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Measurement of the dynamic fracture toughness with notched PMMA specimen under impact loading.

S. Sahraoui(*), A. El Mahi and B. Castagnède

*Laboratoire d'Acoustique de l'Université du Maine, UMR CNRS 6613
Avenue O. Messiaen, 72085 Le Mans Cedex 9 (France)*

Abstract

In the present study three-point-bend impact experiments were conducted using an instrumented Charpy pendulum with a laser displacement measurement to better understand the correlation between impact velocity and the dynamic effects observed on the load-time curves. The experiments were performed at impact velocities ranging from 1 to 4 m/s.

The aim of this work is to measure the dynamic fracture toughness at high impact velocities where the classical method is limited by the inertial effects. The direct measurements of the specimen deflection are successfully used for the toughness evaluation. The results obtained with this method, which are compared to other studies, indicate that this approach seems promising for brittle materials such as PMMA.

keywords : toughness, impact, dynamic effects, brittle fracture, PMMA

1 - Introduction

The mechanical properties of most materials are sensitive to loading rate and the fracture toughness under dynamic loading has received considerable attention [1].

The instrumented impact loading test is rapidly being accepted as a useful tool for evaluating the dynamic response of materials. The modern instrumentation has overcome many of the practical limitations and some studies were focused on electronic devices recommendations [2].

The effects of inertial loading in Charpy tests have been studied by many authors. The dynamic effects were analysed through various modelisations: spring system [3,4], Euler-Bernoulli equation [5], finite element method [6],... The aim of these investigations is to give a simple interpretation of the load-time traces observed during the instrumented impact testing and then to determine the effects of inertial loading.

A procedure obtained from pre-established impact response curve and a measurement of the time-to-fracture were proposed by Kalthoff [7]. The measuring technique is fully dynamic and does not impose any restrictions on the test conditions. This is precisely the objective of the present paper. In fact, we use the classical instrumentation for measuring the impact load and we introduce a novel procedure for determining the dynamic toughness through the specimen deflection measurement by mean of a laser vibrometer. This method can be applied for brittle materials (PMMA in the present study) where the instable fracture can be easily observed on the load-time curve. The direct measurement of the specimen deflection may overcome the inertial effects which are not perturbing the displacement-time curve. The

(*) corresponding author:

email address : sohbi.sahraoui@univ-lemans.fr

determination of the deflection at the fracture initiation during the impact can be considered as good information in the Charpy test. At high velocity, the load-time curves are not exploitable due to the presence of inertial effects, but the calculations with the present method give similar results as Theocaris [8]. Concluding remarks and prospective discussion are given in the end of this paper.

2 - Experimental method

The Charpy pendulum (Zwick 7.5J) is instrumented by strain gages for measuring the load applied on the three point bending specimen of which the deflection is obtained through the laser vibrometer (Polytec OFV302) as indicated on Fig. 1. The load-time and the deflection-time curves are recorded on the numerical oscilloscope and then analyzed on a personal computer. In dynamic loading, the direct measurement of the specimen deflection by this optical equipment (based on Doppler effect) is more adequate than the measurement of the hammer (or projectile) displacement used in the literature [9,10].

The PMMA Charpy specimens (Fig.1) have the dimensions (40x10x10 mm³) and 55 mm for the total length and different crack lengths a (varying between 2 and 5 mm). At each velocity, the tests are performed on ten specimens.

At low velocities (less than 1.5 m/s) the toughness is calculated through the measured fracture load. At higher impact velocities (more than 2 m/s), the experimental conditions are not valid for using the classical approach applied above; the influence of inertial effects are discussed in the next section. The new approach consists on the simultaneous measurements of the specimen deflection and the impact load during the fracture test. The maximum load F_{\max} corresponding to the fracture initiation is related to the fracture deflection u_f as indicated in Fig2. The time to fracture t_f is then obtained by means of the load time curve. The effective fracture load is calculated from the fracture deflection and the stiffness of each specimen. The stiffness K_o of the three point bending specimen is statically measured before each test and allows us to calculate the fracture load F_u from the deflection measurement u as:

$$F_u = K_o u \quad (1)$$

The static stiffness can be used for materials not sensitive to strain rate. Otherwise for the strain rate sensitive materials such as PMMA, we must use the Young modulus values at the strain rate loading. Then the dynamic stiffness $K(\dot{\varepsilon})$ which is proportional to Young modulus can be calculated as $K_o E(\dot{\varepsilon})/E_o$, where E_o and $E(\dot{\varepsilon})$ are the Young modulus in static and dynamic loading given in table from a previous paper [11]. The table 1 indicates also the correspondence between the impact velocity and the strain rate as related below (Eq.2).

