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► **To cite this version:**

R Chowdhury, S Adhikari, F Scarpa, M I Friswell. Transverse vibration of single-layer graphene sheets. Journal of Physics D: Applied Physics, 2011, 44 (20), pp.205401. 10.1088/0022-3727/44/20/205401 . hal-00616913

**HAL Id: hal-00616913**

**<https://hal.science/hal-00616913>**

Submitted on 25 Aug 2011

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# Transverse vibration of single layer graphene sheets

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**Abstract.** We investigate the vibrational properties of zigzag and armchair single-layer graphene sheets (SLGSs) using the molecular mechanics approach. The natural frequencies of vibration and their associated intrinsic vibration modes are obtained. Vibrational analysis is performed with different chirality and boundary conditions. The simulations are carried out for three types of zigzag and armchair SLGS. The universal force field potential is used for the molecular mechanics approach. The first four natural frequencies are obtained for increasing lengths. The results indicate that the natural frequencies decrease as the length increases. The results follow similar trends with results of previous studies for SLGS using a continuum structural mechanics approach. These results have shown the applicability of SLGSs as electromechanical resonators.

PACS numbers: 62.25.-g

Mechanical properties of nanoscale systems

PACS numbers: 68.55.-a

Thin film structure and morphology

PACS numbers: 61.43.Bn

Structural modeling: serial-addition models, computer simulation

**Keywords:** graphene, bending, molecular mechanics, frequencies

Submitted to: *Journal of Physics D: Applied Physics*

## 1. Introduction

In 2003, Gan et. al. [1] described the in-situ exfoliation of a single layer with an intersecting grain boundary in graphite using an STM operated in air. The exfoliation technique has led to the successful production of graphene, opening a new era in the field of nanoelectronics [2–4]. The very high in-plane stiffness of graphene sheets [5] has suggested some possible use of graphite nanosheets for nanosensors and NEMS applications [6, 7], due to their extremely high surface to volume ratio, as well as large deflection capability under point loading [8–13]. In this paper the out-of-plane or transverse vibration of single-layer graphene sheets (SLGSs) is considered. The vibration studies could be useful for graphene based mass and/or gas sensors [6, 14, 14–18].

The vibration of single and multiple layer graphene sheets has been investigated by several authors, using either continuum mechanics approaches [19], equivalent lattice structures made by atomistic-continuum models representing the C-C bonds [20], and molecular dynamics approaches combined with continuum mechanics for thickness identification [21]. The out-of-plane deformation of SLGS has been considered using the continuum mechanics models [13, 22], together with continuum and truss-like structural assemblies [23–30]. In a recent paper a nonlinear mechanical model has been used [31] to take account for large deformations in SLGS. They observed higher resonance frequencies from the nonlinear model compared to the equivalent linear model. A molecular mechanics (MM) approach based on the computation of the Hessian matrix and its eigenvalues has been proposed by some authors to describe the structural dynamics of single wall carbon nanotubes [32], and used to validate a lattice structural mechanics approach in nanoribbons [33]. In this work, we describe the behavior of the natural frequencies and modes shapes of single layer graphene sheets with various boundary conditions using the MM method, and compare the findings with continuum mechanics based on isotropic properties. Kirschoff-based plate formulations consider the graphene sheets as an isotropic continuum, while edge effects and the finite size of the sheets have been demonstrated to provide in-plane special orthotropic properties [5, 29, 34, 35].

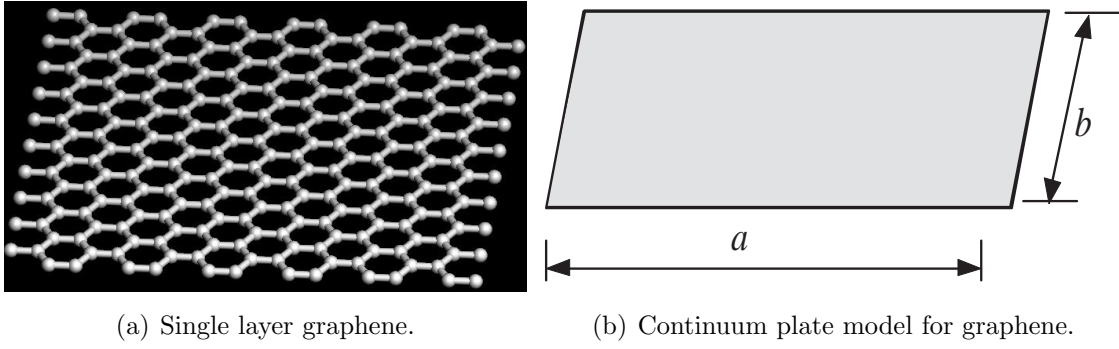
The main novelty of our paper is the molecular mechanics [36, 37], which is a higher fidelity model compared to the previously used models [20, 38] where the structural mechanics (SM) approach is used. In the MM approach, the molecule first finds the lowest energy configuration from its pre-optimized state and then the subsequent calculations are performed. Finding the minimum potential energy surface is a key distinguishing feature in the MM approach. It can be noted that during the optimization the structures may change their configurations. Planar structure such as SLGS may deform to attain the lowest energy configuration. This is ignored in the SM approach [20, 38], which only considers the initial geometry of the molecule. In the SM approach the Euler-Bernoulli beam model is used to represent the C-C bonds and subsequently the finite element method [39] is used to discretise the equation of motion and obtain the natural frequencies. In the formulation presented in ref [20, 38], the length and the diameter of the ‘beams’ representing the C-C bonds are almost similar (length = 0.142nm and diameter = 0.146nm). It is well known [40] that for such ‘short beams’ the Euler-Bernoulli beam model used in [20, 38] produce inaccurate results as the shear deformation is ignored. This can introduce error particularly for the calculation of higher frequencies. The results obtained using the MM approach do not have these drawbacks as neither the Euler-Bernoulli beam model, nor the finite element approach, is used. Contrary to SM approaches, the molecular mechanics method also allows the natural frequencies of the system in equilibrium to be computed without any requirement to use an equivalent thickness value for the nanostructure. In general, the introduction of the thickness concept in nanomaterials is highly contentious, leading to the well-known “Yakobson’s paradox” [41, 42], responsible for the high scattering of Young’s moduli and Poisson’s ratio results in open literature.

We will show that the MM approach is able to capture the equivalent anisotropic properties and their influence in the structural dynamics of the graphene sheets, and therefore provide a valid prediction tool to simulate the resonance behavior of graphene-based NEMS devices. The paper is organized in the following way. The continuum mechanics approach for the frequency analysis of graphene sheets is presented in section 2. Section 3 will be centered on the analysis and calculation of the frequencies using a molecular mechanics model. The numerical results and discussion will be presented in section 4. Finally, the major conclusions of this paper will be drawn in section 5 based on the results and analyses in section 4.

