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Dimitrios Tsalis, Kevin Bonnay, Georges Chatzigeorgiou, Nicolas Charalambakis. Effective properties of multiphase composites made of elastic materials with hierarchical structure. Mathematics and Mechanics of Solids, 2015, 10.1177/1081286515612142. hal-01320767

## HAL Id: hal-01320767 https://hal.science/hal-01320767v1

Submitted on 24 May 2016  $\,$ 

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# Effective properties of multiphase composites made of elastic materials with hierarchical structure

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#### Abstract

In this paper, the analytical solution of the multi - step homogenization problem for multi - rank composites with generalized periodicity made of elastic materials is presented. The proposed homogenization scheme may be combined with computational homogenization for solving more complex microstructures. Three numerical examples are presented, concerning locally periodic stratified materials, matrices with wavy layers and wavy fiber reinforced composites.

#### Keywords

Micromechanics; Generalized periodicity; Multi - step homogenization; Elasticity; Analytical solution.

### 1. Introduction

Natural and manufactured materials may exhibit complex structures with more than one length scale, in a manner that a "hierarchical" structure could be defined (see Fig. 1). This "hierarchy" plays a major role in determining the bulk material properties (Lakes (1993)). Complex micro - architectures can be found in the nature or constructed or accidently obtained in manufacturing processes and the related thermo - electro - mechanical behavior of forming complex structures under different conditions, or the wave propagation in these structures or kinetic techniques and propagation of oscillations are of great technological importance (Jabin & Tzavaras (2009); Moreno et al. (2014)). Microstructures with periodicity are often characterized by the repetition of the same material element with respect to one or two or three - dimensional coordinate axis. However, there are structures that can not be obtained by the repetition of the same micro - volume. Examples of these structures are the structures with cylidrical periodicity or the locally periodic stratified structures.

One of the simplest composite is a stratified material where the material properties vary only in one direction, called the direction of lamination. This laminate is called rank - 1. In multiple - rank laminates, there is a large difference in the scales of the successive laminations (Fig. 1), which are in different directions (Milton (2002)). More complicated, than rank -1, locally periodic multilayer structure is the chevron pattern (see Fig.2), usually used as core in sandwich structures

(Lebee & Sab (2012)). We refer to the concept of "stratified periodic homogenization", introduced and developed by Bensoussan et al. (1978) and Briane, 1990, 1993, and the concept of "generalized periodicity" presented in Tsalis et al. (2012) and Chatzigeorgiou et al. (2012). Applications of multilayer composites are very common. The human compact bone is a characteristic example of structure that belongs to this caterogy. Generally, it consists of osteons, which are hollow fibers composed of concentric lamellae and pores (Lakes (1993)). The cells of the human heart form fibers nearly parallel to the cardiac wall with orientations varying continuously from the endocardium at the epicardium (Briane, 1990, 1993, Chan & Leong (2008)). In construction industry, multilayers fiber reinforced composites are very popular since antiquity. Woven fabrics covered by an appropriate matrix in a multilayer structure are used in a wide - range of applications, such as protective materials for military and law enforcement personnel or structural containment of turbine fragments (one of the most widely used applications is in propulsion engine containment systems where engine containment system is typically constructed by wrapping multiple layers of Kevlar @49 around a thin aluminum encasement (Nilakantan et al., 2008; Rajan et al., 2010; Zhu et al., 2014)).

It is known that finite element analysis (FEA) methods are not sufficient to solve heterogeneous problems, since heterogeneities impose restrictions on the size of elements and make too expensive the discretization of heterogeneous structures (Bensoussan et al., 1978; Guedes & Kikuchi, 1990; Hollister & Kikuchi, 1992), unless they are combined with micromechanical and/or analytical methods (Hashin, 1983; Kalamkarov & Kolpakov, 1997; Nemat-Nasser & Hori, 1999; Aboudi et al., 1999; Nemat-Nasser, 1999; Guinovart-Diaz et al., 2005; Chatzigeorgiou & Charalambakis, 2005; Love & Batra, 2006; Chatzigeorgiou et al., 2007; Kalamkarov et al., 2009; Pindera et al., 2009; Nie et al., 2011; Wu et al., 2014; Chatzigeorgiou et al., 2014; Abd-Alla et al., 2014; Tu & Pindera, 2014; Savvas et al., 2014; Mahmoud et al., 2014).

Mathematical homogenization (Babuska, 1976a,b; Bensoussan et al., 1978; Sanchez-Palencia, 1978; Francfort, 1983; Francfort & Murat, 1986; Allaire, 1992; Murat & Tartar, 1997; Milton, 2002), combined with numerical simulation, finite element or finite volume methods (see for instance Zhigang et al., 2010; Gelebart, 2011; Lee & Yu, 2011; Nackenhorst et al., 2011; Pindera et al., 2012; Birman et al., 2013; Balokhonov et al., 2014; Katz et al., 2015; **?**; Tu & Pindera, 2014; Freund et al., 2014) remain a rigorous theory for describing heterogeneous materials and structures in engineering applications. Multiscale FEA methods are applied, in which the convergence is improved by defining the basis functions in the following macro–micro way: it is used a coarse mesh for defining the nodal values of the unknown functions and a fine mesh for precomputing the locally periodic oscillating basis functions that depend on the microproperties. So the problem dimension is that of the coarse mesh.

Generalized periodicity - based homogenization theory allows for studying the effective properties in elasticity and generalizes the micromechanics Hill's lemma (see Tsalis et al. (2012) for details). In this paper, we assume that microstructure is periodic with respect to linear or non - linear periodicity surfaces, ordered in different completely distinct scales and forming a multi - rank coated laminate, for the case of linear periodicity (Milton (2002)), or a multi - rank stratified material with generalized (for instance wavy or cylindrical) periodicity. The existence of a sequence of scales of decreasing order allows for performing a succession of homogenization steps, in which the "homogenized" material from every step acts as initial ("heterogeneous") material for the next step. In every step, the homogenization scheme requires the solution of an one – dimensional cell problem with data corresponding to the material properties of the previous step and the volume fractions corresponding to the actual size. We note that in Tsalis et al., 2012, 2013 we presented the solution only of a two–step homogenization and we used trapezoidal rule for the second step. In Section 2 we derive the micro- and macroequation that govern the problem during a homogenization step and present the expression of the effective elastic matrix in terms of the microequation during a homogenization step, which is necessary in order to compute the effective elastic matrix. In Section 4, three examples are presented, namely a "chevron" pattern, a composite with

wavy layers and a matrix reinforced by wavy fibers in two directions. The results are crossovered with the results from the Dissipation-Inequality based Periodic Homogenization (DIPH) (see Tsalis et al. (2015)) showing an excellent agreement.



Fig. 1. An example of hierarchical structure, where in every subfigure the structure consists of only two materials  $(\epsilon^V \ll \epsilon^{IV} \sim \epsilon^{III} \ll \epsilon^{II} \ll \epsilon^{I}$ , where  $\epsilon$  is the heterogeneity parameter (see Section 2 for more details)).



Fig. 2. "Chevron" structure. Two microscale composite with laminate structure, forming angles  $\theta_1$  and  $\theta_2$  with  $x_1$  axis in the macrocoordinate system, the one - dimensional cell in the framework of generalized periodicity in microcoordinate system of the first step and the corresponding of the second step. The material to be homogenized of the second step is a two - phase stratified material.

