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Laura Balea, Gilles Dusserre, Gérard Bernhart. Mechanical behaviour of plain-knit reinforced injected composites: effect of inlay yarns and fibre type. Composites Part B: Engineering, 2014, 56, p.20-29. 10.1016/j.compositesb.2013.07.028. hal-00946564

HAL Id: hal-00946564

https://hal.science/hal-00946564

Submitted on 6 Nov 2018

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# Mechanical behaviour of plain-knit reinforced injected composites: Effect of inlay yarns and fibre type

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#### ABSTRACT

This paper investigates the effect of inlay yarns and fibre type (E-glass, basalt, carbon) to improve the mechanical behaviour of plain knit reinforced composites (epoxy matrix). The tensile behaviour of the dry reinforcement was investigated in the wale and course-directions and has shown that the course-wise deformation is drastically reduced whereas the strength is strongly increased when inlay yarns are included. Fibre type has a weak effect on deformation and strength. For the composite material, processed by Liquid Composite Moulding, the carbon fibre reinforcement gives always the best results whatever the testing direction. Inlay yarns decrease the 0° mechanical properties, whereas 45° and 90° are increased. Moreover a quasi-isotropic behaviour is obtained when using two inlay yarns, irrespective of the fibre types.

Keywords:

- A. Fabrics/textile
- B. Mechanical properties
- E. Knitting
- E. Resin transfer moulding (RTM)

#### 1. Introduction

Fibre reinforced composite materials are increasingly used in industrial applications requiring high specific mechanical properties. They cover a large range of mechanical and physical properties due to the numerous parameters involved in their manufacturing. Amongst them, the fibre arrangement in the matrix has an important effect on the mechanical properties of the composite. Up to now, woven and unidirectional reinforcements are mainly used in industrial applications. The low curvature of the fibres confers high performance to the composite. However, issues related to fibrous architecture remain, especially for composite forming. Complex shape parts manufacturing requires cutting and assembly operations, resulting in material wastage and high production costs. This leads also to a local weakening in assembly areas, mainly related to the reinforcement discontinuity. Therefore new reinforcement architectures were designed, and, amongst them, knitted fabrics meet the technical requirements for complex shapes forming. This technology enables the development of netshape fabrics, including holed preforms. Knits are obtained by intermeshing loops of fibre yarns resulting in a high deformation ability compared with woven fabrics [1,2] thanks to the possibility to change the shape of the loops [3,4]. This deformation ability is especially suitable for stretch forming and deep drawing, since it allows to distribute homogeneously the textile deformation, avoiding local material heterogeneities in a single part [5,6]. A low strain energy is required allowing wrinkles to appear lately in comparison to woven fabrics [7,8]. Excellent properties for composite moulding are obtained, especially for stacking.

Knit reinforced composites exhibit also interesting impact behaviour properties. Stitches of adjacent layers overlap each other, reducing inter-laminar resin-rich areas and crack propagation. The inter-laminar fracture toughness is then increased comparing to woven and UD composites [9–12]. Indeed the delamination of knit composites involves both mode I and mode III crack propagation [12]. Compression-After-Impact tests highlight the high damage tolerance of knitted composites: they exhibit a relative compressive strength retention higher than woven composites [9].

However, knitted composites exhibit low in-plane mechanical properties, close to short fibres composites, because of the curvature of the fibres [13,14]. The compressive properties are isotropic and depend strongly on the matrix properties [15]. On the contrary, the tensile properties are anisotropic and related to the reinforcement. Up to now, most studies were focused on basic knitted structures such as plain,  $1 \times 1$  rib and Milano made of glass fibres. These investigations highlighted that the composite properties depend on knit architecture [16,17] and on knit geometrical parameters such as stitch size, stitch density and yarn length [16,18]. An initial fabric deformation before composite processing leads to an improvement of the properties in the stretched direction and to a loss of properties in the transverse direction [19]. However biaxial stretching before composite forming has a negligible effect [20]. It is then demonstrated that an improvement of the tensile properties would require a modification of the fibre distribution in the

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material. No studies on the effect of the fibre type on the knitted composite behaviour are known.

In this paper, the effects of knit construction and fibre type on the composite tensile behaviour are investigated. Previous works [21,22] were focused on the two basic knitted reinforcements, plain and  $1 \times 1$  rib. This latter is made up of front and back stitches, leading to a three-dimensional structure and a higher deformability is then achieved than with plain-knit, only made up of front stitches. Plain structure confers the best tensile properties to the composite, especially in the wale-direction, due to a less isotropic fibre orientation distribution, and has been chosen as the starting point of the optimisation aiming to improve the composite mechanical properties [23–25]. Plain-knit composites are anisotropic with higher properties in the wale-direction. These reinforcements have higher deformation ability in the coursedirection. A correlation between fabric deformation ability and composite properties was carried out. The optimisation of the plain-knit architecture requires to reinforce the course-direction and to find a compromise with the deformation ability (reinforcing means adding straight yarns, what obviously decreases the extensibility). The knitting process allows to add non-knitted yarns in the course-direction into the fabric, thanks to inlay yarns obtained by two techniques: float loops [26] or tuck loops [26,27]. In this paper, the cross effects of inlay yarns, their number, and the fibre type on knitted fabric deformability and composite mechanical behaviour are investigated.

