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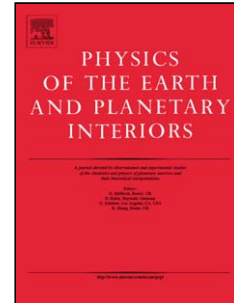
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1 **SLIM3D: A tool for three-dimensional thermomechanical modeling of lithospheric**
2 **deformation with elasto-visco-plastic rheology**

3

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5

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8

9 **Abstract**

10

11 We describe a new technique for three-dimensional lithospheric-scale modeling of solid state
12 deformation including strain localization processes. The new code, SLIM3D, includes a
13 coupled thermo-mechanical treatment of deformation processes and allows for an elasto-
14 visco-plastic rheology with diffusion, dislocation and Peierls creep mechanisms and Mohr-
15 Coulomb plasticity. The code incorporates an Arbitrary Lagrangian Eulerian formulation with
16 free surface and Winkler boundary conditions. SLIM3D is developed and implemented using
17 the C++ object-oriented programming language. We describe aspects of physical models as
18 well as details of the numerical implementation, including the Newton-Raphson solver, the
19 stress update procedure, and the tangent operator. The applicability of the code to
20 lithospheric-scale modeling is demonstrated by a number of benchmark problems that
21 include: (i) the bending of an elastic plate, (ii) the sinking of a rigid cylinder into a viscous
22 fluid, (iii) the initiation of shear bands in the brittle crust, (iv) triaxial compression test, and
23 (v) lithospheric transpressional deformation. Finally, we discuss possible directions of further
24 development.

25

26 *Keywords:* 3D numerical modeling; Geodynamics; Lithospheric deformation; Strain
27 localization; Mohr-Coulomb

28

29 **1. Introduction**

30

31 On geological time scales, lithospheric rocks deform by three fundamentally different
32 phenomenological mechanisms. These are: (i) elastic mechanism (reversible), which on
33 geological time-scales occur mostly during flexural deformation of the lithospheric plates or
34 bending of slabs in subduction zones; (ii) viscous mechanism, typical of the convective

35 mantle, or for lower crustal flow at plate boundaries; and (iii) plastic mechanism, which
36 simulate brittle failure and manifests itself in the formation of narrow shear zones or faults.
37

38 The quantitative proportion between each of these mechanisms depends on many parameters,
39 the most important of which are temperature and stress. Viscosity, for instance, varies by
40 several orders of magnitude with temperature change of a few hundred degrees. At low
41 temperatures, viscous deformation is prohibited which leads to a buildup of elastic stresses
42 during deformation, until the rock fails. This type of behavior characterizes brittle or elasto-
43 plastic deformation. At high temperatures, on the other hand, the rock may accommodate high
44 deformation rates through the viscous mechanism without excessive stress accumulation. This
45 is a typical manifestation of the ductile flow regime.

46
47 The lithosphere is not only a region of large compositional heterogeneity, contrasting in this
48 respect with asthenosphere, but it is also a locus of large temperature variations. In this
49 context, it becomes necessary not only to resolve separate brittle and ductile flow regimes, but
50 also to adequately model the brittle-ductile transition. Apart from many other complexities,
51 this single requirement makes lithospheric-scale modeling a challenging task.

52
53 This is further complicated by several factors, including: the presence of the Earth's free
54 surface, where erosion and sedimentation processes take place; the occurrence of
55 metamorphic reactions in rocks with changing temperature and pressure; the presence of
56 spontaneously evolving large-scale fault zones; by the inherent three-dimensional nature of
57 most lithospheric-scale problems, and many others. All these complexities pose severe
58 conceptual and implementation problems for lithospheric-scale modeling tools.

59
60 In recent decades, the geodynamic modeling community has accumulated significant
61 experience in the development and application of numerical tools designed for modeling
62 various geodynamic processes. These studies include Christensen and Harder (1991),
63 Weinberg and Schmeling (1992), Bercovici (1993), Poliakov et al. (1993), Braun and
64 Sambridge (1994), Bunge and Baumgardner (1995), Fullsack (1995), Trompert and Hansen
65 (1996), Schmalholz et al. (2001), Babeyko et al. (2002), Tackley and Xie (2003), Moresi et al.
66 (2003), Sobolev et al. (2005), Muhlhaus and Regenauer-Lieb (2005), Petrunin and Sobolev
67 (2006), O'Neil et al. (2006), Gerya and Yuen (2007), Braun et al. (this issue), and Petrunin

68 and Sobolev (this issue), which have all contributed to the development of geodynamic
69 modeling techniques.

70

71 In our research we are mostly interested in highly dynamic deformation processes occurring at
72 plate boundaries involving the combination of transform and compressional/extensional
73 deformation. Such processes are essentially 3D and operate on a temporal scale of a few to
74 several hundred million years and a spatial scale of hundreds to thousands of kilometers.
75 Specific examples include continental collision processes in Tibet and transform deformation
76 at the San Andreas Fault System and the Dead Sea Transform. Numerical techniques to
77 handle such processes must be 3D, should operate with an elasto-visco-plastic rheology, as
78 this is the most adequate rheology to model lithospheric deformation, and must be suitable to
79 model geological time-scale processes. Unfortunately, none of the tools available at present
80 can be directly applied to these problems. The most suitable 3D codes either still do not
81 include elastic deformation processes (Braun et al., this issue) or, being explicit codes
82 (Petrunin and Sobolev, 2006, this issue), require very small steps for time integration and
83 therefore are not efficient for modeling long-term deformation. The best candidate among
84 existing tools is the 3D version of visco-elasto-plastic code I2ELVIS, which is currently under
85 development (Gerya and Yuen, 2007). We note, however, that this code is based on a purely
86 Eulerian approach and uses fully staggered finite-difference discretization. Staggered schemes
87 are inherently stable (Shih et al., 1989), but the Eulerian approach substantially complicates
88 treatment of the free surface.

89

90 With this in mind, we propose here a new code for lithospheric-scale modeling (SLIM3D). It
91 is an implicit Arbitrary Lagrangian Eulerian Particle-in-Cell Finite Element code with a free
92 surface, designed specifically for the thermo-mechanical modeling of deformation processes
93 involving an elasto-visco-plastic lithospheric rheology on geologic time scales. SLIM3D is
94 being designed to model deformation at plate boundaries (the focus of the recently formed
95 geodynamic modeling group in GFZ-Potsdam). It is intended to complement the capability of
96 the explicit code LAPEX3D already used in the same group for similar problems (Petrunin
97 and Sobolev, 2006, this issue). SLIM3D is developed and implemented using the C++ object-
98 oriented programming language.

99

100 We emphasize that the potential advantage of SLIM3D over the other similar codes is in its
101 three-dimensional nature combined with the adequate description of lithospheric deformation

102 (rheology, strain localization, free surface etc.). To our knowledge, such a combination is
 103 unique at present. At the same time, we note that the code is still limited in certain aspects,
 104 which are already implemented in the other codes. These aspects are, for example, the
 105 multigrid solver (e.g. Moresi et al., 2003) and the adaptive mesh (e.g. Braun et al., this issue).
 106 Despite the present disadvantages, we suggest that SLIM3D has a good potential to be
 107 substantially improved in the future versions.

108
 109 This paper is not intended to demonstrate the application of the code to certain geodynamic
 110 problems, but rather to introduce the method in general with several benchmarks and
 111 examples. The structure of the paper is as follows. In the next section, we describe the basic
 112 physical and rheological framework that we utilize for lithospheric scale modeling. In section
 113 3 we outline implementation details of the code. In section 4, we test our tool with five
 114 benchmark problems which demonstrate aspects of elastic, viscous and plastic deformation
 115 mechanisms both separately and in combination. Finally, we briefly conclude our work and
 116 describe the direction we are planning to further develop our technique.

117
 118

119 **2. Physical models**

120
 121

122 *2.1 Conservation equations*

123

124 Lithospheric-scale deformation can be effectively characterized as a quasi-static thermo-
 125 mechanically coupled deformation process. Assuming a continuous media approximation, we
 126 can describe this process by the conservation equations of momentum:

$$127 \quad \frac{\partial \sigma_{ij}}{\partial x_j} + \rho g \hat{z}_i = 0, \quad (1)$$

128 and thermal energy:

$$129 \quad \frac{DU}{Dt} = -\frac{\partial q_i}{\partial x_i} + r. \quad (2)$$

130 Here, x_i ($i = 1, 2, 3$) denote Cartesian coordinates, σ_{ij} is the Cauchy stress tensor, ρ is the
 131 material density, g is the gravitational acceleration, \hat{z}_i is the unit vector of the vertical axis
 132 pointing downward, U is internal energy, D/Dt is the material time derivative, q_i is the

133 heat flux vector, r is volumetric heat sources. In the above equations and in the rest of this
 134 paper, we use indicial notation and apply Einstein summation convention over repeated
 135 indices. Since there is no difference between covariant and contravariant components in the
 136 Cartesian coordinate system, all tensor indices are written as subscripts. For convenience we
 137 additionally explain notation, meaning, and dimension of basic quantities used in this paper in
 138 Table 1.

139

140

[Table 1]

141

142

143 2.2 Deviatoric-volumetric decomposition

144

145 The thermo-rheological behavior of the rocks is more conveniently formulated in terms of the
 146 deviatoric-volumetric strain (stress) decomposition (e.g. Bonet and Wood, 1997). For the
 147 Cauchy stress tensor we may write:

$$148 \quad \tau_{ij} = \sigma_{ij} + p\delta_{ij}, \quad p = -\frac{1}{3}\sigma_{ii}, \quad (3)$$

149 where τ_{ij} is the Cauchy stress deviator and p is hydrostatic pressure (positive in

150 compression). The deviatoric strain rate tensor and the rate of volume change, respectively,
 151 may be written directly as:

$$152 \quad \dot{\epsilon}_{ij} = \frac{1}{2}(\partial v_i / \partial x_j + \partial v_j / \partial x_i) - \frac{1}{3}(\partial v_k / \partial x_k)\delta_{ij}, \quad \dot{\theta} = \partial v_i / \partial x_i, \quad (4)$$

153 where v_i is the spatial velocity vector. We adopt the second (Euclidean) norm as the effective
 154 scalar measure of deviatoric tensorial quantities. For an arbitrary tensor a_{ij} the second norm is
 155 expressed as:

$$156 \quad a_{II} = (a_{ij}a_{ij})^{1/2}. \quad (5)$$

157

158

159 2.3 Continuity equation

160

161 We include the effects of elastic compressibility and thermo-elasticity. In this case, the
 162 continuity equation can be conveniently coupled with the constitutive equation for hydrostatic
 163 pressure:

$$164 \quad \frac{Dp}{Dt} = -K \left(\dot{\theta} - \alpha \frac{DT}{Dt} \right). \quad (6)$$

165 Here, K is the bulk modulus and α is the coefficient of thermal expansion. The presence of
 166 volumetric deformations evokes the following corrections of material density:

$$167 \quad \rho = \rho_0 \left[1 - \alpha(T - T_0) + \frac{P}{K} \right], \quad (7)$$

168 where ρ_0 is the density at reference temperature and zero pressure and T_0 is the reference
 169 temperature. Note that Eq. (7) can be easily replaced by the equation of state applicable for
 170 high pressure and temperature for models involving deep portions of the mantle.

171

172

173 *2.4 Additive decomposition and elasticity*

174

175 Taking benefit from the smallness of *elastic* strains, we adopt conventional additive
 176 decomposition (e.g. Simo and Hughes, 2000) of total deviatoric strain rate. The elastic,
 177 viscous and plastic components, respectively, can be written as follows:

$$178 \quad \dot{\epsilon}_{ij} = \dot{\epsilon}_{ij}^{el} + \dot{\epsilon}_{ij}^{vs} + \dot{\epsilon}_{ij}^{pl} = \frac{1}{2G} \hat{\tau}_{ij} + \frac{1}{2\eta_{eff}} \tau_{ij} + \dot{\gamma} \frac{\partial Q}{\partial \tau_{ij}}, \quad (8)$$

179 where G is the elastic shear modulus, $\hat{\tau}_{ij}$ is the objective stress rate (e.g. Bonet and Wood,
 180 1997), η_{eff} is the effective creep viscosity, $\dot{\gamma}$ is the plastic multiplier, and Q is the plastic
 181 potential function (e.g. Simo & Hughes, 2000). Note that for numerical integration of elastic
 182 stresses over the finite time step, we use the incrementally objective scheme of Hughes and
 183 Winget (1980). For more details see the next section of this paper.

184

185

186 *2.5 Ductile creep*

187

188 We adopt a detailed description of the ductile deformation component. The total viscous
 189 strain rate is additively decomposed into three temperature- and stress-dependent creep
 190 mechanisms, namely diffusion creep, dislocation creep and Peierls creep (see e.g. Kameyama
 191 et al., 1999). The corresponding effective creep viscosity is given by:

$$192 \quad \eta_{eff} = \frac{1}{2} \tau_{II} (\dot{\epsilon}_L + \dot{\epsilon}_N + \dot{\epsilon}_P)^{-1}, \quad (9)$$

193 where τ_{II} is effective differential stress (see Eq. 5), and $\dot{\epsilon}_L$, $\dot{\epsilon}_N$ and $\dot{\epsilon}_P$ are the effective scalar
 194 strain rates due to the diffusion, dislocation and Peierls mechanisms, respectively. Specific
 195 expressions for each strain rate can be written as follows:

$$196 \quad \dot{\epsilon}_L = B_L \tau_{II} \exp\left(-\frac{H_L}{RT}\right), \quad (10)$$

$$197 \quad \dot{\epsilon}_N = B_N (\tau_{II})^n \exp\left(-\frac{H_N}{RT}\right), \quad (11)$$

$$198 \quad \dot{\epsilon}_P = B_P \exp\left[-\frac{H_P}{RT}(1-\beta)^2\right] \left(\frac{\tau_{II}}{\beta\tau_p}\right)^s, \quad (12)$$

199 where

$$200 \quad s = 2\beta(1-\beta)\frac{H_P}{RT}. \quad (13)$$

201 In the above equations, B_L , B_N , B_P and H_L , H_N , H_P denote the creep parameter and
 202 activation enthalpy, respectively, of each correspondent mechanism, R is the gas constant, n
 203 is the power law exponent, τ_p is the Peierls stress, and $0 < \beta < 1$ is an adjustable
 204 approximation parameter. Note that in Eq. (12) we have adopted an asymptotic approximation
 205 of Peierls mechanism (see Kameyama *et al.* 1999), since the original equation is inappropriate
 206 for stresses below $O(10^2)$ MPa.

207

208 Here we assume that the creep parameter and activation enthalpy are constant for all creep
 209 mechanisms. In the general case, the creep parameter can depend on grain size (Karato *et al.*,
 210 2001), and activation enthalpy may be sensitive to pressure (Regenauer-Lieb and Yuen,
 211 2004). If necessary, these complexities can be easily introduced in the presented numerical
 212 formulation.

213

214 In Fig. 1, we plot the effective logarithmic viscosity versus temperature and differential stress,
 215 assuming parameters for dry olivine from Kameyama *et al.* (1999). Each creep mechanism
 216 dominates over the others (produces a higher strain rate) in different temperature-stress
 217 domains. In Fig. 1, these domains are separated by black solid lines and labeled. Note that for
 218 olivine, most of effective viscosity reduction from extremely high (10^{27} Pas) to extremely low
 219 (10^{18} Pas) values occurs within a relatively narrow temperature range between 500 and 1000
 220 C for all stress levels.

