

Experimental and Numerical Simulation of the Dynamic Frictional Contact between an Aircraft Tire Rubber and a Rough Surface

Iulian Rosu, Lama Elias-Birembaux, Frédéric Lebon, Hagen Lind, Matthias Wangenheim

► To cite this version:

Iulian Rosu, Lama Elias-Birembaux, Frédéric Lebon, Hagen Lind, Matthias Wangenheim. Experimental and Numerical Simulation of the Dynamic Frictional Contact between an Aircraft Tire Rubber and a Rough Surface. Lubricants, 2016, 4 (29), 10.3390/lubricants4030029. hal-01420862

HAL Id: hal-01420862 https://hal.science/hal-01420862

Submitted on 3 Feb 2018

HAL is a multi-disciplinary open access archive for the deposit and dissemination of scientific research documents, whether they are published or not. The documents may come from teaching and research institutions in France or abroad, or from public or private research centers. L'archive ouverte pluridisciplinaire **HAL**, est destinée au dépôt et à la diffusion de documents scientifiques de niveau recherche, publiés ou non, émanant des établissements d'enseignement et de recherche français ou étrangers, des laboratoires publics ou privés.



Article



Experimental and Numerical Simulation of the Dynamic Frictional Contact between an Aircraft Tire Rubber and a Rough Surface

Iulian Rosu ¹, Hélène L. Elias-Birembaux ^{1,*}, Frédéric Lebon ¹, Hagen Lind ² and Matthias Wangenheim ²

- ¹ LMA CNRS UPR 7051, Aix-Marseille University, Centrale Marseille, 4 impasse Nikola Tesla, CS40006, 13453 Marseille Cedex 13, France; rosu@lma.cnrs-mrs.fr (I.R.); lebon@lma.cnrs-mrs.fr (F.L.)
- ² Institute of Dynamics and Vibration Research, Leibniz University Hannover, Appelstr. 11, D-30167 Hannover, Germany; lind@ids.uni-hannover.de (H.L.); wingenheim@ids.uni-hannover.de (M.W.)
- * Correspondence: elias@lma.cnrs-mrs.fr; Tel.: +33-484-524-269

Academic Editor: James E. Krzanowski Received: 15 January 2016; Accepted: 21 July 2016; Published: 17 August 2016

Abstract: This paper presents a numerical simulation of an aircraft tire in contact with a rough surface using a variable friction coefficient dependent on temperature and contact pressure. A sliding facility was used in order to evaluate this dependence of the friction coefficient. The temperature diffusion throughout the tire cross-section was measured by means of thermocouples. Both frictional heating and temperature diffusion were compared to numerical two- and three- dimensional simulations. An adequate temperature prediction could be obtained. In future simulations, wear should be taken into account in order to have a more accurate simulation especially in the case of high pressures and slipping velocities. A 3D finite element model for a rolling tire at a velocity of 37.79 knots (19.44 m/s) and in a cornering phase was investigated using a variable friction coefficient dependent on temperature and pressure. The numerical simulation tended to predict the temperature of the tire tread after a few seconds of rolling in skidding position, the temperature of the contact zone increases to 140 °C. Further investigations must be carried out in order to obtain the evolution of the temperature observed experimentally. The authors would like to point out that for confidentiality reasons, certain numerical data could not be revealed.

Keywords: tire; modeling; finite element; friction coefficient; contact pressure; temperature

1. Introduction

Aircraft tires are subject to very high performance requirements when providing directional control and braking during the maneuvering of the airplane on the ground. The tires must not only support the aircraft's weight, and transmit acceleration and braking forces to the runway surface, but also supply the forces necessary for the directional control of the aircraft [1]. It is well known that the directional capacity of an aircraft is measured by the friction coefficient [2]; this is the measurement of the contact area forces as a fraction of the aircraft's weight. In the field of aeronautics, NASA carried out the first experimental and related studies [3–5] on the thermal aspect (influence of temperature and its distribution) of the evolution of the friction coefficient between the tire and the ground. Subsequently, many other studies also investigated the interaction of various phenomena in the contact area, with the friction coefficient [6–8]. As friction is the major factor for generating forces on the aircraft through the tire, it is extremely important to have an accurate characterization of the magnitude and direction of the friction and force generated at the ground/tire interface. Generally, the normalized tire friction coefficient $\mu = F_y/F_z$ is used in most tire friction models, where F_y is the tangential friction force at the tire/runway interface and F_z is the vertical loading on the tire (Figure 1). Thus, μ is the sliding friction coefficient,