#### 2. Materials and process

#### 2.1. Fibres and resin

E-glass, basalt and carbon fibres are studied to highlight the effect of the fibre type on the behaviour of the fabric and the composite. Glass and carbon fibres are the most widely used in the composite manufacturing industry. The former are mostly dedicated to mass production materials due to its excellent price-quality ratio and the latter provide high cost structural materials, with excellent mechanical properties. Basalt fibres, considered as emergent, have interesting physicochemical properties and slightly better mechanical properties than glass fibres.

The yarn diameter is a key parameter of a knit, it controls the jamming of the loops [21]. In order to produce equivalent knitted fabrics, roving bobbins with a similar yarn diameter were chosen. This diameter is calculated from diameter and number of filaments by considering a hexagonal fibre packing. Fibres and roving properties are given in Table 1. Glass and basalt fibres have similar properties and both rovings have the same linear density. However a smaller filament diameter confers to basalt yarns a lower bending stiffness. Carbon fibres have thinner filaments and exhibit a

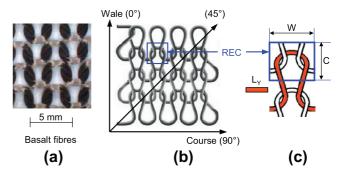
higher Young's modulus and a lower density, that confer to the yarns lower bending stiffness and linear density.

The resin used is a two components epoxy system (SICOMIN SR 1710 resin with SD 8822 hardener) designed for resin transfer moulding processes. This system is characterized by a very low viscosity and high mechanical properties, conferring a high inter-laminar shear strength to the composite (Table 1).

#### 2.2. Knitted fabrics

Knitting consists in intermeshing loops of yarns to produce fabrics from a single or from several yarns. Weft-knitting refers to the knitting direction: the stitches are knitted successively in the weft-direction, also called course-direction. The loops are interconnected in the wale direction called also columns. Plain-knit, only made of front stitches, is the basic architecture (Fig. 1) in which inlay yarns were added to reinforce the course-direction. Every four plain stitches, inlay yarns are linked to the knit by tuck stitches (Fig. 2). Between two consecutive tuck stitches, the inlay yarn forms a float loop. Basic plain-knit (*P*0) and plain-knits with one (*P*1), two (*P*2) and three (*P*3) inlay yarns per course were manufactured and are presented in Figs. 1 and 2.

All knits under study were produced from each fibre type with the same knitting parameters in order to get equivalent stitch sizes, given by the geometrical parameters defined in Fig. 1. W, C and  $L_Y$  are respectively the average wale width, course height and yarn length per stitch. W and C were calculated from fabric dimensions and number of stitches, whereas  $L_Y$  is calculated from fabric weight and yarn linear density. The mean values calculated for all knits under study are  $W = 4.2 \pm 0.4$  mm,  $C = 2.5 \pm 0.1$  mm and  $L_Y = 17.5 \pm 3.9$  mm. The important dispersion of these parameters (5–20%) is related to the properties of the different fibres. The stitch size depends on knitting parameters but also on the loop



**Fig. 1.** Plain-knit basic architecture (*P*0): (a) photograph of a basalt fabric; (b) testing directions and (c) representative elementary cell (REC) and geometrical characteristics of the loop.

**Table 1** Fibre, yarn and resin properties.

Fibre type	E-glass	Basalt	Carbon	Resin type		
	Owens corning Vetrotex T30 111AX23	Basaltex BCF	Tohotenax HTA 5131 6 K Z10	Thermoset epoxy resin SICOMIN SR1710 + SD8822 hardener		
Fibre properties				Tensile properties		
Filament diameter Fibre density Young's modulus	$16\mu m$ $2620kgm^{-3}$ $80GPa$	13 μm 2700 kg m <sup>-3</sup> 84 GPa	7 μm 1760 kg m <sup>-3</sup> 238 GPa	Young's modulus Rupture stress Strain at break	3680 MPa 85 MPa 3.1%	
Yarn properties				Bending properties		
Number of filaments Yarn diameter Linear density (Dy) Bending stiffness	1200 0.58 mm 600 tex 0.31 N mm <sup>2</sup>	1672 0.56 mm 600 tex 0.20 N mm <sup>2</sup>	6000 0.57 mm 400 tex 0.16 N mm <sup>2</sup>	Modulus Maximal stress Strain at break	3720 MPa 136 MPa 5.20%	

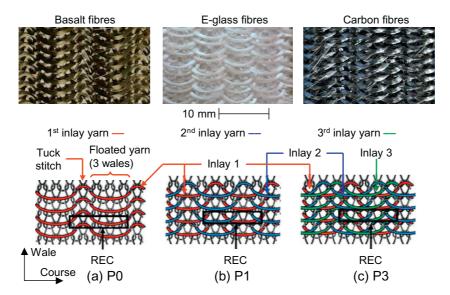


Fig. 2. Photographs, schematic views and representative elementary cell (REC) of the knitted fabrics: (a) with 1 inlay yarn (P1); (b) with 2 inlay yarns (P2) and (c) with 3 inlay yarns (P3).

