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Experimental investigation of the kinematics of post-impact ice fragments

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Abstract

Hail is more hazardous for aircraft engines compared to rain and snow, mainly, because of its solid nature and high water content. In extreme cases it can lead to **engine** flame out. In order to avoid such situations, aero engines should be designed to withstand hail ingestion. For this purpose we have studied the post-impact characteristics of ice, such as particle velocity and directions of travel. To achieve this goal, a large experimental program has been undertaken, in which spherical ice specimens were projected against a rigid plate. Three specimen diameters **(6.2, 12.9 and 27.5 mm)** and four impact angles **(20, 45, 75 and 90°)** were considered, as well as a wide range of impact velocities **(60-200 m/s)**. From this experimental work, we can conclude that the ice fragments formed after impact do not bounce back and that the post-impact ice trajectory angle is lower than 2°. This is in line with observations found in the literature. On the other hand, the ice fragments are mainly organised in a circular cloud, when observed in the

target plane. The center of this cloud has the same velocity as the initial ice ball tangential impact velocity. Furthermore, the cloud radius expands with a rate proportional to the ice ball normal impact velocity. Finally, each fragment inside the cloud has a relative velocity which varies linearly with its distance from the cloud center. These experimental observations should be very helpful **in developing** models and simulations of hail ingestion by aircraft engines.

Key words: Hailstorm, aircraft engines, ice fragmentation, hail ingestion.

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1 Aircraft engines and structures can face several kind of impacts, mainly
2 during take off and landing, such as from birds [1], tyre debris [2, 3, 4],
3 hail [5] or sand [6, 7, 8]. These kinds of impacts can induce damage in
4 some parts of the aircraft structure and lead eventually to **an** aircraft crash.
5 Furthermore, impact-induced fragments can be ingested by aircraft engines
6 and cause engine power loss. Therefore, aero engines have to satisfy safety
7 requirements against hail ingestion to obtain certification.

8 Hail ingestion, as water ingestion [9, 10, 11], can cause engine power loss
9 and in extreme cases engine flame out [12, 13, 14]. In the event of ingested
10 hail passing through the fan into the core engine, changes in temperature,
11 density and specific heat ratio will occur. As a result the engine will lack
12 combustion efficiency or/and stability which can lead to its loss of thrust
13 and eventually to a flame out. Engine manufacturers and aviation regulatory
14 agencies have a common aim: preventing engine flame outs, roll-back or other
15 **operational** effects that could result in an in-flight shut down when flying
16 by stormy weather. These considerations are a major concern for the design,
17 the certification and the **operation** of modern high bypass aircraft engines.

3

₁₈ The engine manufacturers need powerful numerical tools to take into ac-
₁₉ count inclement weather threats in the engine design. A first model was
₂₀ proposed by [15]. However, to be realistic, these simulations must rely on
₂₁ experimental investigations of hail impact in order to understand the phe-
₂₂ nomenon, develop appropriate constitutive equations and identify the rele-
₂₃ vant parameters.

₂₄ A number of references can be found in the literature about ice-impact
₂₅ induced damage e.g. [5, 16, 17], fatigue induced damage e.g. [16] or high
₂₆ velocity ice impacts on composite panels e.g. [5, 17]. On the other hand, low
₂₇ velocity ice impacts, i.e., for velocities lower than $10 \, \mathrm{m.s^{-1}}$, were investigated
₂₈ in Refs. [18, 19, 20, 21, 22, 23]. These latter works aimed at defining the
₂₉ transition between the rebound and fragmentation regimes.

₃₀ In order to understand hail ingestion by aircraft engines, Pan, Render
₃₁ et al. [24, 25, 27, 26, 28, 29, 30] have studied the impact fragmentation
₃₂ characteristics of hailstones on rigid targets. They were interested in post-
₃₃ impact particle size, particle velocity and travel directions. In Ref. [26]
₃₄ a patternator technique was used. A still photography technique was also
₃₅ employed in Ref. [25, 28]. These investigations were extended to study the
₃₆ effect of target curvature [29] and target rotation [27]. Subsequently, the
₃₇ methodology was applied to consider the hail mass distribution behind a
₃₈ rotating fan and spinner assembly [30]. Pan, Render et al. [24, 25, 27, 26,
₃₉ 28, 29, 30] have shown that the Rosin-Rammler [31] distribution, also called
₄₀ Weibull distribution [32], which is also used in rock fragmentation [33], is
₄₁ well suited to represent the particle size and particle velocity distributions.

₄₂ In the present paper, we investigate the kinematics of post-impact ice

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₄₃ fragments. Render and Pan [26] have observed that the fragments do not

₄₄ bounce back, i.e., the post-impact angles are very low with the fragments

₄₅ trajectories remaining almost in the plane of the target. This result was ob-

₄₆ tained by a patternator technique. One aim of the present work is to check

₄₇ this result against high speed photography observations. On the other hand,

₄₈ the fact that the particle velocity distribution was found by [28] to be of the

₄₉ Rosin-Rammler [31] type is not enough to describe the fragments movement

₅₀ completely. Indeed, the distribution only indicates what percentage of mass

₅₁ fragments moves at a given velocity. In the present work, we are also inter-

₅₂ ested in obtaining a better description of the kinematics of fragments in the

₅₃ plane of the target. To this aim a wide experimental program was carried

₅₄ out to assess the influences of the ice projectile diameter, impact angle and

₅₅ impact velocity.

₅₆ 1. Method

₅₇ 1.1. Ice preparation

₅₈ The ice spherical specimens were prepared using moulds made of two

₅₉ hollow half-spheres in order to facilitate the removal of the ice ball. The

₆₀ moulds were filled with cooled water (between 4 and 8 °C), then placed in a

₆₁ freezer (between -25 and -18 °C). Subsequently, the ice balls were removed

₆₂ from the moulds after 24 hours and kept in the freezer until the beginning of

₆₃ the test.

₆₄ Hailstones can have diameters ranging from 2 to 100 mm [26]. In the

₆₅ present study, three diameters (D_0) were considered: 6.2, 12.9 and 27,5 mm

₆₆ corresponding to a mass of approximately 0.1, 1 and 10 g, respectively. It

5

67 must be noted that since the tests were performed at room temperature

68 melting can induce important dimension inaccuracies in the ice of which

69 diameter is lower than 6 mm. On the other hand, it is hard to analyse

70 images of projectiles with large diameters because of the huge number of

71 fragments formed. Note that Pan and Render [26, 28] worked mainly with

72 ice balls 12.7 mm in diameter.

1.2. Experimental set-up

74 The experimental set-up consists of a gas gun, that comprises a high

75 pressure chamber, a cooled chamber and a tube, a target plate and the in-

76 strumentation (Figure 1). The high pressure chamber works with nitrogen

77 gas and can go up to 85 bar (8.5 MPa). The actual applied pressures were

78 between 0.6 and 3.9 bar (0.06 and 0.39 MPa), which results in impact veloci-

79 ties (V_0) ranging between 60 and 200 m.s^{-1}. This velocity range corresponds

80 to that of a low engine power regime, such as during aircraft descent. It is in

81 this regime that inclement weather effects are mostly encountered [14]. It is

82 worth noticing that Pan and Render [26, 28] were limited to impact velocities

83 in the range 100 to 175 m.s^{-1}.

84 The cooled chamber is made out of stainless steel. It is located between

85 the high pressure chamber and the gas tube. Its role is to house the ice

86 projectile while preventing it from melting before the strike. It is cooled

87 by the flow of an antifreeze - water mixture maintained at a temperature

88 between -4 and 0 °C by a cooling generator. The cooled chamber is connected

89 to the canon made out of a stainless **steel** tube. The cooled chamber inner

90 dimensions and the tube inner diameter are adapted to the ice ball diameters,

91 so that no sabot is needed. Since we performed our tests with three ice ball

diameters, three different cooled chambers and three different tubes were used.

