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Analysis of the Dynamic Behaviour of Magnetic Gear with Nonlinear Modelling for Large Wind Turbines

M. Desvaux, R. Le Goff Latimier, B. Multon, S. Sire, and H. Ben Ahmed

Abstract – This article deals with the dynamic behaviour of magnetic gear. This study is based on a nonlinear analytical model of magnetic gear which gives an analytical expression of the magnetic torque for a step disturbance. From this expression, various criteria will be defined in order to reach a good performance of the magnetic gear with step or sinusoidal disturbance. These results will be compared with a nonlinear simulation. Simulations show that it is possible to have, in a transitory regime, a load angle higher than the limit load angle maintaining a coupling between the low speed and the high speed rotor. Simulations will illustrate the importance of defining design rules based on the system application domain. A high power wind turbine (MW) example is proposed. In that case studies, loads generated by wind induce strong disturbances.

Index Terms—Damping factor, disturbance, dynamic, high torque application, load angle, magnetic gear, nonlinear simulation, operating range, transfer function, wind turbine.

I. NOMENCLATURE

T_G : Maximum Gear torque [Nm]: $T_G = \max(T_l)$
 T_h : High speed rotor magnetic torque [Nm]
 T_{in} : Input torque [Nm]
 T_l : Low speed rotor magnetic torque [Nm]
 T_{l0} : Low speed magnetic torque after step disturbance [Nm]
 T_{nom} : Nominal torque of the wind turbine [Nm]
 T_{out} : Output torque [Nm]
 D_{ext} : External diameter of magnetic gear [m]
 f_{dist} : Frequency of sinusoidal disturbance [Hz]
 f_h : Fluid friction of the high speed rotor [Nm/rad/s]
 f_l : Fluid friction of the low speed rotor [Nm/rad/s]
 G_r : Gear ratio
 J_h : Inertia of high speed rotor [kg.m²]
 J_l : Inertia of low speed rotor [kg.m²]
 k : Ratio between nominal torque and maximum gear torque
 N_h : Speed of high speed rotor [rpm]
 N_l : Speed of low speed rotor [rpm]
 n_s : Number of ferromagnetic pole pieces
 OS_φ : Overshoot of the load angle for a step disturbance [%]
 p_h : Number of magnetic pole pairs of high speed rotor
 p_l : Number of magnetic pole pairs of low speed rotor

ΔT : Amplitude of torque step or sinusoidal disturbance [Nm]
 ΔT_l : Torque overshoot on the low speed rotor [Nm]
 ω_{oh} : High speed rotor undamped resonance frequency [rad/s]
 ω_{ol} : Low speed rotor undamped resonance frequency [rad/s]
 ω_h : Speed of high speed rotor [rad/s]
 ω_l : Speed of low speed rotor [rad/s]
 φ : Load angle [rad]: $\varphi = p_l \cdot \theta_l + p_h \cdot \theta_h$
 φ_0 : Final value of load angle (after step) [rad]
 φ_i : Initial value of load angle (before step) [rad]
 θ_h : Angle of high speed rotor [rad]
 θ_{h0} : Angle of high speed rotor after step [rad]
 $\theta_{h,i}$: Angle of high speed rotor before step [rad]
 θ_l : Angle of low speed rotor [rad]
 θ_{l0} : Angle of low speed rotor after step [rad]
 $\theta_{l,i}$: Angle of low speed rotor before step [rad]
 ξ_h : Damping factor of high speed rotor
 ξ_l : Damping factor of low speed rotor

II. INTRODUCTION

FOR wind turbine applications, mechanical gearboxes are frequently used in indirect drive (Fig.1a); this design allows a lower investment cost and a lower weight compared to direct drive designs (Fig.1b). However, failures of mechanical gearboxes and their induced production interruptions, which increase operating costs, are more frequently observed with indirect drive design [1].

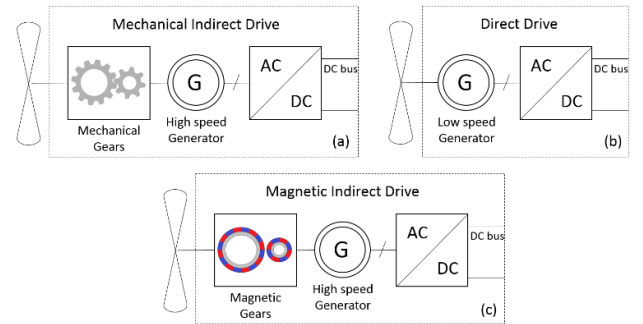


Fig. 1. Wind power conversion chains. (a) Mechanical indirect drive, (b) direct drive, and (c) magnetic indirect drive

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Analysis Of The Dynamic Behaviour of Magnetic Gear with Nonlinear Modelling

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An interesting intermediate solution is to use a generator and combine it with magnetic gear (Fig.1c) to obtain a magnetic indirect drive. An attractive topology of magnetic gear (Fig.2) has been proposed by [2] in the early 2000s and used in different studies [3]-[5]. This magnetic gear includes a high speed rotor, a low speed rotor and fixed ferromagnetic pole pieces. This topology of magnetic gear potentially offers high performance and gives a hope of competitiveness compared to mechanical solutions [6] even for high torque applications (off-shore wind turbine for instance) [7]-[10]. In this article, only the case of fixed ferromagnetic pole pieces will be studied.