#### 2. Definition of the micro- and macroequation problem and the effective elastic matrix

In mathematical homogenization at least two scales are introduced. The first one is the macroscale denoted by  $x \in \Omega$ , where  $\Omega$  is the volume occupied by the heterogeneous body, at which the heterogeneities, characterized by  $\epsilon$ , are very small compared to the whole structure. The second one is the microscale denoted by  $\frac{x}{\epsilon}$ , which is the actual scale for the heterogeneities. At every step of homogenization, the scale is much bigger than the scale of the previous step.

The choice of the RVE is made with respect to the generalized periodicity function  $\rho(x)$  and  $Y = [0, y_1] \times [0, y_2] \times [0, y_3]$  is chosen to be the basic cell, where

$$y = \frac{\boldsymbol{\varrho}(\boldsymbol{x})}{\epsilon},\tag{1}$$

where the parametric equations  $\rho(x) = \text{constant}$  describe the surfaces into the structure, with respect to which a material property  $\varphi$  is periodic (for more details see Tsalis et al. (2012)). Generally, generalized periodicity function  $\rho$  is a vector valued function (i.e. in the case of (wavy) stratified material, it is simplified to a scalar function while in the case of cylindrical periodicity, it is a two component vector). In the general case, the non-linear periodicity (for instance cylindrical periodicity), if solved by cylindrical coordinates, needs a sufficiently large number of cell problems to be solved. On the contrary, using generalized periodicity, the problem needs the solution of only one cell problem (see Fig. 4). In the case



Fig. 3. A typical wavy two - microscale multilayered material. The material to be homogenized of the second step is orthotropic.

of wavy layers, the two-dimensional problem  $\overline{Y}$ , thanks to the generalized periodicity  $\rho$ , becomes one-dimensional (in Fig. 3 the space Y of the first step).



**Fig. 4.** Using the concept of generalized periodicity function we are able to define a unique cell problem for the whole structure instead of a sufficiently large number of cell problems in different radial positions.

Then, if these surfaces are very close each other forming a microstructure (without necessarily geometric periodicity), the property  $\varphi$  is periodic with respect to  $\frac{\varrho(x)}{\epsilon}$ . We write then

$$\varphi^{\epsilon}(\boldsymbol{x}) = \varphi\left(x, \frac{\boldsymbol{\varrho}(\boldsymbol{x})}{\epsilon}\right), \text{ periodic with respect to } \frac{\boldsymbol{\varrho}(\boldsymbol{x})}{\epsilon},$$
 (2)

or

$$\varphi(\boldsymbol{x}, \boldsymbol{y}), \quad y - \text{periodic.}$$
 (3)

The dependence of functions on the microcoordinate is performed (generally in a non - periodic way) via

$$\bar{\boldsymbol{y}} = \frac{\boldsymbol{x}}{\epsilon}.\tag{4}$$

We denote field variables  $\sigma^0$ ,  $\varepsilon^0$  and  $u^1$  as microscopic variables and  $\Sigma$ , E and  $u^0$  as the macroscopic variables where the macroscopic quantities depend only on the macrocoordinate x. Both classes of deformation fields are related to the unit cell located at x. Away from the boundaries  $\partial\Omega$ , stress and strain fields conform at the microlevel to the generalized periodicity conditions:

$$\sigma^0, \varepsilon^0$$
 are Y – periodic functions of y. (5)

This assumption is equivalent to the assertion that the stress belongs to the space of microscopically self - equilibrated fields.

The actual displacement  $u^0$  within Y located at x has two properties: it is oscillating and it has a generalized periodicity. We assume that  $u^0$  can be expressed as a sum of a linear and a periodic part (Tartar, 1978; Suquet, 1985; Tsalis et al., 2012)

$$u_i^0(\boldsymbol{x}, \bar{\boldsymbol{y}}, \boldsymbol{y}) = E_{ij}\bar{y}_j + u_i^1, \tag{6}$$

where

$$u_i^1 = u_i^1(\boldsymbol{x}, \boldsymbol{y}),\tag{7}$$

is periodic with respect to y. We remind that the difference with Suquet (1985) is that, there, the function  $u_i^1$  is periodic with respect to the microvariable  $\bar{y}$ , while here it is periodic only with respect to the generalized periodicity (Tsalis et al. (2012)).

Microstrain is defined from (6) with respect to the microcoordinate,

$$\varepsilon_{ij}^{0} = \operatorname{sym}\left(\frac{\partial u_{i}}{\partial \bar{y}_{j}}\right) = \frac{1}{2}\left(\frac{\partial u_{i}}{\partial \bar{y}_{j}} + \frac{\partial u_{j}}{\partial \bar{y}_{i}}\right),\tag{8}$$

while the macrostrain is defined with respect to the macrocoordinate,

$$E_{ij} = \operatorname{sym}\left(\frac{\partial u_i^0}{\partial x_j}\right) = \frac{1}{2}\left(\frac{\partial u_i^0}{\partial x_j} + \frac{\partial u_j^0}{\partial x_i}\right).$$
(9)

It was prooved (Tsalis et al. (2013)) that

$$\langle \varepsilon_{ij}^0 \rangle = E_{ij}.\tag{10}$$

We consider a rank - 1 composite made of elastic material exhibiting periodicity with respect to the vector valued function  $\rho(x)$ . Two main hypotheses hold. The first is the perfect bonding among layers of the constituents (see in Berrehili (2014) the effect of debonded fibers of a composite under traction). The second one is that there is no cracking in the structure. The elastic coefficients are assumed of the form,

$$C_{ijkh}(\boldsymbol{y}) = C_{ijkh}\left(\frac{\boldsymbol{\varrho}(\boldsymbol{x})}{\epsilon}\right), \quad C_{ijkh}(\boldsymbol{y}) \quad Y - \text{periodic}, \quad C_{ijkh} \in L^{\infty}(\mathbb{R}^m), \tag{11}$$

$$\boldsymbol{\varrho} \in C^2(\mathbb{R}^m), \quad m \le 3, \tag{12}$$

$$C_{ijkh}\xi_{ij}\xi_{kh} \ge \lambda\xi_{ij}\xi_{ij}, \ \forall \ \xi \ \text{symmetric}, \ C_{ijkh} = C_{jikh} = C_{ijhk} = C_{khij},$$
(13)

$$\sum_{i=1}^{n} \left( \sum_{k=1}^{m} \frac{\partial \varrho_k(\boldsymbol{x})}{\partial x_i} \eta_k \right)^2 \ge \beta \sum_{k=1}^{n} \eta_k^2, \quad \forall \quad \boldsymbol{\eta}.$$
(14)

We note that, at every scale  $\frac{1}{\epsilon}$ , the generalized periodicity is related only to the actual scale. The sequentially formed composite allows for a succession of homogenization steps based on one-dimensional cell problems. We note that generalized

periodicity is the ingredient used to reduce the dimension of 2-D or 3-D cell problems in terms of cartesian coordinates (needing a FEM application), to the one-dimensional cell problems mentioned above, that can be solved analytically with smaller computational cost. See also the second example of Section 4.