## 2. Vibration of single layer graphene

A single layer graphene sheet (shown in Fig. 1) may be approximated by a thin elastic plate [11]. The equation of motion of the transverse free vibration of a thin elastic plate [40, 43] can be expressed as

$$D \left( \frac{\partial^4 w}{\partial x^4} + 2 \frac{\partial^2 w}{\partial x^2} \frac{\partial^2 w}{\partial y^2} + \frac{\partial^4 w}{\partial y^4} \right) + \rho \frac{\partial^2 w}{\partial t^2} = 0 \quad (1)$$



**Fig. 1.** A rectangular single layer graphene sheet (SLGS) and its mathematical idealization using a thin continuum plate. The SLGS is assumed to be of dimension  $a \times b$ .

Here  $w \equiv w(x, y, t)$  is the transverse deflection,  $x, y$  are coordinates,  $t$  is the time,  $\rho$  is the mass density per area and the bending rigidity is defined by

$$D = \frac{Eh^3}{12(1 - \nu^2)} \quad (2)$$

$E$  is the Young's modulus,  $h$  is the thickness and  $\nu$  is the Poisson's ratio. We consider rectangular graphene sheets with cantilevered (clamped at one edge) and bridged (clamped at two opposite edges) boundary conditions. Following Blevins [44], the natural frequency (in rad/s) of a rectangular plate of dimension  $a \times b$  can be expressed as

$$\omega_{ij} = \left\{ \frac{\pi^4 D}{a^4 \rho} \right\}^{1/2} \left\{ G_x^4 + G_y^4 \left( \frac{a}{b} \right)^4 + 2 \left( \frac{a}{b} \right)^2 [\nu H_x H_y + (1 - \nu) J_x J_y] \right\}^{1/2} \quad (3)$$

where  $i, j = 1, 2, 3, \dots$  are mode indices. The values of the coefficients  $G_x, H_x, J_x$  and  $G_y, H_y, J_y$  depend on the boundary conditions and the mode indices  $i, j$ . The first set of coefficients depends on the boundary conditions of the edges of width (side  $b$ ) while the second set of coefficients depends on the boundary conditions of the edges of length (side  $a$ ). The boundary conditions on the two edges of length (side  $a$ ) are free. In this paper we consider the lower modes of vibration. For the first three modes the coefficients  $G_y, H_y$  and  $J_y$  are given in Table 1. The coefficients  $G_x, H_x$  and  $J_x$  for both boundary

**Table 1.** Coefficients for the free-free boundary conditions of the edges of length (side  $a$ )

Mode index ( $j$ )	$G_y$	$H_y$	$J_y$
1	0	0	0
2	0	0	1.216
3	1.506	1.248	5.017

conditions on the edges of width (side  $b$ ) are given in Table 2. General expressions of the coefficients for the higher values of  $i$  and  $j$  are given in [44]. The values given in Tables

**Table 2.** Coefficients for the two boundary conditions on the edges of width (side  $b$ )

Mode index ( $i$ )	Clamped-clamped			Clamped-free		
	$G_x$	$H_x$	$J_x$	$G_x$	$H_x$	$J_x$
1	1.506	1.248	1.248	0.597	-0.0870	0.471
2	2.5	4.658	4.658	1.494	1.347	3.284
3	3.5	10.02	10.02	2.5	4.658	7.842

1 and 2 will be used to obtain the natural frequencies and compare with the molecular mechanics simulation described in the next section.

### 3. Molecular simulation approach

Since atomic configurations can have significant impact on the mechanical properties of single layer graphene sheets, zigzag and armchair models are adopted in this study. The zigzag and armchair models of graphene sheets under consideration are:

- Zigzag sheet clamped at one edge (Cantilevered condition)
- Zigzag sheet clamped at two opposite edges (Bridged condition)
- Armchair sheet clamped at one edge (Cantilevered condition)
- Armchair sheet clamped at two opposite edges (Bridged condition)

Different atomic configurations and boundary conditions can be considered in an unified manner within the scope of molecular mechanics. The general expression of total energy is a sum of energies due to valence or bonded interactions and non-bonded interactions [45]

$$\begin{aligned}
 E = & \sum_0^{N_B} \frac{1}{2} k_{IJ} (r - r_{IJ})^2 + \sum_0^{N_A} k_{IJK} (C_0 + C_1 \cos \theta + C_2 \cos 2\theta) \\
 & + \sum_0^{N_T} \frac{1}{2} V_\phi (1 - \cos(n\phi_0) \cos(n\phi)) + \sum_0^{N_I} V_\omega (C_0^I + C_1^I \cos \omega + C_2^I \cos 2\omega) \quad (4) \\
 & + \sum_0^{N_{nb}} R_{IJ} \left[ -2 \left( \frac{x_{IJ}}{x} \right)^6 + \left( \frac{x_{IJ}}{x} \right)^{12} \right] + \sum_0^{N_{nb}} \frac{q_I \cdot q_J}{\varepsilon \cdot x}
 \end{aligned}$$

$N_B$ ,  $N_A$ ,  $N_T$ ,  $N_I$  and  $N_{nb}$  are the numbers of the bond-, angle-, torsion-, inversion- and the non bonded-terms, respectively.  $k_{IJ}$  and  $k_{IJK}$  are the force constants of the bond- and angle-terms, respectively.  $r$  and  $r_{IJ}$  are the bond distance and natural bond distance of the two atoms  $I$  and  $J$ , respectively.  $\theta$  and  $\theta_0$  are the angle and natural angle for three atoms  $I - J - K$ , respectively.  $\phi$  and  $\phi_0$  are the torsion angle and torsion natural angle for three atoms  $I - J - K - L$ , respectively.  $V_\phi$ ,  $n$ ,  $V_\omega$ ,  $\omega$  are the height of the torsion barrier, periodicity of the torsion potential, height of the inversion barrier and inversion- or out-of-plane-angle at atom  $I$ , respectively.  $C_0^I$ ,  $C_1^I$  and  $C_2^I$  are the Fourier

coefficients of the inversions terms.  $x$  and  $x_{IJ}$  are the distance and natural distance of two non bonded atoms  $I$  and  $J$ .  $R_{IJ}$  is the depth of the Lennard-Jones potential.  $q_I$  and  $\varepsilon$  are the partial charge of atoms  $I$  and dielectric constant. For the general nonlinear case, the bend function should have a minimum  $\theta = \theta_0$ , with the second derivative at  $\theta_0$  equal to the force constant ( $k_{IJK}$ ). The Fourier coefficients of the general angle terms  $C_0$ ,  $C_1$  and  $C_2$  are evaluated as a function of the natural angle  $\theta_0$ :

$$\begin{aligned} C_2 &= \frac{1}{4\sin^2\theta_0} \\ C_1 &= -4C_2 \cos \theta_0 \\ C_0 &= C_2 (2\cos^2\theta_0 + 1) \end{aligned} \quad (5)$$