The target is a glass plate ($200 \times 200 \, \text{mm}^2$) $20 \, \text{mm}$ in thickness. The choice of glass as the target material was motivated by two reasons. Firstly, glass has a stiffness comparable to some metallic material such as aluminium (its Young's modulus is about $64 \, \text{GPa}$). Secondly, since glass is transparent, it is possible to use back-lighting which yields more homogeneous light on the target than front-lighting.

This target is mounted on a bearing system allowing two degrees of freedom (Figure 2): a translation along the vertical direction and a rotation around the horizontal axis perpendicular to the (horizontal) gas tube thus allowing changes in the impact angle α. The impact angle is defined as the angle between the projectile trajectory before impact and the normal to the plane of the target. In the present work, four values of the impact angle α were considered: 20, 45, 67 and 90 °.

The experimental set-up is instrumented by two laser barriers to measure the ice ball impact velocity. Furthermore, two high speed video cameras were used in order to record the post-impact characteristics. The first camera was used to film the lateral side, i.e., the plane perpendicular to the target that contains the gas-tube axis, whereas the second camera was used to film the front side, i.e., the target surface. The two cameras were synchronised. The trigger was given by the first laser barrier. The adopted acquisition frequency of the cameras was 15000 frames per second.

The initial conditions of all our tests are given in Table 1.

7

2. Results

2.1. General observations

Figures 3 and 4 show the sequences of images for an ice ball fragmentation obtained with the front camera and the lateral camera, respectively. In this test, the ice ball diameter, impact angle and impact velocity were 12.9 mm, $20\,°$and $120.8\,\mathrm{m.s^{-1}}$, respectively (Test No. 15). Both cameras were synchronised and took 15000 frame per second, so that, for example, photographs (a) on Figures 3 and 4 were taken at the same time. The resolutions of the front and lateral camera were $320\,\times\,832\,\mathrm{pix^2}$ and $1024\,\times\,128\,\mathrm{pix^2}$, respectively. Figures 3(a) and 4(a) show the ice specimen just before impact. The first contact between the ice ball and the plate is shown in Figures 3(b) and 4(b). Fragmentation begins at Figures 3(c) and 4(c).

From Figures 3 and 4, we can divide the fragmentation process into two stages. In a first stage, only the front part of the ice ball fragments (Figure 3(c) to (e) and Figure 4(c) to (e)). The created fragments are ejected first and form a low density chaotic cloud. In a second stage, the remaining ice projectile collapses (Figure 3(f) to (l) and Figure 4(f) to (l)) yielding a high density, almost circular, cloud. We can also notice from Figure 4 that the post-impact angle, which we define as the angle between the fragments trajectories and the plane of the target, is very low. Fragments trajectories lie almost in the target plane.

The above observations are also confirmed in Figure 5 where photos (a) to (f) show images of ice fragmentation for a normal impact at a velocity of $150.4\,\mathrm{m.s^{-1}}$. The two photos in Figure 6, which are magnifications of Figures 5(b) and 5(e), show that the first ejected particles which make the low density

cloud have a high velocity, higher than the initial ice ball impact velocity. **This is consistent with results of Render and Pan [26] who observed also that particles formed immediately after impact moves greatly faster than the approaching ice ball velocity.** Figure 6(a) shows two particles with a post-impact velocities of $335\,\mathrm{m.s^{-1}}$ and $443\,\mathrm{m.s^{-1}}$, respectively, to be compared to the ice ball impact velocity of $150.4\,\mathrm{m.s^{-1}}$. On the other hand, the fragments created during the second stage, that form the high density cloud, have lower velocities. As an illustration Figure 6(b) shows two particles, one with a post-impact velocity of $52\,\mathrm{m.s^{-1}}$ and the other with a post-impact velocity of $18\,\mathrm{m.s^{-1}}$. The difference in the post-impact velocities can be explained as follows. At the beginning of the impact, the ice ball has a high initial velocity and elastic energy is stored in the yet unfragmented material. Only a small ratio of the mass is fragmented. Particles are ejected with a high velocity, which can be even higher than the nominal impact-velocity owing to the energy release. Furthermore, as the first stage cloud has a low density of fragments, the likelihood that collisions occur is low. As the second stage of the fragmentation process begins, the remaining unfragmented ice mass has already lost a part of its kinetic energy during the first stage. In addition, the second stage cloud of fragments has a higher density. Hence, multiple collisions happen that consume part of the kinetic energy of the particles.

It is worth to point out here that the post-impact particles velocities were determined by measuring the particles coordinates (in pixels) in several consecutive images. **This particle tracking is processed manually. In order to have a full comprehension of the post-impact fragment**

166 **kinematics, the tracking is operated on the center of the high den-**
167 **sity cloud and its circular counter and also on three to seven parti-**
168 **cles inside the cloud.** The velocities were first determined in pix.s^{-1} then
169 converted to m.s^{-1}from the calibration coefficient determined from a cali-
170 bration image **which corresponds to a mellimetric grid. Considering,**
171 **the camera settings, the accuracy of displacement measurements**
172 **are at best 0.01 pix and at worst 0.1 pix**.

173 *2.2. Kinematics in the normal plane*

174 In this section, we are mainly interested in the fragments post-impact
175 angle. The ice ball impact angle α and fragments post-impact angle γ are
176 represented in Figure 7. The angle γ is defined as the maximum angle be-
177 tween the fragments trajectories and the plane of the target. This angle is
178 determined from the lateral camera images as shown in Figure 8. Figure 9
179 shows the different values of post-impact angles obtained for all the experi-
180 ments. It seems that the post-impact angle γ is independent of the ice ball
181 impact velocity V_0, of the impact angle α and of the projectile diameter D_0.
182 All the measured values of γ lie in the range between 1 and 2°. Therefore,
183 the normal velocity of fragments can be considered as negligible with respect
184 to their tangential velocity, that is, the fragments kinematics is governed by
185 their sole tangential velocity **which is in line with the findings of Ren-**
186 **der & Pan [26] who observed a low bounce normal to the surface**
187 **of the ice ball fragments**.

10

2.3. Kinematics in the target plane

As explained in section 2.1, the fragmentation process is divided into two stages. The first stage yields a low density, high velocity, chaotic cloud of fragments. The second stage is characterized by a high density, low velocity, circular cloud. In this section, we focus on the kinematics of the fragments created during the second stage. Mainly, we are interested in the velocity of the cloud center, the rate of surface expansion of the cloud and in the velocities of fragments relative to a (moving) reference point in the moving cloud. To this aim we analyse the images obtained by using the front camera (Figure 10). For each image i, the contour of the second-stage cloud of fragments is approximated by a circle of center C_i and radius R_i. By analysing the sequence of images for each test, we can obtain the evolution with time of the cloud center and radius.

Figure 11 shows the schematic of the superposition of two cloud contours, one corresponding to the instant when the specimen hits the target ("first-contact" time; corresponding image i_0), the other at a post-impact time (image i). Note that the "initial" cloud, represented as a circle, should in fact be the elliptical projection of the undamaged spherical specimen onto the target plane (along its trajectory before impact). The y axis corresponds to the projection of the trajectory of the ice ball before impact on the plane of the target. The origin of coordinates is taken at the center C_{i_0} of the initial cloud, so that $x_{i_0} = y_{i_0} = 0$. Our analyse of the different tests has shown that the contours centers remain aligned with the y axis (i.e., in the same vertical plane that contains the ice ball trajectory before impact) (see Figure 12). In other words, for any image i, the center of the fragments cloud

11

213 is such that $x_i = 0$, and is thus defined only by y_i.