**Table 2** Knitted fabrics characteristics.

		P0	P1	P2	Р3
Thx (mm) Fabric thickness (measured under 0.005 MPa pressure)	Basalt	$1.55 \pm 0.09$	$1.83 \pm 0.06$	2.22 ± 0.13 2.20 ± 0.05 2.08 ± 0.03	$2.78 \pm 0.75$
M <sub>S</sub> (g m <sup>-2</sup> ) Areal density	Basalt	797 ± 113	1044 ± 227	1382 ± 167 1337 ± 161 874 ± 152	

static equilibrium, which is determined by the yarn mechanical properties, structure and size of the fabric: inlay yarns insertion increases W and  $L_Y$  by modifying the static equilibrium.

Due to the shape of the needle hooks and the high velocity of the needles, some fibres are damaged during knitting. If glass and basalt fabrics show a very low amount of broken fibres, it is larger in carbon fabrics and increases with the number of inlays. The fabric characteristics are given in Table 2. The thickness is measured under a 5 kPa pressure following the Bueno et al. [28] conditions. For similar yarn diameters, the thickness mainly depends on the fabric texture and less on the fibre type. Indeed the variations are due to the fibre rearrangement, facilitated by a lower diameter and bending stiffness of carbon fibres. The areal density is similar for glass and basalt knits, and lower for carbon fabrics, in agreement with their yarn linear density.

# 2.3. Composite laminates

The reinforcements aforementioned were processed by VARTM in an aluminium rigid closed mould coated with a semi-permanent release agent (Loctite 44-NC Frekote). A vacuum pressure of 1 mbar was applied into the mould by vents located at each corner and the resin was injected through an inlet located at the centre of the mould (Fig. 3). The resin injection was performed at room temperature and the laminates were cured at 60 °C for 16 h on a heating plate. The composite laminates characteristics are given in Table 3. P0 composites were manufactured with 3 reinforcing layers to reach a fibre volume fraction close to 30%. P1 composites were manufactured with two layers providing the same fibre

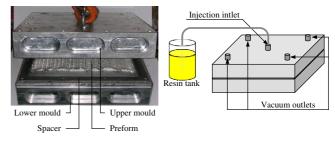


Fig. 3. VARTM processing conditions.

**Table 3** Composite plate characteristics.

	Reinforcement	P0	P1	P2	P3
	Number of plies	3	2	2	2
Plate thickness (mm)	E-glass	3.15	3.18	3.19	3.22
	Basalt	3.15	3.17	3.19	3.21
	Carbon	3.15	3.18	3.17	3.17
Fibre content (%)	E-glass	31	29	38	43
	Basalt	32	31	35	41
	Carbon	32	30	37	42
Porosity (%)	E-glass	2.8	5.4	7.1	5.4
	Basalt	3.4	8.4	7.5	6.6
	Carbon	2.6	10.1	4.1	3.2

content as *P*0. *P*2 and *P*3 composites contain also two layers as *P*1, but exhibit a higher fibre volume fraction. The porosity ratio varies between 2.6% and 10.1%. It is lower in *P*0 laminates, but no correlation can be made with the number of inlays. High ratios are mainly related to air bubbles entrapment during resin mixture. A careful selection of the specimen was carried out, so that the materials tested have the less voids as possible.

### 3. Mechanical behaviour of the knitted fabrics

#### 3.1. Specimen and testing method

Uniaxial tensile tests were performed on knitted preforms in wale and course-directions according to NF EN ISO 4606 stan-

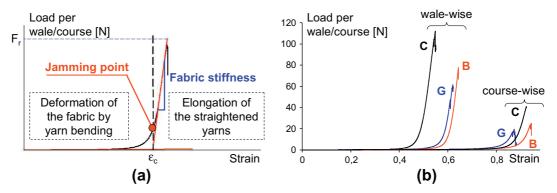


Fig. 4. General behaviour of a plain knit fabric: (a) characteristic parameters, and (b) tensile load-true strain curves of E-glass (G), basalt (B) and carbon (C) plain-knit fabrics in wale and course-directions.

dard dedicated to uniaxial tensile testing of woven fabrics, since no standards concerning knitted fabrics are available. The fabrics were directly knitted at the testing sample size  $(300 \times 100 \text{ mm}^2)$  since the knit cannot be cut without unravelling. A few courses of an elastomeric fibre were knitted on the long edges of the course-direction specimens to avoid unravelling while limiting its intrusiveness. The tests were carried out on an INSTRON 5800R electromechanical testing machine fitted out with pneumatic grips at the constant crosshead displacement rate of 50 mm min<sup>-1</sup>. An acceptable dispersion of the experimental data made it possible to select a representative load versus strain curve for the discussion, among the three available for each test condition. The results are given in load per wale or course versus true strain (Fig. 4).

