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NUMERICAL MODELING OF A 3 POINT BENDING TEST OF A REINFORCED CONCRETE BEAM USING A SECOND GRADIENT THEORY

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Abstract. *Being a quasi-brittle material, concrete exhibits a strain softening behavior that cannot be reproduced with classical continuum mechanics models. To regularize the problem, an internal length should be introduced. Several ways to do so have been proposed in the literature. One way is the so called local second gradient model. It is a local theory as it introduces the internal length by enriching the kinematical description of the continuum adding higher order gradients of the displacement according to the work of Cosserat [8], Toupin [18], Mindlin [19] and Germain [9,10]. The model has been developed by Chambon et al. [11,12] and has this far been used mainly to reproduce the behavior of soils. It is here applied for the first time to a reinforced concrete beam subjected to a 3 point bending test.*

1 INTRODUCTION

Strain localization in quasi-brittle materials, or more generally in materials exhibiting strain softening is a well-known problem [1] [2] [3]. It is a phenomenon that can be clearly observed in experimental tests, and yet cannot be modeled with classical continuum mechanics models. Analytically, the solution is a crack with zero energy dissipation [4]; numerically, it leads to a pathological mesh dependency. These shortcomings are due to the lack of an internal length in the continuum model.

Several ways of introducing an internal length have been proposed. Non local integral models were first formulated by Pijaudier-Cabot et al. [4] and applied to damage theory. The gradient models of Aifantis [6] enrich the conventional plasticity and damage theories with gradients of the internal variables. This type of model can be shown to be equivalent to the integral type models. More recently, strain localization due to damage has been treated using the thick level set approach [7].

A rather natural way of introducing (indirectly) a length parameter in a continuum model is to account for the microstructure of the material. The general class of so called microstructured models or higher order continuum models allows for the description of the kinematics of the microstructure by using an additional tensor in the displacement field. Higher order continuum theories can be traced back to the works of the Cosserat brothers [8] and have been generalized and properly formulated by Germain [9] [10] using the method of virtual power.

The local second gradient model developed by Chambon et al. can be seen as a particular case of a higher order continua and has been used to regularize problems involving localization in geomaterials [11,12]. It is hereafter used to model a three point bending test of a reinforced concrete beam.

2 THE SECOND GRADIENT MODEL

2.1 Theoretical framework

As detailed in the seminal work of Germain, using the virtual power method one can choose a field of virtual displacements to describe the proper kinematics of the continuum (including its microstructure). The internal stresses, limit conditions and equilibrium equations appear naturally as long as the linear form representing the virtual power is correctly defined and that it respects the principle of material independence.

Following this, if we choose the virtual displacement field as the “field of continuous and continuously differentiable velocities”, then the principle of virtual work yields:

$$\int_D \sigma_{ij} D_{ij}^* + \Sigma_{ijk} u_{i,jk}^* d\Omega - \int_D G_i u_i^* d\Omega - \int_{\partial D} p_i u_i^* + P_i Du_i^* ds = 0 \quad (1)$$

Where u_i is the displacement field, D_{ij} is the symmetric part of its gradient, σ_{ij} is the macro stress (conjugate of the first gradient of the velocity field) and Σ_{ijk} is a double stress (conjugate of the second gradient of the velocity field). G_i is the classical body force, p_i is a contact force and P_i a double contact force. Du_i refers to the normal derivative: $Du_i = n_k \partial u_i / \partial x_k$. Here and henceforth * denotes a virtual quantity.

The equilibrium equations and boundary conditions are given by:

$$\frac{\partial \sigma_{ij}}{\partial x_j} - \frac{\partial^2 \Sigma_{ijk}}{\partial x_j \partial x_k} + G_i = 0 \quad (2)$$

$$\sigma_{ij}n_j - n_k n_j D\Sigma_{ijk} - \frac{D\Sigma_{ijk}}{Dx_k} n_j - \frac{D\Sigma_{ijk}}{Dx_j} n_k + \frac{Dn_l}{Dx_l} \Sigma_{ijk} n_j n_k - \frac{Dn_j}{Dx_k} \frac{Dn_l}{Dx_l} \Sigma_{ijk} n_j = p_i \quad (3)$$

$$\Sigma_{ijk} n_j n_k = P_i \quad (4)$$

Where Dq/Dx_j is the tangential derivative.

