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1 Abstract

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3 This study is focused on experimental analysis of the flow patterns inside scraped surface heat exchanger (SSHE) under isothermal and continuous flow conditions. Experimental flow 4 5 pattern analyses are the basis for further experimental explorations of mixing and heat 6 exchange mechanism. The 2D particle image velocimetry (PIV) is used to characterize the 7 flow structure. Synchronization between the laser sheet and blades rotation was achieved in 8 order to allow a statistical analysis of the obtained velocity vectors. Axial, radial and 9 tangential velocity components have been examined. Newtonian and non-Newtonian shear 10 thinning fluids are examined. The statistical results based on the calculation of the relative 11 standard deviation show that the error made in the experiments is less than 2%. The analysis 12 of the axial velocity inside SSHE shows the presence of very large axial heterogeneities. At 13 the entrance and exit of shearing region, axial velocity can reach 10 times the velocity 14 corresponding to the imposed inlet mass-flow. These results illustrate the importance and 15 extent of flow reversal for two used fluids and for N>4-5 rev/s. Tangential velocity dominates 16 around blades, and the maximum is reached near the tip of the blade. In most cases, the 17 obtained velocities results near the walls shows a violating no-slip condition, which is due to 18 the high velocity gradient near to the wall, and not captured by PIV. The radial velocity 19 revealed a poor radial mixture near the rotor wall and the best mixing is localised on both 20 sides of the blades. The results showed that under certain experimental flow conditions of 21 rotating velocity and axial flow rate, a more effective flow can be obtained, which is 22 characterised by low axial velocity dispersion. The obtained results in this work will be used 23 to validate a 3D numerical model.

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Key words: Mixing fluids, non-Newtonian, PIV flow measurement, fluid structure, scraped
 surface heat exchanger.

Nomenclature

d	stator diameter (m)	u.*	μ μ μ ρ R
d.	rotor diameter (m)	*	normalized radial velocity $= u_r / S_r r_r$
u _r		u ₀ *	normalize tangential velocity $= u_{\theta} / 22.K_r$
D_{p}	mean diameter of seeded particles (μ m)	Vz	normalized axial velocity = u_z / U_d
i	coordinate in y-direction	V_b	volume of bowls (l)
j	coordinate in x-direction	\mathbf{V}_{t}	total volume of SSHE (1)
k	frame number in PIV measurements	$\mathbf{V}_{\mathbf{Z}}$	axial velocity (m/s)
Κ	consistency index (Pa.s ⁻ⁿ)	x,y,z	cartesian coordinate (m)
L	stator length (m)		
Ν	impeller rotational speed (rev/s)		
n	flow behaviour index	Greek	letters
Q_{v}	volumetric flow rate (m ³ /s)		
r	radial position (mm)	θ	angular position (°)
r*	normalized radial position $= (r - R_r)/(R_s - R_r)$	δ	relative standard deviation
R _r	rotor radius	ρ	density (kg/m ³)
R _s	stator radius (m)	γ	shear rate (s^{-1})
Re _{axg}	generalized Reynolds number	η	dynamic viscosity (Pa.s)
S	cross –section (m ²) = $\pi (R_s^2 - R_r^2)$	Ω	frequency speed $\Omega = 2\pi N (s^{-1})$
Т	temperature (°C)		
Tag	generalized Taylor number		
Tagc	critical Taylor number		
U_d	velocity based on axial flow rate " Q_v/S "	Y	
$V_{r_{\star}} V_{\theta}$	velocity components in cylindrical coordinate		
	(m/s)		

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1 **1. Introduction**

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3 Scraped surface heat exchangers (SSHE) are used in a wide variety of engineering 4 applications such as the heat treatment of various viscous fluids. SSHE is used in chemical, 5 pharmaceutical and food industries. The particularity of this device is to periodically scrape 6 the surface heat exchange, which allows an improvement in the heat transfer and avoids the 7 problems of the fluid fouling the exchange surface. Several studies have been carried out in 8 order to examine the fluid flow pattern, especially in order to explain the non-uniformity of 9 the temperature inside and at the exit of SSHE apparatus. The non-uniformity of the 10 temperature field and the flow pattern are likely to be coupled, which is the main cause of 11 deterioration of the heat exchanger efficiency. The flow in SSHE is the result of the 12 superposition of a Poiseuille-type flow (axial flow) and a Couette-Taylor-type flow (rotational 13 flow induced by the blades rotation). The presence of the scraping blades makes the fluid flow 14 more complex and does not allow simple experimental measurements using temperature or 15 velocity sensors, which are necessary for comprehension better understanding of the physical 16 phenomena taking place in these devices. In the previous works given by others, two 17 approaches were generally used to investigate the heat transfer and the fluid flow: 18 experimental and numerical ones. In experimental approaches the authors first focused their 19 studies on the determination of the residence time distribution (RTD) [1-4] which gives better 20 information on the axial flow dispersion and eventually the global diagnostic of the presence 21 of dead zones. The information obtained by RTD remains insufficient to describe the flow 22 structure inside the SSHE. Velocity profiles using non intrusive methods such as (LDV, IRM 23 and PIV) were carried out by Wang et al. [5,6] and Stranzinger et al. [7, 8] but only in a 24 simplified geometrical configuration and in batch system (without axial flow rate). They 25 conclude that the velocity profile in the annulus greatly depends on the behavior index of the 26 shear thinning fluids. The flow regimes in SSHE can be identified by the Taylor number, and 27 a critical Taylor number is used to characterize the passage from a laminar flow towards 28 laminar vortical flow [9-12]. Dumont et al. [13] determined, using electrochemical technique, 29 a limit value of the critical Taylor number Tage=60 beyond which vortices appear, and another 30 critical value of 80 when helical vortices start. The theoretical critical Taylor number is equal 31 to 41.3, the difference between the theoretical and SSHE value is explained by the presence of 32 the blades, in fact the presence of the blades delays the onset of Taylor vortices. 33 Numerical investigations have also been made in order to investigate the heat transfer and the