Their competitiveness must be evaluated in wind power context where the system is subjected to strong disturbances that can generate mechanical problems like resonance or loss of control of the magnetic gear after an overload. The uncoupling of the magnetic gear between two rotors due to overload is often described as an advantage (overload protection) but in the context of wind power, it can be dangerous. Indeed, when uncoupling occurs between rotors of the magnetic gear, the low speed rotor and the blades of the wind turbine are no longer subjected to torque breaking ($\bar{T}_l = \bar{T}_h = 0$, where \bar{T}_l and \bar{T}_h are the average value of the low and the high speed rotor respectively) and can go faster and faster until damaging the wind power conversion chains. Some strategies have been proposed to counteract the uncoupling between rotors of the magnetic gear [11]-[12]. These strategies are based on servo control of magnetic gear but these solutions are not adapted to wind application due to the difficulty of the input torque pitch control.

This paper, therefore, aims at providing an analysis of the intrinsic dynamic characteristics of magnetic gear. It proposes some design criteria to make sizing relevant for its application domain without servo control. These criteria are described using analytical analysis and nonlinear simulations. This paper illustrates the importance of defining design rules based on the application domain. An example of application domain, applied to high power wind turbine, is proposed.

III. NONLINEAR MAGNETO-MECHANICAL MODELLING OF MAGNETIC GEAR

Magnetic gear [2] is composed of three rings:

- A high speed rotor ring with p_h pole pairs of permanent magnets and a ferromagnetic yoke,
- A low speed rotor ring with p_l pole pairs of permanent magnets and a ferromagnetic yoke,
- A fixed ferromagnetic ring with n_s pole pieces between the two rotors (an example is given in Fig. 2 with $p_h = 2, p_l = 7, n_s = 9$).

Each rotor generates in airgaps a wave of magneto-motive force. Ferromagnetic pole pieces aim to modulate the magnetic field in the two airgaps to obtain a common harmonic. The result is a magnetic torque with an average different from zero.

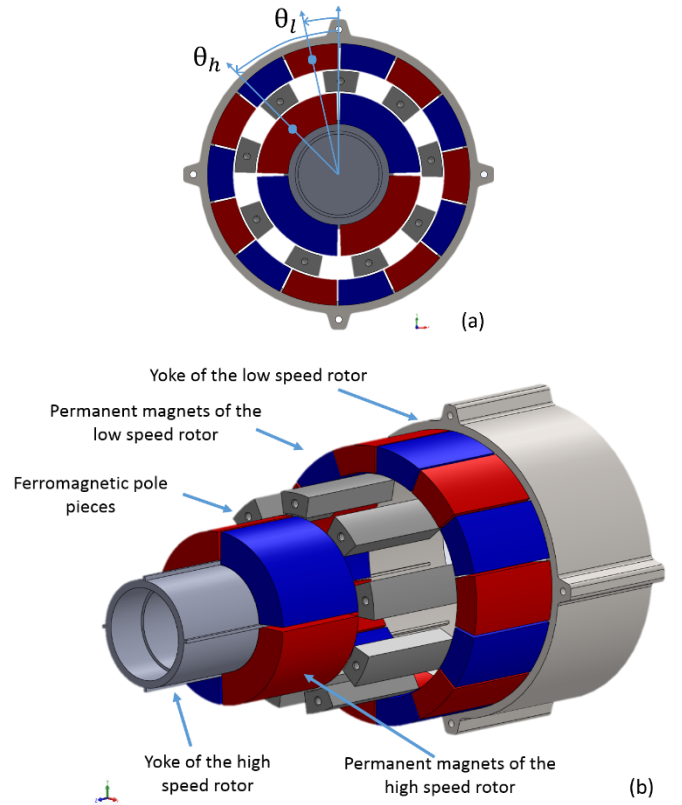


Fig. 2. Magnetic part of the magnetic gear (a) in side drawing (b) in exploded drawing (in this example $p_h = 2, p_l = 7, n_s = 9$)

The three rings parameters must follow the relation (1) and the gear ratio is given by (2) [3]. According to the studied configuration (fixed ferromagnetic pole pieces), the rotational speeds of the two rotors are opposite to each other (the gear ratio is negative). The coupling between the two rotors results in a magnetic torque whose expression is given by (3) with T_G which depends to geometry and materials properties and without torque modulation for the two rotors:

$$p_l + p_h = n_s \quad (1)$$

$$G_m = -\frac{p_l}{p_h} \quad (2)$$

$$\begin{cases} T_l = T_G \cdot \sin(p_l \cdot \theta_l + p_h \cdot \theta_h) \\ T_h = \frac{T_G}{G_m} \sin(p_l \cdot \theta_l + p_h \cdot \theta_h) \end{cases} \quad (3)$$