221

222

[Figure 1]

223

224 *2.6 Brittle failure*

225

226 We describe brittle failure of rocks by the classical Mohr–Coulomb plasticity model (see e.g.
227 Vermeer, 1990). The expression for the Mohr–Coulomb yield surface can be written as:

$$228 \quad F = \frac{1}{2}(\sigma_{\max} - \sigma_{\min}) + \frac{1}{2}(\sigma_{\max} + \sigma_{\min}) \sin \varphi - c \cos \varphi \leq 0, \quad (14)$$

229 where σ_{\max} and σ_{\min} are the maximum and minimum principal stresses (negative in
230 compression), φ is the material angle of friction, c is cohesion, $\frac{1}{2}(\sigma_{\max} - \sigma_{\min})$ is the
231 maximum differential stress, and $\frac{1}{2}(\sigma_{\max} + \sigma_{\min})$ is the normal stress.

232

233 In the principal stress space, the Mohr-Coulomb yield surface can be represented as a
234 hexagonal pyramid with a singular apex point in the tensile domain. Physically, this
235 singularity means that the Mohr-Coulomb yield surface is inappropriate to model tensile
236 failure of the rocks. A more adequate description in the context of lithospheric deformation
237 would require coupling with the continuity equation to account for large plastic dilatation.
238 Such sophisticated treatment of the tensile failure is not yet available. Therefore in this paper
239 we adopt a standard ad-hoc approach and approximate the Mohr-Coulomb yield surface in the
240 tensile domain with the Tresca criterion which is given by:

$$241 \quad F = \frac{1}{2}(\sigma_{\max} - \sigma_{\min}) - c \cos \varphi \leq 0. \quad (15)$$

242 The resulting composite yield surface is shown in Fig. 2.

243

244 [Figure 2]

245

246 Associative Mohr–Coulomb in 3D is essentially a multi-surface plasticity model which has
247 so-called corner regions (Simo et al., 1988) in both yield surface and flow potential.
248 Integration of such a model constitutes a particularly nontrivial task (Sloan and Booker, 1986;
249 Larsson and Runesson, 1996; Borja et al., 2003). Another issue is related to the associativity
250 of the flow rule, which significantly overestimates plastic dilatation of the rocks (e.g. Alejano
251 and Alonso, 2005). We resolve both these issues by adopting the purely deviatoric corner-free
252 Prandtl-Reuss flow rule (e.g. Zienkiewicz and Taylor, 2000), which is non-associative with
253 the Mohr–Coulomb yield surface. In this case the plastic potential function takes the
254 following simple form:

$$255 \quad Q = \tau_{II}. \quad (16)$$

256 The Prandtl-Reuss flow rule assumes complete plastic incompressibility (i.e. dilatation angle
257 is zero), which is a suitable approximation for rocks in the large strain regime. At the same

258 time, the integration of the plasticity model becomes considerably simpler due to a lack of
 259 corners in the plastic potential.

260

261 We approximate the degradation of the strength in faults and gouge zones by the strain
 262 softening model. The friction angle is assumed in the following product form:

$$263 \quad \varphi = \varphi_0 D(\kappa), \quad (17)$$

264 where φ_0 is initial friction and $D(\kappa)$ is the function that controls degradation of the friction
 265 angle with progressive increase of the accumulated plastic strain. We assume $D(\kappa)$ in
 266 piecewise linear form. The accumulated plastic strain is given by:

$$267 \quad \kappa = \int_t \left(\dot{\varepsilon}_{ij}^{pl} \dot{\varepsilon}_{ij}^{pl} \right)^{1/2} dt. \quad (18)$$

268 We note that the adopted plasticity model with strain softening does not incorporate any
 269 length scale for the strain localization (e.g. Muhlhaus and Aifantis, 1991). Thus the theoretical
 270 thickness of the shear band is zero. In the numerical model the thickness of the shear band is
 271 limited from below by the element size. Therefore in the context of an adaptively refined grid
 272 one must incorporate the length-scale or certain regularization (e.g. Belytschko and Tabbara,
 273 1993) to prevent the element size from reducing to very small values. In this paper we do not
 274 explicitly address this mesh-dependence issue, since at present we only consider uniform non-
 275 adaptive grids.

276

277

278 *2.7 Heat flow*

279

280 At present, we neglect latent heat effects due to phase change and assume internal energy in
 281 the form of linear function of temperature, i.e.:

$$282 \quad U = C_p T, \quad (19)$$

283 where C_p is specific heat.

284

285 We define the heat flux vector according to Fourier law as follows:

$$286 \quad q_i = -\lambda \delta_{ij} \frac{\partial T}{\partial x_j}, \quad (20)$$

287 where $\lambda \delta_{ij}$ is the isotropic thermal conductivity tensor.

288

289 The volumetric heat sources include radiogenic heat and heat produced by visco-plastic
 290 deformation, i.e.:

$$291 \quad r = \rho A + \chi \tau_{ij} (\dot{\epsilon}_{ij}^{vs} + \dot{\epsilon}_{ij}^{pl}), \quad (21)$$

292 where A is radiogenic heat per unit mass and $0 \leq \chi \leq 1$ is the constant regulating degree of
 293 thermo-mechanical feedback. This coupling constant is an ad-hoc approach that effectively
 294 accounts for the processes which we do not include in our models, such as heat transport by
 295 the fluids, etc.

296

297

298 *2.8 Boundary conditions*

299

300 The conservation equations (1) and (2) must be complemented with initial and boundary
 301 conditions. This may include specified velocity, temperature, stress and heat flux.

302 Additionally, in the context of lithospheric-scale modeling it becomes necessary to
 303 approximate the boundary conditions at the base of the lithosphere. In this paper we use the
 304 classical Winkler foundation (see e.g. Fig. 5 in Regenauer-Lieb, 2006), which assumes zero
 305 viscous drag forces and takes into account buoyancy forces. The boundary stress tensor of the
 306 Winkler foundation can be parameterized as follows:

$$307 \quad \bar{\sigma}_{ij} = -[p_0 + \rho_{ext} g(z - z_0)] \delta_{ij}, \quad (22)$$

308 where p_0 is pressure at the reference surface, ρ_{ext} is the density of the external material, and
 309 z and z_0 are the vertical coordinates of the bottom boundary and the reference surface,
 310 respectively. Typically, we use the initial position of the bottom boundary as the reference
 311 surface. The pressure at the reference surface is assigned as total weight divided by the depth
 312 of the model.

313

314 Other non-standard boundary conditions include more complicated effects such as erosion on
 315 the free surface and material in-flux and out-flux. These effects are considered in more detail
 316 in the next section.

317

318

319 **3. Numerical algorithms**

320

321

322 3.1 Spatial discretization

323

324 We build numerical models of lithospheric-scale deformation on the basis of the Finite
 325 Element Method (e.g. Hughes, 1987; Zienkiewicz and Taylor, 2000; Belytschko et al., 2000).
 326 Following the Galerkin procedure (e.g. Belytschko et al., 2000), we transform the momentum
 327 balance equation (1) into a corresponding nodal force residual:

$$328 \quad f_{ii} = \int_{\Omega} \frac{\partial N_I}{\partial x_j} \sigma_{ij} d\Omega - \int_{\Omega} N_I \rho g \hat{z}_i d\Omega - \int_{\Gamma} N_I \bar{\sigma}_{ij} n_j d\Gamma = 0. \quad (23)$$

329 Similarly, we approximate the energy balance equation (2) by a nodal power residual:

$$330 \quad w_I = \int_{\Omega} N_I \rho \frac{DU}{Dt} d\Omega - \int_{\Omega} \frac{\partial N_I}{\partial x_i} q_i d\Omega - \int_{\Omega} N_I r d\Omega + \int_{\Gamma} N_I \bar{q}_i n_i d\Gamma = 0. \quad (24)$$

331 Here N denotes the nodal shape function (Zienkiewicz and Taylor, 2000), I is the nodal
 332 index, $\bar{\sigma}_{ij}$ and \bar{q}_i are surface stress and surface heat flux, respectively, n_i is the outward unit
 333 normal, and Ω and Γ stand for the volume and surface of the domain.

334

335 We employ hexahedral finite elements with linear interpolation functions (e.g. Zienkiewicz
 336 and Taylor, 2000) to approximate volume integrals in the above equations, and similar
 337 quadrilateral elements to approximate surface integrals (see Fig. 3). Element integrals are
 338 expressed in terms of parametric coordinates and evaluated over the unit cube or square using
 339 numerical integration (e.g. Belytschko et al., 2000). This stage is directly followed by
 340 assembly of the global residual equations (23) and (24) from element contributions (e.g.
 341 Hughes, 1987).

342

343

[Figure 3]

344

345 In each element, the shape function derivatives with respect to global coordinates are
 346 computed using standard coordinate transformation (summation over repeated nodal indices is
 347 implied):

$$348 \quad \frac{\partial N_I}{\partial x_i} = J_{ij}^{-1} \frac{\partial N_I}{\partial \xi_j}, \quad J_{ij} = \frac{\partial N_I}{\partial \xi_i} x_{jI}, \quad (25)$$

349 where ξ_i are the local coordinates and J_{ij} denotes Jacobian matrix. The unit outward normal
 350 vector is evaluated according to:

$$351 \quad n_i = J^{-1} \hat{n}_i, \quad J = (\hat{n}_i \hat{n}_i)^{1/2}, \quad (26)$$

352 where J stands for surface Jacobian and \hat{n}_i is the cross-product between two surface tangent
 353 vectors, which can be expressed as follows:

$$354 \quad \hat{n}_i = \varepsilon_{ijk} t_j^{(1)} t_k^{(2)}, \quad t_i^{(a)} = \frac{\partial N_I}{\partial \xi_a} x_{iI}, \quad a = 1, 2. \quad (27)$$

355 Here $t_i^{(a)}$ are the surface tangent vectors and ε_{ijk} is the Levi-Civita permutation symbol. To
 356 ensure a positive determinant of the Jacobian matrix and outward orientation of the unit
 357 normal vector, we perform appropriate control over the numbering of nodes in the element.

358

359

360 *3.2 Locking and Hourglass*

361

362 In nearly incompressible problems the finite element mesh is prone to locking, which
 363 manifests itself in severe underestimation of velocities/displacements (e.g. Belytschko et al.,
 364 2000). Locking is cured either by suitable under-integration of stress terms (Malkus and
 365 Hughes, 1978) or by using higher interpolation order for displacements (velocities) than for
 366 the stresses (e.g. Zienkiewicz and Taylor, 2000). Both approaches are essentially equivalent.

367

368 We suppress locking by evaluating constitutive equations only in one point per element
 369 (Flanagan and Belytschko, 1981). This approach implies that strains (strain rates) and other
 370 spatial gradients are evaluated using the following element-average derivatives of the shape
 371 functions:

$$372 \quad b_{ii} = \frac{1}{V_E} \int_{\Omega} \frac{\partial N_I}{\partial x_i} d\Omega. \quad (28)$$

373 where V_E is the element volume. For element integrals, we use standard quadrature.

374

375 The hexahedral elements employed here violate the so-called LBB stability condition named
 376 after Ladyzhenskaya (1969), Babuska (1973) and Brezzi (1974), which leads to mesh artifacts
 377 known as “hourglass modes” (e.g. Flanagan and Belytschko, 1981) (see also Fig. 4a). Despite
 378 this, we use them together with careful monitoring of the stress and displacements fields,
 379 because the more stable quadratic element is computationally very expensive in 3D problems.
 380 With the same element resolution, quadratic interpolation requires roughly an order of
 381 magnitude more nodes (degrees of freedom) than the linear interpolation.

382

383 We have tested some of the anti-hourglass techniques available for hexahedral (quadrilateral)
 384 and tetrahedral (triangular) elements (see e.g. Flanagan and Belytschko, 1981; Liu et al.,
 385 1998; Bonet et al., 2001; Reese, 2003; Puso and Solberg, 2006). In certain problems, such as
 386 the Rayleigh-Taylor instability with large abrupt viscosity variations (see Fig. 4), none of
 387 stabilization technique we tested was able to prevent the hourglass modes. We have inferred
 388 that the critical factor is the relatively high confining pressure compared to deviatoric stress in
 389 the low viscosity domain.

390

391

[Figure 4]

392

393 The magnitude of the confining pressure can be reduced by replacing the material density
 394 with the differential density $\Delta\rho = \rho - \rho_{ref}$. Here ρ_{ref} denotes the arbitrary chosen constant
 395 reference density. With the differential density approach, the numerical solution produces the
 396 dynamic pressure Δp . To evaluate the constitutive equations properly, it becomes necessary
 397 to augment the dynamic pressure with a lithostatic component $\rho_{ref} g (z - z_{free})$. Here z and
 398 z_{free} denote the vertical coordinates of the integration point and the free surface, respectively.

399

400 The density of the external material must also be set to the differential value
 $\Delta\rho_{ext} = \rho_{ext} - \rho_{ref}$. Similarly, the Winkler condition must be applied at the Lagrangian free
 401 surface with the differential density equal to negative reference density i.e. $\Delta\rho_{air} = -\rho_{ref}$.

402

403 According to our experience, this simple rearrangement of the computational scheme is
 404 sufficient to remove the hourglass modes even in the Rayleigh-Taylor instability problem
 405 mentioned above. Figs. 4a and 4b show the mesh distortion during a typical time step for the
 406 total and differential density formulations, respectively. The total density formulation exhibits
 407 uncontrollable oscillation while differential density formulation behaves stably.

408

409 We note, however, that despite the fact that the differential density technique can extend
 410 stability margins in certain problems, there is no guarantee that it will work in other problems.
 411 Therefore, we suggest that this technique should be used only in conjunction with careful
 412 monitoring of the stress and displacements fields. Additionally, one can use spatial averaging
 413 over the neighboring nodes (e.g. Fulsack, 1995; Braun et al., this issue) to enhance the
 414 stability.

415

416 As an alternative to finite elements with bilinear interpolation for velocities and constant
 417 pressure, fully staggered finite difference discretization can be used (Harlow and Welsh,
 418 1965), which appear to be non-oscillatory (for discussion, see Shih et al., 1989). Successful
 419 application of this discretization in geodynamic problems is presented in Gerya and Yuen
 420 (2007). However this approach, despite stability, is restricted to pure Eulerian formulations. In
 421 lithospheric scale modeling, we consider this restriction as unfavorable because it complicates
 422 tracking of the free surface.

423

424

425 *3.3 Time discretization and primary variables*

426

427 We use the backward Euler method as the primary time discretization algorithm for both the
 428 momentum and energy balance equations. This approach is referred to as implicit and first
 429 order accurate. The first order accuracy brings advantageous stability, which is missed by the
 430 second order accurate trapezoidal rule (Ortiz and Popov, 1985). In the adopted context,
 431 integration of all quantities becomes particularly simple, e.g. for velocities we can write:

$$432 \quad \Delta u_i = v_i \Delta t, \quad (29)$$

433 where Δu_i is incremental displacement vector and $\Delta t = {}^{n+1}t - {}^n t$ is the time step. Here the left
 434 superscript indicates the time step index. In the remainder of this paper we omit the index of
 435 the current time step for notational clarity, e.g. we simply write: $\Delta t = t - {}^n t$.

436

437 The simplicity of the integration scheme allows us to choose either velocities or *incremental*
 438 displacements as primary kinematical variables, irrespective of the employed rheology or
 439 kinematical formulation. Indeed, it is clear from the integration formula (29) that both
 440 approaches are essentially equivalent. For convenience we prefer incremental displacements.
 441 Note that all results presented below are readily extendable to the velocity formulation merely
 442 by simple scaling.