214 *2.3.1. Velocity of the cloud center*

215 Figure 13 is a plot of the positions of the centers C_i, defined by the

216 coordinate y_i, as function of time. This figure corresponds to a test performed

217 with an ice ball of diameter $D_0 = 12.9\,\text{mm}$, an impact angle $\alpha = 67.2\,°$and

218 an impact velocity $V_0 = 118.8\,\text{m.s}^{-1}$(Test No. 24). The variation of y_i versus

219 time is almost linear and can be approximated by

$$y(t) = at + b, \tag{1}$$

220 where a and b are two constants. In the case shown in Figure 13, $a =$

221 $42.05\,\text{m.s}^{-1}$and $b = -111.9\,\text{mm}$. a is the velocity of the cloud center in the

222 target plane, i.e., the tangential velocity of the cloud center, which is denoted

223 by V_{C_t} in the following. It is worth to compare V_{C_t} to the ice ball tangential

224 impact velocity V_{0_t} which is such as

$$V_{0_t} = V_0 \cos \alpha, \tag{2}$$

225 where V_0 and α are the ice ball impact velocity and impact angle, respectively.

226 In the case shown in Figure 13, the impact velocity V_0 is $118.8\,\text{m.s}^{-1}$and the

227 tangential impact velocity V_{0_t} equals $46.04\,\text{m.s}^{-1}$. Hence, the ratio V_{C_t}/V_{0_t}

228 is equal to 0.91. This means that the tangential velocity of the cloud center

229 is of the same order of magnitude **as** the normal impact velocity. In order

230 to confirm this conclusion, we have plotted the ratio V_{C_t}/V_{0_t} for the all the

231 experiments in Figure 14(a). The values of the ratio V_{C_t}/V_{0_t} are mostly

232 between 0.9 and 1.1. In order to reduce the effect of experimental noise we

233 have computed the average value for each experimental configuration (see

12

234 Figure 14(b)). This figure shows that V_{C_t}/V_{0_t} is independent of the diameter

235 D_0 and of the impact angle α. **Nevertheless, more scatter is obtained**

236 **with results of $D_0 = 6.2$ mm than those of $D_0 = 12.9$ and 27.5 mm.**

237 **Indeed, the geometry of low-diameter ice balls is more sensitive**

238 **to melting than the geometry of balls with important diameters.**

239 **More precisely, melting can affect the spheric and smooth shape**

240 **of ice balls with low diameter. As the scatter in results is lower**

241 **than 10%, we conclude that melting has an insignificant effect on**

242 V_{C_t}/V_{0_t} **ratio. Namely, the average ratio V_{C_t}/V_{0_t} is between 0.96**

243 **and 1.01, except for two configurations, ($D_0 = 6.2$ mm; $\alpha = 45°$) and**

244 **($D_0 = 27.5$ mm; $\alpha = 67°$). As the average value for all experiments**

245 **is 0.986, we can state** that the tangential velocity of the cloud center V_{C_t}

246 is almost equal to the ice ball tangential impact velocity V_{0_t}. Then we can

247 assume that

$$V_{C_t} \approx V_{0_t}. \tag{3}$$

248 *2.3.2. Surface expansion rate of the fragments cloud*

249 The surface expansion rate of the cloud of fragments is characterised by

250 the rate of expansion of the cloud radius. For each test we study the variation

251 of the circular cloud radius versus time. This variation is almost linear and

252 can be approximated by

$$R(t) = V_R t + R_0, \tag{4}$$

253 where $R_0 = D_0/2$ is the initial radius of the ice ball and $V_R = \dot{R}$ is the radius

254 expansion rate.

255 Figure 15(a) shows the values of V_R for the all our experiments as a

256 function of the ice ball normal impact velocity V_{0_n}. V_{0_n} is the component of

13

₂₅₇ the ice ball impact velocity normal to the target plane:

$$V_{0_n} = V_0 \sin(\alpha). \tag{5}$$

₂₅₈ Figure 15(a) shows that the radius expansion rate increases with increas-
₂₅₉ ing normal impact velocity. Quantitatively, the radius expansion rate can be
₂₆₀ approximated by the linear relation

$$V_R \approx K_n V_{0_n}, \tag{6}$$

₂₆₁ where K_n is a constant. From Figure 15(a), $K_n \approx 0.585$.

₂₆₂ The ratios $K_n = V_R/V_{0_n}$ for the different experiments are plotted in
₂₆₃ Figure 15(b). It seems that K_n is independent of the test configuration:
₂₆₄ there is no special tendency neither as a function of the ice ball diameter nor
₂₆₅ of the impact angle. This is also confirmed by Figure 15(c) where the average
₂₆₆ value for each configuration is represented. **Moreover, we compare the**
₂₆₇ **values of K_n obtained with tests of similar conditions in Table 2.**
₂₆₈ **More precisely, we compare K_n of Tests 14 and 15 ($D_0 = 12.9$ mm,**
₂₆₉ **$\alpha = 20\,^\circ$, $V_0 \approx 120$ m/s), 17 and 18 ($D_0 = 12.9$ mm, $\alpha = 20\,^\circ$, $V_0 \approx$**
₂₇₀ **158.5 m/s) and 19 and 20 ($D_0 = 12.9$ mm, $\alpha = 45\,^\circ$, $V_0 \approx 158$ m/s).**
₂₇₁ **It comes that the difference of the two values of K_n obtained by**
₂₇₂ **two tests in the same configuration is around 10%. Hence, we**
₂₇₃ **can confirm that K_n is independent of the test configuration and**
₂₇₄ **that the scatter observed in the values of K_n are rather due to**
₂₇₅ **eperimental inaccuracies and not to a more complex behaviour of**
₂₇₆ **the fragmentation process.**

2.3.3. Fragments relative velocities

In sections 2.3.1 and 2.3.2, we have derived relations for the fragments center velocity and fragments cloud radius rate, respectively. In other words we can describe the velocity of the fragments near the center of the cloud and at its boundary. In the section, we are interested in the velocity of fragments inside the cloud. This velocity is studied in the mobile reference frame, i.e., we measure the velocity relative to the center of the cloud (Figure 16).

First, it is observed that the relative velocity of fragments V_r at a distance r from the center is dependent only on this distance:

$$\vec{V}_r \equiv \vec{V}_r(r). \tag{7}$$

Second, the relative velocity is along the radius line, i.e.,

$$\vec{V}_r(r) = V_r(r)\vec{e}_r, \tag{8}$$

where $\vec{e}_r$ is defined in Figure 16.

The variation of V_r in terms of the distance r is represented in Figure 17 for different experimental configurations. It comes from this figure that the velocity V_r increases with r, which explains the cloud expansion. The fragments near the center move slower than that at the boundary. Moreover, the variation of of V_r versus r can also be approximated by a linear relation

$$\vec{V}_r(r) \approx \nu r, \tag{9}$$

where ν is a constant which can be determined by considering the radius expansion rate V_R. Indeed, the velocity of fragments at the boundaries $V_r(R)$ should be equal to radius expansion rate V_R, so that the relative velocity of

15

296 a fragment inside the cloud can be expressed as

$$\vec{V}_r(r) \approx V_R \frac{r}{R}. \tag{10}$$

297 **3. Discussion**

298 Despite the large experimental program undertaken, several possibly in-
299 fluencing parameters were neglected in this study, for example the tempera-
300 tures of the ice ball and of the target.

301 Temperature influence several mechanical properties. Hence, there is no
302 reason to neglect its effect on the ice fragmentation process. However, natural
303 hail temperatures are generally between -60 °C and 0 °C. This is a rather nar-
304 row temperature range. In this work, we have considered ice balls at about
305 -20 °C, which is not far from the middle of the temperature range. There-
306 fore, we think that the experimental results obtained at this temperature are
307 representative of what can happen over the total range, i.e., $[-60 - 0]$ °C.