On the basis of this coupling, a block diagram of the magnetic gear can be proposed in Fig. 3:

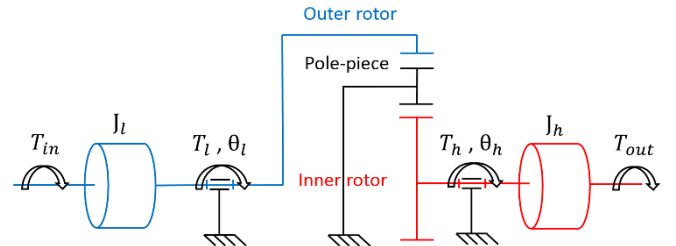


Fig. 3. Block diagram of the magnetic gear

The dynamic equations are then deduced for the two rotors as reported in equations (4) and (5) in agreement with [12]. This dynamic equation has been solved in [12] with a linearization of the magnetic torque around a single position of the load angle. The scope of the present article is to carry out a modelling which takes into account the non-linearity of the magnetic torque.

$$\begin{cases} J_l \frac{d^2\theta_l}{dt^2} = T_{in} - T_l - f_l \frac{d\theta_l}{dt} \\ J_h \frac{d^2\theta_h}{dt^2} = T_h - T_{out} - f_h \frac{d\theta_h}{dt} \end{cases} \quad (4)$$

$$\begin{cases} J_l \frac{d^2\theta_l}{dt^2} = T_{in} - T_G \cdot \sin(p_l \cdot \theta_l + p_h \cdot \theta_h) - f_l \frac{d\theta_l}{dt} \\ J_h \frac{d^2\theta_h}{dt^2} = \frac{T_G}{G_m} \sin(p_l \cdot \theta_l + p_h \cdot \theta_h) - T_{out} - f_h \frac{d\theta_h}{dt} \end{cases} \quad (5)$$

IV. ANALYTICAL EXPRESSION OF THE EVOLUTION OF THE MAGNETIC TORQUE FOR A STEP DISTURBANCE

This section is dedicated to the analytical expression of the evolution of the magnetic torque for a step disturbance. This step can model, in a wind power context, a sudden increase of the wind, resulting in an increased ΔT . The evolution of the torque will be analysed when the disturbance is applied to the low speed rotor (like in a wind turbine context) but can be transposed to the other case (disturbance on the high speed rotor). It is possible to describe two expressions of the evolution of the magnetic torque in function of the damping factor (ξ_l). They correspond to the case where ξ_l is less than one and to the case where ξ_l is higher than one.

A. Transfer function of the magnetic gear with load angle Taylor expansion

In order to obtain an analytical expression of the dynamic behaviour of the system from (5), a Taylor expansion of the load angle is performed at the first order around all positions taken by the load angle. Equations (6) correspond to the Taylor expansion around φ_0 . From the Laplace transform, it is possible to obtain the transfer function $H_l(s)$ and $H_h(s)$ available around φ_0 in (7) and identify the damping factor and the undamped resonance frequency for the two rotors, as shown in (8) and (9):

$$\begin{cases} J_l \frac{d^2\theta_l}{dt^2} = T_{in} - f_l \frac{d\theta_l}{dt} - T_G [\sin(\varphi_0) + \cos(\varphi_0)(p_l \theta_l + p_h \theta_h - \varphi_0)] \\ J_h \frac{d^2\theta_h}{dt^2} = -T_{out} - f_h \frac{d\theta_h}{dt} + \frac{T_G}{G_m} [\sin(\varphi_0) + \cos(\varphi_0)(p_l \theta_l + p_h \theta_h - \varphi_0)] \end{cases} \quad (6)$$

$$\begin{cases} \frac{\theta_l(s)}{T_{in}(s)} = H_l(p) = \frac{1}{J_l \cdot s^2 + f_l \cdot s + p_l \cdot T_G \cos(\varphi_0)} \\ \frac{\theta_h(s)}{T_{out}(s)} = H_h(p) = \frac{1}{J_h \cdot s^2 + f_h \cdot s - \frac{T_G}{G_m} \cdot p_h \cos(\varphi_0)} \end{cases} \quad (7)$$

$$\begin{cases} \omega_{0l} = \sqrt{\frac{p_l \cdot T_G \cos(\varphi_0)}{J_l}} \\ \xi_l = \frac{f_l}{2} \sqrt{\frac{1}{J_l \cdot p_l \cdot T_G \cos(\varphi_0)}} \end{cases} \quad (8)$$

$$\begin{cases} \omega_{0h} = \sqrt{\frac{p_h \cdot T_G \cos(\varphi_0)}{J_h \cdot |G_m|}} \\ \xi_h = \frac{f_h}{2} \sqrt{\frac{|G_m|}{J_h \cdot p_h \cdot T_G \cos(\varphi_0)}} \end{cases} \quad (9)$$