443

444 We do not separate pressure as an independent variable, since we use a compressible
 445 formulation. However, in typical problems the near-incompressible behavior is recovered in
 446 the asthenosphere domain. We treat this additional difficulty during the linear solution stage
 447 by appropriate damping of the stiffness parameters (see section 3.8 for more details). For the
 448 heat balance equation we use temperature as the primary variable.

449

450

451 *3.4 Kinematical formulation and solution scheme*

452

453 We employ the Arbitrary Lagrangian-Eulerian kinematical formulation (Hirt et al., 1974) to
454 account for material advection. A typical calculation step is subdivided into three major
455 stages:

456 (i) In the first stage, we solve the discretized residual equations (23) and (24) by the Newton
457 Raphson iterative method (e.g. Belytschko et al., 2000). During iterations we
458 conveniently treat advection terms implicitly by means of the Updated Lagrangian
459 formulation (Bathe et al., 1975). The advantage is that Lagrangian treatment does not
460 limit the time step by the amount of rigid body motion, in contrast with the Eulerian
461 approach. In the Lagrangian context, only the amount of material straining poses an
462 actual kinematic limit for the time step.

463 (ii) In the second stage, we perform mesh adaptation (regridding) such that the new mesh fits
464 the free surface, the moving (stretching) calculation window and simple material
465 interfaces without overturns or self-intersections. The flexibility of the approach is that
466 regridding can be done either in each step or after a certain number of steps, which is
467 favorable in the case that advection is not the dominant type of nonlinearity in the system
468 (e.g. the onset of plastic localization). Regridding is also the easiest way to implement
469 boundary material fluxes (erosion, sedimentation, etc.).

470 (iii) In the third and final stage, we perform consistent remapping of all solution variables
471 onto the updated mesh. Our algorithm is based on the particle-in-cell approach which was
472 initially developed by Harlow and Welsh (1965) and subsequently has been widely used
473 in geodynamic applications (Moresi et al., 2003; Tackley and Xie, 2003; O'Neil et al.,
474 2006; Gerya and Yuen, 2007).

475 We explain each calculation stage in more detail in the following sections. The overall
476 computational flowchart is summarized in Table 2.

477

478

[Table 2]

479

480 *3.5 Hughes-Winget scheme*

481

482 A constitutive model of advection dominated flow that includes memory effects (e.g.
 483 Fullsack, 1995) should be objective, i.e. it should ensure conservation of tensor quantities
 484 (e.g. Muhlhaus and Regenauer-Lieb, 2005). This can be formally achieved by using time-
 485 continuous (infinitesimal) objective stress rates (e.g. Bonet and Wood, 1997). However, since
 486 our practical interest is implicit large-step integration of the flow, we use the Hughes-Winget
 487 scheme (Hughes-Winget 1980) to ensure objectivity in an incremental, rather than an
 488 infinitesimal, sense.

489

490 The central point of the scheme is the following second order accurate approximation for an
 491 incremental displacement gradient:

$$492 \quad h_{ij} = \frac{\partial \Delta u_i}{\partial x_j^{n+1/2}}, \quad x_j^{n+1/2} = x_j^n + \frac{1}{2} \Delta u_j, \quad (30)$$

493 where $x_j^{n+1/2}$ is the mid-point material configuration. In each element, we compute h_{ij} using
 494 the mid-point element-average derivatives of shape functions (see Eq. 28) as follows:

$$495 \quad h_{ij} = \Delta u_{il} \, b_{jl}^{n+1/2}. \quad (31)$$

496 We note that the mid-point scheme in Eq. (30) effectively eliminates the influence of rigid
 497 body motion on the strain components (Hughes and Winget, 1980). This property contrasts
 498 with pure Eulerian formulation, which is restricted to differentiation only with respect to the
 499 current configuration.

500

501 Deviatoric and volumetric strain *increments* to be used in constitutive equations are
 502 computed, respectively, as follows:

$$503 \quad \varepsilon_{ij} = \frac{1}{2} (h_{ij} + h_{ji}) - \frac{1}{3} h_{kk} \delta_{ij}, \quad \theta = h_{ii}. \quad (32)$$

504 Correspondent rate quantities are approximated by:

$$505 \quad \dot{\varepsilon}_{ij} = \frac{\varepsilon_{ij}}{\Delta t}, \quad \dot{\theta} = \frac{\theta}{\Delta t}. \quad (33)$$

506 According to the Hughes-Winget scheme, rotation of stress from the previous time step is
 507 governed by the following orthogonal tensor (for more details see Hughes and Winget, 1980):

$$508 \quad R_{ij} = \left(\delta_{ik} - \frac{1}{2} \omega_{ik} \right)^{-1} \left(\delta_{kj} + \frac{1}{2} \omega_{kj} \right), \quad \omega_{ij} = \frac{1}{2} (h_{ij} - h_{ji}). \quad (34)$$

509 The approximations (30) – (34), despite their simplicity, produce stable and accurate results
 510 even for large strain and rotation increments (e.g. Rashid, 1993).

511

512

513 3.6 Incremental constitutive equations

514

515 Incremental expansion of the continuity equation (6) gives the following updated pressure, if
516 the reactional (phase transformation) volume changes are neglected:

$$517 \quad p = {}^n p - K\theta + K\alpha(T - {}^n T). \quad (35)$$

518

519 To update the deviatoric stress, we use a two-step predictor-corrector procedure (e.g. Simo
520 and Hughes, 2000). In the first step, the plastic multiplier in equation (8) is assumed to be
521 zero and the correspondent trial visco-elastic stress is evaluated. In the second step, the Mohr-
522 Coulomb yield surface is checked (Eq.14), and if violated, the trial stress is modified to
523 remain on the yield surface. This gives rise to a nonzero plastic multiplier and corresponding
524 plastic strain increment.

525

526 Combining the Hughes-Winget scheme with the analytical integration of equation (8) over the
527 time step we obtain the following expression for trial deviatoric stress:

$$528 \quad \tau_{ij}^{tr} = 2\eta_{CR}\dot{\varepsilon}_{ij} + \alpha_{CR} R_{ik} {}^n \tau_{kl} R_{jl}, \quad (36)$$

529 where ${}^n \tau_{kl}$ is the Cauchy stress deviator from previous time step, R_{ij} is the incremental
530 rotation tensor (see Eq. 34), and η_{CR} and α_{CR} are the effective viscosity and relaxation ratio
531 of visco-elastic creep, respectively, given by:

$$532 \quad \eta_{CR} = \eta_{eff} (1 - \alpha_{CR}), \quad \alpha_{CR} = \exp\left(-\frac{\Delta t}{t_M}\right), \quad t_M = \frac{\eta_{eff}}{G}. \quad (37)$$

533 Here t_M stands for Maxwell time.

534

535 Equation (36) shows that the magnitude of Maxwell time does not pose the upper limit for the
536 time step (see also Bailey, 2006). Indeed, in typical calculations we choose time steps equal to
537 tens of thousands years. In the asthenosphere, where temperature is high enough to
538 sufficiently decrease the viscosity, the following purely viscous limit is recovered:

$$539 \quad \frac{\Delta t}{t_M} \gg 1, \quad \lim_{\alpha_{CR} \rightarrow 0} \tau_{ij}^{tr} = 2\eta_{eff}\dot{\varepsilon}_{ij}. \quad (38)$$

540 In the upper crustal domain, on the other hand, the relatively low temperatures make the
541 viscosity so large that Equation (36) degenerates into the corresponding elastic end-member:

$$542 \quad \frac{\Delta t}{t_M} \ll 1, \quad \lim_{\alpha_{CR} \rightarrow 1} \tau_{ij}^{tr} = 2G\varepsilon_{ij} + R_{ik} {}^n \tau_{kl} R_{jl}. \quad (39)$$

543 However, due to the presence of the Mohr-Coulomb stress limiter, most of deformation in the
544 upper crustal domain manifests itself in the form of plastic localization.

545

546 Apart from temperature dependence, the effective viscosity depends also on stress, and this
547 introduces nonlinearity into the equation (36). This issue can be resolved, in principle, with
548 good stress estimation from the previous time step. However, we have concluded from our
549 experiments that this explicit approach severely restricts stable calculation steps due to stress
550 oscillations. The alternative is, therefore, implicit treatment of nonlinearity by solving the
551 nonlinear equation (36) for a trial deviatoric stress.

552

553 The simplest fixed point iteration cannot be used to this end, since in our case it may produce
554 infinite loops without convergence. Hence, we have adopted the simple and reliable root-
555 solving algorithm FZERO (Shampine and Watts, 1970), which is based on a combination of
556 the bisection and secant methods.

557

558 Prior to solution we rewrite the tensorial equation (36) in scalar residual form by taking the
559 difference between norms of its left- and right hand sides. The solution is achieved when the
560 residual is reduced to a sufficiently small relative tolerance. Typically, very few iterations
561 (around 5-7) are required by FZERO to converge.

562

563 For numerical stability reasons, we remove stress dependence in the effective viscosity at
564 differential stress levels higher than $O(10^3)$ MPa, which is far higher than expected in typical
565 problems. Finally, in the numerical simulations we truncate effective viscosity to the
566 reasonable range $10^{18} - 10^{27}$ Pas.

567

568 After the visco-elastic stress predictor step, we check the yield surface to determine whether a
569 plastic stress correction is required. In the following, we introduce basic elements of the
570 procedure, whereas more details are outlined in Appendix A.

571

572 Plastic stress correction starts with spectral decomposition of the trial deviatoric stress tensor
573 by the optimized 3D Jacobi eigenvalue algorithm (Press et al., 2002):

$$574 \quad \tau_{ij}^{tr} = \sum_{A=1}^3 \tau_A^{tr} m_{ij}^{(A)}. \quad (40)$$

575 Here τ_A^{tr} denotes the trial principal deviatoric stresses, $m_{ij}^{(A)}$ are the correspondent spectral
576 directions.

577

578 In the next stage, we evaluate the trial yield surface (see Appendix A):

$$579 \quad F^{tr} = \frac{1}{2}(\tau_{\max}^{tr} - \tau_{\min}^{tr}) + \frac{1}{2}(\tau_{\max}^{tr} + \tau_{\min}^{tr})\sin\varphi - p\sin\varphi - c\cos\varphi. \quad (41)$$

580 For simplicity, the friction angle in the above equation is computed using the magnitude of
581 accumulated plastic stain from previous time step, i.e. $\varphi = \varphi_0 D({}^n\kappa)$.

582

583 As soon as the trial deviatoric stress departs from the yield surface ($F^{tr} > 0$ condition is met),
584 we undertake the following correction:

$$585 \quad \tau_{ij} = \alpha_{PL} \tau_{ij}^{tr}, \quad (42)$$

586 where $0 < \alpha_{PL} < 1$ is the plastic scaling ratio, which is given by:

$$587 \quad \alpha_{PL} = \frac{2(p\sin\varphi + c\cos\varphi)}{(1 + \sin\varphi)\tau_{\max}^{tr} - (1 - \sin\varphi)\tau_{\min}^{tr}}. \quad (43)$$

588 For simplicity, we assume that both the effective viscosity of visco-elastic creep and the trial
589 stress remain unaltered during plastic stress correction.

590

591 The magnitude of plastic strain rate is readily computed as follows:

$$592 \quad \dot{\gamma} = \frac{1}{2\eta_{CR}}(1 - \alpha_{PL})\tau_{II}^{tr}. \quad (44)$$

593 Integrating the above expression by the Backward Euler algorithm we obtain the following
594 relation for the plastic strain increment:

$$595 \quad \Delta\kappa = \frac{\Delta t}{2\eta_{CR}}(1 - \alpha_{PL})\tau_{II}^{tr}, \quad (45)$$

596 which produces asymptotically exact results for both viscous and elastic cases. Corresponding
597 limit expressions are given by:

$$598 \quad \lim_{\alpha_{CR} \rightarrow 0} \frac{\Delta t}{2\eta_{CR}} = \frac{\Delta t}{2\eta_{eff}}, \quad \lim_{\alpha_{CR} \rightarrow 1} \frac{\Delta t}{2\eta_{CR}} = \frac{1}{2G}. \quad (46)$$

599 Accumulated plastic strain for the next time step is updated by:

$$600 \quad \kappa = {}^n\kappa + \Delta\kappa. \quad (47)$$

601

602 For the Updated Lagrangian kinematical formulation, the material and partial time derivatives
 603 are coincident. We can therefore approximate the rate of internal energy in each integration
 604 point by the following first order difference:

$$605 \quad \frac{DU}{Dt} = C_p \frac{T_I N_I - {}^n T_I N_I}{\Delta t}. \quad (48)$$

606
 607 For consistency with the Hughes-Winget scheme, we evaluate the spatial temperature gradient
 608 with respect to the mid-point material configuration (see Eq. 30). Thus, the finite element
 609 approximation for the heat flux vector can be written as (see Eq. 28):

$$610 \quad q_i = -\lambda T_I^{n+1/2} b_{iI}. \quad (49)$$

611

612 Heat source due to dissipation of mechanical energy can be re-expressed as:

$$613 \quad \tau_{ij} (\dot{\epsilon}_{ij}^{vs} + \dot{\epsilon}_{ij}^{pl}) = \tau_{II} \left(\frac{\tau_{II}}{2\eta_{eff}} + \dot{\gamma} \right). \quad (50)$$

614 The basic steps of the stress update procedure are summarized in Table 3. We note that this
 615 sequence is repeated every iteration of every time step in each integration point.

616

617 [Table 3]

618

619 3.7 Nonlinear solver

620

621 At each time step, we solve the coupled system of discretized residual equations (23) and (24)
 622 by the full Newton-Raphson iterative method (Belytschko et al., 2000). We neglect coupling
 623 terms in the Jacobian matrix. Therefore, at each iteration we solve the following two systems
 624 of linear equations:

$$625 \quad \delta \mathbf{u} = -\mathbf{K}_{(k)}^{-1} \mathbf{f}_{(k)}, \quad \Delta \mathbf{u}_{(k+1)} = \Delta \mathbf{u}_{(k)} + \delta \mathbf{u}, \quad (51)$$

$$626 \quad \delta \mathbf{T} = -\mathbf{E}_{(k)}^{-1} \mathbf{w}_{(k)}, \quad \mathbf{T}_{(k+1)} = \mathbf{T}_{(k)} + \delta \mathbf{T}. \quad (52)$$

627 Here, the right subscript (k) denotes iteration index, \mathbf{f} is the out-of-balance nodal force vector
 628 (Eq. 23), \mathbf{w} is the out-of-balance nodal power vector (Eq. 24), $\Delta \mathbf{u}$ is the incremental
 629 displacement (total velocity) vector, \mathbf{T} is the nodal temperature vector, $\mathbf{K} = \partial \mathbf{f} / \partial \Delta \mathbf{u}$ is the
 630 mechanical tangent matrix, $\mathbf{E} = \partial \mathbf{w} / \partial \mathbf{T}$ is the thermal tangent matrix, and $\delta \mathbf{u}$ and $\delta \mathbf{T}$ are
 631 iterative correction vectors for incremental displacement and temperature, respectively.

632

633 At each iteration, we update nodal coordinates similarly to incremental displacements
 634 $\mathbf{x}_{(k+1)} = \mathbf{x}_{(k)} + \delta \mathbf{u}$, while all the element integrals are evaluated over the latest updated
 635 coordinates (Updated Lagrangian formulation).

