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Comparaison of two homogenization methods using a damage model for a fibrous membrane, based on the fibers' fracture process at the microscale.

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Abstract

The objective of this paper is to evaluate two existing homogenization methods using a simple damage model for fibrous membrane and compare them. A macroscopic damage model of a fibrous membrane tissue based on the knowledge of the behavior and fracture process of a single fiber is constructed. Under some simplifying assumptions (linear behavior of the fibers, brittle fibers, constant angular distribution of the fibers), the evolution of fiber fracture is described by two macroscopic scalar variables; this enables the tissue's damage to be modeled without requiring a second discretization at the microscale. For the homogenization of the fibers contribution, an energetic method and a kinematic method are adapted and compared. Both lead to similar macroscopic behavior in the elastic phase, but the behavior differs in the fracture

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phase of the homogenized material. In the case of biological materials, the proposed law enables taking into account the often observed phenomena of crimping and damage of the fibers.

Key words: fibrous membrane tissue, damage, microstructure, homogenization, biaxial tension

1 1. Introduction

The question of the fracture modeling of fibrous tissues arises in many do-2 mains, including composites, papers and biological tissues. In the latter case, 3 the large presence of fibrous tissues of different natures in the human body 4 lead to an general research effort to explain how the macroscopic behavior 5 of those tissues is related to their microstructure, especially in the case of 6 non-homogeneous fiber distributions. In terms of modeling, two aspects have 7 to be considered : 1) the fibrous nature of the tissue and 2) the evolution 8 of damage in the tissue. Several options are proposed in the literature to 9 assess the first point. Using invariants to take into account the isotropic or 10 anisotropic nature of the fibrous contribution is a first option, introduced in 11 (Holzapfel, 2000). This method does not require any further homogenization 12 as the invariants already are a macroscopic representation of the fiber orien-13 tations and assumes that the fibers are perfectly aligned along one or several 14 directions. If the fibers are distributed around one or several preferred di-15 rections, a statistical distribution of the fibers is generally introduced. From 16 there, two options are suggested in the literature: either a structure tensor 17 (Gasser et al., 2006) can be used to provide a macroscopic representation of 18 the microstructure (the kinematic method) or the individual energetic con-19

tributions of the fibers are summed up (Lanir, 1983) (the *energetic method*). 20 A more recent option is the microsphere-based approach (Menzel and Waf-21 fenschmidt, 2009; Sáez et al., 2012) which uses spherical units in which the 22 homogenization process is conducted. The anisotropy of the tissue is then 23 described by associating different weights to the fiber directions. Using a 24 limited number of integration directions was proved to be enough to provide 25 good accuracy as well as a low computational cost to the two-scale simulation 26 of 3D tissues. Another recent feature is the macromolecular network model 27 proposed by (Kuhl et al., 2005), based on eight-chain unit cells and which 28 can handle fiber reorientation. 29

In this paper, the attention is focused on how the fiber distribution is taken into account: both the kinematic and energetic methods are adapted to the same fibrous tissue model to assess whether and in which domain they are equivalent.

The second point in modeling the fracture of fibrous tissue is the de-34 scription of the damage. In the field of biomechanics, the problem of the 35 fracture of fibrous tissues has received only limited attention. Some authors 36 model damage continuously using an internal damage variable which is to 37 be identified for 1D (Calvo et al., 2007) or 2D tissues (Balzani et al., 2006). 38 (Rodríguez et al., 2006) introduces a 3D model which also describes dam-39 age continuously, but uses two independent variables for the matrix and the 40 fibers. Other papers describe damage in the tissue as the result of fiber 41 fracture on the lower scale: (Hurschler et al., 1997) proposes a tendon-and-42 ligament model involving three scales (the fibrils', the fibers' and the tis-43 sue's) and studies the behavior and damage of the tissue in the case of fibers 44

aligned with a single direction. (Liao and Belkoff, 1999) also proposes a 45 fracture model of a 1D tissue, but takes into account the initial crimping of 46 the fibers. (Gasser and Holzapfel, 2006) models the fracture of arteries by 47 dissection, *i.e.* splitting through the thickness, which constitutes a different 48 fracture mode than surface tearing. (Cacho et al., 2007) proposes a damage 49 model for a fibrous tissue made of initially crimped fibers. More recently, 50 (Sáez et al., 2012) introduced damage in a microsphere model of the blood 51 vessel taking into account its initial anisotropy. 52

The fibrous tissue modeling in the present paper is similar to that from (Cacho et al., 2007). The idea of our model is to describe the evolution of the macroscopic damage in a fibrous membrane and therefore of the tissue anisotropy during a biaxial strain loading; under simplifying assumptions, it will be shown that two scalar variables are sufficient for this purpose. This simple model, including some extensions related to biological tissues, is used to illustrate the comparison of the two homogenization methods.

The outline of this paper is the following: the first part presents the two homogenization methods adapted to a fibrous membrane model; the second part is focused on the description of the simple damage model that we propose; the third part illustrates the comparison of the two homogenization methods using our damage model; the fourth part presents extensions of the damage model to biological phenomena such as damageable fibers and fiber crimp, as well as the consequences on the homogenization methods.

⁶⁷ 2. Modeling the tissue fracture

Let us consider a fibrous membrane and suppose that we know the in-68 fluence of a biaxial tension loading on the fibers breakage. We can proceed 69 to homogenize the behavior. Several homogenization methods are available 70 in the literature; the most commonly used are the ones described in (Lanir, 71 1983) and (Gasser et al., 2006). The first one consists in an energetic ho-72 mogenization of the fibers while the second is based on a kinematic homoge-73 nization. In this paper, these two methods are adapted to the fibrous tissue 74 damage problem and compared. 75

76 2.1. General framework

This section takes up the same general framework as in (Gasser et al., 77 2006). We consider a plane tissue consisting of a matrix and fibers. The 78 free energy ψ of the tissue is the sum of the free energy of the matrix ψ_m 79 and the free energy of the fibers ψ_f . A common way to enforce compress-80 ibility/incompressibility conditions is to split the strain energy of the matrix 81 into a volumetric part U_m and an isochoric part $\bar{\psi}_m$. It is usually assumed 82 that the incompressibility of biological tissues is due to the water content 83 of the ground matrix and that the fibers' contribution is negligible in the 84 volumetric part of the free energy. Therefore, it is not necessary to split the 85 fibers' contribution into volumetric and isochoric parts. 86

⁸⁷ Hence the free energy of the tissue can be written as:

$$\psi = \psi_m + \psi_f = U_m + \bar{\psi}_m + \psi_f \tag{1}$$

First we will address the matrix free energy expression; then, we will examine two homogenization methods for the contribution of the fibers.