B. Case where the damping factor is less than one

In this case, the evolution of the position of low speed rotor (θ_l) valid for a small evolution of the load angle can be described by equation (10):

$$\theta_l(t) = \theta_{l_i} + \Delta T \left(1 - \frac{e^{-\xi_l \omega_{0l} t}}{\sqrt{1 - \xi_l^2}} \sin \left(\omega_{0l} t \sqrt{1 - \xi_l^2} + \cos^{-1}(\xi_l) \right) \right) \quad (10)$$

It is possible to consider that the high speed rotor is not influenced by a small step disturbance. With this hypothesis, the load angle evolution is proportional to the low speed rotor angle (this hypothesis will be confirmed thereafter by simulation). The evolution of the magnetic torque is given by (11). To determine the maximal amplitude of the torque wave, the evaluation of the overshoot of the load angle is necessary. It is given in % by (12). The maximal amplitude of the torque wave can be then obtained by (13).

$$T_l(t) = T_G \cdot \sin(p_l \cdot \theta_l(t) + p_h \cdot \theta_{h_i}) \quad (11)$$

$$OS_\varphi = 100 \cdot e^{\frac{-\pi \xi_l}{\sqrt{1 - \xi_l^2}}} \quad (12)$$

$$\Delta T_l = T_G \sin(\varphi_i + (\varphi_0 - \varphi_i) OS_\varphi) - T_G \sin(\varphi_i) \quad (13)$$

Even if this expression is available for small disturbance, it represents correctly the behaviour of the magnetic gear for bigger disturbance (this result is confirmed by simulation). Moreover, this expression highlights that a disturbance has a direct consequence on the magnetic torque and can produce an uncoupling between the two rotors.

C. Case where the damping factor is higher than one

In the case where $\xi_l \geq 1$, the evolution of the position of low speed rotor (θ_l) can be described by equation (14). With the same hypothesis as in the last case, the new expression of the low speed rotor torque can be proposed in (15):

$$\begin{cases} \theta_l(t) = \theta_{l_i} + \Delta T \left(1 + \frac{1}{\tau_2 - \tau_1} (\tau_1 e^{-t/\tau_1} - \tau_2 e^{-t/\tau_2}) \right) \\ \tau_1 = \omega_{ol} \left(-\xi_l - \sqrt{\xi_l^2 - 1} \right) \\ \tau_2 = \omega_{ol} \left(-\xi_l + \sqrt{\xi_l^2 - 1} \right) \end{cases} \quad (14)$$

$$T_l(t) = T_G \cdot \sin(p_l \cdot \theta_l(t) + p_h \cdot \theta_{h_i}) \quad (15)$$

In that case, the magnetic torque will not overshoot the amplitude of the disturbance. Considering the studied system, this result makes the system more robust to disturbances. Indeed, if there is no overshoot, it will be more difficult for the magnetic torque to reach the maximum gear torque value and create uncoupling between the two rotors.

D. Conclusion for the two cases

The present study highlights the different types of dynamic behaviour when magnetic gear is submitted to a step disturbance. It appears that in order to avoid an excessive magnetic torque peak (which could cause an overload in function of the safety margin), the case where the damping factor is higher than one is preferred.

To have a magnetic gear with a high torque density and a high efficiency, it is similar to minimize the damping factor (according to equation (8)) and the magnetic gear is likely to be in the case where the damping factor is less than one. To avoid any problem like uncoupling between rotors in this case, it will then transfer power with a nominal torque lower than the maximum gear torque to compensate the disturbance.

Considering the nonlinear equation to be solved in order to know the evolution of the magnetic torque for large displacement (equation (5)), simulations are performed to describe more precisely the magnetic gear behaviour and validate the previous hypothesis.

In the next section of this article, the dynamic study of a magnetic gear designed for wind power 3.9 MW, 15 rpm (corresponding to the case where $\xi_l \geq 1$) will be presented. It is possible to carry out the same study for the other case ($\xi_l \leq 1$) with the same protocol as the one presented below.

V. SIMULATIONS OF THE MAGNETIC GEAR BEHAVIOUR WITH THE NONLINEAR MODELS

In order to validate the analytical expressions established previously, it is possible to simulate the nonlinear behaviour of the magnetic gear. To do this, a block diagram is proposed in Fig. 6 where the input torque is imposed and the output speed is imposed. The simulation is performed for a magnetic gear designed for a wind turbine 3.9 MW, 15 rpm. Magnetic gear is obtained with an optimization described in [13] and dynamic parameters are defined in Table I.