fluid flow in SSHE [14-18]. Numerical results show the existence of zones with recirculation

upstream of the scraping blade [14]. According to the various studies carried out, the flow
pattern in a *SSHE* remains unknown. In order to give a better understanding of the flow
pattern in the *SSHE*, velocity measurements using Particle Image Velocimetry (PIV) were
done.

5 Although PIV had its genesis several years ago, its more common usage occurred in the last 6 decade. Owing to the ability of PIV, in principle, to provide information about a flow field, 7 simultaneously in different points, its use is sure to become even more widespread. PIV is an 8 instantaneous whole-field measurement technique, which uses pulsed light laser to illuminate 9 a fluid flow seeded with tracer particles. It can potentially provide more information about the 10 flow mechanism than the conventional technique. The bibliography of Adrian [19] gives an 11 indication of the historical development of PIV, while different PIV techniques are described 12 in the studies made by Adrian [20] and Raffel et al. [21]. Several recent studies [22-24] have 13 demonstrated the application of PIV to complex flows. An important aspect of PIV is its 14 ability to measure the vorticity field and spatial correlations [25, 26]. To some extent, PIV 15 still suffers limitations in the dynamic range of velocity measurement, though significant 16 improvements have been made [27]. Other limitations on the quality of data arise due to the 17 size of seed particles, as well as image quality and size, camera frame rate, processing 18 software, etc.

Despite a number of studies on SSHE hydrodynamics, there have been no studies reported in
the open literature to use PIV in continuous flows conditions in order to obtain information
about the flow fields in this type of equipment.

22 The originality of this work is to use the PIV technique to study the fluid flow inside SSHE 23 under isothermal and real flow conditions. In order to understand the flow structure in more 24 details, PIV technique flow measurement was used. The (r,z) and (r,θ) velocities components 25 were obtained for Newtonian and non Newtonian shear thinning fluids. The results could give 26 the time evolution of the spatial distribution of the velocity. In order to reach these objectives, 27 a series of velocity vectors graphs and profiles at different cross sections inside the SSHE are 28 presented in the case of Newtonian and non-Newtonian shear thinning fluids for different z 29 locations.

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1 2. Experimental Setup and Approach

2 2.1. Experimental Devices

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The PIV technique is a non intrusive flow measurement. The principle of PIV is to illuminate seeded particles added into the fluid with laser sheet in a cross-section. The seeded fluid flow particles are captured by a CCD camera. To have access to the cross-sections inside the device, it was necessary to have transparent walls and fluids.

8 Fig.1 shows the schematic drawing of experimental system, the all dimensions of the pilot9 SSHE are given in Table 1.

The fluid is pumped through the inlet bowl via the volumetric pump with limits ranging from 5.510^{-6} to 5.510^{-5} m³/s and flows back to the storage tank, with volume capacity of 30 l. An electromagnetic flowmeter (Promg 33 type) is placed at the exit of the exchanger, and gives the flow rate in kg/s. The rotating velocity of the rotor (thus of the blades) is regulated with an electronic variator with a range of speed of 1 to 10 rev/s.

15 The particle image velocimetry (PIV) is used to provide instantaneous velocity field 16 measurements in a longitudinal and vertical cross-section of the exchanger. A laser light flash 17 sheet illuminates particles in the flow field and a digital camera captures the two points in 18 time. The resulting flow velocity extracted from this couple of digital images is an 19 instantaneous snapshot of the flow in the area viewed, in stationary conditions. The seeding 20 (PSP, 50 µm diameter, 1.03 g/cm³) was adjusted to 5-10 particles per interrogation window. 21 The PIV system used for the determination of the velocity fields is essentially composed of a 22 CCD camera of 1600x1186 pixels resolution (Dantec Dynamics Flow, Sense M2/E 8 bit) with 23 a Nikon objective (60 mm), a pulsed Nd-Yag 15 mJ laser (New Wave Solo). The PIV 24 recordings from the camera are acquired with the PIV software package FlowManager from 25 Dantec Technology. For these images, a cross-correlation analysis method is used with interrogation, window size of 64x64 pixels², which corresponds in the light-sheet to an area of 26 27 approximately $0.8 \times 0.8 \text{ mm}^2$ and an overlap of 50% on the final pass. For the two frames, 28 cross-correlation and fast Fourier methods are applied in the data processing. For each 29 measurement, 100 image pairs are used and the statistical average vector is calculated in order 30 to obtain the time averaged for this flow field. In order to eliminate optical distortion caused 31 by the curved external surface of the SSHE (cylindrical geometry), a full parallelepiped 32 Plexiglas was dug with the cylindrical tube representing the exchanger stator (Fig.2). The 33 presence of the blades in the annulus requires the installation of a synchronization system. 34 The main objective is to always obtain the same position of the blade for any laser impulsion,

thereby allowing one to apply statistical calculations on the obtained velocity vectors. The synchronization is ensured by a photodiode which delivers a signal for every scraper blade revolution towards the laser controller. The image synchronization was based on the assumption that the scraper blade keeps a constant rotational speed.