636

637 We achieve a coupled thermo-mechanical solution by:

- 638 (i) Calculating effective viscosity and density using the latest updated temperature,
- 639 (ii) Calculating mechanical dissipation using the latest available stress and strain rate,
- 640 (iii) Simultaneously solving the force balance and energy balance equations.

641

642 In the context of the Newton-Raphson method, it is necessary to carefully select the initial
 643 guess, since it can significantly reduce computation time. At the first time step we take zero
 644 displacements and an initial temperature distribution. For each subsequent time step we use
 645 the previous converged solution.

646

647 Constrained degrees of freedom are removed from the tangent matrices. To facilitate
 648 convergence *only* at the first iteration of the first time step, we add corrections related to
 649 removed degrees of freedom to the right hand side vectors.

650

651 We employ the so-called line search procedure (see e.g. Press et al., 2002; Crisfield, 1983) to
 652 stabilize iterative solution of the mechanical equation. The displacement update formula with
 653 line search is modified to the following form:

$$654 \quad \Delta \mathbf{u}_{(k+1)} = \Delta \mathbf{u}_{(k)} + \alpha \delta \mathbf{u} . \quad (53)$$

655 Here $0 < \alpha < 1$ is the damping parameter, which is chosen to satisfy the following condition:

$$656 \quad \|\mathbf{f}_{(k+1)}\| < \|\mathbf{f}_{(k)}\| . \quad (54)$$

657 We bisect the time step and restart the overall equilibrium iteration in case the above criterion
 658 cannot be fulfilled. This technique forms the core of our adaptive time-stepping algorithm.

659 Initially we specify the uniform time step; the algorithm then performs bisections whenever
 660 convergence problems are experienced. When iteration becomes stable again, the algorithm
 661 tries to increase the time step. However it keeps in memory the last converged solution, to
 662 restore stable iteration in the case of unsuccessful time step increases. To further enhance
 663 convergence in the strain localization problems, more sophisticated time-stepping methods
 664 may be used such as the method of subplane control functions (Geers, 1999).

665

666 We terminate the Newton-Raphson iteration as soon as the following criteria are satisfied:

$$667 \quad \|\delta \mathbf{u}\| < \varepsilon \|\Delta \mathbf{u}_{(k+1)}\| \quad \text{and} \quad \|\delta \mathbf{T}\| < \varepsilon \|\mathbf{T}_{(k+1)}\|, \quad (55)$$

668 where ε is a small tolerance, typically $\varepsilon = 10^{-3}$.

669

670

671 3.8 Linearization and linear solver

672

673 We assume the following simplified approximations for the tangent matrices (Appendix B):

$$674 \quad K_{ijkl} = \int_{\Omega} \frac{\partial N_I}{\partial x_j} C_{ijkl} \frac{\partial N_J}{\partial x_i} d\Omega + \int_{\Gamma} N_I (\rho_{ext} \mathbf{g} \cdot n_z \hat{z}_i \hat{z}_k) N_J d\Gamma, \quad (56)$$

$$675 \quad E_{IJ} = \int_{\Omega} N_I \frac{\rho C_p}{\Delta t} N_J d\Omega + \int_{\Omega} \frac{\partial N_I}{\partial x_i} \lambda \frac{\partial N_J}{\partial x_i} d\Omega. \quad (57)$$

676 Here, C_{ijkl} is the material tangent operator (Simo and Taylor, 1985), n_z is the vertical

677 component of outward normal vector, and \hat{z}_i are the components of the unit vector of the

678 vertical axis (downward positive). The first term in Eq. (56) corresponds to linearization of

679 the internal stress term, while the second term is caused by the Winkler boundary condition.

680

681 Depending on type of the flow (visco-elastic or elasto-visco-plastic), the material tangent

682 operator takes the following simple form:

$$683 \quad C_{ijkl} = K \delta_{ij} \delta_{kl} + \begin{cases} 2G_{CR} I_{ijkl}^D & \text{if } F^{tr} \leq 0 \\ 2G_{PL} I_{ijkl}^D + \tau_{ij}^{tr} \hat{g}_{kl} & \text{if } F^{tr} > 0 \end{cases} \quad (58)$$

684 Here $I_{ijkl}^D = \frac{1}{2} (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk}) - \frac{1}{3} \delta_{ij} \delta_{kl}$ is the fourth order unit deviatoric tensor, $G_{CR} = \eta_{CR} / \Delta t$ is

685 the effective visco-elastic shear modulus, and $G_{PL} = \alpha_{PL} G_{CR}$ is the effective elasto-visco-

686 plastic shear modulus. The non-dimensional tensor \hat{g}_{kl} is given in Appendix B.

687

688 We infer from our numerical experiments that anisotropic stress terms in the tangent operator

689 (Eq. 58) have little influence on the convergence rate of the Newton-Raphson iteration.

690 Therefore, we suggest that they can be omitted whenever necessary. With this assumption, the

691 effective elasto-visco-plastic shear modulus (G_{PL}) becomes essentially equivalent to the

692 effective viscosity (see e.g. Eq. 37 of Moresi et al., 2003 or Fig. 4 of Fullsack, 1995).

693

694 Both these papers employ the fixed point method or direct (Picard) iteration (Zienkiewicz and
695 Taylor, 2000) as a nonlinear solver. Even omitting differences in the tangent operator, there
696 are two substantial differences between the Newton-Raphson and Picard methods:

697 (i) The Picard method has only external forces and stresses from the previous time step in
698 the right-hand-side, while the Newton-Raphson method has complete out-of-balance
699 forces.

700 (ii) The Picard method operates on total velocity (incremental displacement), while the
701 Newton-Raphson method operates on iterative corrections.

702 We have tried both Picard and Newton-Raphson methods. Fig. 5 shows a comparison between
703 the two solvers for the shear band initiation benchmark problem, discussed in the next section.
704 We calculated an identical setup by both methods with a gradually increasing number of
705 increments in which we applied a constant amount of extension. Fig. 5 shows the severe
706 sensitivity of the Picard method to increment size. At the same time, the Newton-Raphson
707 method produces accurate results even for comparatively large increments. Other authors (e.g.
708 Muhlhaus and Regenauer-Lieb, 2005) have also pointed out superiority of the Newton-
709 Raphson method over the Picard method.

710

711 [Figure 5]

712

713 At the current development stage, we use the Preconditioned Conjugate Gradient (PCG)
714 iterative method (Hestenes, Stiefel, 1952) for both the mechanical and thermal linear systems
715 (Eq. 51 & Eq.52). For the mechanical system we adopt a symmetric version of the Incomplete
716 LU factorization with Threshold (ILUT) preconditioner (Saad, 1994), equipped with near-
717 optimal nested dissection ordering (George, 1973). The thermal system is preconditioned by
718 the inverse of diagonal of thermal tangent matrix.

719

720 The above algorithm, in general, is not numerically scalable (Farhat et al., 2000), i.e. the
721 number of iterations grows nonlinearly with increasing size of the linear system. At present
722 the algorithm can only run in sequential mode. However, it works well with the roughly 10^5
723 grid nodes in typical 3D regional-scale models on ordinary PCs with 2 GB RAM.

724

725 Note that the linear solver may fail in the low-viscosity domains (e.g. asthenosphere), where
726 effective the visco-elastic modulus becomes many orders of magnitude lower than the bulk
727 modulus. We treat this issue by supplementing the mechanical linear solver with the fictitious

728 compressibility technique (Zienkiewicz and Taylor, 2000). The fictitious bulk modulus is
729 made at maximum two orders of magnitude larger than the shear modulus in the integration
730 point. The algorithm then iteratively removes spurious volumetric strains until the original
731 continuity equation is satisfied.

732

733

734 *3.9 Regridding and Remapping*

735

736 After the Newton Raphson step, we resolve the advection terms of the flow. This step is
737 implemented by means of regridding and remapping. The computational sequence of adopted
738 procedure is summarized in the second part of Table 2.

739

740 We employ a specific type of the Arbitrary Lagrangian Eulerian kinematical formulation (Hirt
741 et al., 1997). It is characterized by using a structured Cartesian grid whose nodes are only
742 allowed to move in the vertical direction. With this flexibility, we track the free surface or
743 certain material interfaces inside the model domain. Using the same mechanism we
744 implement erosion and sedimentation processes on the free surface. The bottom surface of the
745 grid is equipped with the Winkler boundary condition.

746

747 Lateral boundaries of the grid may be either fixed in space or move parallel to the coordinate
748 axis with constant or variable velocity. This is a particularly useful feature, since it creates a
749 flexible computation window which can track some portion of space, or even stretch or
750 contract laterally. When the length of the mesh in a certain direction changes by the size of
751 the original element, we add (or remove) an entire vertical slice of the elements. Our solvers
752 are designed to treat a variable number of elements during computation.

753

754 To track material properties and solution history we use the Particle-in-Cell method (Harlow
755 and Welsh, 1965). Each marker has an individual history, including coordinates, material
756 number, pressure, deviatoric stress tensor, accumulated plastic strain, total displacements, and
757 temperature. During the Newton-Raphson step, the history accumulates in cell centers. When
758 markers are advected by mesh, all markers that belong to the cell obtain the same *increment*
759 of history. When markers are mapped back to the mesh, both material properties and history
760 are averaged among all markers in the updated cell. This incremental approach minimizes
761 spurious numerical diffusion.

762

763 We advect markers in the displacement field by consistent interpolation from the mesh. Since
764 the mesh remains rectangular in horizontal plain, the mapping of markers to updated cell
765 positions becomes particularly simple. Certain difficulties exist in regions where a population
766 of markers becomes too dense or too sparse (Moresi et al., 2003). We treat this issue by
767 inserting or deleting markers in cells where the total number of markers falls beyond the
768 limits. We treat material fluxes on the boundaries with the same methodology. When a marker
769 leaves the domain, we simply delete it. When a marker is generated in the boundary cell it
770 acquires material properties and solution history from the closest neighbors. We suggest,
771 therefore, that material extends beyond all boundaries of the domain except the free surface.

772

773 We note that the remapping procedure inevitably produces perturbations in the global force
774 balance. We treat these perturbations as initial residuals to be removed on the next time step.
775 This explicit approach to a certain extent reduces the accuracy of the numerical solution
776 (stresses, displacements). Nevertheless, it is common in geodynamics to use similar
777 techniques with the argument that the time step can be reduced if higher accuracy in the time
778 integration is required (e.g. Moresi et al. 2003). In section 4.2 we demonstrate that the
779 remapping procedure does not significantly affect the accuracy of the velocity field, which is
780 estimated analytically, even in the benchmark with hundreds of remapping cycles.

781

782

783 **4. Benchmarks and examples**

784

785 In this section, we present five simplified problems to verify computer implementation and
786 applicability of the adopted physical models and numerical techniques. The following
787 problems are included:

788

(i) Bending of an elastic plate

789

(ii) Sinking of a rigid cylinder in a viscous fluid

790

(iii) Initiation of shear bands in the brittle crust

791

(iv) Triaxial compression test

792

(v) Lithospheric transpressional deformation

793

794 The first two problems are designed to test the code's ability to separately handle elastic and
795 viscous rheological mechanisms in 2D. In the first problem, the 2D formulation (plane strain)

796 is chosen to facilitate comparison with the analytical solution, while in the second problem,
 797 the choice is based on the resolution issue. With the sequential solver and without mesh
 798 adaptation, the solution of the free-falling Stokes sphere problem with the same resolution
 799 would require too much computation time. The next two problems demonstrate quasi-brittle
 800 elasto-plastic strain localization in 2D and 3D, respectively. The final problem deals with the
 801 complete elasto-visco-plastic rheology in a typical 3D lithospheric-scale setup. Since the plain
 802 strain formulation is not currently available in SLIM3D, we discretize entire 2D setups by a
 803 single layer of hexahedral elements with zero velocity in the out-of-plane direction. Wherever
 804 necessary we introduce and track markers in the mid-surfaces of the elements.

805

806

807 *4.1 Bending of an elastic plate*

808

809 We consider bending of an elastic plate which has infinite length in one of the lateral
 810 dimensions. The setup of the problem is depicted in Fig. 6, and the geometry, loading and
 811 elastic constants of the plate are summarized in Table 4. Similar to a cantilever beam, the
 812 plate has one end fixed and the other end free. It has slightly negative buoyancy relative to
 813 surrounding material, which allows it to sink and bend. Since we are interested in modeling
 814 elastic deformation of the plate, we exclude the surrounding viscous material. Calculations are
 815 done in the Lagrangian mode without remeshing.

816

817 [Table 4]

818

819 In this benchmark, we compare the maximum vertical deflection of the plate and the
 820 maximum bending stress with the analytical solution of the Euler-Bernoulli beam equation
 821 (see Appendix C). Application of this equation requires displacements and thickness to be
 822 small compared with length of the plate. We fulfill this requirement by appropriately selecting
 823 parameters of the setup. Assuming validity of the Euler-Bernoulli approximation, the
 824 maximum deflection of plate can be expressed in terms of parameters in Table 4 as:

$$825 \quad w = \frac{3\Delta\rho gl^4}{2Eh^2}. \quad (59)$$

826 For comparison with the numerical solution, we calculate analytical bending stresses using the
 827 following relation (see Appendix C):

$$828 \quad \sigma_b = \pm \frac{3\Delta\rho g l^2 (h - \Delta h)}{h^2}, \quad (60)$$

829 where Δh is the element size along the height of the beam.

830

831 We apply and remove the gravity load multiple times in this model. Each time, when the load
 832 is removed, we allow the plate to restore its original shape and simultaneously release stored
 833 elastic stresses. During the solution, we monitor spurious residual stresses, which tend to
 834 accumulate during multiple load cycles. This deficiency is related to the rate formulation of
 835 elasticity adopted in SLIM3D (see Eq. 8). Unlike exact large strain elasticity models (Bonet
 836 and Wood, 1997), this formulation produces nonzero dissipation of elastic energy along the
 837 closed deformation path (Belytschko et al., 2000). The applicability of the model to rocks is
 838 based on the fact that elastic strains remain small, while elastic stresses constantly dissipate
 839 either by plastic or viscous rheological mechanisms. In this benchmark, we numerically assess
 840 spurious dissipation of elastic energy.

841

842 [Figure 6]

843

844 Fig. 6 shows the shape of the deformed plate (exaggerated by factor of 10) along with a
 845 contour plot of normal bending stresses. Key results of the benchmark are presented in Table
 846 4. We investigate influence of discretization on the accuracy of the solution using different
 847 numbers of elements along the plate thickness starting with just two elements (since we have
 848 a single integration point per element). In all cases, the 1 : 2 (thickness vs. length) element
 849 aspect ratio is maintained within the vertical plane. Fig. 7 shows the relative error in the
 850 approximation of stresses and displacements versus number of elements along the plate
 851 thickness. Note that discretization converges relatively fast, providing acceptable accuracy
 852 (around 5.0% error) with just 4-5 elements across the plate. The results in Table 4 correspond
 853 to discretization by 7 elements across the plate. Further increasing the number of elements did
 854 not yield a substantial increase in the accuracy in neither the vertical deflection (2.6% error)
 855 nor in the bending stresses (0.8% error). We attribute this fact partially to inaccuracies of the
 856 approximate analytical solution and partially to the presence of shear locking (e.g. Belytschko
 857 et al., 2000) of the employed elements. After 10 cycles of cyclic loading, the residual stress
 858 remains about 0.1% of peak bending stress (0.3 MPa vs. 300.0 MPa). Thus, we conclude that
 859 the stress integration algorithm has sufficient accuracy for practical applications.