We study the elasticity problem

$$\frac{\partial \sigma_{ij}^{\epsilon}}{\partial x_j} + f_i^{\epsilon}(x) = 0 \quad \text{in} \quad \Omega, \tag{15}$$

$$u_i^{\epsilon} = 0 \quad \text{on} \quad S_d, \tag{16}$$

$$\sigma_{ij}^{\epsilon} n_j = t_i(x) \quad \text{on} \quad S_t, \tag{17}$$

$$\sigma_{ij}^{\epsilon}(x) = C_{ijkh}^{\epsilon}(x)\varepsilon_{kh}(u^{\epsilon}), \tag{18}$$

where  $m{arepsilon}(m{u})=rac{1}{2}(
ablam{u}+^t
ablam{u})$  is the infinitesimal strain tensor. By putting

$$V = \{ \boldsymbol{v} \in (H^1(\Omega))^3 | \boldsymbol{v} |_{S_d} = 0 \},$$
(19)

where  $v|_{S_d}$  is the value of v on the boundary  $S_d$ , the equations of equilibrium (15), in weak formulation, are written

$$\int_{\Omega} \nabla_j \sigma_{ij}^{\epsilon} \omega_i d\Omega + \int_{\Omega} f_i^{\epsilon} \omega_i d\Omega = 0, \quad \forall \ \omega_i \in V.$$
<sup>(20)</sup>

We recall that  $H^1$  stands for the Sobolev space of  $L^2$  - functions having  $L^2$  - derivatives.

The dependence of the solution  $u^{\epsilon}$  in both the macroscopic and the microscopic levels justifies the assumption that the displacement and the related quantities (stress and strains) can be expressed as asymptotic expansions in terms of the microstructural parameter.

We apply the asymptotic expansion homogenization technique and we conclude (Tsalis et al. (2012)) that the microstress satisfies the equation of equilibrium,

$$\frac{\partial \sigma_{ij}^0}{\partial \bar{y}_j} = 0, \tag{21}$$

or

$$\frac{\partial \sigma_{ij}^0}{\partial y_m} \frac{\partial \varrho_m}{\partial x_j} = 0, \tag{22}$$

or

$$-\int_{Y} C_{ijlm} K_{lm}(w^{kh}) K_{kh}(\omega) dY = \nabla_{j} \varrho_{l} \int_{Y} \nabla_{yl} C_{ijkh} \omega_{i} dY,$$
(23)

where K is the symmetric matrix,

$$K(\boldsymbol{u}) = \frac{1}{2} \left( \nabla_y \boldsymbol{u} \nabla \boldsymbol{\varrho} + {}^t \left( \nabla_y \boldsymbol{u} \nabla \boldsymbol{\varrho} \right) \right).$$
(24)

The macrostress is defined as the mean value of the microstress over the unit cell,

$$\langle \sigma_{ij}^0 \rangle = \Sigma_{ij}.$$
(25)

In Tsalis et al. (2013), it is shown that the Mandel - Hill's lemma holds,

$$\langle \sigma^0_{ij} \varepsilon^0_{ij} \rangle = \Sigma_{ij} E_{ij}, \tag{26}$$

while the effective elastic matrix is given by

$$C_{ij\phi\theta}^{eff} = \langle C_{ij\phi\theta} + C_{ijmn} \frac{\partial w_m^{\phi\theta}}{\partial y} \frac{\partial \varrho}{\partial x_n} \rangle, \tag{27}$$

or by using Voigt notation for i=1, k=1, l=6 and m=5, for i=2, k=6, l=2 and m=4, for i=3, k=5, l=4 and m=3

$$C_{p\boldsymbol{a}}^{eff} = \langle C_{p\boldsymbol{a}} + \sum_{i=1}^{3} \left( C_{pk} \frac{\partial \varrho}{\partial x_1} + C_{pl} \frac{\partial \varrho}{\partial x_2} + C_{pm} \frac{\partial \varrho}{\partial x_3} \right) \frac{\partial w_i^{\boldsymbol{a}}}{\partial y}$$
(28)

In (26) and (27),  $\langle ... \rangle$  is the mean value operator. In (27) we use Voigt notation for *p* instead of *ij* and *a* instead of  $\phi\theta$ . The gradient of the generalized periodicity function enters the effective coefficients, reflecting the microstructure properties at the actual scale  $\epsilon$ . If there exists a next scale, the effective coefficients (28), obtained from the previous step, act as "heterogeneous" material coefficients of the next homogenization step. In the next section, the partial derivatives of elastic correctors are computed through the solution of the microequation (21).

#### 3. The solution of the microequation

The cell equation (21) is written as

$$\frac{\partial}{\partial y_l} \left( \frac{\partial \varrho_l}{\partial x_j} C_{ijkz} \frac{\partial \varrho_m}{\partial x_z} \frac{\partial w_k^{\phi\theta}}{\partial y_m} + \frac{\partial \varrho_l}{\partial x_j} C_{ij\phi\theta} \right) = 0$$
<sup>(29)</sup>

with respect to the unknowns  $w_l^{kh}$  (where the indices l, m take the value 1, the indices i, k take the values 1,2,3 and the indices j, z take the values 1,2) under appropriate periodicity and continuity conditions.

By expanding (29) for l, m, j, z, k,

$$\frac{\partial}{\partial y} \left( \frac{\partial \varrho}{\partial x_1} C_{i1\phi\theta} + \frac{\partial \varrho}{\partial x_2} C_{i2\phi\theta} \right) + \sum_{j=1}^3 \frac{\partial}{\partial y} \left( \frac{\partial w_j^{\phi\theta}}{\partial y} \left( \frac{\partial \varrho}{\partial x_1} \frac{\partial \varrho}{\partial x_1} C_{i1j1} + \frac{\partial \varrho}{\partial x_1} \frac{\partial \varrho}{\partial x_2} C_{i1j2} + \frac{\partial \varrho}{\partial x_2} \frac{\partial \varrho}{\partial x_1} C_{i2j1} + \frac{\partial \varrho}{\partial x_2} \frac{\partial \varrho}{\partial x_2} C_{i2j2} \right) \right) = 0.$$

$$(30)$$

Putting

$$a_{j}^{i}(x_{1}, x_{2}) = \frac{\partial \varrho}{\partial x_{1}} \frac{\partial \varrho}{\partial x_{1}} C_{i1j1} + \frac{\partial \varrho}{\partial x_{1}} \frac{\partial \varrho}{\partial x_{2}} C_{i1j2} + \frac{\partial \varrho}{\partial x_{2}} \frac{\partial \varrho}{\partial x_{1}} C_{i2j1} + \frac{\partial \varrho}{\partial x_{2}} \frac{\partial \varrho}{\partial x_{2}} C_{i2j2}, \tag{31}$$

for j=1,2 and 3, equation (30) is written

$$\frac{\partial}{\partial y} \left( \frac{\partial \varrho}{\partial x_1} C_{i1\phi\theta} + \frac{\partial \varrho}{\partial x_2} C_{i2\phi\theta} + a_1^i \frac{\partial w_1^{\phi\theta}}{\partial y} + a_2^i \frac{\partial w_2^{\phi\theta}}{\partial y} + a_3^i \frac{\partial w_3^{\phi\theta}}{\partial y} \right) = 0.$$
(32)

Equations (32) using Voigt notation for the free index i = 1, 2, 3 and by substitute  $\phi \theta$  by a for a = 1, 2, 3, 4, 5, 6, are written

$$\frac{\partial}{\partial y} \left( \frac{\partial \varrho}{\partial x_1} C_{k\boldsymbol{a}} + \frac{\partial \varrho}{\partial x_2} C_{l\boldsymbol{a}} + a_1^i \frac{\partial w_1^{\boldsymbol{a}}}{\partial y} + a_2^i \frac{\partial w_2^{\boldsymbol{a}}}{\partial y} + a_3^i \frac{\partial w_3^{\boldsymbol{a}}}{\partial y} \right) = 0, \tag{33}$$

for i=1, k=1 and l=6, for i=2, k=6 and l=2, for i=3, k=5 and l=4.