For the three point bending configuration the strain rate $\dot{\varepsilon}$ is related to the impact velocity v by:

$$\dot{\varepsilon} = 6Wv / L^2 \quad (2)$$

which is derived from the well known relation [12] :

$$\varepsilon = \frac{6W}{L^2} u \quad (3)$$

where ε is the maximum strain at the mid span of the specimen.

3 - Dynamic effects

It is well known that the conventional impact testers such as Charpy pendulum generate oscillations which can disturb the measurement of the effective load at fracture initiation. The crack initiation point must be determined in order to calculate the dynamic toughness K_{Id} . Crack initiation generally occurs at the maximum load in the case of brittle material and K_{Id} can be calculated by applying linear elastic fracture mechanics; for Charpy specimen the dynamic toughness is given by [13]:

$$K_{Id} = \frac{3}{2} \frac{FL}{BW^2} (\pi a)^{1/2} [1.11 - 1.55(a/W) + 7.71(a/W)^2 - 13.53(a/W)^3 + 14.23(a/W)^4] \quad (4)$$

where F represents the fracture load and the other geometric parameters are defined in Fig. 1. Eq. 4 is valid in a quasistatic loading or if the fracture occurs after times sufficiently large such as a quasistatic loading condition has been reached in the specimen.

Fig. 3 and 4 give typical load-time curve obtained at two different velocities. At slow velocity (Fig.3), the oscillations of the load are relatively small and then the measured fracture load is not perturbed by the dynamics effects. We observe in Fig. 4 (high velocity) that the oscillations have high amplitude which gives an erroneous value of the fracture load. The most important observation is the quasistatic configuration of the specimen during the impact characterized by loading oscillations inherent to inertial effect.

4 - Results and discussion

Fig. 5 provides all the results obtained in this study where each experimental data corresponds to the average value of the calculation given for three specimens at various crack length.

The dynamic toughness K_{Id} is calculated through Eq. 4 by using the maximum load obtained in load-time records (see the typical curves in Fig.3 and Fig.4). These K_{Id} values are shown by full squares in Fig. 5. For high impact velocities (more than 2m/s), this procedure is not valid and we cannot use Eq. 4. The ASTM standard [2] assumes that the time to fracture of the specimen must be larger than three times the period of the oscillations of the impacted specimen. The values indicated inside the dashed line of Fig.5 are calculated with Eq. 4 but the experimental conditions do not respect the above rule. The high amplitude oscillations during the impact have overestimated the fracture load. The objective of this study is to use the measured displacement which is not perturbed by the oscillations during the impact and to determine the corresponding load F_u (Eq.1) which is the effective load responsible of the deflection and then of the fracture itself. Numerical simulations given in a previous paper [5] provide the basis of the physical interpretation of this approach. The K_{Id} values calculated with this method are showed by full circles in Fig.5. Other results taken in the literature are reported in this figure (empty square and circles). The observed gap between the results of Fig.5 can be explained by the experimental conditions (specimen geometry, Charpy and Hopkinson loading, initial cracking).

5 - Conclusion

The present study has pointed out the importance of the measurement of the specimen deflection for calculating the dynamic toughness. At high loading rates, the load-time curve is useful for the determination of the time to fracture and this information can be obtained by electric measurements directly taken at the crack tip of the specimen. The load measurements in this study have illustrated the limitation of the classical approach and give a new performed method for brittle materials. This approach is very simple to apply for non sensitive materials to strain rates. Otherwise we need to characterize the specimen at each impact velocity. Future investigations can study the limitation of this approach for high rates of loading.

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CAPTIONS FOR ILLUSTRATIONS

Fig. 1:

Three points bending Charpy specimen and schematic of the experimental arrangement.

Fig. 2 :

Schematisation of the deflection to fracture measurement.