The bond stretching force constants ( $k_{IJ}$ ) are atom based and are obtained from generalization of Badger's rules. The assumption is that the bonding is dominated by attractive ionic terms plus short-range Pauli repulsions [45]. The force constant (in units of (kcal/mol)/ $\text{\AA}^2$ ) then becomes

$$k_{IJ} = 644.12 \frac{Z_I^* Z_J^*}{r_{IJ}^3} \quad (6)$$

The  $Z_I^*$  is the effective atomic charges, in electron units. Similarly, the angle bend force constants ( $k_{IJK}$ ) are generated using the angular generalization of Badger's rule. The force constant (in units of kcal/mol.rad<sup>2</sup>) then becomes [46]:

$$k_{IJK} = 644.12 \frac{Z_I^* Z_J^*}{r_{IJ}^5} [3r_{IJ} r_{JK} (1 - \cos^2\theta_0) - r_{IK}^2 \cos \theta_0] \quad (7)$$

The torsional constant (kcal/mol) is defined as

$$V_\phi = 5\sqrt{U_I U_J} [1 + 4.18 \ln (BO_{JK})] \quad (8)$$

where,  $BO_{JK}$  is the bond order for Atom- $J$  and Atom- $K$ .  $U_I$  and  $U_J$  are the atomic constants defined with UFF  $sp^2$ . Regarding the inversion term, the coefficients are  $C_0^I = 1$ ,  $C_1^I = -1$  and  $C_2^I = 0$  for  $sp^2$  atom type. In this study, we used the UFF model [45], wherein the force field parameters are estimated using general rules based only on the element, its hybridization and its connectivity. [Hybridization determines the type of bonding of the carbon atoms with its neighbours.](#) The  $sp^3$  hybridization corresponds to the well-known tetrahedral configuration in which carbon binds to four neighbours giving rise to three-dimensional inter-connectivity of carbon atoms that is found in diamond. The  $sp^2$  bonding in which carbon atoms bind to three neighbours also known as trigonal hybridization gives planar structures found in graphite and graphene. The  $sp^2$ -hybridized carbon atoms, which are at their energy minimum in planar graphite (or graphene), must be bent to form the closed sphere (fullerenes) or tube (CNT), which produces angle strain. The characteristic reaction of fullerenes is electrophilic addition at 6,6-double bonds, which reduces angle strain by changing  $sp^2$ -hybridized carbons

into  $sp^3$ -hybridized ones. The change in hybridized orbitals causes the bond angles to decrease from about  $120^\circ$  in the  $sp^2$  orbitals to about  $109.5^\circ$  in the  $sp^3$  orbitals. This decrease in bond angles allows for the bonds to bend less when closing the sphere or tube, and thus, the molecule becomes more stable. The force field functional forms and parameters used in this study are in accordance with [45]. The calculation of frequency and their validation for CNTs were detailed in [32]. In the following section we provide the methodology of the frequency calculation.

### 3.1. Calculation of the natural frequencies

We start with the Hessian matrix  $\mathbf{f}_{CAR}$ , which holds the second partial derivatives of the potential  $E$  with respect to the displacement of the atoms in cartesian coordinates (CAR) [46, 47]:

$$f_{CAR_{ij}} = \left( \frac{\partial^2 E}{\partial \xi_i \partial \xi_j} \right)_{Opt} \quad (9)$$

This is a  $3N \times 3N$  matrix ( $N$  is the number of atoms), where  $\xi_1, \xi_2, \xi_3, \dots, \xi_{3N}$  denote the displacements in cartesian coordinates,  $\Delta x_1, \Delta y_1, \Delta z_1, \dots, \Delta z_N$ . The  $(\ )_{Opt}$  refers to the fact that the derivatives are taken at the equilibrium positions of the atoms. These force constants are then converted to mass weighted cartesian coordinates (MWC) [47]

$$f_{MWC_{ij}} = \frac{f_{CAR_{ij}}}{\sqrt{m_i m_j}} = \left( \frac{\partial^2 E}{\partial c_i \partial c_j} \right)_{Opt} \quad (10)$$

where,  $c_1 = \sqrt{m_1} \xi_1 = \sqrt{m_1} \Delta x_1$ ,  $c_2 = \sqrt{m_1} \xi_2 = \sqrt{m_1} \Delta y_1$  and so on.  $\mathbf{f}_{MWC}$  is diagonalized, yielding a set of  $3N$  eigenvectors and  $3N$  eigenvalues.

The next step is to translate the center of mass to the origin, and determine the moments and products of inertia, with the goal of finding the matrix that diagonalizes the moment of inertia tensor. Using this matrix we can find the vectors corresponding to the rotations and translations. Once these vectors are known, we know that the rest of the normal modes are vibrations. The center of mass  $\mathbf{R}_{COM}$  is found in the usual way:

$$\mathbf{R}_{COM} = \frac{\sum_{\alpha} m_{\alpha} \mathbf{r}_{\alpha}}{\sum_{\alpha} m_{\alpha}} \quad (11)$$

where the sums are over the atoms,  $\alpha$ . The origin is then shifted to the center of mass  $\mathbf{r}_{COM_{\alpha}} = \mathbf{r}_{\alpha} - \mathbf{R}_{COM}$ . Next we have to calculate the moments of inertia (the diagonal elements) and the products of inertia (off diagonal elements) of the moment of inertia tensor ( $\mathbf{I}$ ).

$$\mathbf{I} = \begin{Bmatrix} I_{xx} & I_{xy} & I_{xz} \\ I_{yx} & I_{yy} & I_{yz} \\ I_{zx} & I_{zy} & I_{zz} \end{Bmatrix} = \begin{Bmatrix} \sum m_{\alpha} (y_{\alpha}^2 + z_{\alpha}^2) & -\sum m_{\alpha} (x_{\alpha} y_{\alpha}) & -\sum m_{\alpha} (x_{\alpha} z_{\alpha}) \\ -\sum m_{\alpha} (y_{\alpha} x_{\alpha}) & \sum m_{\alpha} (x_{\alpha}^2 + z_{\alpha}^2) & -\sum m_{\alpha} (y_{\alpha} z_{\alpha}) \\ -\sum m_{\alpha} (z_{\alpha} x_{\alpha}) & -\sum m_{\alpha} (z_{\alpha} y_{\alpha}) & \sum m_{\alpha} (x_{\alpha}^2 + y_{\alpha}^2) \end{Bmatrix} \quad (12)$$