Here we suppose that the model uses two constitutive equations, one linking the macro stress to the first gradient of the displacements and the other linking the double stress to the second gradient of the displacements. The two equations are also supposed to be decoupled. The first gradient law can be any classical constitutive law (involving or not damage, plasticity etc...). The constitutive laws can be written in rate form as:

$$\dot{\sigma}_{ij} = C_{ijkl} \dot{D}_{kl}^e \quad (5.1)$$

$$\dot{\Sigma}_{ijk} = K_{ijklmn} \dot{u}_{l,mn} \quad (5.2)$$

If the second gradient law is an isotropic elastic one, the general form of the constitutive law involves five material parameters.

$$\begin{bmatrix} \dot{\Sigma}_{111} \\ \dot{\Sigma}_{112} \\ \dot{\Sigma}_{121} \\ \dot{\Sigma}_{122} \\ \dot{\Sigma}_{211} \\ \dot{\Sigma}_{212} \\ \dot{\Sigma}_{221} \\ \dot{\Sigma}_{222} \end{bmatrix} = \begin{bmatrix} a^{12345} & 0 & 0 & a^{23} & 0 & a^{12} & a^{12} & 0 \\ 0 & a^{145} & a^{145} & 0 & a^{25} & 0 & 0 & a^{12} \\ 0 & a^{145} & a^{145} & 0 & a^{25} & 0 & 0 & a^{12} \\ a^{23} & 0 & 0 & a^{34} & 0 & a^{25} & a^{25} & 0 \\ 0 & a^{25} & a^{25} & 0 & a^{34} & 0 & 0 & a^{23} \\ a^{12} & 0 & 0 & a^{25} & 0 & a^{145} & a^{145} & 0 \\ a^{12} & 0 & 0 & a^{25} & 0 & a^{145} & a^{145} & 0 \\ 0 & a^{12} & a^{12} & 0 & a^{23} & 0 & 0 & a^{12345} \end{bmatrix} \begin{bmatrix} \dot{u}_{1,11} \\ \dot{u}_{1,12} \\ \dot{u}_{1,21} \\ \dot{u}_{1,22} \\ \dot{u}_{2,11} \\ \dot{u}_{2,12} \\ \dot{u}_{2,21} \\ \dot{u}_{2,22} \end{bmatrix} \quad (6)$$

Where $a^{ijk\dots}$ are functions of five material parameters $a_1, a_2, a_3, a_4,$ and a_5 [19].

A particular form of this latter constitutive equation has been proposed by Chambon et al.[19] with a single modulus B (corresponding to a specific choice of the a_i parameters).

The internal length does not appear explicitly in these equations. However, the analytical solution in a one-dimensional localization problem with a bilinear first gradient law of moduli A_1 and A_2 , exhibits a localized strain band with an internal length l_s given by [13]:

$$\tanh\left(\sqrt{\frac{-A_1}{B}}(L - l_s)\right) = -\sqrt{\frac{-A_1}{A_2}} \tan\left(\sqrt{\frac{-A_2}{B}} l_s\right) \quad (7)$$

It is shown that the internal length l_s is a function of the ratio B over A_2 .

The use of the second gradient model allows for the proper representation of strain localization but it does not restore the uniqueness of the solution. For a one-dimensional tension test or a biaxial compression test, the number of bands and their positions can vary [14]. This is in accordance with experimental results which tend to be poorly reproducible when involving strain localization.

