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# Experimental investigation of the kinematics of post-impact ice fragments

Pierrick Guégan<sup>a</sup>, Ramzi Othman<sup>a</sup>, Daniel Lebreton<sup>a</sup>, Franck Pasco<sup>a</sup>,  
Philippe Villedieu<sup>b</sup>, Jacques Meyssonier<sup>c</sup>, Sylvie Wintenberger<sup>d</sup>

<sup>a</sup>*Institut de Recherche en Génie Civil et Mécanique, Ecole Centrale de Nantes, 1 Rue de la Noe BP 92101, 44321 Nantes cedex, France.*

<sup>b</sup>*ONERA, 2 Avenue Edouard Belin, BP 4025, 31055 Toulouse, France.*

<sup>c</sup>*LGGE, UMR 5183 UJF-CNRS, 54 Rue Molière, BP 96, 38402 Saint Martin d'Hères Cedex, France.*

<sup>d</sup>*SNECMA, Site de Villaroche, Rond Point René Ravau, Réau, 77550 Moissy Cramayel, France.*

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## Abstract

Hail is more hazardous for aircraft engines compared to rain and snow, mainly, because of its solid nature and high water content. In extreme cases it can lead to **engine** flame out. In order to avoid such situations, aero engines should be designed to withstand hail ingestion. For this purpose we have studied the post-impact characteristics of ice, such as particle velocity and directions of travel. To achieve this goal, a large experimental program has been undertaken, in which spherical ice specimens were projected against a rigid plate. Three specimen diameters (**6.2, 12.9 and 27.5 mm**) and four impact angles (**20, 45, 75 and 90°**) were considered, as well as a wide range of impact velocities (**60-200 m/s**). From this experimental work, we can conclude that the ice fragments formed after impact do not bounce back and that the post-impact ice trajectory angle is lower than 2°. This is in line with observations found in the literature. On the other hand, the ice fragments are mainly organised in a circular cloud, when observed in the

target plane. The center of this cloud has the same velocity as the initial ice ball tangential impact velocity. Furthermore, the cloud radius expands with a rate proportional to the ice ball normal impact velocity. Finally, each fragment inside the cloud has a relative velocity which varies linearly with its distance from the cloud center. These experimental observations should be very helpful **in developing** models and simulations of hail ingestion by aircraft engines.

*Key words:* Hailstorm, aircraft engines, ice fragmentation, hail ingestion.

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1     Aircraft engines and structures can face several kind of impacts, mainly  
 2     during take off and landing, such as from birds [1], tyre debris [2, 3, 4],  
 3     hail [5] or sand [6, 7, 8]. These kinds of impacts can induce damage in  
 4     some parts of the aircraft structure and lead eventually to **an** aircraft crash.  
 5     Furthermore, impact-induced fragments can be ingested by aircraft engines  
 6     and cause engine power loss. Therefore, aero engines have to satisfy safety  
 7     requirements against hail ingestion to obtain certification.

8     Hail ingestion, as water ingestion [9, 10, 11], can cause engine power loss  
 9     and in extreme cases engine flame out [12, 13, 14]. In the event of ingested  
 10    hail passing through the fan into the core engine, changes in temperature,  
 11    density and specific heat ratio will occur. As a result the engine will lack  
 12    combustion efficiency or/and stability which can lead to its loss of thrust  
 13    and eventually to a flame out. Engine manufacturers and aviation regulatory  
 14    agencies have a common aim: preventing engine flame outs, roll-back or other  
 15    **operational** effects that could result in an in-flight shut down when flying  
 16    by stormy weather. These considerations are a major concern for the design,  
 17    the certification and the **operation** of modern high bypass aircraft engines.

18 The engine manufacturers need powerful numerical tools to take into ac-  
 19 count inclement weather threats in the engine design. A first model was  
 20 proposed by [15]. However, to be realistic, these simulations must rely on  
 21 experimental investigations of hail impact in order to understand the phe-  
 22 nomenon, develop appropriate constitutive equations and identify the rele-  
 23 vant parameters.

24 A number of references can be found in the literature about ice-impact  
 25 induced damage e.g. [5, 16, 17], fatigue induced damage e.g. [16] or high  
 26 velocity ice impacts on composite panels e.g. [5, 17]. On the other hand, low  
 27 velocity ice impacts, i.e., for velocities lower than  $10 \text{ m.s}^{-1}$ , were investigated  
 28 in Refs. [18, 19, 20, 21, 22, 23]. These latter works aimed at defining the  
 29 transition between the rebound and fragmentation regimes.

30 In order to understand hail ingestion by aircraft engines, Pan, Render  
 31 et al. [24, 25, 27, 26, 28, 29, 30] have studied the impact fragmentation  
 32 characteristics of hailstones on rigid targets. They were interested in post-  
 33 impact particle size, particle velocity and travel directions. In Ref. [26]  
 34 a patternator technique was used. A still photography technique was also  
 35 employed in Ref. [25, 28]. These investigations were extended to study the  
 36 effect of target curvature [29] and target rotation [27]. Subsequently, the  
 37 methodology was applied to consider the hail mass distribution behind a  
 38 rotating fan and spinner assembly [30]. Pan, Render et al. [24, 25, 27, 26,  
 39 28, 29, 30] have shown that the Rosin-Rammler [31] distribution, also called  
 40 Weibull distribution [32], which is also used in rock fragmentation [33], is  
 41 well suited to represent the particle size and particle velocity distributions.

42 In the present paper, we investigate the kinematics of post-impact ice

43 fragments. Render and Pan [26] have observed that the fragments do not  
 44 bounce back, i.e., the post-impact angles are very low with the fragments  
 45 trajectories remaining almost in the plane of the target. This result was ob-  
 46 tained by a patternator technique. One aim of the present work is to check  
 47 this result against high speed photography observations. On the other hand,  
 48 the fact that the particle velocity distribution was found by [28] to be of the  
 49 Rosin-Rammler [31] type is not enough to describe the fragments movement  
 50 completely. Indeed, the distribution only indicates what percentage of mass  
 51 fragments moves at a given velocity. In the present work, we are also inter-  
 52 ested in obtaining a better description of the kinematics of fragments in the  
 53 plane of the target. To this aim a wide experimental program was carried  
 54 out to assess the influences of the ice projectile diameter, impact angle and  
 55 impact velocity.

## 56 **1. Method**

### 57 *1.1. Ice preparation*

58 The ice spherical specimens were prepared using moulds made of two  
 59 hollow half-spheres in order to facilitate the removal of the ice ball. The  
 60 moulds were filled with cooled water (between 4 and 8 °C), then placed in a  
 61 freezer (between -25 and -18 °C). Subsequently, the ice balls were removed  
 62 from the moulds after 24 hours and kept in the freezer until the beginning of  
 63 the test.

64 Hailstones can have diameters ranging from 2 to 100 mm [26]. In the  
 65 present study, three diameters ( $D_0$ ) were considered: 6.2, 12.9 and 27.5 mm  
 66 corresponding to a mass of approximately 0.1, 1 and 10 g, respectively. It

must be noted that since the tests were performed at room temperature melting can induce important dimension inaccuracies in the ice of which diameter is lower than 6 mm. On the other hand, it is hard to analyse images of projectiles with large diameters because of the huge number of fragments formed. Note that Pan and Render [26, 28] worked mainly with ice balls 12.7 mm in diameter.

### 1.2. Experimental set-up

The experimental set-up consists of a gas gun, that comprises a high pressure chamber, a cooled chamber and a tube, a target plate and the instrumentation (Figure 1). The high pressure chamber works with nitrogen gas and can go up to 85 bar (8.5 MPa). The actual applied pressures were between 0.6 and 3.9 bar (0.06 and 0.39 MPa), which results in impact velocities ( $V_0$ ) ranging between 60 and 200 m.s<sup>-1</sup>. This velocity range corresponds to that of a low engine power regime, such as during aircraft descent. It is in this regime that inclement weather effects are mostly encountered [14]. It is worth noticing that Pan and Render [26, 28] were limited to impact velocities in the range 100 to 175 m.s<sup>-1</sup>.

