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In-plane mechanics of soft architected fibre-reinforced silicone rubber membranes

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Abstract

Silicone rubber membranes reinforced with architected fibre networks were processed with a dedicated apparatus, allowing a control of the fibre content and orientation. The membranes were subjected to tensile loadings combined with continuous and discrete kinematical field measurements (DIC and particle tracking). These tests show that the mechanical behaviour of the membranes is hyperelastic at the first order. They highlight the influence of the fibre content and orientation on both the membrane in-plane deformation and stress levels. They also prove that for the considered fibrous architectures and mechanical loadings, the motion and deformation of fibres is an affine function of the macroscale transformation. These trends are fairly well described by the micromechanical model proposed recently in Bailly et al. (*JMBBM*, 2012). This result proves that these materials are very good candidates for new biomimetic membranes, *e.g.* to improve aortic analogues used for *in vitro* experiments, or existing textiles used for vascular (endo)prostheses.

Key words: Micro-mechanical model, Tensile tests, Biomimetism, Fibre-reinforced membrane, Anisotropy, Material design, Silicone rubber

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1 Introduction

Human abdominal aortic tissue (AA) is a complex soft sandwich structure, arranged in three different layers. Within these layers, the distribution of all components displays a double-helix architecture of collagen fibres, elastic fibres, smooth muscle cells and other molecules of the extracellular matrix (Holzapfel, 2008). The fibrous microstructure is characterised by a particular wavy arrangement under stress-free conditions (Freed and Doehring, 2005; Holzapfel, 2008; Rezakhaniha *et al.*, 2011) and by distinctive preferred orientations when subjected to a mechanical loading (Holzapfel, 2006; Horny *et al.*, 2010). These histological features provide the abdominal aortic tissue a highly non-linear, anisotropic and essentially hyperelastic mechanical behaviour (Vande Geest *et al.*, 2006). Over the past decades, several synthetic materials aiming at mimicking the microstructural and/or mechanical specificities of aortic tissue were developed (i) to perform *in vitro* experiments, or (ii) for bioengineering issues:

(i) *In vitro* experiments were carried out to characterise biomechanical factors involved in the evolution of arterial diseases such as Abdominal Aortic Aneurysm (AAA). For that purpose, several deformable AAA analogues were processed. Most of them were placed into vascular flow simulators to investigate endovascular repair (Schurink *et al.*, 1998; Flora *et al.*, 2002; Chong and How, 2004; Gawenda *et al.*, 2004). Few were utilised to quantify biological fluid-structure interactions within AAAs (Ene *et al.*, 2011; Deplano *et al.*, 2013, 2014). Inflation tests were conducted by increasing internal pressure in order to identify AAAs deformation or rupture locations (Morris *et al.*, 2004; Doyle *et al.*, 2009, 2010; Corbett *et al.*, 2010; Meyer *et al.*, 2011). However, all these materials were made up of homogeneous elastomers with isotropic material properties, which are very far from the anisotropic hyperelastic behaviour of aortic tissue and from its fibrous microstructure.

(ii) In parallel, in the field of vascular tissue engineering, numerous implants were

designed so as to be mechanically as close as possible to the host tissue macroscale mechanical behaviour. To this end, polymeric fibrous composites were largely designed as bioinspired scaffolds (Li and Cooper Jr, 2011). Various manufacturing processes were also developed to elaborate fibrous networks (Ratcliffe, 2000; Chen and Hunt, 2007). Currently, a specific interest is provided to fibre electrospinning and textile processing techniques:

- (a) Electrospinning technology is favoured to manufacture nano-structured biomaterials made up of filaments close to collagen/elastin fibrils dimensions (Matthews *et al.*, 2002; Boland *et al.*, 2004; Agarwal *et al.*, 2008; Clure *et al.*, 2010). Recent advances allow a better control of the electrospun fibres morphology, strength and size by adjusting set-up conditions or polymer solution parameters (Kowalczyk *et al.*, 2008; Soliman *et al.*, 2011). However, the control of the fibres position and orientation is still a challenging task to generate the desired macroscopic mechanical anisotropy (Bu *et al.*, 2012).
- (b) Textile processing techniques are also widely used to develop vascular (endo)prostheses (Jacobs *et al.*, 2003; Chakfe *et al.*, 2004; VanLieshout *et al.*, 2006). Knitted, braided or woven fabrics are characterised by a fine regular arrangement of the fibres, therefore allowing a better control of the macroscopic anisotropic properties (Xu *et al.*, 2010; Schreiber *et al.*, 2010; Yang *et al.*, 2012). However, due to their inappropriate fibrous microstructures, the current commercial textile prostheses do not exhibit suitable deformability (Surovtsova, 2005; Sarkar *et al.*, 2006; Demanget *et al.*, 2012, 2013; Lemerrier *et al.*, 2013).

Therefore, the global objective of this work is to design and process new artificial materials, able to mimic the macroscopic anisotropic properties of AA and AAA tissues, their hyperelastic J-shape mechanical response, and their main histological features (fibrous microstructures with distinctive fibre orientations). In the short run, these materials represent appropriate candidates to perform *in vitro* experiments similar to those carried out in Deplano *et al.* (2014), with more relevant models of AAs or

AAAs. In the long run, they could help improving the design of aortic prostheses towards more efficient mechanical performances. To this end, optimal elastomeric fibrous composites were identified in a previous theoretical work using a multi-scale homogenisation process combined with microstructure optimisation (Bailly *et al.*, 2012). The optimised architected materials are composed of a thin and soft hyperelastic membrane reinforced by periodic fibrous lattices displaying four preferred orientations and hyperelastic fibres. The present study aims at (i) providing experimental advances concerning the manufacturing and mechanical characterisation of such architected soft fibre-reinforced membranes, and (ii) validating the assumptions stated to build the micro-mechanical model in Bailly *et al.* (2012). Thus, a summary of the theoretical background is first reported in Section 2. The manufacturing processes of the elastomeric fibrous membranes are presented in Section 3, along with the method used to characterise their mechanical behaviour. Experimental results are detailed in Section 4. A comparison between these results and the prediction of the micromechanical model is proposed and discussed in Section 5.

2 Theoretical background

To mimic the main histological features of the arterial tissues, various fibrous architectures were investigated with the micro-mechanical model proposed in Bailly *et al.* (2012), *i.e.* single or bi-layers lattices of fibres. As a first step in the validation of the model assumptions, the present study focuses on the *one-layer fibrous structure*, as illustrated in Figure 1(a). This structure comprises a single lattice of crossed, identical continuous and straight fibres of initial cross-section S_0 , embedded into a thin and soft elastomer matrix. Each fibrous lattice can be seen as a repetition of a Representative Elementary Cell (REC) lying in the $(\mathbf{e}_x, \mathbf{e}_y)$ plane, as sketched in Figure 1(b) in the initial configuration \mathcal{C}_0 . The microscale geometrical and mechanical assumptions related to the REC are summarised as follows:

- (i) The REC geometry is characterised by two periodic vectors $\mathbf{P}_1 = \mathbf{A}_1\mathbf{A}_4 = \ell_0\mathbf{E}_1$ and $\mathbf{P}_2 = \mathbf{A}_1\mathbf{A}_2 = \ell_0\mathbf{E}_2$ (resp. $\mathbf{p}_1 = \mathbf{a}_1\mathbf{a}_4 = \ell_1\mathbf{e}_1$ and $\mathbf{p}_2 = \mathbf{a}_1\mathbf{a}_2 = \ell_2\mathbf{e}_2$) joining the extremities A_1, A_2 and A_4 (resp. a_1, a_2 and a_4) of the REC in the initial \mathcal{C}_0 (resp. deformed \mathcal{C}) configuration. The angles $(\mathbf{e}_x, \mathbf{P}_1)$ and $(\mathbf{e}_x, \mathbf{P}_2)$ [resp. $(\mathbf{e}_x, \mathbf{p}_1)$ and $(\mathbf{e}_x, \mathbf{p}_2)$] are noted θ_{01} and θ_{02} (resp. θ_1 and θ_2) in \mathcal{C}_0 (resp. in \mathcal{C}). Initial REC configuration is defined so that: $\theta_{02} = -\theta_{01} = \theta_0$.
- (ii) Fibres are supposed to be linked at the REC extremities A_1, A_2, A_3 and A_4 , in such a way that the relative displacement of fibres at these points is zero. Between these points, the initial and deformed lengths of fibre i ($i = 1$ and 2) are expressed as $\ell_{0i}^f = \|\mathbf{P}_i\| = \ell_0$ and $\ell_i^f = \|\mathbf{p}_i\| = \ell_i$, respectively. Furthermore, its elongation λ_i^f and its tension \mathbf{t}_i^f are expressed as $\lambda_i^f = \ell_i/\ell_0$ and $\mathbf{t}_i^f = t_i^f\mathbf{e}_i$ (no summation on the index i). Fibres are assumed to behave as hyperelastic bars, *i.e.* the tensions t_i^f only depend on the fibre elongations λ_i^f via a fibre strain energy function W^f .
- (iii) The mechanical interactions between the fibres lattice and the homogeneous elastomer membrane, and between the fibres are negligible.

