



Implementation of a new discrete Immersed Boundary Method in OpenFOAM

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Context

Vortex Induced Vibration (VIV) affect

Stealth of military aircrafts and cause fatigue damage on the risers used in offshore petroleum production.



Objectives VIV modeling and creation of optimal control laws

Solver

Implementation of a solver able to model **fluid** – **structure interactions**



DGA

Journée des Utilisateurs OpenFoam



Object Oriented Library



- Implementation of an object oriented library « Immersed Boundary Method » (Pinelli & al .JCP 2010) in the Openfoam code (tested in version 2.1.1 ; 2.2.1 ;2.2.2; 2.2.X; 3.0.1)
- Able to handle **complex**, moving and deformable bodies









Immersed Boundary Method

Volume force integrated in N-S equations to model the Geometry :

$$\frac{\partial \mathbf{u}}{\partial t} + \nabla \cdot (\mathbf{u} \cdot \mathbf{u}) = -\nabla p + \frac{1}{Re} \nabla^2 \mathbf{u} + f(\mathbf{u})$$

With :

Force term on a lagrangian grid

 $\mathbf{F}_{s} = \frac{\mathbf{U}_{s}^{d} - \mathcal{I}[\mathbf{u}]_{s}}{\Delta t} \quad \text{and} \quad \begin{array}{c} U^{d} = U \text{ desired on } a \\ lagrangian \text{ point} \end{array}$

Interpolation of the velocity on the lagrangian grid:

$$\mathcal{I}[\mathbf{u}]_{s} = \sum_{j \in D_{s}} \mathbf{u}_{j} \mathbf{W}_{d}(\mathbf{x}_{j} - \mathbf{X}_{s}) \Delta v$$

Spreading the force on the eulerian mesh:

$$\mathcal{S}[\mathbf{F}_k] = \mathbf{f}(\mathbf{x}_j) = \sum_{k \in D_j} \mathbf{F}_k \quad \mathbf{W}_{d}(\mathbf{x}_j - \mathbf{X}_k) \boldsymbol{\epsilon}_k$$



NACA 0016 with Immersed Boundary Method



Aix*Marseille





Immersed Boundary Method

IBM delta function for an uniform cartesian mesh



Eulerian discretization of the IBM force term f around a Lagrangian marker: comparison between 2nd order discretization using a centered scheme and the kernel analytical solution





Aix Marseille

Immersed boundary on an uniform mesh

$$w_d(r) = \begin{cases} \frac{1}{6} \left(5 - 3|r| - \sqrt{-3(1 - |r|)^2 + 1} \right) & 0.5 \le |r| \le 1.5\\ \frac{1}{3} \left(1 + \sqrt{-3r^2 + 1} \right) & |r| \le 0.5\\ 0 & \text{otherwise} \end{cases}$$

where r = (x - s)/d that satisfies the following properties:

1. $w_d(r)$ is continuous $\forall r \in \mathbb{R}$; 2. $w_d(r) = 0$ if $|r| \ge 1.5$; 3. $\sum_l w_d(r-l) = 1, \ \forall l \in \mathbb{N}$; 4. $\sum_l (r-l) w_d(r-l) = 0, \ \forall r, l$;





Immersed Boundary Method

IBM delta function with **rkpm** method for non-uniform mesh

Eulerian discretization of the IBM force term f around a Lagrangian marker: comparison between 2nd order discretization using a centered scheme and the kernel analytical solution



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Zoom on the mesh discretization of an Immersed boundary on a **non-uniform mesh**

Modification of the delta function following Liu et al. 1995:

$$\widetilde{w}_d(x-s) = \sum_{i=0}^n b_i(y,d)(x-s)^i w_d(x-s),$$

where the unknown polynomial coefficients $b_i(y, d)$ are determined by imposing the reproducing conditions

$$\widetilde{m}_0(x) = \int_{\Omega} \widetilde{w}_d(x-s) \, ds = 1$$

$$\widetilde{m}_i(x) = \int_{\Omega} (x-s)^i \, \widetilde{w}_d(x-s) \, ds = 0 \qquad (i=1,...,N),$$



IBM PisoFoam

Step 1: Calculation of a velocity predictor for the (IBM), \hat{u} , by solving the momentum equation without the (IBM) force term:



Step 2: Computation of the (IBM) force term $f^{n+1}(\hat{u}^{n+1})$ from equation (7) on the Eulerian mesh.

Step 3: Compute a first predictor velocity field to initialize the PISO iterative loop, $u^{\star,1}$, $(\nabla \cdot u^{\star,1} \neq 0)$, by solving the discretized momentum equation (1) including the (IBM) force term $f(\hat{u})$ from step 2:

$$\frac{\partial u^{\star,1}}{\partial t} + \nabla \cdot (u^{\star,1}u^{\star,1}) = -\nabla p + \frac{1}{Re} \nabla^2 u^{\star,1} + f(\hat{u})$$

$$\downarrow \downarrow$$
(A.11)

Iterative PISO loop for m = 1, ..., M





Max Piso iterations reached





- Succesfull modification of the PISO solver to import the IBM library
- Submitted paper in « Computers and fluids »
- Good scalability of the solver with the IBM

library



Performance (scability) of the solver as function of the number of processors for a 2D simulation of a flow at Re = 500 past a moving cylinder using 10⁶ Eulerian points and 312 Lagrangian markers.