5 PIV measurements are carried out under the conditions gathered in Table 2.

6 These operating conditions are selected in order to cover flow conditions and to know the 7 influence of each of these parameters such as the flow rate and the rotating velocity on the 8 velocity profiles. They also are selected according to the flow conditions used in food 9 industries. The dimensionless numbers (Reynolds and Taylor numbers) presented in Table 2 10 are generally used to characterize fluid flow regime in SSHE. According to Table 2, we notice 11 that the flow regime in all considered operating conditions is laminar. In addition, for a Taylor 12 number superior to 60 it is possible to have secondary flows, called Taylor vortices [12].

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14 2.2. Working Fluids

15 In this study, two fluids were examined, a Newtonian and a non-Newtonian shear thinning. 16 The study aims to examine the flow pattern of liquid food products; the two selected fluids 17 must be close to those treated in scraped surface heat exchangers. Glycerin has a high 18 viscosity (1.31 Pa.s at 18°C) and 1% CMC solution was a non-Newtonian shear thinning 19 fluid, with rheological behavior similar to yoghourt. The rheological behavior of the working 20 fluids is measured by using a standard rheometer (AR1000) with Couette geometry under the 21 shear rate between 1-2500 1/s and at constant temperature of 18°C. The non-Newtonian fluid 22 used was an aqueous CMC solution (Carboxyl-Methyl-Cellulose), which was a shear thinning 23 fluid and was not perfectly transparent like glycerin. The use of a not perfectly transparent 24 fluid did not prevent its use, but there was a limit depth beyond which the CCD camera of 25 PIV could not capture the particles displacement of the illuminated laser plane (for 1% CMC 26 this limit depth was about 15 cm against 40 cm for glycerin); this depth limit concerns (r, θ) 27 planes. The density of the fluids was measured with a densitometer. Table 3 gives the results 28 of the rheological and density measurements data for fluids used. In the experiment and in 29 order to make sure that the working fluid was not degraded by the action of the blade 30 (shearing effect) or by any other experimental factor (dilution, light,...etc) the measurements 31 of viscosity and density were systematically checked after each experiment test. If the fluid 32 presented a change in viscosity, it was replaced by another preparation for further 33 experimentation. In general no high fluid degradation was observed.

1 2.3. Statistical velocity fields analysis

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3 The collected instantaneous velocity vector maps taken from a regular grid of points are 4 obtained after a correlation processing the particle image data. For each test cross-section, 100 5 frames of transient velocity field were obtained in the experiment. The time averaged velocity 6 field of each cross section test was then obtained through continuous processing. The average 7 time value of the velocity components is given by:

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$$\overline{u}_{i,j} = \frac{\sum_{k=1}^{100} u_{i,j,k}}{100}$$
 (1)
9 $\overline{v}_{i,j} = \frac{\sum_{k=1}^{100} v_{i,j,k}}{100}$ (2)

10 Where u_{i,j,k} is the velocity component in the x-direction of the measured point (i,j) in the frame k and $v_{i,j,k}$ is the y-direction velocity component of measured point (i,j) in the frame k. 11 12 The standard deviation value of velocity components is then defined as:

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$$\Delta u_{i,j} = \sqrt{\frac{\sum_{k=1}^{100} (u_{i,j,k} - \bar{u}_{i,j})^2}{99}}$$
 (3)
14 $\Delta v_{i,j} = \sqrt{\frac{\sum_{k=1}^{100} (v_{i,j,k} - \bar{v}_{i,j})^2}{99}}$ (4)

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16 The relative standard deviation, which is the standard deviation normalized by the average 17 time velocity vector magnitude, will be the most convenient measure to be used and it is 18 indeed used by several authors [28, 29] to quantify the errors generated in the experiments. It 19 is defined by eq(5) where grid indices are dropped for simplicity:

20
$$\delta = \frac{\sqrt{\Delta u_{i,j}^2 + \Delta v_{i,j}^2}}{\sqrt{\frac{-2}{u_{i,j}^2 + v_{i,j}^2}}}$$
(5)

21 The use of the relative standard deviation will enable us to combine measurements errors 22 from different sources. When the flow regime is turbulent, δ represents the turbulence 23 intensity. In this study and under all considered operating conditions, the flow regime remains 24 laminar. The relative standard deviation here represents the uncertainty of the flow 25 measurements.

1 2.4. The selection of test sections

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3 The pulsed laser light sheet is not able to illuminate the whole test section, because the 4 physical dimension of the pilot system is large. The area of the measured velocity field captured by the CCD camera is limited to 90x80 mm². Due to this limitation and in order to 5 6 cover the whole geometry of the exchanger according to z-direction, three laser planes were 7 considered (Fig.3). Hence, in this configuration, axial and radial velocities can be determined 8 in the whole system. For the measurement of the radial and tangential velocities the CCD 9 camera cannot capture the image further than 13 cm deep due to the poor transparency of the 10 working fluids in the case of 1% CMC solution. In the case of glycerin, five (05) laser sheet 11 planes are examined corresponding to normalized z*- locations of 0.2, 0.3, 0.6, 0.9 and 0.97 12 (Fig.3),. To obtain all these planes, two transparent bowls are used: one at the entrance and 13 the second at the exit of the exchanger, thereby allowing the view of all presented planes.