1). Thus, μ is the sliding friction coefficient, this being a nonlinear function depending principally on

the velocity of the tire. Grosch [9] studied different vulcanized rubber tires and described the this being a nonlinear function depending principally on the velocity of the tire. Grosch [9] studied dependence of the friction coefficient on sliding velocity, temperature and surface roughness. He different vulcanized rubber tires and described the dependence of the friction coefficient on sliding velocity, temperature and surface roughness. He explained that friction is due to energy dissipation when rubber is compressed and released by velocity, temperature and surface roughness. The explained that friction is due to energy dissipation when rubber is compressed and released by velocity, temperature and surface roughness. The explained that friction coefficient also depends on several airplane factors (loading, velocity, slip angle) [10–12], tire factors (tire type, inflation pressure, tread design, rubber composition) several airplane factors (loading, velocity, slip angle) [10–12], tire factors (tire type, inflation pressure, tread design, rubber composition) [13], runway surface conditions and contamination) [15, 16].



Figure 4. Forces and moments definitions on a tire.

Persson studied it revealed to the tentern trate of read which and called the unlaced inverse in contact HZ-developed a schere influence of the positive application of the positive application of the positive positi evictive avefilier rashed in the harmonic of the state of slisting history (methods), which is chick is convirable an a description of million friction bor studing oliding corrologi dubugh substrato the bisence depondence of other instiany is una intraduate kristingah hading vertex with the constance grow was a way in the standard of the second standard in the second different magnitudan windun and children generation of the second state of the second disconlations at comparison and the second sinteenvoluscolina the piece lasticup per a the solution of th alopessalento itrist a decase ty catized user the alacelized case pointer universas a in the low stating trease they shiding peloaities it here as per rate in the construction of the second s or otor order, abchilzed/holarahized deratives depta indeate play a scale horres or that in a sepiral camp that any paratase issues a result rive sector reason of fruth an with inner the ingested ing station of the results in the results of th (if the sticing velocity is above 0.01 m/s).

In precedent papers [20-22], FEM models are proposed to simulate the aircraft tire behavior in contact with the unwaver and goted the therme chan had level price at according to the contact with the unwaver and goted the therme changes in a level of the contact with the contact we have a set of the contact with the contact we have a set of the contact with the contact we have a set of the contact w initional is achieved and the total function of the total function of the total of the termination of termin coefficient depending on the sliding velocity, contact pressure and temperature. The present paper presents the numerical simulation of an aircraft tive in contact with a rough surface, using a variable friction coefficient dependent on temperature and contact pressure. Firstly, a slidling facility with rubber block samples under sliding conditions is used, under different fixed vertical pressures, with a constant sliding velocity. The friction coefficient was determined and the thermal evolution of the rubber samples discussed. Secondly, experimental tests were carried out in order to simulate the frictional heating phenomenon developed in the tire tread, using an external heat source. In these studies, experimental parameters corresponding to the real parameters of an aircraft tire in rolling conditions (vertical laadings verterity in instruction research and identical agency and be a series of a second ander transformation and enverted real action between constructions and in the second and the second and the second s Finally-cuthersicial ansignal chains a content of a content of the m/s) in a cornering mode was investigated using a variable friction coefficient dependent on a cornering mode was investigated using a variable friction coefficient dependent on temperature and pressure. The thermal evolution of the tire rubber during this rolling phase was compared to experimental data.

Rubber Block Frictional Heat

Tread rubber characteristics play a crucial role in tire force generation; they describe tread stiffness, friction properties and heat generation. Energy loss and modulus vary, not only from one compound to another, but also depending on stress frequency and on temperature. Tread rigidity is directly related to the rubber shear modulus G", as rubber becomes softer when its bulk temperature increases; tread rigidity will depend on the tire temperature increase. Higher cornering stiffness is obtained at a lower tire tread temperature [23]. Fevrier et al. demonstrate the mechanisms involved in the relative slippage between the tire tread and runway. The first mechanism is the frequency excitation of the material by the runway surface texture. The rubber is distorted when it comes into contact with rough spots on the runway, thus leading to energy loss (hysteresis). The second mechanism is molecular adhesion. As regards the energy balance, both hysteresis and adhesion result from the transformation of mechanical energy in the contact zone. Thus, friction is not only a mechanical process but is also related to the thermal evolution of the rubber, which means that a thermo-mechanical contact model is essential for a reliable description of the frictional behavior.