Fig. 3 :

Typical load-time and displacement-time curves of brittle material ($v=1.2$ m/s, $a = 2$ mm)

Fig. 4 :

Typical load-time and displacement-time curves of brittle material ($v=2.3$ m/s, $a = 2.5$ mm)

Fig. 5 :

Dynamic toughness ($\text{MPa m}^{1/2}$) vs strain rate (s^{-1})

Table :

Young modulus versus strain rate and impact velocity [9]

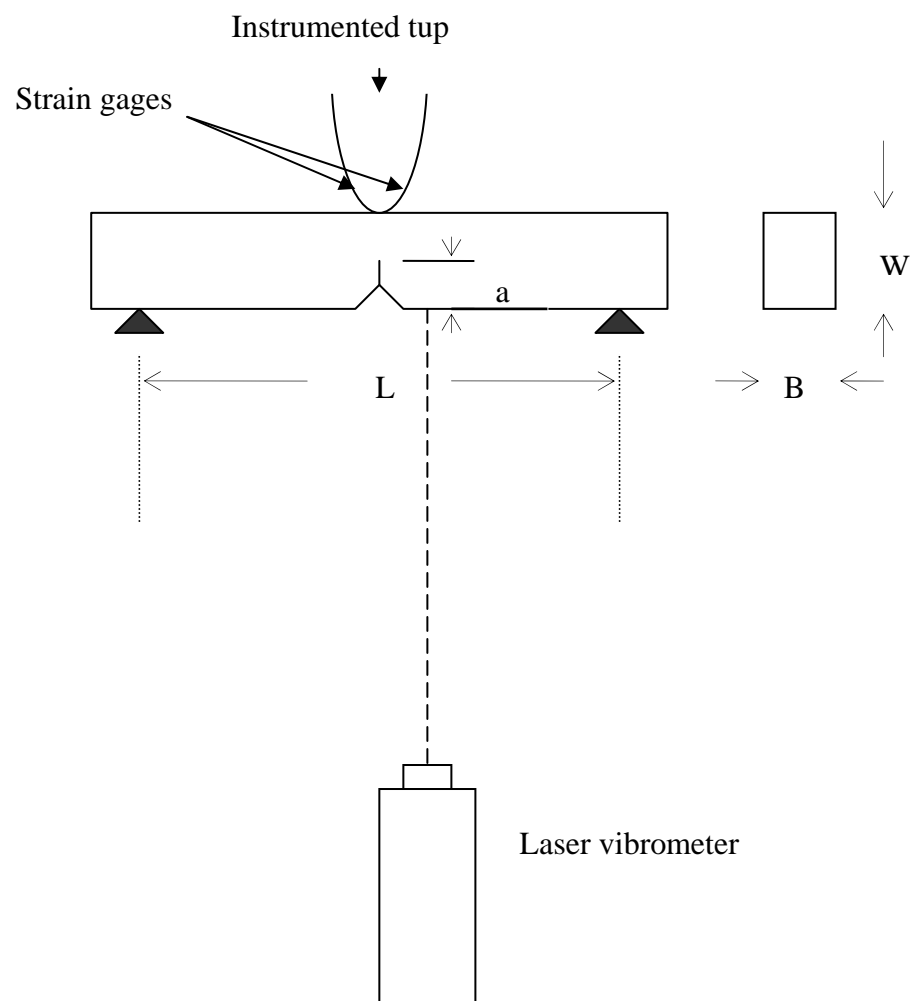


Figure 1

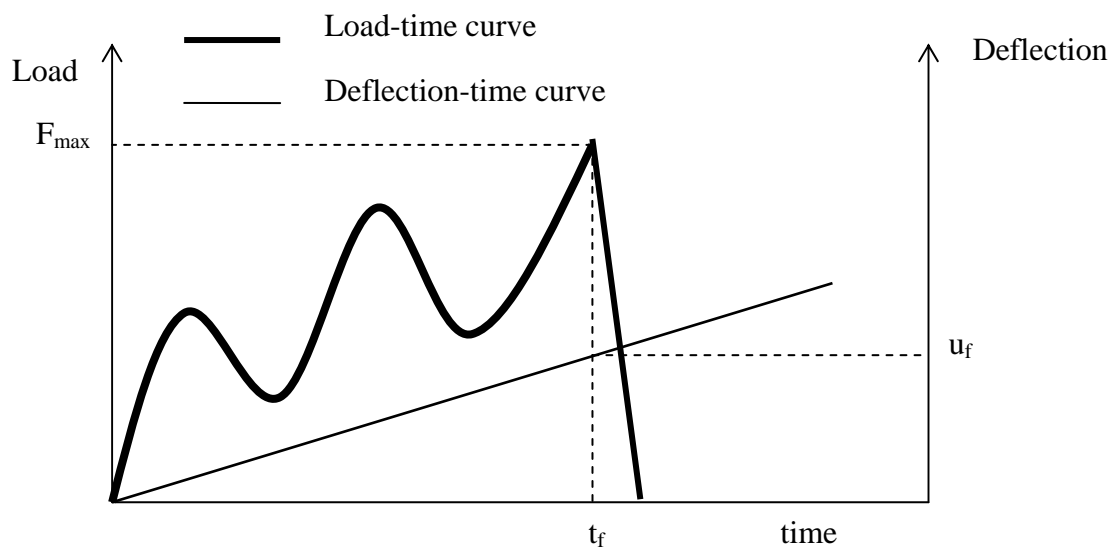


Figure 2

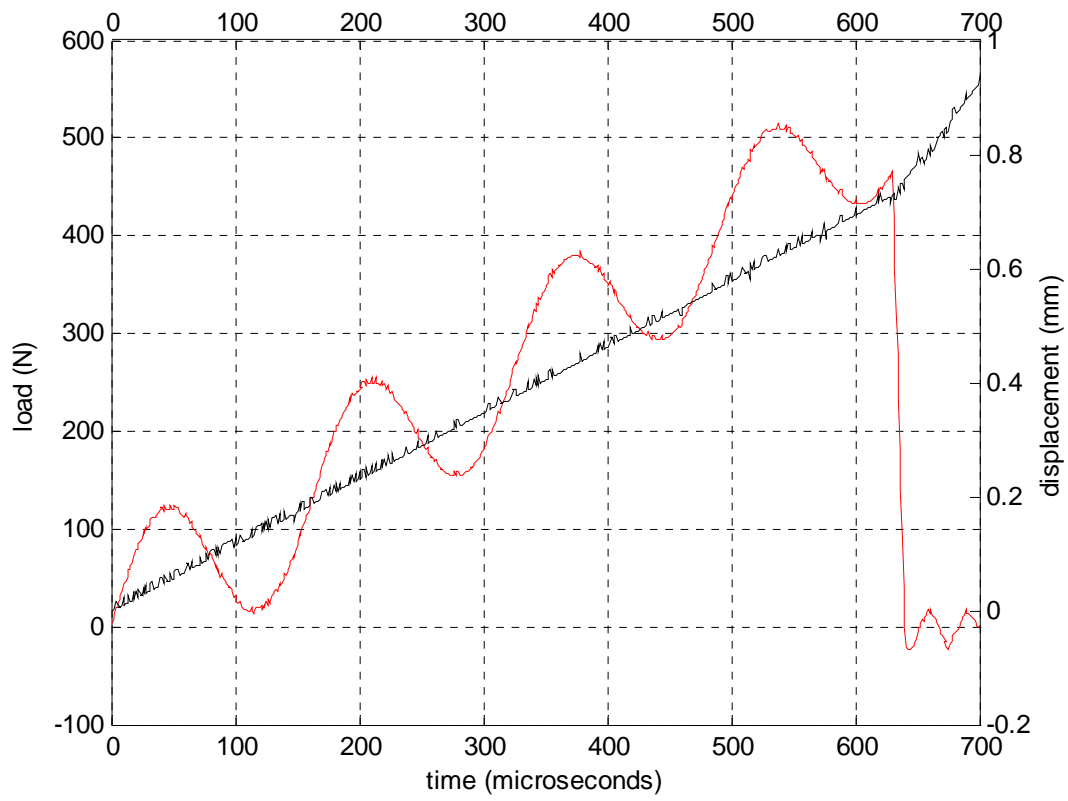


