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F. Weidner, H. Bösch, H. Bovensmann, J. P. Burrows, A. Butz, et al.. Balloon-borne limb profiling of UV/vis skylight radiances, O<sub>3</sub>, NO<sub>2</sub>, and BrO: technical set-up and validation of the method. *Atmospheric Chemistry and Physics*, 2005, 5 (5), pp.1422. hal-00328388

**HAL Id: hal-00328388**

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Submitted on 18 Jun 2008

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# Balloon-borne limb profiling of UV/vis skylight radiances, O<sub>3</sub>, NO<sub>2</sub>, and BrO: technical set-up and validation of the method

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Received: 9 August 2004 – Published in Atmos. Chem. Phys. Discuss.: 24 November 2004

Revised: 2 May 2005 – Accepted: 9 May 2005 – Published: 14 June 2005

**Abstract.** A novel light-weight, elevation scanning and absolutely calibrated UV/vis spectrometer and its application to balloon-borne limb radiance and trace gas profile measurements is described. Its performance and the novel method of balloon-borne UV/vis limb trace gas measurements has been tested against simultaneous observations of the same atmospheric parameters available from either (a) in-situ instrumentation (cf., by an electrochemical cell (ECC) ozone sonde also deployed aboard the gondola) or (b) trace gas profiles inferred from UV/vis/near IR solar occultation measurements performed on the same payload. The novel technique is also cross validated with radiative transfer modeling. Reasonable agreement is found (a) between measured and simulated limb radiances and (b) inferred limb O<sub>3</sub>, NO<sub>2</sub>, and BrO and correlative profile measurements when properly accounting for all relevant atmospheric parameters (temperature, pressure, aerosol extinction, and major absorbers).

## 1 Introduction

In the past two decades remote sensing of the atmosphere by optical methods has evolved into a powerful tool for meteorology, atmospheric photochemistry and climate studies. Most recently, space-borne UV/vis limb observations of the skylight have also become available, c.f. through the SME (Mount et al., 1983), SOLSE/LORE (McPeters et al., 2000), Odin/OSIRIS (Von Savigny et al., 2003; Sioris et al., 2003), Envisat/SCIAMACHY (Burrows et al., 1995) instruments.

The SCIAMACHY (SCanning Imaging Absorption spectroMeter for Atmospheric CHartography) on the ESA-Envisat satellite offers unprecedented possibilities for at-

mospheric remote sensing by monitoring a larger number of atmospheric trace constituents by spectrally resolved UV/vis/near IR limb scattering observations. SCIAMACHY is a national contribution by Germany, the Netherlands and Belgium to the ESA-Envisat satellite, which was launched into a sun synchronous low orbit on 28 February 2002 (Bovensmann et al., 1999). SCIAMACHY simultaneously measures the atmospheric skylight in a variety of viewing directions and the extraterrestrial irradiance in the wavelength range from 220 nm to 2380 nm, at moderate spectral resolution (e.g., Eichmann et al., 2003; Von Savigny et al., 2004 and 2005).

The UV/vis limb measurements of SCIAMACHY, however, involve a number of methods which require careful validation and verification through collocated ground-based, aircraft, and balloon-borne measurements. Among them, the modeling of the atmospheric radiative transfer (RT) – including the investigation of its sensitivities to a larger number of atmospheric parameters (cf., the temperature, pressure, ozone and aerosol profile, ...) – is most challenging for the interpretation of UV/vis limb measurements.

Following the pioneering studies of McElroy (1988), we report here on one of the most stringent tests ever to validate the individual steps (spectral retrieval, RT modeling and profile inversion) – and thus of the whole UV/vis limb technique. For the present study we have chosen a twofold approach.

(1) The validation is performed by deploying a novel balloon-borne, and absolutely calibrated UV/vis scanning limb spectrometer (called mini-DOAS) on a couple of stratospheric balloon flights on the azimuth controlled LPMA/DOAS payload (Laboratoire de Physique Moléculaire et Applications and Differential Optical Absorption Spectroscopy). These measurements allow us (a) to absolutely monitor UV/vis skylight radiances as a function

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of sensor and tangent height, viewing geometry and solar zenith angle (SZA) and (b) by applying the Differential Optical Absorption Spectroscopy (DOAS, Platt, 1994; Platt and Stutz, 2006) technique to infer vertical profiles for a number of UV/vis absorbing atmospheric trace gases ( $\text{O}_3$ ,  $\text{NO}_2$ ,  $\text{O}_4$ ,  $\text{BrO}$ ,  $\text{H}_2\text{O}$ , and possibly in future of  $\text{OCIO}$ ,  $\text{IO}$ ,  $\text{OIO}$ , ...).

(2) Further, the measured mini-DOAS limb radiances are inter-compared to simulations of a novel Monte-Carlo (MC) radiative transfer model, the latter to be used in future studies for forward modeling of the SCIAMACHY limb observations. Secondly, the inferred trace gas profiles from the mini-DOAS instruments are validated against simultaneous profile measurements of the same atmospheric constituents measured by well established in-situ (e.g., ozone by an electrochemical cell) or optical remote sensing techniques (cf., solar occultation) performed on the LPMA/DOAS payload. For details of the latter measurements see e.g., Camy-Peyret et al., 1993; Payan et al., 1998; Ferlemann et al., 1998, 2000; Harder et al., 1998, 2000; Bösch et al., 2003, and the accompanying paper of Gurlit et al., 2004).

The present study is organized as follows: In Sect. 2, the newly developed instrument is described and characterized, and experimental details about the absolute calibration are given. Furthermore, descriptions of the data analysis methods, i.e. DOAS evaluation, RT modeling, profile inversion, and error analysis, are provided. In the Sects. 3 and 4, selected atmospheric observations of the mini-DOAS instrument are described and discussed with respect to the measurements of the other instruments also deployed aboard the LPMA/DOAS payload. Section 5 concludes the study with the lessons learned for future investigations.

## 2 Methods

### 2.1 Technical set-up of the mini-DOAS instrument

The novel mini-DOAS spectrometer has been designed for low weight ( $<7$  kg) and low power consumption (7.5 W), with a particular emphasis being put on stable imaging and a reasonably large signal to noise ratio. While the small size and low weight offers the chance for versatile applications (c.f., a stand-alone operation for time resolved measurements of important stratospheric radicals, trace gas measurements and radiative transfer studies in the cloudy troposphere), a stable imaging is found to be necessary for the detection of  $\text{O}_3$ ,  $\text{NO}_2$  and in particular of the weakly absorbing gases ( $\text{BrO}$ ,  $\text{OCIO}$ ,  $\text{IO}$ ,  $\text{OIO}$  ...), based on the experience with our larger precursor balloon spectrometer.

The mini-DOAS instrument consists of 5 major parts: (a) 2 light intake telescopes for simultaneous nadir and scanning limb observations (the latter being mounted on an automated elevation scanner), (b) glass fibre bundles which conduct the skylight from the telescopes into the spectrometers, (c) two commercial Ocean Optics USB-2000 spectrometers

(d) which are mounted into an evacuated and temperature-stabilized housing, and finally (e) a single board computer for data handling and storage.

(a) The nadir and limb telescopes each consist of a spherical quartz lens (12.7 mm diameter, 30 mm focal length) which focuses the incoming scattered skylight onto the circular or the rectangular entrance of the glass fiber bundles. During the balloon flight, the nadir telescope is mounted at the bottom of the outer frame of the LPMA/DOAS payload structure, which provides an unobscured view into nadir direction. The limb telescope is mounted on an elevation angle scanner (built by Hofmann Meßtechnik, Rauenberg, Germany) which supports limb observations in a range of  $+10^\circ$  to  $-20^\circ$  elevation angle, with step sizes as small as  $0.04^\circ$ . During the balloon flight, the scanner is mounted on the right hand side (i.e., in a  $+90^\circ$  azimuth angle relative to the Sun's azimuth direction) of the azimuth controlled LPMA/DOAS gondola.