In this part, the dynamic behaviour of the magnetic gear is evaluated with a step torque disturbance, a sinusoidal torque disturbance and a representative wind turbine torque disturbance.

TABLE I
PARAMETERS OF THE MAGNETIC GEAR

Symbol	Quantity	Value
P_{nom}	Nominal power	3.9 MW
N_l	Speed of high speed rotor	15 rpm
T_{nom}	Nominal torque	2.5 MNm
T_G	Maximum gear torque	2.7 MNm
G_r	Gear ratio (absolute value)	6.5
p_h	Number of magnetic pole pairs of high speed rotor	20
p_l	Number of magnetic pole pairs of low speed rotor	130
D_{ext}	External diameter	4 m
J_h	Inertia of high speed rotor	$1.0 \cdot 10^5 \text{ kg.m}^2$
J_l	Inertia of low speed rotor	$1.2 \cdot 10^5 \text{ kg.m}^2$
f_h	Fluid friction of the high speed rotor	$3.7 \cdot 10^2 \text{ Nm/rad/s}$
f_l	Fluid friction of the low speed rotor	$1.6 \cdot 10^4 \text{ Nm/rad/s}$

A. Behaviour of the magnetic gear with a step torque disturbance.

For the first part of the study, the behaviour of the magnetic gear is evaluated with a step torque disturbance (16) and a maximum gear torque equal to 2.7 MN.m. This disturbance happens when the magnetic gear is stabilized in its nominal working point. Fig. 4 and Fig. 5 show the evolution of the magnetic torque on low speed rotor for the magnetic gear described in Table I and the step disturbance described above.

$$\Delta T = \frac{3 \cdot T_G}{100} \quad (16)$$

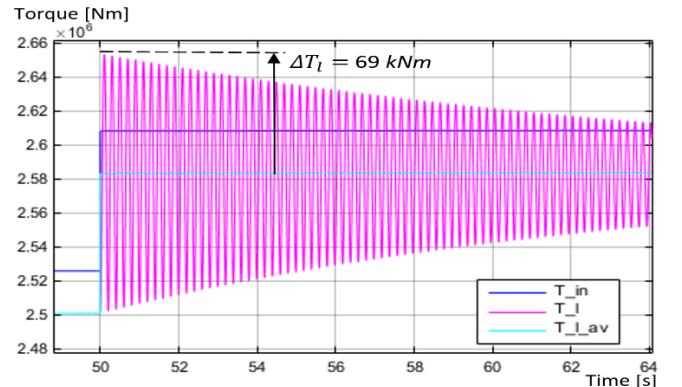


Fig.4 Evolution of low speed rotor torque with the step disturbance (16) and with a torque overshoot on the low speed rotor equal to 69 kN.m.

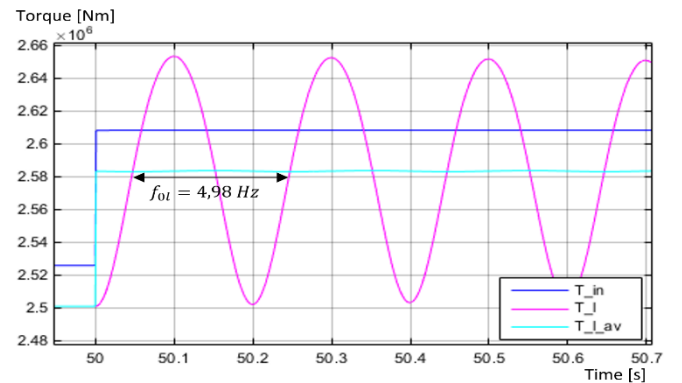


Fig.5 Determination of the undamped resonance frequency of low speed rotor (4.98 Hz) corresponding to the step disturbance (16).

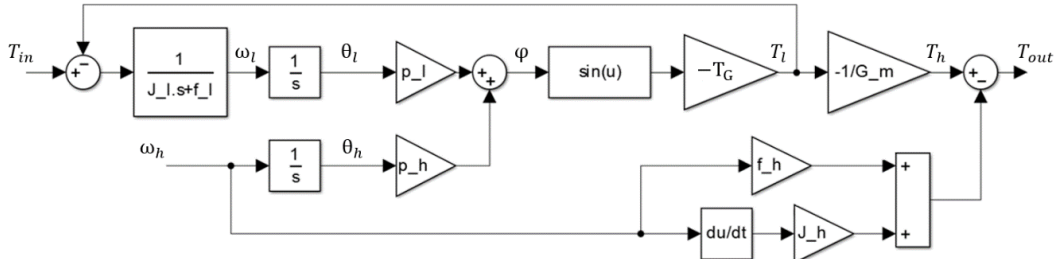


Fig.6 Block diagram of the magnetic gear in the nonlinear simulations

For the same parameters, the analytical expression gives results described in (17). Even if the analytical values are obtained with a Taylor expansion, they are very close to those obtained with nonlinear simulation. Such result validates the analytical expression.

$$\begin{cases} f_{ol} = 4,85 \text{ Hz} \\ \Delta T_l = 64 \text{ kNm} \end{cases} \quad (17)$$

For bigger step torque disturbance, analytical expression gives results close to results obtained with simulations. Then, a comparison can be made for different torque step disturbances with an initial torque equal to $0.8 \cdot T_G$ (2.3 MN.m). For bigger step torque disturbance, analytical expression gives expected results and can describe correctly the behaviour of the magnetic gear as shown in Table II.