#### 3.2. Behaviour of a plain knit fabric

The behaviour of a knitted fabric can be divided into two main steps separated by a jamming transition phase (Fig. 4a). The first step involves a slipping of the yarns at the contact point between adjacent loops. A change of the loop shape occurs by yarn bending and results in a linear behaviour with an important deformation of the fabric under low loads. This phenomenon occurs until the jamming transition phase, in which a transverse compression of the yarns occurs due to fibres rearrangement at the cross-over. The yarns bending tends toward a limit while the tensile load increases. At the end of this phase, the stitch is fully jammed. The load is then carried out by the stitch legs, subjected to tensile stresses, whereas shear stresses increase in the cross-over areas. A fast increase of the load is thus observed, resulting in a second almost linear domain of the load-strain curve. The first step is characterized by a critical strain  $(\varepsilon_c)$  which corresponds to the strain at the intersection of both linear domains. The last step is characterized by the fabric stiffness after full jamming of the stitches, corresponding to the slope of the curve, and by the failure load.

Fig. 4b illustrates the tensile behaviour of the plain-knit according to the loading direction and the fibre type fabrics (E-glass, basalt and carbon). Table 4 reports the values of the characteristic parameters of the tensile behaviour. Irrespective of the fibre type and loading direction, plain-knit fabrics behave similarly. A higher deformation ability and lower mechanical properties are observed in the course-direction. These properties result from the loop geometry related to the knitting process: the yarns run along the courses and are interconnected in the wale-direction. In Fig. 5 are shown schematic views of the fabric structure at initial state and stretched in wale or course-directions. Stretched in the course-direction, the stitches lose their loop shape to take a curvy

form, whereas in the wale-direction, the stitches present a stretched loop shape at jamming. In the first step, considering the low loads involved, the fibre type has a negligible effect on the fabric macroscopic behaviour and the critical strains reached are similar for a given loading direction (Table 4). After jamming, one yarn per course is subjected to tensile stress while two yarns per wale are loaded. It explains the higher rigidity and the higher strength reached by plain-knits loaded in the wale-direction. In this part, the fibre type is the most influent parameter. Best properties are obtained with carbon fibres whereas glass and basalt plain-knits exhibit similar properties despite the slightly higher values reached with basalt fibres.

#### 3.3. Inlay yarn fabric behaviour

The tensile behaviour of plain-knits with inlay yarns is plotted in Fig. 6 and the properties are summarized in Table 4. In the wale-direction, the insertion of inlays tends to decrease the strain at jamming, but this effect is slight compared to the standard deviation of the experimental results. Each yarn diameter being similar to each others, the strain levels reached are of the same order of magnitude, irrespective of the fibre type. Fig. 5 depicts a schematic view of a plain-knit with inlay yarns stretched in the wale-direction. In this case the inlay yarns are not tightened and do not contribute to the fabric behaviour except by slightly modifying the size of the stitches. P1, P2 and P3 fabrics behave similarly to P0. The differences observed are attributed to variations in the stitches dimensions and to inlay yarns that decrease slightly the critical strain. Considering the experimental dispersion, it could be concluded that the mechanical properties are almost the same with glass and basalt fibres. With carbon fibres, the amount of broken fibres increases with the number of inlay yarns, resulting in a loss of properties.

In the course-direction, the pneumatic grips maximal load limit was reached before P2 and P3 knits failure. The inlay yarns control the behaviour of the fabric and the plain-knit performs the function of carrying the inlays. Thus, the critical strain of the fabric is the strain required to tighten the floated yarns (Fig. 6), decreasing sharply the critical strain  $\varepsilon_c$  (Fig. 6). The fibre orientation of the straightened yarns close to the loading direction has for consequence a high increase of the fabric rigidity. The curves related to P1 fabrics show a second increase of the load after the rupture of the inlay yarns: plain stitches are loaded and the fabric behaves like a plain-knit stressed in the course-direction, but at lower strains levels due to the inlay yarns. The fabric rigidity increases systematically with the number of inlays, however this increase is more or less important depending on the fibre type. Comparing P0 and P3, an improvement of the rigidity of 770% is obtained with

