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Gaits Stabilisation for Legged Robots using Energetic Regulation

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Abstract— A new method is used to stabilize periodic cycles for legged robots with fast dynamic gaits. The control objective is regulation of the system energy (using a nominal energetic representation) for stabilization of fast gaits. The Controlled Limit Cycles (CLC) allow to establish a quasi periodic hopping gait by energy shaping and regulation. The resulting control law is simple, efficient and easy to implement. This approach leads simultaneously the control and reference trajectories (implicitly generated). Robustness and effectiveness of the proposed control are illustrated by simulations and experimental results.

Keywords— Robots control, Gaits stabilization, Controlled Limit Cycles, Hopping gaits, Energy regulation.

I. INTRODUCTION

For robots with fast gaits, the crucial problem is gaits design and control, in order to preserve stability despite interaction with an unknown and variable environment. For fast gaits (long flight phases where the robot is not completely controllable and short contact phases), the major problem is to generate trajectories which cope with system structure, the contact features and power optimization.

The CLC approach involves self generated trajectories and stabilizes a quasi periodic hopping motion [1][2]. This emphasizes the fact that the system's behavior and trajectories are consequences of robot's dynamic + the control and ground interactions. Energy shaping is performed by use of an admissible control. We show that control of legged robots can be reduced to study an equivalent energy model involving the robot, the ground and a simple control law. Our main objective is to introduce an approach leading to efficient stabilization of quasi periodic gaits which can be applied to legged robots. The energy of the system is regulated for an automatic generation of trajectories (implicit and deduced from the dynamic behavior), and a rejection of perturbations and unknown environment events. This is realized by means of *Controlled Limit Cycles (CLC)* and regulation of the system energy. For hopping gaits (monitored by gravitation and potential effects), the limit cycles correspond to a motion in the subspace (z, \dot{z}) of the phase space of the system. This can be obtained by projection of the robot's dynamic on the motion subspace (z, \dot{z}) . The projected motion correspond to a mass and spring system as those studied in literature [3][4][5][6][7][8][9][10][11]. Then gaits are obtained by use of natural periodic cycles of the system's dynamic.

The proposed control approach, is based on a Variable Structure Control as presented in Figure (1). The presentation of the approach has been first developed for one

DOF in [10][1][2] and then is extended here for 2 DOF and more. This control approach is applied to generate trajectories and stabilization for hopping gait. Afterward in order to clarify its generalization to more complex systems, a control structure is designed to stabilize periodic cycles for systems with more than 3 DOF.

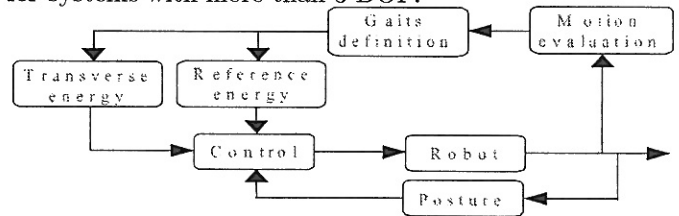


Fig. 1. Control methodology for legged robots

Section 2 presents an invariant periodic orbit for mass spring system and extension of the CLC approach for 2DOF system. In section 3 we present an application of the proposed approach to SAP robot which has 3DOF. Some discussions and comments are presented in section 4. Our future prospects, investigation and some conclusions in this work will be given.

II. VARIABLE STRUCTURE FOR ROBOTS CONTROL

A. The nominal models

Figure (2b) depicts an elementary system for analysis of robots locomotion in fast motions. It is composed by a body with a mass M and inertia I and two massless springs with stiffness k_l, k_r and nominal lengths z_{lo}, z_{ro} . Two controllable displacement u_l and u_r (inputs) are added to spring lengths. Subscripts l and r denote left and right elements. The body has 2 degrees of freedom, a vertical displacement denoted by z (displacement of mass center) and a revolute displacement θ around the center of mass. We assume punctual the contact between the end point of springs and the ground (assumed more rigid than equivalent impedance of the robot legs: $k_r, k_l \ll k_e$).

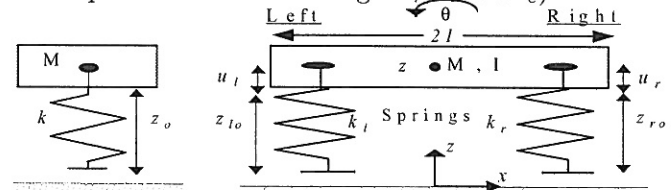


Fig. 2. Mass-spring systems: a) 1 DOF and b) controlled 2DOF. This system has 4 motion phases: 1) stance where both springs are on the ground and compressed, 2) flight phase (no contact with the ground), the robot has a ballistic trajectory and 3) two other phases where only one spring is in