Kainradl et al. [24] also studied the thermal evolution of the tire tread in a rolling phase. They demonstrate that the tire tread temperature rise is due to frictional heating, which does not significantly affect the temperature profiles within the tire carcass structure. However, these tread temperature peaks probably have a significant influence on tread wear rates. It is well known that the maximum friction level developed by a worn tire is shown to be more sensitive to variations in tread temperature and velocity. It was observed that a tire in cornering or braking conditions at a high angle produced molten rubber deposits on the runway surface and tread temperatures high enough to cause the rubber to undergo changes in its chemical composition [3]. For all these reasons, a profound study of tread rubber in skidding positions is primordial, in order to establish a dynamic friction coefficient dependence on contact pressure, sliding velocity and temperature. In the literature, few experiments have been realized to determine the temperature rise in the contact area between an aircraft tire rubber and a rigid, rough runway [4,5].

2. Experimental and Numerical Studies: Materials and Methods

2.1. Experimental Study: Frictional Heating

The aim of this study is to identify tire tread behavior in a sliding situation. It was therefore important to design an experiment that could provide necessary data for comparison with numerical simulation. The experimental investigations were performed on the High Speed Linear Friction Tester (HiLiTe), (Figure 2a) at the Institute of Dynamics and Vibration Research (IDS) at Leibniz Universität Hannover, Germany. With this test rig, sliding friction and sticktion measurements can be performed over a wide velocity range (0.1–10 m/s) and with vertical forces up to 1 kN. The ambient temperature can be controlled between -20 °C and +55 °C. Typically, the sliding friction of tire tread samples is tested on several surfaces, such as asphalt or concrete test tracks, and also on snow and ice. Experiments at positive ambient temperatures can be carried out in dry, damp or wet conditions. Whilst friction and vertical force are measured with standard equipment, additional test equipment must be used to measure the sample temperature and contact temperature. The experiments for this study were performed on a concrete test track, the rubber tread samples measured 20×20 mm on the contact area and 10 mm in thickness. The sample was bonded onto the sample mounting plate and fitted with a thermocouple 6 mm from the contact area, which was implanted into a borehole and thermally connected by heat conductive paste, Figure 2b. The velocity profile is trapezoidal; it is formed in a short acceleration phase, then a phase of constant velocity v = 4 m/s followed by a short

deceleration phase. The overall track length is 2 m (Figure 2c). The experiments were realized at $20^{\circ}C$. As well as the measurement of friction force, normal force, and the sample core temperature $_{14}^{\circ}$ the temperature in the contact area was measured with an infrared camera, at the end of the track, see Figure 2b. Simultaneously, a high speed concertain was used to record attachimage of the deformed block while bidek while



FFigure 2. (a) (A)gHigheeSpecifical-instar (Anter (MilliTest)) (b) itest set-up for software temperature temperatu

Figure 2d,e present, respectively, the final shape and the profile of the rubber sample after sliding. The rubber sample was subject to considerable wear; such a high level degradation could not be achieved without taking into account a strong link between the heating history and wear, which is a highly non-linear phenomenon. Another parameter to take into account is the split coefficient of the heat generated by friction between the rubber, the sample and the track surface.

Figure 2d,e present, respectively, the final shape and the profile of the rubber sample after sliding. The rubber sample was subject to considerable wear; such a high level degradation could not be achieved without taking into account a strong link between the heating history and wear, which is a highly non-linear phenomenon. Another parameter to take into account is the split coefficient of the heating mattered by friction between the rubber, the sample and the track surface. 5 of 14

2.2. Experimental Study: Simulated Frictional Heating of the Tire Treadibyam External Heatt Source

The main objective of this is statist is determine the temperapera and avoid a single time in the interval of the single temperature of the single temperature in the single t





(b)

Figure 3: (a) Heating setting; (b) tire tread and sensors positions, zone (1) presents the tire tread zone where the heating was applied, zone (2) presents the internal part of tire tread where an infrared sensor was installed, zone (3) presents the grover area where an extra sensor substalled, zone (3) presents the grover area where an extra sensor substalled, zone (3) presents the grover area where an extra sensor substalled, zone (3) presents the grover area where an extra sensor substalled, zone (3) presents the grover area where an extra sensor was installed, zone (3) presents the grover area where an extra sensor substalled, zone (3) presents the grover area where an extra sensor was assured to a sensor was an extra sensor was assured to a sensor was a sensor was assured to a sensor was a sen