308 Concerning the target temperature, the problem is slightly different as
309 the target was at room temperature in our study. However, Render and Pan
310 [26] worked at two temperatures (room temperature and -14 °C) and they
311 showed that the target temperature does not induce significant changes in
312 the results.

313 Several other parameters such as the projectile micro-structure and the
314 target surface roughness might influence the fragmentation process and the
315 kinematics of fragments . In our case, the influence of the micro-structure
316 was not considered. However, we prepared the ice projectiles following the
317 same procedure in order to ensure the same micro-structure.

16

We think that the observed tendencies on the fragments kinematics should be also observed when changing some parameters as concerns the ice ball or the target. However, the constants determined here would not be the same. Therefore, the value of constant K_n (Eq. 6) might change if we change the ice micro-structure or the target material for example. Moreover, Eq. (3), giving the cloud center velocity, can be more generally expressed as

$$V_{C_t} \approx K_t V_{0_t}, \tag{11}$$

where K_t is a constant that possibly depends on the target roughness. We think it reasonable that for smooth surfaces, which is the case with our glass target, K_t tends toward 1. However, K_t could be significantly less than 1 for rough surfaces.

4. Conclusion

The kinematics of the ice fragmentation process was investigated experimentally. We have observed that the post-impact angle is lower than $2°$, which confirms the observations reported in the literature. On the other hand, we have observed that the ice fragments are mainly organised in a circular cloud in the target plane. The center of this cloud has the same velocity as the tangential component of the projectile velocity and the cloud radius expands with a rate proportional to the projectile normal velocity. Moreover, fragments inside the cloud have a relative velocity with respect to the cloud center which varies linearly with their distance from this center. We think that these experimental results give valuable information about the ice fragmentation process, which can help modelling and simulating hail ingestion by aircraft engines.

17

5. Acknowledgements

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428   42:177–194.

429 **6. Figures and tables**
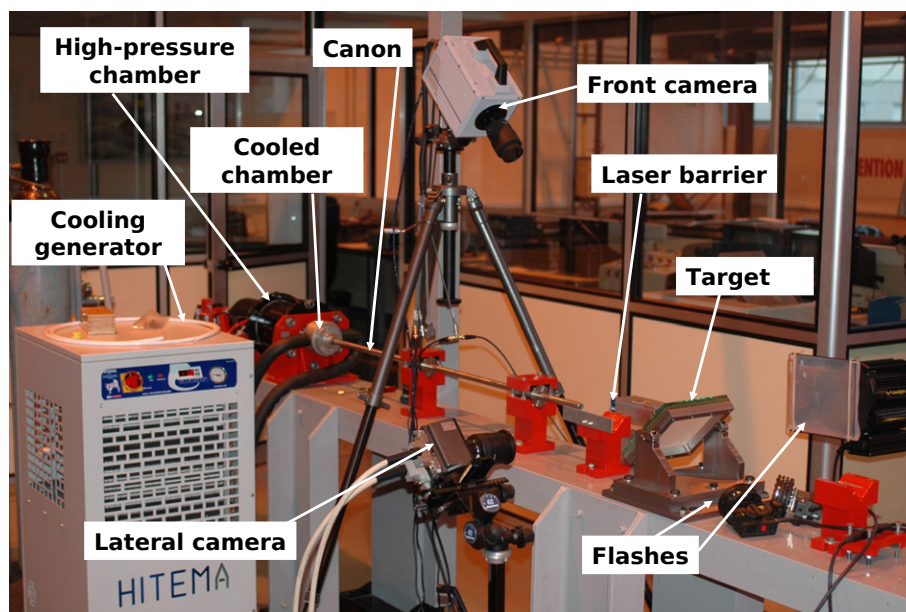
Figure 1: Overview of the experimental setup.

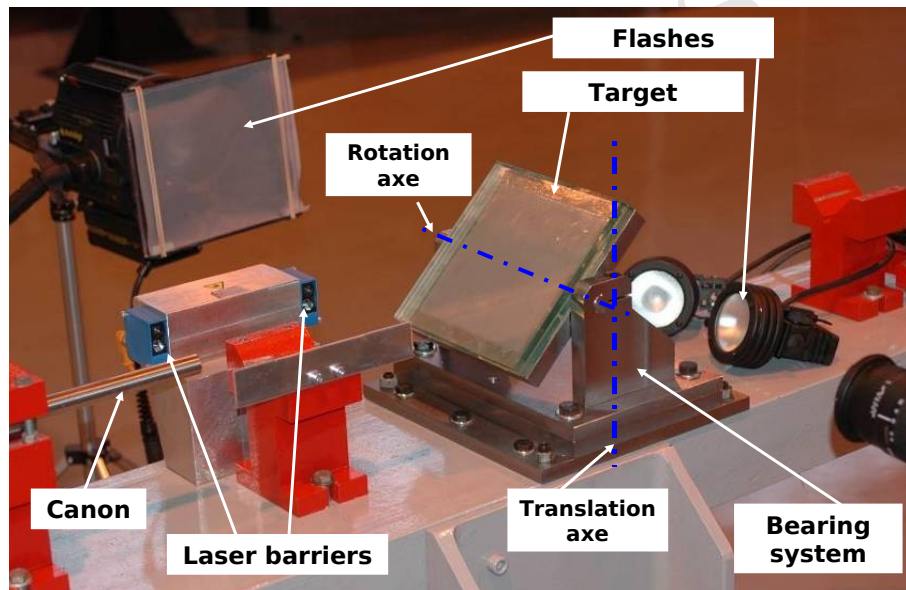Figure 2: Overview of the target and its bearing system.

Test No.	D_0 (mm)	$\alpha(°)$	V_0 (m.s^{-1})
1	6.2	20	135.8
2	6.2	20	163.2
3	6.2	20	117.1
4	6.2	45	185.7
5	6.2	45	138.0
6	6.2	45	141.9
7	6.2	67	174.9
8	6.2	67	121.9
9	6.2	67	105.2
10	6.2	90	131.0
11	6.2	90	152.7
12	6.2	90	97.2
13	6.2	90	202.8
14	12.9	20	119.9
15	12.9	20	120.8
16	12.9	20	112.1
17	12.9	20	159.4
18	12.9	20	157.9
19	12.9	45	158.0
20	12.9	45	157.9
21	12.9	45	164.8
22	12.9	45	120.3
23	12.9	45	116.4
24	12.9	67.2	118.8
25	12.9	67.2	91.1
26	12.9	67.2	130.5
27	12.9	67.2	155.2
28	12.9	90	122.2
29	12.9	90	114.7
30	12.9	90	146.5
31	12.9	90	150.4
32	27.5	20	66.8
33	27.5	20	97.4
34	27.5	20	116.4
35	27.5	45	66.1
36	27.5	45	98.2
37	27.5	45	116.2
38	27.5	67.2	60.5
39	27.5	67.2	100.0
40	27.5	67.2	116.1
41	27.5	90	66.1
42	27.5	90	97.0
43	27.5	90	113.9

Table 1: Entry parameters of the impact tests

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Test No.	D_0 (mm)	$\alpha(^\circ)$	V_0 (m.s^{-1})	K_n
14	12.9	20	119.9	0.716
15	12.9	20	120.8	0.650
17	12.9	20	159.4	0.526
18	12.9	20	157.9	0.574
19	12.9	45	158.0	0.565
20	12.9	45	157.9	0.510