90 2.2. Contribution of the matrix

The volumetric/isochoric decomposition is applied to the gradient of transformation \mathbf{F} , *i.e.* to the right Cauchy-Green strain tensor \mathbf{C} :

$$\mathbf{F} = J^{\frac{1}{3}} \bar{\mathbf{F}} \tag{2}$$

$$\mathbf{C} = J^{\frac{2}{3}} \bar{\mathbf{C}} \tag{3}$$

91

with $J = det(\mathbf{F})$. The cumulative energy decomposition enables the second Piola-Kirchhoff stress tensor (PK2) to be expressed as:

$$\mathbf{S}_m = 2\frac{\partial\psi_m}{\partial\mathbf{C}} = 2\left(\frac{\partial U_m}{\partial\mathbf{C}} + \frac{\partial\bar{\psi}_m}{\partial\mathbf{C}}\right) = \mathbf{S}_m^{vol} + \bar{\mathbf{S}}_m \tag{4}$$

92

We set $p = \partial U_m / \partial J$; we also assume that the matrix follows an isotropic, neo-Hookean behavior expressed by $\bar{\psi}_m = c \left(tr \bar{\mathbf{C}} - 3 \right) / 2$, where c is a material parameter. Then:

$$\mathbf{S}_{m}^{vol} = 2\frac{\partial U_{m}}{\partial \mathbf{C}} = 2\frac{\partial U_{m}}{\partial J}\frac{\partial J}{\partial \mathbf{C}} = pJ\mathbf{C}^{-1}$$
(5)

$$\bar{\mathbf{S}}_{\mathbf{m}} = 2\frac{\partial\bar{\psi}_m}{\partial\mathbf{C}} = 2\frac{\partial\bar{\psi}_m}{\partial\bar{\mathbf{C}}} : \frac{\partial\bar{\mathbf{C}}}{\partial\mathbf{C}} = J^{-\frac{2}{3}}\mathbb{P} : c\mathbf{I}$$
(6)

with the fourth-order operators $\mathbb{P} = \mathbb{I} - \frac{1}{3} \mathbb{C}^{-1} \otimes \mathbb{C}$ and $I_{ijkl} = (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk})/2$, where δ_{ij} is Kronecker's symbol.

⁹⁵ Thus, the contribution of the matrix to PK2 tensor becomes:

$$\mathbf{S}_m = pJ\mathbf{C}^{-1} + J^{-\frac{2}{3}}\mathbb{P} : c\mathbf{I}$$
(7)

97 2.3.1. The concept of fiber density function

In order to describe the strain energy in the tissue, we introduce the concept of angular fiber density, denoted $\rho(\xi)$. This function defines the number of the fibers whose orientation belongs to the interval $[\xi, \xi + d\xi]$. Before damage occurs, this function is normalized, *i.e.*:

$$\frac{1}{\pi} \int_{-\frac{\pi}{2}}^{\frac{\pi}{2}} \rho(\xi) \,\mathrm{d}\xi = \frac{1}{\pi} \int_{A_0} \rho(\xi) \,\mathrm{d}\xi = 1 \tag{8}$$

In the case of a homogeneous distribution, *i.e.* $\rho(\xi) = constant$, $\rho(\xi) = 103$ 1 $\forall \xi \in A_0 = [-\pi/2, \pi/2]$.

Once damage occurs, the density is a function of damage state **D**. The integration domain evolves and is then denoted $A(\mathbf{D})$. Given this description, we can write :

$$\int_{A_0} \rho(\xi, \mathbf{D}) d\xi = \int_{A(\mathbf{D})} \rho(\xi) d\xi$$
(9)

$A(\mathbf{D})$ represents the domain where the fibers are not broken.

108 2.3.2. The Energetic (E) homogenization method

In this method, the strain energy of the fibrous part of the tissue is assumed to be the sum of the strain energies of the strained, but undamaged, fibers. A fiber subjected to a Green-Lagrange strain field **E** is strained only along its longitudinal axis $\mathbf{n}(\xi)$ and its strain energy is $\phi_f = \phi_f(\varepsilon_f)$ *i.e.* $\phi_f(\xi, \mathbf{E})$, whose expression depends on the constitutive relation of the fiber.

114 Therefore, on the tissue's scale, the free energy ψ_f^E of the fibers is:

$$\psi_f^E(\mathbf{E}, \mathbf{D}) = \frac{1}{\pi} \int_{A_0} \rho(\xi, \mathbf{D}) \phi_f(\xi, \mathbf{E}) \,\mathrm{d}\xi \tag{10}$$

¹¹⁵ The expression of stress tensor PK2 is:

$$\mathbf{S}_{f}^{E} = \frac{\partial \psi_{f}^{E}(\mathbf{E}, \mathbf{D})}{\partial \mathbf{E}} = \frac{1}{\pi} \int_{A_{0}} \rho(\xi, \mathbf{D}) \frac{\partial \phi_{f}(\xi, \mathbf{E})}{\partial \mathbf{E}} \,\mathrm{d}\xi \tag{11}$$

Now let us consider the particular case where the fibers behave linearly
before damage. The total free energy of the fibrous part of the tissue is:

$$\psi_f^E(\mathbf{E}, \mathbf{D}) = \frac{1}{\pi} \int_{A_0} \rho(\xi, \mathbf{D}) \frac{1}{2} E \left(\mathbf{M} : \mathbf{E}\right)^2 \mathrm{d}\xi$$
(12)

where *E* is the Young's modulus of the fiber's material and $\mathbf{M} = \mathbf{n}(\xi) \otimes \mathbf{n}(\xi)$ the orientation tensor (see Section 3). Thus, the expression of stress tensor PK2 is:

$$\mathbf{S}_{f}^{E} = \frac{E}{\pi} \int_{A_{0}} \rho\left(\xi, \mathbf{D}\right) \left(\mathbf{M} : \mathbf{E}\right) \mathbf{M} \,\mathrm{d}\xi \tag{13}$$

Eq. (11) is valid if we suppose that the variation of the integral of the density function $\int_{A_0} \rho(\xi, D)$ with strain is negligible compared to the variation of $\phi_f(\xi, \mathbf{E})$, which is the case for the simple model considered in this approach. Therefore only the inner member of the integral is differentiated with respect to \mathbf{E} .