Figure 4 presents the temperature evolution of various tire tread zones. As can be observed, for Figure 4 presents the temperature evolution of various tire tread zones. As can be observed, for the thermal evolutions corresponding to the heated zone (zone (1), triangles), the time shifted increases compared to the temperature inside the tire tread (zone (2)), and smaller values were noted compared to the temperature inside the tire tread (zone (2)), and smaller values were noted (blue solid). Concerning the grooves, the evolution profile appears very similar to that of the tread (concerning the grooves, the evolution profile appears very similar to that of the tread (red solid). However, we can notice that, after the end of the heating, marked by the vertical line (Figure 4), the groove temperature continues to rise for a short time before reaching a plateau. This proves that the temperature is "stored" and "diffused" throughout the tire rubber and, at the same time, that the rubber has a very low conductivity coefficient.



2.3. Numerical Simulation/Heating of the Tire Tread

Figure 5 illustrates a comparison between the experimental measurements and the simulated temperature evolution of the tire tread rubber (zone (2), Figure 3b) for an equivalent external heat source. The numerical simulation was carried out using a numerical specimen identical to the physical specimen.

A good corigination is obtained between the simulated temperature of the simulated temperature of the simulation of the presents the internal thermal evolution of the rubber block observed during stimulation of the good corigination of the tree observed, the core temperature increases during the first figure 3 investigation of the simulation of the simulated temperature increases during the simulated temperature increases during the simulated temperature increases during the first figure 3 investigation of the simulated temperature increases during the simulated temperature evolution of the uncertained the experimental measurements and the simulated temperature evolution of the uncertained the simulated temperature evolution of the uncertained out when the experimental measurements and the simulated temperature evolution of the uncertained out when the experimental measurements and the simulated temperature evolution of the uncertained out when the experimental measurements and the simulated temperature evolution of the uncertained out when the evolution of the uncertained out using a numerical specimental identical to the prevention was carried out using a numerical specimental identical to the prevention of the rubber remains identical.



A good correlation is obtained between the simulated temperature of the tire tread and that actually measured. Figure 6 presents the internal thermal evolution of the rubber block observed during sliding (IDS test, Figure 2b). As can be observed, the core temperature increases during the first instants of sliding; a plateau is rapidly attended. It can be concluded that the temperature increase of rubits of sliding with the temperature increase of sliding attended. It can be concluded that the temperature increase of rubits of sliding attended. It can be concluded that the temperature increase of rubits of sliding attended. It can be concluded that the temperature increase of rubits of sliding attended. It can be concluded that the temperature increase of rubits of sliding at the temperature increase of the time temperature increase of the temperature increase of the time temperature increase of the temperature increase of temperature increase



2.4. Numerical Simulation of a Sliding Rubber Block The sine of lithesistuation is a sliding Rubber Block The sine of lithesistuation is the state of the state of the distingtion of

Aishingshord, diburghtaiae kinitete sind a kiniteterior is bilivestigisplacements and the analysis of the second step, the block is moved with a constant sliding whether the block is moved with a constant slidi



(a)

Figure 7. Cont.





Figure 7: (a) The 2D Animetical specimen (b) specimen shape after sliding in Von Mises Notess Astribution.



Figure 8. (a) Numerical specimen, rubber sample shape with initial croded zone; (b) typical lateral **Figure 8**. (a) Numerical specimen, rubber sample shape with initial croded zone; (b) typical lateral view of deformed, wom sample form? (c) rubber sample shape with initial croded zone; (b) typical lateral view of deformed, wom sample form? (c) rubber sample shape (d) rubber (d) rubber test sample during single difficult and the sample shape with initial croded zone; (b) typical lateral view of deformed, wom sample form? (c) rubber sample final shape; (d) rubber test sample during sliding at the line; friction tester.