Table 2: Values of K_n for some tests.
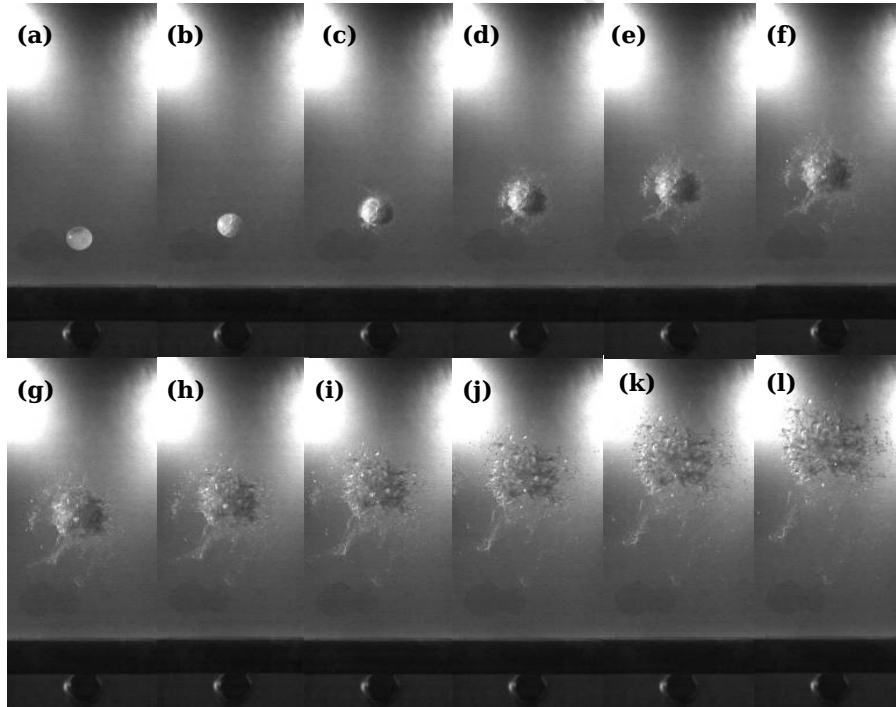
Figure 3: Image sequence of the ice ball fragmentation taken by the front camera (diameter: 12.9 mm, impact angle 20°, impact velocity: 120.8 m.s^{-1}).
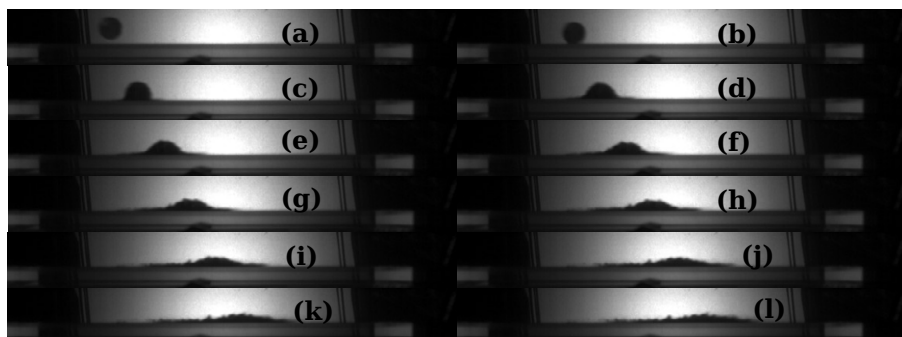
Figure 4: Image sequence of the ice ball fragmentation taken by the lateral camera (diameter: 12.9 mm, impact angle 20 °, impact velocity: 120.8 m.s^{-1}).
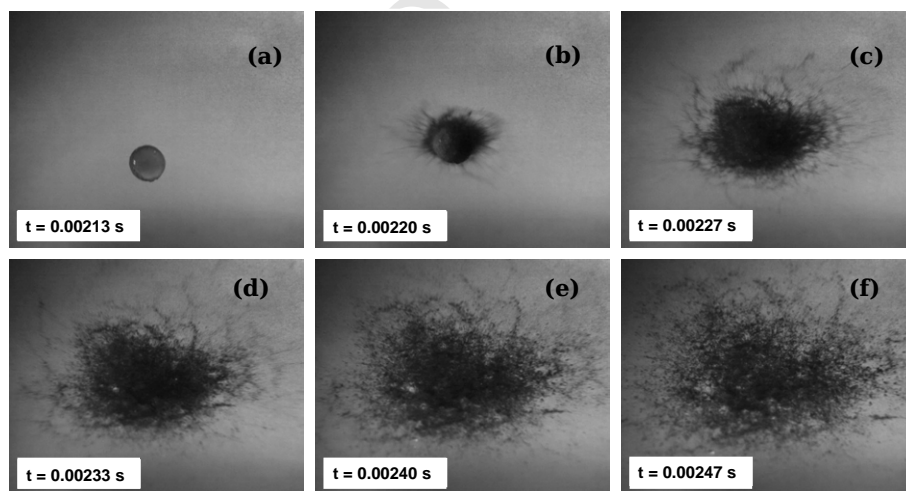
Figure 5: Front view of ice fragmentation (diameter: 12.9 mm, impact angle 90 °, impact velocity: 150.4 m.s^{-1})).

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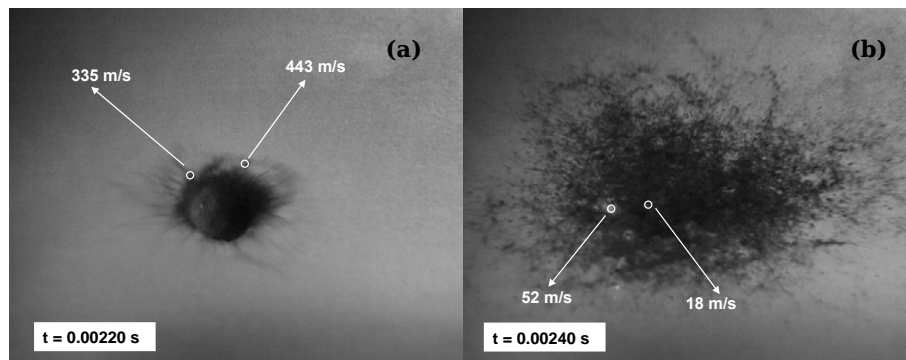
Figure 6: Magnification of the areas shown in Figure 5(b) and Figure 5(e).
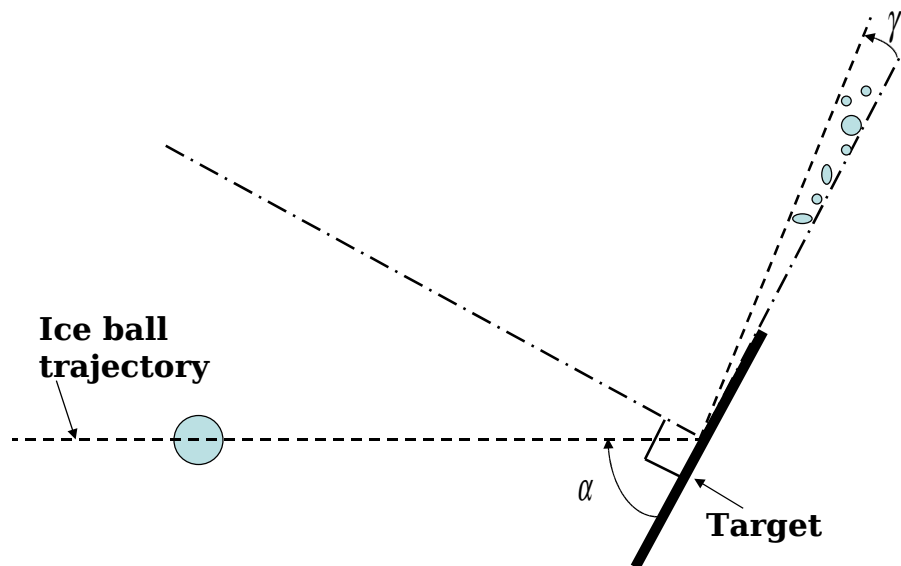
Figure 7: Definition of the post-impact angle.
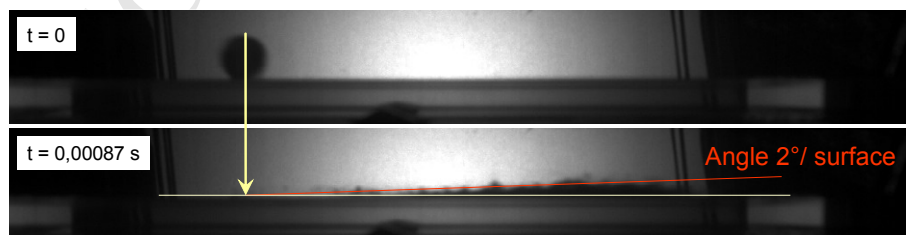
Figure 8: Example of measurement of the post-impact angle.

