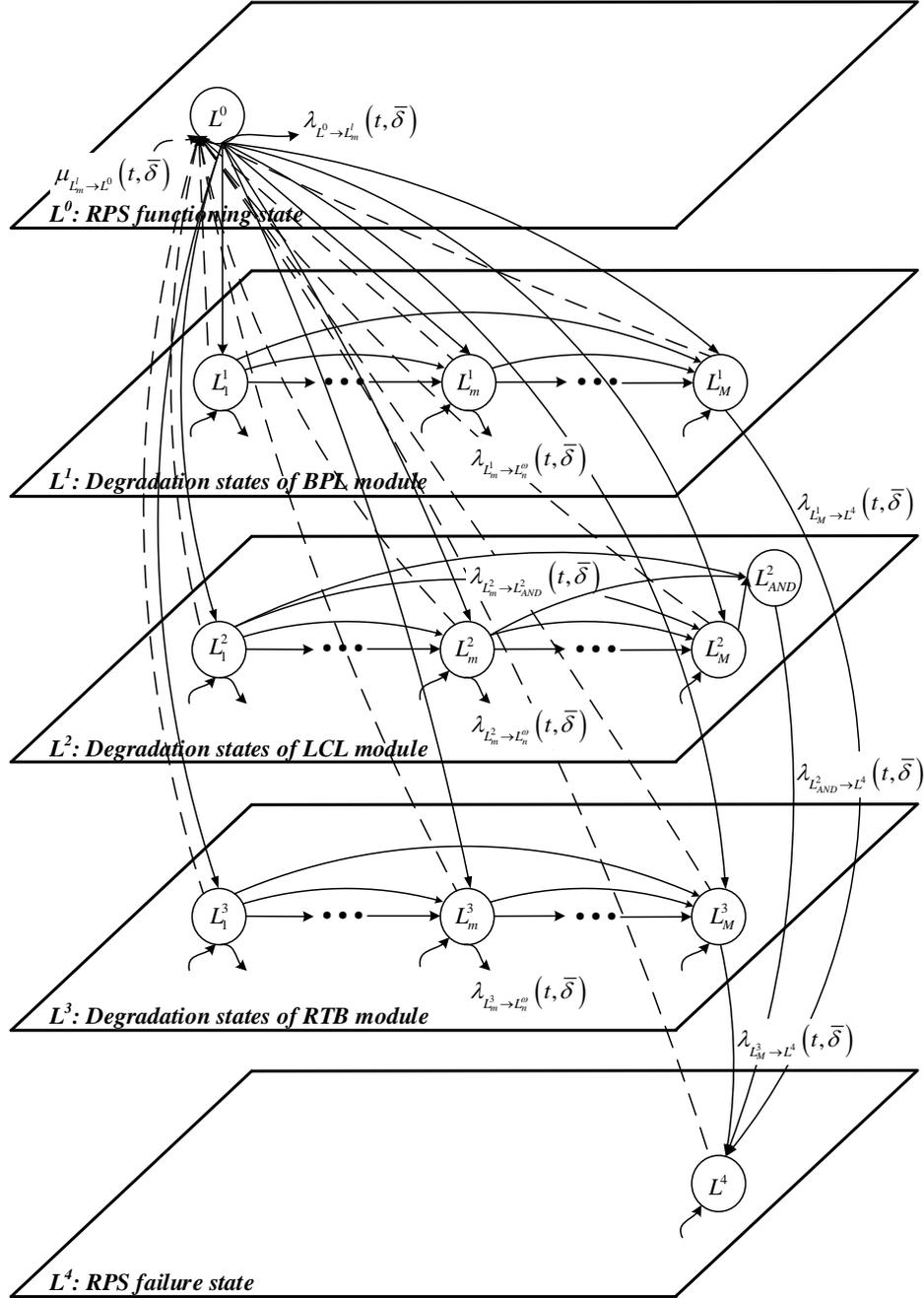


Fig. 17 The system-level degradation model of the RPS digital I&C system

The objective of the system-level degradation modeling effort is the quantification

of the state probability vector $\bar{\mathbf{P}}(t, \bar{\boldsymbol{\delta}}) = \{p_{L^0}(t, \bar{\boldsymbol{\delta}}), p_{L^1_1}(t, \bar{\boldsymbol{\delta}}), \dots, p_{L^l_m}(t, \bar{\boldsymbol{\delta}}), \dots, p_{L^4_M}(t, \bar{\boldsymbol{\delta}}), p_{L^5}(t, \bar{\boldsymbol{\delta}})\}$, which can be obtained by Monte Carlo simulation to generate random walks across layers and within the MSPM describing each module, as illustrated in section 2. Without loss of generality, we will present a system-level MSPM where only the RTD-MSPM of section 2.1 is considered, whereas all the other components are assumed to obey a binary behavior (safe/failed). Therefore, the RPS system-level MSPM sketched in Fig. 19, whose states are described in Table 5, can be built:

- State L^0 is assumed as the system initial state, where all the components are new at $t=0$.
- State L^1_1 corresponds to the onset of the RTDs New-to-drift failure mode that may lead to system failure state L^4 when its response time exceeds the failure threshold (solid edges with transition rates represent the stochastic phenomena herein modeled).
- Absorbing states L^1_2, L^1_3, L^2_2 and L^2_{AND} , deterministically lead without any stochasticity to the RPS system failure.
- Time-varying transition rates of the RTD sensors are taken from those of the RTD-MSPM of Section 2.2 (see Fig. 12), whereas the transition rates of binary-state components (i.e., BPLs, LCLs and RTB) are taken from public databases (BPL and LCL failure rates λ_B and λ_L are equal to 1E-6/hr and 5E-6/hr respectively (US: EPRI, 2008), with the common cause factor $\beta=0.1$; whereas RTB failure rate λ_R is equal to 4.3E-8/hr (IAEA, 1992)).

Table 5 Identification of states in RPS system-level MSPM

State	Description
L^0	RPS functioning state.
L^1_1	Drift onset in one or the other RTD sensor.
L^1_2	Either one or the other BPL fails to send out PTSSs.
L^1_3	Common cause failure of BPL-A and BPL-B.
L^2_1	Either one or the other LCL fails to produce the ESS.
L^2_2	Common cause failure of LCL-A and LCL-B.
L^2_{AND}	Both LCLs fail to produce the ESS.
L^3_1	RTB fails during operation.
L^4	RPS failure state.