The macroscale mechanical behaviour of such periodic fibrous lattice was entirely determined analytically, by using the homogenisation of periodic discrete structures (Caillerie *et al.*, 2006; Bailly *et al.*, 2012). It was shown that the overall macroscopic Cauchy tension tensor $\boldsymbol{\tau}^c$ of the one-layer fibrous membrane can be written as:

$$\boldsymbol{\tau}^c = \boldsymbol{\tau}^m + \frac{1}{\|\mathbf{P}_1 \times \mathbf{P}_2\|} \sum_{i=1}^2 (t_i^f \otimes \mathbf{p}_i), \quad (1)$$

where $\boldsymbol{\tau}^m$ corresponds to the Cauchy tension tensor of the elastomer membrane contribution. Thus, knowing the mechanical behaviour of the elastomeric membrane, that of the fibres, and the geometry of the REC, the mechanical behaviour of the considered architected and fibre-reinforced membranes is fully estimated: no constitutive parameter has to be fitted *a posteriori*.

3 Experimental procedure

3.1 Materials

Matrix - A biocompatible high strength RTV Silicone elastomer dispersion (PN 40021, Applied Silicone Corporation) was selected as matrix material, due to its high deformability and specific physical properties. It allowed an easy impregnation of the fibrous network and cured at room-temperature. To minimise the scattering of the experimental results, a particular attention was brought to subject the processed membranes to the same curing time (24h) and to keep them at the same room temperature (290K) and relative humidity (50%) until mechanical testing.

The silicone in-plane mechanical response was investigated by means of a uniaxial tensile-testing device (Bose Corporation, Eden Prairie, Minnesota) equipped with a load cell of capacity of 450 ± 0.01 N. Displacement-controlled tests were performed at room temperature on 8 rectangular samples of mean initial gauge $L_0 = 21.21$ mm (STD = 1.07 mm), width $W_0 = 10.13$ mm (STD = 0.47 mm) and thickness $T_0 = 1.05$ mm (STD = 0.22mm), STD being the data standard deviation. Note that these dimensions were determined by photographing the undeformed samples with an accuracy of 0.02 mm. Each sample was subjected to 10 load-unload cycles, with a maximal tensile elongation of 1.30 and an elongation rate of 10^{-2} s^{-1} , representative of AA tissue's physiological loadings (Bailly *et al.*, 2014). During the tests, pictures of the sample's gauge zone were recorded with a JAI CCD camera (1624×1236 pixels, 25 Hz). Before, speckles were drawn on the surface of each strip so that strain field measurements could be carried out using 2D Digital Image Correlation (DIC), as illustrated in Figure 1(d). Elongations of the specimen along the stretch \mathbf{e}_y and the transversal \mathbf{e}_x directions, *i.e.* respectively λ_{yy} and λ_{xx} , were calculated using a DIC software (7D) developed by Vacher *et al.* (1999). Longitudinal force F_y was also measured during the tests and corresponding component of the first Piola-Kirchhoff

stress tensor was defined as $\Pi_{yy} = F_y/W_0T_0$.

Typical experimental data are displayed in Figure 1(d), showing both the elongation path $(\lambda_{xx}, \lambda_{yy})$ and the stress-elongation response (Π_{yy}, λ_{yy}) . A low hysteresis was measured during the load/unload sequences. Therefore, as a first reasonable approximation, it can be inferred that the considered silicone rubber behaved nearly as a hyperelastic material. Besides, within the investigated elongation range, its mechanical behaviour could be fairly well modelled with a classical neo-Hookean model (Treloar, 1943), thus characterised by a volumetric strain-energy function $W^m = c^m(I_1 - 3)/2$. The constant c^m is a positive material parameter, and $I_1 = \text{tr}(\mathbf{C})$ denotes the first principal invariant of the right Cauchy-Green tensor \mathbf{C} . This model was adjusted on both curves, as illustrated in Figure 1(d). For all samples, the average material parameter was found around $c^m = 0.24$ MPa (STD = 0.01 MPa).

Fibre - A 50 μm diameter polyamid monofilament was selected as a fibre candidate. This fibre is commonly used as fishing wire (Feeling 36605, Sensas - 0.8 lb, 0.350 kg). Uniaxial tensile tests were performed to characterise the mechanical behaviour of 16 fibre samples cut from this filament on a dedicated testing machine (Instron 3367, 100 ± 0.01 N). Tests were conducted until failure, using an elongation rate of 10^{-2} s^{-1} , and an initial fibre length of 200 mm at rest. The force within each fibre sample t^f was measured during the tests and the corresponding first Piola-Kirchhoff stress was estimated as $\Pi^f = t^f/S_0$, S_0 being the fibre unloaded cross-section. Figure 1(e) presents the mean stress value Π^f , as function of fibre elongation λ^f , measured in average over the 16 samples. The obtained fibre mechanical behaviour is repeatable and non-linear. Failure loads varied from 1.93 to 2.35 N (mean value 2.21 N), and failure stretches ranged between 1.20 and 1.30 (mean value 1.26). Finally, the experimental data could be described by a hyperelastic model proposed in Org as *et al.* (1998) for Shape Memory Alloys (see Figure 1(e)), so that the mean tension within the fibre is expressed as:

$$t^f = \frac{S_0}{\lambda^f} \left\{ \begin{array}{l} \frac{\mu_1}{2} \left[\epsilon^f + \operatorname{sgn}_\epsilon \{ \epsilon_s - \sqrt{[\epsilon^f + \operatorname{sgn}_\epsilon \epsilon_s]^2 + \alpha_1^2} \} \right] + \operatorname{sgn}_\epsilon \sigma_0 \\ + \mu_2 \epsilon^f \\ + \frac{\mu_3}{2} \left[\epsilon^f + \operatorname{sgn}_\epsilon \{ \sqrt{[\epsilon^f - \operatorname{sgn}_\epsilon \epsilon_c]^2 + \alpha_2^2} - \sqrt{\epsilon_c^2 + \alpha_2^2} \} \right], \end{array} \right. \quad (2)$$

$$\text{with } \epsilon_s = \frac{\alpha_1^2 - (2\sigma_0/\mu_1)^2}{4\sigma_0/\mu_1} \quad \text{and} \quad \operatorname{sgn}_\epsilon = \begin{cases} 1 & \text{if } \epsilon^f \geq 0 \\ -1 & \text{if } \epsilon^f < 0. \end{cases}$$

This tension is a function of the logarithmic strain, $\epsilon^f = \ln \lambda^f$, and of 7 constitutive parameters: μ_1 , μ_2 , μ_3 , σ_0 , ϵ_c , α_1 and α_2 (Org as, 1997). The model parameters were identified from the experimental data given in Figure 1(e) over the stretch range $\lambda^f \in [1 : 1.15]$. The latter covers the range of tensile loadings predicted within the fibres in the experiments below. The identification yielded to $\mu_1 = 2444.94$ MPa, $\mu_2 = 223.16$ MPa, $\mu_3 = 38652.62$ MPa, $\sigma_0 = 42.18$ MPa, $\epsilon_c = 16.25 \times 10^{-2}$, $\alpha_1 = 9.64 \times 10^{-3}$ and $\alpha_2 = 5.20 \times 10^{-2}$.

3.2 *Architected soft fibre-reinforced membranes*

Processing route - The as-described fibre and matrix materials were used to manufacture the target fibre-reinforced membranes. For that purpose, a specific device was designed to manufacture lattices of crossed and identical straight fibres, impregnated inside rectangular membranes (160 mm \times 50 mm), as shown in Figure 2(a). The device consists of a pantographic deployable structure and a TeflonTM rectangular mould. Each arm of the pantograph is equipped with a 1 mm-tooth comb. Three parameters of the device were adjusted before each composite membrane's elaboration:

- (i) the number of teeth involved in the manufacturing (see Figure 2(c)). This parameter allows the control of the initial fibre length of the lattice, ℓ_0 , and thus,

the fibre content. Length ℓ_0 can vary by step of 1 mm.

- (ii) the angulation of the pantograph, adjustable by translation of the arm's cross-junctions as illustrated in Figure 2(b). This parameter allows the control of the initial half-angle between fibres, θ_0 . It can be varied between 20° and 70° by step of 5° .
- (iii) the angle β_0 as defined in Figure 2(c), adjustable by rotation of the mould. This parameter allows to control the orientation of the fibre network in the $(\mathbf{e}_x, \mathbf{e}_y)$ plane. Two cases were investigated: $\beta_0 = \{0^\circ, 45^\circ\}$.