**Table 4** Tensile properties of the knitted fabrics.

Knitted fabric	Fibre type	Critical stretched strain $\varepsilon_c$		Fabric rigidity (N/mm/mm)		Ultimate tensile load Fr (N)	
		Wale	Course	Wale	Course	Wale	Course
P0	Glass	0.57 ± 0.02	0.81 ± 0.03	1450 ± 63	443 ± 27	57.2 ± 7.5	18.3 ± 1.9
	Basalt	$0.59 \pm 0.02$	0.91 ± 0.01	1573 ± 384	$482 \pm 71$	$70.0 \pm 8.0$	21.5 ± 3.1
	Carbon	$0.52 \pm 0.03$	$0.88 \pm 0.02$	2292 ± 351	$768 \pm 93$	$111.0 \pm 3.0$	$37.2 \pm 3.9$
P1	Glass	$0.58 \pm 0.02$	$0.19 \pm 0.01$	1740 ± 138	1917 ± 176	54.0 ± 1.6	90.9 ± 12.0
	Basalt	$0.54 \pm 0.01$	$0.15 \pm 0.01$	1866 ± 164	$2315 \pm 94$	57.8 ± 1.9	108.6 ± 5.5
	Carbon	$0.50 \pm 0.01$	$0.15 \pm 0.05$	$1560 \pm 52$	$2349 \pm 366$	56.2 ± 1.8	$97.4 \pm 8.6$
P2	Glass	$0.54 \pm 0.01$	$0.18 \pm 0.01$	1525 ± 150	$3513 \pm 528$	52.2 ± 2.6	_a
	Basalt	$0.52 \pm 0.02$	$0.16 \pm 0.02$	1542 ± 57	2744 ± 177	$58.1 \pm 3.6$	_a
	Carbon	0.51 ± 0.01	$0.16 \pm 0.02$	1536 ± 74	2422 ± 142	$64.1 \pm 0.7$	_a
P3	Glass	$0.42 \pm 0.03$	$0.13 \pm 0.01$	1538 ± 121	3856 ± 269	66.2 ± 3.9	_a
	Basalt	$0.43 \pm 0.03$	$0.09 \pm 0.01$	1756 ± 165	2902 ± 86	$72.0 \pm 3.3$	_a
	Carbon	$0.47 \pm 0.00$	$0.13 \pm 0.01$	1370 ± 217	3137 ± 105	$71.5 \pm 3.8$	_a

<sup>&</sup>lt;sup>a</sup> Allowable load of the pneumatic grips reached before fabric break.

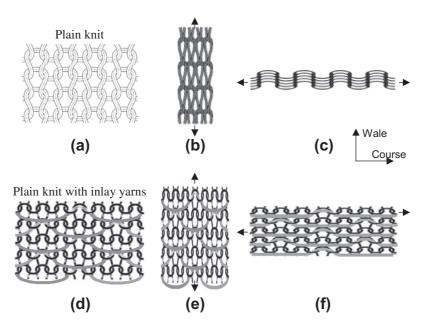


Fig. 5. Scheme of a plain-knit fabric and plain-knit with inlay yarns at (a and d) initial state, (b and e) stretched in wale direction and (c and f) stretched in course direction. This figure was drawn using pictures adapted from [29,30].

glass fibres whereas it is only of 308% with carbon fibres and of 502% with basalt fibres. Although carbon fibres are the most efficient, the properties obtained with the glass fibres are the best. Similarly, between *P*0 and *P*1, the ultimate strength has been multiplied by 5 with glass and basalt fibres and only by 2.5 with carbon fibres. Indeed, it is not possible to ensure a perfectly identical length of the different inlay yarns, preventing them to be tighten simultaneously. These trends are also attributed to the amount of broken fibres in the fabric: producing more complex fabrics induces an increase of damaged fibres. The shear strength depends on the filament cross-section and seems to be the key parameter that controls the ability of the yarn to be knitted without damage.

#### 4. Behaviour of knit fabric reinforced composites

#### 4.1. Specimen and testing method

Tensile tests were performed in the directions  $0^{\circ}$  (wale-direction),  $45^{\circ}$  and  $90^{\circ}$  (course-direction) on  $20 \times 180 \text{ mm}^2$  plain-knit (P0) and on  $25 \times 200 \text{ mm}^2$  P1, P2, and P3 composite specimens according to ASTM D3039/D3039M-00 recommendations. P1, P2 and P3 composite specimens are larger in order to get a full

representative elementary cell of the knitted reinforcement (see Fig. 2.) in the specimen width. Four  $20 \times 40$  or  $25 \times 40$  mm² aluminium tabs were bonded on each specimen in order to limit stress concentration in the clamping areas. The tensile tests were carried out on an INSTRON 5800R electromechanical testing machine at the constant crosshead displacement rate of 0.5 mm min $^{-1}$ . The longitudinal strain is measured with a 50 mm gauge length extensometer. Each configuration was tested three times. In this section a representative test among the three was selected to plot and discuss the results.

#### 4.2. Mechanical testing results

Due to the matrix involved in this study, characterised by a low strain at break, the fibrous architecture does not change during the tests, and the composite behaviour is mainly related to the initial shape of the knitted loop.