126 2.3.3. The Kinematic (K) method

The *kinematic* method is derived from works described in (Gasser et al., 2006), among others. We introduce a generalized second-order structure tensor **H** defined by Eq.(14). This tensor is used as a macroscopic projector of the strain tensor onto the structure of the undamaged fibers.

$$\mathbf{H} = \frac{1}{\pi} \int_{A_0} \rho(\xi, D) \mathbf{n}(\xi) \otimes \mathbf{n}(\xi) \mathrm{d}\xi$$
(14)

 H_{33} equals zero because the fibers are oriented only perpendicularly to Direction 3.

Thus, the constitutive law is applied to the tissue rather than to its constituent fibers, taking the scalar $E_h = \mathbf{H} : \mathbf{E}$ as the strain value to express the macroscopic strain energy ψ_f^K here in the linear case (Eq.15).

$$\psi_f^K = \frac{1}{2} E \cdot E_h^2 = \frac{1}{2} E \left(\mathbf{H} : \mathbf{E} \right)^2$$
 (15)

From that expression, we deduce the PK2 tensor corresponding to the K method:

$$\mathbf{S}_{f}^{K} = \frac{\partial \psi_{f}^{K}}{\partial \mathbf{E}} = E\left(\mathbf{H} : \mathbf{E}\right)\mathbf{H}$$
(16)

Once again, the damage variable D and the Green-Lagrange strain tensor E
are independent variables. So H is not differentiated with respect to E.

¹³⁸ 2.4. Association of contributions from the matrix and the fibers

Let us recall that the total energy of the tissue can be divided into three parts (Eq.1) including two from the matrix contribution (Eq.7). Thus, depending on the homogenization method, tensor PK2 has either
of the two forms:

$$\mathbf{S}^{E} = pJ\mathbf{C}^{-1} + J^{-\frac{2}{3}}\mathbb{P} : c\mathbf{I} + \frac{2E}{\pi} \int_{A_{0}} \rho\left(\xi, \mathbf{D}\right) \left(\mathbf{M} : \mathbf{E}\right) \mathbf{M} \mathrm{d}\xi$$
(17)

$$\mathbf{S}^{K} = pJ\mathbf{C}^{-1} + J^{-\frac{2}{3}}\mathbb{P} : c\mathbf{I} + E\left(\mathbf{H} : \mathbf{E}\right)\mathbf{H}$$
(18)

Since the objective is to implement the model into a finite element calculation code, the tissue is modeled in 3D and plane stresses; in order to apply this condition, we define the Cauchy stress tensor **T**:

$$\mathbf{T} = J^{-1} \mathbf{F} \cdot \mathbf{S} \cdot \mathbf{F}^T \tag{19}$$

The components derived from the matrix contribution are:

$$\mathbf{T}_{m}^{vol} = p\mathbf{I} \tag{20}$$

$$\bar{\mathbf{T}}_{m} = J^{-\frac{5}{3}}c\left(\mathbf{B} - \frac{1}{3}tr\left(\mathbf{C}^{-1}\right)\mathbf{I}\right)$$
(21)

where $\mathbf{B} = \mathbf{F} \cdot \mathbf{F}^T$ is the left Cauchy-Green tensor. Concerning the contribution of the fibers, we distinguish the two homogenization methods:

$$\mathbf{T}_{f}^{E} = J^{-1}\mathbf{F}.\mathbf{S}_{f}^{E}.\mathbf{F}^{T} = \frac{2E}{J\pi}\mathbf{F}.\int_{A_{0}} \rho\left(\xi,\mathbf{D}\right)\left(\mathbf{M}:\mathbf{E}\right)\mathbf{M}\mathrm{d}\xi.\mathbf{F}^{T}$$
(22)

$$\mathbf{T}_{f}^{K} = J^{-1} \mathbf{F} \cdot \mathbf{S}_{f}^{E} \cdot \mathbf{F}^{T} = \frac{E}{J} \left(\mathbf{H} : \mathbf{E} \right) \mathbf{F} \cdot \mathbf{H} \cdot \mathbf{F}^{T}$$
(23)

We can notice that the fibrous contribution to the overall tissue does not create a Cauchy stress components out of the plane; this is due to the formulation of the fiber density function and to the fact that we do not split the fibrous contribution into a volumetric component and an isochoric component.

¹⁵³ The plane stress condition is expressed by:

$$T_{33} = p + J^{-\frac{5}{3}}c\left(\lambda_3^2 - \frac{1}{3}\left(\lambda_1^2 + \lambda_2^2 + \lambda_3^2\right)\right) = 0$$
(24)

We can observe that this equation involves p, the material characteristics of the matrix and the components λ_i of the deformation gradient tensor. Here, as is commonly assumed for soft biological tissues, the matrix is considered to be incompressible. In this case, the strain component across the thickness is defined by Eq.(25) and p is a Lagrange multiplier determined by the plane stress condition (Eq.26).

$$J = 1 \Leftrightarrow \lambda_3 = \frac{1}{\lambda_1 \lambda_2} \tag{25}$$

$$p = -\frac{c}{3} \left(\frac{2}{\lambda_1^2 \lambda_2^2} - \lambda_1^2 - \lambda_2^2 \right) \tag{26}$$

The two homogenization methods have been described; the next section presents the damage model that was constructed to compare them.

¹⁶² 3. A simple kinematic fracturing fiber model

In this section, we propose a simple model to describe the evolution of the macroscopic damage of a plane fibrous tissue subjected to biaxial tension loading, which is assumed to be the result of fiber breakage on the microscale. The two objectives of this section are first, to build a damage evolution law for the macroscale damage, taking into account its anisotropic nature due to the microscale phenomena; second, to compare the two homogenization methods presented in the previous section.

¹⁷⁰ The underlying assumptions of this section are the following:

- the fibers are rectilinear, linear elastic, and brittle;
- prior to damage, the angular distribution function of the fibers is
 known, continuous and nonzero;
- the phenomenon of fiber reorientation during loading is neglected;
- the *principal* directions of the biaxial strain loading do not change.