Figure 3

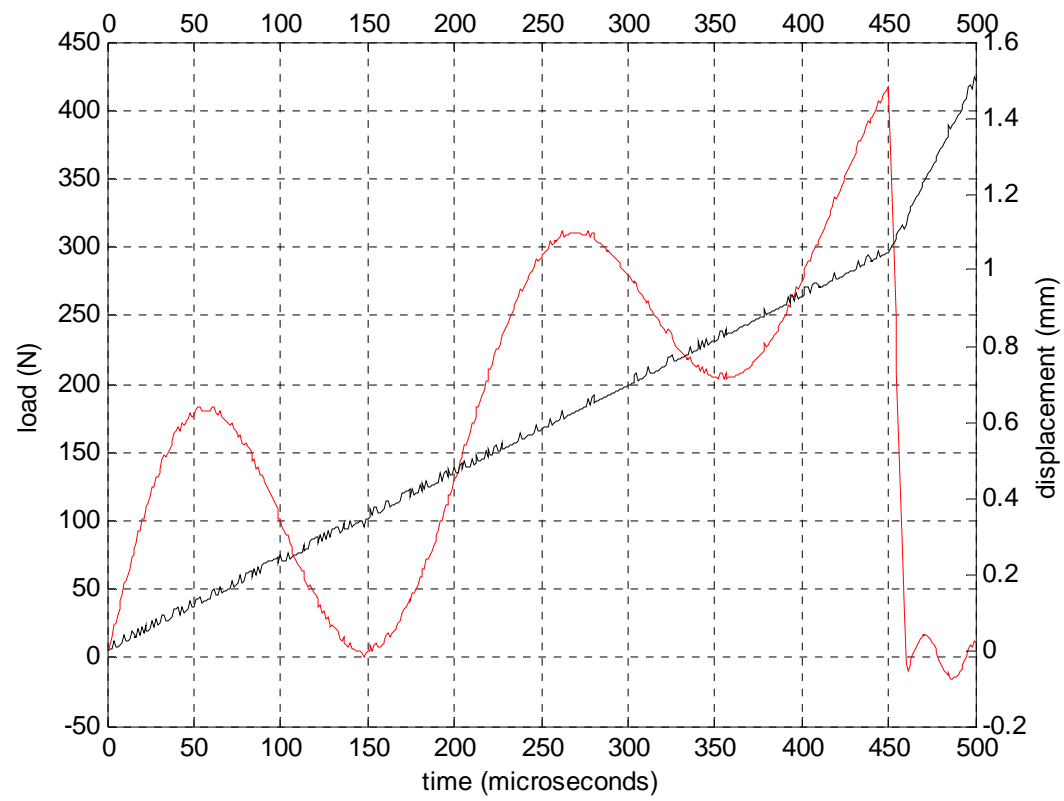


Figure 4

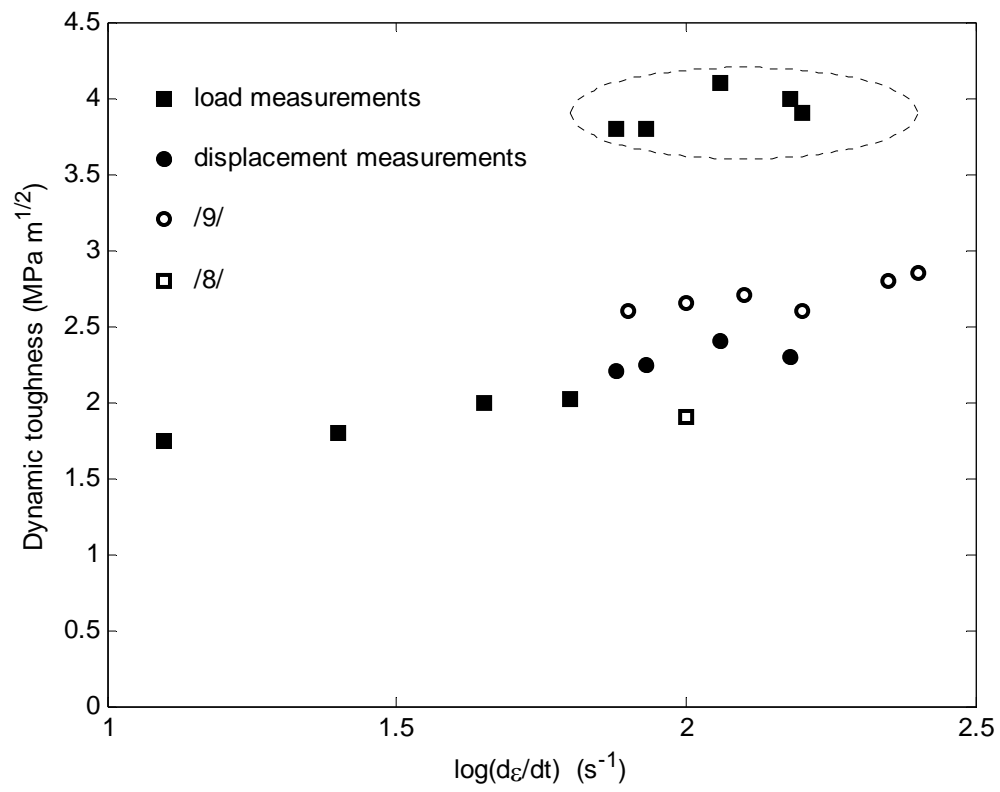


Figure 5

v (m/s)	static loading	2.3	3.1	4
$\log \dot{\epsilon}$ (s ⁻¹)	-2	1.93	2.06	2.18
E (GPa)	3.4	5.3	5.4	5.5

Table