The cooled chamber is made out of stainless steel. It is located between the high pressure chamber and the gas tube. Its role is to house the ice projectile while preventing it from melting before the strike. It is cooled by the flow of an antifreeze - water mixture maintained at a temperature between -4 and 0 °C by a cooling generator. The cooled chamber is connected to the canon made out of a stainless **steel** tube. The cooled chamber inner dimensions and the tube inner diameter are adapted to the ice ball diameters, so that no sabot is needed. Since we performed our tests with three ice ball

92 diameters, three different cooled chambers and three different tubes were  
93 used.

94 The target is a glass plate ( $200 \times 200 \text{ mm}^2$ ) 20 mm in thickness. The  
95 choice of glass as the target material was motivated by two reasons. Firstly,  
96 glass has a stiffness comparable to some metallic material such as aluminium  
97 (its Young's modulus is about 64 GPa). Secondly, since glass is transparent,  
98 it is possible to use back-lighting which yields more homogeneous light on  
99 the target than front-lighting.

100 This target is mounted on a bearing system allowing two degrees of free-  
101 dom (Figure 2): a translation along the vertical direction and a rotation  
102 around the horizontal axis perpendicular to the (horizontal) gas tube thus  
103 allowing changes in the impact angle  $\alpha$ . The impact angle is defined as the  
104 angle between the projectile trajectory before impact and the normal to the  
105 plane of the target. In the present work, four values of the impact angle  $\alpha$   
106 were considered: 20, 45, 67 and 90°.

107 The experimental set-up is instrumented by two laser barriers to measure  
108 the ice ball impact velocity. Furthermore, two high speed video cameras were  
109 used in order to record the post-impact characteristics. The first camera was  
110 used to film the lateral side, i.e., the plane perpendicular to the target that  
111 contains the gas-tube axis, whereas the second camera was used to film the  
112 front side, i.e., the target surface. The two cameras were synchronised. The  
113 trigger was given by the first laser barrier. The adopted acquisition frequency  
114 of the cameras was 15000 frames per second.

115 The initial conditions of all our tests are given in Table 1.



## 2. Results

### 2.1. General observations

Figures 3 and 4 show the sequences of images for an ice ball fragmentation obtained with the front camera and the lateral camera, respectively. In this test, the ice ball diameter, impact angle and impact velocity were 12.9 mm,  $20^\circ$  and  $120.8 \text{ m.s}^{-1}$ , respectively (Test No. 15). Both cameras were synchronised and took 15000 frame per second, so that, for example, photographs (a) on Figures 3 and 4 were taken at the same time. The resolutions of the front and lateral camera were  $320 \times 832 \text{ pix}^2$  and  $1024 \times 128 \text{ pix}^2$ , respectively. Figures 3(a) and 4(a) show the ice specimen just before impact. The first contact between the ice ball and the plate is shown in Figures 3(b) and 4(b). Fragmentation begins at Figures 3(c) and 4(c).

From Figures 3 and 4, we can divide the fragmentation process into two stages. In a first stage, only the front part of the ice ball fragments (Figure 3(c) to (e) and Figure 4(c) to (e)). The created fragments are ejected first and form a low density chaotic cloud. In a second stage, the remaining ice projectile collapses (Figure 3(f) to (l) and Figure 4(f) to (l)) yielding a high density, almost circular, cloud. We can also notice from Figure 4 that the post-impact angle, which we define as the angle between the fragments trajectories and the plane of the target, is very low. Fragments trajectories lie almost in the target plane.

The above observations are also confirmed in Figure 5 where photos (a) to (f) show images of ice fragmentation for a normal impact at a velocity of  $150.4 \text{ m.s}^{-1}$ . The two photos in Figure 6, which are magnifications of Figures 5(b) and 5(e), show that the first ejected particles which make the low density

cloud have a high velocity, higher than the initial ice ball impact velocity. This is consistent with results of Render and Pan [26] who observed also that particles formed immediately after impact moves greatly faster than the approaching ice ball velocity. Figure 6(a) shows two particles with a post-impact velocities of  $335 \text{ m.s}^{-1}$  and  $443 \text{ m.s}^{-1}$ , respectively, to be compared to the ice ball impact velocity of  $150.4 \text{ m.s}^{-1}$ . On the other hand, the fragments created during the second stage, that form the high density cloud, have lower velocities. As an illustration Figure 6(b) shows two particles, one with a post-impact velocity of  $52 \text{ m.s}^{-1}$  and the other with a post-impact velocity of  $18 \text{ m.s}^{-1}$ . The difference in the post-impact velocities can be explained as follows. At the beginning of the impact, the ice ball has a high initial velocity and elastic energy is stored in the yet unfragmented material. Only a small ratio of the mass is fragmented. Particles are ejected with a high velocity, which can be even higher than the nominal impact-velocity owing to the energy release. Furthermore, as the first stage cloud has a low density of fragments, the likelihood that collisions occur is low. As the second stage of the fragmentation process begins, the remaining unfragmented ice mass has already lost a part of its kinetic energy during the first stage. In addition, the second stage cloud of fragments has a higher density. Hence, multiple collisions happen that consume part of the kinetic energy of the particles.

It is worth to point out here that the post-impact particles velocities were determined by measuring the particles coordinates (in pixels) in several consecutive images. **This particle tracking is processed manually. In order to have a full comprehension of the post-impact fragment**

kinematics, the tracking is operated on the center of the high density cloud and its circular counter and also on three to seven particles inside the cloud. The velocities were first determined in  $\text{pix.s}^{-1}$  then converted to  $\text{m.s}^{-1}$  from the calibration coefficient determined from a calibration image which corresponds to a millimetric grid. Considering the camera settings, the accuracy of displacement measurements are at best 0.01 pix and at worst 0.1 pix.

## 2.2. Kinematics in the normal plane

In this section, we are mainly interested in the fragments post-impact angle. The ice ball impact angle  $\alpha$  and fragments post-impact angle  $\gamma$  are represented in Figure 7. The angle  $\gamma$  is defined as the maximum angle between the fragments trajectories and the plane of the target. This angle is determined from the lateral camera images as shown in Figure 8. Figure 9 shows the different values of post-impact angles obtained for all the experiments. It seems that the post-impact angle  $\gamma$  is independent of the ice ball impact velocity  $V_0$ , of the impact angle  $\alpha$  and of the projectile diameter  $D_0$ . All the measured values of  $\gamma$  lie in the range between 1 and  $2^\circ$ . Therefore, the normal velocity of fragments can be considered as negligible with respect to their tangential velocity, that is, the fragments kinematics is governed by their sole tangential velocity which is in line with the findings of Render & Pan [26] who observed a low bounce normal to the surface of the ice ball fragments.

188 *2.3. Kinematics in the target plane*

189 As explained in section 2.1, the fragmentation process is divided into two  
 190 stages. The first stage yields a low density, high velocity, chaotic cloud of  
 191 fragments. The second stage is characterized by a high density, low velocity,  
 192 circular cloud. In this section, we focus on the kinematics of the fragments  
 193 created during the second stage. Mainly, we are interested in the velocity  
 194 of the cloud center, the rate of surface expansion of the cloud and in the  
 195 velocities of fragments relative to a (moving) reference point in the moving  
 196 cloud. To this aim we analyse the images obtained by using the front camera  
 197 (Figure 10). For each image  $i$ , the contour of the second-stage cloud of frag-  
 198 ments is approximated by a circle of center  $C_i$  and radius  $R_i$ . By analysing  
 199 the sequence of images for each test, we can obtain the evolution with time  
 200 of the cloud center and radius.