Fig.2 shows the diagram of the laser and camera positions allowing us to obtain three velocity components. Fig.3 shows the different positions of the measured planes in both directions. By using only one CCD camera for the PIV acquisition, it is not possible to obtain the three velocity components in space at the same time, but it enables us to measure two velocity components at the same time. To obtain three velocity components (v_z , v_r , v_{θ}), two series of measurements were required.

The results obtained by PIV are expressed in a Cartesian plane. In order to represent the velocity components in the (r,θ) plane, a transformation is needed to pass towards polar coordinates. If v_x and v_y are the Cartesian velocity components in the reference plane of (x, y)and v_r , v_{θ} are the polar velocity components in the reference of (r,θ) , the transformation is given by:

$$25 u_r = u_x \cos\theta + u_y \sin\theta (6)$$

(7)

$$26 \qquad u_{\theta} = u_x \sin \theta - u_y \cos \theta$$

27 where θ is angular position.

The calculation of the mean velocity, standard deviation and all presented results were made
 using the computing language MatlabTM.

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31 3. Results and discussion

This study was focused on exploring experimental results obtained by PIV technique, in real flow condition of a rotating machine of SSHE. In this section, a statistical analysis of the obtained velocity vectors is first presented and analyzed. Secondly, when the obtained results
 are reliable, analyses of the obtained velocities vectors are given, and the flow structure in
 SSHE is discussed.

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5 3.1. Statistical analysis results

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7 In a typical PIV experiment, several double-frame images are captured and processed in order 8 to calculate instantaneous velocity fields. These data are directly applicable for the 9 computation time-mean statistics, such as mean and rms (root mean square) fields and several 10 derived quantities such as vorticity, shear rate and streamlines. The 2D-PIV data can also be 11 used to compose a 3D volume for the time-mean statistics by combining several individual 12 measurement planes. A visual examination of PIV data allows observation of the spatial 13 scales and the nature of coherent flow structures. However, a careful analysis of these features 14 is sometimes needed. A statistical analysis of the obtained vectors makes it possible to know 15 the average error made for the velocity measurement using the PIV. Calculation results are 16 presented here for working fluids mentioned in Table 2 and specifically in the case of N=3 rev/s with an axial mass flow rate of $1.38 \ 10^{-5} \ m^3/s$. Fig.4 shows an example of the 17 18 distribution of the relative standard deviation (δ) using equation (5) for the middle r-z plane of 19 the SSHE. Fig.4 shows that the relative standard deviation is less than 0.5 % for the four 20 considered fluids, which is an acceptable error. Globally, the relative standard deviation in 21 (r,θ) and (r,z) cross-sections planes of the time average velocity is less than 2% for all 22 experiments. An example of the same calculation of the relative standard deviation for $(r-\theta)$ 23 plane corresponding to the z location of $z^*=0.2$ is given in Fig.5. It can be noted that in the 24 case of $(r-\theta)$ plane the errors are mainly located in the zone of an existing shadow area created 25 by the blade. According to this statistical analysis, we deduced that the error of measurement 26 made by using PIV is not significant and that the results can be reliable. The presence of the 27 shadowed zone locally increases locally the relative standard deviation, which can reach a 28 maximum of 4.5% for 1% CMC solution and 2.5% of pure glycerin. Therefore, in this study, 29 we only present the velocity vectors corresponding to the half plane located between the 30 angular position of $0 < \theta < 180^\circ$ which corresponds to the relative standard deviation which is 31 less than 2%.

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- 1 3.4. Tangential velocity inside SSHE
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Fig.6 shows the obtained result of the contours velocity magnitude of $v_{\theta}^2 + v_r^2$ in the case of 3 pure glycerin. Before the fluid flowed to the sheared part ($z^{*}=0.16$), the flow had a similar 4 5 behavior to that of a Couette-Taylor, with a maximum velocity on the rotor and zero velocity 6 on the stator, with a linear velocity profile in the case of a Newtonian fluid. Once it was in the 7 sheared zone $z^*>0.16$ (presence of blades) the flow pattern completely changed. The 8 comparison between all obtained velocity contours for each plane shows a perfect similarity 9 of the flow configuration in the presence of the blades. There is an increase of the velocity 10 behind (wake) and in front of the blade. The same observations were obtained with 1% CMC 11 solution. The influence of the axial flow rate does not add a visible change to the contours.

12 Studies about the flow across SSHE were mainly handled by Wang et al. [5, 6, 37, and 38]. 13 They used two methods to determine the axial and tangential velocity components: an 14 analytical method to solve continuity and momentum equations and an experimental method 15 using MRI (Magnetic resonance Imaging). In their study, they used a simple geometry with 16 only one blade. The results of these authors were obtained in annular space far away from the 17 blades. A numerical study was also made by Stranzinger et al. [7, 39] and recently by 18 Yataghene et al. [18] based on the solving of continuity and momentum equations by using a 19 commercial software computation in 2D geometry. The results of the numerical computation 20 showed that the blades greatly affected tangential velocity, but no experimental validations 21 were available until now. The use of 2D numerical models does not allow the authors to 22 examine the axial velocity. In order to supplement these studies already carried out, we 23 suggest examining the tangential velocity profiles for different angular positions (θ). The 24 results given by Wang et al. [37] showed that the flow rate did not significantly affect the 25 tangential velocity profile, which was confirmed in this work. We examined in detail a plane 26 corresponding to $z^*=0.20$, where we can reasonably assume that the flow was mainly 27 controlled by the blades. The plane corresponding to $z^{*}=0.20$ is located exactly at the 28 entrance of the fluid in the sheared region with the presence of the blades (see. Fig.7). The 29 synchronization system allowed the measurement of the velocity by respecting the same 30 position of the blades for every passage. In the same manner, the tangential velocity 31 normalized by $(2\pi R_r N)$, where R_r represents the rotor radius, was plotted.