The third assumption relies on the conclusions of Sacks and Gloeckner (1999), which observed that the closer the loading to equibiaxiality, the lesser the reorientation of the fibers. Therefore, in the framework of the present study which focuses on biaxial loading, it appears acceptable to ignore fiber reorientation.

This model is academic. It is designed to produce clear conclusions when the two homogenization methods proposed in the previous section are compared. It can be extended using a two scale approach to more realistic situations as uncrimping, damage, non isotropic fiber orientations: the price to pay to these extensions is a larger number of internal variables to describe the small scale state.

187 3.1. Parameterization of the problem

Let us consider a plane fibrous tissue. In the material plane $(\mathbf{X}_{\mathbf{m}}, \mathbf{Y}_{\mathbf{m}}, \mathbf{Z}_{\mathbf{m}})$, the direction of a fiber is characterized by the angle $\xi \in [-\pi/2, \pi/2]$ and its initial direction vector **n** defined by:

$$\mathbf{n} = \cos \xi \, \mathbf{X}_{\mathbf{m}} + \sin \xi \, \mathbf{Y}_{\mathbf{m}} \tag{27}$$

The tissue is subjected to a biaxial strain characterized by the macroscopic Green-Lagrange strain tensor **E** described in Cartesian coordinates by:

$$\mathbf{E} = \varepsilon_r k \left(\cos \varphi \, \mathbf{X}_{\mathbf{m}} \otimes \mathbf{X}_{\mathbf{m}} + \sin \varphi \, \mathbf{Y}_{\mathbf{m}} \otimes \mathbf{Y}_{\mathbf{m}} \right) \tag{28}$$

$$= E_1 \mathbf{X}_{\mathbf{m}} \otimes \mathbf{X}_{\mathbf{m}} + E_2 \mathbf{Y}_{\mathbf{m}} \otimes \mathbf{Y}_{\mathbf{m}}$$
(29)

194

where ε_r is the ultimate longitudinal strain of the fibers. φ is the loading angle. From here on, we will assume that $\varphi \in [0, \pi/2]$ and $k \ge 0$, which implies strict biaxial tension.

The fibers constituting the tissue are uniaxial elements which can withstand only solicitations along their axis. The longitudinal Green strain ε_f of a fiber oriented along an angle ξ is defined by:

$$\varepsilon_f = \mathbf{n}(\xi) \cdot \mathbf{E} \cdot \mathbf{n}(\xi) = k \varepsilon_r \left(\cos \varphi \cdot \cos^2 \xi + \sin \varphi \sin^2 \xi \right)$$
(30)

We can observe that for $\varphi = \frac{\pi}{4}$ all the fibers are solicited equally; then, their longitudinal strain is $\varepsilon_f = \frac{k\varepsilon_r}{\sqrt{2}}$. Also, differentiating φ with respect to ²⁰³ ξ shows that the most highly loaded fibers are oriented along the principal ²⁰⁴ directions of the strain tensor, that is $\xi = 0$ or $\xi = \frac{\pi}{2}$.

205 3.2. Initial elasticity range

The elasticity range \mathcal{D} of a fiber is defined in the strain space by:

$$\mathcal{D} = \{ \varepsilon_f \mid F(\varepsilon_f) = \varepsilon_f - \varepsilon_r < 0 \}$$
(31)

The corresponding elasticity range of the tissue, denoted S, is:

$$S = \{ \mathbf{E} / \forall \xi, \mathbf{n}(\xi) \cdot \mathbf{E} \cdot \mathbf{n}(\xi) - \varepsilon_r < 0 \}$$
(32)

The shape of S corresponds to the resolution of the equation $\varepsilon_f - \varepsilon_r < 0$ and is described by:

$$\begin{cases} k < \frac{1}{\cos\varphi} \quad \forall \varphi \in \left[0, \frac{\pi}{4}\right] \\ k < \frac{1}{\sin\varphi} \quad \forall \varphi \in \left[\frac{\pi}{4}, \frac{\pi}{2}\right] \end{cases}$$
(33)

At the boundary of S, at least one fiber breaks as the non-rupture criterion is not respected anymore (Eq.31). The first fiber to break is always the one oriented along $\xi = 0$ if $\varphi \leq \frac{\pi}{4}$ or the one oriented along $\xi = \frac{\pi}{2}$ if $\varphi \geq \frac{\pi}{4}$. The next section describes the damage process of the fibrous tissue.

212 3.3. Damage evolution

We now consider different loading cases to study the evolution of the damage state of this model at the micro scale. Different cases are treated to better understand the potentialities of the model. ²¹⁶ Case of a proportional loading. We assume that the tissue is subjected to ²¹⁷ a proportional strain loading (*i.e.* with φ constant) so as to reach a point ²¹⁸ defined by (k,φ) such that:

$$\begin{cases} k > \frac{1}{\cos\varphi} & if \ \varphi \in \left[0, \frac{\pi}{4}\right] \\ k > \frac{1}{\sin\varphi} & if \ \varphi \in \left[\frac{\pi}{4}, \frac{\pi}{2}\right] \end{cases}$$

The damaged state at the micro scale is defined by two subsets: the 219 subset of healthy fibers and the set of broken ones. These sets are simply 220 defined by two angles ξ_1 and ξ_2 . From here on, the vector of the two damage 221 variables ξ_1 and ξ_2 will be denoted **D** which characterizes the damaged tissue 222 state. These angles are obtained by the solution of equation $\varepsilon_f(\xi) - \varepsilon_r > 0$ 223 (which is detailed in Appendix A). A proportional loading with an intensity 224 k greater than the bounds defined in Eq.(33) leads to the fracture of the 225 fibers as follows: 226

$$\varphi \in \left[0, \frac{\pi}{4}\right] : \begin{cases} \text{all fibers are broken} \forall \xi \in \left[-\xi_1, \xi_1\right] \\ \xi_1 = \arccos\sqrt{\frac{1-k\sin\varphi}{k\left(\cos\varphi - \sin\varphi\right)}} \end{cases}$$
(34)
$$\varphi = \frac{\pi}{4} : \text{all fibers break simultaneously at } k = \sqrt{2} \text{ (Eq.30)}$$
(35)

$$\varphi \in \left[\frac{\pi}{4}, \frac{\pi}{2}\right] : \begin{cases} \text{all fibers are broken} \forall \xi \in \left[-\frac{\pi}{2}, -\xi_2\right] \cup \left[\xi_2, \frac{\pi}{2}\right] \\ \xi_2 = \arccos\sqrt{\frac{1 - k \sin\varphi}{k \left(\cos\varphi - \sin\varphi\right)}} \end{cases}$$
(36)

²²⁷ Micro scale damage state and the angles ξ_1 and ξ_2 are illustrated in Fig.1 ²²⁸ for two typical cases: { $\varphi = \pi/6, k = 1.7$ } and { $\varphi = \pi/2.7, k = 1.2$ }. Fig.2 ²²⁹ displays the evolution of ξ_1 with increasing k values for different angles φ .