Figure 8a presents a full 2D image of the eroded initial shape of the rubber sample corresponding Figure 8a presents a full 2D image of the eroded unitial shape of the rubber sample corresponding to the experimental specimen (Figure 8c). Figure 8b shows the numerical results with respect to the same rubber sample, while taking into a count the final eroded shape of the specimen by means of an erother ising the taking into a count the final eroded shape of the specimen by means of an erother ising hapen length of the rubber block is loaded with 2.5 MP and sliding at 4 m/s. This deformed shape is in good correlation with the rubber block tested using the Hannover facility (Figure 8d). an "eroded initial shape". In this case, the rubber block is loaded with 2.5 MPa and sliding at 4 m/s. This deformed shape is in good correlation with the rubber block tested using the Hannover facility (Figure 8d).

Figure 9 presents a full 3D numerical simulation for a rubber block loaded with 2:5 MPa and sliding on a deformable concrete track at a fixed velocity of 4 m/s. Coupled temperature-displacement elements (E3D8RHT, eight-node tri-linear displacement and temperature, reduced integration with hourglass control, hybrid with constant pressure) were used in this simulation.



Figure 9: The 3B view of a rubber block on a runway track, thermal evolution of the contact zone, the temperatures are presented in $^{\circ}$ C (NF11):

As already mentioned, the numerical simulation of a sliding rubber sample demonstrates the As already mentioned, the numerical simulation of a sliding rubber sample demonstrates the same tendencies as the friction rig results. Figure 10a presents the profile of an experimental specimen same tendencies as the friction rig results. Figure 10a presents the profile of an experimental specimen of a rubber sample observed after the first stiding step. The thermal evolution of the contact zone of of a rubber sample observed after the first sliding step. The thermal evolution of the contact zone of the the rubber sample was measured using an intrared camera (Figure 2b), and the temperature of each rubber sample was measured using an infrared camera (Figure 2b) and the temperature of each area of area of the contact zone is shown in Figure 10b. The numerical simulation presented in Figure 10d, the contact zone is shown in Figure 10b. The numerical simulation presented in Figure 10d, shows the shows the temperature variations between 170°C and 193°C (or ange code) in the tipper part of the contact zone variations can also be observed on the contact zone variations can also be observed on the experimental in the contact presented in the second seco which can also be observed on the experimental block (orange patch, Figure 10b), However, a very However, a very high temperature was noted at the edge of the numerical rubber block (Figure 10b, high temperature was noted at the edge of the numerical rubber block (Figure 10b, red zone); it appears red zone), it appears to reveal the first abrasion zone. The sliding sample is subject to several to reveal the first abrasion zone. The sliding sample is subject to several mechanical and thermal mechanisms: the heating of the sliding surface leads to wear. The contact mechanisms: the heating of the sliding surface leads to wear. The contact rubber layer of the rubber rubber sample wears out and a new undamaged surface comes into contact. This sample wears out and a new undamaged surface comes into contact. This new surface undergoes a new surface undergoes a temperature fise until wear appears again and thus the process of repeated temperature rise until wear appears again and thus the process of repeated (heating/wearing) leads to (heating/wearing) leads to the final eroded shape shown in Figure 10a. The numerical simulation the final eroded shape shown in Figure 10a. The numerical simulation results shown in Figure 10c results shown in Figure 10c do not take into account this erosion mechanism, as the temperature of do not take into account this prosion mechanism, as the temperature of the leading edge rises to the leading edge rises to 1800°C. As previously mentioned, the experimental test shows avery high 1800 °G. As provide and the experimental test shows a very high level of abrasion and the experimental test shows a very high level of abrasion and the phenomenon and the experimental test shows a very high level of the phenomenon and the experimental test of the experim phenomenon needs to be taken into consideration. Within the ABAQUS code [30], a method could be code [30], a method could be adopted, based on a new modified geometry (new mesh) for each step of sliding. This method will be discussed in a future study.

discussed in a future study. From these present studies, it can be concluded that pressure and temperature have a considerable influence on the aircraft tire wear phenomenon which needs to be taken into account in future simulations. It must also be mentioned that a specific mesh could be used in the case of a constant wear ratio and a constant number of elements, and the rubber block could be divided into sub-zones, each zone having its own heating history.



Figure 10. (a) Experimental specimen, a rubber sample shape after the sliding step; (b) experimental measurement, thermal evolution of a rubber sample/contact zone at the end of the sliding step (in °C); (c) numerical specimen, rubber sample shape after first sliding; (d) numerical simulation typical lateral view of the deformed specimen's final shape appearance, thermal evolution in the contact zone, the temperatures are presented in °C (NT11).