32 An example of a seeded flow of the cross-section of $z^*=0.20$ is presented in Fig.7 (a), where 33 the position of the two blades are visible. Fig. 7 (a) also shows that, behind the rotor, a

shadowed zone was created by the blades laser reflexion. For this reason, only the velocity
profiles included between the angular position of 0 and 180° are presented in Fig. 7 (b).

3 The results of the obtained normalized tangential velocity corresponding to the different

angular positions as shown in Fig 7 (b) are given in Fig.8 for the case of the pure glycerin and 1% CMC solution under the flow conditions of N=3 rev/s and $Q_{\nu}=1.38 \ 10^{-5} \ m^3/s$. The positions of $\theta=0^{\circ}$ and $\theta=180^{\circ}$ correspond to a slipstream behind the 1st and the 2nd blade, $\theta=150^{\circ}$ corresponds to the front of the scraping blade, and the other angular positions of $\theta=60^{\circ}, \theta=90^{\circ}$ and $\theta=120^{\circ}$ correspond to the annulus space. The position r*=0 correspond to the rotor wall and r*=1 correspond to the stator wall.

10 The tangential velocity profiles were completely deformed in the presence of the blades, the 11 maximum of the tangential velocity was obtained near the stator located at a distance from 12 $\frac{3}{4}$.r* which corresponds to $1.6 \times \Omega R_r$.

13 In addition, it has been noted that the condition of no-slip is not respected on the stator wall in 14 the case of the passage blade: the passage of the blade completely destroyed the 15 hydrodynamic boundary layer developed on the wall of the stator (Fig.7 (b) for the different angular positions corresponding to $\theta=0^\circ$, 150° and 180°). It also appears in Fig.8 (a,b) that 16 17 near the rotor wall velocity is not zero. This can be explained by the low accuracy of the PIV 18 near the walls, and the rotation of the rotor effect. For the same angular θ -positions of θ =0°, 19 150° and 180°, the tangential velocity at the stator wall is not null; this is perhaps due to the 20 blade entrainment at this wall, which remove the boundary layer. It is also highly probable 21 that no-slip condition is verified but there is a high velocity gradient near to the wall which 22 was not captured by the PIV.

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25 3.2. Axial and radial velocity inside SSHE

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The axial and radial velocity profiles under considered experimental conditions were studied. Two fluids were used, a Newtonian one of pure glycerin, and a 1% CMC solution. Two axial flow rates conditions were examined. The selected axial flow rate was 0.017 and 0.051 kg/s for glycerin, and 0.014 and 0.043 kg/s for 1% CMC solution. Three rotating velocities of 1, 3 and 5 rev/s were investigated. For all these operating conditions the flow remained laminar without Taylor vortices (i.e. Table 2), as already mentioned in the introduction. The most authors who are working on the SSHE compare usually the axial velocity with this of

1 Couette- Poiseuille flow. The axial velocity profile in the case of Couette-Poiseuille is given by $\frac{v}{v_0} = \frac{3n+1}{n+1} \left(1 - \left(\frac{r}{R}\right)^{\frac{n+1}{n}} \right)$, where n represents the flow index behavior of the power law 2 3 fluid. In the case of a Newtonian fluid (n=1) the profile was strictly parabolic, and when 0 < n4 <1 the profile became increasingly plate with decreasing of the index flow behavior n. In the 5 classical case of Couette-Poiseuille flow the axial velocity profile is parabolic for Newtonian 6 fluid, with the maximum axial velocity equal to twice the mean velocity corresponding to the 7 imposed axial flow rate [6, 31]. Wereley and Lueptow [32] have shown that even for a Taylor 8 vortices flow regime, the averaged velocity profiles were remarkably similar to the stable 9 theoretical parabolic velocity profile. These results clearly indicate that, in the case of 10 Couette-Poiseuille flow, the vortical flow regime has only a very small effect on the axial 11 velocity profile and that the effect is largely independent of the characteristics of the vortical 12 flow [32]. This result is particularly interesting when the effect of the Taylor vortices occurs. 13 Fig.9 (a,b) shows an example of the velocity vectors projection on the (r,z) plane, in the case

of the axial flow rate of $1.38 \ 10^{-5} \ m^3/s$ and for rotating velocity of N=5 rev/s, in the entrance 14 15 and the exit planes of the exchanger. The experimental investigation on the flow structures 16 conducted by Härröd [9] in SSHE has shown helical behavior. The same flow pattern as the 17 one observed by Härröd [9] was obtained in the middle of the exchanger, and far from the 18 entrance and exit SSHE-planes, where the flow presents an important change of the axial 19 velocity in the z ranged between 70<z (mm) <110 (0.16<z*<0.25). The reversed flow was 20 also observed under certain flow conditions of axial flow rate, rotating velocity and type of 21 fluids. For glycerin solution, the reversed flow was especially important in the region defined 22 by: 410<z (mm) <440 (0.93<z*<1), as showed in Fig.9 (a,b).

According to the observations obtained in this work, the flow structure or the general tendency of the flow pattern in the r-z plane remains helical, except at the entrance and exit plane, where the axial velocity is highly influenced by the rotating velocity and imposed axial flow rate.