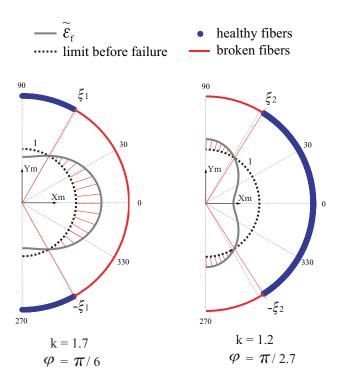


Figure 1: Examples of fiber damage states: if the normalized strain $\tilde{\varepsilon_f} = \frac{\varepsilon_f}{\varepsilon_r}$ is greater than 1 (the red-hatched zones), the fibers break.

Loading case of a constant-amplitude strain but rotating angle φ . It is interesting to treat the case of a constant-amplitude strain (k = cte) separately because it corresponds to an evolution of damage that is not monotonic. If $k < \sqrt{2}$ and φ evolves between 0 and $\pi/2$, fiber fracture occurs near $\xi = 0$ and $\xi = \pi/2$, but the fibers near $\pi/4$ are unaffected (Fig.3 illustrates the evolution of the two angles when φ increases from 0 to $\frac{\pi}{2}$). If $k \ge \sqrt{2}$, all the fibers break at $\varphi = \pi/4$.

²³⁷ Case of an arbitrary loading. We now consider that both the characteris-²³⁸ tic angle φ and the amplitude of the loading vary. This corresponds to the

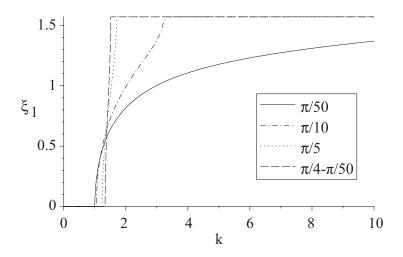


Figure 2: Examples of evolution of the damage variable ξ_1 for different values of φ

general complex case. The solution of evolution of damage state can be predicted with an incremental approach. A special attention has to be paid to the vicinity of singular point $\{\sqrt{2}, \frac{\pi}{4}\}$ at which all the fibers break simultaneously. Therefore, it is necessary to describe the path of the loading point precisely, especially in the vicinity of this point. Detailed explanation on the numerical strategy can be found in Brunon (2011).

245 3.4. Macroscopic evolution of the elasticity range: distorsion

The fracture surface at the macroscopic level evolves as damage increases. We denote A the set of healthy fibers angles. The fiber density function becomes:

$$\rho(\xi, \mathbf{D}) = \begin{cases} 0 & \text{if } \xi \notin A \\ 1 & \text{if } \xi \in A \end{cases}$$
(37)

²⁴⁶ The fiber density function can be observed for a particular case in Fig.4.

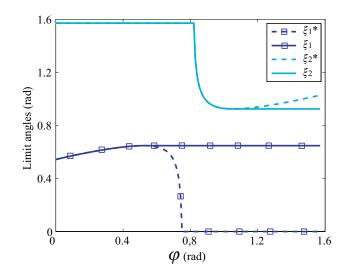


Figure 3: Evolution of the damage variables ξ_1 and ξ_2 as functions of φ in the case of a circular loading (an increasing φ with $k < \sqrt{2}$ constant). ξ_1^* and ξ_2^* are the virtual damage variables due to the current loading without taking the loading history into account.

The elasticity range becomes:

$$\mathcal{S} = \{ \mathbf{E} / \forall \xi \in A, \mathbf{n}(\xi) . \mathbf{E}(k, \varphi) . \mathbf{n}(\xi) - \varepsilon_r < 0 \}$$
(38)

The distortion of the surface is simply the change of the set A. More details for the model are described in Brunon (2011).

249 4. Influence of the homogenization method on the shape of the 250 stress strain curves

This section compares the properties of the two homogenization methods when they are applied to the simple fiber breaking damage model defined in the preceding section. All the graphs of this part correspond to the biaxial loading described on Fig.7(a).

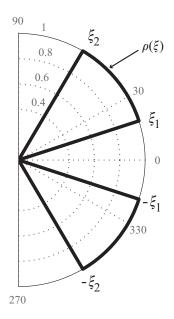


Figure 4: Shape of the density function for a particular damage case $(k = 1.5, \varphi = \pi/10$ and $\pi/3)$.

255 4.1. Macroscopic structure tensor properties

Let us observe some properties of the resulting macroscopic structure tensor **H**. We can immediately observe that $H_{12} = H_{21} = 0$ because the function cos * sin is odd. Indeed, since the integration intervals are always symmetrical with respect to 0, these two components are always zero, even when damage occurs. Besides, damage affects the components of **H** directly. The even nature of functions cos^2 and sin^2 leads to:

$$\mathbf{H} = \frac{2}{\pi} \int_{0}^{\frac{\pi}{2}} \rho(\xi, \mathbf{D}) \mathbf{n}(\xi) \otimes \mathbf{n}(\xi) \mathrm{d}\xi$$
(39)

An example of the evolution of the nonzero components of **H** for a loading up to rupture and for a uniform angular distribution prior to damage, is given

in Fig.5.

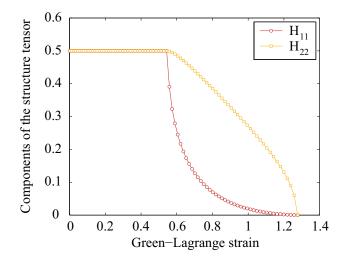


Figure 5: An example of the evolution of the components of the structure tensor ($\varepsilon_r = 0.5$, $\varphi = \pi/10$).