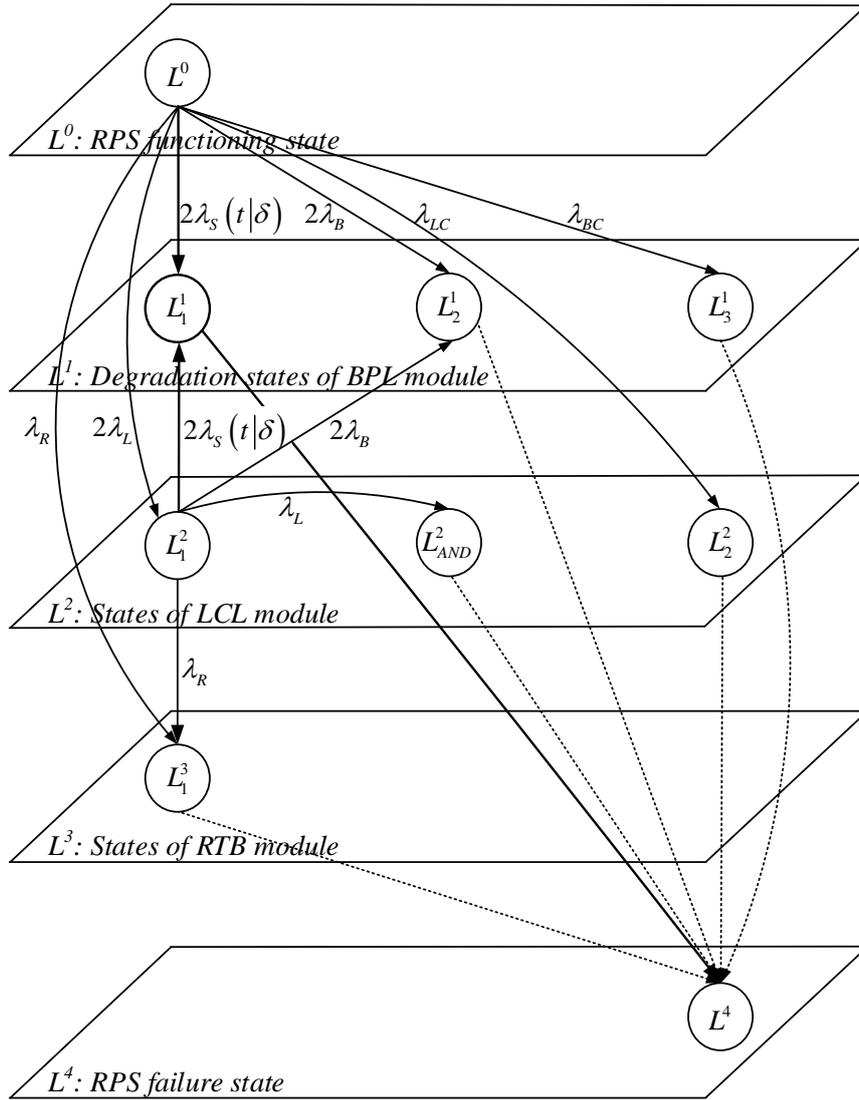


Fig. 19 The RPS system-level MSPM integrating RTD New-to-drift failure mode

After N_M trials of the Monte Carlo simulation, whose flowchart is shown in Fig. 20 (where the inner shadowed loop corresponds to the Monte Carlo simulation adopted in Section 2.2 for the estimation of the RTD drift onset time), the calculated RPS unreliability $P(t|\delta)$ is plotted in Fig. 21, and compared with the system unreliability $P(t)$ obtained by solving a binary-states homogeneous continuous-time discrete-state Markov Chain Model (i.e., the RPS-MCM), where also the RTDs are assumed to have a constant transition rate equal to $1E-4/hr$ (US: EPRI, 2008). It can be seen that the RTD-MCM overestimates the system unreliability, which makes the design over conservative, or the maintenance over demanding, especially at the early stage of the

system life.