201 Figure 11 shows the schematic of the superposition of two cloud contours,  
 202 one corresponding to the instant when the specimen hits the target (“first-  
 203 contact” time; corresponding image  $i_0$ ), the other at a post-impact time  
 204 (image  $i$ ). Note that the “initial” cloud, represented as a circle, should in  
 205 fact be the elliptical projection of the undamaged spherical specimen onto  
 206 the target plane (along its trajectory before impact). The  $y$  axis corresponds  
 207 to the projection of the trajectory of the ice ball before impact on the plane  
 208 of the target. The origin of coordinates is taken at the center  $C_{i_0}$  of the  
 209 initial cloud, so that  $x_{i_0} = y_{i_0} = 0$ . Our analyse of the different tests has  
 210 shown that the contours centers remain aligned with the  $y$  axis (i.e., in the  
 211 same vertical plane that contains the ice ball trajectory before impact) (see  
 212 Figure 12). In other words, for any image  $i$ , the center of the fragments cloud

is such that  $x_i = 0$ , and is thus defined only by  $y_i$ .

### 2.3.1. Velocity of the cloud center

Figure 13 is a plot of the positions of the centers  $C_i$ , defined by the coordinate  $y_i$ , as function of time. This figure corresponds to a test performed with an ice ball of diameter  $D_0 = 12.9$  mm, an impact angle  $\alpha = 67.2^\circ$  and an impact velocity  $V_0 = 118.8 \text{ m.s}^{-1}$  (Test No. 24). The variation of  $y_i$  versus time is almost linear and can be approximated by

$$y(t) = at + b, \quad (1)$$

where  $a$  and  $b$  are two constants. In the case shown in Figure 13,  $a = 42.05 \text{ m.s}^{-1}$  and  $b = -111.9 \text{ mm}$ .  $a$  is the velocity of the cloud center in the target plane, i.e., the tangential velocity of the cloud center, which is denoted by  $V_{C_t}$  in the following. It is worth to compare  $V_{C_t}$  to the ice ball tangential impact velocity  $V_{0_t}$  which is such as

$$V_{0_t} = V_0 \cos \alpha, \quad (2)$$

where  $V_0$  and  $\alpha$  are the ice ball impact velocity and impact angle, respectively. In the case shown in Figure 13, the impact velocity  $V_0$  is  $118.8 \text{ m.s}^{-1}$  and the tangential impact velocity  $V_{0_t}$  equals  $46.04 \text{ m.s}^{-1}$ . Hence, the ratio  $V_{C_t}/V_{0_t}$  is equal to 0.91. This means that the tangential velocity of the cloud center is of the same order of magnitude as the normal impact velocity. In order to confirm this conclusion, we have plotted the ratio  $V_{C_t}/V_{0_t}$  for all the experiments in Figure 14(a). The values of the ratio  $V_{C_t}/V_{0_t}$  are mostly between 0.9 and 1.1. In order to reduce the effect of experimental noise we have computed the average value for each experimental configuration (see

Figure 14(b)). This figure shows that  $V_{C_t}/V_{0_t}$  is independent of the diameter  $D_0$  and of the impact angle  $\alpha$ . Nevertheless, more scatter is obtained with results of  $D_0 = 6.2$  mm than those of  $D_0 = 12.9$  and  $27.5$  mm. Indeed, the geometry of low-diameter ice balls is more sensitive to melting than the geometry of balls with important diameters. More precisely, melting can affect the spheric and smooth shape of ice balls with low diameter. As the scatter in results is lower than 10%, we conclude that melting has an insignificant effect on  $V_{C_t}/V_{0_t}$  ratio. Namely, the average ratio  $V_{C_t}/V_{0_t}$  is between 0.96 and 1.01, except for two configurations, ( $D_0 = 6.2$  mm;  $\alpha = 45^\circ$ ) and ( $D_0 = 27.5$  mm;  $\alpha = 67^\circ$ ). As the average value for all experiments is 0.986, we can state that the tangential velocity of the cloud center  $V_{C_t}$  is almost equal to the ice ball tangential impact velocity  $V_{0_t}$ . Then we can assume that

$$V_{C_t} \approx V_{0_t}. \quad (3)$$

### 2.3.2. Surface expansion rate of the fragments cloud

The surface expansion rate of the cloud of fragments is characterised by the rate of expansion of the cloud radius. For each test we study the variation of the circular cloud radius versus time. This variation is almost linear and can be approximated by

$$R(t) = V_R t + R_0, \quad (4)$$

where  $R_0 = D_0/2$  is the initial radius of the ice ball and  $V_R = \dot{R}$  is the radius expansion rate.

Figure 15(a) shows the values of  $V_R$  for the all our experiments as a function of the ice ball normal impact velocity  $V_{0_n}$ .  $V_{0_n}$  is the component of

the ice ball impact velocity normal to the target plane:

$$V_{0_n} = V_0 \sin(\alpha). \quad (5)$$

Figure 15(a) shows that the radius expansion rate increases with increasing normal impact velocity. Quantitatively, the radius expansion rate can be approximated by the linear relation

$$V_R \approx K_n V_{0_n}, \quad (6)$$

where  $K_n$  is a constant. From Figure 15(a),  $K_n \approx 0.585$ .

The ratios  $K_n = V_R/V_{0_n}$  for the different experiments are plotted in Figure 15(b). It seems that  $K_n$  is independent of the test configuration: there is no special tendency neither as a function of the ice ball diameter nor of the impact angle. This is also confirmed by Figure 15(c) where the average value for each configuration is represented. **Moreover, we compare the values of  $K_n$  obtained with tests of similar conditions in Table 2. More precisely, we compare  $K_n$  of Tests 14 and 15 ( $D_0 = 12.9$  mm,  $\alpha = 20^\circ$ ,  $V_0 \approx 120$  m/s), 17 and 18 ( $D_0 = 12.9$  mm,  $\alpha = 20^\circ$ ,  $V_0 \approx 158.5$  m/s) and 19 and 20 ( $D_0 = 12.9$  mm,  $\alpha = 45^\circ$ ,  $V_0 \approx 158$  m/s). It comes that the difference of the two values of  $K_n$  obtained by two tests in the same configuration is around 10%. Hence, we can confirm that  $K_n$  is independent of the test configuration and that the scatter observed in the values of  $K_n$  are rather due to experimental inaccuracies and not to a more complex behaviour of the fragmentation process.**

277 *2.3.3. Fragments relative velocities*

278 In sections 2.3.1 and 2.3.2, we have derived relations for the fragments  
 279 center velocity and fragments cloud radius rate, respectively. In other words  
 280 we can describe the velocity of the fragments near the center of the cloud and  
 281 at its boundary. In the section, we are interested in the velocity of fragments  
 282 inside the cloud. This velocity is studied in the mobile reference frame, i.e.,  
 283 we measure the velocity relative to the center of the cloud (Figure 16).

284 First, it is observed that the relative velocity of fragments  $V_r$  at a distance  
 285  $r$  from the center is dependent only on this distance:

$$\vec{V}_r \equiv \vec{V}_r(r). \quad (7)$$

286 Second, the relative velocity is along the radius line, i.e.,

$$\vec{V}_r(r) = V_r(r)\vec{e}_r, \quad (8)$$

287 where  $\vec{e}_r$  is defined in Figure 16.

288 The variation of  $V_r$  in terms of the distance  $r$  is represented in Figure  
 289 17 for different experimental configurations. It comes from this figure that  
 290 the velocity  $V_r$  increases with  $r$ , which explains the cloud expansion. The  
 291 fragments near the center move slower than that at the boundary. Moreover,  
 292 the variation of  $V_r$  versus  $r$  can also be approximated by a linear relation

$$\vec{V}_r(r) \approx \nu r, \quad (9)$$

293 where  $\nu$  is a constant which can be determined by considering the radius  
 294 expansion rate  $V_R$ . Indeed, the velocity of fragments at the boundaries  $V_r(R)$   
 295 should be equal to radius expansion rate  $V_R$ , so that the relative velocity of



296 a fragment inside the cloud can be expressed as

$$\vec{V}_r(r) \approx V_R \frac{r}{R}. \quad (10)$$

### 297 3. Discussion

298 Despite the large experimental program undertaken, several possibly in-  
299 fluencing parameters were neglected in this study, for example the tempera-  
300 tures of the ice ball and of the target.