Intergate equations (33) for y,

$$(\text{mat}) \left( a_1^i \frac{\partial w_1^a}{\partial y} \right) + (\text{mat}) \left( a_2^i \frac{\partial w_2^a}{\partial y} \right) + (\text{mat}) \left( a_3^i \frac{\partial w_3^a}{\partial y} \right) = (\text{mat}) con_i^a - \frac{\partial \varrho}{\partial x_1} (\text{mat}) C_{ka} - \frac{\partial \varrho}{\partial x_2} (\text{mat}) C_{la},$$
(34)

for i=1, k=1 and l=6, for i=2, k=6 and l=2, for i=3, k=5 and l=4 and where  $(^{\text{mat})}con_3^a$  are the basic unknowns. Equations (34) are written in matrix form

$$\begin{pmatrix} (\operatorname{mat})a_{1}^{1} & (\operatorname{mat})a_{2}^{1} & (\operatorname{mat})a_{3}^{1} \\ (\operatorname{mat})a_{1}^{2} & (\operatorname{mat})a_{2}^{2} & (\operatorname{mat})a_{3}^{2} \\ (\operatorname{mat})a_{1}^{3} & (\operatorname{mat})a_{2}^{3} & (\operatorname{mat})a_{3}^{3} \end{pmatrix} \begin{pmatrix} (\operatorname{mat})\frac{\partial w_{1}^{a}}{\partial y} \\ (\operatorname{mat})\frac{\partial w_{2}}{\partial y} \\ (\operatorname{mat})\frac{\partial w_{3}}{\partial y} \end{pmatrix} = \begin{pmatrix} (\operatorname{mat})con_{1}^{a} - \frac{\partial \varrho}{\partial x_{1}} (\operatorname{mat})C_{1a} - \frac{\partial \varrho}{\partial x_{2}} (\operatorname{mat})C_{6a} \\ (\operatorname{mat})con_{2}^{a} - \frac{\partial \varrho}{\partial x_{1}} (\operatorname{mat})C_{6a} - \frac{\partial \varrho}{\partial x_{2}} (\operatorname{mat})C_{2a} \\ (\operatorname{mat})con_{3}^{a} - \frac{\partial \varrho}{\partial x_{1}} (\operatorname{mat})C_{5a} - \frac{\partial \varrho}{\partial x_{2}} (\operatorname{mat})C_{4a} \end{pmatrix}.$$

Solving with respect to  ${}^{(mat)}\frac{\partial w_1^a}{\partial y}$ ,  ${}^{(mat)}\frac{\partial w_2^a}{\partial y}$  and  ${}^{(mat)}\frac{\partial w_3^a}{\partial y}$  and intergating for y gives

$$\begin{pmatrix} (\operatorname{mat}) w_{1}^{a} \\ (\operatorname{mat}) w_{2}^{a} \\ (\operatorname{mat}) w_{3}^{a} \end{pmatrix} = \begin{pmatrix} (\operatorname{mat}) A_{1}^{1} & (\operatorname{mat}) A_{2}^{1} & (\operatorname{mat}) A_{3}^{1} \\ (\operatorname{mat}) A_{1}^{2} & (\operatorname{mat}) A_{2}^{2} & (\operatorname{mat}) A_{3}^{2} \\ (\operatorname{mat}) A_{1}^{3} & (\operatorname{mat}) A_{2}^{3} & (\operatorname{mat}) A_{3}^{3} \end{pmatrix} \begin{pmatrix} (\operatorname{mat}) \operatorname{con}_{1}^{a} - \frac{\partial \varrho}{\partial x_{1}} (\operatorname{mat}) C_{1a} - \frac{\partial \varrho}{\partial x_{2}} (\operatorname{mat}) C_{6a} \\ (\operatorname{mat}) \operatorname{con}_{2}^{a} - \frac{\partial \varrho}{\partial x_{1}} (\operatorname{mat}) C_{6a} - \frac{\partial \varrho}{\partial x_{2}} (\operatorname{mat}) C_{2a} \\ (\operatorname{mat}) \operatorname{con}_{3}^{a} - \frac{\partial \varrho}{\partial x_{1}} (\operatorname{mat}) C_{5a} - \frac{\partial \varrho}{\partial x_{2}} (\operatorname{mat}) C_{4a} \end{pmatrix} y + \\ \begin{pmatrix} (\operatorname{mat}) \operatorname{con}_{1}^{*} a \\ (\operatorname{mat}) \operatorname{con}_{2}^{*} a \\ (\operatorname{mat}) \operatorname{con}_{3}^{*} a \end{pmatrix},$$

where matrices  ${}^{(mat)}A_j^i$  are the inverse of the matrices  ${}^{(mat)}a_j^i$  while the constants  ${}^{(mat)}con_1^*a$ ,  ${}^{(mat)}con_2^*a$  and  ${}^{(mat)}con_3^*a$  are computed by the last equation using the boundary and periodicity conditions of the problem.

The periodicity conditions for the first and the last material in the unit cell read

at 
$$y = 0$$
,  ${}^{(1)}w_k^{\phi\theta} = 0$  and at  $y = 1$ ,  ${}^{(n)}w_k^{\phi\theta} = 0$ . (35)

The continuity conditions at the material interfaces in the unit cell read

$$[|w_k^{\phi\theta}|] = 0, \quad [|\frac{\partial\varrho}{\partial x_j}C_{ijlm}\frac{\partial\varrho}{\partial x_m}\frac{\partial w_l^{kh}}{\partial y} + \frac{\partial\varrho}{\partial x_j}C_{ij\phi\theta}|] = 0, \tag{36}$$

where [|...|] denotes the difference at both sides of an interface. The second equation (36) is equivalent to

$$\left[\left|\frac{\partial\varrho}{\partial x_j}\sigma_{ij}^0\right|\right] = 0. \tag{37}$$

From equations (34) and (37) one obtains that  $(mat) con_i^a$  is independent of y and take the same value in every material.

In general, for a *n*-phase composite structure, a system of 18(n-1) equations with 18 basic unknowns and 18(n-2) auxialliary unknowns solves the cell problem

$$\tilde{C}\tilde{U}=\tilde{S},$$
(38)

where the matrix of coefficients  $\tilde{C}$ , the matrix of unknowns  $\tilde{U}$  and the constant matrix  $\tilde{S}$  are shown in Appendix A, Tables 19, 20 and 21.

The novelty of this paper is the introduction of an automated process in the form of a "code" (Fig. 5) with analytical expressions for the effective stiffness of a multiphase material with hierarchical structure, while in Tsalis et al. (2012) the simple case of two phases is treated. The present "code" allows for passing easier from the finer scale to the coarser scale

due to the fact that the result of the coarser scale is the input data of the finer scale . Also, there is no need for  $\rho$  to be the same for all scales, giving the code flexibility.



Fig. 5. A schematic representation of the "code".

The case of wavy layered materials is a good example of a structure exhibiting generalized periodicity. In the coordinate system x, this problem needs a two-dimensional cell problem formulation. This two-dimensional cell problem can be solved only by FEA application. On the contrary, by considering that the structure exhibits periodicity with respect to the wavy surface, the problem can be formulated as one-dimensional cell problem.