TABLE II

RESULTS OF SIMULATIONS AND NUMERICAL VALUES FOR DIFFERENT STEP DISTURBANCE WITH AN INITIAL TORQUE EQUAL TO 2.3 MNM

ΔT	Simulation (torque in kN.m and frequency in Hz)	Analytical expression (torque in kN.m and frequency in Hz)
$3\% \cdot T_G$	$\Delta T_l = 81 \quad f_{ol} = 6.77$	$\Delta T_l = 80 \quad f_{ol} = 6.64$
$5\% \cdot T_G$	$\Delta T_l = 131 \quad f_{ol} = 6.58$	$\Delta T_l = 125 \quad f_{ol} = 6.46$
$8\% \cdot T_G$	$\Delta T_l = 188 \quad f_{ol} = 6.25$	$\Delta T_l = 175 \quad f_{ol} = 6.15$
$12\% \cdot T_G$	$\Delta T_l = 227 \quad f_{ol} = 5.56$	$\Delta T_l = 193 \quad f_{ol} = 5.59$

B. Stability of the magnetic gear

Following different simulations of the behaviour of the magnetic gear with step torque disturbance, it is observed that a load angle value higher than $\pi/2$ can be reached without uncoupling between the two rotors except during a transitory phase as shown in Fig. 7 and Fig. 8. Then, the operating range of a magnetic gear can be increased during the transitory phase or when a magnetic gear is installed in an application with disturbances.

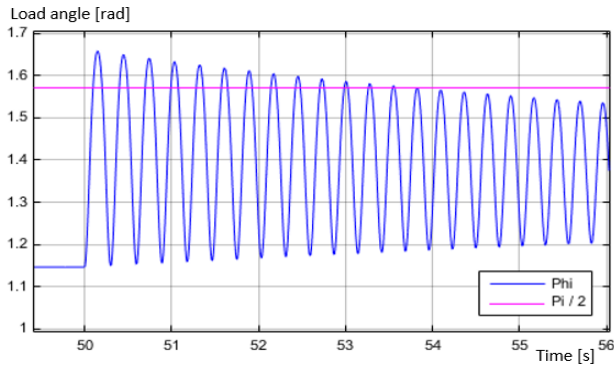


Fig.7 Evolution of the load angle with a step disturbance who generates a load angle value higher than $\pi/2$ without uncoupling between two rotors.

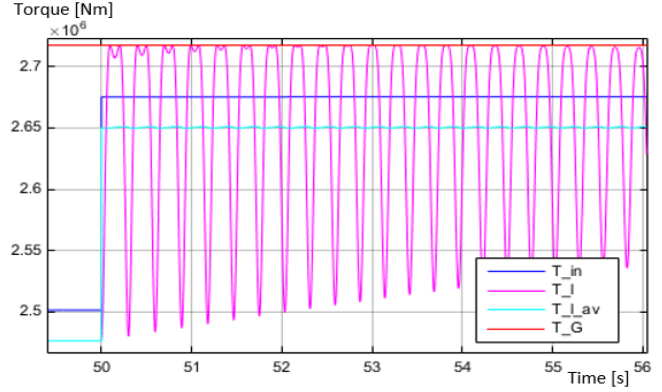


Fig.8 Evolution of the magnetic torque on a low speed rotor with a step disturbance which generates a load angle value higher than $\pi/2$ without uncoupling between two rotors ($T_G = 2.73 \text{ MN.m}$).

According to the result and for magnetic gear characterized in Table I with a step torque disturbance, it is possible to define dynamic limit of stability without uncoupling between the two rotors in function of the initial torque (before step torque disturbance) and ΔT , see Fig. 9.

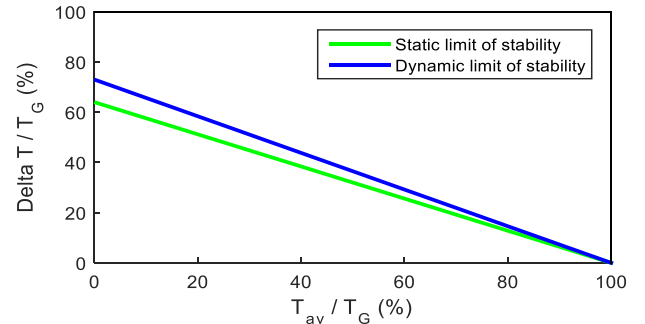


Fig.9 Dynamic and static limit of stability without uncoupling between rotors for a torque step disturbance ($T_G = 2.73 \text{ MN.m}$).