2.2 Numerical implementation

The second gradient of the displacements in the weak formulation of the problem necessitates the use of C^1 elements in a finite element code. This can be avoided by introducing a new field v_{ij} imposed equal to the gradient of u_i through the use of Lagrange multipliers [15]. The new weak formulation of the problem becomes:

$$\int_D \sigma_{ij} D_{ij}^* + \Sigma_{ijk} v_{ij,k}^* d\Omega - \int_D \lambda_{ij} \left(\frac{\partial u_i^*}{\partial x_j} - v_{ij}^* \right) d\Omega - P_e = 0 \quad (8.1)$$

$$\int_D \lambda_{ij}^* \left(\frac{\partial u_i}{\partial x_j} - v_{ij} \right) d\Omega = 0 \quad (8.2)$$

The problem is then discretized using a 9th node second gradient finite element: 8 of the nodes are used for u_i , 4 for v_{ij} and the central node for λ_{ij} .

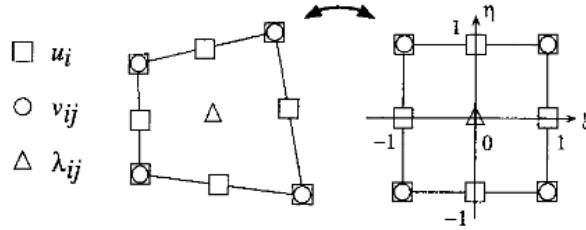


Figure 1: Second gradient finite element [15]

This element has been implemented in the finite element code Lagamine and the problem is solved using the classical Newton-Raphson method.

3 THE THREE POINT BENDING TEST

The experimental test was conducted on a reinforced concrete beam according to the specifications of the CEOS.FR benchmark. The beam is of thickness $b=200\text{mm}$, height $h=500\text{mm}$ and 5000mm span (see: figure 2). It was subjected to a cyclic controlled load.

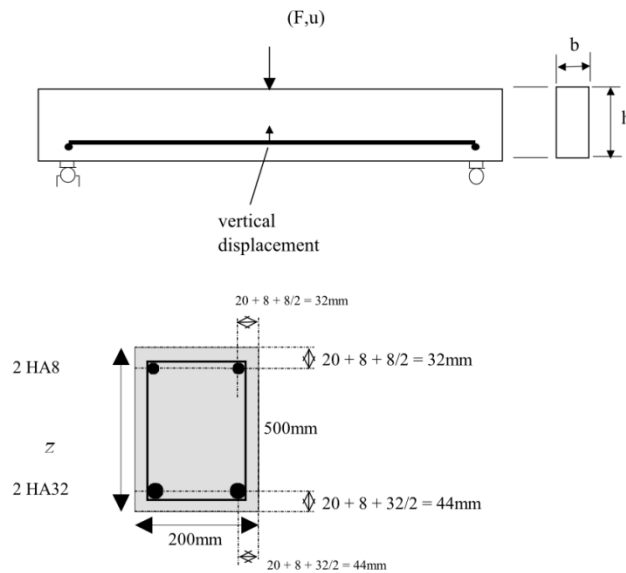


Figure 2: Reinforced concrete beam dimensions [17]

Strains were measured with gauges placed as indicated on figure 3.

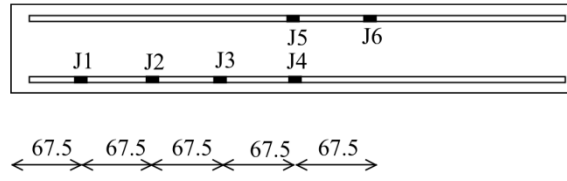


Figure 3: Location of strain gauges [17]

4 MODELLISATION OF THE BENDING TEST

The bending test is modeled as a two dimensional problem using the 9 node finite element described above. Two meshes have been used for the simulations. The first mesh consists of 5180 elements, 4148 of which are second gradient elements representing concrete and 1032 truss elements representing the horizontal reinforcement. The average size of the concrete elements for this mesh is of 0.02mx0.035m. The second mesh consists of 13494 elements with an average size of 0.01mx0.017m for the concrete elements. Concrete and steel elements are supposed to be perfectly bonded. The end nodes at the bottom of the beam are blocked vertically; the right node at the bottom is also blocked horizontally.