(b) Each glass fibre bundle consists of 10 individual quartz glass fibers each (diameter  $100\text{ }\mu\text{m}$ , length 2 m, numerical aperture=0.22). Glass fibre bundles are used, since they not only allow for a more flexible arrangement of the instrument, but are also known for largely reducing the polarization sensitivity of grating spectrometers (Stutz and Platt, 1996, 1997). In fact, laboratory measurements show that by using glass fibre bundles the polarization sensitivity of an Ocean Optics USB 2000 spectrometer is small ( $\leq 1\%$ ). For the nadir observations, the individual glass fibres are arranged in circular geometry at the light intake, a mounting which together with the telescope supports a round field of view (FOV) of  $0.6^\circ$ . For the limb observations the glass fibres are arranged in a "rectangular geometry" light intake set-up i.e., the individual glass fibre entrances are linearly aligned. This arrangement supports a FOV of  $0.19^\circ$  in the vertical and  $1.34^\circ$  in the horizontal direction. Likewise, the glass fibres are linearly aligned at both exits, and the outgoing light is skimmed by a  $50\text{ }\mu\text{m}$  wide and  $1000\text{ }\mu\text{m}$  high spectrometer entrance slit.

(c) The heart of the mini-DOAS balloon instrument consists of two commercial Ocean Optics USB 2000 spectrometers for simultaneous nadir and limb observations. The USB 2000 is a miniature grating spectrometer working in cross Czerny-Turner geometry. Its advantage is the small size ( $86\times 63\times 30\text{ mm}^3$ ), the low weight (270 g) and the high photon detection sensitivity owing to an integrated linear CCD array detector (Sony ILX511). The light enters the spectrometer through an entrance slit from which it is focused by a collimator mirror onto a holographic grating with 1800 grooves/mm. A second mirror focusses the light onto the linear CCD array with 2048 pixels (each pixel is  $14\text{ }\mu\text{m}$  wide and  $200\text{ }\mu\text{m}$  high). Attached onto the CCD array detector is a cylinder lense which focuses the  $1000\text{ }\mu\text{m}$  high entrance slit onto the  $200\text{ }\mu\text{m}$  high detector. Also attached to the CCD array detector is the preamplifier and a control logic unit which handles the pre-amplification of the signals, A/D

conversion to 12 bit data and communication.

The spectrometers cover a spectral range of 327–527 nm at a full width at half maximum resolution (FWHM) of 0.8–1 nm, or 8 to 10 detector pixel/FWHM depending on the wavelength. Based on previous experience, this wavelength coverage and resolution should allow for the detection of the atmospheric trace gases  $O_3$ ,  $NO_2$ ,  $O_4$ ,  $BrO$ ,  $H_2O$ , (and potentially  $OCIO$ ,  $IO$ ,  $OIO$ ,  $CH_2O$ ).

(d) Both spectrometers are kept in a sealed and evacuated container, which itself is immersed in a water-ice reservoir ( $\sim 2$  liters). This ensures a stable spectrometer and CCD array temperature of  $0^\circ C$  during an entire balloon flight.

(e) Data handling and storage is maintained by a single board PC (type National Geode 200 MHz) which is equipped with a flash memory device. The allocated data are transferred from the spectrometers to the PC via a USB data transfer connection. It supports a data transmission rate fast enough to record a single spectrum every 25 ms. Possible integration times per spectrum as provided by the manufacturer of the spectrometers are in the range of 3–65 535 ms. The PC can alternatively be operated under Windows or Linux with our lab-owned DOASIS<sup>1</sup> or XDOAS software packages, respectively. Both software tools support the automatic adjustment of the integration time, recording and storage of the measured spectra and the control of the limb scanning stepper motor.

The total size of the instrument is  $260 \times 260 \times 310 \text{ mm}^3$  (w/o fibers), its weight is  $\sim 4.8 \text{ kg}$  plus 2 kg of water and ice, and its power consumption is  $\sim 7.5 \text{ W}$ .

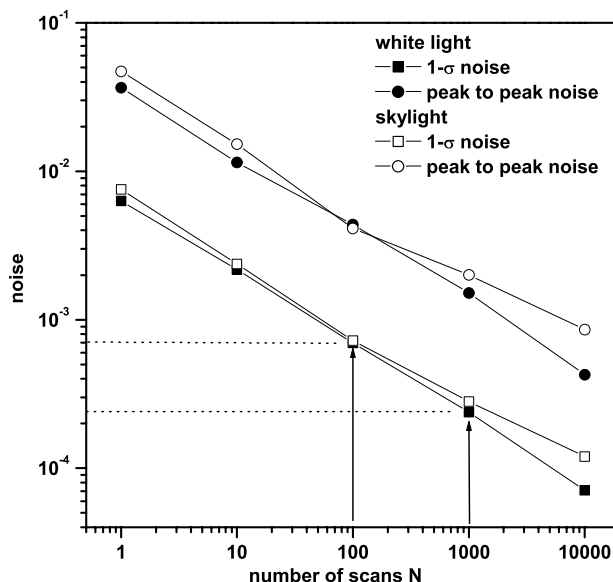
## 2.2 Instrument performance and calibration

**Instrument performance:** The optical performance of the mini-DOAS instrument is tested in a set of laboratory measurements using alternatively Penray Hg, Kr or HgCd emission lamps or white light sources (integrating sphere type BN-102-3 manufactured by Gigahertz). For the temperature stabilized ( $0^\circ C$ ) instrument, the following relevant noise contributions ( $1\text{-}\sigma$  rms) are found for single scans:

(a) an electron shot noise level of 15 binary units (BU) corresponding to a  $1\text{-}\sigma$  noise of 0.474% for a 80% saturation level of the CCD array with a well-depth of 62 500 electrons and an electron to binary unit conversion factor of 15 electrons/BU. As for a DOAS evaluation the ratio of the spectrum being analyzed and a solar reference spectrum is considered, the noise has to be multiplied by a factor of  $\sqrt{2}$ , yielding a photo-electron shot noise of 0.67%.

(b) a total electronic noise of 67.4 electrons or 4.4 BU causing a noise of 0.135% at  $1\text{-}\sigma$ .

(c) the dark current of the SONY ILX511 shows a large pixel to pixel variation e.g., next to pixels with very small dark current there are pixels for which the current



**Fig. 1.** Root mean square ( $1\text{-}\sigma$ ) and peak-to-peak residual structure ( $5\text{-}\sigma$ ) as a function of co-added spectra  $N$  at 80% illumination level of the CCD array detector for white light of an integrating sphere and skylight. The arrows indicate the ( $1\text{-}\sigma$ ) noise actually achieved during the balloon flights.

is 270 electrons/s corresponding to 18 BU/s. The average dark current, however, is 16.6 electrons/s only, corresponding to 1.09 BU/s. The dark current noise is proportional to the square root of the integration time and measured to  $0.42 \text{ BU}/\sqrt{s}$ , much smaller than the other noise contributions.

Since all noise contributions (a) to (c) are inversely proportional to the square root of the number of scans ( $N$ ), the total noise is measured as a function of  $N$  (see Fig. 1). For this purpose the total noise and its scaling with the number of scans is monitored using scattered skylight and a small integrating sphere. In both cases, the  $1\text{-}\sigma$  noise for single scans is  $\sim 0.7\%$ . For a medium large number of scans, the instrument, in fact, operates at the physical limits given by the photo-electron noise. For an actual balloon flight, typically 100–1000 spectra (corresponding to 10–100 s integration time) are co-added for the sake of height resolution. This results for field conditions in a total  $1\text{-}\sigma$  noise of  $7 \times 10^{-4}$  for 100, and  $2.5 \times 10^{-4}$  for 1000 co-added spectra, respectively (as indicated by the arrows in Fig. 1).

The spectrograph stray light is analyzed by laboratory measurements (Weidner, 2005). It is estimated to 0.2–0.3% of full saturation resulting in a contribution of  $< 1\%$  at all occurring saturation levels. Spectral structures arising from spectrograph stray light can be eliminated by including an additive polynomial to be fitted to the measured intensity (intensity offset) during the fitting procedure.

<sup>1</sup>available for download under: <http://www.iup.uni-heidelberg.de/bugtracker/projects/doasis/>

**Absolute calibration:** The radiometric calibration of both spectrometers is performed in two steps: In a first step, a small, but not absolutely calibrated, integrating sphere (type BN-102-3 by Gigahertz) is absolutely cross calibrated against an absolute radiance standard using the mini-DOAS instrument as transfer device. In a second step, this now absolutely calibrated integrating sphere is used for radiometric calibration of each of the spectrometers, shortly before the actual balloon flight is conducted.