If the torque disturbance can be represented with a sinusoid instead of a step, operating range could be defined in function of the average torque T_{av} , the amplitude and the frequency of the sinusoid respectively ΔT and f_{dist} (Fig. 10-12) with an input torque proposed in (18). Fig. 10-12 illustrate that the operating range is reduced when the frequency of the sinusoid is close enough to the resonance frequency of the low speed rotor. Indeed, in regard of the equation (8), the resonance frequency of the low speed rotor depends on the average torque and evolves like (19).

$$T_{in}(t) = T_G \left(\frac{T_{av}}{T_G} + \frac{\Delta T}{T_G} \sin(2\pi \cdot f_{dist} \cdot t) \right) \quad (18)$$

$$\begin{cases} \frac{10.T_G}{100} \leq T_{av} \leq \frac{90.T_G}{100} \\ 8.6 \text{ Hz} \geq f_{ot} \geq 5.5 \text{ Hz} \end{cases} \quad (19)$$

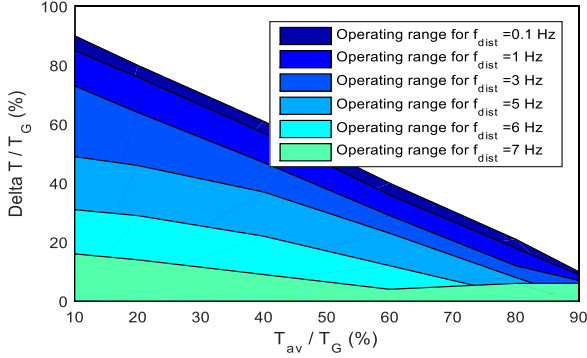


Fig.10 Operating range without uncoupling between rotors for a sinusoidal torque disturbance with frequency up to 7 Hz ($T_G = 2.73 \text{ MN.m}$).

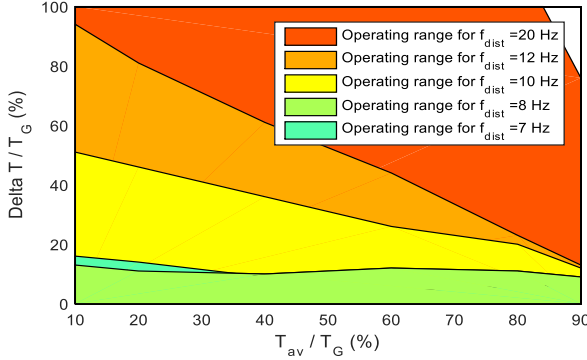


Fig.11 Operating range without uncoupling between rotors for a sinusoidal torque disturbance with frequency from 7 Hz to 20 Hz ($T_G = 2.73 \text{ MN.m}$).

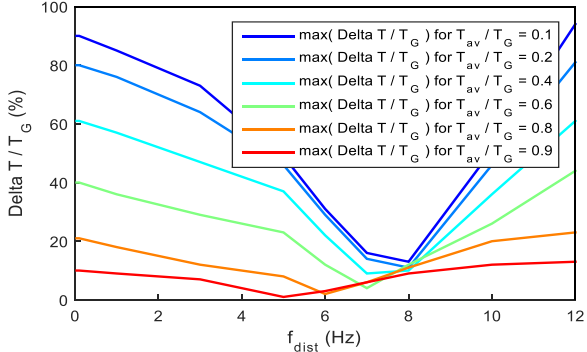


Fig.12 Maximum value of $\Delta T / T_G$ in function of the frequency of the disturbance without uncoupling between rotors for different value of T_{av} / T_G ($T_G = 2.73 \text{ MN.m}$).

C. Behaviour simulation of the magnetic gear with wind turbine disturbance

The previous results are independent of the application field. This section will focus on a wind turbine application with conversion chains proposed in Fig. 1c. Input torque of the magnetic gear follows (20):

$$T_{in}(t) = T_{nom} \left(1 + \frac{5}{100} \sin(3\theta_t(t)) + \frac{0.3}{100} \sin(0.10t) + \frac{3}{100} \sin(0.27t) + \frac{1}{100} \sin(1.29t) + \frac{0.3}{100} \sin(3.66t) \right) \quad (20)$$

In this expression, the first term corresponds to the nominal torque, the second term corresponds to the influence of

passage of the blades in front of the tower for a three blades wind turbine. The other terms correspond to typical wind fluctuations.