301 Temperature influence several mechanical properties. Hence, there is no  
302 reason to neglect its effect on the ice fragmentation process. However, natural  
303 hail temperatures are generally between  $-60^\circ\text{C}$  and  $0^\circ\text{C}$ . This is a rather nar-  
304 row temperature range. In this work, we have considered ice balls at about  
305  $-20^\circ\text{C}$ , which is not far from the middle of the temperature range. There-  
306 fore, we think that the experimental results obtained at this temperature are  
307 representative of what can happen over the total range, i.e.,  $[-60 - 0]^\circ\text{C}$ .

308 Concerning the target temperature, the problem is slightly different as  
309 the target was at room temperature in our study. However, Render and Pan  
310 [26] worked at two temperatures (room temperature and  $-14^\circ\text{C}$ ) and they  
311 showed that the target temperature does not induce significant changes in  
312 the results.

313 Several other parameters such as the projectile micro-structure and the  
314 target surface roughness might influence the fragmentation process and the  
315 kinematics of fragments . In our case, the influence of the micro-structure  
316 was not considered. However, we prepared the ice projectiles following the  
317 same procedure in order to ensure the same micro-structure.

318 We think that the observed tendencies on the fragments kinematics should  
 319 be also observed when changing some parameters as concerns the ice ball or  
 320 the target. However, the constants determined here would not be the same.  
 321 Therefore, the value of constant  $K_n$  (Eq. 6) might change if we change the  
 322 ice micro-structure or the target material for example. Moreover, Eq. (3),  
 323 giving the cloud center velocity, can be more generally expressed as

$$V_{C_t} \approx K_t V_{0_t}, \quad (11)$$

324 where  $K_t$  is a constant that possibly depends on the target roughness. We  
 325 think it reasonable that for smooth surfaces, which is the case with our glass  
 326 target,  $K_t$  tends toward 1. However,  $K_t$  could be significantly less than 1 for  
 327 rough surfaces.

#### 328 4. Conclusion

329 **The kinematics of the ice fragmentation process was investi-**  
 330 **gated experimentally.** We have observed that the post-impact angle is  
 331 lower than  $2^\circ$ , which confirms the observations reported in the literature.  
 332 On the other hand, we have observed that the ice fragments are mainly or-  
 333 ganised in a circular cloud in the target plane. The center of this cloud has  
 334 the same velocity as the tangential component of the projectile velocity and  
 335 the cloud radius expands with a rate proportional to the projectile normal  
 336 velocity. Moreover, fragments inside the cloud have a relative velocity with  
 337 respect to the cloud center which varies linearly with their distance from  
 338 this center. We think that these experimental results give valuable infor-  
 339 mation about the ice fragmentation process, which can help modelling and  
 340 simulating hail ingestion by aircraft engines.

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428 42:177–194.

429 **6. Figures and tables**



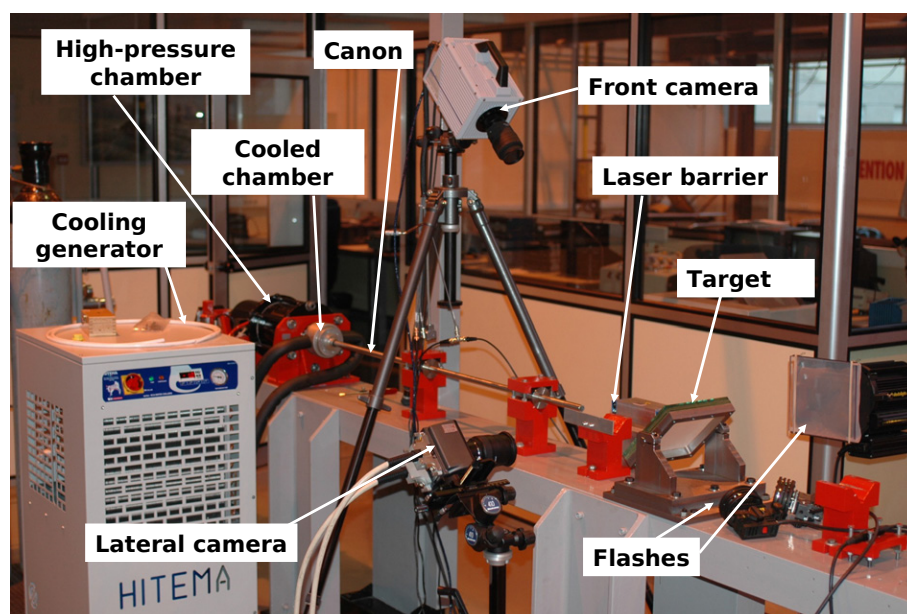


Figure 1: Overview of the experimental setup.

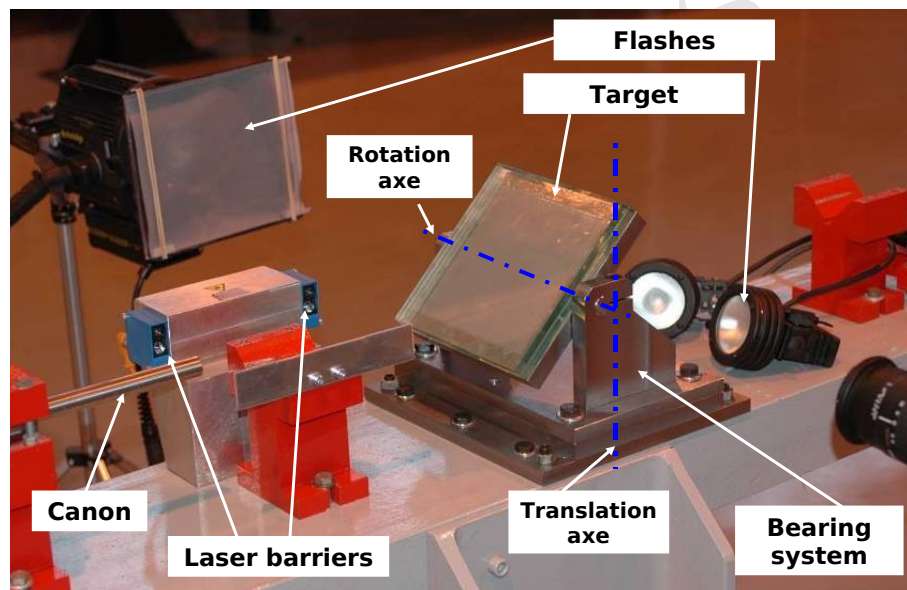
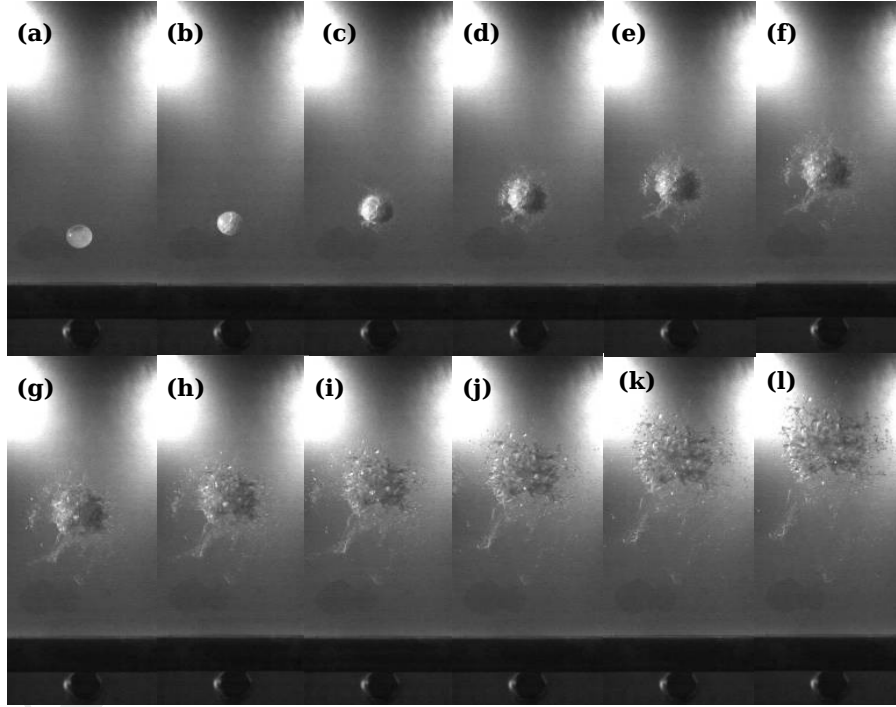


Figure 2: Overview of the target and its bearing system.