At a chosen configuration $(\ell_0, \theta_0, \beta_0)$, a first unidirectional fibre network was built using the same fibre. The fibre was guided alternatively by two combs of the pantograph facing to each other. In the same way, a second unidirectional network was built using the other combs of the device, and superimposed on the first one. So as to create a 1 mm-thick membrane, the lattice was then embedded by a 20 mL-layer of silicone casted into the mould. Note that the depth of the mould was designed (i) to account for the average mass loss (58%) measured after the silicone curing due to solvent evaporation and shrinkage of the silicone during curing, and (ii) to spatially equilibrate the microstructure along the membrane's thickness. Finally, several composite membranes were processed, characterised by six microstructural arrangements (ℓ_0, θ_0) as follows: (1 mm, 45°), (3 mm, 45°), (1 mm, 30°), (3 mm, 30°), (1 mm, 60°), (3 mm, 60°). These target values of both parameters were chosen in line with previous predictions found in Bailly *et al.* (2012). Typical illustrations of manufactured microstructures are presented in Figure 2(d).

Samples - Eight rectangular strips $(S_i, i \in [1, \dots, 8])$ were cut from the composite membranes. Their geometrical dimensions were determined so as to preserve the same separation scale from a given sample to another one, by considering 6 RECs along the width and 9 RECs along the length. Strip's initial dimensions were optically measured and summarised in Table 1. In the following, for the sake of clarity, the composite strip S_i , characterised by the triplet of parameters $(\ell_0, \theta_0, \beta_0)$, will be

denoted as $S_i(\ell_0, \theta_0, \beta_0)$. In practice, the values of (ℓ_0, θ_0) slightly differed from the target ones. Thus, optical measurements were processed using a particle-tracking method applied on strips' images registered at rest. The image processing was based on the tracking of experimental REC extremities, as illustrated in Figure 3 upon one REC. For each strip, the measured values of (ℓ_0, θ_0) were averaged on all the surface's RECs and noted $(\overline{\ell_0}, \overline{\theta_0})$. Values of $(\overline{\ell_0}, \overline{\theta_0})$ are detailed in Table 1. The accuracy of the tracking method was assessed at $(0.05 \text{ mm}, 0.2^\circ)$. Finally, the target values of (ℓ_0, θ_0) were achieved within a mean precision of $(0.07 \pm 0.05 \text{ mm}, 0.4 \pm 0.2^\circ)$ over the whole database.

Tensile tests - The in-plane mechanical response of the S_i samples was investigated using the tensile-testing device and the protocol previously described for the homogeneous membrane. The stretch direction was defined by the vector \mathbf{e}_y . Thus, for $\beta_0 = 0^\circ$, the stretch direction matches the REC diagonal ($\mathbf{A}_4\mathbf{A}_2$ direction). For $\beta_0 = 45^\circ$, the stretch direction matches the fibres one, as illustrated in Figure 3.

Each test was repeated twice in order to ensure its reproducibility, and coupled with image recording of the deformed strip. In the following, focus is given to the first load of each test only. The discrete kinematical field measured at the fibre scale was derived from the tracking of experimental REC nodes A_i on each image, as displayed in Figure 3. The image processing was repeated three times on four centered RECs for each composite sample. This yielded to average values of the microscopic longitudinal and transversal elongations, *i.e.* λ_{yy} and λ_{xx} respectively. The local value of the half-angle between the fibres was also derived. This angle was noted $\widehat{\theta}_0$ and its evolution was calculated along the test. In summary, for each strip S_i , Table 2 compares (i) the target value of angle θ_0 , (ii) the angle $\overline{\theta_0}$ averaged over all surface's RECs, and (iii) the angle $\widehat{\theta}_0$ averaged over the four RECs associate to the measured elongations $(\lambda_{xx}, \lambda_{yy})$.

Finally, some samples' surfaces were additionally covered by speckles to perform 2D DIC measurements. In that case, the continuous kinematical field applied on the

composite membrane was derived in average over a whole REC. Therefore, kinematics measured at the composite scale (global, DIC measurements) and at the fibre scale (local, particle tracking measurements) could be compared.

4 Experimental results on composite membranes

4.1 General trends

Typical measurements are reported in Figure 3, showing the local loading paths $(\lambda_{xx}, \lambda_{yy})$, the derived local half-angle between the fibres $\hat{\theta}$ as a function of the elongation λ_{yy} , and the corresponding tension-elongation response in the composite membrane $(\tau_{yy}, \lambda_{yy})$. Two tests repeated on samples $S_1(1, 45, 0)$ and $S_2(1, 45, 45)$ are illustrated, highlighting important general features:

- whatever the loading angle β_0 , the repeatability of each test is ensured.
- the local kinematical fields are optically measurable, with an accuracy on angle $\hat{\theta}$ being of 0.2° . During the tests, a typical quasi-linear $\hat{\theta}$ -variation is observed for $\beta_0 = 0^\circ$, against zero θ -variation for $\beta_0 = 45^\circ$ (see second row in Figure 3). This angle variation comes up to 15° at the applied maximal strain when $\beta_0 = 0^\circ$. Note that for each sample, a slight change in the value of initial angle $\hat{\theta}_0$ ($\pm 1/2^\circ$) is observed between the first test and the second one: this is probably due to possible fibrous rearrangement (*e.g.* slight rotation of the fibres due to the low hysteresis measured after loading cycles) or local variations in the sample's architecture (see Table 2).
- the mechanical effects of the fibrous reinforcement are highly dependent on the loading direction. For $\beta_0 = 0^\circ$, the fibres progressively align with the stretch direction (see pictures in Figure 3(a)). Hence, the lattice mechanical contribution combines the rotations of the fibres with their tensile response. These mechanisms typically yield to a drastic contraction of the sample along the transverse

direction (up to $\lambda_{xx} = 0.7$), and low values of the mechanical tension in the overall membrane at $\lambda_{yy} = 1.3$ (maximum value of τ_{yy} below 0.5 N/mm). On the contrary, for $\beta_0 = 45^\circ$, the fibres arranged perpendicular to the stretch direction are slightly compressed. They strongly restrain the membrane width's contraction along the transverse direction ($\lambda_{xx} > 0.97$ during the test). In that case, only the fibres aligned with the loading direction contribute to the overall mechanical tension measured in the composite membrane. Therefore, as the silicone rubber matrix displays soft mechanical properties, the lattice mechanical contribution at the macroscale is nearly equivalent to the composite's one. The lattice contribution also comes to the tensile response of individual fibre, weighted by the volume fraction of the fibres parallel to the loading direction, v_f . This volume fraction is defined by the ratio of the volume of fibres parallel to the loading direction within one REC in configuration C_0 ($\pi d_0^2 \ell_0 / 4$), over the volume of the REC ($T_0 \ell_0^2 \sin(2\theta_0)$). This ratio yields to $v_f = \frac{\pi d_0^2}{4 T_0 \ell_0 \sin(2\theta_0)}$, *i.e.* 0.2% for $S_2(1, 45, 45)$.

4.2 Comparison between global and local kinematics

Figure 4(a) confronts the global kinematics measured at the composite scale by 2D DIC, with the local kinematics measured at the fibre scale by tracking the displacements of the REC nodes A_i . Samples $S_3(3, 45, 0)$ and $S_4(3, 45, 45)$ were chosen for illustration. Whatever the loading direction, a fairly good quantitative agreement was measured between global and local kinematics, as shown by the superimposition of the loading paths derived from both techniques in Figure 4(a). In particular, the relative difference e_r between a couple of stretches determined by DIC ($\lambda_{xx}^{DIC}, \lambda_{yy}^{DIC}$), and the one determined by particle tracking ($\lambda_{xx}^t, \lambda_{yy}^t$) was calculated using:

$$e_r = \frac{[(\lambda_{xx}^{DIC} - \lambda_{xx}^t)^2 + (\lambda_{yy}^{DIC} - \lambda_{yy}^t)^2]^{0.5}}{[(\lambda_{xx}^t - 1)^2 + (\lambda_{yy}^t - 1)^2]^{0.5}} \quad (3)$$

Regarding data series with $\beta_0 = 45^\circ$, the average value, $\bar{\epsilon}_r$, comes to 13.8% for $\lambda_{yy} \in [1 : 1.14]$, and drops down to 6% for $\lambda_{yy} > 1.05$. Regarding data series with $\beta_0 = 0^\circ$, $\bar{\epsilon}_r$ comes to 17.7% for $\lambda_{yy} \in [1 : 1.23]$. It falls below 15% for $\lambda_{yy} > 1.05$. Note that this quantitative agreement is nearly unchanged when calculating the global kinematical field in average over a surface of 3×4 RECs (not shown). From these results, the overall membrane and the REC were considered to be subjected to the same macroscopic transformation gradient (see the discussion below). In the following, the discussed loading paths $(\lambda_{xx}, \lambda_{yy})$ refer exclusively to the elongations calculated at the fibre scale.