For absolute radiance calibration, a National Institute of Standards and Technology (NIST) calibrated FEL 1000 W irradiance Quartz Tungsten Halogen (QTH) standard (serial number F-455 from OSRAM Sylvania; Walker et al., 1987) is employed in combination with a calibrated space grade Spectralon diffuser plate manufactured by Labsphere. The same setup is used for absolute radiometric calibration of SCIAMACHY during the SCIAMACHY calibration campaign in 1998 (Dobber, 1999). The bi-directional reflectance distribution function (BRDF) of the diffuser plate is calibrated in 0–23° geometry by the Dutch company TNO TPD. For more details see TNO TPD report of calibration (van Leeuwen, 2003). NIST provides the calibration at a distance of 50 cm. The wavelength dependent radiometric irradiance accuracy of the NIST-FEL lamp ranges between 0.91%–1.09%, and the long term reproducibility is 0.87%–0.96% in the 300–654.6 nm wavelength range (for more details see the NIST report of calibration, 844/25 70 96-96-1, 1997). For the radiance transfer measurements, the NIST-FEL lamp and the Spectralon diffuser plate are positioned into the optical axis of the limb transfer spectrometer. The field of view of the spectrometer light intake telescope is small and completely located inside the characterized lamp irradiance plane on the Spectralon diffuser plate. After the measurement is taken, a not yet absolutely calibrated integrating sphere (type BN-102-3) is cross calibrated with the calibrated transfer spectrometer. The uncertainty of the radiance of the NIST-FEL lamp and Spectralon setup in the 300–700 nm region is 2–3% as indicated by test measurements performed during the SCIAMACHY calibration campaign (Gerilowski, 2004). For the somewhat less ideal conditions in the field, the estimated accuracy of the absolute radiometric calibration of both channels is estimated to 35% at 380 nm, 10% at 440 nm and 4% at 510 nm, including all known sources of uncertainties and errors. The reproducibility of the integrating sphere measurements is better than 1%.

Further, the absolute calibration is checked by comparing the measured radiance at two wavelengths (360 nm vs. 490 nm), for a condition for which the radiative transfer is simple. For example, for the limb observations (90° azimuth angle, +0.5° elevation angle, SZA=88.54° at 30 km) during the Kiruna 23 March 2003 flight, RT calculations show that most photons ((80±5)%) are single Rayleigh scattered. This number is in agreement with the study of Oikarinen et al. (1999). Therefore, by knowing the relative solar irradiance  $F(360\text{ nm})/F(490\text{ nm})=0.588$  (e.g. Gurlić et

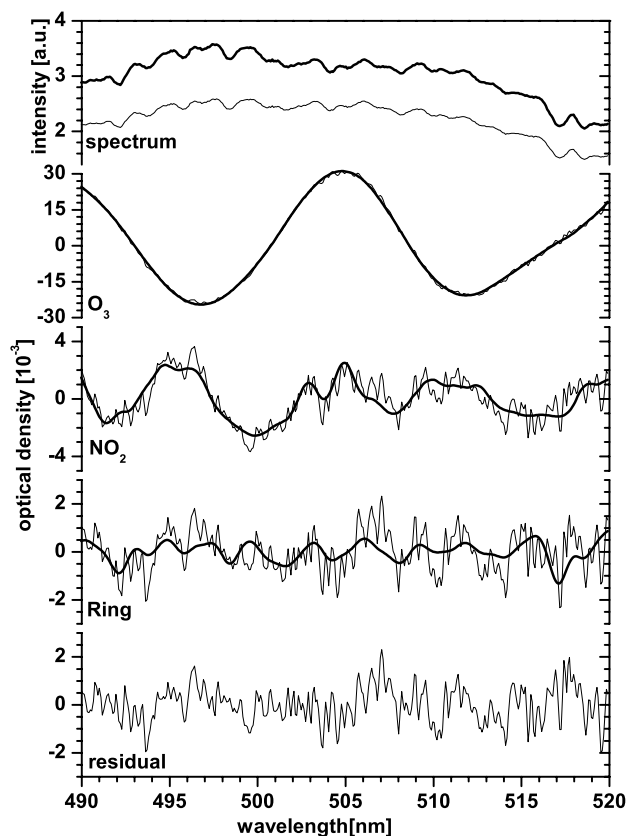
al., 2004) for the two wavelengths and the relative probability for Rayleigh scattering  $(490\text{ nm}/360\text{ nm})^4=3.432$ , the radiance ratio  $I(360\text{ nm})/I(490\text{ nm})=2.018$  can be calculated. In fact, the model ‘Tracy’ calculates a ratio of  $I(360\text{ nm})/I(490\text{ nm})=2.06$ , and our observation yields  $I(360\text{ nm})/I(490\text{ nm})\approx 2.0$ .

### 2.3 DOAS evaluation

For the evaluation of O<sub>3</sub>, NO<sub>2</sub>, BrO, O<sub>4</sub>, and H<sub>2</sub>O slant column densities (SCD), the Differential Optical Absorption Spectroscopy (DOAS) technique is applied to the measured spectra (Platt, 1994; Platt and Stutz, 2006). For the spectral retrieval, the WinDOAS software is used (C. Fayt and M. Van Roozendaal, 2001). Each spectrum is corrected for electronic offset and dark current using an offset and dark current spectrum, respectively, recorded either on the ground shortly before launch or during balloon float after sunset. The spectral retrieval of O<sub>3</sub> is performed in the 490–520 nm wavelength interval. The following cross sections are used: ozone from Voigt et al. (2001) at T=203 K; NO<sub>2</sub> from Harder et al. (1997) at T=217 K and 230 K, while the latter is numerically orthogonalized with respect to the former; water vapor from Rothman et al. (2003) at T=230 K and p=400 hPa; and O<sub>4</sub> at room temperature from Hermans et al. (private communication, 2002<sup>2</sup>). The spectral retrieval of NO<sub>2</sub> is performed in the 400–450 nm wavelength range. The following cross sections are used: O<sub>3</sub> from Burrows et al. (1999) at T=202 K and T=221 K while the latter is numerically orthogonalized with respect to the former; NO<sub>2</sub> from Harder et al. (1997) at T=217 K; water vapor from Rothman et al. (2003) at T=230 K and p=400 hPa; and O<sub>4</sub> at room temperature from Hermans et al. In this wavelength interval, the O<sub>3</sub> cross section from Voigt et al. (2001) recorded with a Fourier Transform spectrometer is rather noisy because the ozone absorption is minimal. Thus, the cross section from Burrows et al. (1999) recorded with the GOME spectrometer at medium resolution but higher signal to noise ratio is preferred. The spectral retrieval of BrO is done as recommended by Aliwell et al. (2002) in the 346–359 nm wavelength interval with the following cross sections: BrO from Wahner et al. (1988) at T=203 K shifted by +0.25 nm to match the wavelength calibration of the BrO reference from the IUP-Bremen<sup>3</sup>; ozone from Voigt et al. (2001) at T=203 K and 223 K while the latter is numerically orthogonalized with respect to the former; NO<sub>2</sub> from Harder et al. (1997) at T=217 K; and O<sub>4</sub> at room temperature from Hermans et al. All high resolution cross sections are convolved with the actual instrumental slit function, determined from recorded line spectra of Hg and Kr. A spectrum correcting for the Ring effect (Grainger and Ring, 1962) is also included in the fitting