265 4.2. Homogenization methods comparison

A plot of the fibers' contribution to the strain energy for each method 266 (Fig.6) shows that under the current assumptions of uniform angular distri-267 bution prior to damage and brittle linear fibers, the two macroscopic models 268 behave differently when damage occurs. This difference can be observed by 269 expanding the expressions of these energies (Eq.40,41): it comes down to 270 the difference between the integral of a square and the square of an integral. 271 Indeed, for ξ_1 and ξ_2 constant (especially prior to damage), the ratio of ψ_f^K 272 to ψ_f^E is constant throughout the loading and independent of the value of 273 the elastic parameter E. 274

$$\psi_f^E = \frac{E}{\pi} \int_{\xi_1}^{\xi_2} \left(\mathbf{M}(\xi) : \mathbf{E} \right)^2 d\xi = \frac{E}{\pi} \int_{\xi_1}^{\xi_2} \left(E_1 \cos^2 \xi + E_2 \sin^2 \xi \right)^2 d\xi$$
(40)

$$\psi_f^K = \frac{E}{2} \left(\mathbf{H} : \mathbf{E} \right)^2 = \frac{2E}{\pi^2} \left(\int_{\xi_1}^{\xi_2} \left(E_1 \cos^2 \xi + E_2 \sin^2 \xi \right) d\xi \right)^2$$
(41)

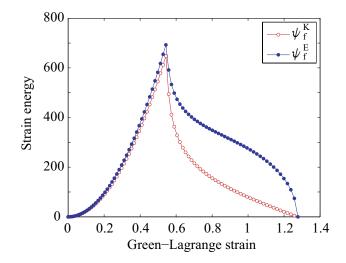


Figure 6: Comparison of the macroscopic strain energies of the fibrous tissue for the two proposed homogenization methods.

Besides, the shape of the components of tensor PK2 (Fig.7, b) show that the K method leads to the same value of the two nonzero components of \mathbf{S}_{f}^{K} prior to damage, whereas the tension applied to the tissue is not equibiaxial. Conversely, with the E method, this unrealistic result is not obtained. This is probably an illustration of the possible non-physical results obtained when the structure tensor theory is applied to a planar isotropic fiber distribution (Holzapfel et al, 2010). However for both homogenization methods, the Cauchy stress components (Fig.7, c) are consistent with the components of the strain tensor. Finally, there is a clear difference between the two methods concerning the concavity of the stress component corresponding to the least solicited direction. The softening part of the constitutive relation is much more anisotropic using the E method than using the K method.

287 5. Extensions of the model

In this part, two extensions related to biological tissues are proposed for the damage model and the two homogenization methods are compared for all of them. In the two cases, we consider that the angular distribution of the fibers is uniform before damage.

²⁹² 5.1. Case of damageable fibers

The general case described in the previous section is based on the assumption that the fibers are brittle. We now assume that they are damageable. This is a realistic assumption as collagen fibers have a substructure consisting of fibrils which can break progressively (Kastelic et al., 1978).

²⁹⁷ We choose here a continuous description of the 1D damage in the collagen ²⁹⁸ fibers. The behavior of a fiber is described in Fig.8: we introduce the damage ²⁹⁹ variable D, whose evolution is linear, and the damage strain ε_D . In the case ³⁰⁰ of a tissue which is initially isotropic and whose fiber density is a binary ³⁰¹ variable, there is a duality relation between ρ and the damage variable D, ³⁰² which enables a simple description of stress homogenization (Eq.43).

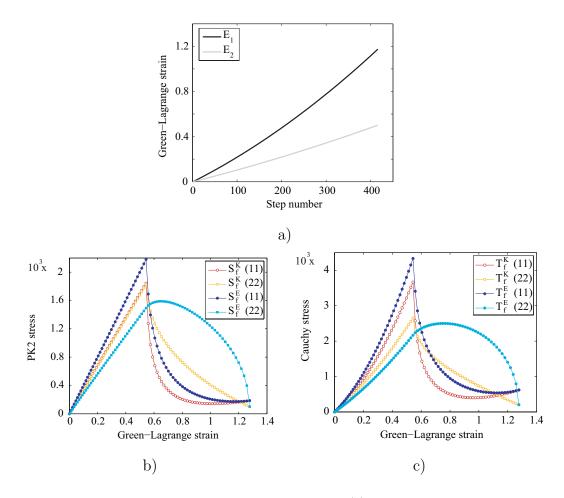


Figure 7: Biaxial strain loading applied to the tissue (a); evolution of the tensor components of PK2 (b) and Cauchy (c) as functions of the loading amplitude. The stress components do not revert back to 0 after complete rupture of the fibers because of the presence of the matrix.

$$E \text{ method:} \begin{cases} \phi_f = \frac{1}{2} E(1 - D(\mathbf{\Gamma}, \xi)) \varepsilon_f^2 \\ \psi_f^E = \int_{A_0} \frac{1}{2} E(1 - D) (\mathbf{n}.\mathbf{E}.\mathbf{n})^2 \, \mathrm{d}\xi \end{cases}$$
(42)
$$K \text{ method:} \begin{cases} \mathbf{H} = \frac{1}{\pi} \int_{A_0} \rho(\mathbf{\Gamma}, \xi) \mathbf{n} \otimes \mathbf{n} \, \mathrm{d}\xi \\ \psi_f^K = \frac{1}{2} E (\mathbf{H} : \mathbf{E})^2 \end{cases}$$
(43)

In these expressions, tensor Γ represents the value of the strain tensor **E** which led to the current macroscopic damage state of the tissue. Since this state is characterized by two independent variables ξ_1 and ξ_2 , we distinguish ${}^{1}\Gamma$ and ${}^{2}\Gamma$ which led to the current values of ξ_1 and ξ_2 respectively. The piecewise linear evolution of the variable D or ρ is given within the range $[0, \pi/4]$ (Eq.44). Within the range $[\pi/4, \pi/2]$, the expression is similar, but involves ${}^{2}\Gamma$.

$$D = \begin{cases} 1 & \text{if } \xi \leq \xi_1 \\ \frac{\varepsilon_f^*(\xi) - \varepsilon_D}{\varepsilon_r - \varepsilon_D} = \frac{\Gamma_1 \cos^2 \xi + \Gamma_2 \sin^2 \xi - \varepsilon_D}{\varepsilon_r - \varepsilon_D} & \text{if } \xi_1 \leq \xi \leq \xi_1' \\ 0 & \text{if } \xi_1' \leq \xi \end{cases}$$
(44)

$$\rho(\xi, {}^{1}\Gamma) = 1 - D(\xi, {}^{1}\Gamma) \ \forall \xi$$
(45)

The shape of the strain energy for each method is given in Fig.9. We obtain a smoothed peak for the evolution of the Cauchy stress in the tissue, corresponding to a more progressive macroscopic damage of the tissue (Fig.10).