The sums appearing in the above expression is over the index  $\alpha$ . This symmetric matrix is diagonalized, yielding the principal moments (the eigenvalues  $\mathbf{I}'$ ) and a  $3 \times 3$  matrix



( $\mathbf{X}$ ), which is made up of the normalized eigenvectors of ( $\mathbf{I}$ ). The eigenvectors of the moment of inertia tensor are used to generate the vectors corresponding to translation and infinitesimal rotation of the molecular system. A Schmidt orthogonalization is used to generate  $N_{vib} = 3N - 6$  remaining vectors, which are orthogonal to the six rotational and translational vectors. The result is a transformation matrix  $\mathbf{D}$  which transforms from mass weighted cartesian coordinates  $\mathbf{q}$  to internal coordinates  $\mathbf{S} = \mathbf{D}\mathbf{q}$ , where rotation and translation have been separated out. Now that we have coordinates in the rotating and translating frame, we need to transform the Hessian,  $\mathbf{f}_{MWC}$ , to these new internal coordinates (INT) [46, 47]. Only the  $N_{vib}$  coordinates corresponding to internal coordinates will be diagonalized, although the full  $3N$  coordinates are used to transform the Hessian. The transformation is straightforward as follows:

$$\mathbf{f}_{INT} = \mathbf{D}^T \mathbf{f}_{MWC} \mathbf{D} \quad (13)$$

The  $N_{vib} \times N_{vib}$  submatrix of  $\mathbf{f}_{INT}$ , which represents the force constants internal coordinates, is diagonalized yielding  $N_{vib}$  eigenvalues  $\lambda = 4\pi^2\omega^2$ , and  $N_{vib}$  eigenvectors. If we call the transformation matrix composed of the eigenvectors  $\mathbf{L}$ , then we have

$$\mathbf{L}^T \mathbf{f}_{INT} \mathbf{L} = \mathbf{\Lambda} \quad (14)$$

where  $\mathbf{\Lambda}$  is the diagonal matrix with eigenvalues  $\lambda_i$ . At this point, the eigenvalues need to be converted to frequencies (in Hz) as

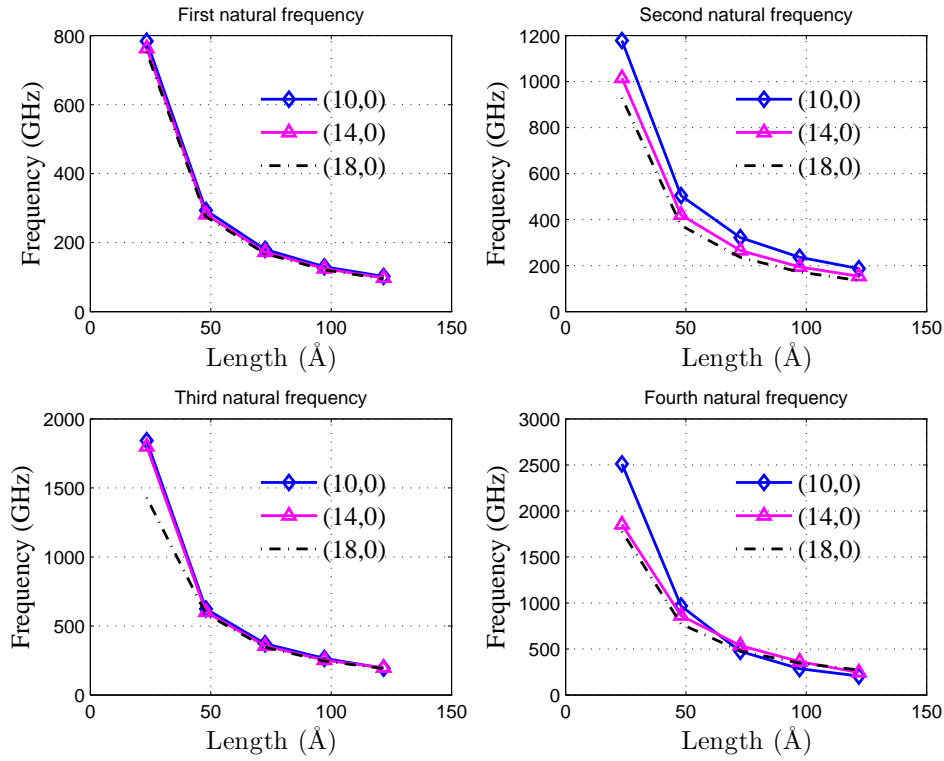
$$\omega_i = \sqrt{\frac{\lambda_i}{4\pi^2}} \quad (15)$$

## 4. Results and discussions

The resonant frequencies of single layer graphene (SLG)-based resonators depend on the geometric configurations. The atomic structures of SLGS could also exert significant influence on their vibration behaviours. Thus, in this work, we analyze two groups of SLG resonators, i.e., three zigzag SLGS (10,0), (14,0), (18,0) and three armchair SLGS (11,11), (15,15), (19,19), with increasing length. In this study, we computed our results using bridged (atoms at the two sides along width are restrained) and cantilevered (atoms at one side along width are restrained) boundary conditions. The computational results of the first four vibrational frequencies of these zigzag SLGS are calculated and presented in Fig. 2 and Fig. 3, respectively, for bridged and cantilevered boundary condition. Similarly, Fig. 4 and Fig. 5, respectively, presents the first four vibrational frequencies of these armchair SLGS, for bridged and cantilevered boundary condition. The widths of the SLGS are given in the figure captions.

### 4.1. Dependence of the length

As shown in Fig. 2 and Fig. 3, for SLGS with the length rising from around 20Å to 120Å, the fundamental frequencies are in the ranges of 100-3000 GHz (ref. Table 3) and