IBM PisoFoam



Improvement of the near wall accuracy of the method, analytical calculation of the

force derivative : **Step 5:** Solve the Poisson problem $\nabla \cdot [\nabla p^{\star,m+1}] = \nabla \cdot [H] + \nabla \cdot f(\widehat{u})$





Verification : Manufactured Aix Marseille solution



Convergence order for the global force and



no slip condition error

Global Force Error :

$$e_{F_{IBM}} = |\sum_{k \in D_j} (\mathbf{F}_k - \mathbf{F}_a) \boldsymbol{\epsilon}_k |$$

where :

$$\mathbf{F}_a = \frac{\mathbf{U}_k^d - \mathbf{U}_a}{\Delta t}$$

No slip condition Error :

$$e_{noslip} = \parallel U_s - \mathcal{I}[\mathcal{S}[U_k]]_s \parallel_{\infty}$$





Validation on welldocumented test cases



1st test case : 2D fixed cylinder

Re = 30

	a/D	b/D	Cd
Non uniform dx 2,5e-2	0,6	0,52	1,84
Non uniform dx 2e-3	0,58	0,52	1,77
Uniform dx 2.10e-2	0,556	0,53	1,78
Uniform dx 10e-2	0,55	0,53	1,77
Pinelli et al.	0,56	0,52	1,8
Blackburn & Henderson			1,74
Coutanceau & Bouard	0,54	0,54	
Tritton			1,74



Geometrical parameters of the wake for a steady flow at Re = 30

Re = 185

	CL rms	Cd mean	St
Non Uniform 2,5.10e-2	0,447	1,509	0,191
Uniform 2.10e-2	0,436	1,387	0,198
Uniform 10e-2	0,427	1,379	0,198
Pinelli et al. 5.10e-3	0,423	1,43	0,196
Pinelli et al. 1.10e-2	0,428	1,509	0,199
Vanella and Balaras	0,461	1,377	
Guilmineau and Queutey	0,443	1,287	0,195
Lu and Dalton	0,422	1,31	0,195
Williamson			0,193

Geometrical parameters of the wake, lift, drag coefficient and strouhal for the configuration of a fixed cylinder. Numerical and experimental data from literature are provided for comparison.



Vorticity countours evidencing the shedding of large-scale vortices in 2D flow past a fixed circular cylinder at Re = 185







Validation on welldocumented test cases



2nd test case : 3D fixed cylinder



Re = 200; $\lambda_z = 4,5$



Iso-surfaces of the instantaneous Q-criterion (-0.8 < Q < 0.8)

Mean separation angle as a function of the Reynolds number. Error bars corresponding to the min/max values achieved during the duration of the averaging process are shown by the vertical lines on either side of the square symbols.





Lift coefficient CL as a function of the cylinder displacement for the 2D flow past an oscillating cylinder at Re=500 : present results (grey line) vs. Results obtained by Blackburn and Henderson 1999.

Comparisons of the instantaneous vorticity contours between the results obtained by H.M.Blackburn and R.D.Henderson (1999) (left column) and present results (right column) on the conguration of moving cylinder at Re = 500. At 5 different instants (f to j)



Validation on well documented test cases



4th test case : 2D moving cylinder

 $m\ddot{y} + b\dot{y} + ky = F_y$



Scheme of the free oscillations model

Re = 100 with free oscillations



Absolute value of vorticity countours evidencing the shedding of large-scale vortices in 2D flow past a moving circular cylinder with free oscillations at Re = 500

Source	Present	Shiels et al.	Shen et al.	Lee et al.
Α	0.539	0.58	0.57	0.64
f	0.17	0.196	0.190	0.180
C _{D.avg}	2.0854	2.22	2.15	2.11



Amplitude, Frequency and Drag for the configuration of a moving cylinder with **free oscillations**. Numerical and experimental data from literature are provided for comparison.





Work in Progress



1st study: turbulent flows

Creation of turbulent dynamics libraries :

- incompressibleIBMRAS.so
- incompressibleIBMLES.so

Includes for now:

- IBM_Spalart_Allmaras
- IBM_DDES_Spalart_Allmaras

IBM on non-uniform meshes

Classical Boundary : Re = 3900



Immersed Boundary : Re = 3900



Vt countours in 2D flow past a fixed circular cylinder







Work in Progress

(0,1,0)

(0,0,0)