contact with the ground ($z_r \leq z_{ro}$ or $z_l \leq z_{lo}$). The beginning of the flight and stance are called lift-off and touch-down, respectively. The apex and bottom correspond to maximum and minimum body height, $g = 9.81 m.s^{-2}$ is the gravitational acceleration. Since the robot corresponds to a variable structure system, let us define commutation contact functions $\xi_r(z_r, u_r)$ and $\xi_l(z_l, u_l)$ which are null during flight and equal to one during contact. They can be defined as: $\xi_i = \xi_i(z_i, u_i) = \frac{1}{2}(1 + \text{sign}(z_i - z_{io} - u_i))$. ($i = r$ or l). This allows us to represent all motion phases as:

$$\begin{cases} \ddot{z} = \xi_l \frac{k}{M} (z_l - z_{lo} - u_l) + \xi_r \frac{k}{M} (z_r - z_{ro} - u_r) - g \\ \ddot{\theta} = \xi_l \frac{I k}{I} (z_l - z_{lo} - u_l) + \xi_r \frac{I k}{I} (z_r - z_{ro} - u_r) \end{cases} \quad (1)$$

B. Invariant periodic orbits

The control proposed in this paper consists first in the characterization of an invariant periodic orbit and then to design a control structure to stabilize such an orbit. To this end we consider first the mass spring system of figure (2a) studied in [10]. It has only vertical displacements (1DOF) and 2 phases, flight and stance, 1 DOF. The system equation can be written

$$\ddot{z} + \xi(z) \frac{k}{M} (z - z_o) = -g \quad (2)$$

During motion, if we assume the landing without rebonds and no energy loss (no friction) [12], system energy evolutions are: Potential (g) \rightarrow kinetic \rightarrow potential accumulation \rightarrow potential restitution \rightarrow kinetic and so on [10].

This shows existence of periodic cycles corresponding to system oscillations and the energy storage (in potential elastic form) for exchange between the ground and the body (mass M). The system dynamic describes a closed periodic orbit or an invariant limit set - $o(x_o, t)$ (where $x_o(z_m, \bar{z}) = 0$) is initial state). This orbit - o can be described by a Lyapunov equation (3):

$$V_f(z, \dot{z}) = \frac{1}{2} \dot{z}^2 + gz + \frac{k}{2M} (z - z_o)^2 = V_{LC}^* \quad (3)$$

The system is conservative: $\dot{V}_f(z, \dot{z}) = 0 \forall z, \forall t \geq 0$. All the trajectories $z(t)$ are in - o , see Figure (3). For 1DOF systems, equation (3) can be used to describe arbitrary motions which cope with the system's ability. Therefore, system (2) has a hopping motion with constant jumps height and a natural frequency depending on $\omega_o = \sqrt{\frac{k}{M}}$. If the system dissipates energy during the stance phase, it is clear that no periodic orbit can exist. However, we can design a control law to stabilize the energy to a reference value V_{LC}^* (defined at the apex z_{max} or the lift off point z_d see figure (3)). If the system has more than one DOF, then its motion depends on more degrees of freedom, namely $(z, \dot{z}, \theta, \dot{\theta})$ for the example in case of the system of figure (2b). The phase space is of dimension 4, but a closed orbit can be obtained only for a motion in the vertical axis (z, \dot{z}) and the other cases are either not obvious or not realizable. Invariant set corresponding to constant energy values V_{LC}^*

does not mean a closed orbit for more than 1DOF. In order to stabilize a periodic motion (vertical hopping) by means of Controlled Limit Cycles, in our approach we first project the system dynamics on (z, \dot{z}) phase subspace and then we stabilize a limit cycle in (z, \dot{z}) .

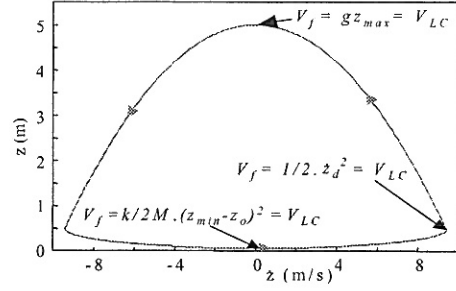


Fig. 3. Periodic orbit for 1DOF free system

C. CLC approach for cycle stabilization

The aim is to evaluate the system motion in the case of hopping and then decompose his energy in two terms, one for the transverse dynamics denoted V_T (i.e: motion in $(\theta, \dot{\theta})$) and a second term for the cycle energy denoted V_{LC} . We can then apply a control u_T to dissipate transverse energy V_T . Then we stabilize a periodic cycle or regulate the energy level V_{LC} to V_{LC}^* in the phase subspace (z, \dot{z}) . Let us consider the Lyapunov function:

$$V = \frac{1}{2} \dot{z}^2 + gz + \frac{I}{2M} \dot{\theta}^2 \quad (4)$$

It can be splitted in two terms as:

$$V = V_{LC} + V_T \quad \text{with:} \quad \begin{cases} V_{LC} = \frac{1}{2} \dot{z}^2 + gz \\ V_T = \frac{I}{2M} \dot{\theta}^2 \end{cases} \quad (5)$$