860

861 [Figure 7]

862

863

864 *4.2 Sinking of a rigid cylinder in a viscous fluid*

865

866 The purpose of this benchmark is twofold: (i) we test the ability to simulate purely viscous
 867 flow with large abrupt changes of the viscosity; (ii) we simultaneously assess the quality of
 868 the remeshing algorithm (treatment of markers and numerical diffusion issue). Setup of the
 869 problem includes one half of a rectangular domain which is 800 km wide and 700 km deep
 870 (see Fig. 8) occupied by viscous fluid. We impose sticking between the domain walls and the
 871 fluid. On the left side of the domain we impose a symmetry condition. An infinite cylinder
 872 with 25 km radius is initially placed at 250 km depth. Due to negative buoyancy the cylinder
 873 sinks into the fluid. We maintain relative rigidity of the cylinder by a 4 orders of magnitude
 874 viscosity contrast with the surrounding fluid. Parameters of the setup are summarized in Table
 875 5. Numerical solution is obtained with a time step 10^4 years.

876

877 [Table 5]

878

879 In this benchmark, we compare analytical and numerical estimation of the cylinder velocity.
 880 The analytical solution for resisting force due to motion of the cylinder in a reservoir of finite
 881 size can be found in Slezkin (1955). By equating resistance force with Archimedes force one
 882 can obtain the following expression for the velocity of the cylinder (see Appendix D):

$$883 \quad v_c = \frac{1}{4} \left(\ln k - \frac{k^2 - 1}{k^2 + 1} \right) \frac{r^2 \Delta \rho g}{\eta_f}, \quad k = \frac{b}{r}, \quad (61)$$

884 where η_f is fluid viscosity, r is the radius of cylinder and b is the characteristic distance
 885 between the cylinder and rigid wall of the domain. In our setup this distance is equal to 375
 886 km.

887

888 [Figure 8]

889

890 We plot the distribution of vertical velocity together with positions of cylinder markers at two
 891 different times in Fig.8 (see also Animation 1 in the online version of the paper). Note that
 892 through the multiple remeshing steps, cylinder shape shows completely no sign of numerical
 893 diffusion. The cylinder remains perfectly circular and undisturbed through the course of

894 computation. We therefore conclude an acceptable quality of adopted remeshing algorithm.
 895 The average velocity of the cylinder constitutes around 8.5 cm per year. The maximum
 896 relative difference between the analytical and numerical estimations is less than 1.7% (see
 897 Table 5). Error in the numerical solution for the velocity field remains relatively low, despite
 898 hundreds of remapping cycles and a large time step (10^4 years). Increase of viscosity contrast
 899 up to 7 orders of magnitude did not result in any substantial changes in the numerical solution.
 900 We conclude that the adopted numerical formulation allows handling of purely viscous flow
 901 with large abrupt variations of the viscosity.

902
 903

904 *4.3 Initiation of shear bands in the brittle crust*

905

906 Formation of zones of localized shear (shear bands) during planar deformation has been the
 907 subject of extensive research (see e.g. Arthur et al., 1977; Muhlhaus and Vardoulakis, 1987;
 908 Vermeer, 1990; Borja and Aydin, 2004; Buitter et al., 2006; Kaus and Podladchikov, 2006).
 909 One of the most important questions concerns preferred orientation of the shear bands. Three
 910 basic models are generally discussed (Coulomb, 1773; Roscoe, 1970; and Arthur et al., 1977).
 911 For a pressure-sensitive material with friction angle φ and dilatation angle ψ , these models
 912 predict the following angles between the shear band and direction of minor principal stress:

$$\begin{aligned}
 &\alpha = 45^\circ + \varphi / 2 && \text{(Coulomb),} \\
 913 \quad &\alpha = 45^\circ + \psi / 2 && \text{(Roscoe) ,} \\
 &\alpha = 45^\circ + (\varphi + \psi) / 4 && \text{(Arthur).}
 \end{aligned} \tag{62}$$

914 The theoretical analysis for the onset of shear banding presented by Vermeer (1990) also
 915 yields the Roscoe and Coulomb angles as limits for possible shear-band orientation. However,
 916 due to elastic unloading the shear band can, in principle, assume any orientation within this
 917 range.

918

919 In this benchmark, we tested the capability to model initiation of shear bands and compare
 920 inclination angles of modeled shear bands with analytical estimations and numerical solutions
 921 obtained by other codes. Since only the onset of shear bands is considered, calculations are
 922 done in the small strain regime. The large strain plastic localization in 3D is additionally
 923 demonstrated in the next subsection.

924

925 The model setup includes a crustal domain which is 40 km wide and 7 km thick (see Fig. 9).
 926 On the top boundary we impose a free surface, while the bottom boundary has zero normal
 927 velocity and tangential free-slip. We consider two loading cases. On the lateral boundaries,
 928 we prescribe 1 cm per year normal velocity such that it causes either shortening or extension
 929 of the domain. We assume an elasto-plastic rheology typical of mafic crust (see Table 6).
 930 Since we employ the purely deviatoric Prandtl-Reuss flow rule, the dilatation angle is zero in
 931 our formulation.

932

933 [Table 6]

934

935 We measure inclination angles of the shear bands from the horizontal axis. In the case of
 936 compression the orientation of the minor principal stress is vertical. Therefore the angle with
 937 respect to horizontal axis should be computed with a minus sign; for example, the Coulomb
 938 angle should be $\alpha = 45^\circ - \varphi/2$. In the case of extension, the minor principal stress is
 939 horizontal; therefore sign should be plus i.e. $\alpha = 45^\circ + \varphi/2$.

940

941 [Figure 9]

942

943 Fig. 9 shows the contour plot of effective strain rate for both extension (a, b) and compression
 944 (c, d) cases at different stages of the process (see also Animation 2 in the online version of the
 945 paper). The shear bands start to propagate from the surface where strength is minimal. We
 946 note that it takes significantly less extension than shortening for shear bands to cut the entire
 947 domain. The initial pattern of shear bands is highly chaotic. Some of them have inclination
 948 angles close to Coulomb angles (30° , 60°) at this stage. At later stages, only a few shear
 949 bands survive. They begin to accommodate entire deformation caused by shortening /
 950 extension such that other shear bands become completely inactive. The mature post-failure
 951 shear bands precisely match the Arthur angles (38° , 53°). In both cases (compression or
 952 tension), shear bands at the Roscoe angles (45°) did not occur in the model.

953

954 Our results are in good agreement with results of the other codes. For example, Poliakov and
 955 Herrmann (1994) solved a similar problem using a non-associative Mohr-Coulomb rheology
 956 and explicit FLAC technique. In their solution, the shear bands formed spontaneously and
 957 assumed various inclination angles ranging between the Roscoe and Coulomb limits. The
 958 results obtained with I2ELVIS (Gerya and Yuen, 2007) for the numerical sandbox experiment

959 (Buiter et al., 2006) revealed that the majority of shear bands in the numerical solution tend to
960 assume the Arthur angle. We conclude that our code is able to effectively simulate origination
961 and propagation of localized shear zones.

962

963

964 *4.4 Triaxial compression test*

965

966 Elasto-plastic strain localization in the large strain regime is further demonstrated by
967 numerical simulation of the triaxial compression test. We consider a laterally unconfined
968 parallelepiped specimen of size $1 \times 1 \times 3$ km discretized into $20 \times 20 \times 60$ hexahedral elements,
969 subjected to kinematical axial compressional loading (Fig. 10a and Fig. 11a). Material
970 parameters are assumed from the previous benchmark (see Table 6). We specify two different
971 types of heterogeneity: either small random seeds uniformly distributed over the specimen
972 (Fig. 10a) or an oblique gash located near the edge as shown in Fig. 11a. In both cases,
973 localization is induced by relatively low cohesion of the seeds with respect to surrounding
974 material (down to 20% of nominal).

975

976

[Figure 10]

977

978 The type of heterogeneity significantly impacts the pattern of deformation observed in the
979 model. Thus, the random seeding induces formation of multiple shear bands as shown in Figs.
980 10b – 10d (see also Animation 3 in the online version of the paper). Despite that the specimen
981 exhibits fragmentation into small pieces, one can easily distinguish major failure plane, which
982 gradually evolves though the course of loading. In contrast, the gash-type seeding
983 immediately produces a complex curvilinear failure plane, which separates the entire
984 specimen into two parts (Figs. 11b – 11d). The majority of the shear band inclination angles
985 also coincide with the Arthur angles. Results of the triaxial compression test support the
986 conclusions of previous benchmark for the 3D case and the large strain regime.

987

988

[Figure 11]

989

990

991 *4.5 Lithospheric transpressional deformation*

992

993 In this example, we present a typical 3D lithospheric-scale setup which includes effects of
994 elasto-visco-plastic rheology with a temperature controlled brittle-ductile transition. The
995 model consists of a lithospheric domain which has lateral dimensions 200 by 600 km and
996 depth 80 km, as shown in Fig. 12a. The domain is discretized into $36 \times 108 \times 16$ hexahedral
997 elements (68561 grid points). Compositional heterogeneity includes equally thick (15 km)
998 felsic upper crust and mafic lower crust and a 50 km thick peridotite mantle. Rheological and
999 thermal parameters used in the modeling are presented in Table 7. At elongated vertical
1000 boundaries we impose zero normal velocities as well as 3.5 cm/year of right-lateral strike-slip
1001 velocity. The upper and lower boundaries are the free surface and the Winkler support
1002 boundary, respectively. At the short vertical boundaries we impose lithostatic pressure along
1003 with a free-slip condition.

1004

1005 [Table 7]

1006

1007 The initial temperature distribution is shown in Fig. 12b. We use different temperature
1008 gradients in the upper crust, and the lower crust with lithospheric mantle to account for
1009 radiogenic heat. In the asthenosphere, we assume a temperature of 1300°C . The lithosphere-
1010 asthenosphere boundary is initially placed at 70 km depth. To initiate the state of
1011 transpression in the model, we use a zone of slightly thinned lithosphere (10 km of thinning),
1012 which is oriented obliquely to the strike-slip direction.

1013

1014 [Figure 12]

1015

1016 Fig. 12c shows a contour plot of the effective strain rate on the free surface and in a typical
1017 cross-section after 3 Myr of evolution and 105 km of strike-slip (see also Animation 4 in the
1018 online version of the paper). The far-field strike-slip deformation tends to localize in a narrow
1019 zone located above the initially thinned lithosphere where strength is minimal. This
1020 mechanism leads to the self-consistent formation of the boundary between two lithospheric
1021 plates. The structure of the plate boundary changes with depth from a very narrow shear zone
1022 in the brittle upper crust to progressively more diffuse and wide shear zone in the ductile
1023 lower crust and upper mantle. The brittle upper crust tends to form regular secondary
1024 structures (en-echelon faults). Fig. 12d shows a contour plot of material phases and isolines of
1025 vertical displacements. The transpressional character of strike-slip deformation leads to

1026 surface uplift on pressure ridges related to en-echelon structures as well as corresponding
1027 crustal thickening along the entire plate boundary.

1028

1029 Fig. 12e shows the influence of temperature and stress on the effective viscosity distribution.
1030 The overall viscosity variation in this model is about 7 orders of magnitude, with most of this
1031 variation caused by gradual temperature increase with depth and compositional
1032 heterogeneities. Among the local effects, we note the effects of shear heating and nonlinear
1033 creep which come into play across the entire plate boundary. Relative stress peaks associated
1034 with both the upper-lower crust boundary and the Moho, commonly referred to as “christmas-
1035 trees” (e.g. Brace and Kohlstedt 1980), can be recognized in Fig. 12f. Note that due to strain
1036 softening, the strength of major faults becomes low, which is also visible from the stress
1037 distribution on the surface (Fig. 12f).

1038

1039 Fig. 13 show a 6 Myr evolution of the Lagrangian mesh embedded in the material (see also
1040 Animation 5 in the online version of the paper). We note that material fluxes on the boundary
1041 of computational box are nonzero. Incoming material acquires the properties and solution
1042 history from the closest material inside the box. The magnitude of the strike-slip displacement
1043 is large (210 km) and comparable to the characteristic size of the model. The transpressional
1044 plate boundary accumulates significant shear strains during the evolution. Summarizing these
1045 results we suggest that our code, in principle, can be applied for 3D modeling of geological
1046 scale evolution of lithospheric plate boundaries.

1047

1048 [Figure 13]

1049

1050

1051 **5. Summary and discussion**

1052

1053

1054 *5.1 What is next?*

1055

1056 Currently, the code SLIM3D already includes many techniques necessary for modeling
1057 deformation processes on lithospheric and geologic scales. Namely, we have implemented
1058 coupled thermo-mechanical solutions, realistic elasto-visco-plastic rheology, and a kinematic
1059 framework that is able to handle large deformations. However, in this section we discuss the

1060 techniques which are still missing in the code. We also point out the directions we plan to
1061 develop our code to satisfy the growing needs of the new modeling section in GFZ-Potsdam.

1062

1063 The major disadvantage is that code is still severely restricted in problem size. This fact is
1064 explained mostly by the following: (i) the code is still sequential (ii) the code utilizes a linear
1065 solver which is not numerically scalable (Farhat et al., 2000). To remove these disadvantages,
1066 we plan to extend our code in the following mainstream directions of linear solver
1067 technology:

- 1068 (i) Direct parallel multifrontal solvers (Gupta et al., 1997; Amestoy et al., 2000)
- 1069 (ii) Multigrid methods with adaptation for plasticity (Adams, 2000; Ekevid et al., 2004)
- 1070 (iii) Dual-primal domain decomposition methods (Farhat et al., 2000)

1071 At the same time, we are planning to parallelize the entire finite element routines using the
1072 MPI package.

1073

1074 Deformation processes in the lithosphere occur on highly variable spatial scales. To
1075 sufficiently resolve the material length scale in lithospheric models, the element size should
1076 be of the order of hundreds of meters (Regenauer-Lieb and Yuen, 2003). Therefore, a
1077 successful code for lithospheric-scale modeling should allow variable spatial resolution. In
1078 SLIM3D we are presently restricted to a structured non-adaptive grid. A possible way to
1079 introduce adaptivity is to switch to unstructured tetrahedral mesh generators (Rassineux,
1080 1998) or octree-based methods (Shephard et al., 1991; Braun et al., this issue). These will be
1081 investigated in the future.

1082

1083 Apart from advancements in numerical methods, we are planning to further develop physical
1084 models of rocks introducing advanced damage rheological models and incorporating
1085 metamorphic reactions and melting as well as coupled porous flow. Furthermore, we plan to
1086 introduce coupling between deformation processes at regional and global spatial scales as
1087 well as at seismic-cycle and geological time scales.

1088

1089

1090 *5.2 Summary remarks*

1091

1092 This paper presents a new tool (SLIM3D) for three-dimensional lithospheric-scale modeling.
1093 The code includes a coupled thermo-mechanical treatment of deformation process and allows

1094 complicated elasto-visco-plastic rheology with diffusion, dislocation and Peierls creep and
1095 Mohr-Coulomb plasticity. The code incorporates an Arbitrary Lagrangian Eulerian
1096 formulation with free surface and Winkler boundary conditions.

1097

1098 We have developed practical implementation of elasto-visco-plastic rheology and shown that
1099 Maxwell time does not pose an upper limit for the integration time step in the numerical
1100 scheme.

1101

1102 We have applied the full Newton-Raphson method for solution of discrete balance equations
1103 and demonstrated its superiority over the Picard method, which is widely used in
1104 geodynamics.