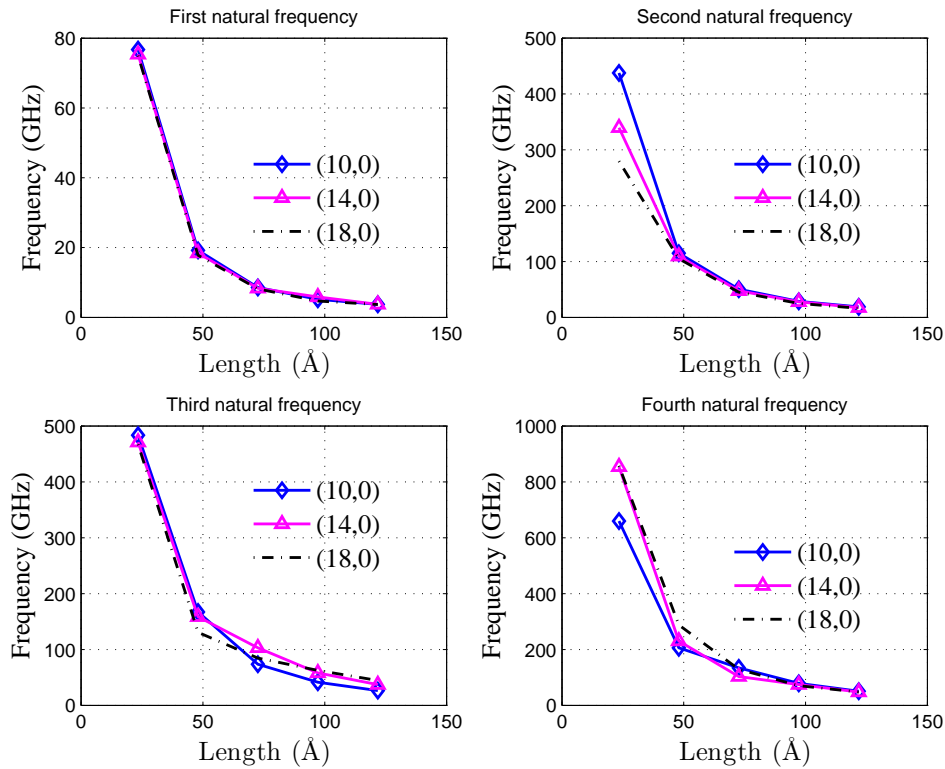


**Fig. 2.** Bridged boundary condition - First four vibrational frequencies of zigzag SLGS as a function of the length of SLGS. The widths are - (10,0): 9.317Å; (14,0): 13.554Å; (18,0): 17.803Å.

4-1300 GHz (ref. Table 4) for the zigzag SLGS with bridged and cantilevered boundary conditions, respectively. While for the armchair SLGS, the variation of frequencies is between 40-1070 GHz (ref. Table 5) and 2-415 GHz (ref. Table 6) for bridged and cantilevered boundary conditions, respectively, with the length rising from around 40Å to 210Å. The trend of the frequency changes with length are generally in accordance with that given in the literature [20]. The discrepancy is primarily a result of the different end constraints, geometric configurations of SLGS and the differences in the simulation approaches. Recall that here the molecular mechanics approach is used as opposed to the finite element method employed in [20, 38].

#### 4.2. Dependence of the geometric configuration

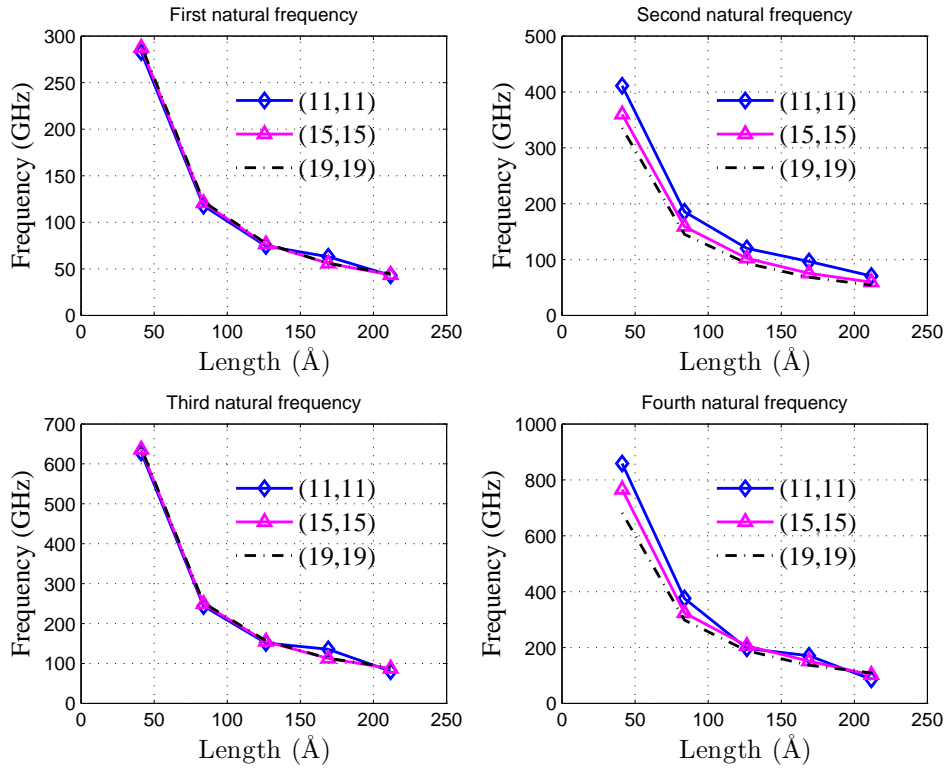
For both zigzag and armchair SLGS, the frequencies of all five modes generally decrease with increasing length. The curves of frequency becomes steeper for the SLGS of smaller sheet length ( $\leq 50\text{Å}$ ). This shows that the dependence on the aspect ratio is stronger for the frequencies of shorter SLGS. In the meantime, it is also seen from Fig. 2 and Fig. 4 that, for a given length the frequencies of SLGS always decline with rising sheet width. This issue can be further clarified from Tables 3 to 6. The frequencies of small-width SLGS are always higher than the corresponding frequencies of large-width SLGS. This is especially so when the length is relatively small. However, the effect of width on the



**Fig. 3.** Cantilevered boundary condition - First four vibrational frequencies of zigzag SLGS as a function of the length of SLGS. The widths are - (10,0): 9.317Å; (14,0): 13.554Å; (18,0): 17.803Å.

frequencies diminishes for SLGS with larger widths. As an example, for zigzag SLGS (ref. Table 3) with width 9.317Å the frequency decreases from 784.10 to 101.88 GHz when the length increases from 23.391Å to 121.822Å, while for certain length of 23.391Å it only varies from 784.10 GHz to 752.30 GHz, when width increases from 9.317Å to 17.803Å. Similar behavior is also observed for armchair SLGS in Table 3. Thus we see that when the aspect ratio of SLGS grows, the difference in frequency due to the variation of length decreases significantly whereas the ratio between the frequencies remains almost unchanged. This observation suggests that the influence of width on the vibration frequency of SLGS does not significantly change with increasing length. Here the decreasing frequencies with increasing length and width observed in Fig. 2, and Fig. 4 can be attributed to the fact that SLGS of larger length and width possess lower dynamic structural stiffness in both longitudinal and transverse directions. In particular, for SLGS of small width their transverse stiffness is high. The frequency of such SLGS thus becomes more sensitive to their longitudinal rigidity, which finally leads to stronger effect of the length for SLGS with smaller width.