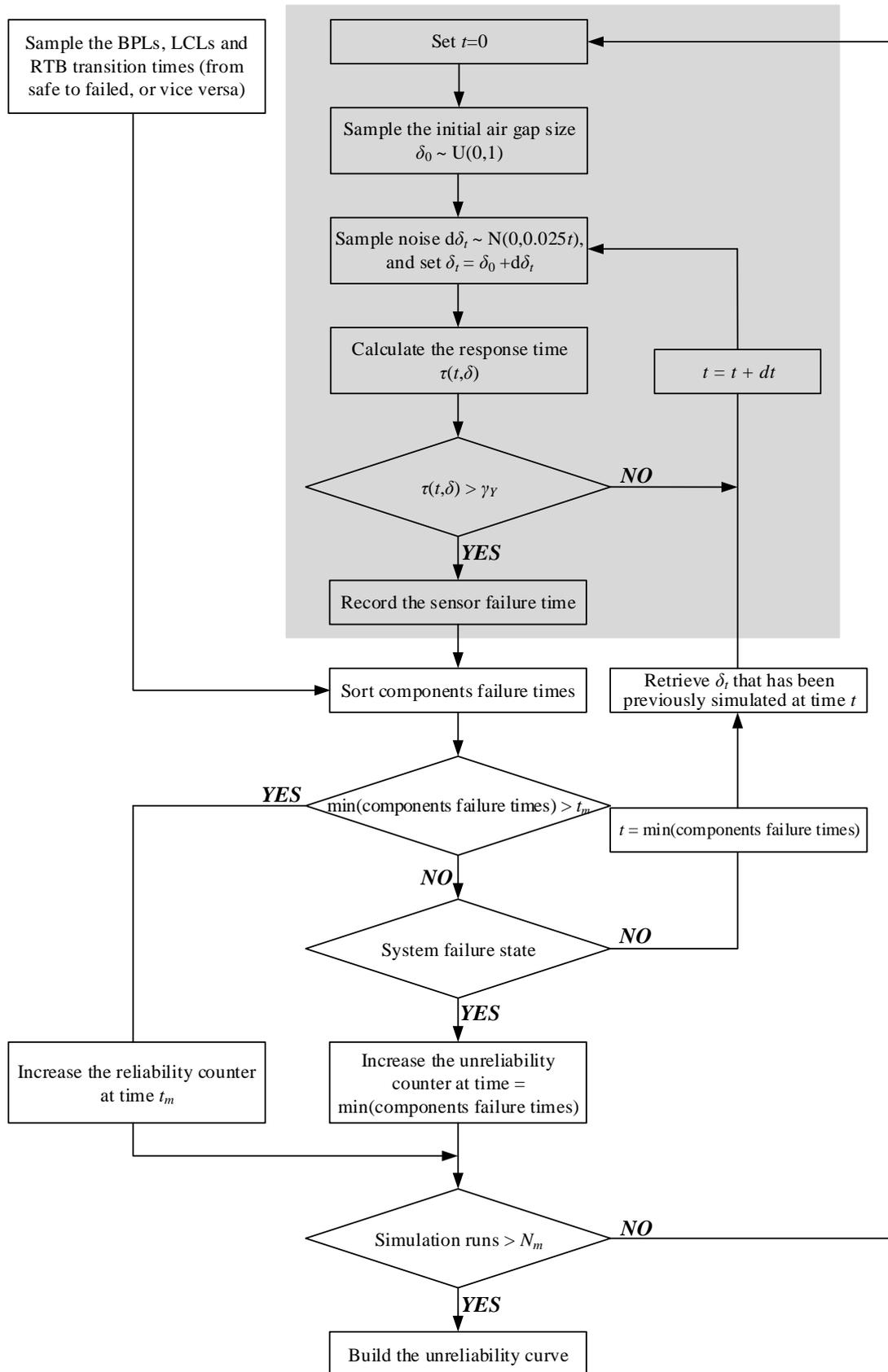


Fig. 20 The flowchart of the Monte Carlo Simulation for the system-level MSPM

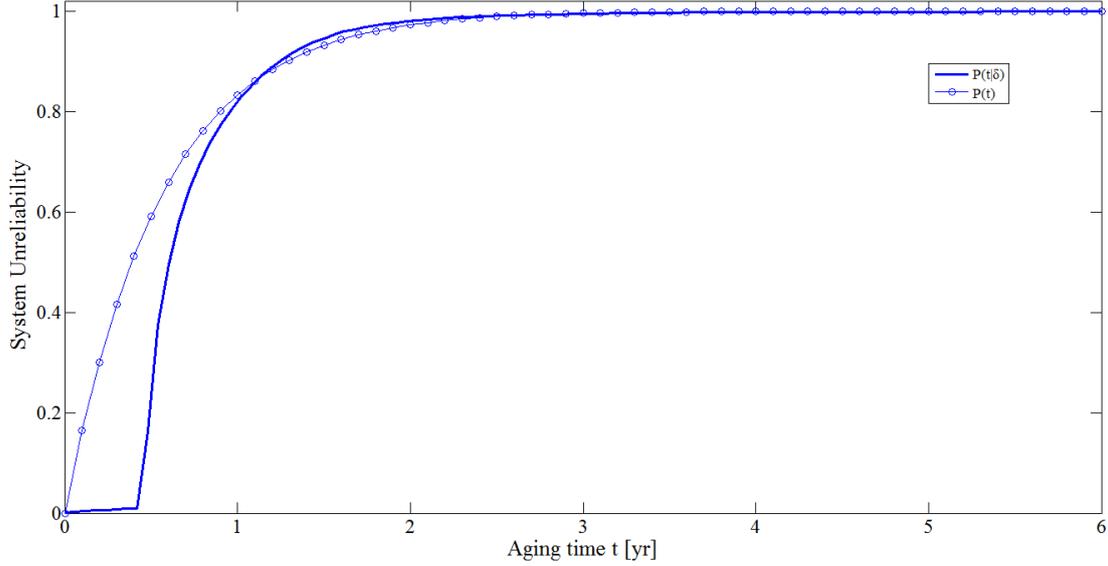


Fig. 21 System unreliability obtained from RPS-MSPM and RPS-MCM framework

4. Conclusions

A MSPMs framework has been proposed for reliability modeling and assessment of digital I&C systems in NPPs. The designers and operators efforts needed to gather the necessary, and quite detailed, physics of failure data that are to be treated within the MSPM framework makes the method primarily applicable to critical NPPs components. Indeed, in this work a RTD was considered to develop the component-level MSPM model, being a typical signal input source of digital I&C systems that critically affects its reliability. Monte Carlo simulation has been used for estimating the degradation state probability (i.e. drift-failure state probability) in the RTD-MSPM, by sampling the stochastic evolution of the air gap size, under deterministic aging. The comparison of MCM and MSPM results shows that with realistic assumptions and available knowledge, MSPM provides a better and more complete representation of the component degradation progression.

The modular modeling approach proposed has been applied to develop a system-level MSPM model taking for a typical Reactor Protection System (RPS) as benchmark.

The results of the illustrative application demonstrate that the proposed MSPM approach can well explain environmental conditions, aging and degradation of failure events, besides timing and sequencing which can also be solved by traditional dynamic methods (e.g., MCM).

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