The stress–strain curves are plotted in Fig. 7 and the mean values of the mechanical properties are given in Table 5. The behaviour of knit reinforced composites is elastic damageable with various behaviours in the non-linear domain. They exhibit a high behaviour diversity depending on the loading direction, fibrous

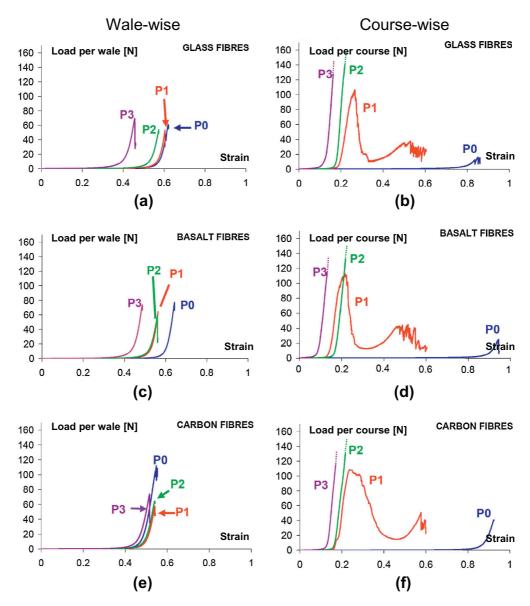


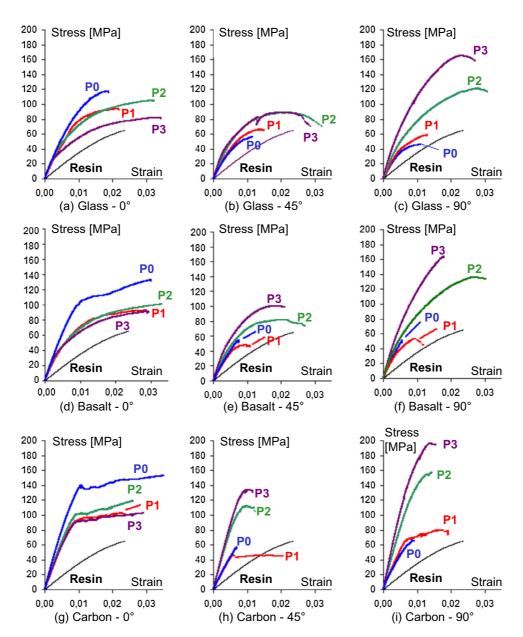
Fig. 6. Wale and course direction load versus true strain curves for dry glass, basalt and carbon fibre knitted fabrics. The dotted lines for P2 and P3 knits in the course direction indicate that the failure has not been reached.

architecture and fibre type. In all cases the fibre reinforcement increases the material elastic modulus when comparing to the net resin sample behaviour. *P*0 reinforced composites exhibit an anisotropic behaviour with higher properties in 0° direction and similar properties in 45° and 90° directions. In this latter direction, the ultimate strength is lower than the net resin strength. The behaviour of the *P*0 composites reinforced with basalt and carbon is close to a brittle elastic one, while the *P*0 composite reinforced with glass fibres, whose properties are the lowest, exhibit a wider non-linear domain before failure (45° and 90°).

Inserting one inlay yarn per course in plain-knit reinforcement has for consequence a slight decrease of the composite properties in the 0° direction attributed to a different arrangement of the fibres distribution in the material at similar fibre volume fractions. The insertion of one inlay yarn has no significant effect in the directions 45° and 90°. A slight improvement is obtained with glass fibres whereas a small decrease of modulus and strength is observed in the direction 45° with basalt and carbon fibres. Similar properties are obtained in the direction 90° except a wider non-linear domain and a slight increase of strength with carbon fibres.

Indeed, the floated yarns are not straight but curved, limiting a major orientation of the fibres in the direction 90°, and limiting the expected improvement of the properties in this direction. In addition, a higher porosity is observed in *P*1 (Table 3).

Increasing the number of inlay yarns per course leads to an important improvement of the properties in the direction 45° and an even larger increase in the direction 90° whereas no significant change is observed in the direction 0°. However, in the loading direction 0°, after a loss of properties with the insertion of the first inlay yarn (P1), an increase of the properties is observed with the insertion of the second inlay yarn (P2), attributed to the increase of the fibre content, and again a decrease in properties is observed with the third inlay yarn (P3). In the others loading directions, between PO and P3, according to the fibre type, an increase of 60-105% of the modulus and 61-134% of the ultimate stress is experienced in the testing direction 45°, and an increase of 76-134% of the modulus and 212-288% of the ultimate stress is reached in the loading direction 90°. This is due to a cross-effect of fibre content and fibre distribution. In a general way, the stressstrain curves show that the non-linear domain of composites



 $\textbf{Fig. 7.} \ \ Representative engineering stress-engineering strain curves of the tensile behaviour of the knit-reinforced composites in the directions $0^{\circ}$ (wale), $45^{\circ}$ and $90^{\circ}$ (course).$ 