After analyzing the different velocity vectors results, it should be noted that for a higher rotating velocity (N>5 rev/s) and under a low axial flow ($<1.38 \ 10^{-5} \ m^3/s$), there is a reverse flow at the exit of SSHE.

30 To quantify this flow tendency, axial velocity profiles were plotted in Figs. 10 & 11 for 31 various z*-locations of 0.2, 0.3, 0.6, 0.9 and 0.97, and for θ =90° under all considered 32 operating conditions. These z* locations correspond to the limits between the entrance and 33 exit of the fluid in the sheared zone (presence of the blades) respectively. The other working

1 fluids (0.5% CMC solution or 15% water glycerin) are not presented here, but results are 2 commented. Fig.10 shows the results of the normalized average axial velocity (v_z^*) profiles 3 with dimensionless radial position (r*) of pure glycerin.

4 To demonstrate the existence of the reverse flow at the exchanger exit, a criterion based on 5 the axial and radial velocity components should be defined. For example, in the case of the glycerin fluid, and for $v_z(z=1, \theta=90^\circ, r=0.25)$ and $u_r(z=0.9, \theta=90^\circ, r=0.25)$ and except 6 7 for 1 rev/s, the negative sign observed on the velocity given in Figs. 10 & 11 reflects the 8 reverse flow character in the axial flow direction. In the great majority of the obtained axial 9 velocities profiles, it can be seen for the low rotating shaft velocity (N=1 rev/s), that the 10 profiles are almost parabolic with a maximum ranged between 2 and 2.5. Important changes 11 on the axial velocity are observed with increasing the blade rotation.

12 Due to the significant differences of obtained results between the glycerin and 1% CMC 13 solution, it is difficult to discuss both fluids at the same time. Hereafter, we will first discuss 14 first the results of the glycerin and then 1% CMC solution.

15

16 According to the obtained results, four observations can be given:

17 The axial velocity v_z^* is highly influenced (Fig. 10) by the rotating velocity: 18 $v_z^*(z^*=0.19, \ \theta=90^\circ, r^*=0.5, N=5 \ rev/s, \ Q_v=1.38 \ 10^{-5} \ m^3/s) =11.8$. The maximum axial 19 velocity measured is about 10 times higher than the velocity of imposed axial flow 20 rate at inlet plane. The increase of the rotational speed promotes axial velocity 21 heterogeneity at the entrance of the exchanger, for example: $N=5 \ rev/s, \ v_z^*(z^*=0.19, \ \theta=90^\circ, \ r^*=0.5, \ Q_v=1.38 \ 10^{-5} \ m^3/s) =11.8 \ and \ N=3 \ rev/s, \ v_z^*(z^*=0.19, \ \theta=90^\circ, \ r^*=0.5, \ Q_v=1.38 \ 10^{-5} \ m^3/s) =6.2.$

24

- The increase of the rotating velocity facilitates the appearance of a preferential way of
 the fluid near to the rotor. The channelling phenomena can be explained by the action
 of the axial pressure induced by the rotating speed of the rotor.
- For N=5 rev/s reverse flow at the exit is important, it can reach almost 10 times the velocity of the imposed axial flow rate: $v_z^*(z^*=1, \theta=90^\circ, r^*=0.25, Q_v=1.38 \ 10^{-5} \ m^3/s,$ $N=5 \ rev/s) \approx -10$ (Figs. 10&11).
- 32

1 The increase in the axial flow rate tends to stabilize the axial flow heterogeneity in 2 SSHE, especially at the entrance and exit of the heat exchanger. For example, according to data given in Fig.11: $v_z^*(z^*=0.97, \theta=90^\circ, r^*=0.25, Q_v=1.38 \ 10^{-5} \ m^3/s,$ 3 $N=5 \ rev/s$) $\cong 12 \ and \ v_z^*(z^*=0.97, \ \theta=90^\circ, \ r^*=0.25, \ Q_v=4.2 \ 10^{-5} \ m^3/s, \ N=5 \ rev/s) \cong 5.$ 4 5 At the exit plane of the exchanger, the influence of the axial flow rate plays an important role to considerably reduce the reverse flow: $v_z^*(z^*=0.97, \theta=90^\circ, r^*=0.25, \theta=90^\circ)$ 6 $Q_{\nu}=1.38 \ 10^{-5} \ m^3/s, \ N=5 \ rev/s) \cong -10 \ against \ v_z^{*}(z^{*}=0.97, \ \theta=90^{\circ}, \ r^{*}=0.25, \ Q_{\nu}=4.2 \ 10^{-5}$ 7 m^3/s , N=5 rev/s) $\cong -1$. 8

9

Fig.11 (a,b) shows the normalized axial velocity component under considered operating flowconditions, in the case of non-Newtonian shear thinning solution of 1% CMC.

12

13 The velocity profile shows the complexity of the fluid flow. The acceleration previously 14 observed in the case of glycerin was also obtained for 1% CMC solution, with a maximum 15 velocity 12 times greater than the mean velocity. At the exit of the exchanger, no important 16 reverse flow was obtained. For $z^*=0.3$, a major change in the axial velocity profiles was 17 observed in the case of N=3 & 5 rev/s. For example, the axial velocity profiles of $v_z^*(z^*=0.3,$ 18 $\theta=90^\circ$, r^* , $Q_v=4.2 \ 10^{-5} \ m^3$ /s, $Q_v=1.38 \ 10^{-5} \ m^3$ /s, $N=3 \ rev/s$, $N=5 \ rev/s$) present (Fig. 11) 19 important deformations.