We propose the following non linear control structure:

$$\xi_r u_r = \frac{1}{2} (u_{LC} + u_T) \quad \text{and} \quad \xi_l u_l = \frac{1}{2} (u_{LC} - u_T) \quad (6)$$

where u_T should dissipate the transverse energy, in order to project the motion on the phase subspace (z, \dot{z}) and u_{LC} stabilizes a cycle in (z, \dot{z}) . u_T and u_{LC} will be defined by stability analysis.

D. Stability analysis

First, we have to dissipate the transverse energy V_T . Let us take the Lyapunov function $V_1 = \frac{1}{2} V_T^2$, its derivative is $\dot{V}_1 = V_T \dot{V}_T$ with: $\dot{V}_T = \frac{I}{M} \dot{\theta} \ddot{\theta}$. Substituting $\ddot{\theta}$ (equation 1) and using control functions (6), we obtain:

$$\dot{V}_1 = \frac{I k}{M} (\xi_l (z_l - z_{lo}) + \xi_r (z_r - z_{ro}) + u_T) \dot{\theta}^2 \quad (7)$$

We take as transverse control function u_T :

$$u_T = -\Gamma_1 \psi(V_T) \dot{\theta}^2 - \xi_l (z_l - z_{lo}) + \xi_r (z_r - z_{ro}) \quad (8)$$

Γ_1 is a positive control gain and $\psi(\xi)$ is positive function of ξ such as $\psi(\xi) \cdot \xi > 0, \forall \xi \neq 0$ and $\psi(0) = 0$ (namely sign or saturation function). Then we have: $\dot{\bar{V}}_T = \square \frac{k_l}{M} \Gamma_1 \psi(V_T) \theta^2$,

$$\bar{V}_1 = V_T \bar{V}_T = \square \Gamma_1 V_T \psi(V_T) \theta^2 \leq 0 \quad (9)$$

Thus $V_T \bar{V}_T < 0$ and therefore the transverse energy V_T converges to zero and is such as $\forall \epsilon_1 > 0, \exists t_0 \geq 0$, such as $|V_T| < \epsilon_1, \forall t > t_0$. Then $\bar{\theta}$ is bounded and tend to zero, then we can choose a constant λ such as $\Gamma_1 \bar{\theta}_{\max}^2 \leq \frac{\lambda}{\sqrt{2}}$, with $\bar{\theta}_{\max} = \max(\bar{\theta})$. Using (9), $\bar{V}_1 \leq \square \frac{\lambda}{\sqrt{2}} |V_T|$ and $V_1 = \frac{1}{2} V_T^2 \Rightarrow V_T = \sqrt{2V_1}$ then $\bar{V}_1 \leq \square \lambda \sqrt{V_1}$. We can conclude that:

$$V_1(t) = 0, \quad t \geq t_1 = t_0 + 2 \frac{\sqrt{V_1(t_1)}}{\lambda}$$

therefore the sliding surface is attractive and $V_T = 0$ is reached in a finite time t_1 , the motion is then reduced to $(z, \dot{z}, 0)$. Now to ensure convergence of V_{LC} to reference value V_{LC}^* (defining a desired cycle), let us choose as Lyapunov function candidate:

$$V_2 = \frac{1}{2} V_T^2 + \frac{1}{2} (V_{LC} \square V_{LC}^*)^2 \quad (10)$$

V_{LC}^* is a constant value and can be defined at the lift-off (\bar{z}_d) or at the apex (z_{\max}) as:

$$-o = \{(z, \dot{z}) \in \mathbb{R}^2: V_{LC} = \frac{1}{2} \dot{z}^2 + gz = gz_{\max}\} \quad (11)$$

At $t > t_1$ we have $(V_T, \dot{V}_T) = (0, 0)$, then:

$$\dot{V}_2 = (V_{LC} \square V_{LC}^*) \dot{V}_{LC} \quad \forall t > t_1 \quad (12)$$

$$\text{with: } \dot{V}_{LC} = (\ddot{z} + g) \dot{z} \quad \forall t > t_1 \quad (13)$$

substituting \ddot{z} (1) and control (6) in \dot{V}_{LC} , we have:

$$\dot{V}_{LC} = \frac{k}{M} (\square \xi_l (z_l \square z_{l0}) \square \xi_r (z_r \square z_{r0}) + u_{LC}) \dot{z} \quad (14)$$