**Table 5**Tensile properties of the knit reinforced composites.

Fibre type	Knitted fabric	Longitudinal elastic modulus (GPa)			Ultimate stress (MPa)		
		<b>0</b> °	45°	90°	0°	45°	90°
Glass	P0	11.2 ± 0.1	8.1 ± 0.2	8.2 ± 0.0	112.2 ± 4.8	54.1 ± 2.2	42.3 ± 3.4
	P1	$9.8 \pm 0.3$	$9.0 \pm 0.4$	$8.8 \pm 0.2$	$90.7 \pm 3.2$	$66.4 \pm 0.2$	57.1 ± 3.0
	P2	$10.8 \pm 0.2$	$11.3 \pm 0.3$	$12.2 \pm 0.2$	$102.8 \pm 2.4$	86.7 ± 2.1	124.1 ± 5.1
	Р3	$12.1 \pm 0.2$	$13.3 \pm 0.4$	$17.2 \pm 0.6$	$82.4 \pm 0.5$	$87.3 \pm 2.7$	164.2 ± 1, 2
Basalt	P0	12.8 ± 0.1	$8.5 \pm 0.1$	$9.3 \pm 0.8$	136.9 ± 5.4	$54.8 \pm 0.9$	51.1 ± 0.3
	P1	$9.7 \pm 0.4$	$8.3 \pm 0.3$	$9.0 \pm 0.2$	$84.2 \pm 9.2$	$47.7 \pm 1.0$	52.5 ± 2.6
	P2	$10.8 \pm 0.2$	$10.8 \pm 0.2$	$12.0 \pm 0.8$	97.1 ± 3.7	85.8 ± 4.5	130.0 ± 6.6
	P3	$11.8 \pm 0.3$	$13.5 \pm 0.1$	$16.4 \pm 0.8$	$87.9 \pm 4.6$	$95.8 \pm 6.0$	$160.9 \pm 5.4$
Carbon	P0	17.5 ± 0.7	$8.9 \pm 0.0$	$9.3 \pm 0.7$	152.1 ± 0.2	55.6 ± 3.2	63.3 ± 3.6
	P1	$12.7 \pm 0.1$	$8.8 \pm 0.2$	10.6 ± 1.3	107.9 ± 15	$46.0 \pm 0.7$	$75.2 \pm 7.6$
	P2	$13.7 \pm 0.7$	$15.0 \pm 0.8$	19.2 ± 1.4	121.5 ± 4.6	$106.6 \pm 6.8$	$163.4 \pm 6.3$
	Р3	$12.7 \pm 0.8$	$18.2 \pm 1.0$	$21.8 \pm 0.2$	104.9 ± 3.0	130.2 ± 5.2	$197.4 \pm 0.0$
Resin		3.62			65.5		

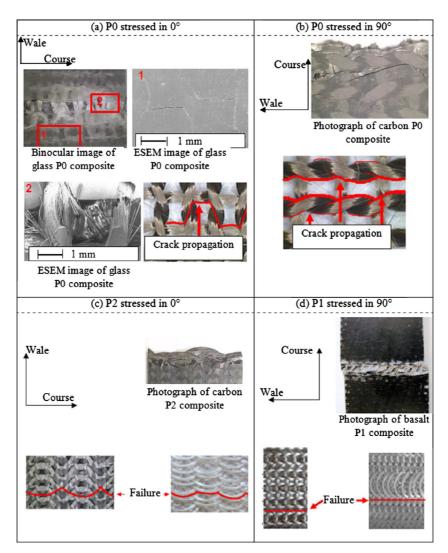


Fig. 8. Failure of plain-knit with and without inlay yarns reinforced composites stressed in  $0^{\circ}$  (wale) and  $90^{\circ}$  (course) directions.

reinforced by plain-knits with inlay yarns is relatively strong in comparison to *P*0 reinforced composites. Increasing the number of floated yarns leads to more diffuse damages and to higher strain at break.

The effect of the fibre type depends on the fibrous architecture and the loading direction. In the direction 0° the fibre type effect on the properties decrease is similar irrespective of the fibre types: it is significant for the first inlay yarn whereas a low decrease is noticed when increasing the number of inlay yarns. On the contrary, the properties in the directions 45° and 90° are directly related to the number of inlays. In these directions, the more the number of inlay yarns is, the higher the properties of carbon fibres increase, as shown by the increasing gap between the results obtained with carbon fibres and those obtained with glass and basalt fibres. In all cases, best properties are obtained with carbon fibres and this is in agreement with the fibre properties given in Table 1.

#### 4.3. Fracture analysis

Figs. 8 and 9 show photographs and images of cracked specimens as well as a schematic view of the fracture surface on the knit samples. In the loading directions 0° and 90° (Fig. 8a and b), the failure occurs systematically in a plane perpendicular to the loading direction. For *P*0 composite specimens tested in the direction 0°, the cracks propagate along a crenel-shape pattern in the vicinity of the fracture (Fig. 8a; ESEM image 1). The crack initiates

perpendicularly to the loading direction by a separation between a fibre tow and the matrix and propagates between a fibre-rich area and a resin-rich area up to the cross-over points. When the crack meet the cross-over area, a bifurcation occurs and the material damages under shear stresses up to the leg of the stitch. The energy required by this mechanism allow the same phenomenon to take place elsewhere before to go on. At least one crack propagates along adjacent stitches until the specimen failure throughout the width of the sample. The two parts of the specimen are then connected by strands of fibres that are then loaded up to the failure (Fig. 8a; SEM image 2). This explains the high strain values reached with *P*0 composites in the direction 0° in contrast to 45° and 90° (Fig. 7).