4.3 Influence of the loading angle β_0

The graphs of Figure 5(a) compare the loading paths and the Cauchy tension-elongation responses measured on three samples: the isotropic homogeneous silicone membrane, samples $S_1(1, 45, 0)$ and $S_2(1, 45, 45)$. All samples are characterised by a similar initial thickness (relative STD being 6.7 %), which enables a comparison relying on tension curves.

Both the mechanical structural reinforcement and the anisotropy induced by the fibrous lattice are emphasised in the left graph of Figure 5(a). For the silicone membrane, the shrinkage of the sample width is about 10% at $\lambda_{yy} = 1.25$. For composite $S_1(1, 45, 0)$, the progressive alignment of the fibres along the loading direction favours the sample contraction along the x -direction, and yields to a width-shrinkage of about 30% at $\lambda_{yy} = 1.25$. On the contrary, for sample $S_2(1, 45, 45)$, the width's decrease is restrained to about 2% beyond $\lambda_{yy} = 1.05$.

The anisotropy is also highlighted in the right graph of Figure 5(a), showing a global stiffening of the mechanical behaviour induced by the fibrous lattice. The more the fibres are aligned along the stretch direction, the higher the stiffness of the sample and stress levels. A tangent modulus was assessed at a tension value typically

measured in aortic tissue for tensile loadings (Vande Geest *et al.*, 2006; Bailly *et al.*, 2012), *i.e.* $\tau_{yy} = 0.1$ N/mm. The stiffening was characterised by a modulus of 1.13 N/mm for $S_1(1, 45, 0)$ [resp. 19.76 N/mm for $S_2(1, 45, 45)$], thus increased by a ratio 1.3 (resp. 22.5) in comparison to the silicone case (0.88 N/mm). At last, the mechanical responses of the silicone and the sample $S_1(1, 45, 0)$ were initially superimposed. This suggests that, for this fibrous architecture, *i.e.* for $\beta_0 = 0^\circ$ and for values of θ_0 below 45° , the contribution of the lattice is negligible, at least for $\lambda_{yy} < 1.05$.

4.4 Influence of the initial fibre length ℓ_0

The graphs of Figure 5(b) compare the loading paths and the tension-elongation responses measured on five samples of comparable initial thickness (relative STD being 13.7 %): the silicone membrane, samples $S_1(1, 45, 0)$, $S_2(1, 45, 45)$, $S_3(3, 45, 0)$ and $S_4(3, 45, 45)$. The influence of initial fibre length ℓ_0 can be assessed by comparing the mechanical response of $S_1(1, 45, 0)$ and $S_3(3, 45, 0)$, and that of $S_2(1, 45, 45)$ and $S_4(3, 45, 45)$.

Regarding the elongation paths shown in Figure 5(b), the influence of the initial fibre length is noticeable when the loading direction is not aligned with the fibres ones. Indeed, for $\beta_0 = 0^\circ$, the contraction of the sample initial width (15% for $\lambda_{yy} = 1.25$) is twice lower for $\ell_0 = 3$ mm than that observed for $\ell_0 = 1$ mm. By contrast, for $\beta_0 = 45^\circ$, the impact of fibre length ℓ_0 on the sample contraction is negligible.

Regarding the tension-elongation responses shown in Figure 5(b), all composites exhibit a global stiffening as compared to the silicone membrane, except for case $S_3(3, 45, 0)$. The similarity between the responses of $S_3(3, 45, 0)$ and the silicone is mainly ascribed to the differences in thickness samples (ratio of values being 0.76). This similarity disappears when tension data are converted into first Piola-Kirchhoff stresses (not shown). For a given stretch direction β_0 , the samples with

higher volume fraction of fibres (*i.e.* for $\ell_0 = 1$ mm) are stiffer. In particular, the ratio of the tensions achieved for $S_4(3, 45, 45)$ by the ones obtained for $S_2(1, 45, 45)$ varies between 0.20 and 0.41, around a mean value of 0.36. Beyond low elongations ($\lambda_{yy} > 1.03$), this ratio rises up to 0.39 in average. This value comes close to the ratio of the volume fractions of the fibres parallel to the loading direction within each sample (*i.e.* 0.40).

4.5 Influence of the initial half-angle between fibres θ_0

The graphs plotted in Figure 5(c) compare the loading paths and the tension-elongation responses measured on five samples of similar initial thickness (relative STD being 8.2 %) for various values of θ_0 . All composite samples are characterised by an identical initial fibre length ($\ell_0 = 1$ mm) and the same loading direction ($\beta_0 = 0^\circ$).

For $\theta_0 = 30^\circ$, the decrease of the sample's width is very close to the one of the matrix alone, as shown in the left graph of Figure 5(c). Tension-elongation responses are also rather similar. These results imply that, for such lattice architectures and such elongation range, the fibres are not loaded and they weakly disturb the deformation of the silicone.

By contrast, for $\theta_0 = 60^\circ$, the fibres tend to align along the loading direction as soon as $\lambda_{yy} > 1$. Therefore, the reduction of the sample width is much higher than in the previous case ($\theta_0 = 30^\circ$): $\lambda_{xx} = 0.68$ versus $\lambda_{xx} = 0.90$ for $\lambda_{yy} = 1.25$. Furthermore, the elongation paths being similar for cases $\theta_0 = 45^\circ$ and $\theta_0 = 60^\circ$, the influence of the fibres' initial orientation is clearly demonstrated on the corresponding tension-elongation responses. The more the fibres are aligned along the stretch direction, the higher the stiffness of the sample.

Note that above results are consistent with those obtained by comparing samples

$S_3(3, 45, 0)$, $S_7(3, 60, 0)$ and $S_8(3, 30, 0)$ exhibiting a larger initial fibre length, $\ell_0 = 3$ mm (see Figure 6). In particular, the low tension-elongation response measured for $S_8(3, 30, 0)$ is linked with the small thickness of the sample as compared to the matrix (ratio of values being 0.67, *i.e.* the lowest in the database).

5 Validation of the micro-mechanical model

5.1 Relevance of some of the model assumptions

As detailed in sections 2 and 3, some major microscale geometrical and mechanical assumptions were formulated in order to apply the micro-mechanical model. Several experimental observations gained from results shown in section 4 ensure the relevance of these assumptions:

- (i) For the tensile loadings under study, the REC geometry is assumed to be diamond-shaped in the initial (resp. actual) configuration \mathcal{C}_0 (resp. \mathcal{C}). Besides, the displacements of nodes A_i are assumed to be affine functions of the macroscale gradient of the geometrical transformation the samples are subjected to.

The tracking of the experimental RECs extremities showed that their diamond pattern remained topologically unaffected by the transformation (see Figure 3). Besides, Figure 4 also proved that the motion of nodes A_i were identical to the displacement field recorded at the macroscale with the DIC technique. These observations thus validate above assumptions.

- (ii) The fibre mechanical behaviour is predicted by a hyperelastic model (Eq. 2), adjusted on experimental tensile tests performed on fibres (Figure 1(d)). This model assumes a symmetrical behaviour under compressive loading.

It was shown that the fibres are subjected to compressive loading for cases

with $\beta_0 = 45^\circ$. No buckling effect was observed during the considered tests (because the compressive strains remained small and because compressed fibres are confined within the matrix). Thus, the above assumption is reasonable for the considered configurations.

- (iii) Finally, the mechanical interactions between the fibres and the matrix, and between the fibres forming the lattice are assumed negligible. Therefore, at the first order, the tension decomposition in Eq. (1) can be considered as a reasonable approximation of the membrane mechanics.

The relevance of this last assumption is detailed in the following.

5.2 Comparison between experimental data and theoretical predictions

Theoretical predictions derived from Eq. (1) were confronted to the experimental results presented in section 4, and displayed along in Figures 5 and 6. All these predictions were calculated by prescribing the experimental elongation paths $(\lambda_{xx}, \lambda_{yy})$ as input loadings, instead of using an additional assumption of simple tensile test conditions. Note that each plotted experimental elongation path was associated to a local value of the initial half-angle between fibres, $\widehat{\theta}_0$ (see bold values in Table 2). Therefore, in the following, the comparison between experiments and predictions is focused on the Cauchy tension - elongation curves $(\tau_{yy}, \lambda_{yy})$.