Based on the molecular mechanics method, mode shapes of the SLGS are obtained. The first six mode shapes of zigzag and armchair sheets are demonstrated in the supplementary document (see supplementary materials). The first mode shape plays a significant role in the design of the nanomechanical resonators. It is perceived that

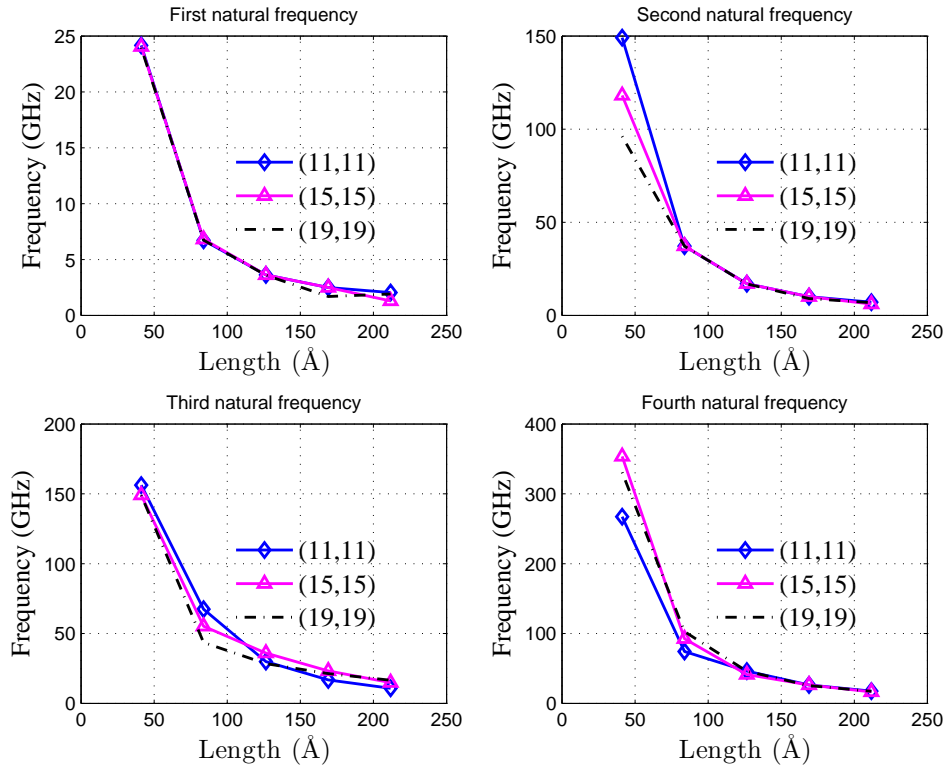


**Fig. 4.** Bridged boundary condition - First four vibrational frequencies of armchair SLGS as a function of the length of SLGS. The widths are - (11,11): 12.31Å; (15,15): 17.23Å; (19,19): 122.15Å.

the SLGS with different boundary conditions has a sinusoidal and/or hyperbolic sine and cosine configuration. These configurations guarantee the ease of detection of any small deflection in the SLGS. In addition, mode shapes of the SLGS, in contrast to the CNTs, are not changed by the length or aspect ratio.

#### 4.3. Dependence of the atomic structure

Next we examine the effect of atomic structures on the frequency of SLGS. To this end we consider the calculated frequencies in Table 5 and Table 3 respectively, for armchair and zigzag SLGS with the bridged boundary condition. Similarly, we consider the calculated frequencies in Table 6, and Table 4 respectively, for armchair and zigzag SLGS with the cantilevered boundary condition. The results in Fig. 6 show that the chirality does not have a significant influence on the natural frequencies of vibration. It is shown that for almost same width and length the fundamental frequencies of zigzag SLGS are only higher than those of armchair SLGS for bridged case, whereas for cantilevered case, it is almost comparable. In the context of single wall carbon nanotubes (SWCNTs), the difference between the frequencies of the two types (e.g., zigzag and armchair) are also not very large [32]. This may be expected as SWCNTs are effectively rolled up SLGS. The frequency of SLGS is primarily determined by their geometry, i.e., length



**Fig. 5.** Cantilevered boundary condition - First four vibrational frequencies of armchair SLGS as a function of the length of SLGS. The widths are - (11,11): 12.31Å; (15,15): 17.23Å; (19,19): 122.15Å.

and the aspect ratio, and cannot be substantially changed by varying their atomic structure. This finding demonstrates that the continuum models can produce a good approximation for the vibration of SLGS with different atomic structures.

#### 4.4. Comparison with the continuum theory

In this section we investigate whether the vibrational frequencies obtained from the simple continuum plate model are comparable with the vibrational frequencies computed using the molecular mechanics approach. The density per unit area on the SLGS is computed from the total mass divided by the total area on the SLGS. We use the first natural frequency to obtain the bending rigidity  $D$  using the natural frequency equation (3) with the values of the coefficients corresponding to  $i = 1$  and  $j = 1$  in Tables 1 and 2. We then use this value of  $D$  to compute the higher natural frequencies to understand if the simple plate model is applicable. The first natural frequency is not shown in the figures because the constant  $D$  is calculated using the first natural frequency.

In Fig. 7, the second and third bending mode frequencies are compared for the bridged boundary condition. We used (10,0) zigzag SLGS as an example. The results for the other types of SLGS used in this study show similar behavior. The analytical results shown in Fig. 7 are obtained from equation (3) with the values of the coefficients

**Table 3.** Vibrational frequencies of zigzag SLGS in GHz - Bridged boundary condition.

Index and Length ( $\text{\AA}$ ) Width ( $\text{\AA}$ )	$\omega_1$	$\omega_2$	$\omega_3$	$\omega_4$	$\omega_5$	
(10,0) 9.317	23.391	784.10	1178.43	1842.16	2511.38	2960.40
	47.995	293.30	504.48	623.61	967.17	1006.04
	72.603	180.15	321.84	370.86	474.31	580.16
	97.213	130.12	236.47	264.56	285.31	406.86
	121.822	101.88	186.91	194.26	205.92	313.98
(14,0) 13.554	23.391	763.00	1014.94	1799.25	1850.99	2220.43
	47.995	281.72	420.08	600.04	861.22	983.66
	72.603	172.47	265.86	355.04	536.70	558.29
	97.213	124.42	194.62	252.85	361.47	387.81
	121.822	97.34	153.53	196.67	244.08	299.77
(18,0) 17.803	23.391	752.30	929.62	1430.21	1777.36	2059.21
	47.995	275.09	374.56	587.23	772.02	897.41
	72.603	168.02	235.48	345.94	477.20	544.22
	97.213	121.12	171.99	246.01	346.27	379.20
	121.822	94.72	135.45	191.26	272.03	291.82