1105

1106 For plasticity model we have obtained simplified expressions for the tangent operator.
1107 Anisotropic stress terms in tangent operator have little influence on the convergence rate of
1108 the Newton-Raphson iteration. Therefore they can be omitted whenever necessary. With this
1109 assumption, the difference between effective viscosity and tangent operator concepts
1110 vanishes.

1111

1112 With linear sequential solvers, which are not numerically scalable, we are able to treat
1113 problems with up to hundred thousand of grid points on ordinary PCs. It is expected that
1114 capabilities of the method can be drastically expanded by implementing numerically scalable
1115 parallel solvers (e.g. multigrid) as well as an adaptive mesh.

1116

1117

1118 **Acknowledgments**

1119

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1122 appreciate constructive and helpful reviews offered by Stefan Schmalholz and an anonymous
1123 reviewer.

1124

1125

1126 **Appendix A. Plastic stress correction**

1127

1128 We recall the spectral decomposition of Cauchy stress tensor (e.g. Borja et al. 2003):

$$1129 \quad \sigma_{ij} = \sum_{A=1}^3 \sigma_A m_{ij}^{(A)}, \quad m_{ij}^{(A)} = \hat{n}_i^{(A)} \hat{n}_j^{(A)}, \quad (A1)$$

1130 where σ_A denotes principal stresses (negative in compression), $m_{ij}^{(A)}$ are the correspondent
1131 spectral direction tensors, and $\hat{n}_i^{(A)}$ are the principal direction vectors.

1132

1133 Deviatoric-volumetric stress decomposition holds in principal stress space as well, therefore
1134 we can write:

$$1135 \quad \sigma_A = \tau_A - p, \quad (A2)$$

1136 where τ_A are the principal deviatoric stresses.

1137

1138 The Mohr-Coulomb yield surface can be rewritten using (A2) as follows:

$$1139 \quad F = \frac{1}{2}(\tau_{\max} - \tau_{\min}) + \frac{1}{2}(\tau_{\max} + \tau_{\min}) \sin \varphi - p \sin \varphi - c \cos \varphi \leq 0. \quad (A3)$$

1140

1141 The plastic stress correction according to the Prandtl-Reuss flow rule reduces to the simple
1142 radial stress return (e.g. Wilkins, 1964). Practical implementation consists of the following
1143 proportional scaling of trial principal deviatoric stresses:

$$1144 \quad \tau_A = \alpha_{PL} \tau_A^{tr}, \quad (A4)$$

1145 where $0 < \alpha_{PL} < 1$ is the plastic scaling ratio, which can be directly determined by substituting
1146 (A4) into the yield surface expression (A3) and enforcing the condition $F = 0$. The result is:

$$1147 \quad \alpha_{PL} = \frac{2(p \sin \varphi + c \cos \varphi)}{(1 + \sin \varphi) \tau_{\max}^{tr} - (1 - \sin \varphi) \tau_{\min}^{tr}}. \quad (A5)$$

1148

1149 According to the Prandtl-Reuss flow rule, the updated stress deviator remains co-axial with
1150 the trial stress deviator, therefore we can update directly (omitting spectral decomposition) as:

$$1151 \quad \tau_{ij} = \alpha_{PL} \tau_{ij}^{tr}. \quad (A6)$$

1152

1153 Using the identities:

$$1154 \quad \tau_{II} = \alpha_{PL} \tau_{II}^{tr}, \quad \tau_{II} = \tau_{II}^{tr} - 2\eta_{CR} \dot{\gamma}, \quad (A7)$$

1155 we can readily solve for the magnitude of plastic strain rate:

$$1156 \quad \dot{\gamma} = \frac{1}{2\eta_{CR}} (1 - \alpha_{PL}) \tau_{II}^{tr}. \quad (A8)$$

1157

1158

1159 **Appendix B. Linearization**

1160

1161 To complete formulation of the Newton-Raphson method, it is necessary to obtain an
 1162 expression for the tangent matrices $\partial \mathbf{f} / \partial \Delta \mathbf{u}$ and $\partial \mathbf{w} / \partial \mathbf{T}$. Unfortunately, in general, it poses a
 1163 difficult algebraic task. The resulting expressions are always non-symmetric except for trivial
 1164 cases (Simo and Hughes, 2000). Moreover, in geodynamical modeling it is necessary to
 1165 simulate nonlinear effects like erosion, sedimentation or melt emplacement, which often
 1166 cannot be linearized analytically. Numerical differentiation may overcome this difficulty (see
 1167 e.g. Perez-Foguet et al., 2000). In this paper, however, we admit certain assumptions to obtain
 1168 approximate linearization, which still possesses acceptable convergence properties.

1169

1170 First, we assume that the integrals in the discrete balance equations Eq. (23) & Eq. (24) are
 1171 configuration-independent. This assumption substantially simplifies linearization and avoids
 1172 additional non-symmetry in the tangent matrices. In this case, the linearization of the energy
 1173 equation becomes trivial. The result reads:

$$1174 \quad E_{,I} = \frac{\partial w_I}{\partial T_J} = \int_{\Omega} N_I \frac{\rho C_p}{\Delta t} N_J d\Omega + \int_{\Omega} \frac{\partial N_I}{\partial x_i} \lambda \frac{\partial N_J}{\partial x_i} d\Omega. \quad (\text{B1})$$

1175 We note that during Newton-Raphson iteration, all integrals in the above equation are
 1176 evaluated over latest available coordinates.

1177

1178 Linearization of internal stress term yields:

$$1179 \quad \frac{\partial}{\partial \Delta u_{k,l}} \left(\frac{\partial N_I}{\partial x_j} \sigma_{ij} \right) = \frac{\partial N_I}{\partial x_j} (C_{ijkl} + L_{ijkl}) \frac{\partial N_J}{\partial x_i}. \quad (\text{B2})$$

1180 The term $C_{ijkl} = \partial \sigma_{ij} / \partial \hat{\varepsilon}_{kl}$ is referred to as the (material) tangent operator (Simo & Taylor
 1181 1985). The term $L_{ijkl} = \partial \sigma_{ij} / \partial \omega_{kl}$ can be called the rotation tangent operator. Here $\hat{\varepsilon}_{kl}$ denotes
 1182 the total strain increment in contrast with the deviatoric strain increment ε_{kl} , which appears in
 1183 Eq. (32). Recalling the expression for trial deviatoric stress (Eq. 36), the rotation operator can
 1184 be expanded as follows:

$$1185 \quad L_{ijkl} = \frac{\partial \sigma_{ij}}{\partial \omega_{kl}} = \frac{\partial}{\partial R_{pq}} \left(\alpha_{CR} R_{im}^n \tau_{mn} R_{jn} \right) \frac{\partial R_{pq}}{\partial \omega_{kl}}. \quad (\text{B3})$$

1186 After straightforward, but quite cumbersome algebra, which we omit here
 1187 (<http://matrixcookbook.com>), one obtains the following expression for the rotation tangent
 1188 operator:

$$1189 \quad L_{ijkl} = \frac{1}{4} \alpha_{CR} (W_{ik} Q_{jl} + Q_{il} W_{jk} - W_{il} Q_{jk} - Q_{ik} W_{jl}), \quad (B4)$$

1190 where the auxiliary tensors are given by:

$$1191 \quad W_{ij} = (\delta_{ij} - \frac{1}{2} \omega_{ij})^{-1}, \quad Q_{ij} = R_{ik} \sigma_{kl} (R_{jl} + \delta_{jl}). \quad (B5)$$

1192 Expression (B4) lacks major symmetry i.e. $L_{ijkl} \neq L_{klij}$. Moreover, we note that it becomes
 1193 identically zero in the viscous case since $\alpha_{CR} \rightarrow 0$.

1194

1195 We note that a similar approach was presented e.g. by Fish and Shek (1999). They rigorously
 1196 linearized all geometrical nonlinearities related to the Hughes-Winget scheme together with
 1197 configuration dependence. From their results, we conclude that potential advantages from
 1198 completely consistent linearization are still marginal, taking into account additional burdens
 1199 related to the solution of non-symmetric systems. Therefore, we suggest omitting the rotation
 1200 operator from the tangent matrix approximation. In the case that a non-symmetric solver is
 1201 available, one can use the unmodified expression (B4). Alternatively, some kind of
 1202 symmetrization can be employed, for example:

$$1203 \quad L_{ijkl}^{\text{sym}} = \frac{1}{2} (L_{ijkl} + L_{klij}). \quad (B6)$$

1204

1205 The material tangent operator $C_{ijkl} = \partial \sigma_{ij} / \partial \widehat{\varepsilon}_{kl}$ forms two cases depending on type of the
 1206 flow, either visco-elastic or elasto-visco-plastic. It is convenient to decompose the material
 1207 tangent operator into volumetric and deviatoric parts. Applying the chain rule we can write:

$$1208 \quad C_{ijkl} = \frac{\partial \sigma_{ij}}{\partial \widehat{\varepsilon}_{kl}} = \frac{\partial \tau_{ij}}{\partial \varepsilon_{mn}} \frac{\partial \varepsilon_{mn}}{\partial \widehat{\varepsilon}_{kl}} - \delta_{ij} \frac{\partial p}{\partial \theta} \frac{\partial \theta}{\partial \widehat{\varepsilon}_{kl}}. \quad (B7)$$

1209 First, we consider a nonlinear visco-elastic deformation process. In this case, we use
 1210 expressions for updated pressure (Eq. 35) and trial deviatoric stress (Eq. 36). Motivated by the
 1211 moderate stress dependence of the effective creep viscosity, we assume $\partial \eta_{CR} / \partial \tau_{ij} = 0$ to
 1212 avoid cumbersome differentiation of the expressions (9) – (12) together with expressions (37).
 1213 The resultant material tangent operator related to nonlinear visco-elastic deformation reads:

$$1214 \quad C_{ijkl} = K \delta_{ij} \delta_{kl} + 2G_{CR} I_{ijkl}^D. \quad (B8)$$

1215 Here, I_{ijkl}^D is the fourth order unit deviatoric tensor:

$$1216 \quad I_{ijkl}^D = \frac{1}{2}(\delta_{ik}\delta_{jl} + \delta_{il}\delta_{jk}) - \frac{1}{3}\delta_{ij}\delta_{kl}, \quad (B9)$$

1217 and G_{CR} is the effective shear modulus of visco-elastic creep:

$$1218 \quad G_{CR} = \eta_{CR} / \Delta t. \quad (B10)$$

1219 We note that G_{CR} has appropriate viscous and elastic limits given by:

$$1220 \quad \lim_{\alpha_{CR} \rightarrow 0} G_{CR} = \eta_{eff} / \Delta t, \quad \lim_{\alpha_{CR} \rightarrow 1} G_{CR} = G. \quad (B11)$$

1221

1222 Next, we consider the elasto-visco-plastic deformation process. Recalling the plastic stress
1223 correction formula (Eq. 42) and applying the chain rule, we expand the deviatoric term in the
1224 following form:

$$1225 \quad \frac{\partial \tau_{ij}}{\partial \hat{\epsilon}_{kl}} = \underbrace{2G_{PL} I_{ijkl}^D}_{\text{isotropic}} + \underbrace{\tau_{ij}^{tr} \hat{g}_{kl}}_{\text{anisotropic}}, \quad (B12)$$

1226 where G_{PL} is the effective elasto-visco-plastic shear modulus (similar to the corresponding
1227 effective viscosity; see e.g. Moresi et al., 2003; Fullsack, 1995)

$$1228 \quad G_{PL} = \alpha_{PL} G_{CR}, \quad (B13)$$

1229 and \hat{g}_{kl} is non-dimensional second order tensor which is given by:

$$1230 \quad \hat{g}_{kl} = \frac{\partial \alpha_{PL}}{\partial \epsilon_{mn}} I_{mnlk}^D. \quad (B14)$$

1231 We note that the elasto-visco-plastic tangent operator has both isotropic and anisotropic
1232 components. The latter destroys major symmetry similar to the rotation operator, so that the
1233 non-symmetric linear solver is again required. From our numerical experiments we infer that
1234 symmetric isotropic approximation (the first term in Eq. B12) possesses acceptable
1235 convergence properties together with the full Newton-Raphson method. Therefore, we
1236 suggest that the anisotropic term can be either dropped out or symmetrized in a standard way
1237 (see Eq. B6).

1238

1239 For completeness, we derive the anisotropic stiffness terms in this Appendix. Recalling the
1240 expression for plastic scaling ratio (Eq. 43) and invoking the chain rule once again, we
1241 perform the following expansion:

$$1242 \quad \frac{\partial \alpha_{PL}}{\partial \epsilon_{mn}} = 2G_{CR} \left(\frac{\partial \alpha_{PL}}{\partial \tau_{\min}^{tr}} \frac{\partial \tau_{\min}^{tr}}{\partial \tau_{mn}^{tr}} + \frac{\partial \alpha_{PL}}{\partial \tau_{\max}^{tr}} \frac{\partial \tau_{\max}^{tr}}{\partial \tau_{mn}^{tr}} \right). \quad (B15)$$

1243 Entire derivatives in the above equation are readily available. For convenience, we denote:

$$1244 \quad D = \frac{(p \sin \varphi + c \cos \varphi)}{\left[(1 + \sin \varphi) \tau_{\max}^{tr} - (1 - \sin \varphi) \tau_{\min}^{tr} \right]^2}, \quad (\text{B16})$$

1245 then derivatives of plastic scaling ratio can be expressed as:

$$1246 \quad \alpha_{\min} = \frac{\partial \alpha_{PL}}{\partial \tau_{\min}^{tr}} = -2(\sin \varphi - 1)D, \quad (\text{B17a})$$

$$1247 \quad \alpha_{\max} = \frac{\partial \alpha_{PL}}{\partial \tau_{\max}^{tr}} = -2(\sin \varphi + 1)D. \quad (\text{B17b})$$

1248 The orthogonality of principle directions yields the following expressions for derivatives of
1249 principal stresses:

$$1250 \quad \frac{\partial \tau_{\min}^{tr}}{\partial \tau_{mn}^{tr}} = m_{mn}^{(\min)}, \quad \frac{\partial \tau_{\max}^{tr}}{\partial \tau_{mn}^{tr}} = m_{mn}^{(\max)}, \quad (\text{B18})$$

1251 where $m_{mn}^{(\min)}$ and $m_{mn}^{(\max)}$ are the spectral direction tensors (see Appendix A) associated with
1252 minimum and maximum principal stresses, respectively. Combining the above results and
1253 applying deviatoric projection (see Eq. B14), we obtain the final expression for non-
1254 dimensional tensor:

$$1255 \quad \hat{g}_{kl} = 2G_{CR} \left(\alpha_{\min} m_{kl}^{(\min)} + \alpha_{\max} m_{kl}^{(\max)} \right) - \frac{2}{3} G_{CR} \left(\alpha_{\min} + \alpha_{\max} \right) \delta_{kl}. \quad (\text{B19})$$

1256 Here, we have used the fact that trace of the spectral direction tensor is equal to unity.

1257

1258 Linearization of the Winkler boundary condition yields:

$$1259 \quad \frac{\partial}{\partial \Delta u_{kJ}} \left(N_I \bar{\sigma}_{ij} n_j \right) = N_I \frac{\partial \bar{\sigma}_{ij}}{\partial \Delta u_{kJ}} n_j. \quad (\text{B20})$$

1260 Here we have assumed that the outward normal vector does not vary with displacements.

1261 Rewriting the expression for boundary stress tensor (Eq. 22) in incremental form we have:

$$1262 \quad \bar{\sigma}_{ij} = {}^n \bar{\sigma}_{ij} - \rho_{ext} g \Delta u_z \delta_{ij}, \quad (\text{B21})$$