Test No.	$D_0$ (mm)	$\alpha(^{\circ})$	$V_0$ (m.s <sup>-1</sup> )
1	6.2	20	135.8
2	6.2	20	163.2
3	6.2	20	117.1
4	6.2	45	185.7
5	6.2	45	138.0
6	6.2	45	141.9
7	6.2	67	174.9
8	6.2	67	121.9
9	6.2	67	105.2
10	6.2	90	131.0
11	6.2	90	152.7
12	6.2	90	97.2
13	6.2	90	202.8
14	12.9	20	119.9
15	12.9	20	120.8
16	12.9	20	112.1
17	12.9	20	159.4
18	12.9	20	157.9
19	12.9	45	158.0
20	12.9	45	157.9
21	12.9	45	164.8
22	12.9	45	120.3
23	12.9	45	116.4
24	12.9	67.2	118.8
25	12.9	67.2	91.1
26	12.9	67.2	130.5
27	12.9	67.2	155.2
28	12.9	90	122.2
29	12.9	90	114.7
30	12.9	90	146.5
31	12.9	90	150.4
32	27.5	20	66.8
33	27.5	20	97.4
34	27.5	20	116.4
35	27.5	45	66.1
36	27.5	45	98.2
37	27.5	45	116.2
38	27.5	67.2	60.5
39	27.5	67.2	100.0
40	27.5	67.2	116.1
41	27.5	90	66.1
42	27.5	90	97.0
43	27.5	90	113.9

Table 1: Entry parameters of the impact tests

Test No.	$D_0$ (mm)	$\alpha(^{\circ})$	$V_0$ (m.s <sup>-1</sup> )	$K_n$
14	12.9	20	119.9	0.716
15	12.9	20	120.8	0.650
17	12.9	20	159.4	0.526
18	12.9	20	157.9	0.574
19	12.9	45	158.0	0.565
20	12.9	45	157.9	0.510

Table 2: Values of  $K_n$  for some tests.Figure 3: Image sequence of the ice ball fragmentation taken by the front camera (diameter: 12.9 mm, impact angle  $20^{\circ}$ , impact velocity:  $120.8 \text{ m.s}^{-1}$ ).

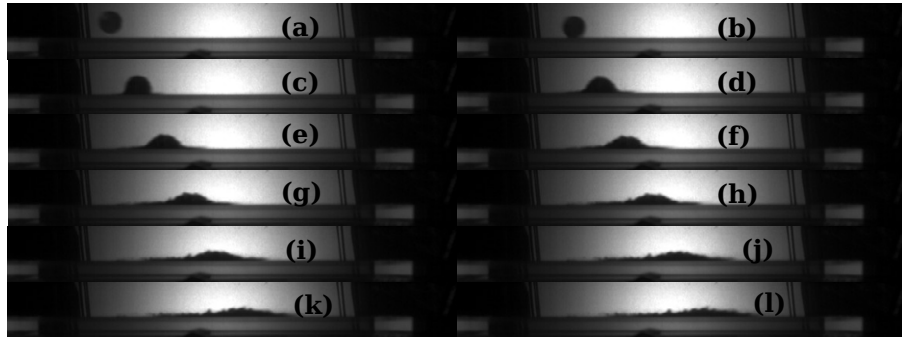


Figure 4: Image sequence of the ice ball fragmentation taken by the lateral camera (diameter: 12.9 mm, impact angle  $20^\circ$ , impact velocity:  $120.8 \text{ m.s}^{-1}$ ).

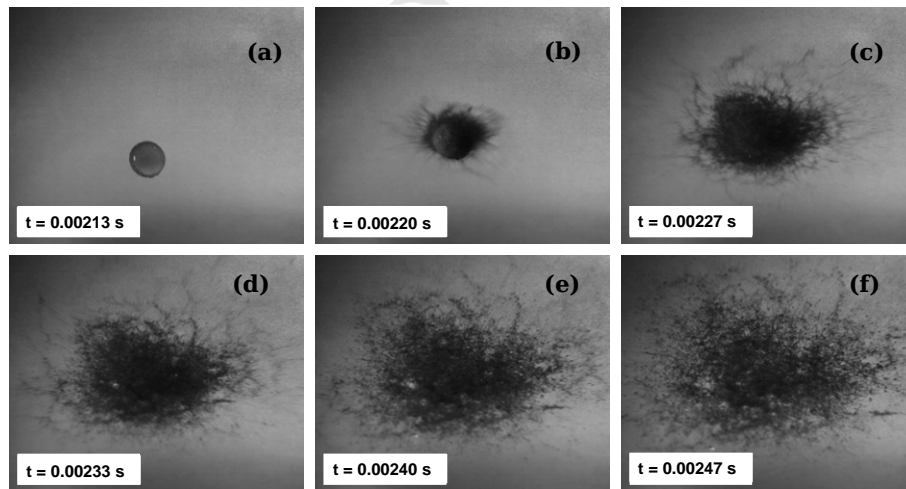


Figure 5: Front view of ice fragmentation (diameter: 12.9 mm, impact angle  $90^\circ$ , impact velocity:  $150.4 \text{ m.s}^{-1}$ ).

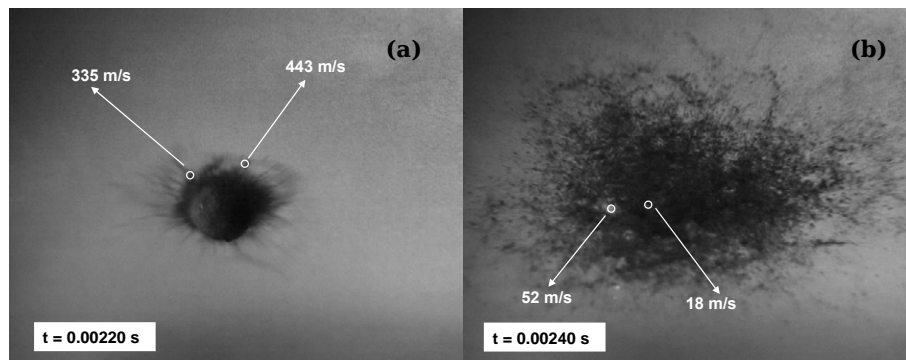


Figure 6: Magnification of the areas shown in Figure 5(b) and Figure 5(e).

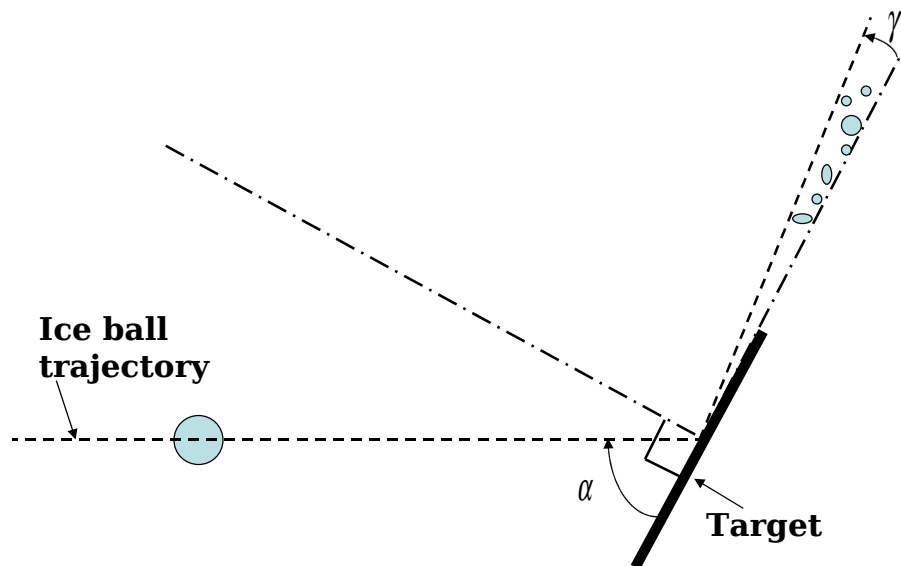


Figure 7: Definition of the post-impact angle.