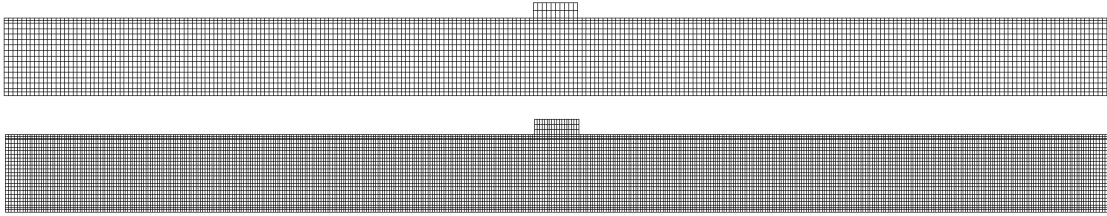


Figure 4: Finite element meshes [15]

The concrete behavior is reproduced using the classical Mazars damage law [16]. The equivalent strain is defined as:

$$\varepsilon_{eq} = \sqrt{\langle \varepsilon \rangle_+ : \langle \varepsilon \rangle_+} \quad (9)$$

Where $\langle \cdot \rangle_+$ denote the positive part and $:$ is the double dot product. The damage is separated in two parts D_t and D_c , one due to the tension, and one due to compression.

$$D_c = 1 - \frac{\varepsilon_{d0}(1 - A_c)}{\varepsilon_{eq}} - A_c e^{-B_c(\varepsilon_{eq} - \varepsilon_{d0})} \quad (10.1)$$

$$D_t = 1 - \frac{\varepsilon_{d0}(1 - A_t)}{\varepsilon_{eq}} - A_t e^{-B_t(\varepsilon_{eq} - \varepsilon_{d0})} \quad (10.2)$$

Where ε_{d0} , A_c , A_t , B_c and A_t are material parameters provided in table 1:

E (GPa)	ε_{d0}	A_t	B_t	A_c	B_c	β
37.2	9.10^{-5}	0.7	6800	0.42	780	1.1

Table 1: Concrete material parameters

On the upper part where the loading is applied and at both supports at the bottom an elastic linear law is introduced to prevent from unwanted damage. An elasto-plastic law with isotropic hardening is used for steel.

Upper trusses		
E (GPa)	σ (MPa)	Area (cm ²)
195	466	16.085 (2HA32)
Lower trusses		
E (GPa)	σ (MPa)	Area (cm ²)
195	466	1.0053(2HA8)

Table 2: Steel material parameters

The second gradient constitutive law for concrete depends on the elastic modulus B taken equal to 1.5.MN.

5 RESULTS

Overall, the numerical results are in good accordance with the experimental data. Figure 5 shows the numerical global force versus displacement at the center of the beam compared to the experimental results (note: for the experimental test the beam was loaded and unloaded cyclically whereas in the simulation the beam was loaded with a monotonic increasing displacement).