1263 where Δu_z is the vertical component of the displacement increment vector (downward

1264 positive). We can write, upon linearization of (B21), the following:

$$1265 \quad \frac{\partial \bar{\sigma}_{ij}}{\partial \Delta u_{kJ}} = -\rho_{ext} g N_J \hat{z}_k \delta_{ij}. \quad (\text{B22})$$

1266 We note that \hat{z}_k denote components of unit vector of vertical axis. Substituting (B22) into

1267 (B20) and reducing to symmetrical form, we can build the following approximation for the

1268 Winkler stiffness term:

$$1269 \quad \frac{\partial}{\partial \Delta u_{kj}} (N_I \bar{\sigma}_{ij} n_j) = N_I (-\rho_{ext} g n_z \hat{z}_i \hat{z}_k) N_J. \quad (B23)$$

1270 Here, n_z denotes vertical component of outward normal vector.

1271

1272 Finally, the approximate mechanical tangent matrix takes the following form:

$$1273 \quad K_{ijkl} = \int_{\Omega} \frac{\partial N_I}{\partial x_j} C_{ijkl} \frac{\partial N_J}{\partial x_l} d\Omega + \int_{\Gamma} N_I (\rho_{ext} g n_z \hat{z}_i \hat{z}_k) N_J d\Gamma. \quad (B24)$$

1274

1275

1276 Appendix C. Cantilever beam

1277

1278 We consider a cantilever beam of length l with rectangular cross-section of width b and
 1279 thickness h . Normal stress related to bending in the cross-section is generally computed using
 1280 the following formula (see e.g. Blake, 1985):

$$1281 \quad \sigma_b = \frac{M}{I} z. \quad (C1)$$

1282 Here, M is the bending moment, z is the distance between the point where we want to
 1283 calculate stress and center of the cross-section, and I is the second moment of inertia of the
 1284 cross-section, which for rectangular shapes is given by:

$$1285 \quad I = \frac{1}{12} b h^3. \quad (C2)$$

1286 Maximum compressive and tensile stresses occur on the top and bottom faces of beam where
 1287 $z = \pm \frac{1}{2} h$. In the benchmark problem, however, we calculate analytical stresses in slightly
 1288 different locations. The points offset from the top and bottom face by half the size of the
 1289 element to match location of integration points in the numerical scheme. Thus, we
 1290 use $z = \pm \frac{1}{2} (h - \Delta h)$, where Δh is the element size along the height of beam.

1291

1292 In a cantilever beam loaded by gravity, the maximum bending moment on the fixed end is
 1293 computed as follows:

$$1294 \quad M = \frac{q l^2}{2}, \quad (C3)$$

1295 where q is the weight of unit length of the beam, which is given by:

$$1296 \quad q = b h \Delta \rho g, \quad (C4)$$

1297 where $\Delta \rho$ is the differential density of the beam relative to surrounding material. Combining
 1298 equations (C1) – (C4) we obtain an analytical estimation for the bending stress:

$$1299 \quad \sigma_b = \pm \frac{3\Delta\rho gl^2 (h - \Delta h)}{h^2}. \quad (C5)$$

1300

1301 Maximum vertical deflection of the free end of the beam is given by (see e.g. Blake, 1985):

$$1302 \quad w = \frac{ql^4}{8EI}, \quad (C6)$$

1303 where E is Young's modulus of the material. In the case that elastic properties are specified
 1304 in terms of bulk and shear moduli, the Poisson's ratio and Young's modulus can be expressed,
 1305 respectively, as follows:

$$1306 \quad \nu = \frac{3K - 2G}{6K + 2G}, \quad E = 2G(1 + \nu). \quad (C7)$$

1307 In the benchmark problem, we use typical values for mafic crust: $K = 630$ kbar ,1308 $G = 400$ kbar , which roughly gives us the value of Young's modulus $E = 990$ kbar .

1309 Substituting equations (C2) and (C4) into (C6), we obtain analytical estimation for maximum

1310 vertical deflection of the beam:

$$1311 \quad w = \frac{3\Delta\rho gl^4}{2Eh^2}. \quad (C8)$$

1312

1313

1314 **Appendix D. Rigid cylinder**

1315

1316 We consider penetration of a rigid cylinder with radius r into a viscous fluid with viscosity1317 η_f . An analytical solution for the infinite fluid reservoir does not exist, which is clearly stated

1318 in the well-known Stokes paradox (see e.g. Slezkin, 1955). If we assume, however, that the

1319 reservoir has a finite characteristic dimension, the analytical solution can be obtained.

1320 Resistance force per unit length can be estimated using the following expression (Slezkin,

1321 1955):

$$1322 \quad R = \frac{4\pi}{\ln k - \frac{k^2 - 1}{k^2 + 1}} \eta_f v_c, \quad k = \frac{b}{r}, \quad (D1)$$

1323 where v_c is the velocity of the cylinder, and b is the distance between cylinder and rigid wall

1324 of the reservoir. We note that a sticking boundary condition must be enforced on the reservoir

1325 walls.

1326

1327 The steady-state velocity of the cylinder can be estimated by equating resistance force with
 1328 negative buoyancy force, which is given by:

$$1329 \quad F = \pi r^2 \Delta \rho g, \quad (D2)$$

1330 where $\Delta \rho$ is the differential density of cylinder relative to surrounding fluid. The resulting
 1331 expression reads:

$$1332 \quad v_c = \frac{1}{4} \left(\ln k - \frac{k^2 - 1}{k^2 + 1} \right) \frac{r^2 \Delta \rho g}{\eta_f}. \quad (D3)$$

1333

1334

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1608

1609

1610 **Figure captions**

1611

1612 **Figure 1.** Logarithm of effective viscosity for dry olivine calculated using parameters from
1613 Kameyama et al. (1999). Temperature-stress domains in which each particular creep
1614 mechanism produces the largest strain rate are labeled. Black solid lines separate the domains.
1615 Viscosity is truncated to a reasonable range $10^{18} - 10^{27}$ [Pa · s].

1616

1617 **Figure 2.** Schematic representation of the yield surface in principal stress space. In the tensile
1618 domain we use Tresca criterion.

1619

1620 **Figure 3.** Finite element discretization of volume (hexahedrons) and surface (quadrilaterals).
1621 Shown are the local coordinate systems and local numbers of the nodes.

1622

1623 **Figure 4.** Grid distortion for a single time step of the Rayleigh-Taylor instability problem
1624 with large (3 orders of magnitude) abrupt viscosity variation. (a) total density formulation, (b)
1625 differential density formulation. Gradient curve indicates boundary between the layers.

1626 Shown are the viscosity and density of upper and lower layers.

1627

1628 **Figure 5.** Contour plots of accumulated plastic strain for the shear band initiation problem
1629 (see section 4); (a) calculated by Newton-Raphson solver in 20 increments, (b) calculated by
1630 Picard solver in 360 increments, (c) calculated by Picard solver in 120 increments. The same
1631 amount of extension is applied in all models. In case (a) the shear band has minimum possible
1632 width constrained by the cell size. In cases (b) and (c) the shear band is significantly wider
1633 and more diffuse slowly decreasing its width with increasing number of time increments.

1634

1635 **Figure 6.** Initial (dashed line) and deformed (solid line) shape of the elastic plate. Vertical
1636 displacements are exaggerated by a factor of 10. Contour plot shows distribution of normal
1637 bending stress. Shown are calculated values for maximum bending stress and vertical
1638 displacement.

1639

1640 **Figure 7.** Relative error in approximation of stresses (rectangles) and vertical displacements
1641 (circles) versus number of elements across the elastic plate thickness.

1642

1643 **Figure 8.** Isolines of vertical velocity for the cylinder problem. (a) after 0.25 Myr, (b) after
1644 2.25 Myr. Black circles indicate positions of the particles which represent the cylinder.

1645

1646 **Figure 9.** Contour plot of effective logarithmic strain rate for the shear band initiation
1647 problem. (a) 0.05 km of extension, (b) 0.08 km of extension, (c) 0.15 km of compression (d)
1648 0.2 km of compression. Dashed lines indicate expected orientation of shear bands according
1649 to Arthur et al. (1977) (expression shown in the figure). Also shown are the values of Arthur
1650 angles between the shear bands and horizontal axis.

1651

1652 **Figure 10.** Setup (a) and results (b – d) of the triaxial compression numerical test with
1653 uniform random seeding (see text). Cases b – d correspond to 2.5%, 4.2%, 5.8% of axial
1654 deformation, respectively. Magnitude of accumulated plastic strain is shown in colors.

1655

1656 **Figure 11.** Setup (a) and results (b – d) of the triaxial compression numerical test with oblique
1657 gash seeding (see text). Cases b – d differ by amount of specimen rotation around vertical axis
1658 (0° , 90° and 180° , respectively) and all correspond to 2.7% of axial deformation. White lines
1659 on the top face show coordinate axis for rotational reference. Magnitude of accumulated
1660 plastic strain is shown in colors.

1661

1662 **Figure 12.** Lithospheric transpressional deformation problem. (a) setup and discretization, (b)
1663 initial thermal model, (c) distribution of effective strain rate on the surface and in the cross-
1664 section after 3Myr , (d) material phases (colors) and vertical displacements (white lines), (e)
1665 distribution of effective logarithmic viscosity (f) effective stress.

1666

1667 **Figure 13.** Evolution of the Lagrangian mesh embedded in the material for the lithospheric
1668 transpressional deformation problem. View from the top of the model. Thick dashed lines
1669 show the lateral boundaries of calculation domain depicted in Fig. 12. Note that material
1670 fluxes on the boundaries of the domain are nonzero.

Table 1
Nomenclature

Notation	Meaning	Dimension
x_i	Cartesian coordinates	m
t	Time	s
v_i	Velocity vector	$\text{m} \cdot \text{s}^{-1}$
T	Temperature	K
g	Gravitational acceleration	$\text{m} \cdot \text{s}^{-2}$
ρ	Density	$\text{kg} \cdot \text{m}^{-3}$
K	Bulk modulus	Pa
G	Shear modulus	Pa
B_L	Diffusion creep constant	$\text{Pa}^{-1} \cdot \text{s}^{-1}$
H_L	Diffusion creep activation enthalpy	$\text{J} \cdot \text{mol}^{-1}$
B_N	Dislocation creep constant	$\text{Pa}^{-n} \cdot \text{s}^{-1}$
H_N	Dislocation creep activation enthalpy	$\text{J} \cdot \text{mol}^{-1}$
n	Dislocation creep exponent	–
B_P	Peierls creep constant	s^{-1}
H_N	Peierls creep activation enthalpy	$\text{J} \cdot \text{mol}^{-1}$
τ_P	Peierls stress	Pa
φ	Friction angle	(°)
c	Cohesion	Pa
α	Thermal expansivity	K^{-1}
C_p	Specific heat	$\text{J} \cdot \text{kg}^{-1} \cdot \text{K}^{-1}$
λ	Thermal conductivity	$\text{W} \cdot \text{m}^{-1} \cdot \text{K}^{-1}$
A	Radiogenic heat production	$\text{W} \cdot \text{kg}^{-1}$
q_i	Heat flux vector	$\text{W} \cdot \text{m}^{-2}$
U	Internal energy	J
p	Pressure	Pa
θ	Volumetric strain	–
τ_{ij}	Deviatoric stress tensor	Pa
$\dot{\epsilon}_{ij}$	Deviatoric strain rate tensor	s^{-1}
η_{eff}	Effective viscosity	$\text{Pa} \cdot \text{s}$

Table 2
Overall computational flowchart

Newton-Raphson solution

1. Update stresses & heat fluxes in the elements
2. Assemble & solve linear systems

$$\delta \mathbf{u} = -\mathbf{K}_{(k)}^{-1} \mathbf{f}_{(k)}, \quad \delta \mathbf{T} = -\mathbf{E}_{(k)}^{-1} \mathbf{w}_{(k)}$$

3. Update displacement and temperature

$$\Delta \mathbf{u}_{(k+1)} = \Delta \mathbf{u}_{(k)} + \delta \mathbf{u}, \quad \mathbf{T}_{(k+1)} = \mathbf{T}_{(k)} + \delta \mathbf{T}$$

4. Update coordinates

$$\mathbf{x}_{(k+1)} = \mathbf{x}_{(k)} + \delta \mathbf{u}$$

5. Check convergence

$$\|\delta \mathbf{u}\| < \varepsilon \|\Delta \mathbf{u}_{(k+1)}\|, \quad \|\delta \mathbf{T}\| < \varepsilon \|\mathbf{T}_{(k+1)}\|$$

6. go to step (1) if necessary

Regridding and Remapping

1. Save history increment to markers
2. Advect markers by mesh
3. Adapt mesh to fit the free surface
4. Apply erosion boundary condition
5. Map markers onto adapted mesh
6. Count number of markers per cell
7. Insert/delete markers where necessary
8. Interpolate properties and history to mesh

Table 3

Stress update in integration point

1. Compute displacement gradient

$$h_{ij} = \Delta u_{il} {}^{n+1/2} b_{jl}$$

2. Evaluate strain increments

$$\varepsilon_{ij} = \frac{1}{2}(h_{ij} + h_{ji}) - \frac{1}{3} h_{kk} \delta_{ij}, \quad \theta = h_{ii}$$

3. Update pressure

$$p = {}^n p - K\theta + K\alpha(T - {}^n T)$$

4. Find trial deviatoric stress by FZERO

$$\tau_{ij}^{tr} = 2\eta_{CR} \dot{\varepsilon}_{ij} + \alpha_{CR} R_{ik} {}^n \tau_{kl} R_{jl}$$

5. Decompose trial deviatoric stress

$$\tau_{ij}^{tr} = \sum_{A=1}^3 \tau_A^{tr} m_{ij}^{(A)}$$

6. Check trial yield surface

$$F^{tr} < 0 \text{ exit}$$

7. Compute plastic scaling ratio,

$$\alpha_{PL} = \frac{2(p \sin \varphi + c \cos \varphi)}{(1 + \sin \varphi) \tau_{\max}^{tr} - (1 - \sin \varphi) \tau_{\min}^{tr}}$$

8. Update stress

$$\tau_{ij} = \alpha_{PL} \tau_{ij}^{tr}$$

9. Update plastic strain

$$\kappa = {}^n \kappa + \frac{\Delta t}{2\eta_{CR}} (1 - \alpha_{PL}) \tau_{II}^{tr}$$

Table 4
Plate benchmark

Variable	Meaning	Value	Dimension
$\Delta\rho$	Differential density	100	$\text{kg} \cdot \text{m}^{-3}$
E	Young's modulus	990	kbar
$l \times h$	Size of the domain	10×1	km
$n_l \times n_h$	Discretization	35×7	elements
w_{theor}	Theoretical deflection	0.151	km
w_{calc}	Calculated deflection	0.155	km
err_w	Rel. deflection error	2.6	%
σ_{theor}	Theoretical stress	257.2	MPa
σ_{calc}	Calculated stress	259.2	MPa
err_σ	Rel. stress error	0.8	%
σ_{cycl}	Cyclic stress residual	0.1	%