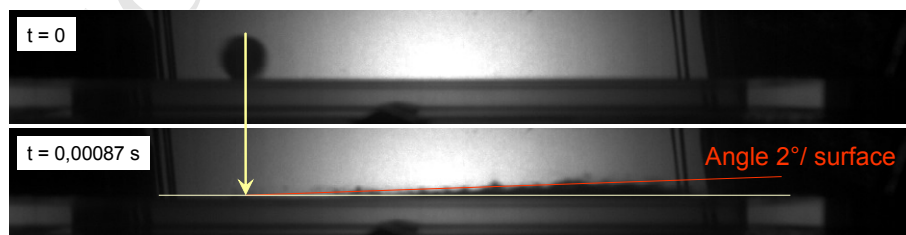


Figure 8: Example of measurement of the post-impact angle.

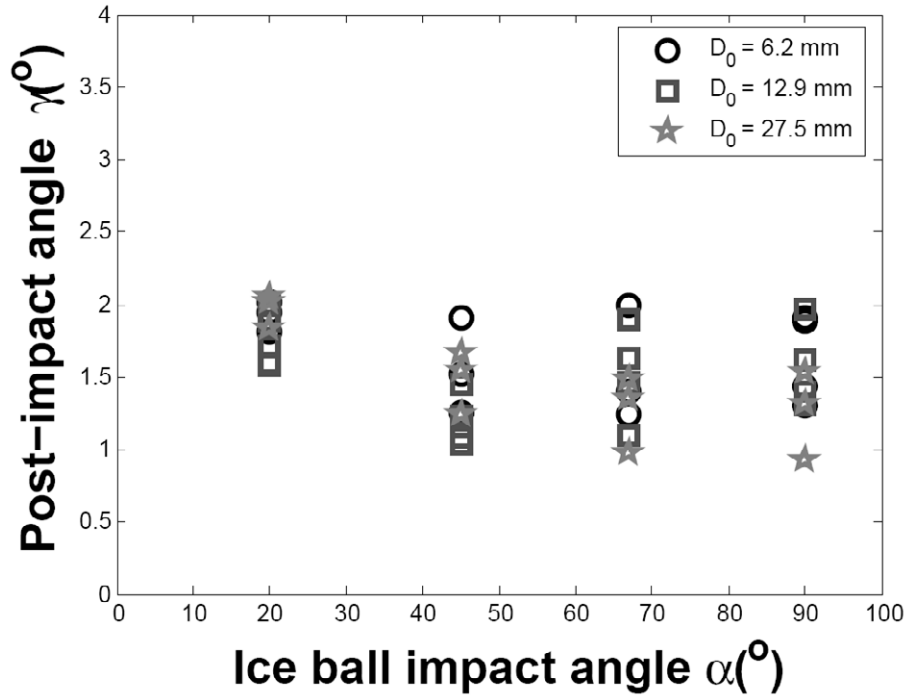


Figure 9: Values of the post-impact angle  $\gamma$  for the different experiments.

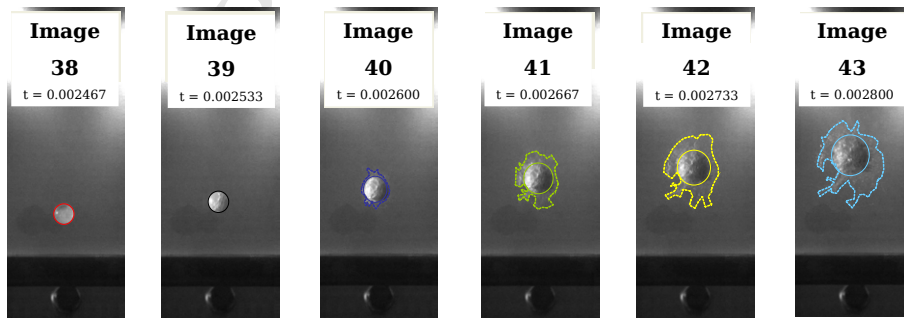


Figure 10: Analysis of images from the front camera: definition of the high density cloud contour (diameter: 12.9 mm, impact angle  $20^{\circ}$ , impact velocity:  $120.8 \text{ m.s}^{-1}$ ).

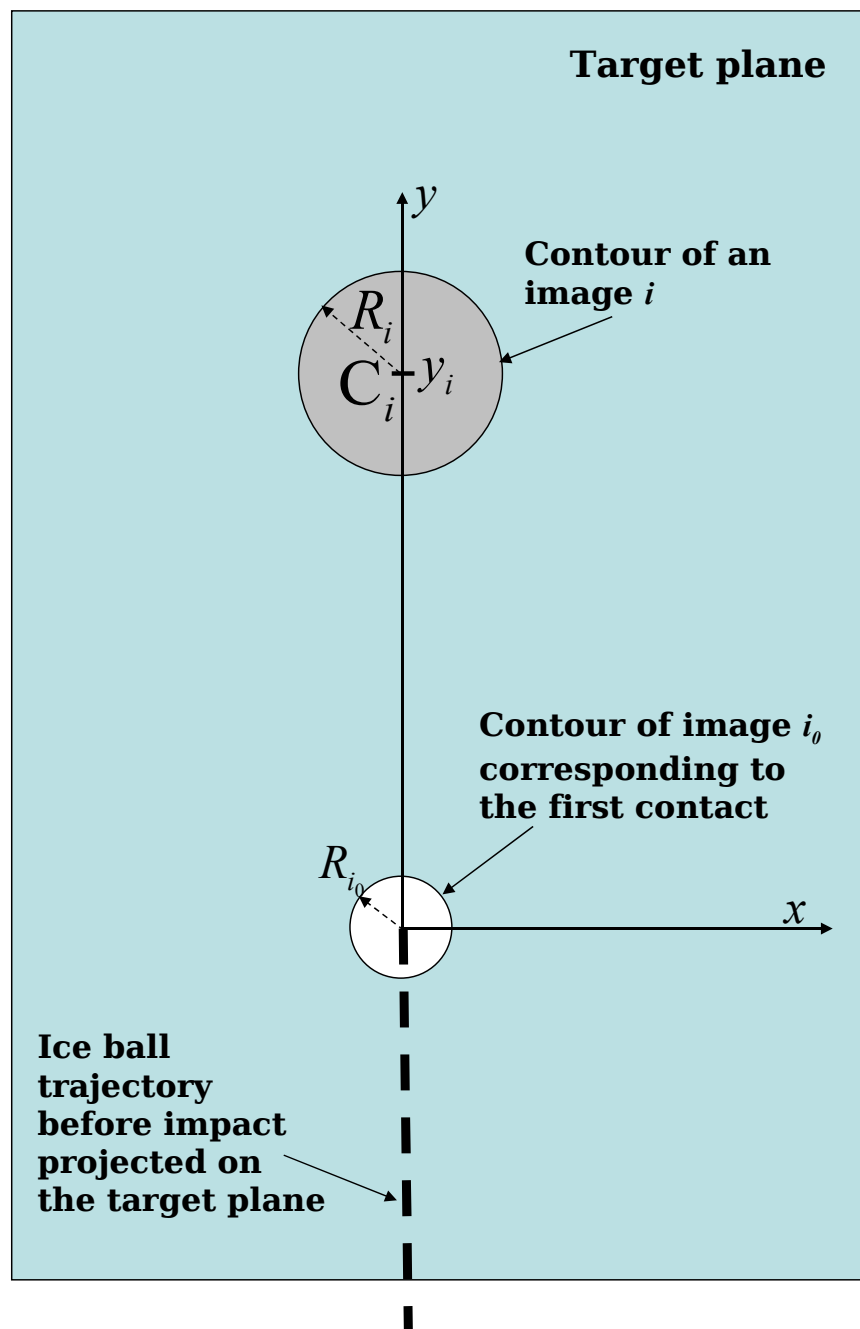


Figure 11: Schematic of the superposition of two circular contours on the image  $i_0$  corresponding to the first (initial) contact and to a subsequent image  $i$ .