Over the whole database, whatever the considered triplet of parameters $(\ell_0, \theta_0, \beta_0)$, Figures 5 and 6 demonstrate that the micro-mechanical model predictions fairly well fit the experimental data (cases of discrepancies observed in the database are detailed thereafter). This result is obtained despite the model simplicity and bearing in mind that none arbitrary macroscale adjustment was made *a posteriori*, *i.e.* all the model data were acquired at the microscale (the REC scale). In particular, the model is able to predict very precisely the mechanical behaviour of composites $S_1(1, 45, 0)$, $S_3(1, 45, 0)$ and $S_6(1, 30, 0)$. The average relative difference between data

and predicted tensions is equal to 5.8%, 9.5% and 9.7% respectively. This mean difference remains below 20% for most other composites, with a value of 11.7% for $S_2(1, 45, 45)$, 16.4% for $S_4(3, 45, 45)$, 18.7% for $S_7(3, 60, 0)$ and 15.8% for $S_8(3, 30, 0)$. Note that this quantitative agreement between predictions and experimental data is similar when considering the second local value of $\widehat{\theta}_0$ in Table 2 (not shown). Such a good accordance validates model assumption (iii).

However, some discrepancies can be observed in two particular cases:

- the first case is pointed out in Figure 7(a). Two tests were performed on sample $S_1(1, 45, 0)$, with a good repeatability in both loading path and tension-elongation curves (see Figure 3(a)). The unique input difference between both test conditions relied on the applied experimental path and the associate value of initial angle $\widehat{\theta}_0$, as measured to determine the local kinematical field. The local angle $\widehat{\theta}_0$ was assessed to 47.5° in one case, and to 46.4° in the other (see Table 2). The impact of the weak difference between both $\widehat{\theta}_0$ -values on the macroscopic tension τ_{yy} is illustrated in Figure 7(a), along with the model predictions (solid lines). The load t^f predicted within REC fibre i ($i = 1$ and 2) in each case is plotted in Figure 7(b). The elongations λ^f predicted at the fibre scale are also plotted as function of the macroscopic loading λ_{yy} in Figure 7(c). Thus, it is shown that a very small change in the lattice fibrous architecture can induce very different tensile loadings at the fibre scale: in case $\widehat{\theta}_0 = 46.4^\circ$, the fibre elongation λ^f stayed close to 1 up to $\lambda_{yy} \in [1 : 1.05]$. This implies that the fibres in the REC were subjected to tension later than the ones of the sample with $\widehat{\theta}_0 = 47.5^\circ$. Such a delayed stretching altered the overall mechanical behaviour of the composite in a noticeable manner. From this point of view, the initial half-angle between fibres appears as a critical parameter in case of loading direction far from the fibre orientation.
- the second important discrepancy of the database is highlighted in Figure 7(d).

Two tests were performed on sample $S_5(1, 60, 0)$, with a good repeatability in both loading path (not shown) and tension-elongation curves. Again, the unique difference between both tests relied on the measurement of the local angle $\widehat{\theta}_0$, *i.e.* 61.9° in one case, and 62.7° in the other. The impact predicted on the macroscopic tension τ_{yy} , the fibre load t^f and the fibre elongation λ^f is displayed in Figures 7(d), 7(e) and 7(f) respectively. In that lattice configuration, both theoretical predictions are rather close to each other despite the $1\text{-}2^\circ$ of difference in $\widehat{\theta}_0$ assessment. The REC fibres were subjected to tension from the beginning of the test ($\lambda^f > 1$ once $\lambda_{yy} > 1$ in each case). Thus, for comparable macroscopic loadings (*e.g.* $\lambda_{yy} = 1.18$), the load achieved within the REC fibres is predicted four times greater than in previous lattice configuration ($\widehat{\theta}_0 \approx 45^\circ$), as shown when comparing Figures 7(e) and 7(b). However, the predictions largely deviate from the experimental tensions (mean relative difference being 86.7%). These discrepancies could be induced either by local stress heterogeneities generated within the composite, or by a misalignment of the sample's microstructure with respect to the load direction.

6 Conclusion

The process of artificial membranes able to mimic the mechanical and histological specificities of aortic tissue is needed to perform *in vitro* experiments as carried out in Deplano *et al.* (2014), using more relevant models of aortic wall. This will help improving the current knowledge on AAA evolution, and can also contribute to bioengineering challenges (*e.g.* building aligned and orderly fibrous structures with a controlled positioning, improving the deformability of future prostheses). In this global context, the present study proposed an original technical device to design architected soft-fibre reinforced membranes. With non-linear, anisotropic mechanical properties at the macroscale, and favoured fibrous orientations at the microscale, these membranes represent appropriate candidates for biomimetic goals.

The control of a macroscale mechanical anisotropy was allowed only by changing two microstructural parameters: the initial half-angle between the fibres θ_0 , and the fibre length ℓ_0 . The particular case of a single lattice of crossed, identical and straight monofilaments of $50 \mu\text{m}$ in diameter, embedded into a 1 mm-thick soft matrix in silicone was investigated in this work. Eight fibrous membranes of different microstructures were manufactured. The target values of (ℓ_0, θ_0) were achieved within a mean precision of $(0.07 \pm 0.05 \text{ mm}, 0.4 \pm 0.2^\circ)$ over the whole database. The fibrous membranes were then characterised using a tensile testing device, for two loading directions β_0 . Optical recordings of displacement fields were realised at both macro-and microscale. These experiments allowed the validation of the assumptions of the micro-mechanical model presented in a previous work (Bailly *et al.*, 2012), and based on the homogenisation of periodic discrete structures. This study demonstrates that the model is able to describe the experimental database in a good quantitative agreement, whatever the triplet of parameters $(\ell_0, \theta_0, \beta_0)$.

The present experimental work must be completed by new biaxial tensile loadings, in order to assess the micro-mechanical model for loadings closer to the ones that arterial tissues are subjected to. Further tests will be coupled with 3D DIC measurements, in order to capture possible out-of-plane displacement fields of the deformed membranes. Finally, manufacturing of wavy substructures in the fibrous lattice would allow the biomimetism of the membranes, as expected in the target solution identified in Bailly *et al.* (2012).

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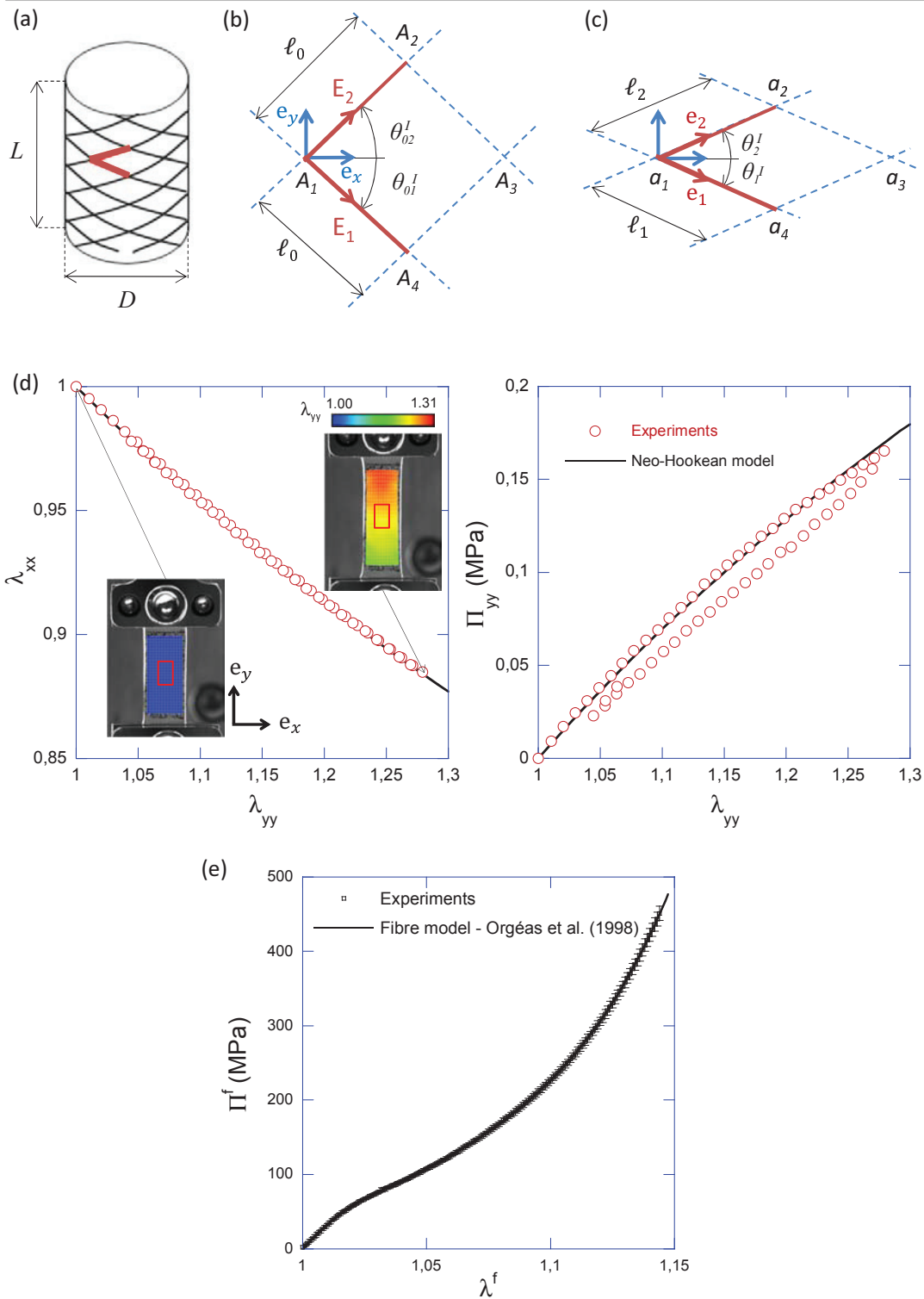