(1,0,0) S



2nd study: deformable bodies

Finite Element Method

Small displacement assumption :

$$\begin{split} \mathbf{E} &= \frac{1}{2} \left(\mathbf{F}^{\mathrm{T}} \mathbf{F} - \mathbf{1} \right) \\ &= \frac{1}{2} \left(\mathbf{D}_{\mathbf{X}} \boldsymbol{\varphi}^{\mathrm{T}} \mathbf{D}_{\mathbf{X}} \boldsymbol{\varphi} - \mathbf{1} \right) \\ &= \frac{1}{2} \left(\mathbf{D}_{\mathbf{X}} \left(\mathbf{u} + \mathbf{X} \right)^{\mathrm{T}} \mathbf{D}_{\mathbf{X}} \left(\mathbf{u} + \mathbf{X} \right) - \mathbf{1} \right) \\ &= \frac{1}{2} \left(\mathbf{D}_{\mathbf{X}} \mathbf{u}^{\mathrm{T}} \mathbf{D}_{\mathbf{X}} \mathbf{u} + \mathbf{D}_{\mathbf{X}} \mathbf{u} + \mathbf{D}_{\mathbf{X}} \mathbf{u}^{\mathrm{T}} \right) \\ &\approx \frac{1}{2} \left(\mathbf{D}_{\mathbf{X}} \mathbf{u} + \mathbf{D}_{\mathbf{X}} \mathbf{u}^{\mathrm{T}} \right) \end{split}$$

Behaviour law : generalized Hook's law for isotropic material

S

P1 Elements in 1,2 and 3D

a´ (0,0,1)

(1,0)

$$\sigma_{ij} = \frac{\mathrm{E}}{1+\nu} \left(\varepsilon_{ij} + \frac{\nu}{1-2\nu} \varepsilon_{kk} \delta_{ij} \right)$$

(0,0)

Solver validated on a cantilever beam.

Target geometries :



0









Work in Progress



3rd study: compressible flows

Modification of the solver sonicFoam by the team of Pprime* with the IBM library



U contours calculated by IBM_sonicFoam on a circular and square cylinder

In collaboration with : RIAHI Hamza, MELDI Marcello, GONCALVES DA SILVA Eric



*-Inst PPrime, UPR 3346, CNRS, Université de Poitiers, ENSMA



Include IBM in your solver

```
34 #include "fvCFD.H"
35 #include "singlePhaseTransportModel.H"
36_#include "turbulenceModel.H"
                                     Before Main to declare the IBM structure
37 #include "Interpolation IBM.
38 //
                                                                     * * * * * * //
39
40 int main(int argc, char *argv[])
41 {
      #include "setRootCase.H"
42
43
      #include "createTime.H"
44
      #include "createMesh.H"
45
      #include "createFields.H"
46
                                                 After main to declare the variable
      #include "initContinuitvErrs.H"
47
                                                 and read in the controlDict file
48
      #include "Set Immersed
                                                                     * * * * * * //
49
      11
50
```





Include IBM in your solver

50		
51 i	interpolation_ibm.Create_IBM_case(&mesh, runTime	e,Info);
52 53	<pre>Infoss "\nStarting time loop\n" << endl:</pre>	
54	in oss (notal ting tine toop(n ss chut),	Add common d line hefers the loss of the select
55	while (cuplime loop())	Add command line before the loop of the solver
56		to create the structure IBM
57	Info<< "Time = " << runTime.timeName() <	< nl << endl:
58		
59	<pre>#include "readPISOControls.H"</pre>	
60	<pre>#include "CourantNo.H"</pre>	
61		
62	<pre>// Pressure-velocity PISO corrector</pre>	
63	{	
64	// Momentum predictor	
65		
66	fvVectorMatrix UEqn	
67	(
68	fvm::ddt(U)	
69	+ fvm::div(phi, U)	
70	<pre>+ turbulence->divDevReff(U)</pre>	Add command line during the loop to calculate the forceIB (interpolation
71);	of the force) div kernel force (analytical gradient of the force) and d the
72		divergence of the fold
73	//UEqn.relax();	divergence of the held
74		
75		Can be any scalar field to calculate
76	solve(UEqn == -Tvc::grad(p));	on the wall
70		
78		
80	internolation ibm Force Velocity	Interpolation (&mesh cupTime Info II p forceIB div kernel force d);
81		$receipt contraction (arean, running, runo, o, p, runcerb, uv_k (v_k (v_k (v_k (v_k)),$







Futher efforts



• Implementation of an adjoint solver coupled with the present IBM Method to achieve the control analysis of the VIVs.









Do you have any question ?





M2P2 MELLOR A March Dansis Fair Com	de Varselle	Input Files				(Aix*Marseille		
Coordinates.body_ 0.csv	Coor	Label 1 0.4 2 0.4 1.csv Label	of the lagrangian j	points beç	gin by 0			
	A	В	С		D			
	1 Label	X	Y		Z			
2	2 0	0.5		0	()		
:	з 1	0.4998986145	0.0100685266		()		
4	4 2	0.4995944991	0.0201329701		()		
:	5 3	3 0.4990877771	0.0301892487		()		
(6 4	0.4983786541	0.0402332844		()		
	7 5	0.4974674175	0.0502610037		()		
8	8 6	0.496354437	0.0602683401	2	()		
9	9 7	0.495040164	0.0702512352		()		
1	0 8	0.4935251313	0.0802056404		()		