The main objective is to make $-o$ invariant and attractive. To stabilize this orbit, we propose the feedback u_{LC} :

$$u_{LC} = \square \Gamma_2 \psi(V \square V_{LC}^*) \dot{z} + \xi_l (z_l \square z_{l0}) + \xi_r (z_r \square z_{r0}) \quad (15)$$

Γ_2 is a positive control gain and then \dot{V}_{LC} becomes: $\dot{V}_{LC} = \square \frac{k}{M} \Gamma_2 \psi(V \square V_{LC}^*) \dot{z}^2$. From equation (12), we obtain the orbit stabilization condition:

$$\bar{V}_2 = \square \frac{k}{M} \Gamma_2 (V_{LC} \square V_{LC}^*) \psi(V_{LC} \square V_{LC}^*) \dot{z}^2 \leq 0 \quad (16)$$

This stability condition gives the following two cases:

Case 1: $V_{LC} > V_{LC}^* \Rightarrow \bar{V}_2 = \square \frac{k}{M} \Gamma_2 \psi(V_{LC} \square V_{LC}^*) \dot{z}^2 < 0$, the height is greater than z_{\max} (orbit $-o$), then system energy is decreased and the control u_{LC} has dissipative effect.

Case 2: $V_{LC} < V_{LC}^* \Rightarrow \bar{V}_2 > 0$, the jump height is less than this of V_{LC}^* (z_{\max}), then u_{LC} is active and supply energy to system.

We can then conclude, for the controlled limit cycle, that the periodic orbit $-o$ is a stable invariant set of the controlled system. Trajectories $z(t)$ converge asymptotically to orbit $-o$ defined above (11). $\theta(t)$ and $\dot{\theta}(t)$ converge to zero due to a control structure (8) and (15). u_T will then dissipate transverse energy and u_{LC} can be either active or passive (dissipative) for cycle stabilization. Note that the CLC control is enabled only in the stance phase (the controllable region) and is frozen in the flight phase (uncontrollable region).

E. Simulation of CLC

For the simulations, we use as ψ a saturation function (with maximum and minimum values ± 0.2). The initial conditions are $\bar{z}(0) = 0, z(0) = 5.5m, z_{l0} = z_{r0} = 1m$. The control gains are $\Gamma_1 = 0.2$ and $\Gamma_2 = 0.008$ and parameters are $k_l/M = k_r/M = 500$ and $I = 2kg \cdot m^2$. The desired height of hopping is $z_{\max} = 3.5m$. For the first simulation, the initial condition is $\theta_o = 0, \dot{\theta}_o = 0$. Figure (4) shows a stabilization of body height to desired value z_{\max} (figure (4a)) and convergence of cycle energy to the reference value (4b). The transverse energy remain equal to zero. This situation is equivalent to systems with 1DOF, where only u_{LC} acts to stabilize the limit cycle (system motion is in (z, \dot{z}) phase plane).

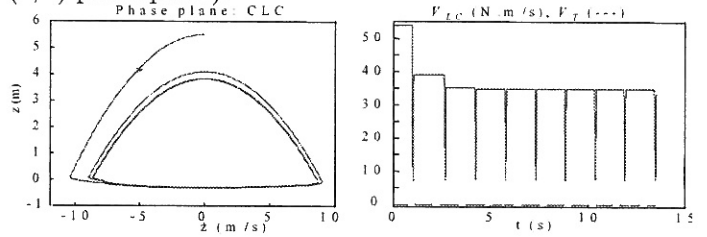


Fig. 4. CLC: $\theta_o = 0$

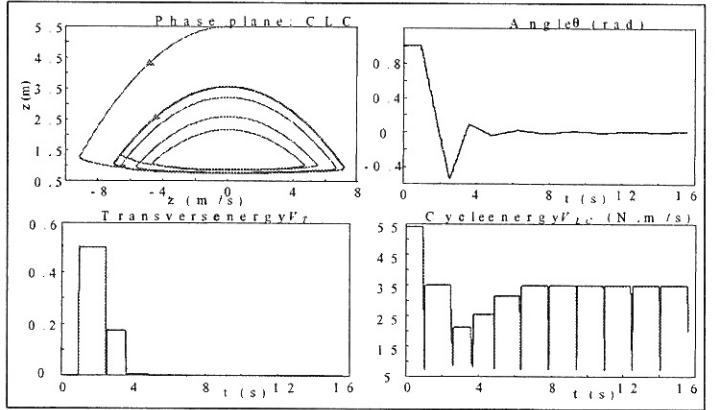


Fig. 5. Transverse and cycle energy, CLC: $\theta_o = 1rad$

We consider now $\theta_o = 1rad$. We keep the same conditions (gains value, initial z and desired height z_{\max}). Figure (5) shows that limit cycle is stabilized after four oscillations and convergence of angle θ to zero in finite time ($t_1 = 8$ sec). This corresponds to the period necessary to dissipate transverse energy V_T and to stabilize V_{LC} to reference value (figure (5)). For robustness illustration, we consider $\theta_o = 1.5rad$ (near to the limit value $\theta_o = \frac{\pi}{2}$), Figure (6) shows a stable CLC. Figure (6) illustrates system behavior when he strike the ground. System, under control

action, deaden his motion to preserve his equilibrium and to dissipate transverse energy. Note that the controller is active only in stance phase.