Figure 1. **(a)** Rhombic lattice and corresponding periodic Representative Elementary Cell of the *one-layer fibrous structure*: **(b)** initial undeformed configuration C_0 , **(c)** actual configuration C . **(d)** Mechanical behaviour of the silicone matrix: typical elongation path $(\lambda_{xx}, \lambda_{yy})$ and corresponding nominal stress Π_{yy} versus elongation λ_{yy} . The continuous line gives the prediction of the neo-Hookean model with $c^m = 0.24$ MPa. **(e)** Mechanical behaviour of the fibre: mean nominal stress Π^f versus fibre elongation λ^f (average value over 16 samples). Error bars represent the data standard deviation.

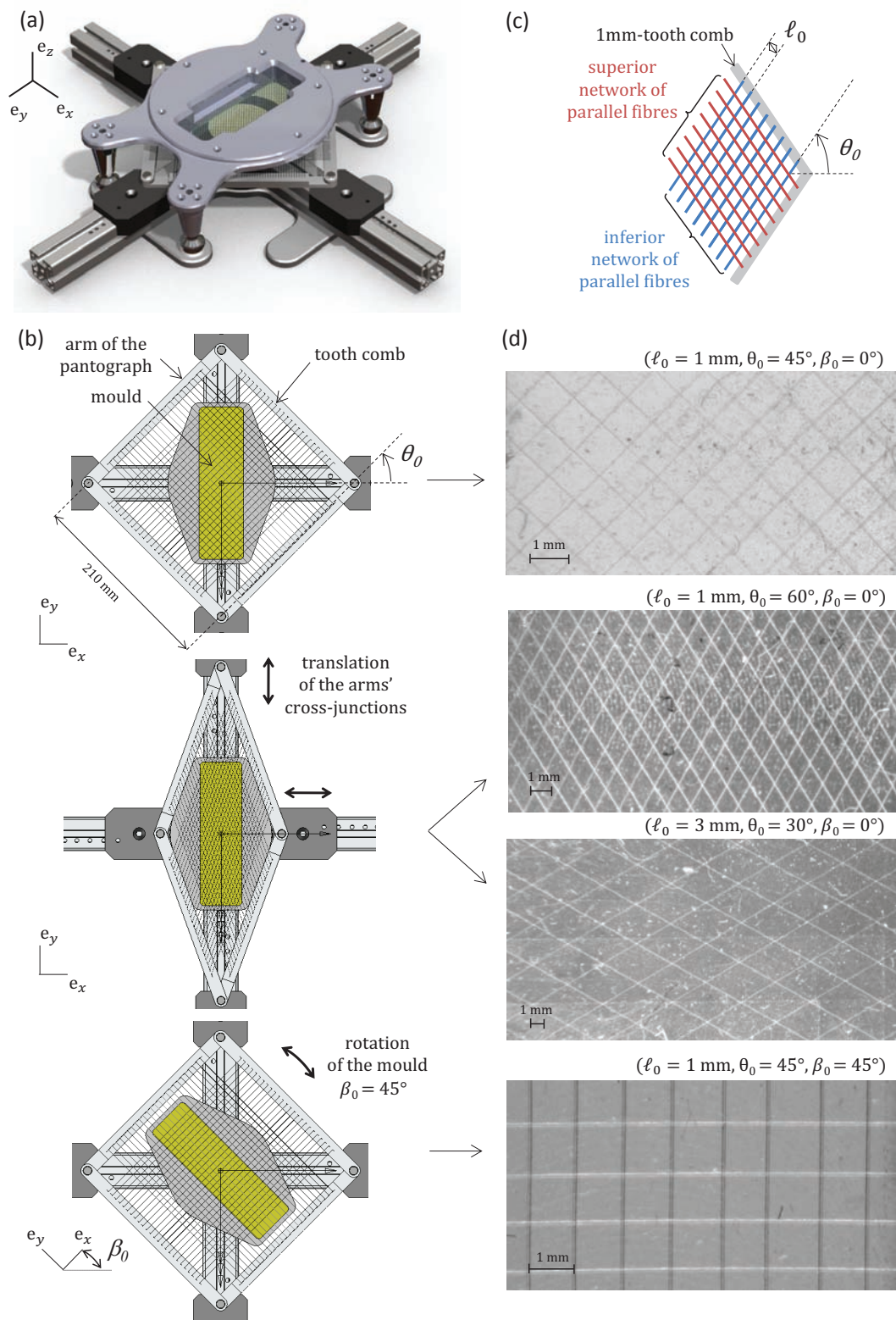


Figure 2. (a) 3D-scheme of the device designed to manufacture the composite membranes. (b) 2D-view of the device's motion possibilities. The yellow part represents the lower part of the mould dedicated to the membrane casting (superior part of the device being not shown here). (c) Schematic zoom of a 2D-view of the pantographic structure for the lattice elaboration. (d) Pictures of some resulting composite membranes. Composites with perpendicular fibres could be created by two different manufacturing processes (see first and last rows).

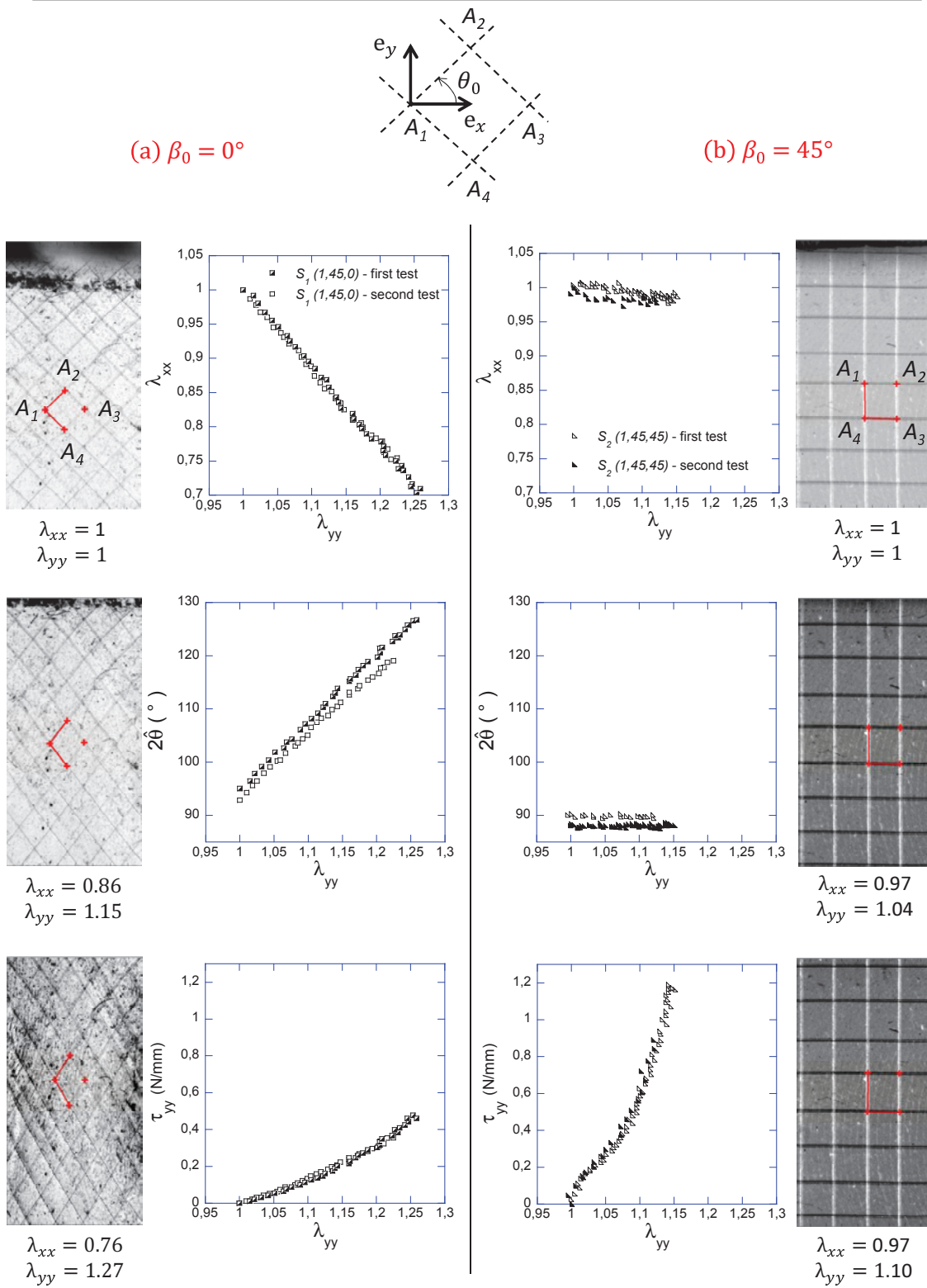


Figure 3. Typical post-processing of the measured data, using the tracking of the experimental REC nodes A_i . Illustration for two distinct loading directions β_0 , with $S_1(1, 45, 0)$ (a) and $S_2(1, 45, 45)$ (b).