The flow complexity can be explained by the shear rate depending fluid. The axial velocity profiles were not parabolic like pure glycerin, and were much more influenced by rotating velocity.

So far, the experimental analysis of the radial velocity in a SSHE has never been handled. This study allows us to know the effectiveness of the radial mixing in the SSHE. With *r*-*z* planes, we can access to single radial velocity profile, but at different z-locations, and for (r, θ) planes it was possible to have flow visualisation of cross-section inside SSHE. Note that the radial velocity profiles obtained by the two planes (r, z) and (r, θ) were similar.

Fig. 12 (*a,b*) shows the radial velocity profiles normalized by $(2\pi R_r N)$ in each considered (*r,z*) plane corresponding to *z**-locations of 0.2, 0.3, 0.6 and 0.97, which represent the entry, the middle and the exit of the sheared zone respectively. Three rotating velocities are compared (*N*=1, 3 and 5 rev/s, with a flow rate of $Q_v=4.2 \ 10^{-5} \ m^3/s$). For Newtonian fluid, the radial velocity profiles for all the considered conditions show an identical general shape (Fig. 12). For the exit plane the presence of the reverse flow has an influence on the radial

1 velocity which is visible on Fig.12. In the case of 1% CMC solution, the general layout of the 2 obtained radial velocity strongly depends on the operating conditions (such as rotating 3 velocity and axial flow rate). This dependence can be explained by the non Newtonian 4 behaviour of the fluid which is influenced by the shearing generated in the fluid. At the exit of the exchanger with z*-location ranging between $0.77 < z^* < 1$ and in the case of the two 5 considered fluids, the radial mixture is poor and corresponds to 5% of $2\pi R_r N$. The reverse 6 7 flow phenomenon at the exit plane is not observed for 1% CMC solution under a rotating 8 velocity of N=1 rev/s. The radial velocity in the case of 1% CMC solution is more significant 9 than in the case of pure glycerin, which can reach for a velocity of N=5 rev/s 40% of 10 $(2\pi R_r N)$ against 15% $(2\pi R_r N)$ for pure glycerin. It is noted that the increase in the rotating 11 velocity in the case of 1% CMC solution increases the effectiveness of the radial mixture, but 12 also the presence of a preferential way close to the rotor. In the case of flow rate of $Q_v=1.38$ 10^{-5} m³/s and under the same operating conditions as previously, it is shown that the lower the 13 14 flow rate, the better radial mixture.

In addition, Fig.13 (a,b) gives the radial velocity profiles for the two considered operating flow conditions. It can be seen that the velocity curves greatly depend on the blades position, and near the blades the radial velocity becomes more important. The profile near the blade is parabolic, and far from the blade the radial velocity is almost zero.

Fig.14 shows an example of radial velocity distribution in the cross-section located at $z^*=0.6$. For all examined operating conditions and fluid type, a similar radial velocity distribution is obtained. The most important radial mixing is located around the blades. The blades periodically scrape the fluid contained in the handled sheared volume, and a layer without mixing is present near the rotor (Fig.14). Outside the blades zones the radial velocity remains almost null, which leads to very poor radial mixing.

25

26 5. Conclusions

27

As flow patterns are important in understanding the mixing and heat transfer mechanisms in the scraped surface heat exchanger, the experimental PIV technique was used to measure different velocities components; such as radial, tangential, and axial ones in real flow conditions. In the food industries that use the SSHE, the main fluids are non Newtonian shear thinning, e.g. yoghurt, cheese melt, mayonnaise, etc. This study is the first one that analyses three velocities components (v_r, v_θ, v_z) on different cross-sections inside the SSHE. One

1 reason for this lack of data is the complexity of the experimental configuration - several 2 arrangements had to be made, such as the synchronisation system which is needed to obtain 3 the average velocity. A statistical analysis was carried out to see the degree of confidence of 4 the obtained PIV results. The statistical parameter which allowed us to identify errors of 5 measurement was the standard deviation, and it proved that the presence of the shadowed 6 zones created by the blades made the particles less visible with the camera, which gives 7 erroneous vectors. Therefore, these zones were excluded from the study. The study of axial 8 velocity under large operating conditions showed strong axial velocity heterogeneities, for N>4-5 rev/s and for a flow rate of $Q_v = 1.38 \ 10^{-5} \ m^3/s$, which can reach 10 times the velocity of 9 10 the imposed flow rate. These conditions show the existence of a large axial dispersion. The 11 increasing in the flow rate stabilizes the flow. Under the above operating conditions, there are 12 very significant axial velocities close to the rotor, which allows assuming the presence of two 13 layers of fluids that run out at two different speeds in simple shearing. The analysis of radial 14 velocity gives information on the radial mixture. After the examination of the obtained 15 vectors in the (r,θ) planes, it seems that the radial mixture is mainly imposed by the blades, 16 and is more effective on both sides of the blades, and null in the remainder of the exchanger. 17 The radial mixture is poor, which can generate significant differences on the quality of the 18 heat treatments inside the SSHE. With regards to the tangential velocity components, the fluid 19 contained in the presence of the blades is controlled by the tangential velocity; the maximum 20 speeds can reach $1.6.u_{\theta(rotor)}$ close to the tip of the blade. The flow rate has no influence on the 21 tangential velocity; this result is not given in this study, but it confirms the one given by 22 Wang et al. [6]. The condition of no slip at the wall is not to be respected in the case of 23 passage of the blade close to the stator characterized by a non null velocity. It is highly 24 probable that no-slip condition is verified but there is a high velocity gradient near to the wall. 25

26 The presence of the blades completely changes the velocity distribution. The flow in a SSHE 27 is very complex, and highly depends on operating conditions. The origin of the temperature 28 heterogeneities observed in these apparatuses strongly depends on the axial dispersion and on 29 the weakness of radial mixture. Improvements for these devices are necessary to allow a good 30 mixing process, and therefore the heat treatment of the product. To get further explanations on 31 the achieved axial velocity profiles, other measurements should be undertaken to fully explain 32 the observed phenomena at the inlet and outlet of the exchanger, together with an analysis of 33 the velocity gradient based on these results.