When inserting inlay yarns in the reinforcement, the rupture plane remains transverse to the loading direction between the rows of stitches (Fig. 8c). Whatever the fibre type or the number of inlays, the composite damages similarly. The cracks are only located in the failure area. The cracks initiate between resin-rich and fibre-rich areas and propagate along floated yarns. As for *PO* composites, the two parts of the specimen are then connected by strands of fibres of the plain knit stitches which are then loaded up to the failure.

In the loading direction  $90^{\circ}$ , the fracture occurs between resinrich and fibre-rich areas, by debonding between fibre tows and matrix whatever the fibrous architecture (Fig. 8b and d). The cracks follow perfectly the stitches legs of a single wale. The inlay yarns

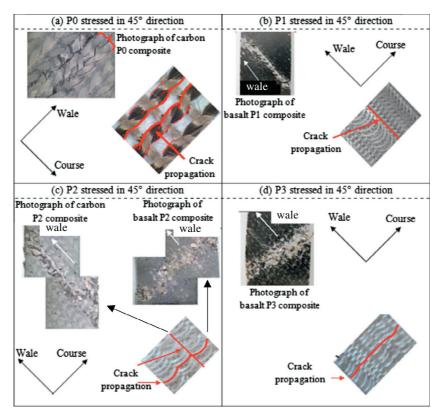


Fig. 9. Failure of plain-knit with and without inlay yarns reinforced composites stressed in 45° direction.

plays the role of slowing down the cracks propagation by increasing the fibre volume fraction perpendicular to the load in this area, which leads to increase the material strength.

In the PO specimens tested in the direction 45°, the fracture occurs in the same way as for those tested in the direction 90° (between a rich and a poor fibre region). A 45° oriented fracture is then obtained (Fig. 9a), and this explains the equivalent properties obtained with P0 composites tested at 45° and 90°. When inserting one inlay yarn, the behaviour is quite similar (Fig. 9b). By inserting two inlay yarns, two cases may arise (Fig. 9c). In the first case, the crack propagation is similar to those of the composites tested in the direction 90° (perpendicular to the inlay yarns, Fig. 8d). This is observed with carbon fibres reinforced composites. In the second case, observed with basalt and glass fibres composites, the crack propagation is similar to the P2 composites tested in the direction 0° (Fig. 8c), i.e. along the floated yarns, leading to a 45° oriented fracture surface occurring along the inlay yarns. This latter failure mode systematically occurs with P3 specimens tested in the direction 45°. With one inlay yarn, the cross-over area is strongly sheared, especially in the tuck stitch area since the floated yarn strengthens the material. With three inlays, the floated yarns constitute a continuous unidirectional stripe. High shear strains arise thus in the vicinity of these stripes in areas where the fibres are mainly wale-wise oriented. The failure occurs between these two areas. With two inlays, the tucked wale is not efficiently reinforced by the floated yarns since they are not perfectly parallel to the course (see Fig. 2). Both failure mechanisms are thus possible depending on the fibres properties and on the amount of broken fibres.

#### 5. Conclusion

Insertion of up to three inlays yarns in a plain knit fibre architecture was investigated in order to improve the mechanical

behaviour of knit-reinforced composites. Each knit fabric was manufactured with E-glass, basalt and carbon fibres, using the same stitch size and yarn diameter and with a tuck every four wale for the inlaid knits.

The tensile behaviour of each knit in the wale-direction shows that the wale-wise behaviour is not significantly affected by the inlays, except by modifying the cross-over areas with the tucks. At the contrary, the course-wise direction is significantly reinforced and about five times less deformable with inlay yarns. Indeed, the behaviour of the knit in this case is mainly the behaviour of the inlays. The effect of the fibre type on the behaviour of the knit is negligible compared to the effect of the inlays.

The stress–strain behaviour of the knit reinforced composite material depends on the fibre type but also strongly on the knit construction. Carbon fibres provide the best mechanical properties but the results are not proportional to the intrinsic properties of the fibres when comparing with E-glass and basalt. Inlaying floated yarns does not allow to improve the mechanical behaviour of the composite since it decreases the strength in the direction 0°, but it makes possible to control the anisotropy of the material properties. Highest mechanical properties are reached in the direction 0° for plain-knit reinforced composites, in 90° direction for *P*3 knitted composites and an almost isotropic material is obtained for the *P*2 knitted composites.

#### Acknowledgements

This study was carried out with the financial support of the Region Midi-Pyrénées in the framework of the project CORTEX. The authors would like to thank JTT Composite, and especially MM. Nicolas Dumont and Francis Planel for the design and manufacturing of the knits.

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