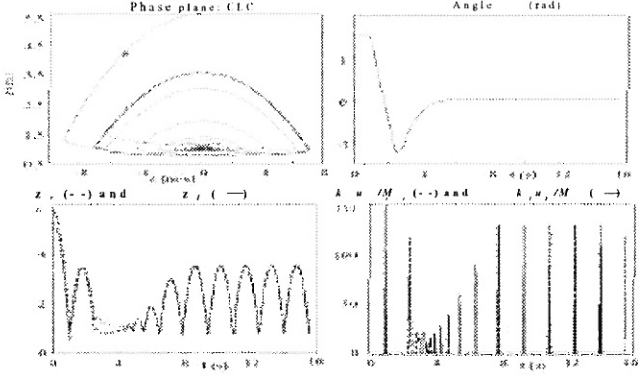


Fig. 6. Hopping heights and input forces, CLC and rotation angle:

Robustness is illustrated by Figure (7), we have changed stiffness k_l and k_r separately and $\theta_o = 0.5rad$. The orbit is reached despite the use of k_l/M six times greater and k_r/M six times smaller. We can observe appearance of transverse energy quantity at each contact with the ground due to the difference between stiffness and the controller action which reduces this quantity to zero.

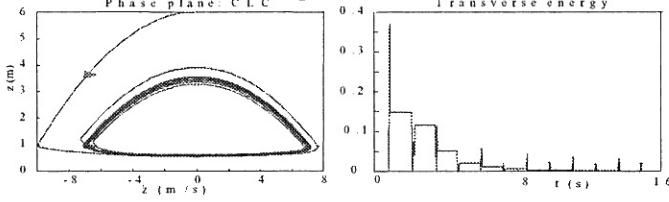


Fig. 7. CLC and transverse energy: changed parameters

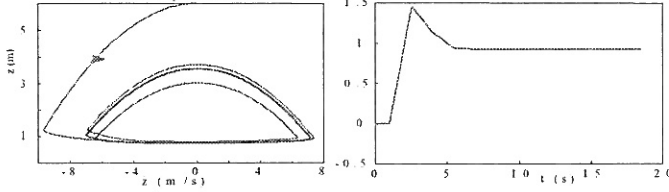


Fig. 8. CLC and rotate angle θ : changed of springs original lengths

Proceeding in a same manner, we have changed the original lengths z_{lo} and z_{ro} ($z_{ro} = 1.2$ and $z_{lo} = 0.8$). Simulation shows that θ is maintained around a value not null due to difference between springs lengths. Therefore the system reach a favourable posture for a stable hopping motion (Figure (8)). To analyse robustness of the controller versus frictions, we introduce a viscous friction in the mechanism $\beta \dot{z}$. Then we obtain the closed loop equation on vertical axis:

$$\ddot{z} = - \left(\frac{k}{M} \Gamma_2 \psi(V - V_{LC}^*) + \frac{\beta}{M} \right) \dot{z} - g$$

The previous equation shows that when the amplitude of control input u_{LC} is limited and if,

$$\beta > \max(k \Gamma_2 \psi(V - V_{LC}^*)) \quad \forall t \in \mathbb{R}_+ \quad (17)$$

then the system will be asymptotically stable near the origin and then no limit cycle exists. Introduction frictions produces deformation of the orbit (symmetry is lost), rebounds reappears and the height of jumps is reduced. The

height reduction is important for high frictions level until the periodic orbit disappears, the system becomes asymptotically stable. To preserve limit cycle, we need a great feedback gain Γ_2 .

III. APPLICATION TO THE SAP ROBOT

In order to generalize the proposed approach, let us consider the legged robot SAP presented in Figure (9). The cartesian variables can be defined, for the end point of robot, as $x = [x_r, z, \phi]^T$ with respect to a reference frame fixed to robot wheel. x_r and z are respectively the horizontal and vertical coordinates and ϕ the orientation. The system dynamics in cartesian space is obtained by use of geometric model $x = L(q)$, where $q = [q_0, q_1, q_2]^T$ is generalized coordinates (with respect to a reference frame fixed to robot wheel), and the corresponding jacobian matrix $J(\bar{x}) = J\dot{q}$:

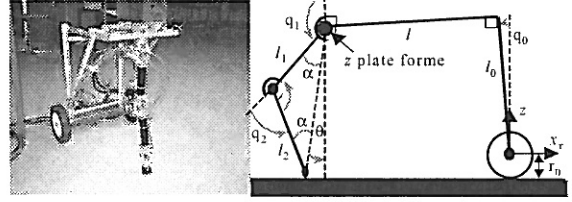


Fig. 9. SAP Robot of LRP

$$M^* \ddot{\bar{x}} + C^* \dot{\bar{x}} + g^* = F + F_e \quad (18)$$

$\tau = J^T F = [0, \tau_1, \tau_2]^T$ is the control input vector and F_e , g^* and $C^* \dot{\bar{x}}$ are respectively vectors of ground reaction, gravitational forces and centripetal and Coriolis forces. M^* is the inertia matrix and J is the jacobian matrix. Note that only q_1 and q_2 are actuated.

A. Stabilizing feedback

The first thing we have to deal with is definition and stabilization of a posture (nominal robot position) when in contact with the ground in order to keep robot in equilibrium. The robot configuration (posture) is presented in Figure (9) and defined by the desired position vector $x_d = [x_{rd}, z_d, \phi_d]^T$. This posture can be stabilized by a position feedback:

$$F = K_p(x_d - x) \quad (19)$$

We consider also the ground very stiff ($K_e \gg K_p$), the contact stiffness is $K = \frac{K_p \cdot K_e}{K_p + K_e} \simeq K_p$, then the system closed loop equation is [1]:

$$M^* \ddot{\bar{x}} + C^* \dot{\bar{x}} + g^* + Kx = Kx_d \quad (20)$$

The equation (20) describes the legged system with a primary feedback loop which is able to have energy exchanges with the environment (introduction of elastic storage).

P1: The elements of the system's model (20) are such that the matrix $N = \frac{1}{2} \dot{M}^* - C^*$ is skew symmetric [13].

P2: The mapping $v = Kx_d \mapsto y = \dot{\bar{x}}$ verifies the passivity property $\int_0^t y^T v dt = \int_0^t \dot{\bar{x}}^T Kx_d dt \geq -\gamma^2$ [14].

Motion Evaluation and gait definition: The robot SAP has 3 DOF and can be considered as an association of 3 coupled second order subsystems:

$$\begin{pmatrix} \ddot{x}_r \\ \ddot{z} \\ \ddot{\phi} \end{pmatrix} = \begin{pmatrix} f_x(\chi, \chi_d, K) \\ f_z(\chi, \chi_d, K) \\ f_\phi(\chi, \chi_d, K) \end{pmatrix} \quad (21)$$

where $\chi = [x^T, \dot{x}^T]$ is the model state vector. For motion evaluation in the case of hopping, the preceding analysis shows that the phase space (x, \dot{x}) have to be decomposed in two subspaces, one for the robot hopping (z, \dot{z}) and its transverse. Proceeding in a same way, we consider the Lyapunov function:

$$V(x, \dot{x}) = \frac{1}{2} \dot{x}^T M^* \dot{x} + P_g(t) \quad (22)$$

where potential energy P_g due to gravity is $P_g(t) = \int_0^x g^*(s) ds$. Then this Lyapunov function can be splitted in two parts (one for the limit cycle and the second for the transverse motion):

$$V(x, \dot{x}) = \frac{1}{2} \dot{x}^T M^* \dot{x} + P_g(t) = V_{LC} + V_T \quad (23)$$

For hopping, we can specify, for reference energy, either the height of the jumps (z_m) or the lift off velocity (\dot{z}_d) .

$$V_{LC}^*(z, \dot{z}) = \frac{1}{2} m \dot{z}_d^2 = mgz_m \quad (24)$$

where m is the equivalent mass of the robot. We then must stabilize two subsystems at $(x_r = 0, \dot{x}_r = 0)$ and $(\phi = 0, \dot{\phi} = 0)$ asymptotically and control the one of vertical motion (z, \dot{z}) to obtain a CLC.

B. Controlled Limit Cycles for SAP robot

Let us consider the nominal position $x_o = [x_{rdo}, z_o, \phi_{do}]^T$ around which the cycles will be stabilized, and the control functions:

$$x_d = [x_{rd}, z_d, \phi_d]^T = [x_{rdo} + u_x, z_o + u_z, \phi_{do} + u_\phi]^T. \quad (25)$$