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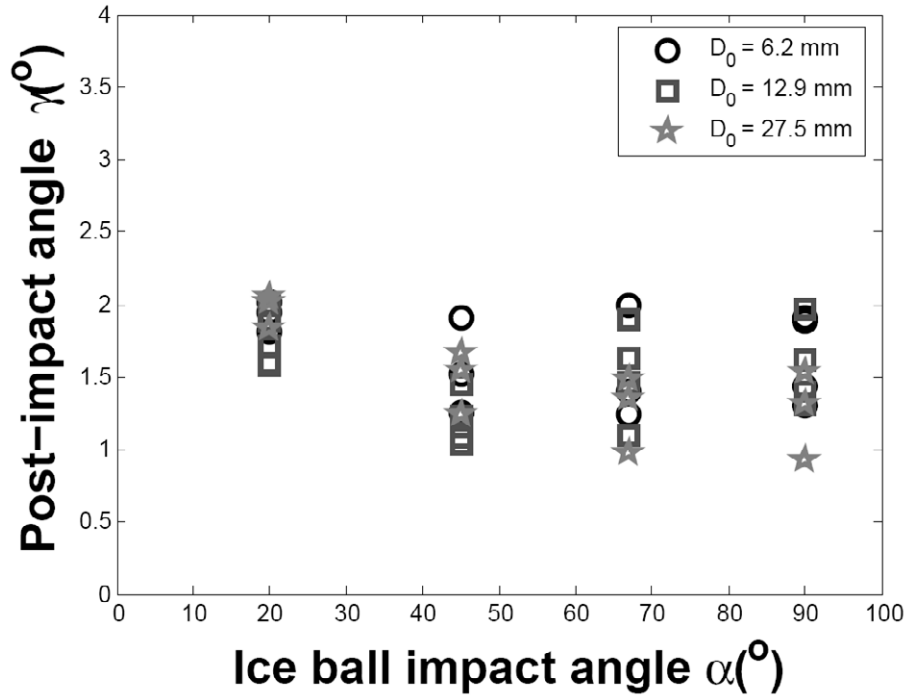
Figure 9: Values of the post-impact angle γ for the different experiments.
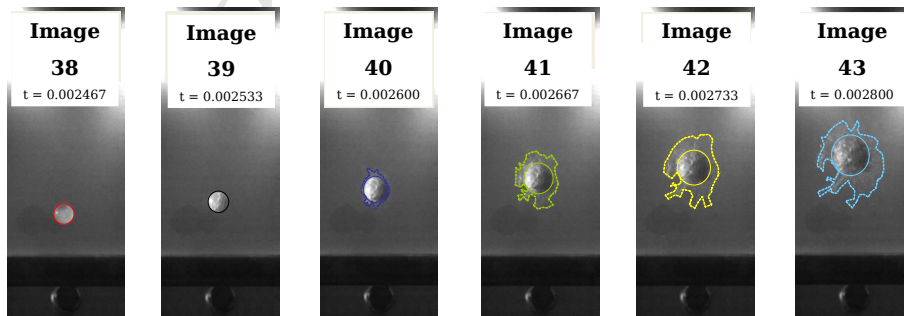
Figure 10: Analysis of images from the front camera: definition of the high density cloud contour (diameter: 12.9 mm, impact angle $20\,^\circ$, impact velocity: $120.8\,\mathrm{m.s}^{-1}$).