From these present studies, it can be concluded that pressure and temperature have a considerable influence on the aircraft tire wear phenomenon which needs to be taken into account in future simulations. It must also be mentioned that a specific mesh could be used in the case of a constant wear ratio and a constant number of elements, and the rubber block could be divided into sub-zones, each zone having its own heating history.

3. Finite Element Model of a Rolling Tire

FEM of a Rolling Tire Using a Variable Friction Coefficient

In a precedent study, a purely mechanical finite element analysis of a rolling tire was investigated [22]. The model was able to predict the cornering behavior of an aircraft tire in contact with a flat rigid body. The Coulomb friction was modeled using a stiffness method [12] that permits some relative motion of the surfaces.

In this section, a thermo-mechanical model was proposed in order to predict the friction coefficient μ , the self-aligning torque M_z and the thermal evolution of a rolling tire. A classical rolling (CR) method based on the Lagrangian formulation was used along with a variable friction coefficient μ (P,T) dependent on the contact pressure and temperature. The contact aspect is modeled in a soft-contact

manner using exponential pressure-overclosure regularization, where the clearance C_0 and the contact pressure at zero clearance λ_0 are defined, based on a numerical sensitivity study. A Prony model was used for the viscoelastic behavior of the rubber. Concerning the friction coefficient, the IDS facility (High Speed Linear Friction Tester, Figure 2b) was used once more in order to identify the dependence of the friction coefficient on temperature and pressure. The experiments for this study were performed on an asphalt test track, and rubber tread samples measuring 20 × 20 mm on the contact area and 10 mm in thickness were used. Experimental results are shown in Figure 11 which presents the variation of the friction coefficient as a function of the ambient temperatures of the test track, for different given contact pressures (corresponding to the real pressures applied on the aircraft tire). As can be noted, for a given contact pressure, the temperature increase results in a decrease in the friction coefficient values. Figure 12 illustrates the variation of the friction coefficient as a function of the contact pressure for different given ambient temperatures of the test track. Here, for a given contact temperature, the friction coefficient decreases with the increase of the contact pressure. These variations of the friction coefficient were taken into account and applied in the numerical simulations.



Figure 11. IDS experimental results, variation of friction coefficient as function of the ambient temperature of the test track, for different given contact pressures.



Figure 12. IDS experimental results, variation of friction coefficient as function of contact pressure for different given ambient temperatures.

A cornering tire loaded at vertical force F_z , running at a velocity of 37.79 knots (19.44 m/s) and a fixed skidding angle, is presented. The temperature of the tire tread increases significantly after a few

seconds of cornering, and the rubber reaches an external high temperature which is close to the critical temperature at which the rubber is deposited on the concrete surface [3]. An experimental test was realized on an aircraft tire, and the thermal evolution of the contact zone of the aircraft tire tread was recorded during both rolling and skidding phases using an infrared camera (Figure 13a). Figure 13b shows the experimental data of the thermal evolution of the contact zone in the cornering conditions

of an aircraft tire under load, running at a velocity of 37.79 knots and for a given slip angle β_1 for a given time (t₆). The blue solid presents the three stages of the skidding test. In the first stage, the tire is skidded from 0° to β_1 °. In the second stage, the tire is kept rolling at β_1 ° for a given time, and in the last stage, the tire is moved from the skidding position (at β_1 °) to the initial position (at 0°). The red line (Figure 13b,d) reveals the heating of the tire tread after a few seconds of skidding, during the first stage of the skidding test.



Figure 13. (a) Experimental measurement; (b) experimental data, thermal evolution of the contact zone during the skidding test for a given skidding angle β_1 ; (c) numerical simulation, thermal evolution of the contact zone of the tire during a cornering stage (the temperatures are presented in °C (TEMP); (d) thermal evolution of the contact zone during the first seconds of skidding (first stage of skidding test), comparison between experimental test and numerical simulation.

Figure 13d presents the thermal evolution in contact zone of tire tread during the first seconds of skidding (between t_0 and t_1 seconds). As can be noted, the numerical simulation tends to predict the maximum temperature of the tire tread after a few seconds of rolling in skidding mode (first stage of the skidding test). Even if the temperature of the contact zone increases to 140 °C, the experimental data shows a different tendency for the initial gradient. That difference could be related to the rolling

test conditions, to the thermal conductivity of the runway surface and to the inelastic deformation, one of the most important mechanisms which leads to self-heating. Further investigations must be carried out in order to obtain the same evolution of the temperature observed experimentally.