The energetic control and the CLC are realized by the control functions $u = u_T + u_{LC} = [u_x, u_z, u_\phi]^T$ defined as $u_{LC} = [0, u_z, 0]^T$ and $u_T = [u_x, 0, u_\phi]^T$. u_z is used to control the hopping motion in the phase plane (z, \dot{z}) and (u_x, u_ϕ) maintain the system motion in this phase subspace. This reduces the energy to the cycles one in a finite time t_1 by means of u_T . u_{LC} stabilizes an orbit - o by regulation of energy $V_{LC}(z, \dot{z})$ to a reference value V_{LC}^* . Then we propose the control structure (Γ_1, Γ_2 and Γ_3 are positive control gains):

$$u_T : \begin{cases} u_x = x_r - x_{rdo} - \frac{\Gamma_1}{k_{11}} \psi(V_T) \dot{x}_r \\ u_\phi = \phi - \phi_{do} - \frac{\Gamma_3}{k_{33}} \psi(V_T) \dot{\phi} \end{cases} \quad (26)$$

$$u_{LC} : u_z = z - z_o - \frac{\Gamma_2}{k_{22}} \psi(V_{LC} - V_{LC}^*) \dot{z} \quad (27)$$

C. Stability analysis for SAP robot

Let us consider the Lyapunov function (23), its time derivative is then (using the skew symmetry of M^*):

$$\begin{aligned} \dot{V} &= \dot{x}^T K (x_d - x) = \dot{V}_{LC} + \dot{V}_T \\ &= \dot{x}_r^T k_{11} (x_{rd} - x_r) + \dot{\phi}^T k_{33} (\phi_d - \phi) + \dot{z} k_{22} (z_d - z) \end{aligned} \quad (28)$$

The transverse dynamics of the two subsystems f_x and f_ϕ are defined by transverse energy variation \dot{V}_T and \dot{V}_{LC} describes the limit cycle of f_z . Conditions to ensure transverse energy dissipation, can be obtained with the Lyapunov function V_1 and its time derivative is (see 28):

$$V_1 = \frac{1}{2} V_T^2 \quad (29)$$

$$\dot{V}_1 = V_T \dot{V}_T = V_T \left(\dot{x}_r^T k_{11} (x_{rd} - x_r) + \dot{\phi}^T k_{33} (\phi_d - \phi) \right)$$

Substituting the desired trajectories (25) and control (26), we obtain:

$$\dot{V}_1 = -\lambda V_T \psi(V_T) \left(\dot{x}_r^2 \Gamma_1 + \dot{\phi}^2 \Gamma_3 \right) \leq 0 \quad (30)$$

$V_T \dot{V}_T$ is negative (convergence condition is satisfied), we can conclude that \dot{x}_r and $\dot{\phi}$ are bounded and tend to zero. We can choose λ such as $\dot{x}_r^2 \Gamma_1 + \dot{\phi}^2 \Gamma_3 \leq \frac{\lambda}{\sqrt{2}}$.

Thus equation (30) becomes $\dot{V}_1 \leq -\frac{\lambda}{\sqrt{2}} |V_T|$ and from $V_T = \sqrt{2V_1}$, we have $\dot{V}_1 \leq -\lambda \sqrt{V_1}$. Then $V_1(t) = 0$ for $t \geq t_1 = t_0 + 2\frac{\sqrt{V_1(t_1)}}{\lambda}$.

We can conclude that $V_T = 0$ in finite time t_1 and $\dot{V}_T = 0$ and at $t > t_1$, the system energy is reduced to cycle energy (24): $V = V_{LC} \Rightarrow \dot{V} = \dot{z} k_{22} (z_d - z)$

Proceeding in a same manner, we consider the Lyapunov function candidate (for $t > t_1$):

$$V_2 = \frac{1}{2} V_T^2 + \frac{1}{2} (V_{LC} - V_{LC}^*)^2 \quad (31)$$

$$\dot{V}_2 = (V_{LC} - V_{LC}^*) \dot{V}_{LC} \text{ at } t > t_1. \quad (32)$$

substituting (28) in \dot{V}_{LC} and using the control function (27), we can write:

$$\dot{V}_2 = -\Lambda (V_{LC} - V_{LC}^*) \psi(V_{LC} - V_{LC}^*) \dot{z}^2 \leq 0 \text{ at } t > t_1. \quad (33)$$

$\dot{V}_2 \leq 0$ leads to the periodic orbit stabilization in the same way as described before. We then conclude that transverse energy converge to zero in finite time and that the limit cycle defined by the invariant orbit - o is asymptotically stable. Thus the system energy converge to the reference one and achieve a controlled limit cycle.