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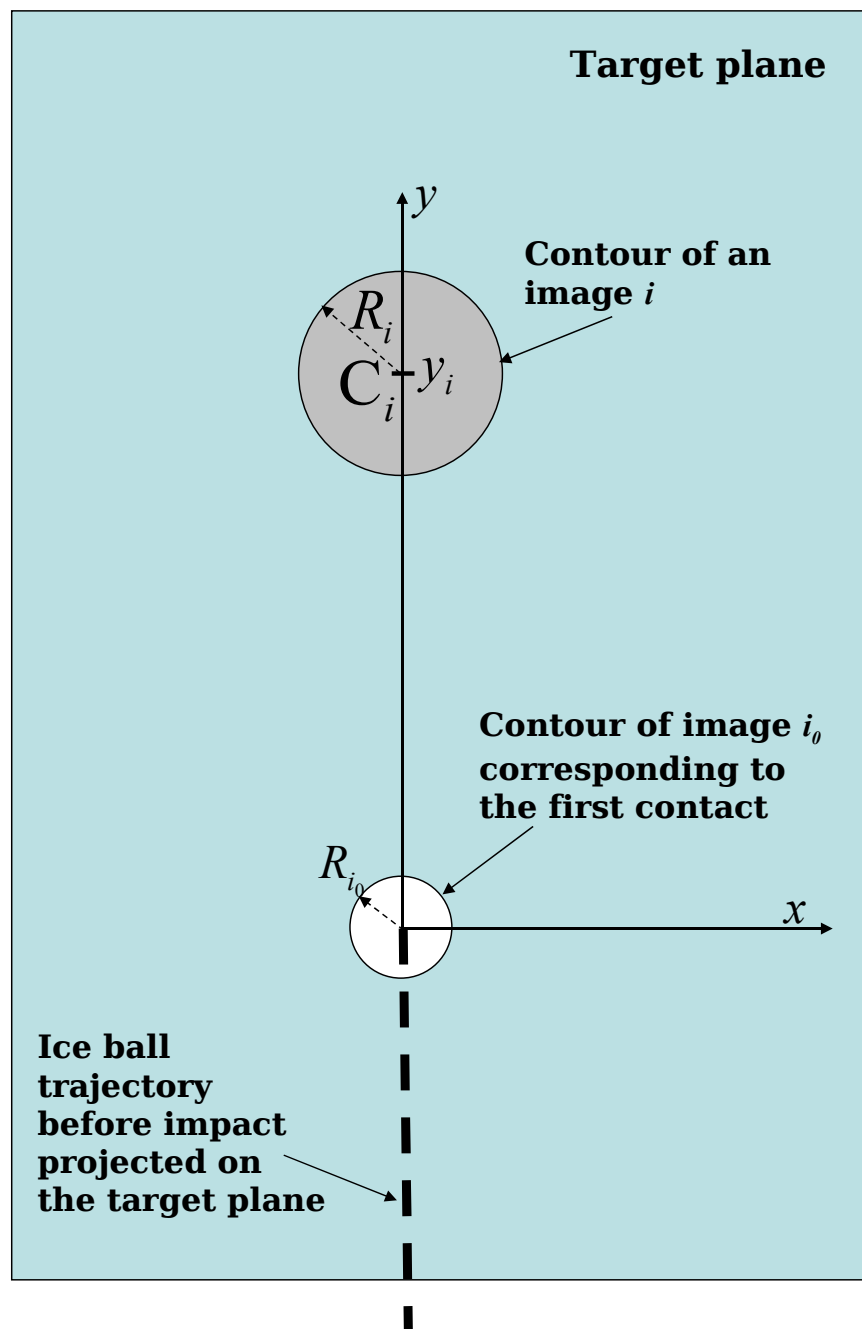
Figure 11: Shematic of the superposition of two circular contours on the image i_0 corresponding to the first (initial) contact and to a subsequent image i.
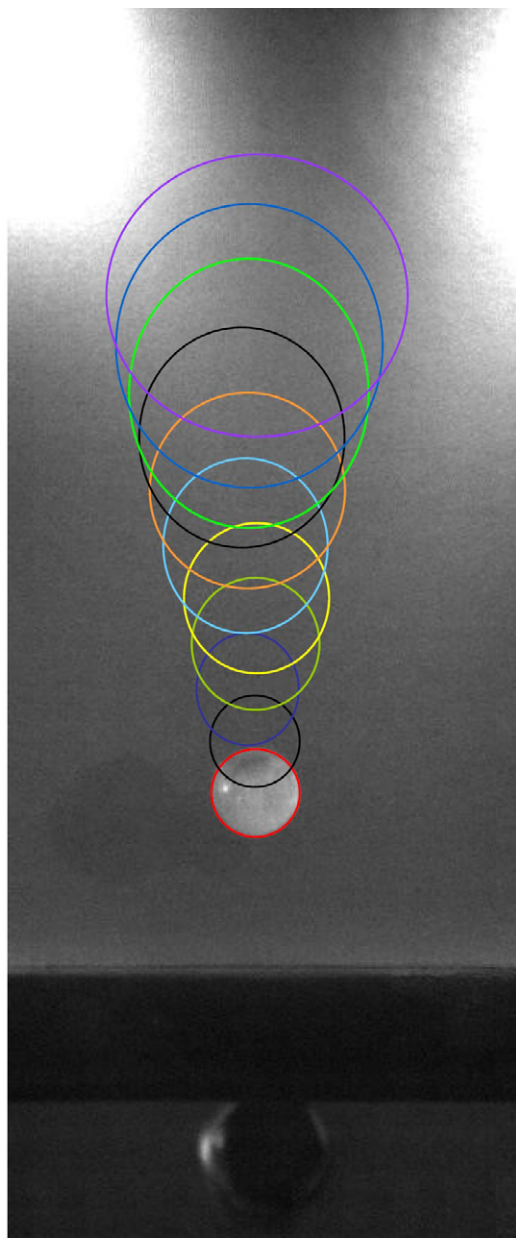
Figure 12: Superposition of circular contours: experimental result (diameter: 12.9 mm, impact angle $20\,^\circ$, impact velocity: $120.8\,\mathrm{m.s}^{-1}$).
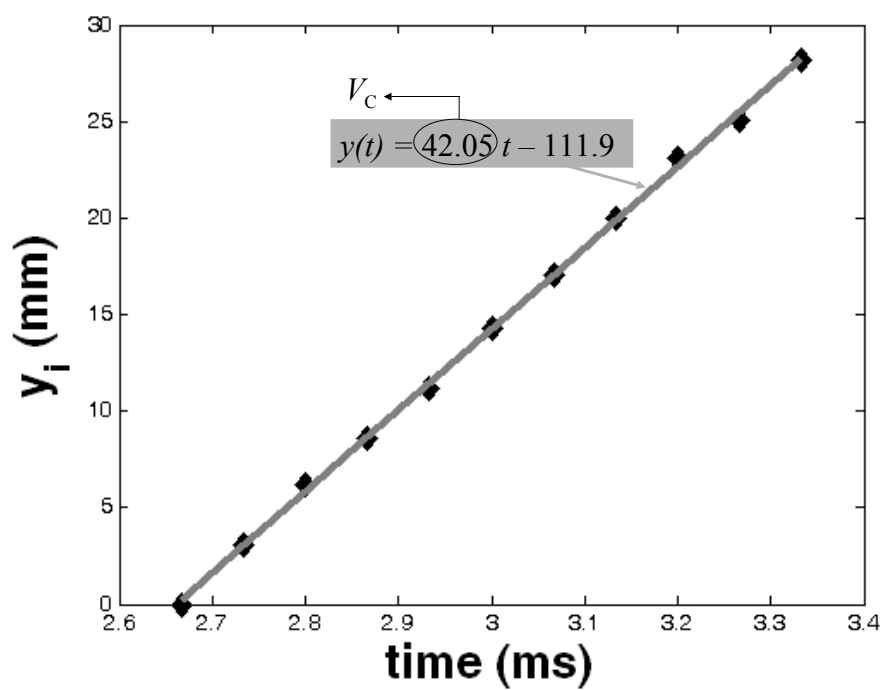
Figure 13: Evolution of the position of the cloud center with time (diameter: 12.9 mm, impact angle $67°$, impact velocity: $118.8\,\mathrm{m.s^{-1}}$).