Table 1

Dimensions and microstructural parameters of composite samples $S_i(\ell_0, \theta_0, \beta_0)$. †Values measured over all RECs of the strip's surface: mean value (minimum - maximum).

Samples	$\bar{\ell}_0(\text{mm})^\dagger$ ± 0.05	$\bar{\theta}_0(^\circ)^\dagger$ ± 0.2	$L_0(\text{mm})$ ± 0.02	$W_0(\text{mm})$ ± 0.02	$T_0(\text{mm})$ ± 0.02
$S_1(1, 45, 0)$	0.96 (0.86 - 1.08)	45.3 (43.0 - 47.6)	11.22	7.75	1.06
$S_2(1, 45, 45)$	0.96 (0.87 - 1.10)	45.2 (43.8 - 46.4)	10.29	5.99	0.98
$S_3(3, 45, 0)$	2.93 (2.60 - 3.08)	45.8 (43.8 - 48.1)	28.31	21.12	0.85
$S_4(3, 45, 45)$	2.85 (2.76 - 2.96)	45.1 (44.2 - 46.5)	22.28	15.34	0.81
$S_5(1, 60, 0)$	0.95 (0.79 - 1.10)	60.4 (57.6 - 63.5)	18.04	6.39	0.93
$S_6(1, 30, 0)$	0.97 (0.86 - 1.07)	30.2 (28.6 - 31.8)	20.82	10.31	1.11
$S_7(3, 60, 0)$	2.90 (2.68 - 3.07)	60.2 (57.8 - 62.6)	47.96	17.54	1.05
$S_8(3, 30, 0)$	2.92 (2.61 - 3.16)	31.1 (29.1 - 33.3)	37.91	25.29	0.77

Table 2

Differences between the target value of the initial half-angle between fibres θ_0 , the mean value measured on the whole sample surface $\bar{\theta}_0$, and the value measured over 4 RECs to determine the local kinematical field $\hat{\theta}_0$ ([†]bold values are the ones used for theoretical predictions illustrated in Figures 5 and 6).

Samples $S_i(\ell_0, \theta_0, \beta_0)$	$\theta_0(^{\circ})$	$\bar{\theta}_0(^{\circ})$ <i>mean (min, max)</i> ± 0.2	$\hat{\theta}_0(^{\circ})$ <i>1st test, 2nd test[†]</i> ± 0.2
$S_1(1, 45, 0)$	45	45.3 (43, 47.6)	47.5 , 46.4
$S_2(1, 45, 45)$	45	45.2 (43.8, 46.4)	44.2 , 45.1
$S_3(3, 45, 0)$	45	45.8 (43.8 - 48.1)	45.6 , 45.6
$S_4(3, 45, 45)$	45	45.1 (44.2 - 46.5)	44.2 , 43.7
$S_5(1, 60, 0)$	60	60.4 (57.6, 63.5)	61.9 , 62.7
$S_6(1, 30, 0)$	30	30.2 (28.6, 31.8)	31.3 , 32.2
$S_7(3, 60, 0)$	60	60.2 (57.8, 62.6)	60.5 , 60.3
$S_8(3, 30, 0)$	30	31.1 (29.1, 33.3)	31.3 , 30.9

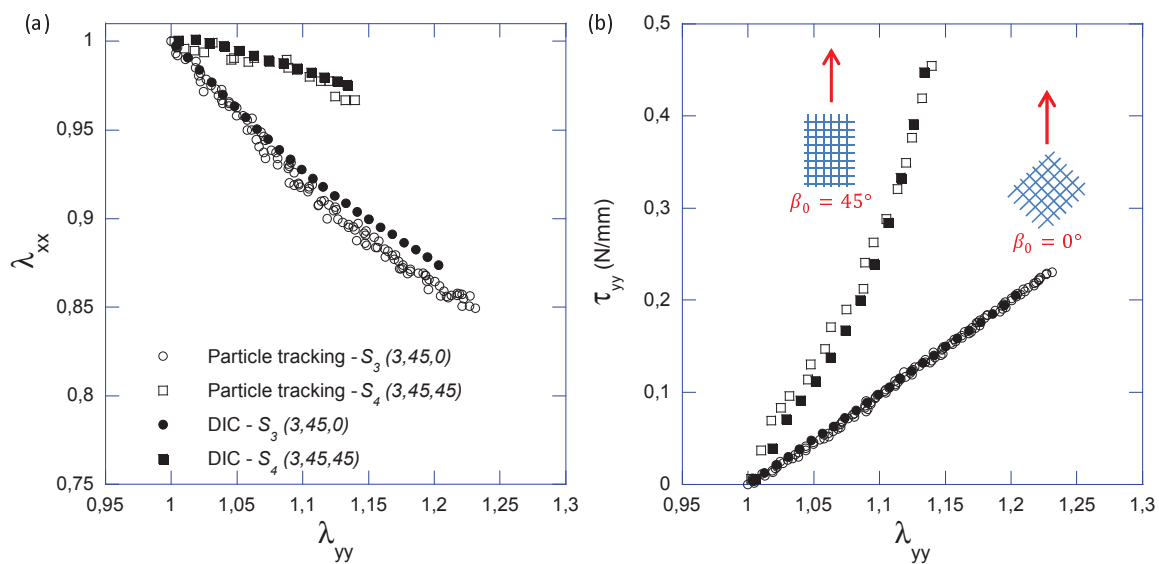


Figure 4. Comparison between global kinematics at the macroscale determined by 2D DIC and local kinematics at the microscale determined by tracking of experimental REC extremities. Illustration for two distinct loading directions, with samples $S_3(3, 45, 0)$ and $S_4(3, 45, 45)$. Same legend for (a) and (b).