4. Conclusions

In this paper, a numerical simulation of an aircraft tire in contact with a rough surface, using a variable friction coefficient depending on temperature and contact pressure, is presented. Firstly, a sliding facility was used in order to evaluate this dependence of the friction coefficient. The thermal evolution of rubber block samples under the sliding condition is also discussed. Additional experimental tests were realized in order to simulate the frictional heating phenomenon developed in the tire tread using an external heat source. In these studies, all experimental parameters were chosen, being similar to the real parameters of aircraft tires in rolling and skidding conditions (vertical loading, velocity, inflation pressure and skidding angle). Both frictional heating and temperature diffusion were compared to numerical 2D and 3D simulations and similar temperature prediction could be obtained. The final shapes of the rubber samples at the end of the experiment sliding procedures clearly show a high-level wear which needs be taken into consideration in future simulations.

A 3D finite element model for a rolling tire in the cornering phase was investigated using a variable friction coefficient dependent on temperature and contact pressure. The numerical simulation tended to predict the temperature of the tire tread after a few seconds of rolling in skidding position; however, further investigations must be carried out in order to obtain the evolution of the temperature observed experimentally.

Author Contributions: Hagen Lind and Matthias Wangenheim, from the Institute of Dynamics and Vibration Research (IDS), Leibniz Universität Hannover, performed the experiments concerning the determination of the friction coefficient and the frictional heating of the tire tread rubber; Iulian Rosu and Hélène L. Elias-Birembaux carried out the numerical and experimental simulations and analyzed the data. All authors were involved in the preparation of the manuscript.

Conflicts of Interest: The authors declare no conflict of interest.

References

- 1. Sakai, H.; Araki, K. Thermal engineering analysis of rubber vulcanization and tread temperature during severe sliding of a tire. *Tire Sci. Technol.* **1999**, *27*, 22–47. [CrossRef]
- 2. Moore, D.F. Friction and wear in rubbers and tyres. Wear 1980, 61, 273–282. [CrossRef]
- 3. Tanner, J.A.; Dreher, R.C.; Strubb, S.M.; Smith, E.G. *Tire Tread Temperatures during Antiskid Braking and Cornering on a Dry Runway*; NASA Technical Paper 2009; National Aeronautics and Space Administration: Washington, DC, USA, 1982.
- 4. McCarty, J.L.; Tanner, J.A. *Temperature Distribution in Aircraft Tire at Low Ground Speed*; NASA Technical Report 2195; NASA Lewis Research Center: Cleveland, OH, USA, 1983.
- Clark, S.K.; Dogde, R.N. *Heat Generation in Aircraft Tire Under Free Rolling Conditions*; NASA Technical Report 3629; Michigan University Department of Mechanical Engineering and Applied Mechanics: Ann Arbor, MI, USA, 1982.
- 6. Eichhorn, U.; Roth, J. Prediction and monitoring of tyre/road friction. In Proceedings of the XXIV FISITA Congress, London, UK, 7–11 June 1992; Volume 2, pp. 67–74.
- Bui, Q.V.; Ponthot, J.P. Estimation of rubber sliding friction from asperity interaction modeling. *Wear* 2002, 252, 150–160. [CrossRef]
- 8. Lebon, F. Contact problems with friction: Models and simulations. *Simul. Model. Pract. Theory* **2003**, *11*, 449–463. [CrossRef]
- 9. Grosch, K.A. The relation between friction and visco-elastic properties of rubber. *Proc. R. Soc. Lond. Ser. A Math. Phys. Sci.* **1963**, 274, 21–39. [CrossRef]
- 10. Kongo Kondé, A. Modélisation du Roulement D'un Pneumatique D'avion. Ph.D. Thesis, Aix-Marseille University, Marseille, France, 2011.