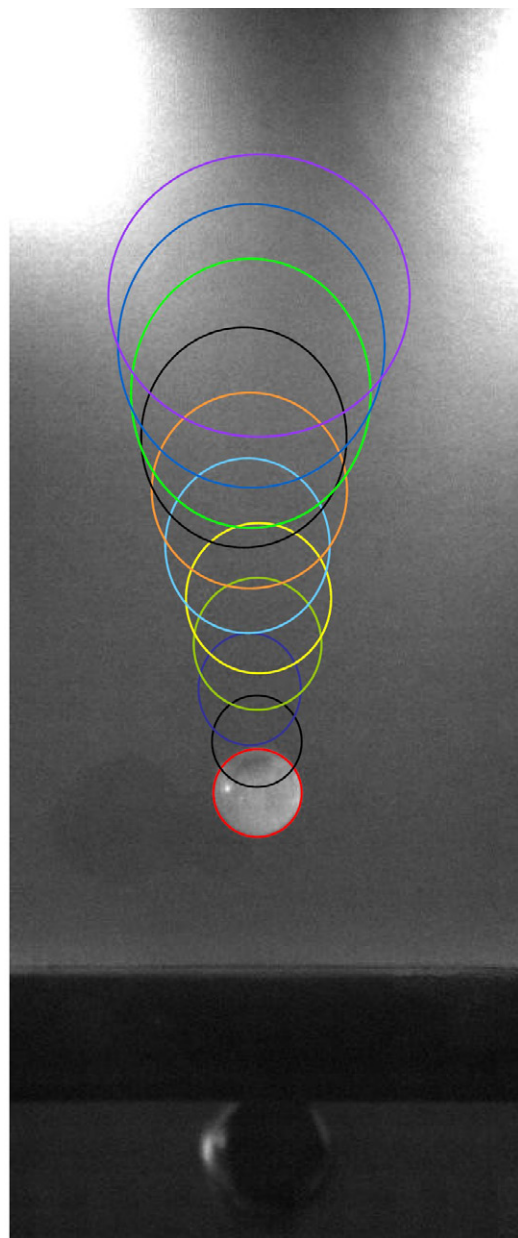


Figure 12: Superposition of circular contours: experimental result (diameter: 12.9 mm, impact angle  $20^\circ$ , impact velocity:  $120.8 \text{ m.s}^{-1}$ ).

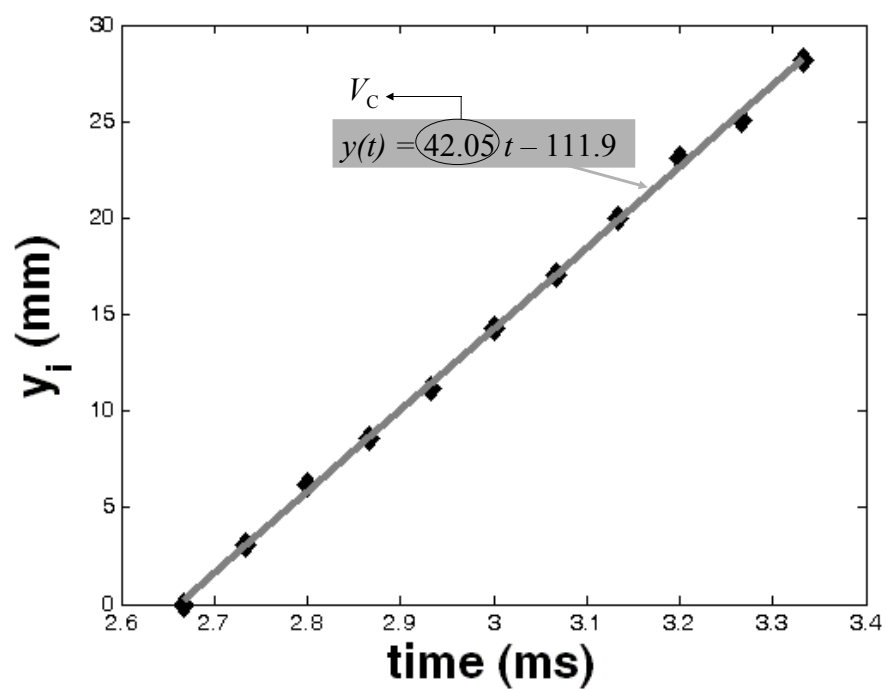


Figure 13: Evolution of the position of the cloud center with time (diameter: 12.9 mm, impact angle  $67^\circ$ , impact velocity:  $118.8 \text{ m.s}^{-1}$ ).