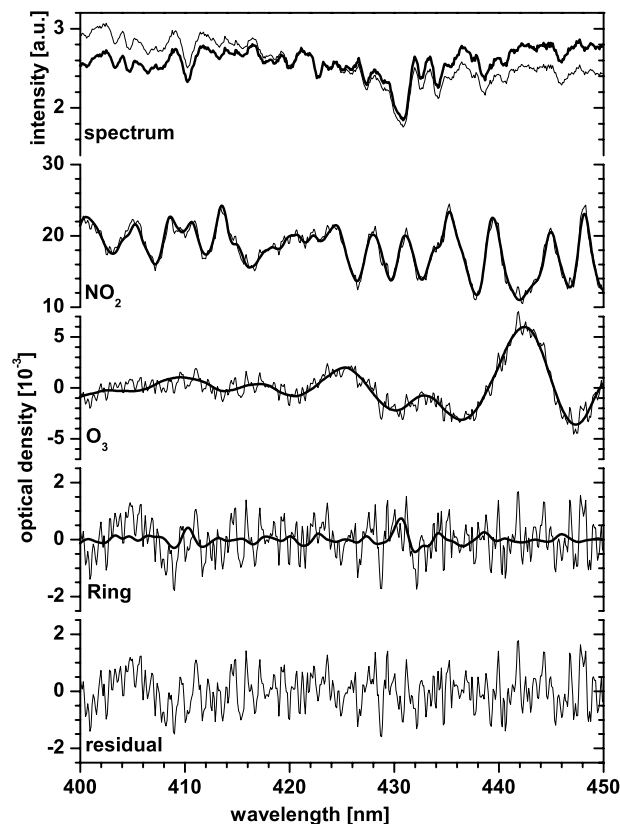
<sup>2</sup>for details see: <http://www.oma.be/BIRA-IASB/Scientific/Topics/Lower/LaboBase/acvd/O4Info.html>

<sup>3</sup>for details see: [http://www.iup.physik.uni-bremen.de/gruppen/molspec/bro2\\_page.html](http://www.iup.physik.uni-bremen.de/gruppen/molspec/bro2_page.html)



**Fig. 2.** Sample DOAS evaluation of ozone in the wavelength interval 490–520 nm for a limb observation at float altitude in limb scanning mode (at 31.6 km altitude, an elevation angle of  $-5.5^\circ$ , azimuth angle of  $90^\circ$ , and  $\text{SZA}=89.9^\circ$ ) for the Kiruna flight on 23 March 2003. Shown are the retrieved optical densities of  $\text{O}_3$ ,  $\text{NO}_2$ , Ring (thick lines) and the latter plus the residual structure (thin lines). The upper two traces show the measured (thick line) and the Fraunhofer (thin line) spectra, the latter is recorded at an altitude of 29.7 km, an elevation angle of  $0.5^\circ$ , an azimuth angle of  $90^\circ$ , and  $\text{SZA}=88.5^\circ$ .

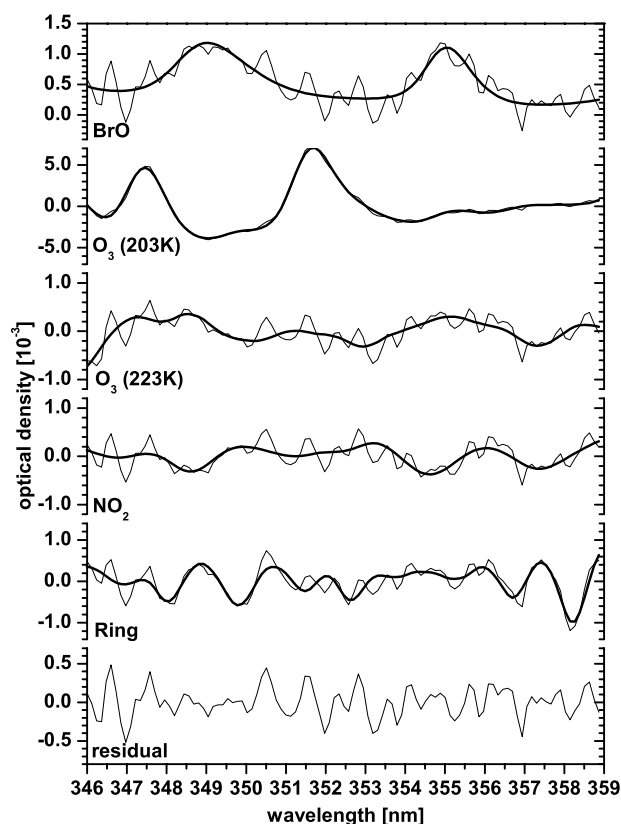
routine:  $\sigma_{\text{Ring}} = I_{\text{Raman}} / (I_{\text{ref}} - I_{\text{Raman}})$ . For  $I_{\text{ref}}$ , the same solar reference spectrum as for the DOAS evaluation is used, and  $I_{\text{Raman}}$  is calculated with the DOAS tool MFC (Gomer et al., 1996). To correct for structures arising from spectrograph stray light, an additive polynomial (intensity offset) of 2nd order is fitted to the measured spectra during the DOAS evaluation procedure. The obtained residuals are somewhat lower if an additive polynomial is included into the fit. Both the residuals and the fit results are almost identical and well within the given uncertainties for a degree of the additive polynomial between 0 and 2. For the Fraunhofer reference spectrum, a limb measurement from balloon float altitude is used, for which the residual trace gas absorptions are expected to be minimal. For example, for the flight from Kiruna on 23 March 2003, the Fraunhofer reference is recorded at 29.7 km of altitude,  $\text{SZA}=88.5^\circ$ , elevation angle  $+0.5^\circ$  and



**Fig. 3.** Same as Fig. 2 but for the  $\text{NO}_2$  evaluation in the 400–450 nm wavelength interval and a limb observation (at 30.9 km altitude,  $-3.5^\circ$  elevation angle (corresponding to a tangent height of  $\sim 19.5$  km),  $90^\circ$  azimuth angle and  $\text{SZA}=89.4^\circ$ ) for the Kiruna flight on 23 March 2003.

azimuth angle  $90^\circ$  to the sun. Figures 2 and 3 show the spectral retrievals for the inferred differential slant column densities of ozone measured at 31.6 km for  $\text{SZA}=89.9^\circ$ , and elevation angle  $-5.5^\circ$ , and of  $\text{NO}_2$  measured at 30.9 km for  $\text{SZA}=89.4^\circ$ , and  $-3.5^\circ$  elevation angle, respectively. Figure 4 shows the inferred BrO absorption for an observation at 26.4 km altitude ( $-1.5^\circ$  elevation angle,  $90^\circ$  azimuth angle and  $\text{SZA}=82.9^\circ$ ) for the Kiruna flight on 24 March 2004.

One particular problem is that the DOAS method only retrieves differential Slant Column Densities (dSCD), i.e. the trace gas absorption of the spectrum to be evaluated minus that of the solar reference spectrum. So the trace gas absorption of the spectrum used as solar reference has to be determined separately and added to all measured dSCD values. One possibility is to use the solar spectrum from Kurucz et al. (1984) convolved to the instrument's resolution. However, this leads to significantly larger residuals caused by improperly removed Fraunhofer structures causing rather large systematic errors in the retrieved SCD values. Accordingly, this method only works for the strong absorber  $\text{O}_3$  but cannot be used in the case of  $\text{NO}_2$  or BrO as their optical densities



**Fig. 4.** Same as Fig. 2 but for the BrO evaluation in the 346–359 nm wavelength interval and a limb observation (at 26.4 km altitude,  $-1.5^\circ$  elevation angle (corresponding to a tangent height of  $\sim 25.5$  km),  $90^\circ$  azimuth angle and  $\text{SZA}=82.9^\circ$ ) for the Kiruna flight on 24 March 2004.

are much smaller than the residual structures in that case. A different approach is undertaken for BrO. As the BrO concentration above balloon float altitude ( $\sim 32$  km) is comparatively small and also well known, a BrO concentration profile from the 3D-CTM Slimcat (Chipperfield, 1999) and the RT modeled BoxAMF are used to simulate the BrO absorption contained in the solar reference spectrum. In the case of  $\text{NO}_2$ , this method cannot be applied because the wintery stratospheric  $\text{NO}_2$  concentrations are not well reproduced by CTMs and the solar reference is recorded in a region of very high  $\text{NO}_2$  concentrations implying large absorptions in the reference spectrum itself so it has to be determined very precisely. For the  $\text{NO}_2$  measurements shown in this paper, the  $\text{NO}_2$  absorption of the solar reference spectrum is determined by a correlation plot of the measured dSCDs and RT simulated SCDs using the profile obtained from the direct sunlight DOAS measurements as input.