**Table 4.** Vibrational frequencies of zigzag SLGS in GHz - Cantilevered boundary condition.

Index and Length ( $\text{\AA}$ ) Width ( $\text{\AA}$ )	$\omega_1$	$\omega_2$	$\omega_3$	$\omega_4$	$\omega_5$	
(10,0) 9.317	23.391	76.71	437.63	483.34	659.82	1323.45
	47.995	19.18	115.03	166.98	206.05	321.91
	72.603	8.58	50.46	73.69	134.18	140.80
	97.213	5.20	28.57	41.25	78.61	99.77
	121.822	3.75	18.70	26.34	50.53	79.32
(14,0) 13.554	23.391	75.35	339.71	471.07	854.05	1035.23
	47.995	18.41	109.56	158.49	230.04	311.10
	72.603	8.34	47.75	102.90	103.10	133.99
	97.213	5.84	27.91	58.08	74.40	77.05
	121.822	3.74	17.52	37.11	47.57	60.62
(18,0) 17.803	23.391	75.00	279.65	467.10	857.12	883.47
	47.995	17.80	104.88	130.68	287.79	305.18
	72.603	8.15	45.04	84.69	128.84	131.38
	97.213	4.68	24.89	62.71	70.16	74.43
	121.822	3.67	16.34	44.52	47.76	49.86

**Table 5.** Vibrational frequencies of armchair SLGS in GHz - Bridged boundary condition.

Index and Length (Å) Width (Å)	$\omega_1$	$\omega_2$	$\omega_3$	$\omega_4$	$\omega_5$	
(11,11) 12.31	41.21	282.84	411.22	628.37	858.54	1072.30
	83.83	117.86	185.65	243.91	375.70	385.14
	126.46	74.47	120.09	151.31	197.16	232.77
	169.08	63.05	96.48	135.33	170.34	176.12
	211.71	42.94	70.46	80.55	86.31	130.72
(15,15) 17.23	41.21	287.54	359.84	636.08	765.41	953.01
	83.83	120.58	158.87	248.98	323.45	392.11
	126.46	76.32	102.08	154.90	205.84	237.93
	169.08	55.85	75.30	112.58	151.24	151.67
	211.71	43.43	59.43	87.37	101.74	119.02
(19,19) 22.15	41.21	290.01	335.32	640.24	681.45	720.46
	83.83	122.14	145.19	251.80	297.89	395.86
	126.46	77.40	93.09	156.92	188.19	240.78
	169.08	55.91	68.18	112.83	136.68	171.43
	211.71	44.70	54.26	89.80	108.84	123.21

corresponding to  $i = 2, 3$  for the clamped-clamped case in Table 2 and  $j = 1$  in Table 1. The trend in the variation of the frequencies with respect to the length is similar for both the methods. The difference between the two theories is more prominent for SLGS with smaller dimension. This is expected as the continuum theory may not be very suitable for SLGS in this case. The continuum model tends to overestimate the MM predictions (17 % and 37 % for the second and third mode respectively). The continuum mechanics formulation assumes an isotropic equivalent material for the graphene. However, edge effects have been demonstrated to play a significant role in the static [29, 34, 35, 48] and dynamic [33] mechanical properties of SLGS, leading to an equivalent orthotropic, rather than isotropic material model for the graphene. Although the dimensions of the SLGS considered in this work are dissimilar, we notice a general agreement in terms of magnitude between the eigenvalues calculated with our MM approach, and the results using the MM3 potential in [21].

Figure 8 shows equivalent plots for the SLGS with cantilevered boundary condition. The analytical results shown in Fig. 8 are obtained from equation (3) with the values of the coefficients corresponding to  $i = 2, 3$  for the clamped-free case in Table 2 and  $j = 1$  in Table 1. The results predicted by the two approaches agree more closely for this boundary condition. The results obtained here shows that the boundary condition has an effect on the accuracy of the predictions from the continuum theory.

**Table 6.** Vibrational frequencies of armchair SLGS in GHz - Cantilevered boundary condition.

Index and Length ( $\text{\AA}$ ) Width ( $\text{\AA}$ )	$\omega_1$	$\omega_2$	$\omega_3$	$\omega_4$	$\omega_5$	
(11,11) 12.31	41.21	24.18	149.23	156.26	267.23	416.07
	83.83	6.72	37.14	67.35	74.23	102.89
	126.46	3.63	17.02	29.88	46.14	48.69
	169.08	2.50	10.08	16.82	26.42	36.25
	211.71	2.04	7.06	10.81	17.54	28.85
(15,15) 17.23	41.21	24.06	118.14	149.10	353.65	387.94
	83.83	6.82	37.24	55.23	92.43	102.87
	126.46	3.62	16.98	36.00	41.29	46.08
	169.08	2.50	10.02	23.26	26.35	26.76
	211.71	1.29	6.03	14.86	16.47	21.08
(19,19) 22.15	41.21	23.96	96.16	148.95	330.70	413.55
	83.83	6.73	37.20	43.58	102.76	116.43
	126.46	3.59	16.90	28.47	45.93	52.51
	169.08	1.91	9.04	21.27	25.46	29.60
	211.71	1.69	6.82	16.53	17.27	19.03