In many cases, the first step of homogenization procedure requires the solution of a three - dimensional cell problem. In this case, cell equations form a system of three partial equations (for illustration purposes, we show only the first equation in Appendix B, the two others resulting in a similar manner). Contrarily to the one - dimensional problem that we have dealt with up to now, this problem has no analytical solution. In order to solve that kind of problems we use FEA commercial softwares, such as DS Simulia Abaqus. By using DS Simulia Abaqus we can solve it completely, but in this case we should take under consideration the computational cost. As an alternatice solution, in this paper we combine the commercial software with the homogenization method mentioned before, following a two - step strategy. In Fig. 1 (a), (b) and (d) the problem is one–dimensional and can be solved using the solution that is presented in this paper. In subfigure (c) the problem is two–dimensional, whithout an analytical solution, and a FEA method in required. The combination of analytical solutions with FEA solution leads to the global problem solution. It should be noticed that despite the fact that generalized periodicity function  $\rho$  is not the same in subfigures (a), (b) and (d), we are able to use the same code due to the fact that this code is parametrized in terms, among other, of the function  $\rho$ . This homogenization scheme will be clarified in the third example of the next section.

#### 4. Numerical examples

In this section we present three examples of multi - step homogenization: the first example is characterized by a linear generalized periodicity, while the second and the third examples exhibit a non - linear generalized periodicity.

#### 4.1. "Chevron" structure

The first example is the two - dimensional, two - phase, second - rank laminate shown in Fig. 2. As mentioned, the widths  $\frac{\epsilon^{II}}{2}$  of the slabs should be much larger than the thicknesses of the layers within each slab. The layers within each slab form angles  $\theta_1$  and  $\theta_2$ , not necessarly equal. The case  $\theta_1 = 30^\circ$ ,  $\theta_2 = 60^\circ$  is considered. The constituents are assumed isotropic with mechanical properties shown in Tables 1 and 2.

In order to solve this problem we will follow the procedure of two - step homogenization. According to this procedure, the problem will be solved first to the direction normal to the layers of the initial composite and as a result we will obtain a "new intermediate" composite. Next we will apply the same procedure along  $x_1$  axis for the constituents of the "new intermediate" composite, resulting from the first step. It is obvious that two generalized functions are required. The generalized periodicity function for the first step of the homogenization is  $\varrho^I = x_2 - x_1 \tan \theta$  with gradients  $\frac{\partial \varrho^I}{\partial x_1} = -\tan \theta$  and  $\frac{\partial \varrho^I}{\partial x_2} = 1$ . The generalized periodicity function for the second step is  $\varrho^{II} = x_1$  with gradients  $\frac{\partial \varrho^{II}}{\partial x_1} = 1$  and  $\frac{\partial \varrho^{II}}{\partial x_2} = 0$ .

The unit cell of the first step of homogenization (parameter of heterogeneity  $\epsilon^I$ ) consists of 90% metal and 10% ceramic, while the volume fractions for the second step depend on the geometry and are depicted in Table 3. We compute the effective elastic matrix from equation (28) using a *MATLAB* code that we have developed.

This is the simplest case since in both steps only two materials (the initial constituents in the first step and the results of the first step as constituents in the second step) are used.

Table 1. "Chevron" structure. Mechanical properties of constituents (Khatam & Pindera, 2009).

Property	Metal	Ceramic
Young Modulus (GPa)	72.400	420.000
Poisson Ratio	0.330	0.250

Table 2. "Chevron" structure. Elastic (symmetric) matrix coefficients of the two constituents (in GPa).

Mechanical Properties	Metal	Ceramic
$C_{11}, C_{22}, C_{33}$	107.271	504.000
$C_{12}, C_{13}, C_{23}$	52.835	168.000
$C_{44}, C_{55}, C_{66}$	27.218	168.000
$C_{14}, C_{15}, C_{16}$	0	0
$C_{24}, C_{25}, C_{26}$	0	0
$C_{34}, C_{35}, C_{36}$	0	0

Table 3. "Chevron" structure. Volume fractions of constituents of the second step of homogenization.

		1 0
Geometry	Phase 1 volume fraction	Phase 2 volume fraction
$(\theta_1 \text{ vs } \theta_2 \text{ angle})$	$(\theta_1 \text{ angle})$	$(\theta_2 \text{ angle})$
30 vs 60	75.000 %	25.000 %

The results of the first step of homogenization are depicted in Tables 4 and 5, while the effective elastic matrix for the whole unit cell is depicted in Table 7. Comparing Table 7 with Table 8, we conclude that the results of the present approach are in very good agreement with the results of the DIPH (see Tsalis et al. (2015)).

From Table 7 it is obvious that, despite the fact that the composite is made of isotropic elastic materials, it behaves like a monoclinic material. In other words, as it is expected, homogenization causes anisotropy.

**Table 4.** "Chevron" structure. Effective elastic matrix for  $\theta_1 = 30^\circ$  from the first step of homogenization (in GPa).

$$C^{eff\ I\ \theta_1} = \left( \begin{array}{cccccccc} 131.579 & 61.306 & 60.210 & 0 & 0 & 9.403 \\ 61.306 & 117.610 & 57.067 & 0 & 0 & 2.694 \\ 60.210 & 57.067 & 144.373 & 0 & 0 & 2.722 \\ 0 & 0 & 0 & 32.605 & 5.018 & 0 \\ 0 & 0 & 0 & 5.018 & 38.399 & 0 \\ 9.403 & 2.694 & 2.722 & 0 & 0 & 35.518 \end{array} \right)$$

**Table 5.** "Chevron" structure. Effective elastic matrix for  $\theta_2 = 60^\circ$  from the first step of homogenization (in GPa).

$$C^{eff\ I\ \theta_2} = \begin{pmatrix} 117.610 & 61.306 & 57.067 & 0 & 0 & -2.694 \\ 61.306 & 131.579 & 60.210 & 0 & 0 & -9.403 \\ 57.067 & 60.210 & 144.373 & 0 & 0 & -2.722 \\ 0 & 0 & 0 & 38.399 & -5.018 & 0 \\ 0 & 0 & 0 & -5.018 & 32.605 & 0 \\ -2.694 & -9.403 & -2.722 & 0 & 0 & 35.518 \end{pmatrix}$$

**Table 6.** "Chevron" structure. Effective elastic matrix for  $\theta_2=30^\circ$  from the first step of homogenization (in GPa).

$$C^{eff\ II\ \theta_2} = \begin{pmatrix} 131.579 & 61.306 & 60.210 & 0 & 0 & -9.403 \\ 61.306 & 117.610 & 57.067 & 0 & 0 & -2.694 \\ 60.210 & 57.067 & 144.373 & 0 & 0 & -2.722 \\ 0 & 0 & 0 & 32.605 & -5.018 & 0 \\ 0 & 0 & 0 & -5.018 & 38.399 & 0 \\ -9.403 & -2.694 & -2.722 & 0 & 0 & 35.518 \end{pmatrix}$$

**Table 7.** "Chevron" structure,  $\theta_1 = 30^\circ$ ,  $\theta_2 = 60^\circ$ . Effective elastic matrix from the second step of homogenization by the proposed method (in GPa).

$$C^{eff\ I} = \begin{pmatrix} 127.017 & 60.536 & 59.010 & 0 & 0 & 6.119 \\ 60.536 & 120.330 & 57.506 & 0 & 0 & -0.328 \\ 59.010 & 57.503 & 144.202 & 0 & 0 & 1.303 \\ 0 & 0 & 0 & 33.499 & 2.189 & 0 \\ 0 & 0 & 0 & 2.189 & 36.766 & 0 \\ 6.119 & -0.328 & 1.303 & 0 & 0 & 35.291 \end{pmatrix}$$