With a mechanical torque on the input of the magnetic gear following (20), it is not possible to have a nominal torque too close to the maximal gear torque. Also, to maintain a high torque density of the magnetic gear, the ratio between nominal torque and maximum gear torque (21) must be maximized. With this load, iterative simulations are performed; the maximum value of the ratio is $k = 92.5\%$. Fig.13-14 illustrate the evolution of the torque and the load angle with this value.

$$k = \frac{T_{nom}}{T_G} \quad (21)$$

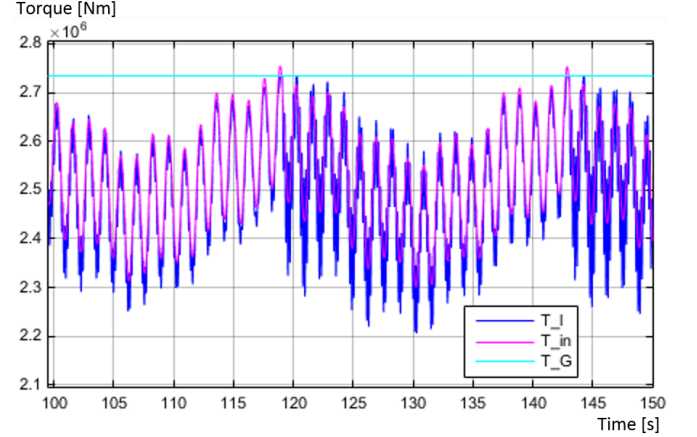


Fig.13 Evolution of the magnetic torque on low a speed rotor and the input torque of the magnetic gear with $k = 92.5\%$ and the load described in (20) ($T_G = 2.73 \text{ MN.m}$ and $T_{nom} = 2.5 \text{ MN.m}$).

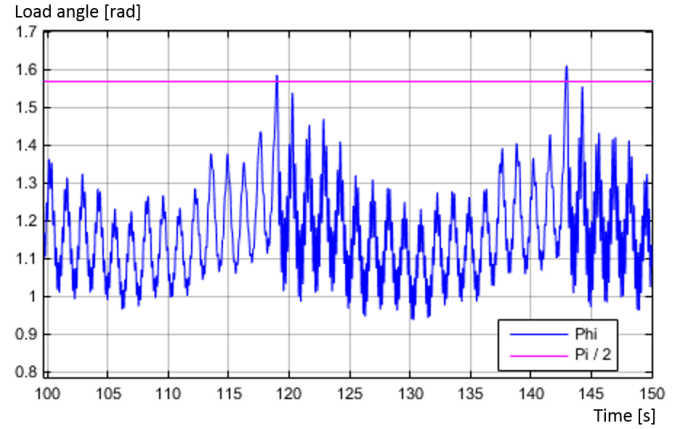


Fig.14 Evolution of the load angle of the magnetic gear with $k = 92.5\%$ and the load described in (20) ($T_G = 2.73 \text{ MN.m}$ and $T_{nom} = 2.5 \text{ MN.m}$).

VI. CONCLUSION

This article brings several contribution to the modelling of magnetic gear:

A. An analytical expression has been proposed for small torque step disturbance on magnetic gear. Because of the non-linearity of the equation, the analytical expression does not seem valid for bigger step disturbance. However, simulations have permitted to observe that the analytical expression describes with good agreement the magnetic gear behaviour for bigger disturbance. In function of the geometry of the magnetic gear, the damping factor can be small and the

overshoot torque expression can be simplified, as shown in equation (22):

$$\begin{cases} \theta_l(t) = \theta_{li} + \Delta T \left(1 - e^{-\xi_l \omega_{0l} t} \cdot \sin \left(\omega_{0l} t + \frac{\pi}{2} \right) \right) \\ T_l(t) = T_G \cdot \sin(p_l \cdot \theta_l(t) + p_h \cdot \theta_{hi}) \end{cases} \quad (22)$$

B. It is possible to define an operating range of magnetic gear in function of the average torque and the amplitude of the disturbance. For sinusoidal torque disturbance with the frequency lower than the resonance frequency of the low speed rotor, the operating range is similar than the operating range with a step disturbance. The operating range is strongly reduced for frequency similar than the resonance frequency of the low speed rotor as shown in Fig. 10. Given that the magnetic gear behavior is similar than a low pass filter, magnetic gear allow sinusoidal torque disturbance with frequency higher than there resonance frequency of the low speed rotor (Fig. 11).

C. Another important observation is the necessity to have a nominal torque away from the maximal gear torque in function of the application typical disturbance is planned to be integrated a magnetic gear. For wind power application, with the input torque described in equation (18), the maximal ratio between the nominal torque and the maximal gear torque is equal to 92.5%.

D. The last observation corresponds to load angle value higher than $\pi/2$ without uncoupling between the two rotors during a transitory phase. This result permits to increase the operating range of the magnetic gear when the magnetic gear is integrated in an application with lot of disturbances.

In order to improve the dynamic behaviour analysis of the magnetic gear, it could be interesting to develop an analytical model of the induced current loose. The damping factors will be different and the coupling between the two rotors of the magnetic gear could be more robust. On the other hand, the different dynamic criteria developed in this article will be used in a future mechatronic optimization of this magnetic gear.

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