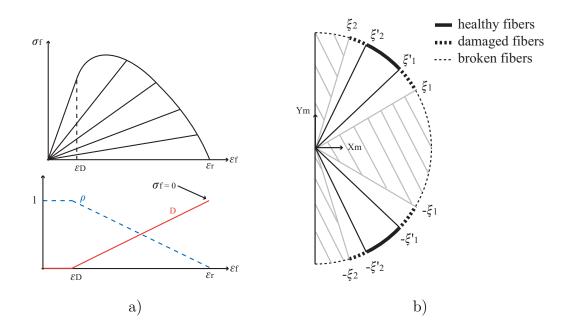


Figure 8: Constitutive law of a damageable fiber and evolution of the damage variable D and density variable ρ throughout the loading (a); damage state of the tissue's fibers (b).

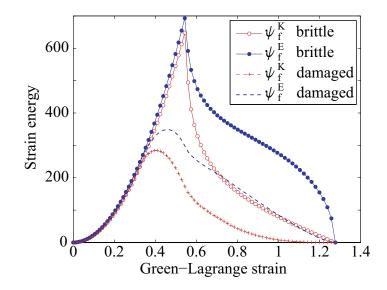


Figure 9: Strain energy during biaxial tension loading for the K and E methods in the case of brittle fibers or damageable fibers.

³¹⁴ 5.2. Introduction of fiber uncrimping

In many applications (biological tissues, elastomers, etc.) the loaded fibers uncrimp before being stretched. To take this phenomenon into account, we adapt the method described in (Cacho et al., 2007) and introduce the strain value ε_t beyond which the fiber starts becoming stretched (Fig.11).

The function $t(\xi)$ is introduced to describe the stretched or unstretched state of the fibers: $t(\xi) = 0$ if $\varepsilon_f < \varepsilon_t$, $t(\xi) = 1$ otherwise. Then, the description of the two macroscopic models is that of Eq.(46,47).

$$E \text{ method:} \begin{cases} \phi_f(\mathbf{E},\xi) = \frac{1}{2} Et(\xi) \left(\varepsilon_f(\xi) - \varepsilon_t\right)^2 = \frac{1}{2} Et(\xi) \left(\mathbf{n}(\xi).\mathbf{E}.\mathbf{n}(\xi) - \varepsilon_t\right)^2 \\ \psi_f^E = \frac{1}{\pi} \int_{A_0} \rho(\xi) \phi_f(\mathbf{E},\xi) \,\mathrm{d}\xi \end{cases}$$

$$\tag{46}$$

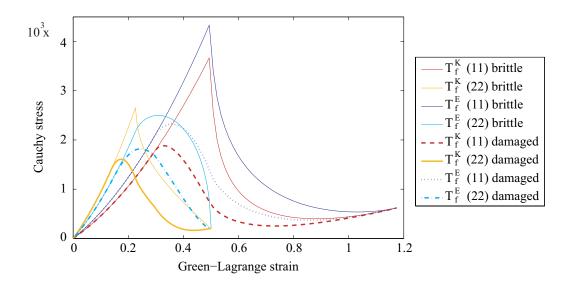


Figure 10: Components of the Cauchy stress tensor of the tissue for the K and E methods in the case of brittle fibers or damageable fibers, as functions of the corresponding components of **E**.

$$K \text{ method:} \begin{cases} \mathbf{H} = \frac{1}{\pi} \int_{A_0} \rho(\xi) t(\xi) \mathbf{n} \otimes \mathbf{n} \, \mathrm{d}\xi \\ \psi_f^K = \frac{1}{2} E \left(\mathbf{H} : \mathbf{E} \right)^2 \end{cases}$$
(47)

Fig.12 shows the evolution of the components of the stress tensor: at the 322 beginning of the loading, they are null since no fiber is stretched. The strain 323 in the fibers gradually increases to ε_t and the components of **H** increase to 324 their maximum values. Then, rupture including the uncrimping phenomenon 325 in our model leads to an unrealistic shape of the strain energy. Indeed, when 326 the fibers uncrimp, the strain energy increases because both the number of 327 stretched fibers and the amplitude of the loading increase. Then, once all the 328 fibers have uncrimped, the energy continues to increase only because of the 329 loading. Thus, the slope of the energy curve is more gentle once all the fibers 330

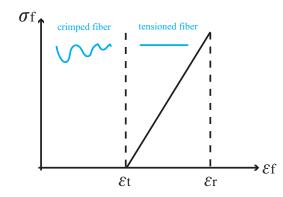


Figure 11: Behavior of a fiber taking uncrimping into account.

are stretched than when some are still crimped. When the fibers uncrimp, the 331 global behavior of the tissue actually "catches up" with the behavior without 332 taking uncrimping into account, leading to steeper slopes of the energy and 333 stress curves. This is a limitation of the model: if all the fibers uncrimp 334 before damage begins, it leads to an energy trend which is never observed 335 experimentally. In order to obtain the "toe region" which is often observed 336 experimentally and which is attributed to the uncrimping of the fibers, only 337 a fraction of the fibers must be stretched when damage begins; this could be 338 obtained using a random distribution of the fibers' crimping, which is not in 330 the scope of this paper as it requires a two-scale computation. 340

341 6. Conclusions

The objective of this work was to compare the response of two homogenization methods, available in the literature, in the case of damaging fibers within a fibrous membrane. For that purpose we propose a very simple anisotropic damage model able to account for several phenomena observed

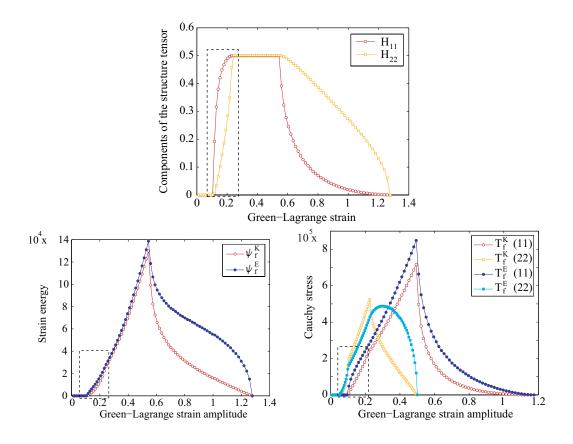


Figure 12: Evolution of the structure tensor, the strain energy and the components of the Cauchy tensor under biaxial tension for initially crimped fibers ($\varepsilon_t = 0.15$, $\varepsilon_r = 0.5$). The box emphasizes the area where the obtained behavior is not realistic.