**Table 8.** "Chevron" structure,  $\theta_1 = 30^\circ$ ,  $\theta_2 = 60^\circ$ . Effective elastic matrix by DIPH (in GPa) (Tsalis et al. (2015)).

$$\tilde{C}^{eff\ I} = \begin{pmatrix} 128.087 & 61.306 & 59.424 & 0 & 0 & 6.379 \\ 61.306 & 121.102 & 57.853 & 0 & 0 & -0.330 \\ 59.424 & 57.853 & 144.373 & 0 & 0 & 1.361 \\ 0 & 0 & 0 & 35.053 & 2.509 & 0 \\ 0 & 0 & 0 & 2.509 & 36.950 & 0 \\ 6.379 & -0.330 & 1.361 & 0 & 0 & 35.518 \end{pmatrix}$$

#### 4.2. Matrix reinforced by wavy layers

The second example presented in this paper is a matrix with wavy multilayered inclusions (see Fig. 3), made of the isotropic materials with properties depicted in Tables 9 and 10 and a waviness value 0.20. We recall that the waviness is defined as the ratio of the heigh to the length of the wave. The volume fraction of the reinforced part is 50% and the rest 50% is pure matrix, while the volume fractions in the reinforced part are 50% for both phases. We follow the same approximation as in the "chevron" examples. The main difference is located in the second step: the result of the first step of homogenization depends on the choice of the mesh width following the  $x_1$  axis and this choice governs the second step of homogenization, where more than two phases exist.

Property	Inclusions	Matrix
Young Modulus (GPa)	200.000	20.000
Poisson Ratio	0.300	0.220

Table 10. Matrix with wavy layers. Elastic (symmetric) matrix coefficients of the two constituents (in GPa).

Mechanical Properties	Inclusions	Matrix
$C_{11}, C_{22}, C_{33}$	269.231	22.834
$C_{12}, C_{13}, C_{23}$	115.385	6.440
$C_{44}, C_{55}, C_{66}$	76.923	8.197
$C_{14}, C_{15}, C_{16}$	0	0
$C_{24}, C_{25}, C_{26}$	0	0
$C_{34}, C_{35}, C_{36}$	0	0

The generalized periodicity function for the first step of the homogenization is  $\rho^I = x_2 + H \sin(\frac{2\pi}{L}x_1)$  with gradients  $\frac{\partial \rho^I}{\partial x_1} = 2\pi \frac{H}{L} \cos(\frac{2\pi}{L}x_1)$  and  $\frac{\partial \rho^I}{\partial x_2} = 1$ . The generalized periodicity function for the second step of the homogenization is  $\rho^{II} = x_1$  with gradients  $\frac{\partial \rho^{II}}{\partial x_1} = 1$  and  $\frac{\partial \rho^{II}}{\partial x_2} = 0$ . We consider that the length of the unit cell is L = 1 and we choose a number of 21 points in each of which the wave

We consider that the length of the unit cell is L = 1 and we choose a number of 21 points in each of which the wave curve can be considered having a specific slope with the  $x_1$  axis. The strategy is initially to solve 21 cell problems for a unit cell made of two isotropic elastic materials and, in the sequel, to solve 1 cell problem for a unit cell made of 21 anisotropic elastic materials resulting from the first step. In Fig. 6 we see the distribution of the elastic coefficients of the material resulting from the first step of homogenization, that behaves like a monoclinic material. The coefficients from the second step of homogenization are depicted in Table 11 and correspond to an orthotropic material. We can compare them with with the results of DIPH in Table 12 (Tsalis et al. (2015)) and we observe that the second method overestimates 8% the shear coefficients, responsible for the shear stress in  $x_1x_2$  - plane. In Fig. 7 we compare the effective stresses obtained by the two methods. Recalling the results for the shearing in Tsalis et al. (2015), it seems that the present method gives a more realistic effective behavior for the simple shearing.

#### 4.3. Matrix reinforced by wavy fibers in two directions

Wavy fiber reinforced composite exhibit a nice mechanical behavior (see Zesers & Krumins (2014) for applications of this type of composites). In this example we consider a three - dimensional, two - phase composite consisting of a matrix reinforced by wavy fibers in two directions (see Fig. 8 and 9), parallel to  $x_2$ - and  $x_1$ - axis respectively, ordered vertically to  $x_3$ - axis and forming two successive layers of thickness  $\epsilon^{II}$  much larger than the distance  $\epsilon^{I}$  between the wavy fibers. Then, the composite can be considered as a multilayered with two types of successive reinforced layers.



**Fig. 6.** Matrix with wavy layers. Distributions of coefficients (a)  $C_{11}$ ,  $C_{22}$  and  $C_{33}$  (b)  $C_{44}$ ,  $C_{55}$  and  $C_{66}$ , (c)  $C_{12}$ ,  $C_{23}$  and  $C_{31}$  and (d)  $C_{16}$ ,  $C_{16}$  and  $C_{26}$  of "effective" elastic matrix along  $x_1$  axis from the first step of homogenization.

Table 11. Matrix with wavy layers. Effective elastic matrix from the second step of homogenization (in GPa).

$$C^{eff} = \begin{pmatrix} 34.731 & 12.568 & 11.966 & 0 & 0 & 0\\ 12.567 & 37.739 & 12.888 & 0 & 0 & 0\\ 11.966 & 12.888 & 71.040 & 0 & 0 & 0\\ 0 & 0 & 0 & 17.044 & 0 & 0\\ 0 & 0 & 0 & 0 & 13.956 & 0\\ 0 & 0 & 0 & 0 & 0 & 12.754 \end{pmatrix}$$

Table 12. Matrix with wavy layers. Effective elastic matrix from the second step of homogenization by DIPH (in GPa).

$$\tilde{C}^{eff} = \begin{pmatrix} 35.299 & 11.848 & 12.006 & 0 & 0 & 0 \\ 11.848 & 38.337 & 12.889 & 0 & 0 & 0 \\ 11.920 & 12.851 & 71.416 & 0 & 0 & 0 \\ 0 & 0 & 0 & 16.928 & 0 & 0 \\ 0 & 0 & 0 & 0 & 14.800 & 0 \\ 0 & 0 & 0 & 0 & 0 & 13.735 \end{pmatrix}$$



**Fig. 7.** Matrix with wavy layers. Macrostresses (a)  $\Sigma_1$ , (b)  $\Sigma_2$ , (c)  $\Sigma_3$  vs macrostrain  $E_1$  and (d)  $\Sigma_6$  vs macrostrain  $E_6$ , where MGPH (Multi - step Generalized Periodic Homogenization) denotes the approach presented in this paper.

Each layer consists of a matrix reinforced by fibers of waviness 0.20. The volume fraction of fiber into each layer is 10%. The first layer is piled with angle  $0^{\circ}$  and the other with angle  $90^{\circ}$  and so on (see Fig. 8 and 9). The properties of both matrix and fiber are shown in Tables 13 and 14.

Property	Fiber	Matrix
Young Modulus (GPa)	276.000	3.000
Poisson Ratio	0.300	0.300

Table 14. Matrix reinforced by wavy fibers in two directions. Elastic (symmetric) matrix coefficients of the two constituents (in GPa).