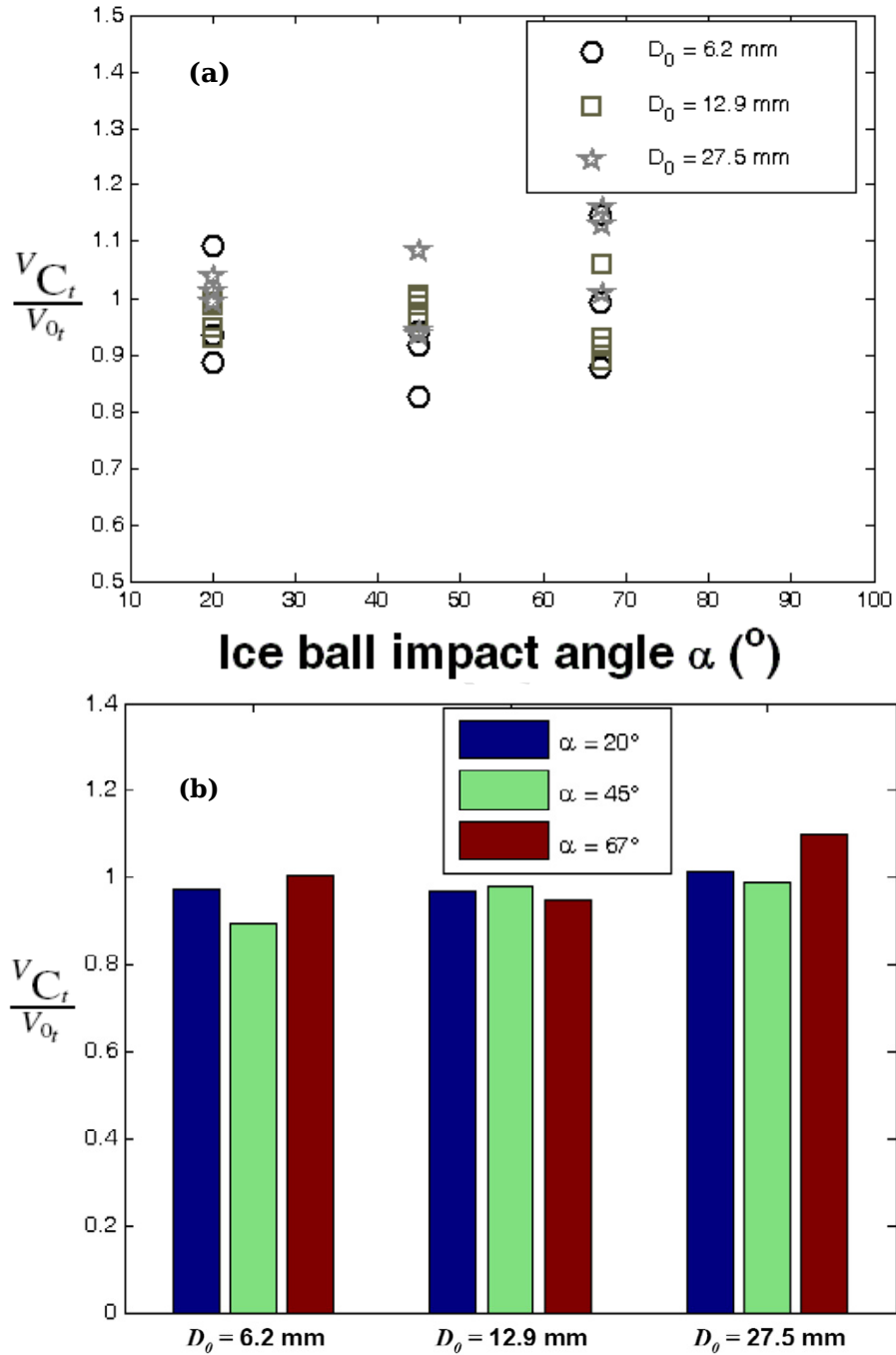
Figure 14: Values of the ratio V_{C_t}/V_{0_t}: (a) for all experiments; (b) average values for each configuration.
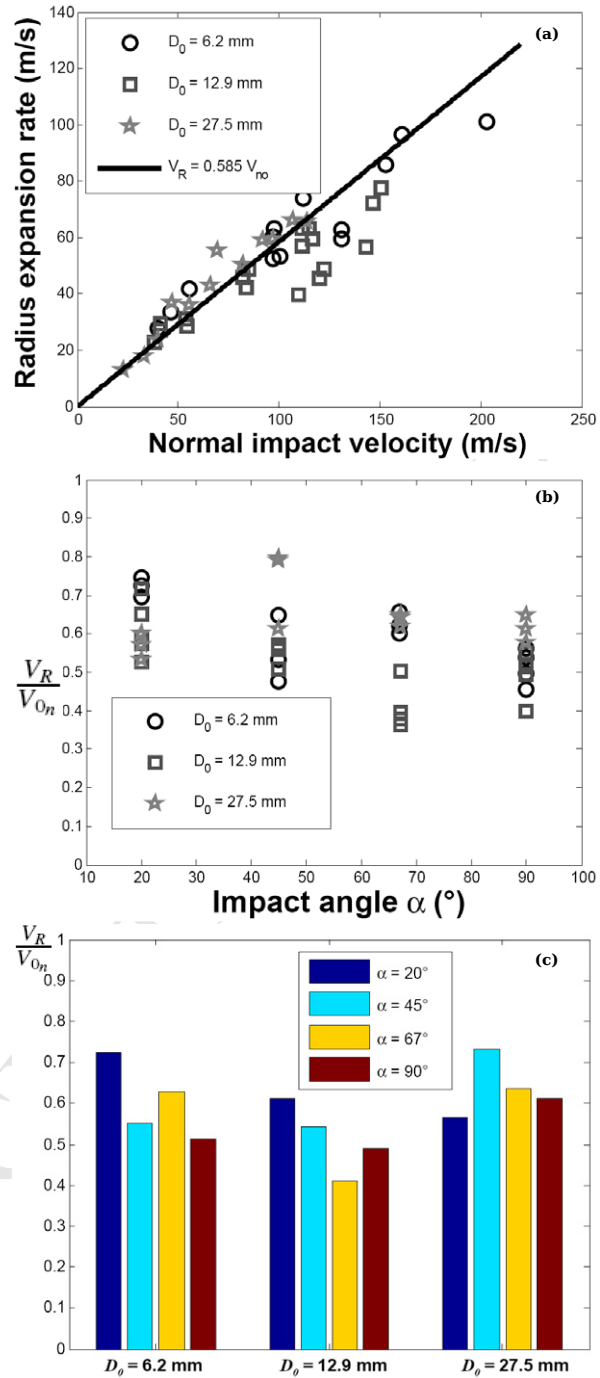
Figure 15: Radius expansion: (a) expansion rate V_R as a function of normal impact velocity V_{0_n}; (b) $K_n = V_R/V_{0_n}$ as a function of impact angle α and (c) averaged values of K_n.
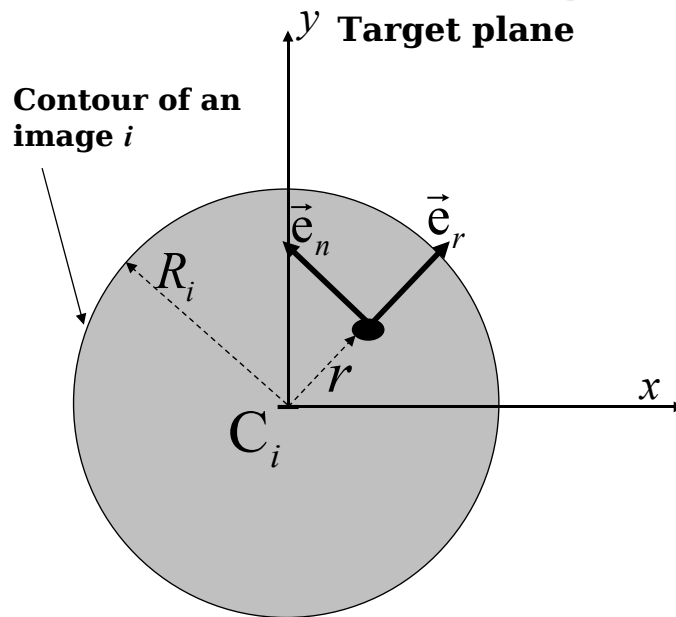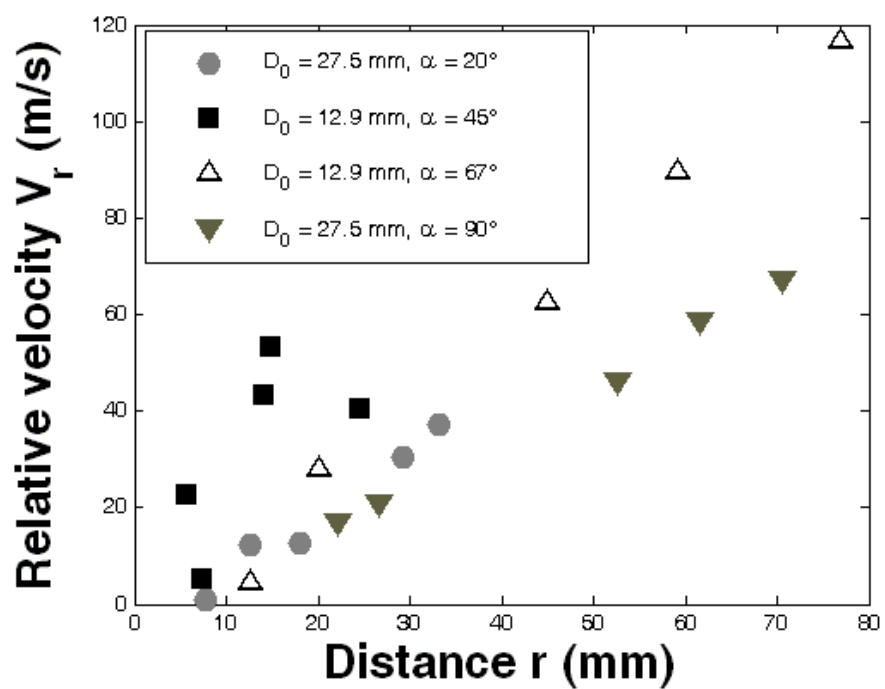
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Figure 16: Mobile reference frame.

Figure 17: Relative velocity of fragments inside the cloud as a function of their distance to the cloud center.