D. Experimental results

In this section, we present 3 experimental results achieved with the SAP robot. The Figure (9), illustrates an equilibrium posture and define the following angular relation: $q_o + q_1 + \frac{q_2}{2} + \theta = \pi$ and vertical body's position $z = r_o + l_o \cos(q_o) - l \sin(q_o)$. An appropriate choice of posture allow us to consider q_o and θ small and then $z \simeq r_o + l_o + l \left(q_1 + \frac{q_2}{2} - \pi \right)$. Therefore we can deduce desired angulars q_1^d and q_2^d from the desired trajectories $z_d = l \left(q_1^d + \frac{q_2^d}{2} \right)$. This is used to implement the control functions. The control input is $\tau = k_p (q^d - q) - k_v \dot{q}$, the control function $z_d = \frac{\Gamma_2}{k_{22}} \psi(V_{LC} - V_{LC}^*) \dot{z}$. As energy function, we use $V_{LC} = \frac{1}{2} m \dot{z}^2 + mgz$ and a reference energy

value $V_{LC}^* = mgz_{\max}$. Note that SAP robot has a potentiometer and an encoder to measure angular positions q_1 and q_2 and gyroscope to measure the velocity of the body \dot{z} . The desired height of jumps is $z_{\max} = 0.03m$. The control gains are $k_p = 0.7$, $k_{v1} = 0.124$, for angular position q_1 , $k_p = 0.7$, $k_{v2} = 0.091$ for q_2 and $\Gamma_2 = 5$ for control function u_z and $\psi \pm 0.2$. We introduce a saturation to avoid geometric model singularities (to limit angle variation of α (Figure (9))).

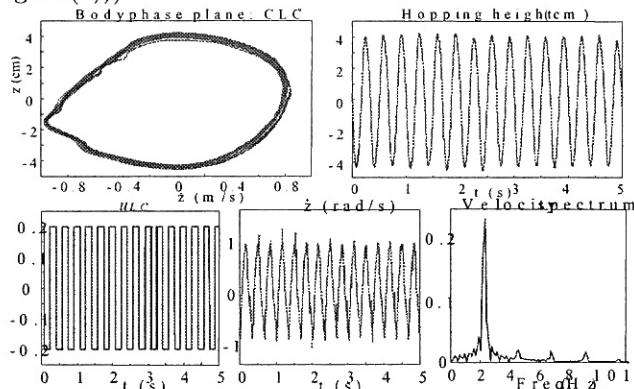


Fig. 10. Limit Cycles (Expérience n°1)

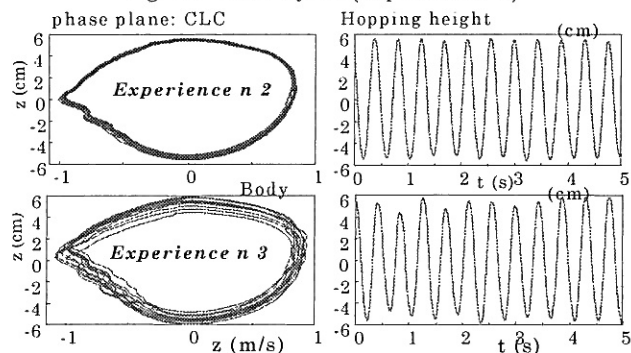


Fig. 11. Controlled Limit Cycles (Expérience N2 and N3)

The first results (Figure (10)) show a stabilized limit cycle. The cycle is distorted when the leg strikes the ground. This corresponds to the position $z = 1cm$. The body's jump height is $4cm$ and is stabilized around the nominal position $z_0 = 0$. We consider as hopping range, the difference between the maximum height and the contact point, then it is $5cm$. This error or inaccuracy is caused by difficulties to calibrate sensors and to deduce the absolute position z . The motion period is 3 jump per second. Figure (10) shows the applied control function CLC, the hopping velocity and the hopping frequency. In the second experience we have changed the robot inertia by adding, on robot body, a mass equivalent to the leg mass. Figure (11) shows a stable limit cycle. The hopping height is $5cm$ and the motion period is $0.442sec$. We can conclude that the additional mass changes only system frequency and the cycle becomes more stable with thin orbit (the coupling is less important). Thus control approach is robust versus parameters variation. Figure (11) illustrates the third experience. We have changed initial conditions (position & velocity) and the measurements show the transient. The body's height (Figure (11)) shows that cycle is set up after a time period equal to $3.6sec$. This period of time depends

on control gains and the saturation value ψ . After that, the jumping height becomes stable and equal to $5.8cm$. Then the hopping range is $5cm$.

IV. CONCLUSION

We have presented a new control strategy for legged robots with fast gaits. It exploits energetic aspects and passivity properties of system. The control methodology stabilizes a periodic hopping cycle by regulation of system energy to a reference value. For more than 1 DOF systems, the control approach decomposes the system energy, for hopping motion, in two terms. One of these two terms describes the transverse dynamics. The second one describes the limit cycle to stabilize. Thus a robust control structure was proposed. It depends on system number of degrees of freedom VCS (Variable Control Structure), dissipates transverse energy in finite time and stabilizes cycle energy to a reference value and therefore a Controlled Limit Cycle is obtained. The method was successful in simulation for 1DOF and 2 DOF systems and in experimentation for SAP robot with 3DOF. It is robust for large parameters variation and initial conditions variation and disturbances. The control approach allow us to stabilize a gait, an implicit trajectory generation and energy optimization by controlling system energy. It is easy to implement.

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