Mechanical Properties	Fiber	Matrix
$C_{11}, C_{22}, C_{33}$	371.538	4.038
$C_{12}, C_{13}, C_{23}$	159.231	1.731
$C_{44}, C_{55}, C_{66}$	106.154	1.154
$C_{14}, C_{15}, C_{16}$	0	0
$C_{24}, C_{25}, C_{26}$	0	0
$C_{34}, C_{35}, C_{36}$	0	0



**Fig. 8.** Matrix reinforced by wavy fibers in two directions. Laminate stucture consists of fiber reinforced layers, piled with angles  $0^{\circ}$  (layer A) and  $90^{\circ}$  (layer B). For more details for layers A and B see Fig. 9.

In order to solve this problem, we follow a two - step strategy. In the first step, we compute the effective elastic modulus in the RVE of each layer, that consists of a fiber reinforced matrix cube (see Fig. 10, 11, 12 and 13). This is a three - dimensional problem that has no analytical solution, as mentioned before. In this case, we use the FEA commercial software DS Simulia Abaqus v.13 - 1 in order to solve the cell problem.

In the second step, where we will use the result of the first step as input data, we will apply the homogenization method as has been mentioned above (see "chevron" example). The volume fraction for each layer is 50.0%. This is an one -



**Fig. 9.** Matrix reinforced by wavy fibers in two directions. Layer A (angle  $0^{\circ}$ ) is shown on the left and layer B (angle  $90^{\circ}$ ) is shown on the right.



Fig. 10. Matrix reinforced by wavy fibers in two directions. Meshing of the fiber for the first step of homogenization.



**Fig. 11.** Matrix reinforced by wavy fibers in two directions. (a) Displacement loading of the fiber and (b) the corresponding shear stress field of the fiber for the first step of homogenization.

**Table 15.** Matrix reinforced by wavy fibers in two directions. Layer A (angle  $0^{\circ}$ ). Effective elastic matrix from the first step of homogenization by DS Simulia Abaqus (in GPa).

$$C^{eff\ I\ A} = \begin{pmatrix} 20.320 & 2.010 & 2.260 & 0 & 0 & 0 \\ 2.010 & 4.810 & 2.000 & 0 & 0 & 0 \\ 2.260 & 2.000 & 5.040 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1.390 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1.620 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1.500 \end{pmatrix}$$

**Table 16.** Matrix reinforced by wavy fibers in two directions. Layer B (angle 90°). Effective elastic matrix from the first step of homogenization by DS Simulia Abaqus (in GPa).

$$C^{eff\ I\ B} = \begin{pmatrix} 4.810 & 2.010 & 2.000 & 0 & 0 & 0 \\ 2.010 & 20.320 & 2.260 & 0 & 0 & 0 \\ 2.000 & 2.260 & 5.040 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1.620 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1.390 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1.500 \end{pmatrix}$$

dimensional problem with two equal volume phases. The generalized periodicity function for the second step is  $\rho^{II} = x_3$  with gradient  $\frac{\partial \rho^{II}}{\partial x_3} = 1$ . The results of the first step of homogenization are shown in Tables 15 and 16 while the results of the second step are shown in Table 17.

This example illustrates the advantages of the proposed method, namely the lower computational cost, the flexibility and the great adaptativity to many cases of "hierarchical" multi - scale composites, comparing to a full FE method.

We observe that with an amount of total reinforcement 10% we managed a significant improvement of the behaviour of composite as shown in Table 22.

#### 5. Conclusions

In this paper, we presented a multi - step homogenization scheme for a sequentially laminated composite made of elastic isotropic materials. The microstructure exhibits a sequence of linear or non - linear generalized periodicities. We formulated the microproblem and gave the analytical solution of it, as well as the effective coefficients at every step of the homogenization process. Moreover, we presented a combination of this method with computational homogenization techniques in order to reduce the computational cost of three - dimensional problems. The advantages of the proposed method are the lower computational cost, the flexibility and the great adaptativity to many cases of multiscale composites, comparing to a full FE method. Finally, we presented three numerical examples of engineering microstructures, namely a "chevron" - like multilayer composite, a matrix reinforced by wavy layers and a matrix reinforced with wavy fibers in two directions.

Table 17. Matrix reinforced by wavy fibers in two directions. Effective elastic matrix from the second step of homogenization (in GPa).

$$C^{eff\ II} = \begin{pmatrix} 12.565 & 2.010 & 2.130 & 0 & 0 & 0 \\ 2.010 & 12.565 & 2.130 & 0 & 0 & 0 \\ 2.130 & 2.130 & 5.040 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1.505 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1.505 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1.500 \end{pmatrix}$$

**Table 18.** Matrix reinforced by wavy fibers in two directions. Percent improvement of the effective elastic matrix comparing to pure matrix.

(	+211.169%	+16.118%	+23.050%	0	0	0 )	•
	+16.118%	+211.169%	+23.050%	0	0	0	Ì
	+23.050%	+23.050%	+24.814%	0	0	0	
	0	0	0	+30.416%	0	0	
	0	0	0	0	+30.416%	0	
	0	0	0	0	0	+29.983%	I



Fig. 12. Matrix reinforced by wavy fibers in two directions. Meshing of the RVE for the first step of homogenization.



(b)

**Fig. 13.** Matrix reinforced by wavy fibers in two directions. (a) Displacement loading of the fiber and (b) the corresponding shear stress field of the RVE for the first step of homogenization.

Appendix A. The system of equation (38) in matrix form.