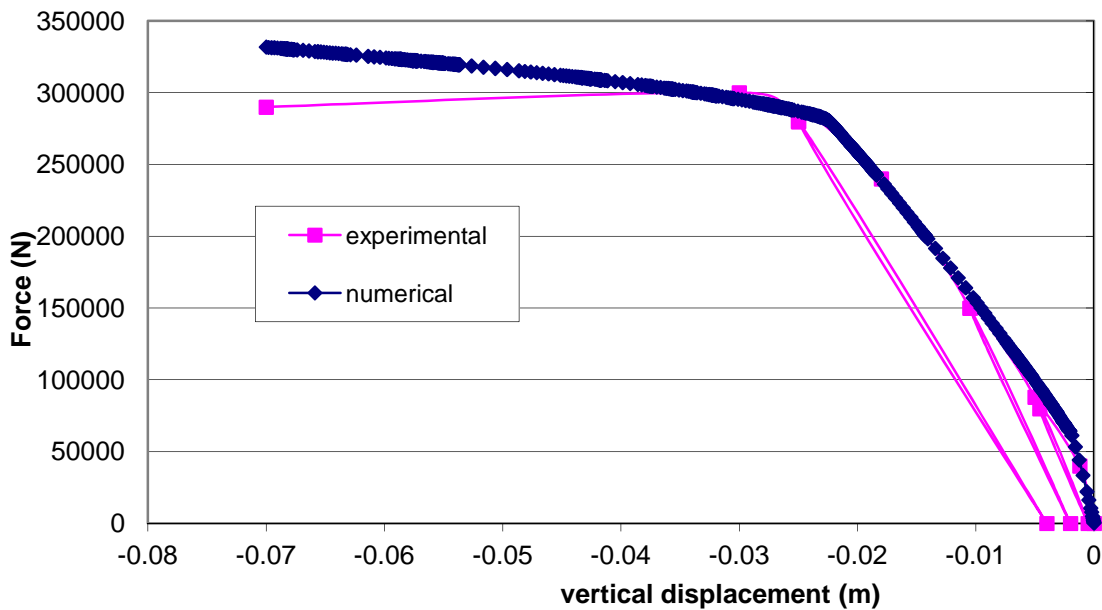


Figure 5: Numerical results Vs. Experimental data: Force-Displacement at the center of the beam

The force displacement graph exhibits the classical reinforced concrete behavior in three stages: In the first stage, concrete and steel stay both in the elastic regime; then concrete starts to damage and the slope on the force displacement curve slightly changes. Finally, steel enters in the plastic phase and the second change in the slope appears. The final slope of the numerical simulation doesn't match the experimental one, but results can be improved by doing a parametrical study on the influence of the elastic modulus B .

Figure 6 shows the pattern of the damage variable in concrete for different stages of loading and for the two meshes.

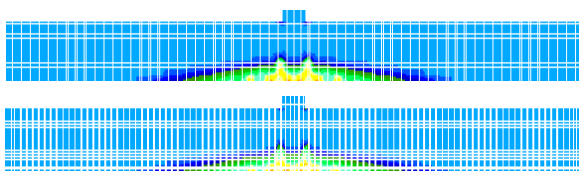


Figure 6.1. Damage distribution (2mm displacement)

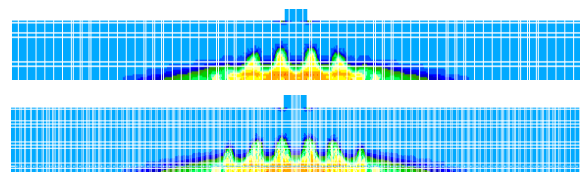


Figure 6.2. Damage distribution (3mm displacement)

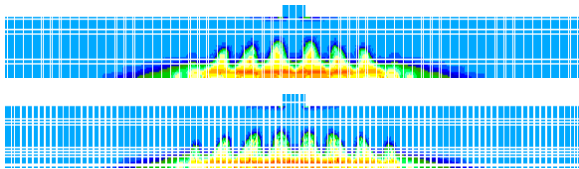


Figure 6.3. Damage distribution (4mm displacement)

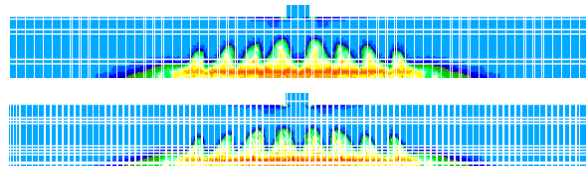


Figure 6.4. Damage distribution (5mm displacement)

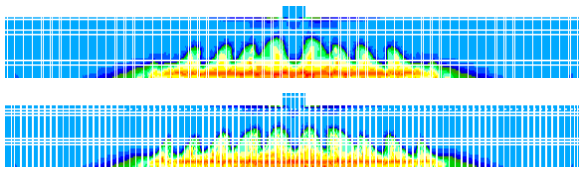


Figure 6.5. Damage distribution (6mm displacement)

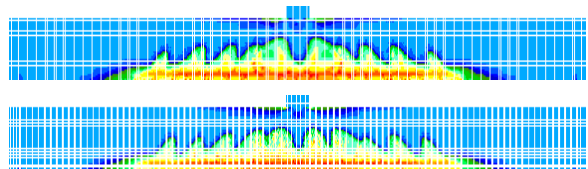


Figure 6.6. Damage distribution (7mm displacement)