## 2.4 Radiative transfer modeling

A Monte Carlo (MC) Radiative Transfer model (called “Tracy”) has been developed by our group (Von Friedeburg, 2003). It allows the forward simulation of the mini-DOAS and Envisat/SCIAMACHY limb observations. In particular, it can simulate the measured limb radiances and slant column densities (SCD) of the trace gases under consideration, including sensitivity tests for varying atmospheric parameters such as  $T$ ,  $p$ , the ozone and aerosol profiles.

“Tracy” solves the radiative transfer equation by backward Monte Carlo simulations in a fully spherical, 3-dimensional and refractive atmosphere. However, it does not consider the polarization of the scattered skylight, which was found necessary in a recent theoretical study on scattered light ozone measurements (Hasekamp et al., 2002). Further, it supports an arbitrary spatial discretization, a tool which permits to account for spatially strong varying aerosol and trace gas concentrations, and it takes into account multiple scattering with arbitrary scattering phase functions.

For the RT simulations, “Tracy” uses atmospheric temperature and pressure profiles and, if available, profiles of the atmospheric aerosol extinction and ozone concentration (Hönniger et al., 2004). For the present simulations, the stratospheric aerosol data version 2.00 of the SAGE III instrument is used (Thomason and Taha, 2003). The used aerosol phase functions are calculated for a standard scenario using Mie-theory (Hendrick, pers. communication). Mie scattering due to tropospheric clouds is not considered further. In the current version, “Tracy” runs with either the “Kurucz” (Kurucz et al., 1984) or the SOLSPEC solar spectrum (Thuillier et al., 1997, 1998a and b).

“Tracy” outputs are wavelength dependent limb radiances, slant column densities (SCD) of the trace gases of interest which the actual measurements can be compared to, and so called box air mass factors (BoxAMFs). The latter give the ratio of slant to vertical paths through each atmospheric layer, i.e.  $\text{AMF} = \text{SCD} / \text{VCD}$ , where  $\text{AMF}$  is the BoxAMF,  $\text{SCD}$  the slant and  $\text{VCD}$  the vertical column density of the considered layer.

## 2.5 Profile retrieval

Profile inversion from measured SCDs is performed by the Maximum A Posteriori (MAP) solution technique described in Rodgers (2000), also used for the inversion of solar occultation measurements performed by the other optical spectrometers (DOAS and LPMA) deployed on the gondola. The two-step approach of forward modeling of the atmospheric RT, and optimal estimation for profile inversion of measured SCDs is chosen since it offers the chance to cross validate each step of the newly developed limb technique. Sensitivity tests and validation can thus be undertaken on the simulated and measured limb radiance level, by comparison of simulated and measured SCDs of the gases of interest, or

**Table 1.** Compendium of balloon-borne mini-DOAS measurements

Date Time (UT)	Location	Geophys. Cond. SZA range	Instrument	Observation Mode
18/19 Aug. 2002 15:15–2:38	Kiruna 67.9° N, 21.1° E	high lat. sum. 69.75–94.4° 94.6–88.1°	LPMA/ mini-DOAS	nadir fixed limb
4 March 2003 12:55–15:25	Kiruna 67.9° N, 21.1° E	high lat. spring 77.6–88.8°	LPMA/DOAS mini-DOAS	nadir fixed limb
23 March 2003 14:47–17:35	Kiruna 67.9° N, 21.1° E	high lat. spring 78.9–94.7°	LPMA/DOAS mini-DOAS	nadir fixed limb during ascent scanning limb at float
9 Oct. 2003 15:39–17:09	Aire-sur-l'Adour 43.7° N, 0.25° W	mid-lat fall 66–88°	LPMA/DOAS mini-DOAS	nadir fixed limb
24 March 2004 13:55–17:35	Kiruna 67.9° N, 21.1° E	high lat. spring 72–98°	LPMA/DOAS mini-DOAS	fixed limb during ascent scanning limb at float

by inter-comparison of otherwise measured profiles c.f., obtained from the well validated solar occultation technique. At the same time, the approach allows us to validate various radiative transfer codes used in upcoming studies addressing SCIAMACHY limb observations.

Using the definition of the BoxAMF as stated above and the standard notation as used in Rodgers (2000), the profile retrieval problem can be written as:

$$\mathbf{y} = \mathbf{K} \mathbf{x} + \boldsymbol{\varepsilon}, \quad (1)$$

where  $\mathbf{y}$  is a vector of the measured SCDs,  $\mathbf{x}$  a vector of the desired VCDs for each altitude layer, and  $\mathbf{K}$  the weighting function matrix given by the RT calculated BoxAMF for each measurement and altitude layer.  $\boldsymbol{\varepsilon}$  describes the inevitable measurement errors. The best estimate solution  $\hat{\mathbf{x}}$  of this problem based on the measurements  $\mathbf{y}$  and available a priori information  $\mathbf{x}_a$  is given by the linear MAP inversion algorithm as described in Rodgers (2000), chapter 4:

$$\hat{\mathbf{x}} = (\mathbf{K}^T \mathbf{S}_{\boldsymbol{\varepsilon}}^{-1} \mathbf{K} + \mathbf{S}_a^{-1})^{-1} (\mathbf{K}^T \mathbf{S}_{\boldsymbol{\varepsilon}}^{-1} \mathbf{y} + \mathbf{S}_a^{-1} \mathbf{x}_a), \quad (2)$$

where  $\mathbf{S}_a$  and  $\mathbf{S}_{\boldsymbol{\varepsilon}}$  are the covariances of the a priori profile and the measurements, respectively. The covariance of the retrieved profile is given by:

$$\hat{\mathbf{S}} = (\mathbf{K}^T \mathbf{S}_{\boldsymbol{\varepsilon}}^{-1} \mathbf{K} + \mathbf{S}_a^{-1})^{-1}. \quad (3)$$

All covariance matrices are diagonal. The covariance of the measurements is given by the errors of the DOAS evaluation  $\varepsilon$  corresponding to each measurement, i.e.  $S_{\varepsilon,ii} = \varepsilon_i^2$ . The covariance of the a priori profile is taken as constant percentage  $p$  of the profile itself, i.e.  $S_{a,i} = (p x_{a,i})^2$ . The error of the retrieved profile is obtained from its covariance matrix by  $\sigma_i = \sqrt{\hat{S}_{ii}}$ .

The inversion is characterized by the averaging kernel matrix  $\mathbf{A}$  given by:

$$\mathbf{A} = (\mathbf{K}^T \mathbf{S}_{\boldsymbol{\varepsilon}}^{-1} \mathbf{K} + \mathbf{S}_a^{-1})^{-1} \mathbf{K}^T \mathbf{S}_{\boldsymbol{\varepsilon}}^{-1} \mathbf{K}. \quad (4)$$

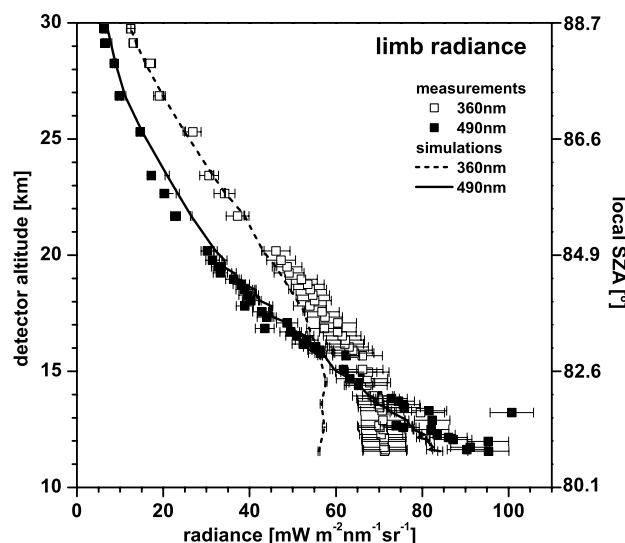
Averaging kernels being close to unity indicate a height resolution of (or better than) the grid spacing used for the profile retrieval and independence of the retrieved profile point from the a priori. For all the profiles shown in this study, only the altitudes levels with averaging kernels close to one are considered.