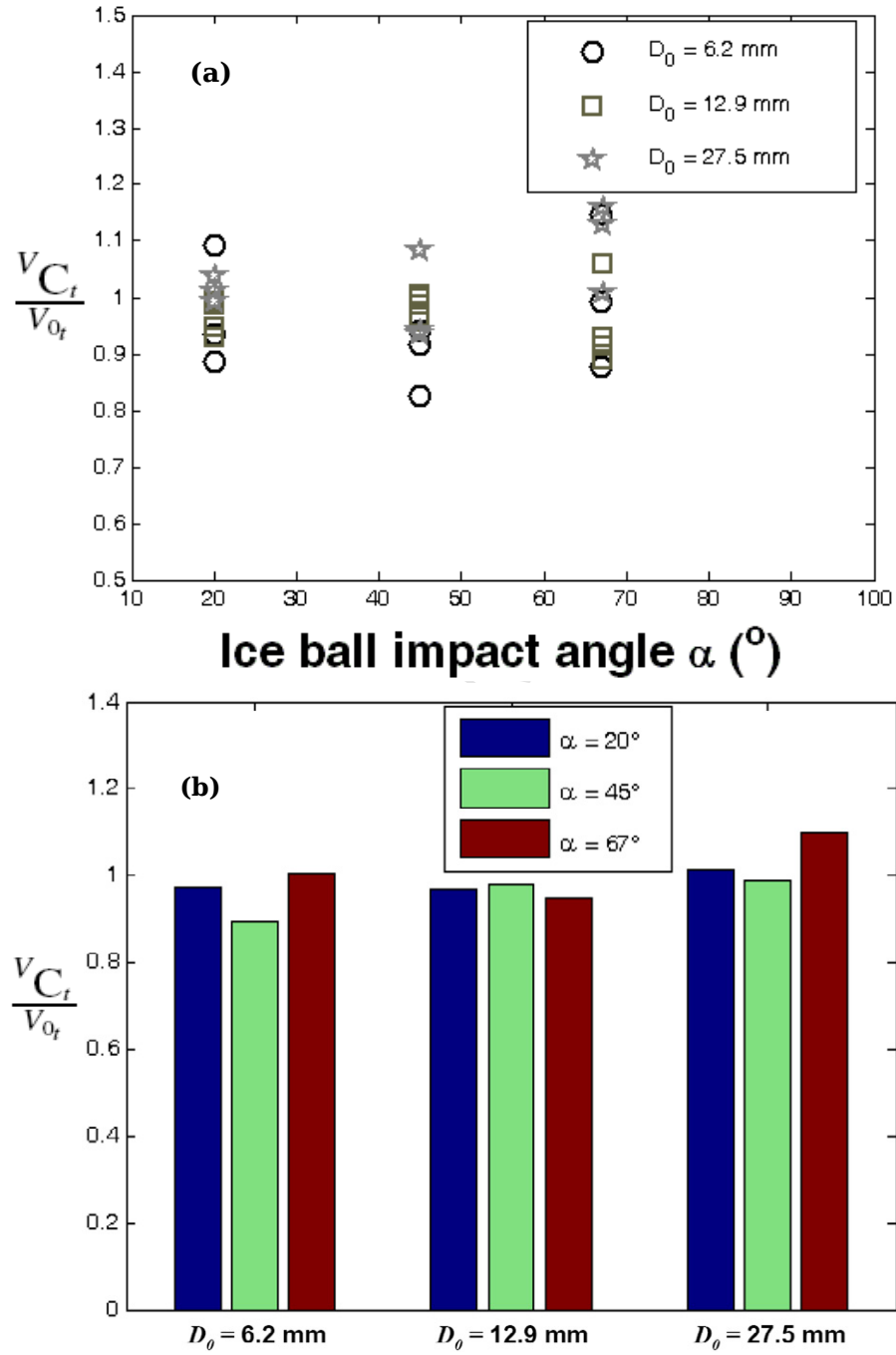


Figure 14: Values of the ratio  $V_{C_t}/V_{0_t}$ : (a) for all experiments; (b) average values for each configuration.

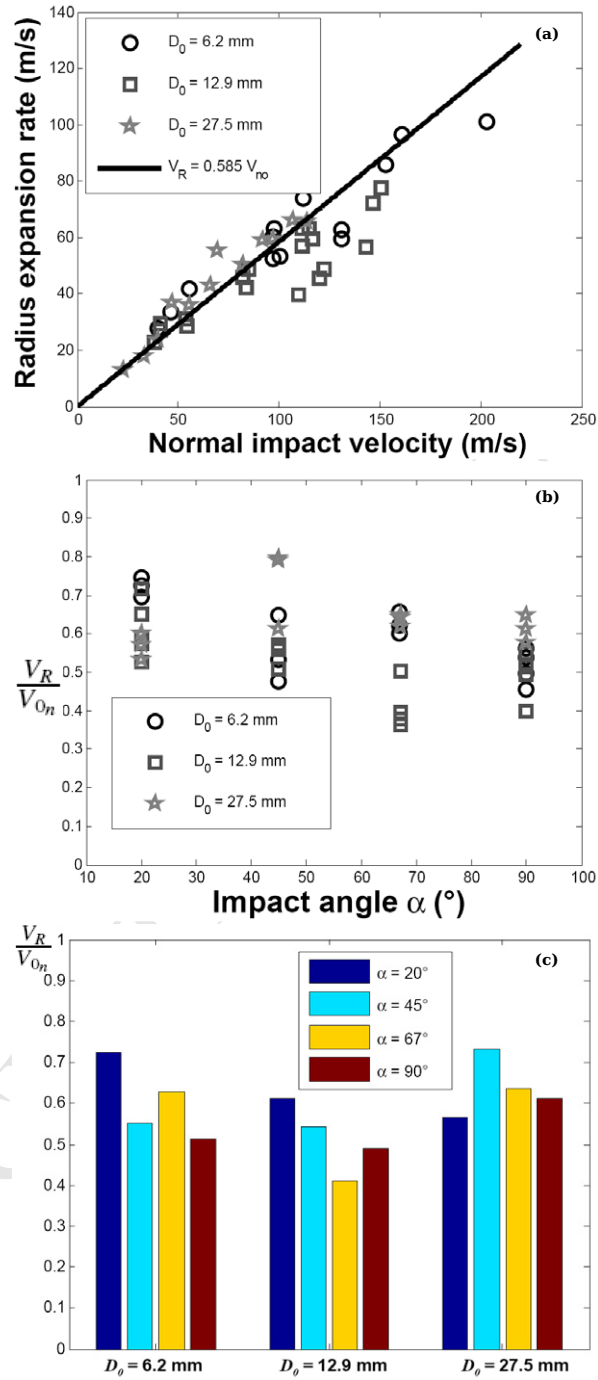


Figure 15: Radius expansion: (a) expansion rate  $V_R$  as a function of normal impact velocity  $V_{0n}$ ; (b)  $K_n = V_R/V_{0n}$  as a function of impact angle  $\alpha$  and (c) averaged values of  $K_n$ .

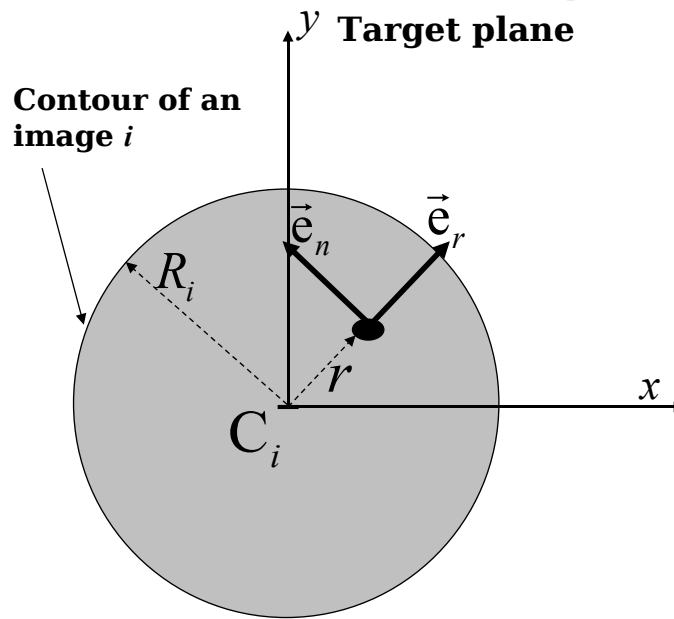


Figure 16: Mobile reference frame.

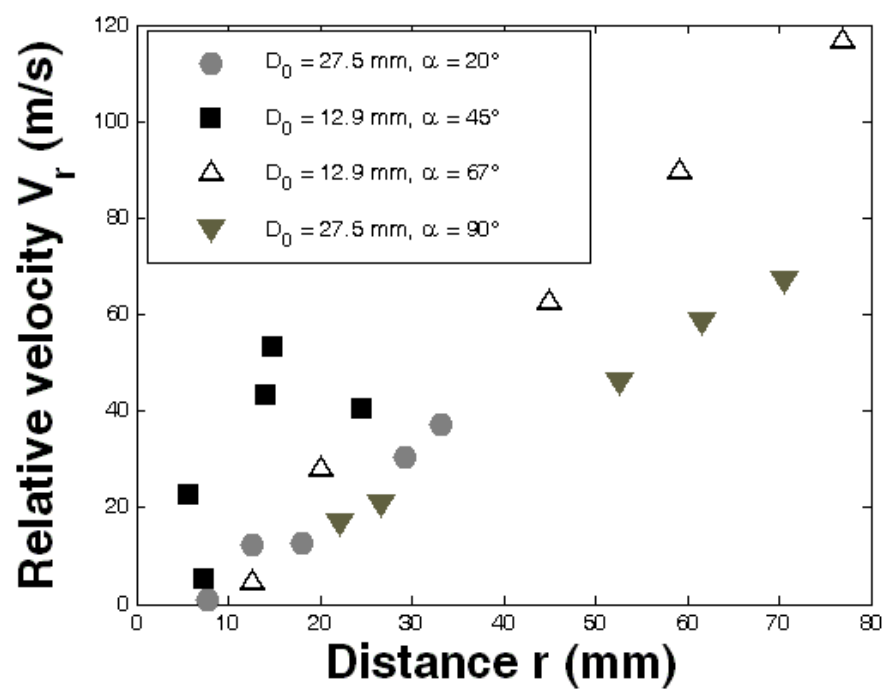


Figure 17: Relative velocity of fragments inside the cloud as a function of their distance to the cloud center.