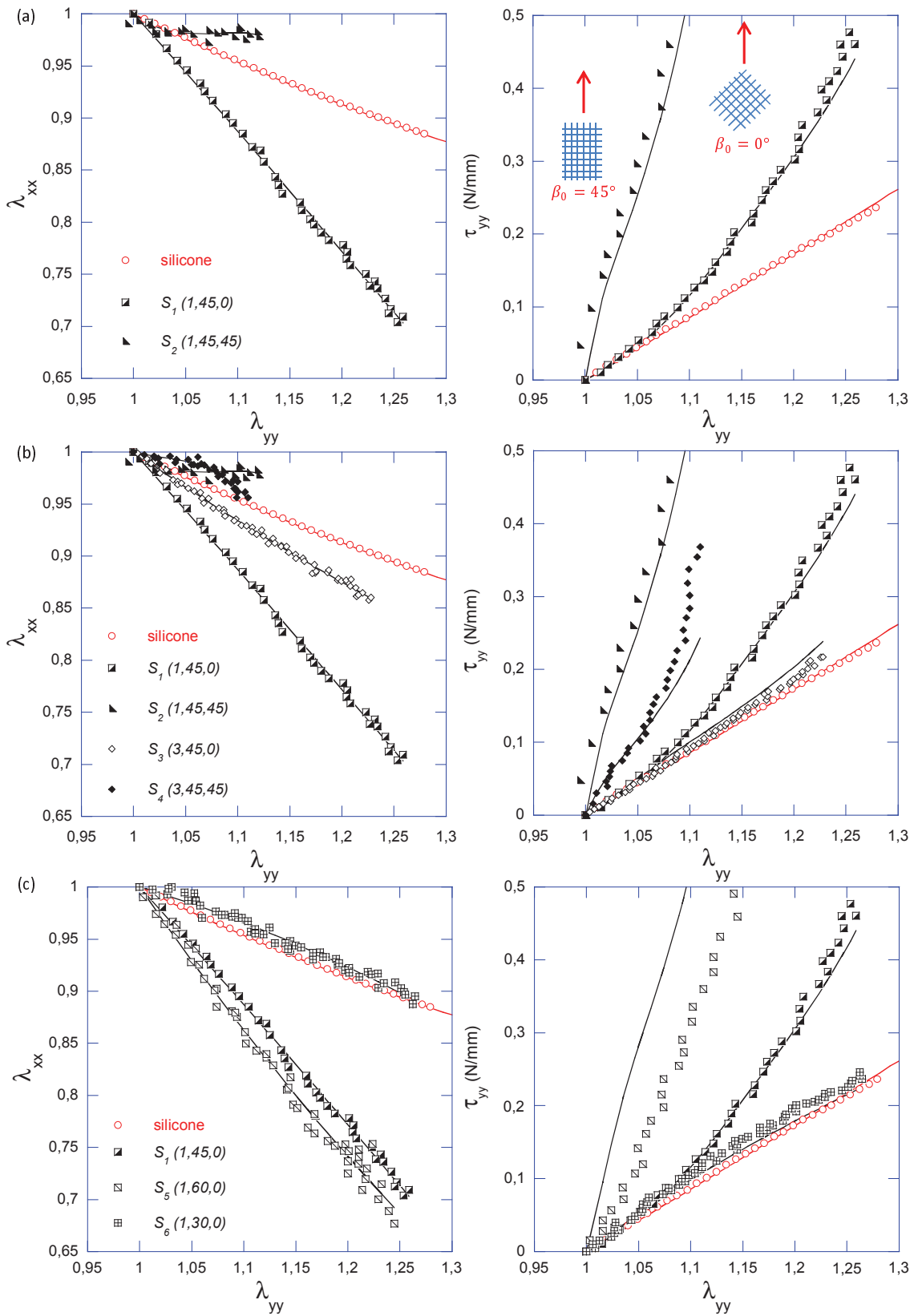


Figure 5. Influence of (a) the loading direction β_0 , (b) the initial fibre length ℓ_0 and (c) the initial half-angle between fibres θ_0 on the mechanics of the composite membranes. Comparison between the predictions of the micro-mechanical model (solid lines) and experimental data (symbols).

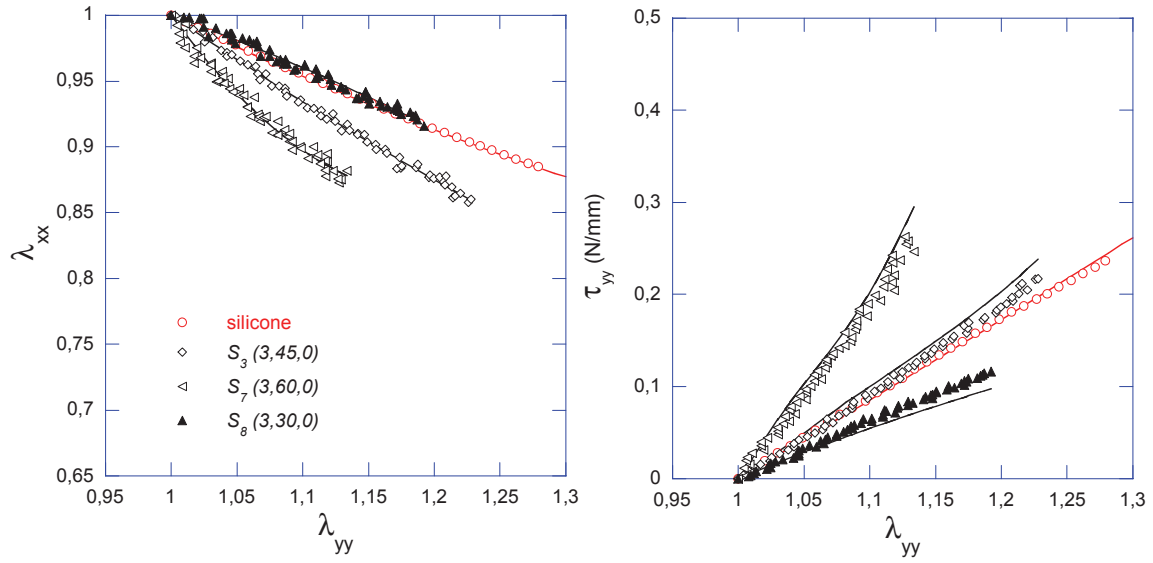


Figure 6. Same as in Figure 5 for composites with $\ell_0 = 3$ mm and $\beta_0 = 0^\circ$. Influence of the initial half-angle between fibres, θ_0 .

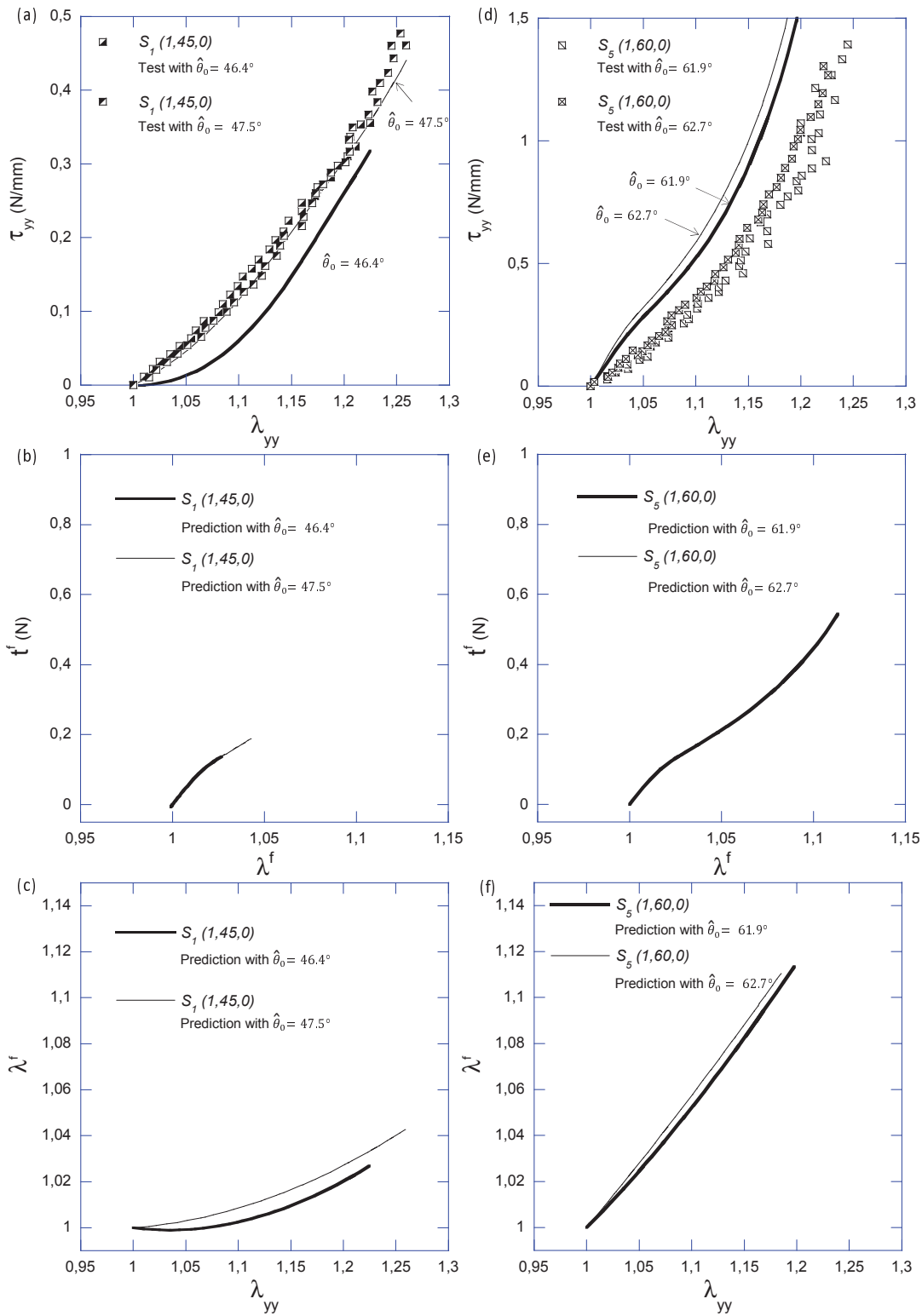


Figure 7. **(a, d)** Critical impact of the angle $\hat{\theta}_0$ on the theoretical predictions of the composite's Cauchy tension-elongation response (solid lines). Illustration on two samples, $S_1(1, 45, 0)$ **(a)**, and $S_5(1, 60, 0)$ **(d)**. **(b, e)** Corresponding predictions of the load within REC fibre i ($i=1,2$), as function of fibre elongation. **(c, f)** Predictions of the fibre elongation, as function of the macroscopic loading in the loading direction.