Constraining the inversion by an a priori profile is necessary as the measurements contain no profile information about the altitude levels above float altitude. In order not to lose profile information about the altitude levels below float, the covariance of the a priori is chosen as high as possible, usually to  $p=100\%$ . As a result, the retrieved profiles are independent from the a priori in the altitude range traversed by the balloon. This independence has been checked by varying the a priori over a large range and by analysis of the averaging kernels.

## 2.6 Error analysis

For the  $\text{O}_3$  and  $\text{NO}_2$  DOAS evaluation, the statistical retrieval error determined by the residual noise is usually very small. In this case, the largest error of the retrieved SCDs arise from systematical effects like uncertainties of the used cross sections and the solar reference offset to be added to the measured dSCDs. These systematical error contributions are assumed to be 5% of the retrieved dSCD and solar reference offset, respectively. They are accounted for in the error analysis by retrieving an additional profile with a set of SCDs with the respective error added. The difference of the two profiles yields the error of the retrieved profile arising from the respective effect and is added to the error given by Eq. (3).



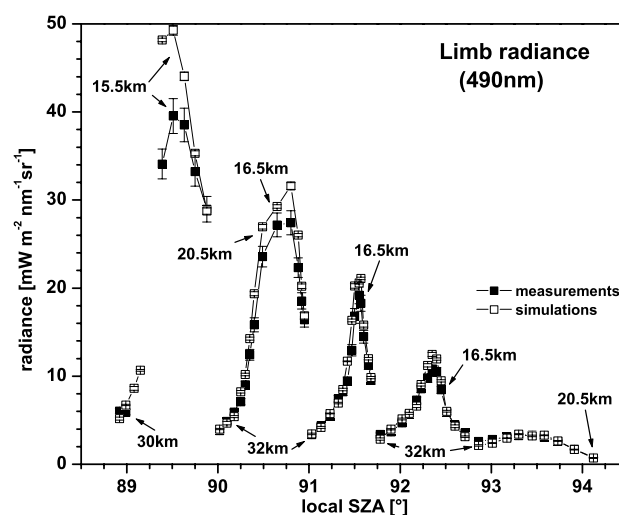


**Fig. 5.** Comparison of measured and simulated limb radiance at 360 nm (open squares and dashed line) and 490 nm (filled squares and full line, respectively) for an azimuth angle of  $90^\circ$  and an elevation angle of  $+0.5^\circ$  during balloon ascent over Kiruna on 23 March 2003. Also denoted is the Solar Zenith Angle (SZA) at the balloon position.

### 3 Observations and flights

For the actual balloon flights (see Table 1) the instrument is mounted on the azimuth controlled LPMA/DOAS payload. One of the telescopes is directed into the nadir, whereas the other telescope is mounted into the limb scanner, which points perpendicular to the Sun's azimuth direction (when the gondola is perfectly azimuth controlled). The elevation of the limb telescope is kept fixed during balloon ascent at an angle around  $0^\circ$ . At balloon float, the instrument performs scanning limb measurements at elevation angles between  $-6^\circ$  and  $0^\circ$ .

To date 5 mini-DOAS flights have been performed; and overall the instrument performed well during all of them. Since the instrument and method are novel and need some testing, the instrument has been constantly refined, e.g., by operating a limb scanner from the second flight on, by subsequently improving the instrument's housing and cooling and its software, and finally by adding a vacuum sealed housing surrounding the spectrometers. Therefore, the quality of the limb and nadir spectra are not directly comparable from one balloon flight to another. Nevertheless, the functionality, and the feasibility of the method, as well as its potential for novel atmospheric observations are well demonstrated by the results discussed in the following section.

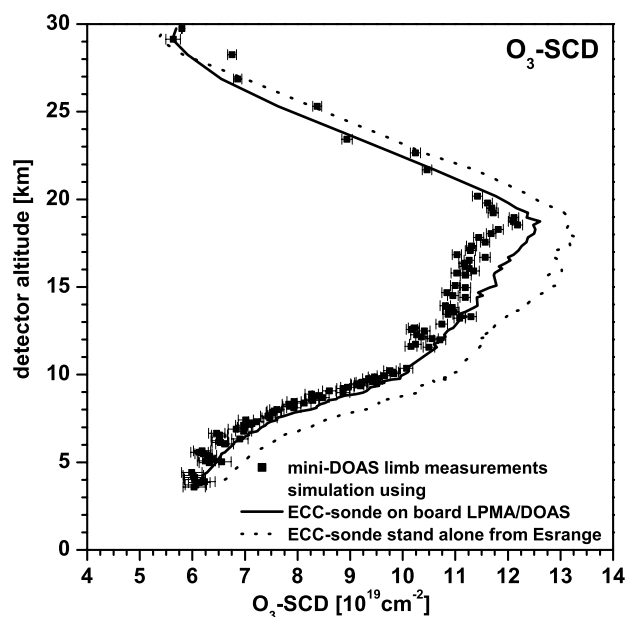


**Fig. 6.** Comparison of measured (filled squares) and RT modeled (open squares) limb radiances at 490 nm vs. local SZA at the balloon position for the Kiruna 23 March 2003, balloon flight at balloon float altitude (30.3–32.2 km) for an azimuth angle of  $90^\circ$  and elevation angle varying from  $+0.5^\circ$  to  $-5.5^\circ$  with  $+0.5^\circ$  steps. The figure shows 5 limb profiling scans consisting of up to 13 individual observations each. The height labels denote the tangent heights of the observations.

### 4 Results and discussion

In the following, we provide some sample results obtained from the data collected during 2 of the 5 balloon flights:

(a) In a first exercise, the measured absolute skylight radiances (azimuth angle  $90^\circ$ , fixed elevation of  $+0.5^\circ$ ) are compared with RT calculated skylight radiances for two wavelengths ( $\lambda=360$  nm and  $\lambda=490$  nm) for the balloon ascent and the scanning limb observations at balloon float altitude (30.3–32.2 km) at Kiruna on 23 March 2003 (see Figs. 5 and 6, respectively). For the RT modeling the actually measured atmospheric parameters (profiles of T and p, and of ozone) are used. The ground albedo is set to 0.6 (which accounts for the still existing snow cover over Northern Scandinavia by late March). No further assumptions are made for the tropospheric aerosols and cloud scattering and absorption, mainly because they are not known for the actual sounding and are known to be quite variable. Figure 5 demonstrates that with the given assumptions the measured stratospheric limb radiances at 360 nm and 490 nm are reproduced reasonably well by the RT model (within the given error bars). At  $\lambda=360$  nm, however, the measured limb radiances are exceedingly larger than the simulation below 17 km. Also not unexpected, in the tropopause region the measured limb radiances are much more variable at 490 nm than at 360 nm. This effect is due to the predominance of Mie scattering by aerosols and clouds over Rayleigh scattering for the skylight radiances in the visible (at 490 nm) compared to UV light (at 360 nm).

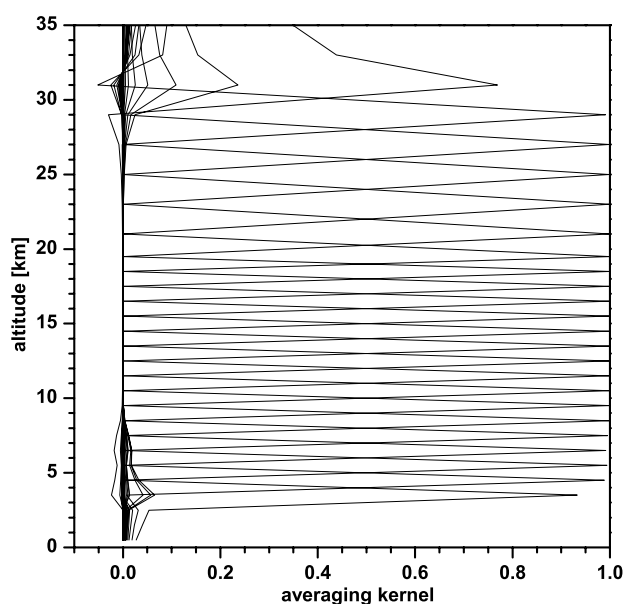


**Fig. 7.** Comparison of limb ( $90^\circ$  azimuth angle and  $0.5^\circ$  elevation angle) measured (filled squares) and forward modeled  $\text{O}_3$ -SCDs using the ozone measurements from two ECC-sondes deployed on the same gondola (full line), and flown stand-alone  $\sim 3$  h after the LPMA/DOAS launch (dotted line) for the Kiruna 23 March 2003 flight.