346 experimentally.

The evolution of macroscopic damage based on fiber fracture at the microscopic scale is described using two scalar variables. The explicit expression of these variables as functions of the material parameters and of the macroscopic strain loading enables a rapid implementation of the fracture behavior law of the tissue. Besides, this paper does not deal with any phenomenon which would require reverting to the microscopic scale.

The main limitation of this model is that it considers the tissue's damage to be borne by the fibers alone independently of the components of the tissue (namely the matrix) with the assumption that the fibers are isolated from one another. This leads to a drastic nature of the constitutive law, that could be tone down by introducing fiber interactions, a distribution of the fibers properties or a possible decohesion between the fibers and the matrix.

The implementation of the two homogenization methods enabled us to 359 compare them: in the case of an isotropic tissue, the difference between the 360 two methods appears during damage and is due to the different formulation of 361 the tissue's macroscopic strain energy. The mathematical expression of this 362 difference makes us believe that a heterogeneous angular distribution of the 363 fibers (for an initially anisotropic tissue) would reveal a disparity between the 364 two methods right from the elastic zone. Another fundamental difference is 365 that the K method cannot take into account the behavior on the fibers' scale 366 because the constitutive law is applied directly to the macroscopic tissue. 367

Depending on the homogenization method, the principle of the identification of the damage model is different: indeed, the E method based on the strain energy of a fiber requires knowing the behavior and fracture strain

of a fiber; these parameters can be determined independently of the fibrous 371 tissue (see, for example, (Sasaki and Odajima, 1996; Svensson et al., 2010) 372 for collagen fibers). Nevertheless, while the response of the fibrous tissue can 373 be determined *a priori*, the response of the matrix is still to be identified. 374 In the case of the K method, the constitutive law is applied to the tissue 375 rather than to the fibers; consequently, the identification of the macroscopic 376 parameters of the tissue is required both for the fibrous tissue and for the ma-377 trix; only the fracture strain of the fibers can be determined experimentally 378 a priori. In practice, the identification of material parameters associated 379 with damage requires being able to control the evolution of damage so that 380 rupture does not to occur too rapidly. This constitutes the main difficulty of 381 the experimental characterization of damage. 382

The model proposed in this study can be applied to initially isotropic membrane tissues. Therefore, the prospective applications of this study concern the identification of the model's parameters in practical cases. In the case of biological tissues, the fracture modeling of the liver capsule could enhance a model of the human body and improve the prediction of liver injuries during an impact.

³⁸⁹ Appendix A: Details about the description of damage

The objective of this development is to define, for a radial loading with an amplitude greater than the bounds defined in Eq.(33), the angular sectors in which the fibers break. This corresponds to the resolution of $\varepsilon_f - \varepsilon_r > 0$. In the general case, since $\sin^2 \xi = 1 - \cos^2 \xi$:

$$\cos\varphi\cos^2\xi + \sin\varphi \left(1 - \cos^2\xi\right) > \frac{1}{k} \tag{48}$$

$$\cos^2 \xi \ (\cos\varphi - \sin\varphi) > \frac{1 - k \sin\varphi}{k} \tag{49}$$

Two cases need to be distinguished:

$$\varphi \in \left[0, \frac{\pi}{4}\right] \Rightarrow \cos\varphi - \sin\varphi \ge 0 \Rightarrow \cos^2 \xi > \frac{1 - k \sin\varphi}{k \left(\cos\varphi - \sin\varphi\right)} \tag{50}$$

$$\varphi \in \left[\frac{\pi}{4}, \frac{\pi}{2}\right] \Rightarrow \cos\varphi - \sin\varphi \le 0 \Rightarrow \cos^2 \xi < \frac{1 - k \sin\varphi}{k \left(\cos\varphi - \sin\varphi\right)}$$
(51)

394

In the following, only the case $\varphi \in \left[0, \frac{\pi}{4}\right]$ is explained.

These equations hold provided that the right-hand side is positive, *i.e.*:

$$1 - k\sin\varphi \ge 0 \iff k \le \frac{1}{\sin\varphi} \tag{52}$$

Therefore, one can write:

$$\cos\xi > \sqrt{\frac{1 - k \sin\varphi}{k \left(\cos\varphi - \sin\varphi\right)}} \,\forall \, k \in \left[\frac{1}{\cos\varphi}, \frac{1}{\sin\varphi}\right] \tag{53}$$

This expression is valid if the second term (which is positive) is between on and 1, which is verified for $k \ge \frac{1}{\cos \varphi}$.

³⁹⁷ Finally, the damage in the tissue can be described as follows:

$$\forall k \in \left[\frac{1}{\cos\varphi}, \frac{1}{\sin\varphi}\right] : \cos\xi \ge \sqrt{\frac{1 - k\sin\varphi}{k\left(\cos\varphi - \sin\varphi\right)}} \tag{54}$$

398

For $\xi \in A$, the solutions of inequality $\varepsilon_f(\xi) - \varepsilon_r > 0$ are:

$$\varphi \in \left[0, \frac{\pi}{4}\right] : \begin{cases} \xi \leq \arccos\sqrt{\frac{1-k\sin\varphi}{k\left(\cos\varphi - \sin\varphi\right)}} \\ ou \\ \xi \geq -\arccos\sqrt{\frac{1-k\sin\varphi}{k\left(\cos\varphi - \sin\varphi\right)}} \\ \xi \geq \arccos\sqrt{\frac{1-k\sin\varphi}{k\left(\cos\varphi - \sin\varphi\right)}} \\ ou \\ \xi \leq -\arccos\sqrt{\frac{1-k\sin\varphi}{k\left(\cos\varphi - \sin\varphi\right)}} \\ \zeta \leq -\arccos\sqrt{\frac{1-k\sin\varphi}{k\left(\cos\varphi - \sin\varphi\right)}} \end{cases}$$
(56)

This leads to the definition of the two angles ξ_1 and ξ_2 described in the paper.

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