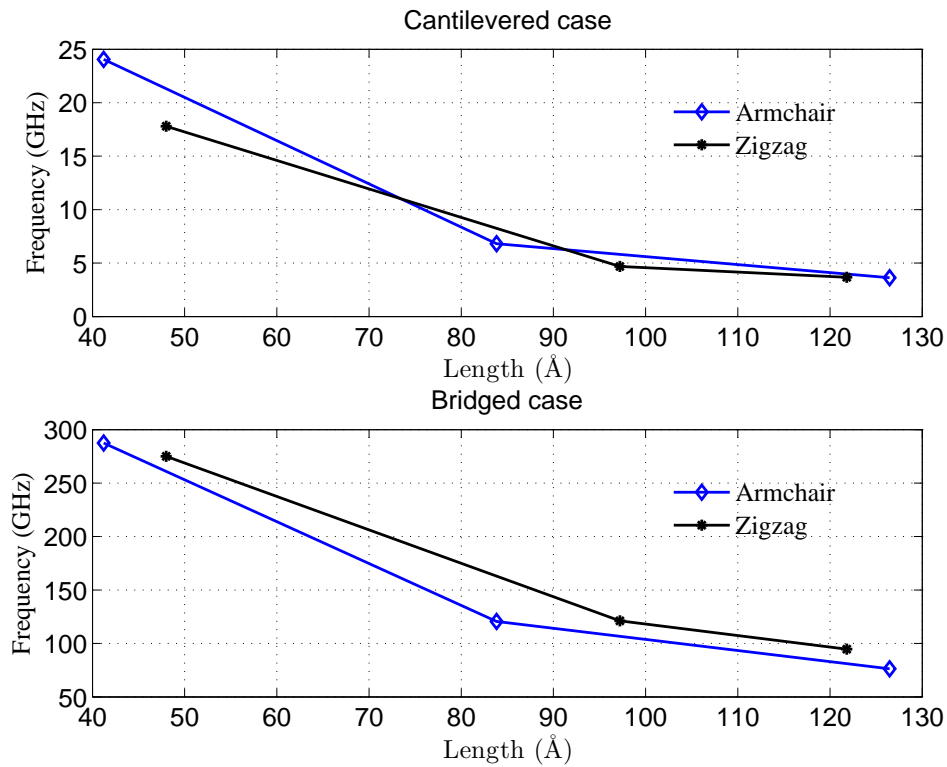
## 5. Conclusions

The vibrational properties zigzag and armchair single wall graphene sheets (SLGS) are studied. A molecular mechanics based approach is used to estimate the frequencies. We used the UFF model, wherein the force field parameters are estimated using general rules based only on the element, its hybridization, and its connectivity. Two types of boundary conditions are considered, namely, cantilevered and bridged. First five Natural frequencies are calculated for three zigzag, namely, (10,0), (14,0) and (18,0) and three armchair, namely, (11,11), (15,15) and (19,19) SLGS. The natural frequencies of SLGS decrease with length but they are generally insensitive to the atomic structure. Results obtained from the molecular mechanics approach are compared with the same obtained using the continuum plate theory. The continuum mechanics results in general tend to overestimate the natural frequencies. The two approach agree more for the cantilevered boundary condition compared to the bridged boundary condition. The results obtained in the paper may be useful for the design and analysis of vibrating SLGS based NEMS and sensor devices.

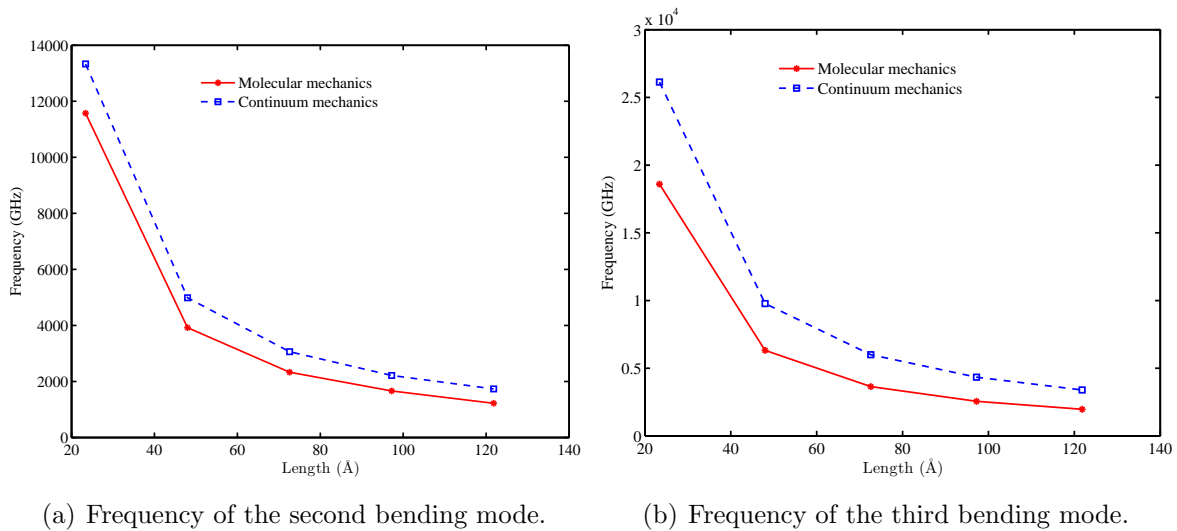
## Acknowledgements

RC acknowledges the support of Royal Society through the award of Newton International Fellowship. SA gratefully acknowledges the support of The Leverhulme





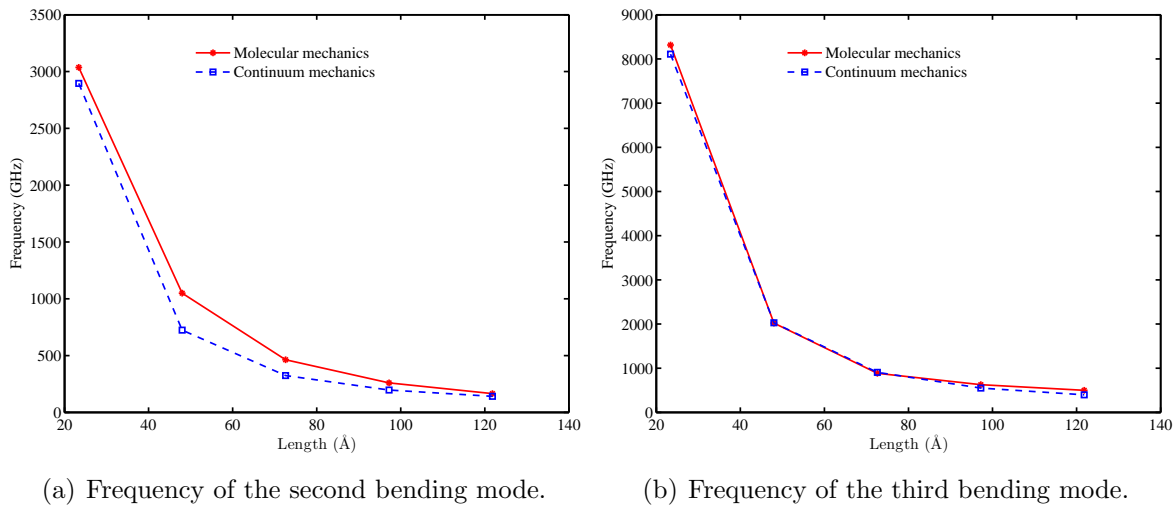
**Fig. 6.** Dependence of the molecular structure on the fundamental frequency of SLGS. It is found that, natural frequencies of zigzag SLGS (18,0) are higher compared to armchair SLGS (15,15) for bridged case, whereas for cantilevered case, it is almost comparable.



(a) Frequency of the second bending mode.

(b) Frequency of the third bending mode.

**Fig. 7.** Bridged boundary condition - vibrational frequencies of the zigzag SLGS (10,0) for the second and third bending mode. Molecular mechanics results and continuum mechanics results are compared for different values of the length.



**Fig. 8.** Cantilevered boundary condition - vibrational frequencies of the zigzag SLGS (10,0) for the second and third bending mode. Molecular mechanics results and continuum mechanics results are compared for different values of the length.

Trust for the award of the Philip Leverhulme Prize.

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