Figure 6 shows the measured and RT modeled limb radiance at  $\lambda=490$  nm for the limb scanning observations at balloon float altitude (30.3–32.2 km). During each limb scan, the elevation angle is varied from  $+0.5^\circ$  to  $-5.5^\circ$  in steps of  $0.5^\circ$  corresponding to about 10 min. Overall, a good agreement in the measured and modeled skylight radiance is found for limb scans number 2 to 5 (when counting from left to right), and only for the 1st run, the disagreement is larger at lower altitudes than expected. We believe that the gap around  $\text{SZA}=90^\circ$  is due to the notorious difficulties of RT models to deal numerically correct in this regime. Sensitivity tests showed that stratospheric aerosols and tropospheric clouds only have a small impact on the RT calculated radiances and cannot explain the discrepancies.

(b) After gaining some confidence in the RT modeling, we further evaluate the measured spectra during balloon ascent and at balloon float to obtain profile information of some relevant atmospheric absorbers in the UV/vis wavelength range i.e., of  $\text{O}_3$ ,  $\text{NO}_2$ , and  $\text{BrO}$ .

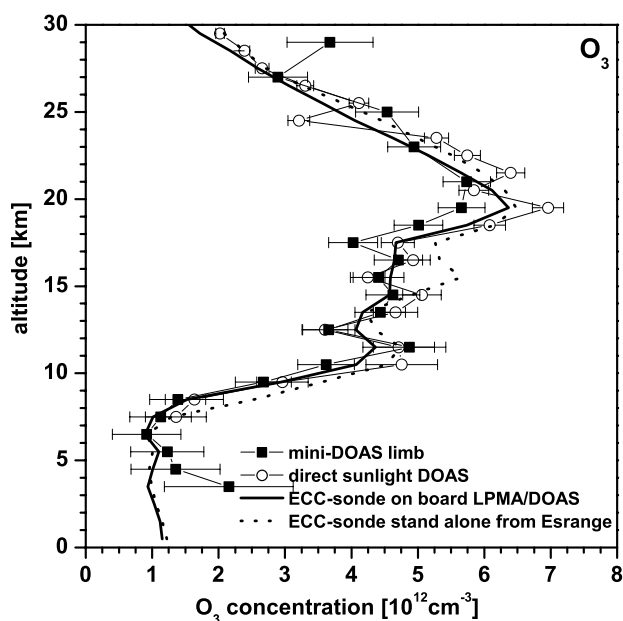
In a first exercise, we tested the  $\text{O}_3$  profile retrieval from the mini-DOAS observations for a fixed limb viewing geometry (at an azimuth angle of  $90^\circ$  and an elevation angle of  $0.5^\circ$ ) performed during the balloon ascent of the Kiruna 23 March 2003 flight. Note that the balloon flight is undertaken around the vortex edge, implying large horizontal gradients in the trace gas concentrations.



**Fig. 8.** Averaging kernels for the ozone profile retrieval shown in Fig. 9.

In a first step, the slant column densities of ozone ( $\text{O}_3$ -SCD) are inferred from the measured spectra, and compared with the same parameter simulated by Tracy RT calculations using as input either the  $\text{O}_3$  profile simultaneously measured on-board by an electrochemical cell (ECC), or by a stand-alone ECC  $\text{O}_3$  sonde, launched  $\sim 3$  h after the LPMA/DOAS gondola. Figure 7 reveals that the measured and simulated  $\text{O}_3$ -SCDs compare reasonably well, however only for the simulations using the ozone profile measured by the ECC-sonde aboard (the full line in Fig. 7). Conversely taking the  $\text{O}_3$  profile in the simulation from the measured stand-alone launched ECC sonde launched  $\sim 3$  h after the LPMA/DOAS payload, larger  $\text{O}_3$  concentrations are obtained in the 12–21 km height range with a corresponding overestimation in the simulated  $\text{O}_3$ -SCDs (dotted curve in Fig. 7). This comparison clearly demonstrates the quality of the limb  $\text{O}_3$  measurements and its sensitivity towards the shape and  $\text{O}_3$  concentration of the profile.

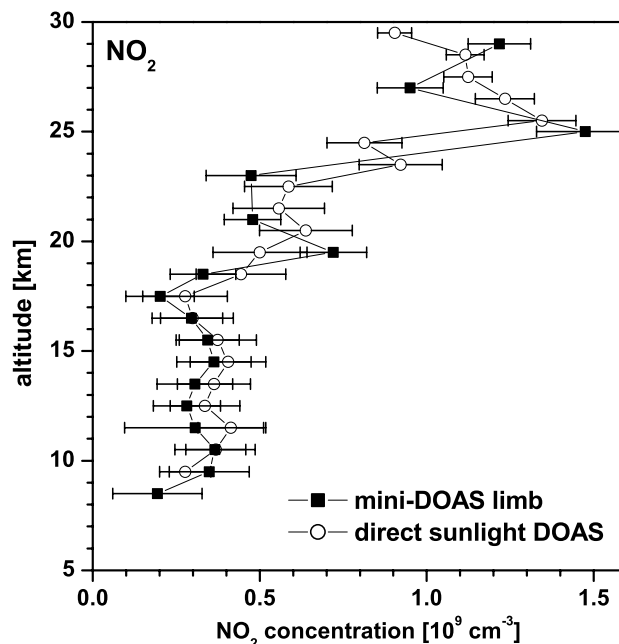
In a second step, the measured  $\text{O}_3$ -SCDs are mathematically inverted into an  $\text{O}_3$  concentration profile, using the RT calculated BoxAMF and the inversion routine described in Sect. 2.5 (see Fig. 9). Overall, a good agreement found between the inferred limb ozone profile and the simultaneously measured  $\text{O}_3$  profiles either from the on-board ECC sonde or inferred from the direct sunlight DOAS observations indicates the feasibility of the balloon-borne limb method. The error bars are calculated as described in Sect. 2.6. The relative total error of the inferred  $\text{O}_3$  concentrations is 5–10% in the 10–25 km range increasing to 15% above as concentrations are lower. Below 10 km, the relative error is strongly increasing due to very low tropospheric concentrations and,



**Fig. 9.** Comparison of inferred  $\text{O}_3$  profiles from (a) limb observations (filled squares) at an azimuth angle of  $90^\circ$  and elevation angle of  $0.5^\circ$ , (b) the direct sunlight DOAS measurements (open circles), and two ECC ozone sondes (c) one deployed on the same gondola (full line) and (d) from the stand-alone launched ECC-sonde (dotted line) for the Kiruna, 23 March 2003 flight.

thus, stronger contribution of the stratospheric  $\text{O}_3$  to the measured SCDs. The rows of the corresponding averaging kernel matrix **A** are shown in Fig. 8. The averaging kernels are close to 1 in the altitude range where measurements are available, i.e. between 3 and 30 km, on the chosen altitude grid of 1 km below 20 km and 2 km above owing to the lower number of measurements due to lower limb radiances there.