Table 5
Cylinder benchmark

Variable	Meaning	Value	Dimension
$\Delta\rho$	Differential density	100	$\text{kg} \cdot \text{m}^{-3}$
r	Radius of cylinder	25	km
$l \times h$	Size of the domain	400×700	km
$n_l \times n_h$	Discretization	200×350	elements
η_f	Viscosity of fluid	10^{20}	$\text{Pa} \cdot \text{s}$
η_c	Viscosity of cylinder	10^{24}	$\text{Pa} \cdot \text{s}$
v_{theor}	Theoretical velocity	8.47	$\text{cm} \cdot \text{yr}^{-1}$
v_{calc}	Calculated velocity	8.61	$\text{cm} \cdot \text{yr}^{-1}$

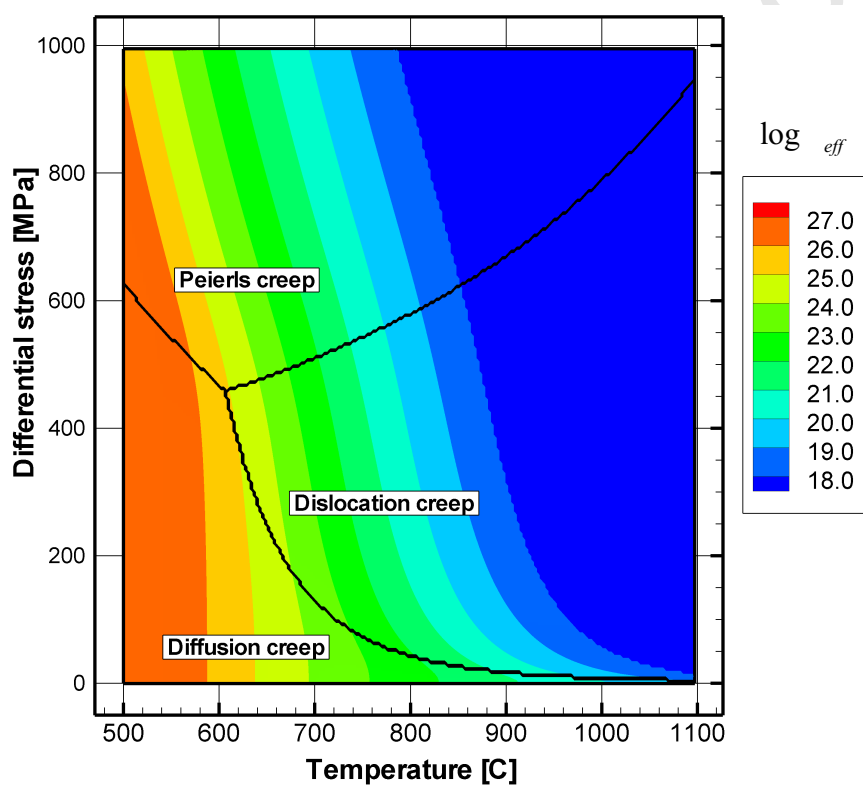
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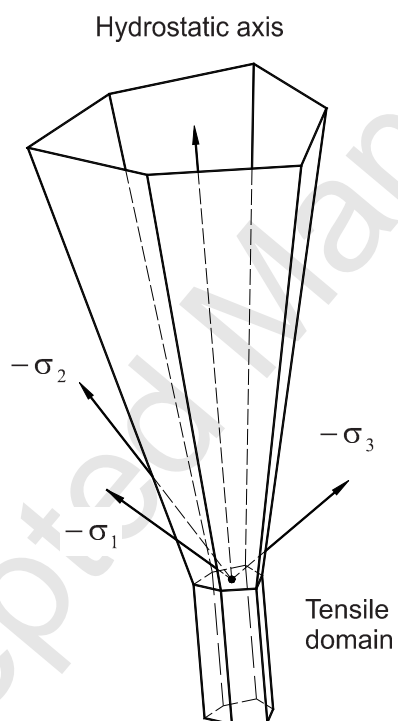
Table 6
Shear bands benchmark

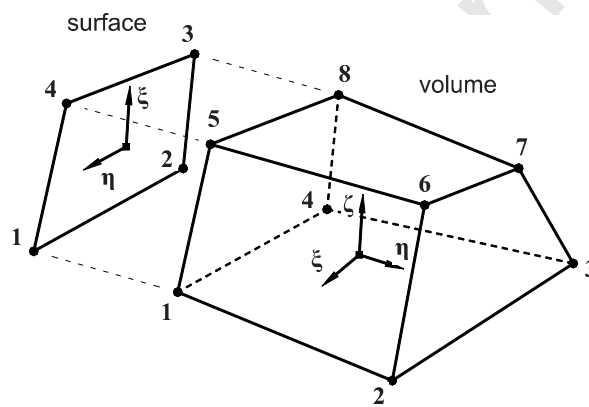
Variable	Meaning	Value	Dimension
ρ	Density	3000	$\text{kg} \cdot \text{m}^{-3}$
K	Bulk modulus	630	kbar
G	Shear modulus	400	kbar
φ	Friction angle	30	($^{\circ}$)
ψ	Dilatation angle	0	($^{\circ}$)
c	Cohesion	20	MPa
$l \times h$	Size of the domain	40×7	km
$n_l \times n_h$	Discretization	200×35	elements
α_{ext}	Inclination angle (extension)	53	($^{\circ}$)
α_{comp}	Inclination angle (compression)	38	($^{\circ}$)

Table 7
Rheological and thermal parameters used in transpression problem

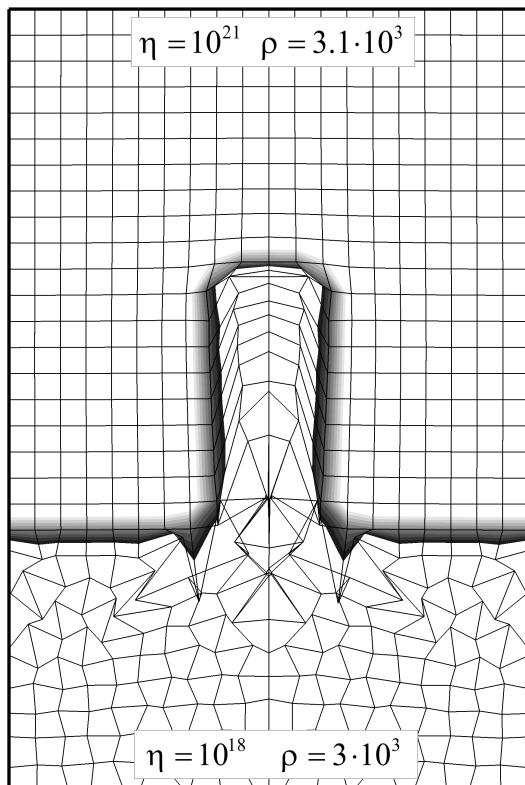
Parameter	Dimension	Felsic Upper Crust	Mafic Lower Crust	Mantle peridotite
ρ	$\text{g} \cdot \text{cm}^{-3}$	2.7	3.0	3.3
K	kbar	550	630	1220
G	kbar	360	400	740
$\log B_N$	$\text{Pa}^{-n} \cdot \text{s}^{-1}$	-28.0	-21.05	-16.3
H_N	$\text{kJ} \cdot \text{mol}^{-1}$	223	445	535
n	–	4.0	4.2	3.5
φ	(°)	30	30	30
c	MPa	20	20	20
α	K^{-1}	$2.7 \cdot 10^{-5}$	$2.7 \cdot 10^{-5}$	$3.0 \cdot 10^{-5}$
C_p	$\text{kJ} \cdot \text{kg}^{-1} \cdot \text{K}^{-1}$	1.2	1.2	1.2
λ	$\text{W} \cdot \text{m}^{-1} \cdot \text{K}^{-1}$	2.5	2.5	3.3
A	$\text{nW} \cdot \text{kg}^{-1}$	1.0	0.1	0.0



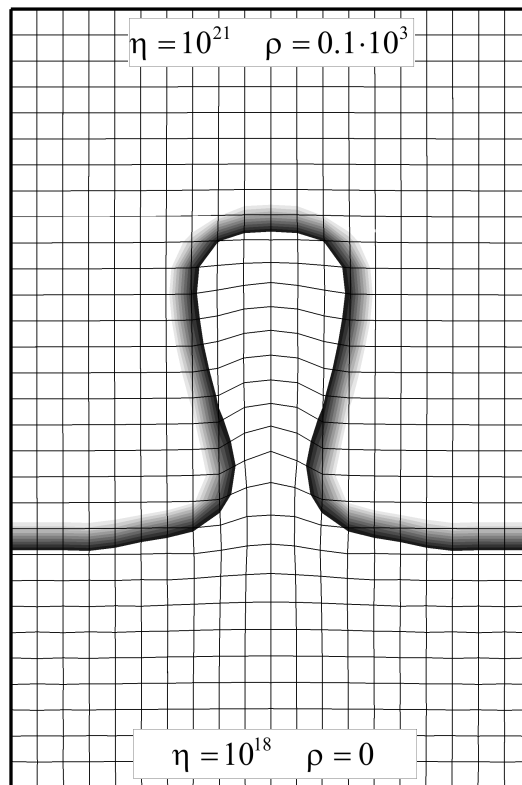


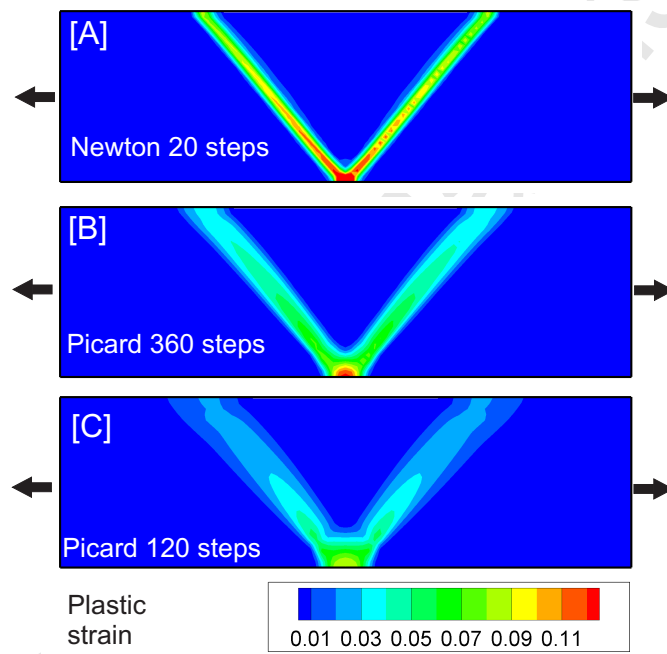


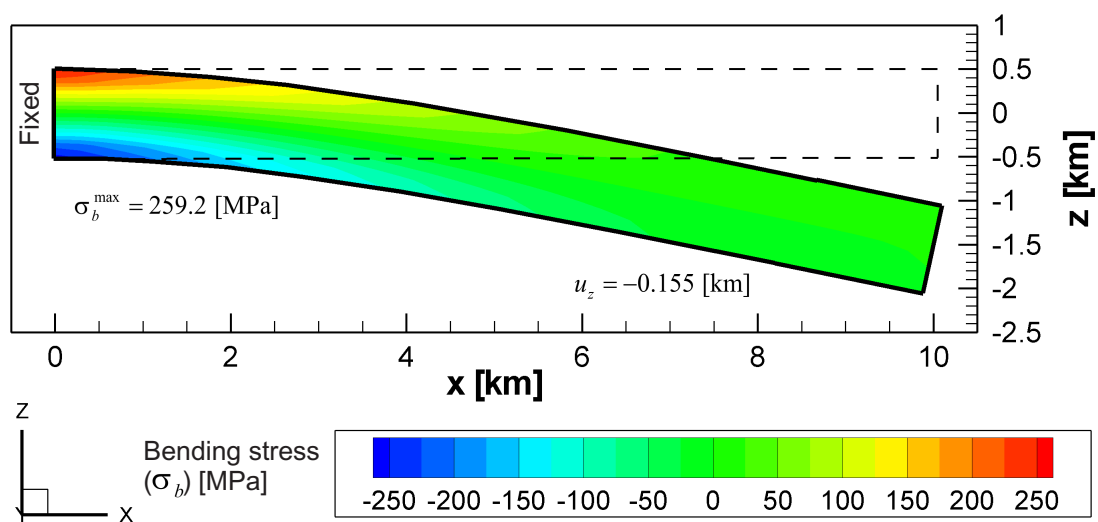
[A] total density formulation

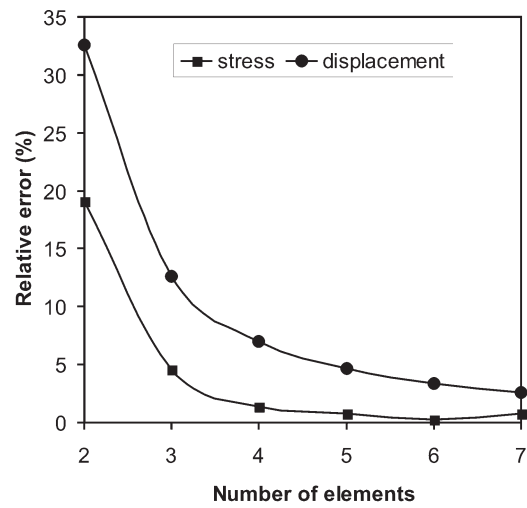


[B] differential density formulation

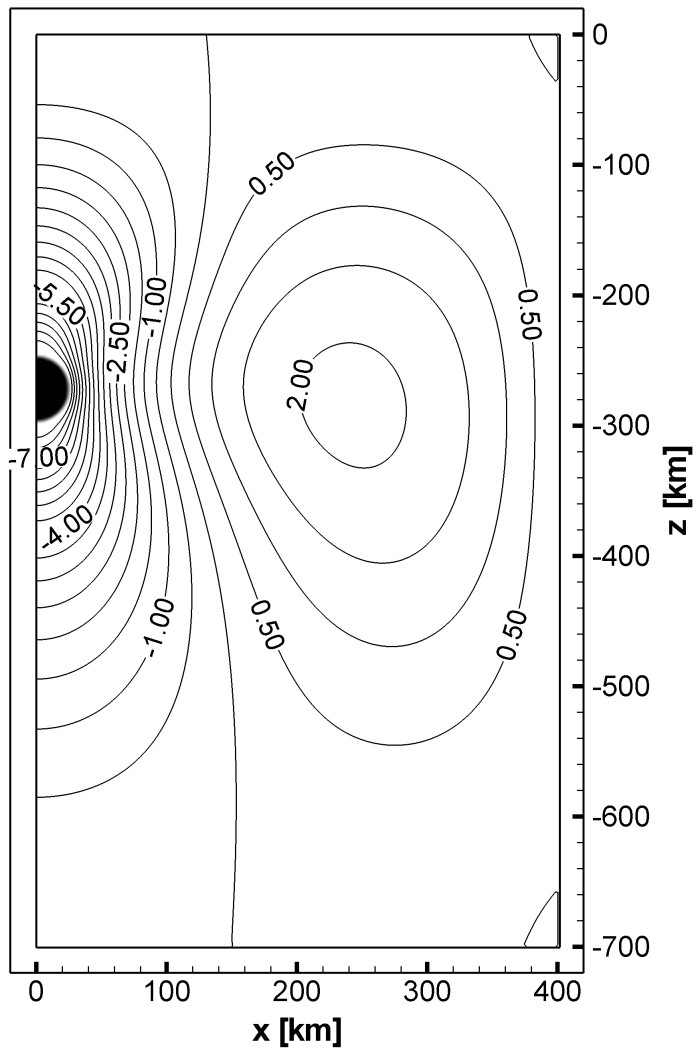








[A] 0.25 Myr



[B] 2.25 Myr