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1 <u>Tables list</u>

3 Table 1 Apparatus dimensions

Table 1 Apparatus unitensions

	Dimensions of
	used scale down
	model SSHE
Stator radius R _s	0.0325 m
Stator length (L)	0.40 m
Rotor radius (R _r)	0.020 m
Rotor plate radius (R _{rp})	0.0168 m
Total length with blows (Lt)	0.530 m
Area of cross-section (S)	2.10^{-3} m^2
Total volume of SSHE (Vt)	10^{-3} m^3
Bowl volume (V _b)	$1.25 \ 10^{-4} \ m^3$
Ratio (V_t/V_b)	4

9 Table 2: Operating Conditions

	Flow rate (l/h)			Rotating velocity (rev/s)		
	0	50	150	1	3	5
	$\operatorname{Re}_{axg} = 8^1$	$-n\left(\frac{4n}{3n+1}\right)\frac{\mu}{2}$	$\frac{\partial U_d^{2-n} d_h^n}{k}^{**}$	$Ta_g = \sqrt{\frac{R_s}{N}}$	$\frac{\overline{-R_r}}{R_r} \frac{\rho d_h^n}{2^n} \left(\frac{1}{2^n}\right)$	$\frac{(\Omega R_r)^{2-n **}}{k}$
Pure Glycerine	0	0,16	0.49	1.19	3.58	5.97
Glycerine 15% [*]	0	1.87	5.60	10.10	30.30	50.50
CMC 1%	0	0.024	0.13	0.66	4.07	9.50
CMC 0.5 %	0	0.32	1.52	5.37	25.86	53.69

10 * 15% volume of water

11 ** For Newtonian fluid n=1

2 **Table 3** : Fluids Properties

Fluids	Viscosity at 18°C	Density	
		(kg/m^3)	
	(Pa.s)		
Pure glycerin	1,31	1260	
Glycerin 15 % of water	0.11	1220	
	(Pa/s ⁿ) n [-]		
CMC 1%	12.83 0.34	1048	
CMC 0.5 %	0,65 0.57	1043	

Figures list: Level of the fluid sheared zone Ρ 1- Fluid tank 5- Fastening screw 6- Scraper blade 2- Volumetric pump 7- Bowl of fluid outlet 3- Bowl of fluid inlet 4- Rotating shaft (rotor) 8- Volumetric flow meters

21 Fig.1. Schematic representation of experimental setup.

-24-





Fig.3. Test sections used in the PIV experiments: a) radial and tangential velocity components, b) axial and radial velocity components.







- **Fig.6.** Magnitude of tangential velocity contour of $(v_{\theta}^2 + v_r^2)^{0.5}$ at z*-locations of 0.16, 0.2 and 0.3 respectively, pure glycerin, N=5 rev/s, $Q_v=1.38 \ 10^{-5} \ m^3/s$. 4
- 5





- 25 plots for different angular positions (right).
- 26



Fig.8. Normalized tangential velocity plots in different angular positions for N=3 rev/s, $Q_v=1.38 \ 10^{-5} \ m^3/s$, a) pure glycerin, b) 1% CMC solution.



- **Fig.9.** Mean velocity vectors plots in *r*-*z* planes in the case of *a*) pure glycerin and *b*) 1% CMC solution, N=5 rev/s and $Q_v=1.38 \ 10^{-5} \ m^3/s$, entrance plane and exit plane, $\theta=90^\circ$. 1
- 2



Fig.10. Normalized axial velocity profiles for pure glycerin at different z^* positions and rotating velocity: *a*) $Q_v = 1.38 \ 10^{-5} \ m^3/s$, *b*) $Q_v = 4.2 \ 10^{-5} \ m^3/s$, $\theta = 90^{\circ}$.



Fig.11. Normalized axial velocity profiles for 1% CMC solution at different z*-locations of and rotating velocity: a) $Q_v = 1.38 \ 10^{-5} \ m^3/s$, b) $Q_v = 4.2 \ 10^{-5} \ m^3/s$, $\theta = 90^\circ$.

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Fig.12. Normalized radial velocity plots across (r, z) planes for N=1, 3 and 5 rev/s, $Q_v=4.210^{-5} m^3/s$: a) pure glycerine, b) 1% CMC solution, $\theta=90^{\circ}$.



6 **Fig.13.** Normalized radial velocity plots in different *θ*-angular positions for N=3 rev/s, 7 $Q_v=1.38$ 10⁻⁵ m³/s: a) pure glycerine, b) 1% CMC solution.



<u>Highlights</u>

- 1. The PIV was used to add a new knowledge on the flow structure within a SSHE
- 2. Synchronization of laser and rotating blades was made allowed statistical analysis
- **3.** Axial heterogeneities at the entrance and exit of SSHE were observed
- 4. Radial velocity was showed a good mixing around the blades
- **5.** Tangential velocity have shown a maximum localized near the tip of the blade