- 11. Kongo Kondé, A.; Rosu, I.; Lebon, F.; Brardo, O.; Devésa, B. On the modeling of aircraft tire. *Aerosp. Sci. Technol.* 2013, 27, 67–75. [CrossRef]
- 12. Kongo Kondé, A.; Rosu, I.; Lebon, F.; Seguin, L.; Brardo, O.; Troude, F.; Devésa, B. Thermomechanical couplings in aircraft tire rolling/sliding modeling. *Adv. Mater. Res.* **2011**, 274, 81–90. [CrossRef]
- McCarty, J.L.; Yager, T.J.; Riccitiello, S.R. Wear, Friction and Temperature Characteristics of an Aircraft Tire Undergoing Braking and Cornering; NASA Technical Paper 1569; NASA Langley Research Center: Hampton, VA, USA, 1979.
- 14. McCarty, J.L. Wear and Related Characteristics of an Aircraft Tire during Braking; NASA Technical Note, D-6963; National Aeronautics and Space Administration: Washington, DC, USA, 1972.
- 15. Henry, J.J. Evaluation of Pavement Friction Characteristics: A Synthesis of Highway Practice, NCHRP Synthesis 291; Transportation Research Board, National Research Council: Washington, DC, USA, 2000.
- 16. Hall, J.W.; Smith, K.L.; Titus-Glover, L.; Wambold, J.C.; Yager, T.J.; Rado, Z. *Guide for Pavement Friction*; National Cooperative Highway Research Program, Transportation Research Board of the National Academies: Washington, DC, USA, 2009.
- 17. Persson, B.N.J. On the theory of rubber friction. Surf. Sci. 1998, 401, 445–454. [CrossRef]
- Persson, B.N.J.; Albohar, O.; Tartaglino, U.; Volokitin, A.I.; Tosatti, E. On the nature of surface roughness with application to contact mechanics, sealing, rubber friction and adhesion. *J. Phys. Condens. Matter* 2005, 17, R1–R62. [CrossRef] [PubMed]
- 19. Persson, B.N.J. Rubber friction: Role of the flash temperature. *J. Phys. Condens. Matter* **2006**, *18*, 7789–7823. [CrossRef] [PubMed]
- 20. Elias-Birembaux, L.; Rosu, I.; Lebon, F. Thermo-mechanical modelling of the aircraft tyre cornering. *Mach. Dyn. Res.* **2013**, *37*, 29–36.
- 21. Rosu, I.; Elias-Birembaux, L.; Lebon, F. Thermo-viscoelastic modeling of the aircraft tire cornering. *Adv. Mater. Res.* **2015**, *1099*, 80–86. [CrossRef]
- 22. Elias-Birembaux, L.; Rosu, I.; Lebon, F. Thermo-viscoelastic modeling of aircraft tire cornering. In Proceedings of the 30th International Conference of Polymer Processing Society, Cleveland, OH, USA, 6–12 June 2014.
- Fevrier, P.; Le Maitre, O. Tire temperature modeling: Application to race tires. In Proceedings of the VDI 13th International Congress Numerical Analysis and Simulation in Vehicle Engineering, Wurzburg, Germany, 27–28 September 2006.
- 24. Kainradl, P.; Kaufmann, G. Heat generation in pneumatic tires. *Rubber Chem. Technol.* **1976**, *49*, 823–861. [CrossRef]
- 25. Mooney, M. A theory of large elastic deformation. J. Appl. Phys. 1940, 11, 582–592. [CrossRef]
- 26. Rivlin, R.S. Large elastic deformations of isotropic materials. IV. Further developments of the general theory. *Philos. Trans. R. Soc. Lond.* **1948**, 241, 379–397. [CrossRef]
- 27. Lin, Y.J.; Hwang, S.J. Temperature prediction of rolling tires by computer simulation. *Math. Comput. Simul.* **2004**, *67*, 235–249. [CrossRef]
- 28. Lahellec, N.; Mazerolle, F.; Michel, J.C. Second-order estimate of the macroscopic behaviour of periodic hyperelastic composites: Theory and experimental validation. *J. Mech. Phys. Solids* **2004**, *52*, 27–49. [CrossRef]
- 29. Yanjin, G.; Guoqun, Z.; Gang, C. Influence of belt cord angle on radial tire under different rolling states. *J. Reinf. Plast. Compos.* **2006**, *25*, 1059–1077. [CrossRef]
- 30. Dassault Systèmes. *ABAQUS/Standard*; Theory Manual and Example Problems Manual; Release 6.12; Dassault Systèmes: Waltham, MA, USA, 2011.



© 2016 by the authors; licensee MDPI, Basel, Switzerland. This article is an open access article distributed under the terms and conditions of the Creative Commons Attribution (CC-BY) license (http://creativecommons.org/licenses/by/4.0/).