	; ;	0 : :		1
	: :	0::	- : :	0
	: :	⊣ : :	o : :	0
	0 -1	000	000	0
	0 0	: 0 0	: 1 -	:
	0 0	: 1 -	: 0 0	:
	$^{-1}$	: 0 0	: 0 0	:
	$\sum_{i}^{c_{1}} \binom{(1)A_{3}^{1} - (2)A_{3}^{1}}{c_{i}((2)A_{3}^{1} - (3)A_{3}^{1})}$	$ \begin{split} & \sum_{i}^{n-1} c_{i}((n-1)A_{3}) &+ (1-\sum_{i=1}^{n-1} c_{i}^{(n)}A_{3}^{(n)}) \\ & c_{1}((1)A_{3}^{2} - (2)A_{3}^{2}) \\ & \sum_{i}^{2} c_{i}((2)A_{3}^{2} - (3)A_{3}^{2}) \end{split} $	$ \begin{split} & \sum_{i}^{n-1} c_{i}((n-1)A_{3}^{2}) \cdots \\ & c_{i}(1)A_{3}^{2} - (2A_{3}^{2}) \\ & c_{1}(1)A_{3}^{2} - (2A_{3}^{2}) \\ & \sum_{i}^{2} c_{i}(2A_{3}^{2} - (3)A_{3}^{3}) \end{split} $	$\sum_{i}^{n-1} c_{i}((n-1)A_{3}^{3}) + (1 - \sum_{i}^{n-1} c_{i}^{(n)}A_{3}^{3}))$
	$\sum_{i=1}^{c_1} {(1)}_{A_2} {1 \atop -(2)}_{A_2} {-(2)}_{A_2} {1 \atop -(3)}_{A_2} {1 \atop A_2}$	$ \begin{split} & \sum_{i}^{n-1} c_i ((n-1)A_1^1) + (1-\sum_{i}^{n-1} c_i^{(n)}A_1^1)) \\ & \qquad \qquad$	$ \begin{split} & \sum_{i}^{n-1} c_{i}((n-1)A_{2}^{2}) + (1-\sum_{i}^{n-1} c_{i}(n)A_{2}^{2})) \\ & \sum_{i=1}^{n} ((1)A_{2}^{2} - (2)A_{2}^{2}) \\ & \sum_{i}^{2} c_{i}((2)A_{2}^{2} - (3)A_{2}^{2}) \end{split} $	$\sum_{i}^{n-1} c_i(^{(n-1)}A_2^3) + (1 - \sum_{i}^{n-1} c_i^{(n)}A_2^3))$
$\mathfrak{c}$ of coefficients $ ilde{\mathcal{C}}$	$\Big( \sum_{i}^{c_1} \frac{(1)}{c_i} \frac{A_1^1}{A_1^1} - \frac{(2)}{-(3)} \frac{A_1^1}{A_1^1} \Big)$	$ \begin{array}{l} \sum_{i}^{n-1} c_{i} \left( \left( n-1 \right) A_{1}^{\dagger} \right) + \left( 1- \sum_{i}^{n-1} c_{i}^{\dagger} \left( n \right) A_{1}^{\dagger} \right) ) \\ \sum_{c_{1} \left( \left( 1 \right) A_{1}^{2} - \left( 2 \right) A_{1}^{2} \right) \\ \sum_{i}^{2} c_{i} \left( \left( 2 \right) A_{1}^{2} - \left( 3 \right) A_{1}^{2} \right) \end{array} $	$ \begin{array}{l} \sum_{i}^{n-1} c_{i} \left( \left( n - 1 \right) A_{1}^{2} \right) + \left( 1 - \sum_{i}^{n-1} c_{i}^{2} \left( n \right) A_{1}^{2} \right) \\ \sum_{c_{1} \left( \left( 1 \right) A_{1}^{2} - \left( 2 \right) A_{1}^{2} \right) \\ \sum_{i}^{2} c_{i} \left( \left( 2 \right) A_{1}^{2} - \left( 3 \right) A_{1}^{2} \right) \end{array} $	$\left(\begin{array}{c} \sum_{i}^{n-1} c_{i}((n-1)A_{1}^{3}) + (1-\sum_{i}^{n-1}c_{i}^{(n)}A_{1}^{3})) \\ \end{array}\right)$
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Ta				

Table 20. Matrix of coefficients  $\tilde{U}$ 

$$\tilde{\boldsymbol{U}} = \left( \begin{array}{ccc} con_1^{\boldsymbol{a}} & con_2^{\boldsymbol{a}} & con_3^{\boldsymbol{a}} & ^{(2)}con_1^* \, \boldsymbol{a} & ^{(2)}con_2^* \, \boldsymbol{a} & ^{(2)}con_3^* \, \boldsymbol{a} & \dots & ^{(n-1)}con_1^* \, \boldsymbol{a} & ^{(n-1)}con_2^* \, \boldsymbol{a} & ^{(n-1)}con_3^* \, \boldsymbol{a} \end{array} \right)^T,$$

$$\tilde{\mathcal{S}} = \begin{pmatrix} c_1^{(1)}A_1^{1}(1)d_1^a + c_1^{(1)}A_2^{1}(1)d_2^a + c_1^{(1)}A_3^{1}(1)d_3^a - c_1^{(2)}A_1^{1}(3)d_1^a - c_1^{(2)}A_1^{1}(3)d_1^a - c_1^{(2)}A_1^{1}(3)d_3^a - c_1^{(2)}A_3^{1}(3)d_3^a - c_1^{(2)}A_3^{1}(3)d_3^a - c_1^{(2)}A_3^{1}(3)d_3^a - c_1^{(2)}A_3^{1}(3)d_3^a - c_1^{(2)}A_3^{1}(3)d_3^a - c_1^{(2)}A_2^{1}(3)d_3^a - c_1^{(2)}A_2^{1}(3)d_3^a - c_1^{(2)}A_2^{1}(3)d_3^a - c_1^{(2)}A_2^{1}(3)d_3^a - c_1^{(2)}A_2^{1}(3)d_3^a - c_1^{(2)}A_2^{2}(3)d_3^a - c_1^{(2)}A_2^{2}(3)d_3^a - c_1^{(2)}A_3^{2}(3)d_3^a - c_1^{(2)}A_2^{2}(3)d_3^a - c_1^{(2)}A_2^{2}(3)d_3^a - c_1^{(2)}A_3^{2}(3)d_3^a - c_1^{(2$$

Table 21. Matrix of coefficients  $\tilde{S}$ 

**Table 22.** Sets of indices for the coefficients of  $a_{ij}$ ,  $b_{ij}$  and  $c_{ij}$  of Appendix B.

1	i	b	k	l	m	n	p	q
ľ	1	1	1	1	1	6	1	5
	1	2	6	1	6	6	6	5
	1	3	5	1	5	6	5	5
ľ	2	1	1	6	1	2	1	4
	2	2	6	6	6	2	6	4
	2	3	5	6	5	2	5	4
	3	1	1	5	1	4	1	3
	3	2	6	5	6	4	6	3
ſ	3	3	5	5	5	4	5	3

## Appendix B. The first equation of the cell problem for the three dimensional case

The first equation for a = 1, 2, 3, 4, 5, 6 and for i = 1, 2, 3

$$\frac{\partial}{\partial y_i} d^a_{1i} = -\frac{\partial}{\partial y_1} \left( \sum_{k=1}^3 \left( \sum_{l=1}^3 \frac{\partial w^a_k}{\partial y_l} a_{kl} \right) \right) - \frac{\partial}{\partial y_2} \left( \sum_{k=1}^3 \left( \sum_{l=1}^3 \frac{\partial w^a_k}{\partial y_l} b_{kl} \right) \right) - \frac{\partial}{\partial y_3} \left( \sum_{k=1}^3 \left( \sum_{l=1}^3 \frac{\partial w^a_k}{\partial y_l} c_{kl} \right) \right), \quad (B.1)$$

where

$$d_{1i}^{a} = \frac{\partial \varrho_{i}}{\partial x_{1}} C_{1a} + \frac{\partial \varrho_{i}}{\partial x_{2}} C_{6a} + \frac{\partial \varrho_{i}}{\partial x_{3}} C_{5a}, \tag{B.2}$$

and for j=1,2,3

$$a_{ij} = \sum_{b=1}^{3} \frac{\partial \varrho_1}{\partial x_b} \left( \frac{\partial \varrho_j}{\partial x_1} C_{kl} + \frac{\partial \varrho_j}{\partial x_2} C_{mn} + \frac{\partial \varrho_j}{\partial x_3} C_{pq} \right), \tag{B.3}$$

$$b_{ij} = \sum_{b=1}^{3} \frac{\partial \varrho_2}{\partial x_b} \left( \frac{\partial \varrho_j}{\partial x_1} C_{kl} + \frac{\partial \varrho_j}{\partial x_2} C_{mn} + \frac{\partial \varrho_j}{\partial x_3} C_{pq} \right), \tag{B.4}$$

$$c_{ij} = \sum_{b=1}^{3} \frac{\partial \varrho_3}{\partial x_b} \left( \frac{\partial \varrho_j}{\partial x_1} C_{kl} + \frac{\partial \varrho_j}{\partial x_2} C_{mn} + \frac{\partial \varrho_j}{\partial x_3} C_{pq} \right), \tag{B.5}$$

where the combination of coefficients are shown in Table 22.

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