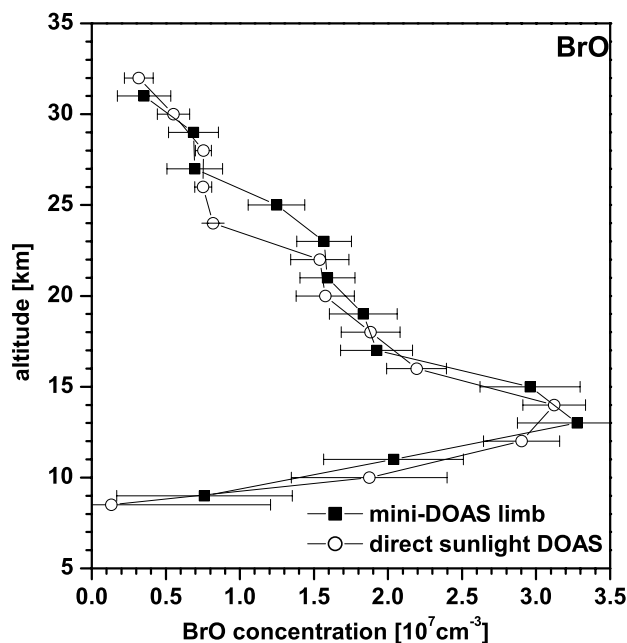
In a second exercise, we inferred an  $\text{NO}_2$  profile from the measured SCDs during balloon ascent on 23 March 2003 (see Fig. 10). As before the comparison of the  $\text{NO}_2$  profiles inferred either from the limb or direct Sun observations is excellent. This result again demonstrates the equally large sensitivity of the balloon-borne ascent limb measurement compared to the solar occultation techniques. The higher errors of the mini-DOAS limb measured  $\text{NO}_2$ -SCDs compared to the SCDs of the direct sunlight DOAS measurements due to a lower number of analyzed photons are compensated by the better conditioning of the inversion problem (i.e. Box-AMFs with sharp maximum at the detector altitude) so that both profiles have similar uncertainties. For the shown error bars, 5% uncertainty of the solar reference offset is assumed. No systematical error of the cross section is assumed for this comparison as the same cross sections are used for the DOAS evaluation of the mini-DOAS limb and solar occultation spectra. The total relative errors of the mini-DOAS profile is 5–10% above 25 km where  $\text{NO}_2$  concentrations are maximum and 20–30% below.



**Fig. 10.** Comparison of inferred  $\text{NO}_2$  profiles from (a) limb observations (filled squares) for an azimuth angle of  $90^\circ$  and elevation angle of  $0.5^\circ$ , and (b) from direct sunlight DOAS measurements (open circles) during balloon ascent at Kiruna on 23 March 2003. The SZA was between  $80^\circ$  at 10 km, and  $88.9^\circ$  at 30 km detector altitude.

In a third exercise, we inferred a BrO profile from the limb measured SCDs during balloon ascent on 24 March 2004 (see Fig. 11). As for the  $\text{NO}_2$  comparison, the limb measured BrO compares excellently with the BrO measured with the solar occultation instrument, thus providing confidence in both methods. The total relative error calculated as described in Sect. 2.6 is  $\sim 12\%$  in the 12–25 km range and significantly higher above and below as BrO concentrations are much lower.

Finally, in a fourth exercise, we intercompare measured and simulated  $\text{O}_3$ -SCDs for the scanning limb observations at balloon float (30.3–32.2 km) for the Kiruna 23 March 2003 flight (see Fig. 12). As before, the measured and simulated  $\text{O}_3$ -SCDs compare well, in particular for tangent heights above the ozone maximum concentration ( $> 25$  km). For lower altitudes, simulated  $\text{O}_3$ -SCD are larger by  $\sim 10\%$  than those measured, a finding for which the reason is not totally clear. However, it is worthy noting that the simulated  $\text{O}_3$ -SCDs are less sensitive to RT errors than the simulated limb radiances. Therefore the small error found in the simulated radiance (compare Fig. 3) can hardly provide an explanation for the discrepancies. For a given ozone profile, a larger simulated than measured limb  $\text{O}_3$ -SCD could mean a longer simulated than real viewing distance (or a better visibility) over which the limb radiance received by the instrument is integrated. Conversely, if the RT model simulations



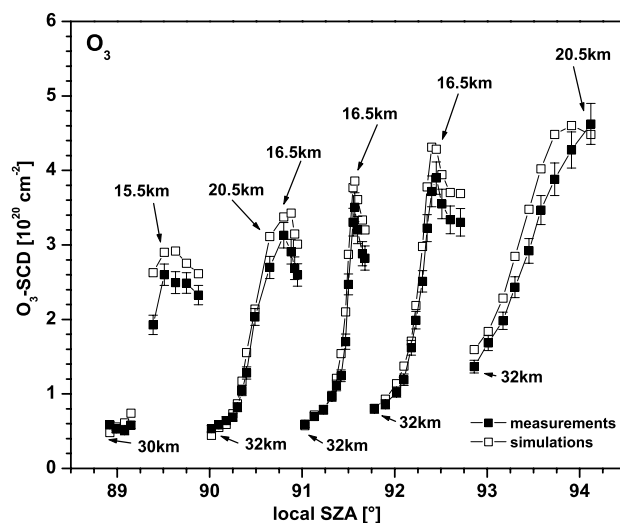
**Fig. 11.** Comparison of inferred BrO profiles from (a) limb observations (filled squares) for an azimuth angle of  $90^\circ$  and elevation angle of  $-1.5^\circ$ , and (b) from direct sunlight DOAS measurements (open circles) during balloon ascent at Kiruna on 24 March 2004. The SZA was between  $76.5^\circ$  at 9 km, and  $85.4^\circ$  at 33 km detector altitude.

fit well with the actual atmospheric RT, smaller measured than simulated  $O_3$ -SCDs would imply smaller ozone concentrations than assumed in the simulation (here as before, the on-board measured ECC-sonde ozone profile is taken). In fact, there is a good reason to assume that the latter explanation is more likely than the former since the limb scanning measurement touched the vortex edge. The ozone-poor polar air-masses were increasingly occupying the instrument's field of view as the gondola followed (in clockwise direction) the increasing solar azimuth angle with progressing balloon flight duration.

## 5 Conclusions

We describe the first application of a novel balloon-borne UV/vis spectrometer and its limb radiance and trace gas measurements. The study discusses the instrument's performance in a series of balloon flights, and the novel method of balloon-borne UV/vis limb trace gas measurements is tested against simultaneous observations of the same atmospheric parameters and RT model results.

Overall, reasonably good agreement is found in inter-comparisons of simulated and measured limb radiance performed in fixed observing geometry during balloon ascent, and in limb scanning geometry at balloon float altitude dur-



**Fig. 12.** Comparison of measured (filled squares) and forward modeled (open squares) limb absorption of ozone vs. local SZA at the balloon position for the Kiruna flight on 23 March 2003 for azimuth angle of  $90^\circ$  to the sun and elevation angles varying from  $+0.5^\circ$  to  $-5.5^\circ$  with  $0.5^\circ$  steps. The height labels denote the tangent heights of the observation.

ing sunset. Due to potential problems arising from the Earth sphericity and atmospheric refraction, the latter provides a stringent test for forward RT modeling. Further, accurate and cross validated RT modeling and measurements, however, permit to calculate more reliably photolysis frequencies of UV/vis absorbing atmospheric trace gases, known to be particularly difficult to calculate at large solar zenith angles (e.g., Bösch et al., 2001; Uhl and Reddmann, 2004). The inter-comparison demonstrates that the UV/vis optical properties of the stratosphere for volcanically quiescent and polar stratospheric cloud free periods are reasonably well understood.

The novel limb technique is also cross validated by simultaneous measurements of the same atmospheric trace gases (here  $O_3$ ,  $NO_2$ , BrO) and, overall, a good agreement is found. It is shown that the UV/vis limb measurements are particularly sensitive in the upper troposphere/lowermost stratosphere (UT/LS) for moderately large SZA during balloon ascent. In the future, this particularly large sensitivity may allow more accurate studies on the UT/LS  $NO_2$  or halogen oxide photochemistry than presently available from remote sensing instrumentation. Moreover, scanning UV/Vis limb profiling may provide a rather powerful tool to investigate the time dependent photochemistry of stratospheric radicals, for example of the ClO+BrO ozone loss cycle by simultaneous observations of the OClO and BrO profiles in the wintertime polar stratosphere (WMO, 2002; Canty et al., 2005; Salawitch et al., 2005).

Finally, our study demonstrates the overall feasibility of the various steps (spectral retrieval, forward RT modeling,

profile inversion, ...) in the emerging technique of satellite-borne remote sensing of atmospheric constituents by spectroscopic UV/vis limb observations.

**Acknowledgements.** Support of the project by BMBF through grants 50FE0017 and 50FE0019. We are grateful to the support given by the team of CNES in particular 'l'équipe nacelles pointées' and the balloon launching team from Aire-sur-l'Adour/France for the assistance given to perform successfully the balloon flights. We thank in particular C. Randall (University of Colorado, Boulder, USA) for providing the SAGE III aerosol data. Thanks to M. Long for proofreading the manuscript with regard to the English.

Edited by: M. Dameris

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