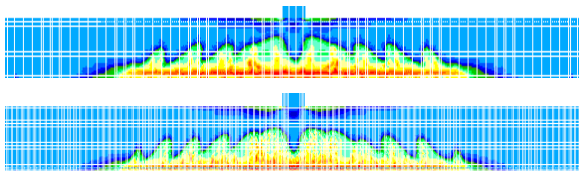


Figure 6.7. Damage distribution (8mm displacement)

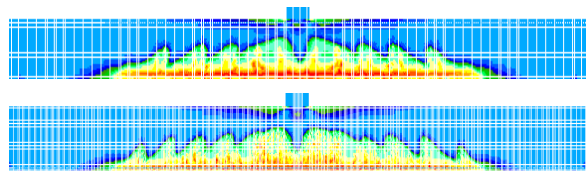


Figure 6.8. Damage distribution (9mm displacement)

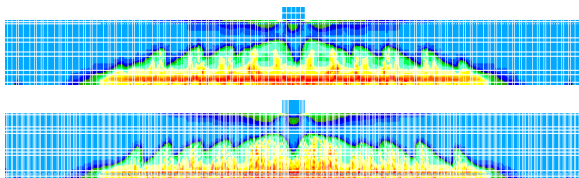


Figure 6.9. Damage distribution (10mm displacement)

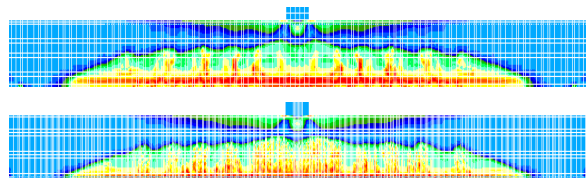


Figure 6.10. Damage distribution (15mm displacement)

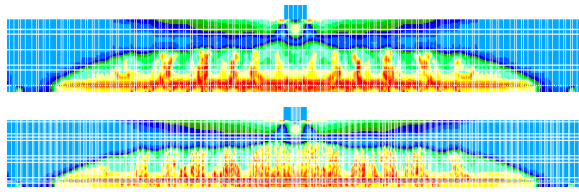


Figure 6.11. Damage distribution (18mm displacement)

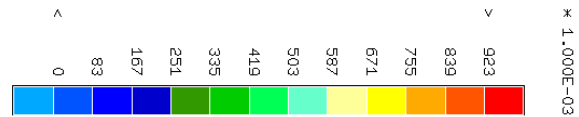


Figure 6.12. Damage scale

The damage pattern develops with sudden “peaks” which experimentally correspond to developing cracks. The crack opening is not modeled directly in this simulation as the displacement field remains continuous, but it can be calculated from the damage model either through energetic equivalence or by simply measuring the jump in displacements between two points located on the opposite sides of the damaged zone. This obviously works only when the damaged bands are clearly separated. The width and separation of the damage bands can be controlled by changing the internal length, which in our case would mean changing the slope of either or both the first gradient and second gradient constitutive laws.

For concrete bending tests, non-local damage models which define an equivalent strain by averaging over a certain distance l_c may develop damage on the upper compressed part of the beam even when the local strain is not high enough to trigger damage. This is due to the averaging over an area. There is no such problem with this model as all the variables are local, including the equivalent strain.

6 CONCLUSIONS

A second gradient model has been used to model the three point bending test of a reinforced concrete beam. The results show that the model is able to reproduce the force-displacement curve obtained experimentally. The damage localizes into bands and their width is controlled by the model parameters. The uniqueness of the solution is however not restored as shown by the results obtained with two different meshes. In these two cases the bands are of the same size but their numbers can vary (figure 6). Being a local theory, the second gradient method avoids the limitations caused by the use of a non-local definition of the equivalent strain used in the non-local damage theory.

These results are encouraging and represent the first steps toward a wider use of the local second gradient method for concrete structures.

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